

737-700/800 WIRING DIAGRAM MANUAL

Hapag-Lloyd Flug GmbH

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DOCUMENT D280A103



This manual is applicable to the aircraft on this list:

		Oper	ator		Manufacturer			
	Model-Series	Identification Code	Effectivity Code	Block Number	Serial Number	Line Number	Registration Number	
	737-8K5	HAP	006	YC001	27981	7	D-AHFA	
	737-8K5	HAP	007	YC002	27982	8	D-AHFB	
	737-8K5	HAP	001	YC003	27977	9	D-AHFC	
	737-8K5	HAP	002	YC004	27978	40	D-AHFD	
	737-8K5	HAP	003	YC005	27979	44	D-AHFE	
I	737-8K5	TRX	004	YC006	27980	45	EI-EDZ	
I	737-8K5	TRX	005	YC007	27989	59	EI-EEA	
	737-8K5	HAP	800	YC008	27983	218	D-AHFH	
	737-8K5	HAP	009	YC009	27984	220	D-AHFI	
	737-8K5	JFU	010	YC010	27990	246	CN-RPE	
	737-8K5	HAP	011	YC011	27991	248	HA-LKC	
	737-8K5	HAP	012	YC012	27985	470	HA-LKD	
	737-8K5	HAP	013	YC013	27986	474	D-AHFM	
	737-8K5	HAP	015	YC015	27987	499	D-AHFO	
	737-8K5	HAP	016	YC016	27988	508	D-AHFP	
	737-8K5	HAP	017	YC017	27992	523	D-AHFQ	
	737-8K5	HAP	019	YC018	30593	528	D-AHFR	
	737-8K5	HAP	018	YC019	28623	556	D-AHFS	
	737-8K5	HAP	020	YC020	30413	636	D-AHFT	
	737-8K5	HAP	021	YC021	30414	703	D-AHFU	
	737-8K5	HAP	022	YC022	30415	719	D-AHFV	
	737-8K5	HAP	023	YC023	30882	760	D-AHFW	
	737-8K5	HAP	024	YC024	30416	778	D-AHFX	
	737-8K5	HAP	025	YC025	30417	781	D-AHFY	
	737-8K5	HAP	026	YC026	30883	783	D-AHFZ	
I	737-8K5	TYM	028	YC028	32905	1046	VQ-BDN	
I	737-8K5	TYM	029	YC029	32906	1087	VQ-BDO	
	737-8K5	HAP	030	YC030	32907	1117	D-AHLR	
	737-8K5	HAP	031	YK901	34684	1870	D-ATUC	
	737-8K5	HAP	032	YK902	34685	1901	D-ATUD	
	737-8K5	HAP	033	YK903	34686	1903	D-ATUE	
	737-8K5	HAP	034	YK904	34687	1907	D-ATUF	

EFFECTIVE AIRCRAFT



		Oper	ator				
r	Model-Series	Identification Code	Effectivity Code	Block Number	Serial Number	Line Number	Registration Number
	737-8K5	HAP	035	YK905	34688	1909	D-ATUG
	737-8K5	HAP	036	YK906	34689	1935	D-ATUH
	737-8K5	BRI	037	YK907	34690	2184	G-FDZJ
1	737-8K5	JFU	039	YK908	34691	2246	CN-RPF
	737-8K5	JFU	040	YK909	34692	2249	CN-RPG
	737-8K5	TLB	041	YK910	35133	2313	OO-JAF
	737-8K5	TLB	047	YK911	35142	2660	OO-JBG
	737-8K5	TLB	049	YK912	35148	2790	OO-JAQ
	737-8K5	HXL	054	YK918	35100	2424	PH-TFA
	737-8K5	HXL	050	YK919	35149	2820	PH-TFB
	737-8K5	HAP	048	YL401	35143	2763	D-AHLK
	737-8K5	BRI	038	YL421	35134	2152	G-FDZA
	737-8K5	BRI	042	YL422	35131	2242	G-FDZB
	737-8K5	BRI	043	YL423	35132	2276	G-FDZD
	737-8K5	BRI	044	YL424	35137	2482	G-FDZE
	737-8K5	BRI	045	YL425	35138	2499	G-FDZF
	737-8K5	BRI	046	YL426	35139	2538	G-FDZG
	737-8K5	BRI	051	YL427	35145	2849	G-FDZR
	737-8K5	BRI	052	YL428	35147	2866	G-FDZS
	737-8K5	HXL	053	YL429	35146	2875	PH-TFC
	737-7K5	HAP	101	YM643	34693	2260	D-AHXC
	737-7K5	HAP	102	YM645	35135	2451	D-AHXE
	737-7K5	HAP	103	YM646	35136	2465	D-AHXF
	737-7K5	HAP	104	YM647	35140	2575	D-AHXG
	737-7K5	HAP	105	YM649	35141	2603	D-AHXI
1	737-7K5	TLB	106	YM651	35144	2652	OO-JAS
	737-7K5	TLB	107	YM652	35150	2825	OO-JAR

EFFECTIVE AIRCRAFT



Hapag-Lloyd Flug GmbH Revision No. 42

Aug 10/2009

To: All holders of this Boeing Document D280A103

Attached is the current revision to the 737 Wiring Diagram Manual (WDM).

The manual is available either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the Effective Pages. The pages which are revised will be identified on the Effective Pages by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the Effective Pages is identified by Chapter-Section-Subject number, page number and page date. Pages replaced or deleted by this revision should be removed and destroyed.

All pages are included in this revision. Revision bars on the pages identify current revision changes.

NOTE: IF YOU RECEIVE PRINTED REVISIONS, PLEASE VERIFY THAT YOU HAVE RECEIVED AND FILED THE PREVIOUS REVISION. BOEING MUST BE NOTIFIED WITHIN 30 DAYS IF YOU HAVE NOT RECEIVED THE PREVIOUS REVISION. REQUESTS FOR REVISIONS OTHER THAN THE PREVIOUS REVISION WILL REQUIRE A COMPLETE MANUAL REPRINT SUBJECT TO REPRINT CHARGES SHOWN IN THE DATA AND SERVICES CATALOG.

Location of Change Description of Change

SERVICE BULLETIN LIST 24A1141 R02 Effectivities updated

24A1141 R02 Title updated 30-1047 Title updated 31-1185 Title updated 31-1362 Status Updated

34-2194 Added

56-1017 R01 Status Updated

CUSTOMER CHANGE LIST MR 39273-4 Added

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EQUIPMENT LIST 24A1141 R02



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2	BLANK	9	Oct 20/2006	WIRING DIAGRA	MS
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A 8	BLANK	6	Feb 12/2008	12	BLANK
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A = Added, R = Revised, D = Deleted, O = Overflow



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	Chapter	Date	Title
	00	FEB 09/2009	GENERAL
	21	MAY 11/2009	AIR CONDITIONING
R	22	AUG 10/2009	AUTOFLIGHT
R	23	AUG 10/2009	COMMUNICATIONS
R	24	AUG 10/2009	ELECTRICAL POWER
	25	FEB 09/2009	EQUIPMENT / FURNISHINGS
	26	FEB 09/2009	FIRE PROTECTION
	27	MAY 11/2009	FLIGHT CONTROLS
R	28	AUG 10/2009	FUEL
	29	MAY 11/2009	HYDRAULIC POWER
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	49	MAY 11/2009	AIRBORNE AUXILIARY POWER
	52	FEB 09/2009	DOORS
	73	FEB 09/2009	ENGINE FUEL AND CONTROL
R	74	AUG 10/2009	IGNITION
	75	MAY 11/2009	AIR
R	76	AUG 10/2009	ENGINE CONTROLS
	77	MAY 11/2009	ENGINE INDICATING
	78	MAY 11/2009	EXHAUST
	79	FEB 09/2009	OIL
	80	FEB 09/2009	STARTING
R	91	AUG 10/2009	CHARTS



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0	Basic	Dec 12/1997	YC001-YC007				
1		Mar 06/1998					
2		Jun 12/1998					
3	3 Follow-On		YC008-YC011				
4		Mar 23/1999					
5	5						
6	Post Delivery	Jul 27/1999					
7	Follow-On	Oct 27/1999	YC012-YC013				
8	Follow-On	Jan 26/2000	YC015-YC019				
9	Follow-On	Apr 25/2000	YC020				
10	Follow-On	Jul 21/2000	YC021				
11	Follow-On	Oct 19/2000	YC022-YC023, YC026				
12	Follow-On	Jan 17/2001	YC024-YC025, YC027				
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15	Follow-On	Nov 09/2001	YC028-YC029				
16	Follow-On	Feb 08/2002	YC030				
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19	Post Delivery	Nov 15/2002					
20	Post Delivery	Mar 03/2003					
21	Post Delivery	Jun 06/2003					
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26	Follow-On	Oct 20/2005	YK901-YK903				
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30		Oct 20/2006	YL421				

BOEING REVISION RECORD



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31	Follow-On	Jan 18/2007	YK907-YK909, YL422, YM643
32	Follow-On	Apr 18/2007	YK910, YL423
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34	Follow-On	Oct 15/2007	YL424-YL425, YM645-YM646
35	Follow-On	Jan 14/2008	YL426
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38	Follow-On	Aug 13/2008	YK912, YK919, YL401, YM652
39	Follow-On	Nov 11/2008	YL427-YL429
40	Follow-On	Feb 09/2009	YK918
41		May 11/2009	
42		Aug 10/2009	



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RECORD OF TEMPORARY REVISION



Number	Incorporated	Started/ Completed	Effectivity	ATA	Subject
21-1133	Nov 09/2001	S	YC001-YC013 YC015-YC022	21-51-12 21-51-22 24-61-22 24-61-24	AIR CONDITIONING - ZONE TEMPERATURE LIGHTS - CIRCUIT BREAKER CHANGE TO THE AIR CONDITIONING ACCESSORY UNIT
23-1107	Jan 26/2000	С	YC001-YC007	23-71-01 33-12-11 73-22-31	COMMUNICATIONS - AUDIO AND VIDEO MONITORING - COCKPIT VOICE RECORDER MANUAL - ON SWITCH INSTALLATION
23-1139 R01	May 13/2008	S	YC001-YC013	23-31-01 23-34-01 24-61-24 24-61-32	COMMUNICATIONS - PASSENGER ADDRESS AND ENTERTAINMENT - INSTALLATION OF THE PASSENGER ADDRESS AMPLIFIER AND IN-FLIGHT ENTERTAINMENT AUDIO CIRCUIT BREAKERS
23-1155	Nov 09/2001	С	YC001-YC002	23-31-06	COMMUNICATIONS - PASSENGER ADDRESS - AND ENTERTAINMENT - ANNOUNCEMENT AND BOARDING MUSIC CONTROL PANEL - MUSIC LOW VOLUME WIRING CHANGE
23-1299	May 13/2008	S	YK907-YK909	23-11-11 23-11-21 23-12-41 23-22-11 24-51-13 31-31-17	COMMUNICATIONS - HIGH FREQUENCY (HF) COMMUNICATION SYSTEM - INSTALLATION OF HF TRANSCEIVER AND HF ANTENNA COUPLER INTO EXISTING PROVISIONS
23A1169 R02	Jul 17/2007	S	YC012	23-32-31 23-32-32 23-32-41 23-32-42 23-32-43 23-32-51 23-32-52 23-32-53 23-32-61 23-32-62 23-32-63	COMMUNICATIONS - PASSENGER ADDRESS
23A1170 R02	Aug 13/2008	С	YC001-YC013 YC015-YC016	23-12-11 23-12-21 91-01-07	COMMUNICATIONS - VHF COMMUNICATION SYSTEM - VHF-1 AND VHF- 2 ANTENNAS INTERCHANGE



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Number	Incorporated	Completed	Effectivity	ATA	Subject
24-1142	Nov 09/2001	S	YC001-YC013 YC015-YC018	24-61-22 24-61-61 32-64-11	ELECTRICAL POWER - DISTRIBUTION - CARGO LEADER SYSTEM CIRCUIT BREAKER CHANGE
24-1176 R01	May 11/2009	S	YM643	24-11-11 24-11-21 24-21-11 24-21-21 24-24-11 24-24-21 77-12-21	ELECTRICAL POWER - AC GENERATION SYSTEM - INSPECTION AND RE-PIN WIRES AT TWO WING-TO- STRUT ELECTRICAL CONNECTORS AT EACH ENGINE STRUT BRACKET
24-1180	May 13/2008	S	YK907-YK909	24-51-15 25-29-11	ELECTRICAL POWER - LOAD SHED RELAY - INSTALLATION OF TWO 110V AC, 60 HZ AUXILLARY ELECTRICAL POWER OUTLETS IN THE FLIGHT DECK
24A1141 R02	Oct 15/2007	S	YC001-YC007	33-12-11 33-12-12	ELECTRICAL POWER - GENERAL - P5 OVERHEAD PANEL - BRACKET REPLACEMENT AND WIRE INSTALLATION
24A1141 R02	Aug 10/2009	S	YC008-YC011	33-12-11 33-12-12	ELECTRICAL POWER - GENERAL - P5 OVERHEAD PANEL - BRACKET REPLACEMENT AND WIRE INSTALLATION
25-1542	Sep 03/2004	S	YC001-YC013 YC016-YC018 YC020-YC026 YC028-YC030	23-32-11 23-32-12 23-32-13 23-32-21 23-32-31 23-32-51 33-25-21 33-51-12 33-51-21	EQUIPMENT AND FURNISHINGS - PASSENGER COMPARTMENT - G7 GALLEY REMOVAL
26A1083 R01	Aug 13/2008	С	YC001-YC007	24-61-21 24-61-23 24-61-51 26-11-21 26-11-31 26-16-21 26-16-22 26-23-11 31-31-15 33-11-33 33-18-62 33-18-64	FIRE DETECTION - GENERAL - CARGO COMPARTMENT SMOKE DETECTION AND FIRE EXTINGUISHING INSTALLATION -737-600/- 700/-800 AIRPLANES
	24-1142 24-1176 R01 24-1180 24-1141 R02 24-1141 R02 25-1542	24-1142 Nov 09/2001 24-1176 R01 May 11/2009 24-1180 May 13/2008 24A1141 R02 Oct 15/2007 24A1141 R02 Aug 10/2009 25-1542 Sep 03/2004	Number Incorporated Completed 24-1142 Nov 09/2001 S 24-1176 R01 May 11/2009 S 24-1180 May 13/2008 S 24A1141 R02 Oct 15/2007 S 24A1141 R02 Aug 10/2009 S 25-1542 Sep 03/2004 S	Number Incorporated Completed Effectivity 24-1142 Nov 09/2001 S YC001-YC013 YC015-YC018 24-1176 R01 May 11/2009 S YM643 24-1180 May 13/2008 S YK907-YK909 24A1141 R02 Oct 15/2007 S YC001-YC007 24A1141 R02 Aug 10/2009 S YC008-YC011 25-1542 Sep 03/2004 S YC001-YC013 YC026-YC026 YC028-YC030	Number Incorporated Completed Effectivity ATA



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27-1220	Jan 26/2000	С	YC001-YC007	27-31-11 27-31-37 27-32-11 27-32-21 27-32-31	FLIGHT CONTROLS - ELEVATOR AND TAB - CHANGES TO THE STALL IDENTIFICATION SYSTEMS TO SUPPORT THE EQUIREMENTS OF THE JAA
27-1247	Mar 31/2005	S	YC001-YC013 YC015-YC026 YC028-YC029	27-23-11 27-25-11	FLIGHT CONTROLS - RUDDER AND RUDDER TRIM CONTROL SYSTEM - WIRE PROVISIONS FOR RUDDER SYSTEM ENHANCEMENT
27-1253 R03	Feb 09/2009	С	YC001-YC013 YC015-YC026 YC028-YC030	22-23-11 24-61-24 27-23-11 27-25-11 29-35-11 91-04-15	FLIGHT CONTROLS - RUDDER AND RUDDER TRIM CONTROL SYSTEM - INSTALLATION OF AN ENHANCED RUDDER CONTROL SYSTEM (RSEP)
27-1262	May 11/2009	С	YC001-YC013 YC015-YC026 YC028-YC030		FLIGHT CONTROLS - RUDDER AND RUDDER TRIM CONTROL SYSTEM - REPLACEMENT OF THE P5- 3 PANEL (RSEP)
27-1285	Feb 09/2009	С	YK907	27-62-11	FLIGHT CONTROLS - SPEEDBRAKE CONTROL SYSTEM - "SPEED BRAKE DO NOT ARM" LIGHT NUISANCE INDICATION CORRECTION
27A1219	Oct 27/1999	С	YC001-YC007	27-53-11	FLIGHT CONTROLS - FLAP, SLAT ELECTRONICS UNIT REVISION TO THE AIR/ GROUND SIGNAL SOURCE
27A1228 R01	Jan 26/2000	С	YC001-YC011	27-41-12	FLIGHT CONTROLS - TRAILING EDGE FLAP AND HORIZONTAL STABILIZER TRIM SYSTEMS - S245 FLAP LIMIT SWITCH INSPECTION AND R850 RELAY INSTALLATION



