

COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

CONTROL CABIN FOOT OUTLET ASSEMBLY

PART NUMBER 65-55300–11, –12, –13, –14, –21, –22, –23, –24, –29, –30, –35, –36, 69-42775–41, –42, –45, –46, –49, –50, –53, –54, –55, –56

BOEING PROPRIETARY, CONFIDENTIAL, AND/OR TRADE SECRET

Copyright © 1995 The Boeing Company Unpublished Work - All Rights Reserved

Boeing claims copyright in each page of this document only to the extent that the page contains copyrightable subject matter. Boeing also claims copyright in this document as a compilation and/or collective work.

This document includes proprietary information owned by The Boeing Company and/or one or more third parties. Treatment of the document and the information it contains is governed by contract with Boeing. For more information, contact The Boeing Company, P.O. Box 3707, Seattle, Washington 98124.

Boeing, the Boeing signature, the Boeing symbol, 707, 717, 727, 737, 747, 757, 767, 777, 787, Dreamliner, BBJ, DC-8, DC-9, DC-10, KC-10, KDC-10, MD-10, MD-11, MD-80, MD-88, MD-90, P-8A, Poseidon and the Boeing livery are all trademarks owned by The Boeing Company; and no trademark license is granted in connection with this document unless provided in writing by Boeing.

PUBLISHED BY BOEING COMMERCIAL AIRPLANES GROUP, SEATTLE, WASHINGTON, USA A DIVISION OF THE BOEING COMPANY PAGE DATE: Jul 01/2009



Page 1 Jul 01/2009



Revision No. 9 Jul 01/2009

To: All holders of CONTROL CABIN FOOT OUTLET ASSEMBLY 21-40-01.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

ATTENTION

IF YOU RECEIVE PRINTED REVISIONS, PLEASE VERIFY THAT YOU HAVE RECEIVED AND FILED THE PREVIOUS REVISION. BOEING MUST BE NOTIFIED WITHIN 30 DAYS IF YOU HAVE NOT RECEIVED THE PREVIOUS REVISION. REQUESTS FOR REVISIONS OTHER THAN THE PREVIOUS REVISION WILL REQUIRE A COMPLETE MANUAL REPRINT SUBJECT TO REPRINT CHARGES SHOWN IN THE DATA AND SERVICES CATALOG.



COMPONENT MAINTENANCE MANUAL

Location of Change

Description of Change NO HIGHLIGHTS





Subject/Page	Date	Subject/Page	Date	Subject/	Page	Date
TITLE PAGE		21-40-01 CLEAN	NG (cont)	21-40-01	ILLUSTI	RATED PARTS LIST
0 1	Jul 01/2009	402	BLANK	(cont)		
2	BLANK	21-40-01 CHECK		1007		Mar 01/2006
21-40-01 TRANS	MITTAL LETTER	501	Mar 01/2006	1008		Mar 01/2006
0 1	Jul 01/2009	502	BLANK	1009		Mar 01/2006
2	BLANK	21-40-01 REPAIF	- GENERAL	1010		Mar 01/2007
21-40-01 HIGHLI	GHTS	601	Mar 01/2007	1011		Mar 01/2007
0 1	Jul 01/2009	602	BLANK	1012		BLANK
2	BLANK	21-40-01 REPAIF	1-1			
21-40-01 EFFEC	TIVE PAGES	601	Mar 01/2006			
1	Jul 01/2009	602	Mar 01/2006			
2	BLANK	21-40-01 REPAIF	2-1			
21-40-01 CONTE	NTS	601	Mar 01/2006			
1	Mar 01/2006	602	Mar 01/2006			
2	BLANK	21-40-01 REPAIF	3-1			
21-40-01 TR AND) SB RECORD	601	Mar 01/2006			
1	Mar 01/2006	602	Mar 01/2006			
2	BLANK	21-40-01 REPAIF	4-1			
21-40-01 REVISI	ON RECORD	601	Mar 01/2006			
1	Mar 01/2006	602	Mar 01/2006			
2	Mar 01/2006	21-40-01 REPAIF	5-1			
21-40-01 RECOR	D OF TEMPORARY	601	Mar 01/2007			
REVISIONS		602	BLANK			
1	Mar 01/2006	21-40-01 ASSEM	BLY			
2	Mar 01/2006	701	Mar 01/2006			
21-40-01 INTROE	DUCTION	702	BLANK			
1	Mar 01/2009	21-40-01 FITS AN	ID CLEARANCES			
2	BLANK	801	Mar 01/2006			
21-40-01 DESCR	IPTION AND	802	BLANK			
1	Mar 01/2006	21-40-01 SPECIA AND EQUIPMEN	L TOOLS, FIXTURES, IT			
2	BLANK	901	Mar 01/2006			
21-40-01 TESTIN ISOLATION	G AND FAULT	902				
101	Mar 01/2006	21-40-01 ILLUST				
102	BLANK	1001	NOV U1/2008			
21-40-01 DISASS	SEMBLY	1002	Jui U1/2006			
301	Mar 01/2006	1003	War 01/2007			
302	BLANK	1004	War 01/2007			
21-40-01 CLEAN	ING	1005	War 01/2007			
401	Mar 01/2006	1006	war 01/2000			