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	28-1194 R02	Oct 15/2007	S	YC001-YC013 YC015-YC026 YC028-YC030	24-21-11 24-21-21 24-24-11 24-24-21 26-12-11 27-81-11 28-23-11 29-11-11 30-11-11 33-45-11 74-11-11 77-31-11	FUEL - FUEL BOOST PUMP - RELOCATION OF CONNECTOR PINS AT WING/BODY CONNECTORS
	28-1257 R01	Aug 13/2008	S	YC001-YC013 YC015-YC026 YC028-YC030 YK901	28-23-11 91-02-05	FUEL - ENGINE FUEL FEED SYSTEM - BOOST PUMP CONTROL RELAY GROUND LOCATION CHANGE
R	30-1047	Oct 19/2000	С	YC001-YC007	30-41-11 30-41-12 31-52-56 33-18-34	ICE AND RAIN PROTECTION - WINDOWS, WINDSHIELDS AND DOORS - NO.4 AND NO.5 WINDOW HEATING WIRING AND INDICATION CHANGE
	31-1136	Jul 16/2001	С	YC001-YC011	22-11-53 23-27-11 23-32-21 31-32-15 31-35-01 31-62-15 31-62-25 34-61-18 34-61-22 34-61-26 49-62-11	INDICATING/RECORDING SYSTEMS - DATA LOAD SELECTOR SWITCH REPLACEMENT
	31-1136 R01	Apr 18/2006	С	YC001-YC011	22-11-53	INDICATING/RECORDING SYSTEMS - DATA LOAD SELECTOR SWITCH REPLACEMENT
	31-1136 R02	Jul 17/2007	S	YC001-YC011	22-11-53	INDICATING/RECORDING SYSTEMS - DATA LOAD SELECTOR SWITCH REPLACEMENT
R	31-1185	May 09/2002	С	YC001-YC013 YC015-YC019	34-61-19	INDICATING AND RECORDING SYSTEMS - CENTRAL DISPLAY SYSTEMS - REPLACEMENT OF COMMON DISPLAY SYSTEM SOFTWARE
	31-1267	Apr 18/2007	С	YC001-YC013 YC015-YC026 YC028-YC030	23-22-11 31-62-41	COMMON DISPLAY SYSTEM (CDS) - OPTIONAL FEATURES - ACTIVATION



ber	Incorporated	Started/ Completed	Effectivity	АТА	Subject
284	Feb 12/2008	С	YC001-YC013 YC015-YC026 YC028-YC030	26-11-21 31-52-11 33-18-51	INDICATING/RECORDING SYSTEMS - MASTER CAUTION - ADDITION OF QUICK DISCONNECTS TO THE MASTER CAUTION AND FIRE WARNING SWITCHLIGHT WIRES
362	Feb 09/2009	S	YK918	31-35-03	INDICATING/RECORDING SYSTEMS - RECORDERS - WIRING REVISION FOR TELEDYNE WIRELESS QUICK ACCESS RECORDER
362	Nov 11/2008	S	YL421-YL423	31-35-03	INDICATING/RECORDING SYSTEMS - RECORDERS - WIRING REVISION FOR TELEDYNE WIRELESS QUICK ACCESS RECORDER
362	Aug 10/2009	С	YM645-YM646	31-35-03	INDICATING/RECORDING SYSTEMS - RECORDERS - WIRING REVISION FOR TELEDYNE WIRELESS QUICK ACCESS RECORDER
116	Jul 16/2001	С	YC001-YC007	24-53-11	LIGHTS - PASSENGER COMPARTMENT LIGHTS - PASSENGER SERVICE UNIT NO-SMOKING FASTEN SEAT BELT SIGN WIRING CHANGE
461	Jan 26/2000	С	YC001-YC007	34-58-21	NAVIGATION - DEPENDENT POSITION DETERMINING - GLOBAL POSITIONING SYSTEM (GPS) MASTER CAUTION DISPLAY CHANGE
767 R01	Apr 18/2006	С	YC001-YC013 YC015-YC026 YC028-YC030	22-11-51 34-21-13 34-21-23 34-53-11 34-53-21 34-58-11 34-58-21 34-61-14	NAVIGATION - DEPENDENT POSITION DETERMINING - AIR TRAFFIC CONTROL (ATC) SYSTEM - WIRING PROVISIONS FOR FUTURE ACTIVATION OF ELEMENTARY SURVEILLANCE (ELS), ENHANCED SURVEILLANCE (EHS) AND EXTENDED SQUITTER (ESTARRAMETERS
	284 362 362 116	Feb 12/2008 Feb 09/2009 Nov 11/2008 Aug 10/2009 Jul 16/2001 Jan 26/2000	Incorporated Completed Reb 12/2008 C Reb 12/2009 S Reb 19/2009 S Reb 19/2009 S Reb 19/2009 C Rep 19/2009 C	ber Incorporated Completed Effectivity 284 Feb 12/2008 C YC001-YC013 YC015-YC026 YC028-YC030 362 Feb 09/2009 S YK918 362 Nov 11/2008 S YL421-YL423 362 Aug 10/2009 C YM645-YM646 362 Aug 16/2001 C YC001-YC007 361 Jan 26/2000 C YC001-YC007 367 R01 Apr 18/2006 C YC001-YC013 YC015-YC026	ber Incorporated Completed Effectivity ATA 284 Feb 12/2008 C YC001-YC013 YC015-YC026 31-52-11 YC028-YC030 33-18-51 362 Feb 09/2009 S YK918 31-35-03 362 Nov 11/2008 S YL421-YL423 31-35-03 362 Aug 10/2009 C YM645-YM646 31-35-03 362 Aug 10/2009 C YC001-YC007 24-53-11 461 Jan 26/2000 C YC001-YC007 34-58-21 767 R01 Apr 18/2006 C YC001-YC013 YC013 YC015-YC026 YC028-YC030 34-21-33 YC028-YC030 34-23-334-53-11 YC028-YC030 34-23-334-53-11 YC038-YC031 YC028-YC030 34-23-334-53-21 YC028-YC030 34-23-334-53-21 YC028-YC030 34-23-334-53-21 YC028-YC030



	Number	Incorporated	Started/ Completed	Effectivity	ATA	Subject
	34-1915	Apr 18/2007	С	YC008-YC013 YC015-YC026 YC028-YC030	34-49-11	NAVIGATION - INDEPENDENT POSITION DETERMINING SYSTEMS- GROUND PROXIMITY WARNING SYSTEMS (GPWS)-ENCHANCED GPWS PEAKS AND OBSTACLES OPTION ACTIVATION
	34-1916	Apr 18/2007	С	YC001-YC013 YC015-YC026 YC028-YC030	34-61-19	FLIGHT MANAGEMENT COMPUTER SYSTEM (FMCS) - OPTIONAL FEATURES - REVISION
	34-2082	Jan 14/2008	S	YK907-YK909	34-49-11	NAVIGATION - ENHANCED GROUND PROXIMITY WARNING SYSTEM - ALTITUDE CALLOUT - REVISION
	34-2083	Jan 14/2008	S	YC001-YC002 YC010-YC013 YC015-YC016 YK901-YK909	34-61-19	NAVIGATION - FLIGHT MANAGEMENT COMPUTER SYSTEM - CHANGE FROM JAA TO FAA FLIGHT RULES AND IN TRANSPORT CANADA FORMAT
	34-2107 R01	Nov 11/2008	S	YC003-YC004 YC008-YC009 YC017-YC026 YC028-YC030 YL421-YL426	34-61-19	NAVIGATION - FLIGHT MANAGEMENT COMPUTER SYSTEM - CHANGE FROM JAA TO FAA FLIGHT RULES AND IN TRANSPORT CANADA FORMAT
	34-2107 R01	Feb 09/2009	S	YK918	34-61-19	NAVIGATION - FLIGHT MANAGEMENT COMPUTER SYSTEM - CHANGE FROM JAA TO FAA FLIGHT RULES AND IN TRANSPORT CANADA FORMAT
Α	34-2194	Aug 10/2009	S	YC028-YC029	33-11-41 33-18-62 34-57-21	NAVIGATION - AUTOMATIC DIRECTION FINDER SYSTEM - SECOND AUTOMATIC DIRECTION FINDER INSTALLATION
R	56-1017 R01	Aug 10/2009	С	YC001-YC013 YC015-YC026 YC028-YC030	30-41-11 30-41-12	WINDOWS - FLIGHT COMPARTMENT - CONTROL CABIN FIXED WINDOWS - DELETION OF NUMBER 4 AND NUMBER 5 WINDOWS



Number	Incorporated	Started/ Completed	Effectivity	АТА	Subject
71-1461	Nov 15/2002	S	YC001-YC013 YC015-YC017	34-61-19	CFM56-7 ENGINES 7B26 RATING
71-1605	May 13/2008	S	YK907-YK909	34-61-19	MODIFY AIRPLANE OPERATING THRUST - CHANGE FROM 7B26 RATING TO A 7B27 RATING
71-1612	May 13/2008	S	YM643 YM645-YM647 YM649 YM651	34-61-19	FMC ENGINE THRUST FOR CFM56-7B22 RATINIG IN LIEU OF -7B20
77-1455	May 09/2002	S	YC018-YC026	34-61-19	CFM56-7 ENGINE - 7B26 RATING IN LIEU OF 7B27/ B1 RATING
PD 0802664	May 13/2008	S	YC001-YC010 YC012-YC013 YC015-YC026 YC028-YC030	24-61-24 27-32-11 27-32-21 27-62-11 27-62-41 33-43-11 33-44-12	APB RETROFIT WINGLETS
SL-33-042	Sep 24/2003	S	YC021-YC026	33-21-12	SIDEWALL LIGHTING COMPONENTS - CASE GROUNDING



Number	Incorporated	Started/ Completed	Effectivity	ATA	Subject
230106	Oct 20/2006	С	YC001-YC013 YC015-YC026 YC028-YC030	23-70-11 24-61-23 33-11-41	COCKPIT DOOR SURVEILLANCE SYSTEM
737-EB34-0155 R04	Aug 07/2002	C	YC001-YC007	22-11-51 24-61-21 27-32-12 27-32-22 31-62-15 31-62-25 33-11-21 33-18-22 34-21-13 34-21-23 34-21-24 34-31-11 34-33-21 34-33-21 34-34-31-11 34-35-21 34-41-11 34-58-11 34-58-21 34-61-15	INSTALLATION OF MARK V ENHANCED GROUND PROXIMITY WARNING SYSTEM (EGPWS)
737-EB34-0192	Aug 07/2002	С	YC001-YC007	34-21-14 34-21-24 34-41-11 34-49-11 34-53-11 34-53-21	NAVIGATION - WXR SYSTEM, PWS PERIPHERAL SYSTEMS CONNECTIONS
737-EB34-0193	Nov 15/2002	С	YC001-YC007	34-41-11	NAVIGATION - WXR PWS SYSTEMS ACTIVATION
737-EB34-0428	May 11/2009	С	YC001-YC007	34-49-11	CHANGE EGPWS PROGRAM PIN STRAPPING
EO 23-80059-6	Oct 15/2007	S	YC001-YC013 YC015-YC026 YC028-YC030	23-12-31 23-27-14 23-27-16 34-31-11 34-53-11 34-53-21 34-58-11	WIRE PROVISION FOR ACARS VDL MODE 2 OPERATION
EO 23-80113-8	Feb 09/2009	С	YK907-YK909	23-24-11	ELT CODING
EO 31-80019-8	Oct 20/2006	С	YC001-YC013 YC015-YC026 YC028-YC030		ADL INSTALLATION
EO B7378-33-04- 01	Sep 03/2004	S	YC001-YC013 YC015-YC026 YC028-YC030		CHANGE OF EMERGENCY LIGHT POWER SUPPLIES
A MR 39273-4	Aug 10/2009	С	YC005-YC007	23-11-11	ADD HF COM SYSTEM TO DOCUMENTATION

CUSTOMER CHANGE LIST



CH-SC-SU	Title
24-52-11	115V AC GROUND SERVICE BUS
24-51-12	115V AC MAIN BUS 1
24-51-22	115V AC MAIN BUS 2
24-54-11	115V AC STANDBY BUS
24-51-11	115V AC TRANSFER BUS 1
24-51-15	115V AC TRANSFER BUS 1 IFE/PASS SEAT
24-51-13	115V AC TRANSFER BUS 1 SECTION 1
24-51-14	115V AC TRANSFER BUS 1 SECTION 2
24-51-21	115V AC TRANSFER BUS 2
24-51-25	115V AC TRANSFER BUS 2 IFE/PASS SEAT
24-51-23	115V AC TRANSFER BUS 2 SECTION 1
24-51-24	115V AC TRANSFER BUS 2 SECTION 2
24-53-11	28V AC BUSES
24-61-11	28V DC BUSES
24-61-31	28V DC BUSES BATTERY BUS SECTION 1
24-61-32	28V DC BUSES BATTERY BUS SECTION 2
24-61-33	28V DC BUSES BATTERY BUS SECTION 3
24-61-21	28V DC BUSES BUS 1 SECTION 1
24-61-22	28V DC BUSES BUS 1 SECTION 2
24-61-23	28V DC BUSES BUS 2 SECTION 1
24-61-24	28V DC BUSES BUS 2 SECTION 2
24-61-61	28V DC BUSES GROUND SERVICE BUS
24-61-51	28V DC BUSES HOT BATTERY BUS
24-61-41	28V DC BUSES STANDBY BUS SECTION 1
24-61-42	28V DC BUSES STANDBY BUS SECTION 2
24-61-52	28V DC BUSES SWITCH HOT BATTERY BUS
24-28-11	AC INDICATION P5-13
24-28-21	AC SYSTEM GENERATOR AND APU INDICATION P5-4



CH-SC-SU	Title
24-23-31	AC TIE BUS
23-27-13	ACARS (724B) - CONTROL AND DISPLAY
23-27-12	ACARS (724B) - DFDAU AND E11 INTERFACES
23-27-14	ACARS (724B) - FMC, PRINTER AND DISCRETE INTERFACES
23-27-11	ACARS (724B) - POWER, COMM AND ANNUNCIATION INTERFACES
23-27-16	ACARS (724B) - REGISTRY AND AIRLINE ID CODE
23-27-15	ACARS (724B) SATCOM AND AIRSHOW INTERFACES
34-57-11	ADF-1
34-57-21	ADF-2
34-21-16	ADIRS - NO COOLING AND "ON DC" OPERATION WARNING
91-04-15	AFT BODY BRACKETS
26-16-22	AFT CARGO COMPARTMENT SMOKE DETECTION
27-18-11	AILERON POSITION INDICATION
27-11-11	AILERON TRIM CONTROL
21-61-51	AIR CONDITIONING TEMPERATURE INDICATION
34-21-14	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - LEFT ADR OUTPUTS
34-21-11	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - LEFT CONTROL & WARNING
34-21-13	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - LEFT IR OUTPUTS
34-21-12	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - LEFT SYSTEM INPUTS
34-21-24	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - RIGHT ADR OUTPUTS
34-21-21	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - RIGHT CONTROL & WARNING
34-21-23	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - RIGHT IR OUTPUTS
34-21-22	AIR DATA INERTIAL REFERENCE SYSTEM (ADIRS) - RIGHT SYSTEM INPUTS
34-21-15	AIR DATA INERTIAL REFERENCE SYSTEM - IR SIGNAL SWITCHING
34-16-11	AIR DATA INERTIAL REFERENCE SYSTEM OVERSPEED TEST LEFT
34-16-21	AIR DATA INERTIAL REFERENCE SYSTEM OVERSPEED TEST RIGHT
32-31-11	AIR/GND SYSTEM 1 & LEVER LATCH
32-31-12	AIR/GND SYSTEM 2 AND NOSE GEAR GROUND SENSING RELAYS



CH-SC-SU	Title
91-00-01	AIRPLANE STATION BODY & VERTICAL STABLISER
91-00-02	AIRPLANE STATION WING & HORIZONTAL STABLIZER
32-35-11	ALTERNATE LANDING GEAR EXTENSION SYSTEM
27-53-11	ALTERNATE TRAILING AND LEADING EDGE FLAP DRIVE
91-01-07	ANTENNA LOCATION
32-41-11	ANTISKID SYSTEM
49-52-31	APU BLEED AIR AND SURGE CONTROL SYSTEM
49-62-11	APU CONTROL- AIRPLANE INTERFACE
49-62-12	APU CONTROL- APU INTERFACE
49-71-21	APU EGT SYSTEM
26-11-31	APU FIRE DETECTION
49-41-11	APU IGNITION AND STARTING SYSTEM
49-94-21	APU OIL SYSTEM
23-27-35	ARINC 716 VHF 3/ SINGLE FMC 702 INTERFACES
34-53-31	ATC ANTENNA SELECT
34-53-11	ATC TRANSPONDER 1
34-53-21	ATC TRANSPONDER 2
33-26-11	ATTENDANT WORK LIGHTS
23-34-02	AUDIO ENTERTAINMENT DISTRIBUTION
23-34-01	AUDIO ENTERTAINMENT SYSTEM POWER AND CONTROL
23-51-11	AUDIO INTEGRATING CAPTAIN
23-51-51	AUDIO INTEGRATING ELECTRONIC SHELF
23-51-21	AUDIO INTEGRATING FIRST OFFICER
23-51-31	AUDIO INTEGRATING OBSERVER
31-53-11	AURAL WARNING - TAKEOFF WARNING
31-51-11	AURAL WARNING SYSTEM
32-42-11	AUTOBRAKE SYSTEM
27-62-11	AUTOMATIC GROUND SPEEDBRAKE CONTROL



CH-SC-SU	Title
24-28-41	AUTOMATIC LOAD SHED GALLEYS AND MAIN BUSES
27-83-11	AUTOSLAT SYSTEM NO. 1
27-83-21	AUTOSLAT SYSTEM NO. 2
22-31-22	AUTOTHROTTLE SYSTEM - DIGITAL INPUTS FROM SMYDC'S AND RA'S
22-31-51	AUTOTHROTTLE SYSTEM - ENGAGE/DISENGAGE
22-31-52	AUTOTHROTTLE SYSTEM - MODE ANNUNCIATION
22-31-31	AUTOTHROTTLE SYSTEM - SERVO MOTOR 1
22-31-41	AUTOTHROTTLE SYSTEM - SERVO MOTOR 2
22-31-11	AUTOTHROTTLE SYSTEM- POWER, TOGA, PROGRAM PINS AND TEST BUSSES
24-31-11	BATTERY AND BATTERY CHARGER
28-21-21	BATTERY AND CONTROL POWER ENGINE SPAR AND APU FUEL VALVES
24-31-12	BATTERY BUS
36-21-11	BLEED AIR PRESSURE INDICATION
36-11-11	BLEED AIR VALVE CONTROL
24-21-52	BUS POWER CONTROL UNIT BLOCK (G15)
21-25-11	CABIN AIR RECIRCULATION SYSTEM
23-42-11	CABIN INTERPHONE
23-42-12	CABIN INTERPHONE - ATTENDANT HANDSETS
21-33-11	CABIN PRESSURE WARNING
33-11-31	CAPTAIN'S AND FIRST OFFICER'S CENTER INSTR PANEL LIGHTING
33-11-11	CAPTAIN'S INSTRUMENT PANEL LIGHTING
26-23-11	CARGO COMPARTMENT FIRE EXTINGUISHERS
25-51-21	CARGO LOADER SYSTEM AFT CARGO COMPARTMENT
31-62-13	CDS - CONTROL PANEL INTERFACES - CAPTAINS
31-62-23	CDS - CONTROL PANEL INTERFACES - F/O
31-62-15	CDS - DEU 1 AVIONICS INTERFACES
31-62-14	CDS - DEU 1 ENGINE HYDRAULIC, APU AND FUEL INTERFACES
31-62-25	CDS - DEU 2 AVIONICS INTERFACES



CH-SC-SU	Title
31-62-24	CDS - DEU 2 ENGINE HYDRAULIC, APU AND FUEL INTERFACES
31-62-41	CDS - DEU SELECT AND PROGRAM PINS
31-62-11	CDS - POWER DISTRIBUTION AND INSTRUMENT LIGHTING CAPTAINS
31-62-21	CDS - POWER DISTRIBUTION AND INSTRUMENT LIGHTING FIRST OFFICER
31-62-12	CDS - VIDEO COAX SPLITTERS 1 AND 3 DU STATUS
31-62-22	CDS - VIDEO COAX SPLITTERS 2 AND 4 DU STATUS
31-62-42	CDS - VIDEO MONITORING
34-61-23	CDU/MCDU INTERFACE
34-61-26	CDU/MCDU/DATA LOADER INTERFACE
33-22-22	CEILING LIGHTS - LEFT AFT
33-22-21	CEILING LIGHTS - LEFT FORWARD
33-22-32	CEILING LIGHTS - RIGHT AFT
33-22-31	CEILING LIGHTS - RIGHT FORWARD
33-22-11	CEILING LIGHTS CONTROL
91-02-00	CIRCUIT BREAKER LIST
23-27-38	CMU 000I, CREW ADVISORIES AND OUTPUT 8
23-27-39	CMU INTERFACES - DATA LOADER PROGRAM PINS
23-27-32	CMU OUTPUT 1 INTERFACES - FMC 1/MCDU 1/PRINTER
23-27-33	CMU OUTPUT 2 INTERFACES - FMC 2/MCDU 2/DFDAU/MIDU
23-27-37	CMU OUTPUT 6/7 INTERFACES - COMMUNICATIONS
23-27-31	CMU POWER & CONTROL
23-71-01	COCKPIT VOICE RECORDER
52-51-11	CONTROL CABIN DOOR LOCK
35-11-11	CREW OXYGEN SYSTEM
31-32-15	DATA LOADER INTERFACE
24-33-13	DC BUS INDICATION DFDAU
24-33-11	DC VOLTAGE AND CURRENT INDICATIONS
22-12-31	DFCS - A A/P PITCH SENSORS AND ACTUATORS



CH-SC-SU	Title
22-11-31	DFCS - A AND B A/P ROLL SENSORS AND ACTUATORS
22-11-11	DFCS - A AND B FCC POWER AND 26V AC EXCITATION
22-11-75	DFCS - A AND B OPTION PINS
22-13-11	DFCS - A AND B SPEED AND STABILIZER TRIM
22-11-12	DFCS - A AND B SYSTEM INTERLOCKS
22-14-11	DFCS - ANNUNCIATION AND WARNING
22-12-41	DFCS - B A/P PITCH SENSORS AND ACTUATORS
22-11-53	DFCS - DIGITAL BUS INTERFACES - DATA LOADER
22-11-51	DFCS - DIGITAL BUS INTERFACES - OUTPUT
22-11-14	DFCS - INTERSYSTEM SWITCHING
22-18-11	DFCS - MACH TRIM
22-11-81	DFCS - RUDDER COMMAND AND CONTROL
22-11-18	DFCS INTERSYSTEM SWITCHING - VHF NAV ANTENNAS
27-81-41	DFDAU AND TEST CONNECTOR INTERFACE
24-24-31	DIFFERENTIAL CURRENT PROTECTION APU GEN
24-24-11	DIFFERENTIAL CURRENT PROTECTION GEN NO. 1
24-24-21	DIFFERENTIAL CURRENT PROTECTION GEN NO.2
31-35-02	DIGITAL FLIGHT DATA RECORDER SYSTEM - ACMS INTERFACE
31-35-04	DIGITAL FLIGHT DATA RECORDER SYSTEM - ACMS INTERFACE
31-35-03	DIGITAL FLIGHT DATA RECORDER SYSTEM - ACMS INTERFACE 3
31-31-14	DIGITAL FLIGHT DATA RECORDER SYSTEM - ANALOG INTERFACE
31-35-01	DIGITAL FLIGHT DATA RECORDER SYSTEM - DATA LOADER INTERFACE
31-31-13	DIGITAL FLIGHT DATA RECORDER SYSTEM - DIGITAL INTERFACE
31-31-11	DIGITAL FLIGHT DATA RECORDER SYSTEM - INTERFACE
31-31-16	DIGITAL FLIGHT DATA RECORDER SYSTEM AIRPLANE CODING
31-31-15	DIGITAL FLIGHT DATA RECORDER SYSTEM DISCRETE INTERFACE
31-31-17	DIGITAL FLIGHT DATA RECORDER SYSTEM MANDATORY OPTIONS
91-04-00	DISCONNECT BRACKET LIST



CH-SC-SU	Title
91-01-06	DISCONNECT BRACKET LOCATION
34-55-11	DME NO. 1
34-55-21	DME NO. 2
33-14-12	DOME LTG, CONTROL STAND FLOOD LTG, STBY COMPASS LTG, & C/B PNL LTG
21-42-11	DOOR AREA HEATERS
52-71-13	DOOR WARNING OVERWING - LEFT OVERWING
52-71-14	DOOR WARNING OVERWING - RIGHT OVERWING
52-71-12	DOOR WARNING SYSTEM - CARGO AND EQUIPMENT
52-71-11	DOOR WARNING SYSTEM - ENTRY AND SERVICE
30-71-11	DRAIN HEATERS
91-03-01	E1 RACK
91-04-04	E1 SHELF BRACKETS
91-03-02	E2 RACK
91-04-05	E2-1 AND E2-2 BRACKETS
91-04-06	E3 BRACKETS
91-03-03	E3 RACK
91-03-04	E4 RACK
91-04-07	E4-1 AND E4-2 BRACKETS
91-04-08	E5 BRACKETS
91-03-05	E5 RACK
91-03-06	E6 RACK
91-04-11	E8 BRACKET
91-03-08	E8 RACK
49-62-51	ECU PINOUT
29-11-12	ELECTRIC HYDRAULIC PUMP CONTROL
31-22-11	ELECTRONIC CLOCK
73-21-12	ELECTRONIC ENGINE CONTROL ALTERNATE MODE
73-22-11	ELECTRONIC ENGINE CONTROL ELECTRICAL POWER



CH-SC-SU	Title
27-31-37	ELEVATOR FEEL DIFFERENTIAL PRESSURE
27-38-11	ELEVATOR POSITION INDICATION
27-31-11	ELEVATOR TAB CONTROL
33-51-21	EMERGENCY EXIT LIGHT - FWD - FLOOR PROXIMITY
33-51-14	EMERGENCY EXIT LIGHTS - AFT
33-51-22	EMERGENCY EXIT LIGHTS - AFT - PROXIMITY
33-51-11	EMERGENCY EXIT LIGHTS - CONTROL
33-51-12	EMERGENCY EXIT LIGHTS - FORWARD
33-51-13	EMERGENCY EXIT LIGHTS - MID
23-24-11	EMERGENCY LOCATOR TRANSMITTER
73-24-11	ENGINE 1 DATABUS OUTPUT
26-11-11	ENGINE 1 FIRE DETECTION
76-21-11	ENGINE 1 FUEL CONDITION CONTROL
73-25-11	ENGINE 1 FUEL CONTROL
30-21-11	ENGINE 1 NACELLE ANTI-ICE
78-34-11	ENGINE 1 THRUST REVERSER CONTROL
78-36-11	ENGINE 1 THRUST REVERSER FLIGHT DECK INDICATION
78-35-11	ENGINE 1 THRUST REVERSER POSITION / THRUST LEVER INTERLOCK
78-32-51	ENGINE 1 THRUST REVERSER SYNCHRONOUS SHAFT LOCKS
75-31-11	ENGINE 1 TURBINE CLEARANCE / TEMPERATURE
75-31-12	ENGINE 1 VARIABLE STATOR VANE / BLEED CONTROL
77-31-11	ENGINE 1 VIBRATION MONITORING SYSTEM
73-24-21	ENGINE 2 DATABUS OUTPUT
26-11-21	ENGINE 2 FIRE DETECTION
76-21-21	ENGINE 2 FUEL CONDITION CONTROL
73-25-21	ENGINE 2 FUEL CONTROL
30-21-21	ENGINE 2 NACELLE ANTI-ICE
78-34-21	ENGINE 2 THRUST REVERSER CONTROL



CH-SC-SU	Title
78-36-21	ENGINE 2 THRUST REVERSER FLIGHT DECK INDICATION
78-35-21	ENGINE 2 THRUST REVERSER POSITION / THRUST LEVER INTERLOCK
78-32-61	ENGINE 2 THRUST REVERSER SYNCHRONOUS SHAFT LOCKS
75-31-21	ENGINE 2 TURBINE CLEARANCE / TEMPERATURE
75-31-22	ENGINE 2 VARIABLE STATOR VANE / BLEED CONTROL
77-31-21	ENGINE 2 VIBRATION MONITORING SYSTEM
26-21-11	ENGINE AND APU FIRE EXTINGUISHING SYSTEM
73-21-31	ENGINE BLEED AIR THRUST CONTROL
73-32-11	ENGINE CONTROL EEC FAULT INDICATION
73-24-12	ENGINE DATABUS INPUT
77-21-11	ENGINE EXHAUST GAS TEMPERATURE INDICATION
73-21-11	ENGINE EXTERNAL RESET AND CONFIGURATION CONTROL
28-21-11	ENGINE FUEL SHUT-OFF VALVES
73-31-11	ENGINE FUEL SYSTEM INDICATION
29-11-11	ENGINE HYDRAULIC PUMP CONTROL
73-23-11	ENGINE IDLE CONTROL
74-31-11	ENGINE IGNITION CONTROL
74-11-11	ENGINE IGNITION POWER
77-12-11	ENGINE N1 SPEED INDICATION
77-12-21	ENGINE N2 SPEED INDICATION
79-33-11	ENGINE OIL FILTER BYPASS WARNING
79-32-11	ENGINE OIL PRESSURE INDICATION
79-31-11	ENGINE OIL QUANTITY INDICATION
79-34-11	ENGINE OIL TEMPERATURE INDICATION
73-22-31	ENGINE RUNNING CONTROL
80-11-11	ENGINE STARTING SYSTEM
73-21-21	ENGINE THRUST-LEVER-ANGLE RESOLVER
33-29-11	ENTRY LIGHTS

CH-SC-SU	Title
21-27-21	EQUIPMENT COOLING EXHAUST
21-27-31	EQUIPMENT COOLING SUPPLY
	EQUIPMENT LIST
33-44-11	EXTERIOR LIGHTS - ANTICOLLISION - RED
33-44-12	EXTERIOR LIGHTS - ANTICOLLISION - WHITE
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CH-SC-SU	Title
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CH-SC-SU	Title
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CH-SC-SU	Title
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CH-SC-SU	Title
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1. APPLICABILITY

This Wiring Diagram Manual is applicable only to those Boeing airplanes listed on the Effective Aircraft page. The instructions and information contained herein apply solely to those airplanes and are not suitable for use with any other Boeing airplane(s).

2. GENERAL DESCRIPTION

The Boeing Wiring Diagram Manual (WDM) is a collection of diagrams, drawings, and Lists which define the wiring and hookup of associated equipment installed on the listed Boeing airplanes. These data are prepared essentially in accordance with the ATA Specification No. 2200, revision 2001.1.

This manual may also contain data and information provided by the customer. The Boeing Company assumes no responsibility for the accuracy and validity of data and information provided by a customer.

The WDM document number is unique to the customer whose name appears on the title page. Each chapter is preceded by its own Table of Contents (TOC), List of Effective Pages (LEP), and Alphabetical Index.

NOTE: System Schematics reside in a separate System Schematics Manual. Standard Wiring Practices—Chapter 20 reside in a separate Standard Wiring Practices manual (D6-54446).

All Wiring Diagrams are shown, unless otherwise specified, with the airplane on the ground, after normal flight, with the shutdown checklist complete (power off).

3. PROCESS CONTROLS

Control of the various manufacturing and installation processes used for wiring the airplane is covered in D6-36911 - Electrical Wiring Assembly and Installation Processes.

4. BOEING CHANGE DEFINITIONS

Changes used by Boeing to implement airplane changes that may affect this manual are listed below.

A. Customer Originated Changes (COC)

Customer Originated Changes are requests to incorporate airplane data, information, changes and modifications authorized by a customer into the WDM.

MOTE: Boeing will not undertake to test or evaluate, in any form, the validity or the technical accuracy of Customer Originated Changes. This will remain the sole responsibility of the customer submitting the Customer Originated Change request.

B. Service Bulletin (SB)

Service Bulletins provide information for accomplishing an engineering change on in-service airplanes. Service Bulletins are incorporated into this manual only upon customer request.

C. Service Letter (SL)

Service Letters notify customers of unique maintenance or operational items.



D. Master Change (MC)

An engineering change is classified as a Master Change if the change appreciably affects the terms and conditions of the purchase agreement and/or the customer detail specification. Consequently, a Master Change must be negotiated with the customer to revise the airplane delivery schedule, contract price, performance, weight and balance, or any other design affecting specification language.

E. Production Revision Record (PRR)

A Production Revision Record is an engineering change initiated by Boeing which is nonnegotiable and is used to make airplane changes such as design improvements.

F. Rapid Revision (RR)

A Rapid Revision is a minor change to an airplane, requested by a customer, which is too late in the production process to generate a Master Change and is not applicable for PRR action.

G. Modification Revision (MR)

A Modification Revision is used by Boeing to describe, negotiate, control and record changes to a customer's airplane configuration after an airplane has been certified and used in revenue service, then returned to Boeing jurisdiction for rework. Modification Revisions may also be used to modify an airplane directly off the assembly line.

Modification Revisions are negotiable changes and may include engineering, fabrication, assembly and/or installation revisions. They may be used by Boeing to incorporate customer furnished kits and parts, or to incorporate Boeing furnished parts which are not in kit form.

Modification Revisions may include SB's, PRR's (represented by a SB), MC's, RR's, etc. On a modification program, whenever a SB number is available, it will be used in preference to the associated MR number.

H. Electrical Liaison Change Commitment Record (ELCCR)

An Electrical Liaison Change Commitment Record is used to process miscellaneous changes on an expedited basis (out of sequence), and to incorporate these changes into the airplane in sequence in the shortest possible time.

I. Boeing Change Reason (BCR)

Boeing Change Reason provides tracking of a change made to the content of the manual that apply to all users of the manual.

5. DESCRIPTION OF SERVICE BULLETIN LIST AND CUSTOMER CHANGE LIST

A. Number Field

The service bulletin or customer change number with it's revision level

B. Incorporated

The date of the manual revision which incorporated the change.



C. Started/Completed

The status of the change. An 'S' is used in the Started/Completed column to indicate Start (Dual) configuration, a 'C' is used to indicate Complete (Final) configuration and a 'X' indicates canceled changes that have been removed from the manual.

D. Effectivity

The aircraft affected by the referenced change.

E. ATA

The list of drawings affected by the referenced change.

F. Subject

The title of the service bulletin or customer change.

6. BOEING COMMERCIAL PUBLICATION CHANGE REQUEST (PCR)

Communications concerning this manual should be directed to:

The Boeing Commercial Airplane Group Attention: Supervisor, Commercial Publications PO Box 3707 M/S 2H-61 Seattle, WA 98124-2207

Or access MyBoeingFleet website and complete the online PCR form.

To facilitate uniform handling and to provide direct routing of questions to the proper Boeing organization, use of the Publication Change Request is encouraged. Boeing makes this form available through the customer's publications organizations.

7. REFERENCE LINES

The Reference Lines provide an exact position within the airplane to aide in locating equipment.

A. Station Line (STA)

Edge view of vertical reference plane which divides the body, wing, nacelle etc., into sections.

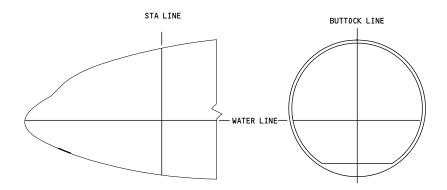
B. Waterline (WL)

Edge view of longitudinal horizontal reference plane.



C. Buttock Line (BL)

Edge view of longitudinal vertical reference plane.



8. WIRE SEPARATION

Airplane wiring installation complies with Federal Aviation Regulation (FAR) 25. Special emphasis is placed on wire separation of redundant systems for safety and other considerations. Wire separation is also utilized to minimize electromagnetic interference.

For further information see Codes section of the Introduction.



The following is a list of abbreviations and acronyms used in this manual. Where marked with an asterisk (*), see the GENERAL INFORMATION section, in the Wiring Diagram manual, for additional definition information.

A/C Air Conditioning

A/C Aircraft
A/R Altitude Rate

ACARS ARINC Communications Addressing and Reporting System

ACE Actuator Control Electronics

ACESS Advance Cabin Entertainment and Service System

ACM Air Cycle Machine

ACMP Alternating Current Motor Pump (See also EMP)

ACMS Airplane Conditioning Monitoring System

ACP Audio Control Panel

ADF Automatic Direction Finder
ADI Attitude Director Indicator

ADIRS Air Data Inertial Reference System
ADIRU Air Data Inertial Reference Unit

ADL Airborne Data Loader ADM Air Data Module

ADP Air Driven Pump
ADRS Address

ADS Air Data Systems
ADU Air Drive Unit

AEM Audio Entertainment Multiplexer

AFDC Air Flight Data Control

AFDS Autopilot Flight Director System

AFL Air Flow

AIDS Airborne Integrated Data System

AIMS Airplane Information Management System

AMU Audio Management Unit

ANCMT Announcement
ANCPT Anticipate
ANCPTR Anticipator

ANS Ambient Noise Sensor

ANTI-COLL Anti-Collision
AOA Angle of Attack
AOC Air/Oil Cooler

APB Auxiliary Power Breaker
APID Airplane Identification
APU Auxiliary Power Unit



ARINC Aeronautical Radio Incorporated
ASA Autoland Status Annunciator

ASCPC Air Supply Cabin Pressure Controller
ASCTS Air Supply Control and Test System

ASCTU Air Supply Control and Test Unit

ASP Audio Select Panel

AVM Airborne Vibration Monitor

BDY BLK Burndy Block

BFE Buyer Furnished Equipment
BPCU Bus Power Control Unit
BSCU Brake System Control Unit

BST Boost

BTB Bus Tie Breaker

BTLCS Brake Torque Limiting Control System
BTMU Brake Temperature Monitor Unit

C Cold

CACTS Cabin Air Conditioning & Temperature Control System

CADS Central Air Data System

CALIB Calibrator CAP Capture

CAP Contact Authorized Proposal
CAPC Cabin Area Control Panel

CAPT Captain

CCA Central Control Actuator
CCL Cargo Control Logic
CCM Cargo Control Module
CCU Cargo Control Unit
CDU Control Display Unit

CFDS Centralized Fault Detection System
CFE Customer Furnished Equipment

CHKPT Checkpoint

CHSP Course Heading Select Panel
CIC Cabin Interphone Controller

CIWS Central Instrument Warning System
CMC Central Maintenance Computer

CMD Command

CMM Component Maintenance Manual
CMS Cabin Management System
COC* Customer Originated Change

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COF MKR Coffee Maker
COLL Collision

COM/NAV Communication/Navigation

COR Corrector
CP Control Panel

CPCS Cabin Pressure Control System

CRKG Cranking

CSB Compressor Stability Bleed
CSMU Cabin System Management Unit

CT Control Transformer

CTC Cabin Temperature Controller
CTS Cabin Temperature Selector
CTS Conversational Terminal System

CVR Cockpit Voice Recorder **CWS** Control Wheel Steering DAA Digital/Analog Adapter DADC Digital Air Data Computer DAR Digital Aids Recorder **DED** Dead Ended Shield DEL Diagram Equipment List **DFCS** Digital Flight Control System **DFDAU** Digital Flight Data Acquisition Unit

DFDR Digital Flight Data Recorder

DH Decision Height
DIU Digital Interface Unit
DMU Data Management Unit
DP Differential Protection
DPA Digital Pre-Assembly

DPCT Differential Protective Current Transformer

DPLY Deploy

DSP Display Select Panel E/E Electrical/Electronics

EADI Electronic Attitude Director Indicator
ECS Environmental Control System
EDIU Engine Data Interface Unit

EDP Engine Driven Pump

EEC Electronic Engine Control (Unit)

EFIS Electronic Flight Instrument System

EHSI Electronic Horizontal Situation Indicator



EICAS Engine Indicating and Crew Alerting System

EIU EFIS/EICAS Interface Unit

ELCCR* Electrical Liaison Change Commitment Record

ELCU Electrical Load Control Unit

ELMS Electrical Load Management System

EMC Electromagnetic Compatibility

EMP Electric Motor Pump (See also ACMP)

ENTMT Entertainment **ENWY** Entryway

EPR Engine Pressure Ratio

EPRL Engine Pressure Ratio Limit

ESCC Electrical Supply and Control Center

ESNTL Essential **ESS** Essential

ETC Electronic Temperature Control ETOPS Extended Twin (Engine) Operations

EXCHR Exchanger **EXTD** Extend

F/D Flight Director Flight Engineer F/E F/F **Fuel Flow** F/O First Officer

FADEC Full Authority Digital Engine Control

FAFC Full Authority Fuel Control **FAR** Federal Aviation Regulations

FBW Fly-by-Wire

FCC Flight Control Computer

FCU Flap Control Unit

FDAU Flight Data Acquisition Unit

FLMTR Flowmeter

FMC Flight Management Computer

FMCS Flight Management Computer System

FMU Fuel Metering Unit **FMV** Fuel Metering Valve FOC Fuel/Oil Cooler

FQIS Fuel Quantity Indication System **FQPU** Fuel Quantity Processor Unit **FSEU** Flap/Slat Electronics Unit GCB Generator Circuit Breaker



GCR Generator Control Relay
GCU Generator Control Unit

GPWS Ground Proximity Warning System

GS Glide Slope

GSB Ground Service Bus

GSPR Gasper H Hot

HLCU High Lift Control Unit
HMU Hydromechanical Unit

HND Hand

HPC High Pressure Compressor (N2 Rotor)

HPSOV High Pressure Shutoff Valve
HPT High Pressure Turbine
HYDIM Hydraulic Interface Module

HYQUIM Hydraulic Quantity Interface Module

HZ Hertz (Cycles Per Second)

IBIT Initiated Built In Test

IBVSU Instrument Bus Voltage Sense Unit

IDG Integrated Drive Generator
IDS Integrated Display System
ILES Inboard Leading Edge Station
INS Inertial Navigation System

INTC Interconnect

IOEU Inboard Overhead Electronics Unit

IPC Illustrated Parts Catalog
IPL Illustrated Parts List
IRS Inertial Reference System

JPR Jumper KHZ Kilohertz

KVA Kilovolt Ampere

LGHTNG Lightning LMP Lamp LO Lock Out

LP Lightning Protector
LPT Low Pressure Turbine
LRRA Low Range Radio Altimeter
LRU Line Replaceable Unit

LSDA Low Speed Digital To Analog

M Mach



M MUX Main Multiplexer

MAI Multiplexer Action Item

MAWEA Modularized Avionics and Warning Electronics Assembly

MC* Master Change

MCDP Maintenance Control and Display Panel MCDU Multipurpose Control and Display Unit

MCP Mode Control Panel

MGSCU Main Gear Steering Control Unit
MHRS Magnetic Heading Reference System

MHZ Megahertz

MIDU Multipurpose Interactive Display Unit

MKR BCN Marker Beacon

MLS Microwave Landing System

MNFST Manifest

MOSFET Metallic Oxide Semiconductor Field Effect Transistor

MR* Modification Revision

MTCHG Matching
MTG Muting
NBR Number

ND Navigation Display

NGT Night

OAP Output Audio Processor

OFCR Officer
OFL Outflow

OMS Onboard Maintenance System

OOEU Outboard Overhead Electronics Unit
OPAS Overhead Panel ARINC 629 System
OPBC Overhead Panel Bus Controller

OVDR Overdoor
OVFL Overfill
OVHT Overheat
OVWG Overwing

PA Passenger Address

PA/Cl Passenger Address/Cabin Interphone

PCH Patch PCT Percent

PDU Power Drive Unit

PES Passenger Entertainment System

PFC Primary Flight Computer



PFD Primary Flight Display

PFIDS Passenger Flight Information Display System

PIS Passenger Information Sign

PKG Parking

PMA Permanent Magnet Alternator
PMG Permanent Magnet Generator
PMS Performance Management System

POR Point of Regulation

PRCLR Precooler PROT Protection

PRR* Production Revision Record

PRSOV Pressure Regulating Shut-Off Valve

PSA Power Supply Assembly

PSEU Proximity Switch Electronics Unit

PSU Passenger Service Unit
PTT Press To Talk/Push To Talk

PVD Paravisual Display

PYL Pylon

QAM Quadrature Amplitude Modulation Unit

QAR Quick Access Recorder

QDT Quadrantal

RAT Ram Air Turbine

RDMI Radio Distance Magnetic Indicator

RDP Roller Drive Power
RDU Remote Display Unit

REP Repellent
RFLNG Refueling
RGLTN Regulation

RMCP Radio Management Control Panel

RR* Rapid Revision

RST Reset
RSV Reserve

RTC Rudder Trim Control

RVSG Reversing

RVT Rotational Variable Transformer

SAARU Standby Attitude/Air Data Reference Unit

SAT Static Air Temperature
SATCOM Satellite Communications

SB* Service Bulletin



SCF System Cardfile

SCM Spoiler Control Module

SCU Seat Control Unit

SDI Source Destination Identifier

SEB Seat Electronics Box

SEB/ST Seat Electronics Box With Self Test

SEI Standby Engine Instruments

SEU Seat Electronics Unit

SHVR Shaver

SL* Service Letter

SN Sign SO Shut-off

SO Standard Option

SPL Splice List

SRM Stabilizer Trim/Rudder Ratio Module

SUP-NUM Supernumerary SVU Seat Video Unit

SWDL Software Data Loader

SWL Sidewall

T/M Torque Motor
T/R Thrust Reverser
TAI Thermal Anti-Ice

TAT Total Air Temperature
TBV Turbine Bypass Valve
TCA Turbine Cooling Air

TCAS Traffic Collision Avoidance System

TCC Turbine Case Cooling
TDL Time Delay Logic

TDX Torque Differential Transmitter

TERM BLK Terminal Block

TGT Turbine Gas Temperature

THSHD, THRSH Threshold

TL Tilt

TLA Thrust Lever Angle

TMC Thrust Management Computer
TMS Thrust Management System

TO Turn-off

TPIS Tire Pressure Indication System
TPMU Tire Pressure Monitor Unit



TR Torque Receiver
TR Transformer Rectifier
TRA Thrust Resolver Angle
TRC Thermatic Rotor Control
TRU Transformer Rectifier Unit

TS Terminal Strip
TTG Time To Go
TURB Turbulence

TX Torque Transmitter

UNLK Unlock

VBV Variable Bypass Valve VCC Video Control Center

VES Video Entertainment System
VGH Velocity, Gravity, Height
VIGV Variable Inlet Guide Vane

VLV Valve

VSI Vertical Speed Indicator VSV Variable Stator Vane

VTY Vanity

W/A Wrap Around WAI Wing Anti-Ice

WBA Wire Bundle Assembly
WEU Warning Electronic Unit
WF Fuel Flow (Weight of Fuel)

WF or wf Weight of Fuel

WHCU Window Heat Control Unit WIU Wire Integration Unit

WXR Weather Radar
XFD Crossfeed
XNT Transient

XPC External Power Contactor

XPNDR Transponder

ZMU Zone Management Unit

Where marked with an asterisk (*), see the GENERAL INFORMATION section, in the Wiring Diagram manual, for additional definition information.



1. EQUIPMENT LIST—GENERAL

Electrical and electronic equipment are shown on wiring diagrams and schematics with alphanumeric designators. These designators are used as cross-reference symbols to the Equipment List where the Part Numbers and Part Descriptions are shown. Splices, grounds, terminals and wire bundles are not included in the Equipment List.

A. Equipment List Data Fields

(a) EQUIP Field (Equipment Number)

The Equipment Number field may contain up to ten (10) alphanumeric characters. The Equipment Number always begins with a letter and may contain a space followed by another character.

NOTE: Equipment numbers 9000 through 9999 and 90000 through 99999 are reserved for customer use. Using these customer assigned equipment numbers facilitates identification of customer installed equipment. Customers should use only customer assigned equipment numbers, not Boeing assigned equipment numbers reported in the Equipment List.

The following list shows the categories assigned to the Basic Equipment Designators.

EQUIPMENT DESIGNATOR	TYPE OF EQUIPMENT
Α	Anti-Icing Controls
	Anti-Icing Equipment
	De-Icing Boots
	Ice Detector
	NESA Windows
	Pitot Heater
В	Black Box
С	Circuit Breakers
	Protective Equipment
D	Connectors
E	Equipment Racks
F	Fuel system components
G	Generator
GD	Grounds (Airframe)
Н	Overflow Categories, Miscellaneous Bundle Equipment
J	Junction boxes
L	Lamp Assemblies
	Lights, Lamps



EQUIPMENT DESIGNATOR	TYPE OF EQUIPMENT
continued	
М	Accessory Units
	Amplifiers
	Antenna
	Antenna Tuners
	Ballast Assemblies
	Batteries
	Bells
	Capacitors
	Chimes
	Computers
	Control Units
	Directional Gyros
	Filters
	Flux Valves
	Horns
	ILS Racks
	Lavatory Assemblies
	Motors
	Phase Adaptors
	Power Units
	Pumps
	Receivers
	Selcal
	Selcal Chimes
	Tape Recorders
	Transmitters
	Vertical Gyros
	Water Heaters
	Other Equipment
N	Indicators
P	Panels
R	Potentiometer
	Rectifiers
	Relays
	Resistors
	Rheostats
S	Switches



EQUIPMENT DESIGNATOR	TYPE OF EQUIPMENT
continued	
SM	Splices (Within a bundle)
SP	Splices (Between bundles)
Т	Temperature Bulbs
	Terminal Strips
	Transformer
	Transmitters
ТВ	Terminal Blocks
V	Valves
Υ	Line Replaceable Units

1) An Equipment Fixture Code is identified when the Equipment Number contains a space followed by another character. The Equipment Fixture Code is typically an alphabetic character.

TYPE OF FIXTURE	FIXTURE CODES
Equipment Mating Connector Fixture (Clamps, Adapters, Backshell, etc.)	T, U, V
Inline Connector Fixtures: ●Receptacle Fixture ●Plug Fixture	K, L, M Q, R, S
Receptacle Identification Fixture: •Decal •Adapter •Plug Button •Cover	W X Y Z
Coaxial Tee's, Connector Fixture	T, U, V
Components of Purchased Assemblies	A-Z
Lights: Bulbs Connectors Caps Ballasts which are non-optional parts (unique part numbers) but serve alternate functions.	T, U, V D C A, B
Customer Designated Fixtures	Y, Z

(b) OPT Field (Option)

The Option field indicates if optional part numbers may be used, and if so, the order of preference.



1) Boeing "as delivered" Options:

When the OPT field is empty or a zero appears, no options are permitted due to systems or physical restrictions. A "1" indicates that options are available.

OPT	Option Available
	No
0	No
1	First Option
2	Second Option
3	Third Option

2) Customer Requested Options:

Customer requested options are used for the incorporation of post-delivery changes, such as Boeing Service Bulletins and Customer Originated Changes. Option "9" is used for the incorporation of these changes. When more than one option is available, the four allowed options are:

OPT	Option Available
9	First Option
8	Second Option
7	Third Option
6	Fourth Option

(c) PART NUMBER Field

The Part Number field provides the part number for the equipment item. It will be a unique number defined by either vendor, Military or Boeing drawing and/or specification.

(d) PART DESCRIPTION Field

The Part Description is derived from the actual use of the item or the title of its associated drawing or specification.

(e) USED ON DWG Field (Used On Drawing)

The Drawing that contains installation details for that part.

(f) VENDOR Field (Vendor Code)

For Vendor Code translation, refer to:

H4-1: Federal Supply Code for Manufacturers-Name to Code H4-2: Federal Supply Code for Manufacturers-Code to Name



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SPECIAL VENDOR CODE ASSIGNMENT

See the Vendor Codes section.

(g) QTY Field (Quantity)

The Quantity shown reflects the number of times each item is used on the airplane.

(h) DIAGRAM Field

The Diagram listed is the diagram or schematic on which the item appears. However, one item may be shown on more than one diagram or schematic.

(i) STATION/WL/BL Field (Location)

Location information is shown directly below the Vendor Code. If the equipment is located on or near a panel, equipment rack, disconnect panel or terminal block, that item is referenced for locations. All other equipment locations are shown by station line, water line, and buttock line. See the Manual Usage section for a more detailed explanation.

(j) EFFECTIVITY Field

This is a Boeing or a Customer assigned number to keep track of airplanes. An effectivity is presented as a single airplane or in a range, e.g., PP001-PP099 or 001-999 or AAA-ZZZ, covering several airplanes. The word "ALL" means that the item is applicable for all aircraft listed in the Effective Aircraft section.

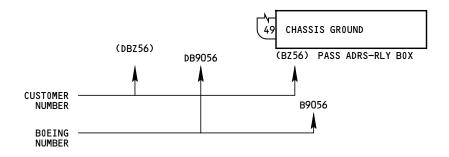
2. CUSTOMER ASSIGNED INCOMPATIBLE EQUIPMENT, WIRE AND BUNDLE NUMBERS

Customer assigned equipment item, wire and/or bundle numbers that are not compatible with the Boeing computer program will have an additional number assigned. This Boeing assigned number, will appear in the computer generated listings, i.e., Equipment, Wire, Ground, Splice, Terminal Strip and Hookup Lists.

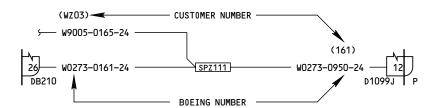
<u>NOTE:</u> On the wiring diagram, the customer assigned equipment item, wire and/or bundle numbers will be placed in parenthesis next to the Boeing assigned numbers.



CUSTOMER ASSIGNED EQUIPMENT ITEM NUMBER



CUSTOMER ASSIGNED WIRE AND BUNDLE NUMBER



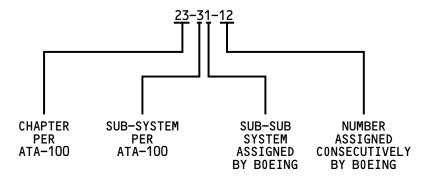


1. BASIC INFORMATION ABOUT WIRING DIAGRAMS

A. Wiring Diagram And Page Numbering

(a) Wiring Diagram Numbering

The Diagram numbering is in accordance with ATA Specification 2200 Revision 2001.1



The first three digits will be identical on diagrams and schematics.

NOTE: When a diagram is referenced to another, only the diagram number is used. Therefore, where there is more than one page of the same diagram, it is necessary to refer to the effectivity block to make certain the diagram applies to the airplane of interest.

(b) Diagram Page Numbering

Diagram page numbering begins at 1 then 2, 3 etc. Each page reflects different delivered configurations between aircraft. See the following example.

DIAGRAM	PAGE	EFFECTIVITY
21-31-12	1	001-004
21-31-12	2	005-999

The Page numbers (Page 101, 102, etc.) are used to represent different delivered configurations of a given schematic which may be applicable to different airplanes within the customer's fleet. When a schematic page number has a suffix (e.g., 101A, 102A for Customer Originated Changes or 101.1, 102.1, etc. for Service Bulletins) it reflects a post-delivery configuration for the same airplane(s). Both the configuration delivered by Boeing and the configuration after modification remain in the manual until the airline notifies Boeing that the post-delivery change has been incorporated in the customer's entire fleet of that model, and requests Boeing to delete the obsolete configurations.



(c) Diagram Sheet Numbering

If Diagrams of the same circuit can not be shown on one sheet, they are shown on additional sheets having the same title, diagram number and page number.

When reference is made to a multisheet diagram, the sheets will be included in the reference.

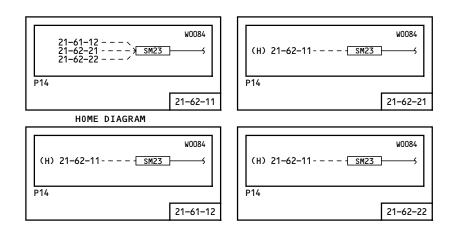
EXAMPLE:

34-11-11	34-11-11
SH 1	SH 3
34-11-11	34-11-11
SH 2	SH 4

B. Home Diagram

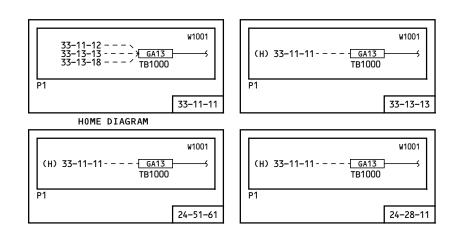
Any wire termination which is used for three or more wires shown on three or more wiring diagrams shall be assigned a "Home" diagram. The "Home" diagram shall indicate all connections on the termination and reference the diagram that shows the connection.

(a) Splices

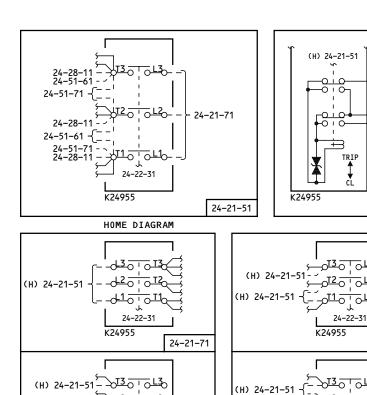




(b) Terminal Blocks



(c) Relays (The Mechanical Linkage is referenced to the diagram containing the control circuit)



(H) 24-21-51 - OTTO O LTO

24-22-31

24-51-61

K24955

24-22-31

24-51-71

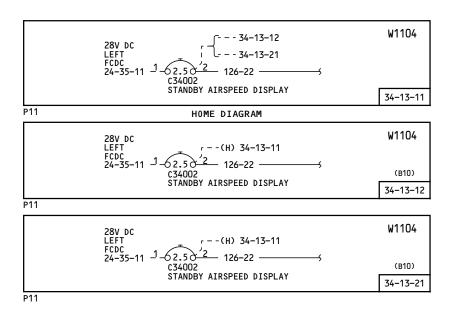
24-28-11

24-22-31

K24955



(d) Circuit Breaker

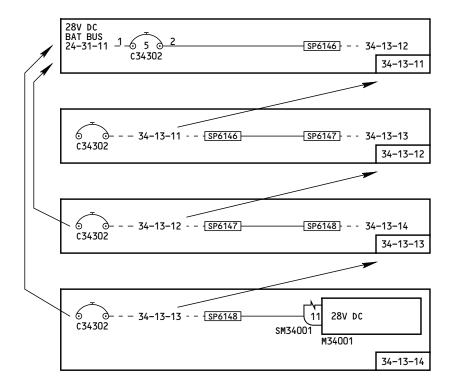




C. Power and Ground Indication

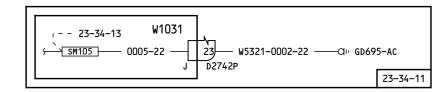
(a) Power Source Indication

The circuit breaker symbol, equipment designator, nomenclature and grid location, and diagram number of power source, are shown for circuits which are routed through connectors, splices and terminals.



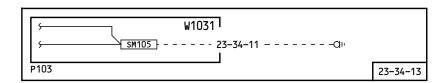
(b) Ground Indication

The diagram detailing the complete ground circuit may contain references to other diagrams.



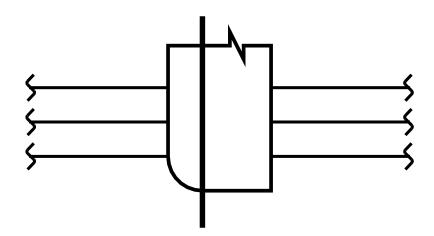


The referenced diagrams contain the ground symbol and a reference to the diagram which depicts the complete ground circuit.



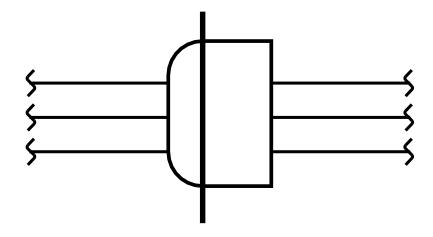
D. Connector Symbols

Connector symbols are shown broken when the same connector is shown elsewhere on that or another diagram.





Connector symbols are shown complete when all used contacts are shown on one diagram.

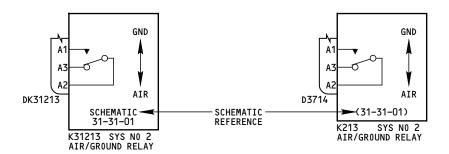


E. Galleys and Lavatories

Wiring Diagrams will show Boeing wiring to the interface with the galley and lavatory units.

F. Schematic References Shown on Wiring Diagrams

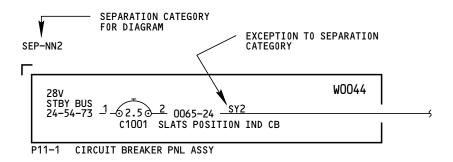
An ATA number on Wiring Diagrams shown with the word SCHEMATIC, SCHEM, or in parentheses within modules, is a Schematic reference. See the following example.





G. Wire Separation Identification

- (a) The wire separation category assigned to the majority of wires on a diagram is shown in the upper left hand corner of each diagram, e.g., SEP-NN2. Wires on a diagram not part of that category are individually labeled.
- (b) The actual code definitions may be found in the Codes section of the Introduction.
- (c) These wire separation codes are intended for production use. Chapter 91-21-13 Master Bundle List includes the wire separation codes. The lack of a wire separation code on a diagram does not indicate there is no wire separation required. Consult the Master Bundle List for the wire separation assigned to the bundle.





1. CHARTS

The Chapter 91 Charts contain airplane station arrangements, wire zones, major wire bundle pathways, panel and equipment shelf locations, circuit breaker panel charts, disconnect bracket charts and Master Bundle information.

2. LISTS

The Chapter 91 Lists are numbered as follows:

91-02-00	 Circuit Breaker List
91-04-00	 Bracket List
91-21-11	 Wire List
91-21-12	 Spare Wire List
91-21-13	 Master Bundle List
91-21-21	 Ground List
91-21-31	 Splice List
91-21-41	 Terminal Strip List
91-21-51	 Hookup List

The following paragraphs in this section define the contents of Chapter 91 Lists. The Wire List is the Primary source for Spare Wire through Hookup Lists.

A. Circuit Breaker List—Chapter 91-02-00

- (a) The Circuit Breaker List reflects all the circuit breakers within an airplane and is derived from data contained in the Equipment List. It lists, in alphanumeric order, each Panel/Access Door, the Description and the Diagram of that panel.
- (b) For each Panel/Access Door the grid location (Grid No), the circuit breaker number (Ckt Bkr), circuit breaker label (Description), Diagram and Effectivity are listed.
- (c) Unused grid locations are not listed.
- (d) The Circuit Breaker List is used as supplemental data for all Chapter 91-02-XX Panel Charts containing circuit breakers.

B. Bracket List—Chapter 91-04-00

- (a) The Bracket List reflects all the disconnect brackets within an airplane and is derived from data contained in the Equipment List. It lists, in alphanumeric order, each disconnect bracket (BRACKET NO.), title (DESCRIPTION), EFFECTIVITY, and where the information is available: maximum number of positions (MAX POS), and location (STATION/WL/BL).
- (b) Each POSITION within a bracket, that is being used, is listed, followed by the mounted receptacle number (RECEPTACLE) and its wire bundle number (BUNDLE), the mating plug (PLUG) and wire bundle number (BUNDLE) and the EFFECTIVITY.

- (c) Positions not containing connectors are not listed but can be determined by viewing the graphical representation of the bracket in the 91-04-XX Disconnect Bracket Charts. In general, brackets with numerically numbered positions (001, 002, 003...) are consecutively numbered and any omitted number is likely present on the bracket but unused. On brackets with alphanumeric grid positions (A01, A02, B04, C07...) are probably approximately rectangular with the alpha part representing one axis of the grid and the numeric part representing the other axis of the grid. In those cases, the positions are generally going to be numbered in a consistent rectangular grid method (A01, A02, ... A05, B01, B02, ... B05, ... F01, F02, ... F05). Any positions in this pattern that are not listed as used are likely on the bracket but unused. Some bracket positions are also numbered with strictly alphabetical values (A, B, C, F, G...). In all of these cases, it is necessary to reference the graphical representation of the bracket to be sure of the existence of a particular unused bracket position.
- (d) The Bracket List is used as supplemental data for all 91-04-XX Disconnect Bracket Charts.

C. Wire List—Chapter 91-21-11

The Wire List reflects all the wire bundles within an airplane. It lists, in alphanumeric order, each Bundle Number (Bundle No.), Part Number, Description and the wires within each bundle.

(a) BUNDLE NO. Field (Wire Bundle Number)

Each wire bundle is given an item number beginning with "W". This item number is the first part of the wire number and is derived from the wire bundle drawing number. The four digits XXXX of the item number is the wire bundle number. The first part of the wire number thus becomes WXXXX.

Wire bundle numbers W9001-W9999 are reserved for customer use.

(b) PART NUMBER and DESCRIPTION Fields

The wire bundle part number is derived from the wire bundle drawing number. A description of the bundle follows the part number.

(c) WIRE No./GA/CO Field (Wire Number/Gauge/Color)

A typical wire number consists of the wire identifier and gauge. A color designator may appear at the end of the wire identifier or gauge number. See Paragraph 3, for wire number details.

The wire numbers are physically printed or stamped on each wire and are also used on the Wiring Diagrams to identify the wires.

Wire numbers 901-999 and 9001-9999 are reserved for customer use.

(d) TY Field (Wire Type)

The Wire Type code is a two-character identifier for the type of wire used. These codes are described in 20-00-13 of the Standard Wiring Practices (Chapter 20).



(e) Fam Field (Wire Family)

Multi-conductor wires such as twisted or shielded wires are grouped as a family of wires. Each family of wires is given a code which is unique per bundle. Therefore, the family code is used to denote that some wires are physically related to each other either by being twisted together or by sharing the same shield or jacket. Family codes are assigned a code between A-ZZ.

(f) FT-IN Field (Wire Length)

1) The length of the wire is shown in this field.

Critical lengths and tolerances will be shown on diagrams for applicable wires or bundles.

Measure the wire length from the connector face or terminal ring centerline to the connector face or terminal ring centerline at the opposite wire end.

2) A wire without a length will appear if it is part of a family of wires. The length for this particular wire is usually specified on the lowest wire number in that family group.

(g) DIAGRAM Field (Diagram Reference)

1) The numbers in this field reflect the diagram on which the wire appears. The diagram references apply to any functional, ARINC Spare, or System Spare wire. These wires will not be shown on the wire diagram and are not spare wires.

MOTE: The diagram reference "99-99-99" is used for the wire bundle manufacturing process (e.g., Stub wires or wires to maintain pin circuit separation).

- 2) Spare wires may be found in the Chapter 91 Wire List, Ground List, Terminal Strip List, Splice List and Hookup List with "SPARE" in the DIAGRAM field. These spare wires may be used by the customer for Kit and Service Bulletin incorporations by assigning diagram numbers to those particular wires used.
- 3) Boeing identifies wires as spare that are no longer functional but may remain in a bundle. These wires will be identified in the Chapter 91 Lists as "SPARE*", and the wires will be deleted from the diagrams. These wires are available for customer use on an individual airplane basis. Boeing may choose to delete or re-use these wires.

(h) EQUIP From and To Field (Equipment Number)

There are two EQUIP fields, one for each end of the wire. Any item designator found in this field, other than splices and grounds, will be defined in the Equipment List.

(i) TERM From and To Field (Terminal Number)

There are two TERM fields, one for each end of the wire. The contact identifiers in these fields are typically alphanumeric characters.

- 1) An equal sign (=) appearing ahead of the identifier means no identification is to be found on the part and the identifiers used are for hookup and test purposes.
- "REF" is used to indicate a termination in the vicinity of, but not in, the splice or terminal indicated in the EQUIP field. It may be used with an SP number instead of SPREF in the EQUIP field.



- 3) "DED" indicates the shield is dead ended and not terminated by pigtail or jumper wire.
- 4) "CAP" indicates a wire terminated by an end cap near the equipment shown.
- 5) Ground terminations are shown as A., AC Ground; D., DC Ground; S., Static or Shield or Special Ground.
- 6) Color codes represent the pin identification codes of colored pigtails from vendor furnished equipment. See Paragraph 3.B(4) for TERM color codes. The wire number associated with the pigtails are in the form A-A thru Z-9.
- 7) Pin and socket lower case letter identifiers are indicated by an upper case letter followed by a minus sign (-), (e.g. F- = f).
- 8) The following are special shield terminations:
 - PER-S280W605 Backshell Zero Inch Termination
 - PERB-S280W603 Backshell Zero Inch Termination
 - PERG-S280W601 Multi Insert Ground Block 3 Inch Maximum
 - = CC-Strain Relief 2 Inch Maximum
 - G-Terminal Track Ground 2 Inch Maximum

(j) Type Field From and To (Terminal Type)

The Terminal Type codes appearing in this field are defined in the CODES section identifying:

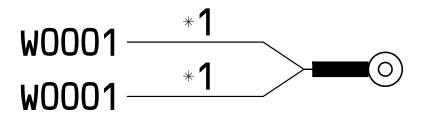
- 1) Codes for Lug (Stud) size (diagram symbol depicted).
- 2) Codes for Special Terminals.
- 3) Codes for Special Contacts.

(k) SPLICE Field From and To (Splice)

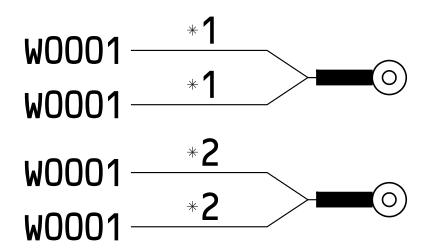
This field shows conditions under which the connection is made:



1) * In the splice field indicates two or more wires terminate in the same attaching device, i.e., lug, pin, or solder terminal (not used on SP or SM splices).



2) * 1, * 2, * 3 designate which wires appear in which single device, i.e., * 1 wires are lugged together; * 2 wires are lugged together in a second lug and * 3 wires are lugged together in the third lug.



3) Ferrule groups or shielded wires with shields tied together with jumpers are identified as a common terminus by a two letter F() code. All shields with an "FA" in either SP field are common.



Sometimes a jumper wire number FR-() is used to connect the FA shields and a termination such as a pin in a connector or to a ground. Jumper FRAA is required to terminate FA, jumper FRAB is required for FB and so on.

4) Shields terminating in a splice will show "A", "B", and so on, in the SP field. The wire from the splice to another termination will be numbered "JPA", "JPB", and so on.

(I) EFFECTIVITY Field

This is a Boeing or a Customer assigned number to keep track of airplanes. An effectivity is presented as a single airplane or in a range, e.g., PP001-PP099 or 001-999 or AAA-ZZZ, covering several airplanes. The word "ALL" means that the item is applicable for all aircraft listed in the Effective Aircraft section.

D. Spare Wire List—Chapter 91-21-12

- (a) The Spare Wire List reflects all spare wires within the airplane. Spare Wires are reported in order by From Matewith Equipment, Position Number on that Equipment, the Wire Bundle Number, then Wire Bundle Separation Code, From Terminating Equipment Number, and Wire Number.
- (b) Additional information provided in the Spare Wire List is Wire Bundle Description, Terminal Number on the From Equipment, Terminal Type on the From Equipment, the Wire Gauge, Wire Type, To Equipment Number, Terminal Number on the To Equipment, Terminal Type on the To Equipment, the To Matewith Equipment, the Position on the To Matewith Equipment, and Effectivity.

E. Master Bundle List—Chapter 91-21-13

- (a) The Master Bundle List reflects all wire bundles within the airplane. Master Bundle List is reported in Bundle number order with the corresponding Separation Code and Description.
- (b) (h) The Equipment field contains the Equipment Item Number of the ends of that Wire Bundle. The Matewith field contains the Equipment Item Number where the end equipment mates with. The LOCATION may contain position, Panel numbers or Station/Water/Buttock line information. The MW Connector field contains the Equipment number of the mating connector. The MW Bundle field contains the Bundle number for the MW Connector.

F. Ground List—Chapter 91-21-21

- (a) The Ground List reflects two types of airframe grounds used within an airplane. Ground Blocks (GB) are used only in pressurized areas, while Ground Studs (GD) are used in both pressurized and non-pressurized areas. Grounds on shelves and panels (GDM, GDX, GDY, GDZ, GBX, GBY, and GBZ) are not listed as they are unique within a specific shelf or panel.
- (b) Ground List reflects all grounds within the airplane. The Ground List is reported in alphanumeric order, with the corresponding Part Number and Location.
- (c) The detail rows below the Ground number contain Termination number, Termination type, Wire bundle number, Wire number within that bundle, wire gauge and color of that wire, diagram reference and effectivity.



G. Splice List—Chapter 91-21-31

- (a) The Splice List reflects all splices (SP) unique within an airplane. Splices (SP) are used when connecting wires from other wire bundles and vendor wires within the same wire bundle. Smooths (SM) are not listed as they are unique only to a wire bundle.
- (b) Splice List reflects all splices within an airplane. The Splice List is reported in alphanumeric order, with the Location (Station/WL/BL).
- (c) The detail rows below the Splice number contain Wire bundle number, Wire number within that bundle, wire gauge, color of that wire, type of wire and the diagram reference, and effectivity.

H. Terminal Strip List—Chapter 91-21-41

- (a) The Terminal Strip List reflects all the terminal strips within an airplane. The Terminal Strip List is reported in alphanumeric order with their part number and location (STATION/WL/BL).
- (b) The detail rows below the Terminal Strip number contain the terminal block fixture identifier, each terminal on a strip, the terminal type, the wire bundle number, the wire number, the gauge and color, the diagram depicting each wire and the effectivity for each wire.
- (c) The fixture (FIX) field reflects the type of terminal block installed on a terminal strip. The various types of terminal blocks are shown in the Symbols section.
- (d) The term G reflects the wire terminating at the integrated grounding module part of the terminal track.

I. Hookup List—Chapter 91-21-51

(a) The Hookup List reflects all wire terminating devises except grounds, splices, terminal strips and single-phase circuit breakers within an airplane. The Hookup List reports the Equipment in alphanumeric order with their location (STATION/WL/BL) and Description.

Single-phase (one circuit) circuit breakers are not listed for two primary reasons:

- Each unit is shown complete on the affected diagram. Hookup List data would be redundant.
- The benefits to be derived from listing the units would not justify the increased size of the manual.
- (b) The detail rows below the Equipment contain the terminals on these devises and the terminal type. The wire bundles, their wire numbers, gauge and color terminating at each terminal is listed along with the diagram on which the terminal is depicted and its effectivity.



3. EXPLANATION OF WIRE NUMBERING AND COLOR CODES FOR THE CHAPTER 91 LISTS (91-21-11 THRU 91-21-51)

A. Wire Numbering

Wire serial numbers are wire bundle unique and are generally chosen from the following categories:

NOTE: Wire numbers reserved for Airline use are 901-999 and 9001-9999, based on the existing wire number patterns in a given Wire bundle. For example: Wire numbers with three numerics will use 901-999 and Wire numbers with four numerics will use 9001-9999.

CATEGORY	WIRE NUMBER
Single (Unshielded)	001-199
Single Shielded Wire	001-199
Shield	SHA-SPZ
Twisted Pair Unshielded	201-299
Twisted Pair Shielded	201-299
Shield	SHA-SPZ
Twisted Triplet Unshielded	301-399
Twisted Triplet Shielded	301-399
Shield	SHA-SPZ
Twisted Quads Unshielded	401-450
Twisted Quads Shielded	401-450
Shield	SHA-SPZ
Special Wires-High Temperature, Thermocouple, etc.	451-500
Overflow of above (does not apply to sheild)	501-899 8000-8999
Reserved for Airlines	900-999 9000-9999
Coaxial Cable	451-500
Shield	SHA-SPZ
Twinax Cable	451-500
Shield	SHA-SPZ
Triaxial Cable	451-500
Inner Shield	SHI
Outer Shield	SHO
Bundles with more than 900 wires ("A" and "W" may be used in any combination with numerics-no other alphas can be used)	01A-A99 01W-W99
Vendor Furnished Wires	A-A thru Z-Z
Reserved for Airline	9-1 thru 9-99
Sheild Pigtail Ferrule group to a pin, ground, splice, etc.	FRA-FR9 GRA-GR9



CATEGORY	WIRE NUMBER
continued	
Sleeving: Engineering Mockup	S01-S99 P01-P99
Braid Over Bundle	YAA-YZZ
Polarizing plug in relay sockets	PA1-PY9
Reserved for Airline	PZI-PZZ
Bus Bars: For Terminal Strips Insulated for Circuit Breakers	ZZA-ZZ9 YAA-YZZ

B. Wire Color Codes

The following is a list of single character color codes used on the wire.

WIRE COLOR	CODE
Gray	A
Blue	В
Green	G
Black	K
Purple	L
Brown	N
Orange	0
Pink	Р
Red	R
Tan	T
Uninsulated	U
Violet	V
White	W
Yellow	Υ

Wires may be identified by color instead of wire numbers.

WIRE COLOR	WIRE NUMBER
Black/Blue	KB
Black/Brown	KN
Black/Green	KG
Black/Gray	KA
Black/Orange	KO
Black/Red	KR
Black/Violet	KV



WIRE COLOR	WIRE NUMBER
continued	
Black/Yellow	KY
Blue/Black	BK
Blue/Brown	BN
Blue/Green	BG
Blue/Yellow	BY
Blue/Orange	ВО
Blue/Pink	BP
Blue/Purple	BL
Blue/Green/Black	AC
Brown/Orange	NO
Orange/Black	OK
Orange/Brown	ON
Orange/Yellow	OY
Red/Black	RK
Red/Blue	RB
Red/Brown	RN
Red/Green	RG
Red/Orange	RO
Red/Pink	RP
Red/Purple	RL
Red/Violet	RV
Red/Yellow	RY
Red/Yellow/Green	AB
White/Black	WK
White/Blue	WB
White/Brown	WN
White/Gray	WA
White/Green	WG
White/Orange	WO
White/Pink	WP
White/Purple	WL
White/Red	WR
White/Violet	WV
White/Yellow	WY
White/Black/Red	AG
White/Black/Violet	AL
White/Black/Yellow	AM

CHARTS AND LISTS



WIRE COLOR	WIRE NUMBER
continued	
White/Blue/Black	AK
White/Blue/Yellow	AA
White/Brown/Red	AH
White/Green/Red	AJ
White/Red/Blue	AD
White/Yellow/Red	AE
White/Purple/Red	AF
Yellow/Black	YK
Yellow/Green	YG
Yellow/Orange	YO
Yellow/Purple	YL

PSU Feeder Bundle Coloring:

In most airplanes, the wires in the PSU Feeder Bundles are identified by wire.

C. PSU Feeder Bundle Coloring

PSU Feeder Bundle Coloring	
COLOR	
Black	
Brown	
Green	
Orange	
Purple	
White	
White/Blue	
White/Green	
White/Orange	
White/Red	
White/Violet	
White/Yellow	



1. WIRE TYPE CODE

This information is covered in Chapter 20, Standard Wiring Practices, Section 20-00-13.

2. VENDOR CODE

For Vendor Code translation, refer to:

H4-1: Federal Supply Code for Manufacturers-Name to Code H4-2: Federal Supply Code for Manufacturers-Code to Name

H4-3: Nato Supply Code for Manufacturers-Name to Code/Code to Name

Published by:

Defense Supply Agency Defense Logistics Services Center Federal Center Building Battle Creek, Michigan 49016

VENDOR CODE	SPECIAL VENDOR CODE ASSIGNMENT
V96906	Parts having Military Part Numbers
VAAL	American Airlines Inc. Tulsa, Oklahoma
VAB	Coastal Mfg. Co. Santa Monica, California
VAC	Safety Industries Inc. Glen Ellyn, Illinois
VAD	Glarban Corp. Gordonville, N.Y.
VAE	Ucinite Co., The Los Angeles, California
VAF	Air France 1 Square Max Hymans 75, Paris 15, France
VAI	Industrial Products Co. Gardena, California
VAJ	Bozak Sales Co. Salisbury, Connecticut
VAO	Teddington Controls Ltd. Tydfil, South Wales
VARINC	Arinc
VARMED	Airmed Ltd. Edinburgh Way Harlow, Essex, England
VAZ	Murphy Radio Ltd. Welwyn Garden City Hertfordshire, England



WIRING DIAGRAM MANUAL

INTRODUCTION

continued...

VBB United Data Control Inc.

Pasadena, California

VBC Holmberg and Co.

Ohlaur Strausse 5-11 Berlin, S036

VBD John E. Lindberg Co.

Berkely, California

VBF The Firewall Co.

Subsidiary of Aro Corpl

Los Angeles, California

VBFE Buyer Furnished Equipment
VBG Societe Française D'Equipments

(SFENA) Siege

Social 25A 20 Rue Du Point Nevilly, Seine, France

VBH Ackerman, Albert, Firma

Akerman Albert,

Gummersback/Rhld., Germany

VBJ Smiths Aviation Division

Cricklewood, London, England

VBJ Standard Telephones and Cables Ltd.

Connaught House, 63 Aldwich West Central 2, London, England

VBM Associated Industries

Seattle, Washington

VBO H.K. Wilson Co.

Bellevue, Washington

VBP Brook Part Laboratories, Inc.

Cleveland, Ohio

VBR Amplivox Ltd., Industrial Div.

Beresford, Av. Wembley,

Middlesex, England

VBRZVA Bronzavia-S.A.

207 Blvd. Saint-Denis 92 Courbevoie, France

VCELRD Cosser Electronics Limited

Radar Division

The Pinnalces, Elizabeth Way Harlow, Essex, England

VDELTA Delta Air Lines, Incorporated

Harsfield-Atlanta International Airport Atlanta, Georgia 30320

VEIA Electronic Industries Association



VENDOR CODE	SPECIAL VENDOR CODE ASSIGNMENT
continued	
VELNO	Elno 18 Rue Du Val Notre Dame 95 Argenteuil, France
VEPS	Electric Power Storage, Ltd. P.O. Box 5 Clifton Junction Swenton, Manchester England, M272LR
VFO241	Filotex 140 Rue Eugene-Delacroix 97210 Draveil France
VGRVNR	Graviner Inc. 1121 Bristol Rd Mountainside, N.J. 07092
VIMP	Inflight Motion Pictures, Inc. 23-06 31st Ave. Long Island City, New York 11106
VJAEGR	Jaeger 2 Rue Baudin Levallois-Perret 92 France

3. TERMINAL INFORMATION

The following index of "Term Type Codes" lists the code, a description of the code and, as applicable, the terminal stud size and/or part number. The code index is arranged in the following order:

- Single alphabetical letter
- Two character code with leading alphabetical letters
- Numbers
- Symbols

See Standard Wiring Practices (Chapter 20) for maintenance or repair information.

A. Single alphabetical letter

	TYPE		
_	CODE	DESCRIPTION OF THE CODE	PART NUMBER
	Α	General Purpose Lug, Standard/Narrow, #2 Stud	BACT12AC43
	В	General Purpose Lug, Standard, #4 Stud	BACT12AR() or 2-323914-2 (24 Gage)
	С	General Purpose Lug, #6 Stud	BACT12AR() or BACT12AC()
	D	General Purpose Lug, #8 Stud	BACT12AR() or BACT12AC()
	Ε	General Purpose Lug, #10 Stud	BACT12AR() or BACT12AC()
	F	General Purpose Lug, 1/4 Stud	BACT12AR() or BACT12AC()



TERM TYPE		
CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
G	General Purpose Lug, 5/16 Stud	BACT12AR() or BACT12AC()
Н	General Purpose Lug, 3/8 Stud	BACT12AR() or BACT12AC()
1	General Purpose Lug, Narrow #6 Stud	BACT12AR()
J	High Temperature Lug, #2 Stud	BACT12M()
K	High Temperature Lug, #4 Stud	BACT12M ()
L	High Temperature Lug, #6 Stud	BACT12M ()
M	High Temperature Lug, #8 Stud	BACT12M ()
N	High Temperature Lug, #10 Stud	BACT12M ()
0	General Purpose Lug, 1/2 Stud	BACT12AC()
Р	High Temperature Lug, 1/4 Stud	BACT12M ()
Q	High Temperature Lug, 5/16 Stud	BACT12M ()
R	High Temperature Lug, 3/8 Stud	BACT12M ()
S	Splice, In-line;Install See Chapter 20	
V	Splice, Shielded;Install See Chapter 20	
W	Install Moisture Seal Splice (Closed End) See Chapter 20	
Υ	End Cap & Stow After Test	
Z	Faston, 1/4 Stud	2-520184-4

B. Two character code with leading alphabetical letters

TERM TYPE		
CODE	DESCRIPTION OF THE CODE	PART NUMBER
AB	White;Faston, 1/4 Stud	42640-2 & 1-480416-0
AC	Blue;Faston 1/4 Stud	42640-2 & 1-480416-3
AD	Red;Faston, 1/4 Stud	42640-2 & 1-480416-4
AG	Thermocouple Lug (Chromel) #6 Stud	AN5548-1
AH	Thermocouple Lug (Alumel) #6 Stud	AN5548-2
AW	General Purpose Lug, Thick Tounge, #10 Stud	150247
AY	60 Degree Lug, 3/8 Stud	324103
AZ	Miscellaneous Lug	YAES18-L85
BC	Splice, End Cap	328308
BD	High Temperature Splice (Parallel)	2-34318-1
BH	Terminal Block Contact	M39029/1-()-()
BJ	Terminal Block Contact for BACM153**	M39029/11-145
BK	Contact (Manufacturer)	60-1541-1
	AB AC AD AG AH AW AY AZ BC BD BH BJ	AB White;Faston, 1/4 Stud AC Blue;Faston 1/4 Stud AD Red;Faston, 1/4 Stud AG Thermocouple Lug (Chromel) #6 Stud AH Thermocouple Lug (Alumel) #6 Stud AW General Purpose Lug, Thick Tounge, #10 Stud AY 60 Degree Lug, 3/8 Stud AZ Miscellaneous Lug BC Splice, End Cap BD High Temperature Splice (Parallel) BH Terminal Block Contact BJ Terminal Block Contact for BACM153**



TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
BM	Copalum Lug, 1/4 Stud	277150-1
BN	Faston, .062 Stud	60900-1
BP	Parallel Splice, Install see Chapter 20	NAS1387-3
BS	Copalum Lug, Narrow, #10 Stud	277147-5
BT	Copalum Lug, 3/8 Stud	277147-2
BU	Copalum Lug, 1/4 Stud	277152-1 or 1-52521-2
BV	Faston, .110 Stud	3-520370-2 OR 640927-1
BX	Terminal, Faston	55319-1
BY	White;Faston, 1/4 Stud	61873-1 & 1-480416-0
CA	Coax Contact	48-1226-02
CK	Flag Lug, 1/4 Stud	YBM25-L1
CS	Stow with Protective Sleeving, See Chapter 20	
СТ	Install, Prepare End, See Chapter 20, and Stow	M39029/1-102
CX	Faston, .205 Stud	640909-1
D2		YAV14H1 NO.8 STUD
DG	Terminal Block Contact	M39029/22-191
DN	Coax Contact	BACC47EU4
E*	Flag Lug, #10 Stud	BACT12G-81
E/	Copalum Lug, Narrow, #10 Stud	277154-1
E2	General Purpose Lug, #10 Stud	324111
E3		YAV14H NO.10 STUD
E6	Coax Contact	48-1227-02
E8	Copalum Lug, #10 Stud	277147-1
FA	Ferrule for Fiber Optics	454819-162
FB	Ferrule for Fiber Optics	454900-162
FD	Ferrule for Fiber Optics	454819-145
F4	Copalum Lug, 1/4 Stud	277149-2
F6	Copalum Lug, Narrow, 1/4 Stud	277148-7 or 277154-2
GD	Copalum Splice (08-10 AL/CU, 08-08 AL/AL)	277156-1
GJ	Copalum Splice (*2-*1 AL/CU, *2-*2 AL/AL)	277161-1
GS	Special Contact	1841-1-5620
GU	Copalum Splice (04-08 AL/CU,)	277164-1
GW	Nickle Lug, #10 Stud	321894, 322338 or 323750
GX	Install Copalum In-Line Splice	



TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
GY	Nickle Lug, 1/4 Stud	322320, 322341 or 323751
G3	Nicle Lug, #8 Stud	321893, 322337 or 323749
G5	General Purpose Lug, 5/16" Stud	324112
G7	Nickle Lug, #6 Stud	321892 or322332
HA	Flag Lug, 3/8 Stud	BACT12G-24
HC	Thermocouple Contact (Socket, Chromel)	MS39029/86-512
HD	Spectial Contact (Pin)	030-8400-500
HE	Special Contact	318-1616-253
HF	Thermocouple Contact (Socket, Alumel)	MS39029/86-511
HG	Crimp Sleeve	BACA14AB-164
HH	Coax Contact	349-0005-000
HL	Special Contact (Socket)	031-8014-800
H*	Copalum Lug, Long, 3/8 Stud	277152-4
H1	Copalum Lug, Long, 3/8 Stud	277151-2
H2	Copalum Lug, 3/8 Stud	277150-3
H3	Copalum Lug, 3/8 Stud	277153-1
H4	Copalum Lug, 3/8 Stud	277149-4
H6	Copalum Lug, 3/8 Stud	277148-4
H7	Flag Lug, 3/8 Stud	BACT12G-44
H8	Flag Lug, 3/8 Stud	BACT12G-102
IA	Dual Stud-hole, 3/8 Stud	YAV4C-2L38-NK
IX	Dual Stud-hole, Copalum H/T, 1/4 Stud	55836-1
JB	Install & Stow, See Chapter 20	BACC47CN1A (PIN)
J1	Coax Contact	BACC47EU3
KA	General Purpose Lug, Short Tounge, #8 Stud	BACT12AL-15 OR 331456
LD	Miscellaneous Lug, Hook Tounge, #8 Stud	32456
LF	Miscellaneous Lug, Hook Tounge, #8 Stud	320381
LR	90 Degree Lug, #10 Stud	BACT12E()
LV	45 Degree Lug, 3/8 Stud	277069-1
LW	Removable Splice	BACT12A()
RW	Moisture seal Splice, Closed-end	D436-60
TA	Thermocouple Contact (Socket, Alumel)	031-1041-009



TERM TYPE		
CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
TC	Thermocouple Contact	ZZL-4020-10R (PIN, ALUMEL) OR ZZL-4120-10R (SKT, ALUMEL)
TD	Thermocouple Contact	ZZL-4020-10P (PIN,CHROMEL) OR ZZL-4120-10P (SKT, CHROMEL)
TE	Thermocouple Contact (Socket, Const)	ZZL-4112-10N
TF	Thermocouple Contact (Pin, Const.)	ZZL-4012-10N
TG	Thermocouple Contact (Socket,Chromel)	ZZL-4112-10P
TH	Thermocouple Contact (Socket, Alumel)	ZZL-4112-10R
TJ	Thermocouple Contact (Pin, Chromel)	ZZL-4012-10P
TK	Thermocouple Contact (Pin, Alumel)	ZZL-4012-10R
TL	Thermocouple Contact (Pin, Alumel)	030-1879-009
TM	Thermocouple Contact (Pin, Chromel)	030-1879-010
TU	Thermocouple Contact (Socket, Alumel)	031-1113-009
TV	Thermocouple Contact (Socket,Chromel)	031-1041-010
TW	Thermocouple Contact (Socket,Chromel)	031-1113-010
VA	Coax Contact	48-1227-54
VB	Coax Contact	48-1226-54
VC	Special Contact	66143-2LP
VG	Removable Splice (Blue) with BACC47CN2 Socket	48-7190 OR AIS16P
VJ	Special Contact	BACC47CP1T
VL	Moisture Proof Shielded Splice	
VM	Install Contact, Prepare End and Stow, See Chapter 20	BACC47EF1
VN	Shielded Contact	10-60479-44
VP	Install Contact and Stow, See Chapter 20	
VR	Removable Splice (Red) with BACC47CN2 Pin	48-7190-1 OR AIS16P-1
VS	Removable Splice (Red) With BACC47CP2T Socket	48-7191-1 or AIS16R-1
VT	Removable Splice (Blue) with 48-100-5021P Oversize Pin	48-7190 OR AIS16P
VU	Removable Splice (Blue) with 248-136- 1614S-02 Oversize Socket	48-7191 OR AIS16R



TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
VV	Removable Splice (Red) with 48-10-5021-P Oversize Pin	48-7191 OR AIS16P-1
VW	Removable Splice (Red) with 248-136- 1614S-02 Oversize Socket	48-7191-1 OR AIS16R-1
VX	Removable Splice	TSE-20-01 & 1841-1-5620 (PIN)
VY	Removable Splice (Blue) with BACC47CP2T Socket	48-7191 OR AIS16R
V1	High Temperature Splice (Moisture Seal and Shielded), Install See Chapter 20	D-150-0251
WI	Splice, Closed-end; Install See Chapter 20	
XQ	Miscellaneous Lug, Split Spade Toungue, #6 Stud	52420
XR	Amp, Spring Spade, #6 Stud	52409
ZA	Removable Splice (Red) and BACC47CN2(Pin). Stow with Plastic Cap, See Chapter 20	48-7190-1
ZB	Removable Splice (Red) and BACC47CO2T (Socket). Stow with Plastic Cap, See Chapter 20	48-7191-1
Z1	Faston, .110 Stud	2-520081-2
Z2	Faston, .058 Stud	60789-2
Numbers		
TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
1	Terminal Block Contact, blue/blue/blue stripe	S280W555-916
2	Terminal Block Contact	BACC47DE()

IERIVI I TPE		
CODE	DESCRIPTION OF THE CODE	PART NUMBER
1	Terminal Block Contact, blue/blue/blue stripe	S280W555-916
2	Terminal Block Contact	BACC47DE()
4	90 degree lug, 1/4 stud	BACT12E()
5	90 degree lug, 3/8 stud	BACT12E()
8	Install Hi-temp splice (Moisture seal) see chapter 20	
9	Install Moisture seal splice, see chapter 20	
10	Oversize Contact #12	48-100-5020P-02 or P204540 (pin), 248- 136-1210S-02 or P204541 (socket)
14	Oversize Contact #16	48-100-5021P-02 (pin), 248-136-1614S-02 or P208575-S (socket)

C.



TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
16	Oversize Contact #20	48-100-5007P-02 or 48-100-5012P-02 (pin), 248-136-2016S-02 or 318-2016-035 (socket)
18	Oversize Contact #20	P209553 (pin), 248-136-2018S-02 or P209541 (socket)
1A	Terminal Block Contact, Red/Red/Red	S280W555-920
1C	Terminal Block Contact	M39029/1-103
1D	Terminal Block Contact, Red/White/Red	S280W555-918
1F	Terminal Block Contact	M39029/58-360
1M	Socket	M39029/57-358
1N	Socket	M39029/57-359
20	Oversize Contact #22 (socket)	100-2020S
22	Oversize Contact #22M (socket)	MS27491-22D
4A	Thermocouple Contact (Pin, Alumel)	5000-070-116
4B	Thermocouple Contact (Socket, Alumel)	M39029/10-521
4C	Thermocouple Contact (Pin, Chromel)	5000-070-216
4D	Thermocouple Contact (Socket, Chromel)	M39029/10-522
4G	Thermocouple Contact (Pin, Alumel)	ZZL-4016-10R
4H	Thermocouple Contact (Socket, Alumel)	ZZL-4116-10R
4K	Thermocouple Contact (Pin, Chromel)	ZZL-4016-10P
4L	Thermocouple Contact (Socket, Chromel)	ZZL-4116-10P
4U	Thermocouple Contact Lug (Chromel) #8 stud	1-321897-0
4V	Thermocouple Contact Lug (Alumel) #10 stud	1-321898-0
5E	Coax Contact	BACC47EN1
5M	Coax Contact, D-sub	CQMEM-200()
5N	Coax Socket, D-sub	CQMEF-200
5P	Coax Contact, Right Angle	CMX010-P502
5Q	Coax Contact, Right Angle	CMX010-S502
62	Thermocouple Contact (Pin, Alumel)	030-1975-009
63	Thermocouple Contact (Pin, Chromel)	030-1975-010
69	Customer Installed Contact	
70	Flag Lug, 1/4 stud	BACT12G-82
7T	Thermocouple Contact (Pin, Alumel)	030-1878-007



TERM TYPE CODE	DESCRIPTION OF THE CODE	PART NUMBER
continued		
7U	Thermocouple Contact (Pin, Alumel)	030-1878-006
98	Oversize Contact #22	66169-2

D. Symbols

TERM TYPE		
CODE	DESCRIPTION OF THE CODE	PART NUMBER
#	Install & Stow, see chapter 20	BACC47DE
\$	Solder Connection: all size lugs	
*	indicated location	
%	Install & Stow, see chapter 20	M39029/11-145
%A	Install & Stow, see chapter 20	M39029/1-101
*	Install & Stow, see chapter 20 (#6 stud)	BACT12AR () or BACT12AC()
+	Install & Stow, see chapter 20 (#8 stud)	BACT12AR() or BACT12AC()
-	Install & Stow, see chapter 20 (#4 stud)	BACT12AR()
=	Install & Stow, see chapter 20 (#10 stud)	BACT12AR() or BACT12AC()

4. WIRE SEPARATION CATEGORY CODES

The Wire Separation Category is a three character code. Each character position is defined as follows:

- Power Separation
- Redundancy Separation
- EMC Separation

Power Separation Values	Redundancy Separation Values	EMC Separation Values
N-Neutral Circuits	N-No redundancy required	1–Source of Interference Circuits & Equipment
L-Left Power Circuits-Left Engine	A-1st, left or system	2-Passive Circuits & Equipment
R–Right Power Circuits-Right Engine	B-2nd (or Right) Redundant System Circuit	3–Sensitive (Susceptible) Circuits & Equipment
C-Center Power Circuits	C-3rd (or Center) Redundant System Circuit	
A–APU Control and Electric Power Circuits	D-4th Redundant System Circuit	
H–Hydraulic Motor Driven Generator		



Redundancy Separation

Power Separation Values

Values

EMC Separation Values

continued...

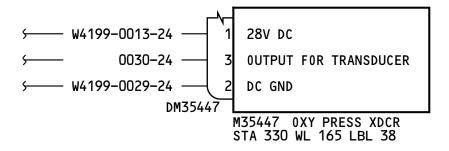
S-Standby Power Circuits. Circuits powered by Battery, Hot Battery, and AC Standby Bus



1. METHODS USED TO FIND INFORMATION

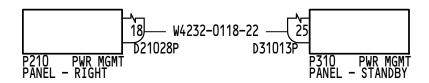
A. How To Locate A Diagram From A Wire Found In The Airplane

- (a) As an example, take wire number W4199-0013-24.
- (b) Knowing the wire bundle number W4199, refer to the Wire List in Chapter 91
- (c) Using Self Indexing wire list shown, locate wire bundle W4199.
- (d) Locate wire number 0013-24 and on the same line under the "Diagram" heading locate the diagram number 35-11-11.
- (e) Refer to Chapter 35 and locate the information needed on diagram 35-11-11.



B. How To Locate, In The Airplane, A Wire Found On A Wiring Diagram

- (a) Wire number W4232-0118-22 appears on Wiring Diagram 28-21-11.
- (b) Both ends of the wire are identified, disconnect D21028P at P210 panel and disconnect D31013P at P310 panel.



(c) Another method is to locate the bundle and wire number in the Wire List. The title of the bundle usually provides Station or Area for Airframe Bundles.



C. How To Locate Spare Wires Within A Wire Bundle

- (a) As an example, use wire bundle W487.
- (b) Knowing the wire bundle, refer to the Wire List in Chapter 91.
- (c) Using Self Indexing Wire List shown, locate wire bundle W487. In the "DIAGRAM' column you will find the wires that are spare wires are labeled "SPARE".
- (d) On the same line as the wire number, both end terminations will be indicated for all spare wires. (Check applicable airplane effectivity under the effectivity column.) Ground, Terminal Strip, Splice, and Hook-Up Lists will also show spare wires.
- (e) Spare wires are not shown on the wiring diagrams, only Airline Reserved Wires are shown, which are reserved for customer modifications.
- (f) Wires that become spare through an engineering change are identified by "SPARE" and are not part of the Customer spares equipment. Their use should be on an individual airplane basis.

D. How To Locate Unused Pins In A Connector

- (a) As an example use connector number DM31202A.
- (b) Knowing the connector number, refer to the Hook-Up List located Chapter 91
- (c) Using Self Indexing Hook-UP List shown, locate connector DM31202A.
- (d) Pin numbers 4-9 in the example are unused pins.

E. How To Determine The Number Of Pins In A Connector

- (a) To determine the number of pins in connector D01339J refer to the Equipment List.
- (b) Using Self Indexing Equipment List shown, locate connector D01339J.
- (c) To the right of D01339J the Part Number BACC45FN18-8P is shown. The number preceding "P" is 8 indicating there are a total of 8 pins in the connector.
- (d) For example, the graphic in the preceding section 1.D, connector DM31202A has 12 pins 6 used and 6 unused. For Non BACC45 type connectors, the number of pins in a connector can be determined by referring to:

Vendor Catalog

The number of pins in a BACC45 and a Non BACC45 type connector can also be determined by referring to the Hook-Up List.

F. Electrical Wiring Assembly and Installation Process How To Identify Lower Case Lettered Pins

Lower case lettered pins in disconnects are identified as A-.



G. How To Locate And Identify All Terminating Wires To A Terminal Strip

For a terminal module, the designator that represents all of the contact cavities in specific bus is the contact cavity with the lowest alphanumeric value. For example, if the bus indicator of the terminal module groups contact cavities A, B, and C, the item designator A is used to identify all three contact cavities.

- (a) As an example use Terminal Strip TB4301, and wires W5103-0002-*1 and W8100-0023-*1.
- (b) Knowing the terminal strip number TB4301, refer to Terminal Strip List in Chapter 91.
- (c) Using Self Indexing Terminal Strip List shown, locate terminal TB4301 at STA, WL, BL.
- (d) Locate the Wires terminating at Pins 2 as W5103-0002-*1 and W8100-0023-*1 and they are shown to be on diagram 24-21-11.
- (e) Refer to the Equipment List for complete information regarding TB4301.

H. How To Find The Remaining Wires Passing Through A Connector That Are Not Shown On The Particular Diagram Being Used.

- (a) As an example use connector DM31202A.
- (b) Note the connector equipment number and find that equipment number in the Hook-Up Charts as shown.
- (c) All wires in the connector are listed. The diagram for each wire is also shown.

I. How To Determine Splice Numbers Of Splices Found On The Airplane.

- (a) Determine the wire number of any wire terminating at the splice in question. As an example use wire number W0022-A-A.
- (b) Knowing the wire number, refer to the Wire List in Chapter 91-21-11.
- (c) Using the Self Indexing Wire List shown below, locate wire number W0022-A-A. Under the "EQUIP" column note the splice number SP00211.
- (d) The wiring diagram the splice is depicted on is listed under the "DIAGRAM" column. Refer to the Charts & Lists Section for more information.

J. How To Locate A Splice On The Airplane Which Is Shown On A Wiring Diagram

- (a) There are several types of splices shown on wiring diagrams.
 - "SP" and "SPZ" splices connect wire from different wire bundles.
 - "SM" and "SMZ" splices connect wires of the same wire bundle.
 - "SPZ" and "SPZ" reflect customer assigned splice numbers.
- (b) "SP" and "SPZ" splice location information is derived from the Splice List in Chapter 91. For example, use "SP25005" and locate it in the self indexing Splice List shown.
 - 1) Read the station "STA", water line "WL", and buttock line "LBL" or "RBL".



- 2) SP25005 is located at STA 1450, WL 300 and RBL 030. If more than one splice is at that location compare the airplane wire numbers to those listed in the Splice List to assure the correct splice has been found.
- (c) "SM" and "SMZ" splice location information is determined by noting the wire number of a wire terminating at the splice in question.
 - 1) Knowing the wire number W1071-2027R-22 locate it in the self indexing wire list shown.
 - 2) The end points of W1071-2027R-22 are listed under the "EQUIP" column as SM00006 and DS31105. The length of the wire is given under "FT/IN" column.
 - 3) Locate DS31105 in the Equipment List shown. Note the location of the equipment under "STATION-WL-BL" column. In this example the equipment is located in the P007 panel.
 - 4) If the location of the P007 panel is not known refer to the Equipment List. Locate P007 in the Equipment List example shown.
 - 5) Knowing P007 is at STA 0174, WL 250, LBL 000 go to the aircraft and locate it.
 - 6) Locate DS31105 inside the P007 panel. Refer to the wire list example below and note the wire length between SM00006 and DS31105 is 2 ft. The wire length given in the wire list is the finished wire length when feet and inches are listed. If only feet are listed then the finished length can be +/- 18 inches.
 - 7) Locate wire W1071-2027R-22 at DS31105 and trace it 2 ft +/- 6 inches and locate SM00006. Verify the wire leading away from splice is W1071-2027R-22.

K. How To Determine Part Numbers For Equipment Depicted On A Wiring Diagram

Electrical and electronic equipment shown on wiring diagrams are assigned alphanumeric designators. These designators are used as cross reference symbols to the Equipment List where the part numbers and part descriptions are shown. Splices, grounds, wire bundles and vendor controlled equipment numbers are not included in the Equipment List.

(a) Part Numbers For Special Contacts Used On Connectors

Special contacts for connectors are identified by a code in the Term Type (TT) column of the wire list. These codes and contact part number for each are listed in the Codes section.

(b) Part Numbers For Standard Contacts Used On Connectors

Standard contacts for connectors are not assigned term type codes. Part numbers are determined by locating the connector equipment number (i.e. D02516) in the Equipment List and its part number (i.e. BACS16W1). The connector part number and assembly are located in the Chapter 20 Standard Wiring Practices Manual, D6-54446. The contact number can then be determined by reviewing the connector assembly section.

(c) Part Numbers For Terminals

Part numbers for terminals depicted on wiring diagrams can be found in Chapter 20 Standard Wiring Practices Manual, D6-54446. The part numbers are based on the gage and type of wire.



(d) Part Numbers For Contacts Used On Terminal Blocks

The terminal block module part number must be determined by locating it in the Equipment List. Once this is known, refer to the Cross Reference Index in Chapter 20 to locate the contact part number and installation information.

L. How To Determine The Location Of Equipment Listed In The Equipment List

Each equipment location is defined by the following three dimensional coordinate system: Station Line, Water Line and Buttock Line.

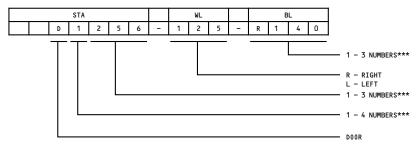
The airplane is divided into six different areas, each with its unique set of coordinates: (1) Body, (2) Wing, (3) Wing Tip, (4) Stabilizer, (5) Fin and (6) Nacelle.

The coordinate interrelations are defined in Chapter 91.

(a) The format of characters as indicated below represent the possible alpha-numeric combinations which define the different area coordinates.

1) Door Location

DOOR LOCATION

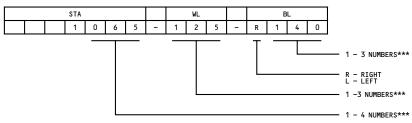


***INCLUDE LEADING ZEROS



2) Body Location

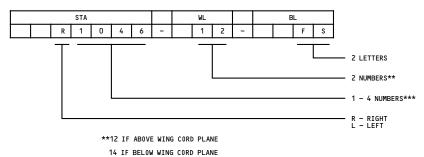
BODY LOCATION



***INCLUDE LEADING ZEROS

3) Wing Location

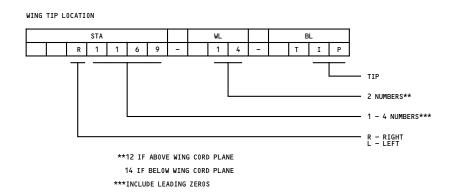
WING LOCATION



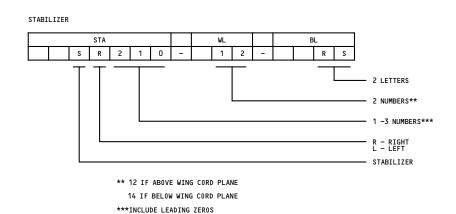
***INCLUDE LEADING ZEROS



4) Wing Tip Location

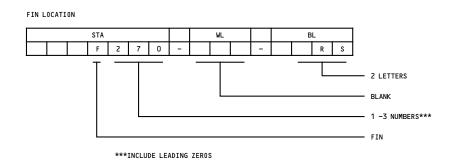


5) Stabilizer Location



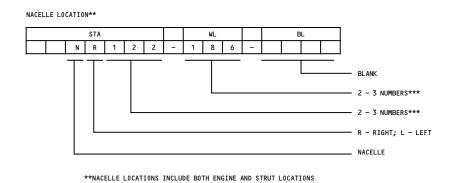


6) Fin Location



7) Nacelle Location

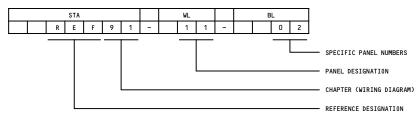
***INCLUDE LEADING ZEROS





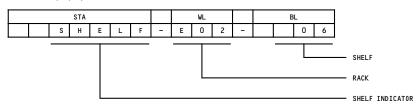
- (b) In addition to the aforementioned location definitions, the following special location definitions are necessary:
 - 1) Chapter 91 Cross Reference For Circuit Breaker

CHAPTER 91 CROSS REFERENCE FOR CIRCUIT BREAKER



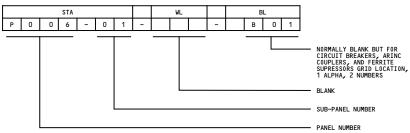
2) Shelves -E1, E2, E3, etc.

SHELVES - E1,E2,E3,ETC.



3) Panels -P1, P2, P3, etc.

PANEL - P1,P2,P3,ETC.

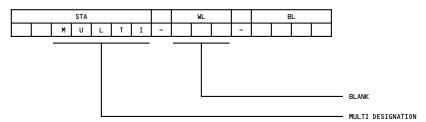




4) Multiple Location For Identical Units

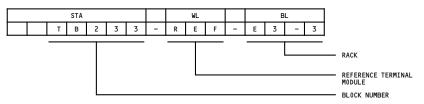
Nomenclature Of Item Denotes Location Usage

MULTI USAGE ITEM LOCATION



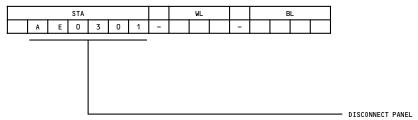
5) Diode-Rack

DIODE - RACK



6) Disconnect Panel Location

DISCONNECT PANEL LOCATION





7) The following areas and their codes give forward and aft positioning on wings, stabilizers, and fins.

IDENTIFIED AREA	CODE
Leading Edge	LE
Front Spar	FS
Mid Spar	MS
Rear Spar	RS
Trailing Edge	TE
Extended Trailing Edge	ET
Plus	Р
Minus	M

P or M define position, above or below respectively, of the wing or stabilizer chord plane.

2. METHODS USED TO ORDER A WIRE BUNDLE

A. How to Determine Wire Bundle (Harness) Information for Ordering a Wire Bundle

The Wire Bundle part numbers have been added to the Wire List for use when ordering a Wire Bundle form Boeing.

- (a) Find the Wire Bundle (ie. W0041) of interest in the Wire List.
 - Find the Part Number of the Wire Bundle, the second entry of the header row of each Wire Bundle.
- (b) Verify the Wire Bundle is applicable to the airplane of interest (Effectivity)
- (c) Provide Boeing the Wire Bundle Part Number and the airplane's Boeing Variable Number when ordering.



Chapter 20 consists of standard practices used to remove, repair and/or install wiring and equipment associated with wiring and its termination.

(This information is covered in D6-54446, the CHAPTER 20 STANDARD WIRING PRACTICES MANUAL.)