A = Added, R = Revised, D = Deleted, O = Overflow





TABLE OF CONTENTS

Paragraph Title		Page
CONTROL CABIN FOOT OUTLET ASSEMBLY - DESCRIPT OPERATION	TION AND	1
TESTING AND FAULT ISOLATION		101
DISASSEMBLY	(Not Applicable)	
CLEANING	(Not Applicable)	
CHECK	(Not Applicable)	
REPAIR		601
ASSEMBLY	(Not Applicable)	
FITS AND CLEARANCES	(Not Applicable)	
SPECIAL TOOLS, FIXTURES, AND EQUIPMENT	(Not Applicable)	
ILLUSTRATED PARTS LIST		1001





TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR 32118	Jun 05/92
		PRR 33330	Jun 05/92
		PRR 35137	Mar 01/97





All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Rev	vision	Fi	led	Revision		Filed	
Number	Date	Date	Initials	Number	Date	Date	Initials





Rev	/ision	Fi	led	Revision Fi		led	
Number	Date	Date	Initials	Number	Date	Date	Initials

21-40-01 REVISION RECORD Page 2 Mar 01/2006



All temporary revisions to this manual will be accompanied by a cover sheet bearing the temporary revision number. Enter the temporary revision number in numerical order, together with the temporary revision date, the date the temporary revision is inserted and the initials of the person filing.

When the temporary revision is incorporated or cancelled, and the pages are removed, enter the date the pages are removed and the initials of the person who removed the temporary revision.

Temporary	Revision	Ins	serted	Rei	moved	Tempora	ary Revision	Inser	ted	Rer	noved
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials

21-40-01 RECORD OF TEMPORARY REVISION Page 1 Mar 01/2006



Temporary	Revision	Ins	serted	Rei	moved	Tempora	ary Revision	Inser	ted	Rer	noved
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials

21-40-01 RECORD OF TEMPORARY REVISION Page 2 Mar 01/2006



INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.





CONTROL CABIN FOOT OUTLET ASSEMBLY - DESCRIPTION AND OPERATION

1. Description and Operation

A. The control cabin foot outlet assembly consists of a glass reinforced plastic valve body and valve assemblies. The valve assemblies are controlled by cranks. The cranks are operated by push-pull rods. The valve assemblies control the amount of cooling air that is directed on to the pilot's and co-pilot's feet.

2. Leading Particulars (Approximate)

- A. Weight 1 pound
- B. Diameter 8 inches
- C. Thickness 3 inches





TESTING AND FAULT ISOLATION

1. Functional Test

A. Visually check that each valve makes contact fully around on the seal with no more than 1 pound force applied to the crank arm.





DISASSEMBLY





CLEANING





<u>CHECK</u>





REPAIR

1. Content

A. Instructions for repair, refinish and replacement procedures are divided into procedures as follows:

	Table 601:	
P/N	NAME	REPAIR
69-42780	RETAINER	1-1
65-55300	VALVE ASSEMBLY	2-1
69-42775-7, -8	VALVE BODY ASSEMBLY	3-1
69-42775-41, -42, -45, -46, -49, -50, -53, · 55, -56	-54, - VALVE BODY ASSEMBLY	4-1
_	MISC PARTS REFINISH	5-1

2. Standard Practices

A. Refer to these standard practices for details of the procedures in the repairs.

- SOPM 20-30-02 Stripping of Protective Finishes
- SOPM 20-30-03 General Cleaning Procedure
- SOPM 20-41-01 Decoding Table for Boeing Finish Codes
- SOPM 20-50-12 Application of Adhesives
- SOPM 20-60-02 Finishing Materials
- SOPM 20-60-04 Miscellaneous Materials

3. Materials

- A. adhesive, A00028 BAC 5010, Type 70 Adhesive
- B. adhesive, A50091 BAC5010, Type 82 Adhesive
- C. primer, C00259 BMS 10-11, Type 1 Primer
- D. sealant, A00551 BAC 5010, Type 44 Sealant





RETAINER - REPAIR 1-1

69-42780-3, -4

1. <u>General</u>

- A. This procedure has the data to repair the initial finish of the retainer (150).
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 1 for the item numbers.

2. Finish Repair

A. Refer to REPAIR 1-1, Figure 601 for retainer repair.



BOEING"

COMPONENT MAINTENANCE MANUAL





Retainer Repair (150) Figure 601

> 21-40-01 REPAIR 1-1 Page 602 Mar 01/2006



VALVE ASSEMBLY - REPAIR 2-1

65-55300-3, -4, -9, -10, -15, -16, -17, -18

1. General

- A. This procedure has the data to repair the initial finish of the valve assembly (10, 20).
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 1 for the item numbers.

2. Finish Repair

A. Refer to REPAIR 2-1, Figure 601 for valve assembly repair.



BOEING"

COMPONENT MAINTENANCE MANUAL



Valve Assembly Repair (10,15) Figure 601

> 21-40-01 REPAIR 2-1 Page 602 Mar 01/2006



VALVE BODY ASSEMBLY - REPAIR 3-1

69-42775-7, -8

1. General

- A. This procedure has the data to repair the initial finish of the valve body assembly (80).
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 1 for the item numbers.

2. Refinish

A. Refer to REPAIR 3-1, Figure 601 for valve body assembly repair (80).



BOEING®

COMPONENT MAINTENANCE MANUAL



Valve Body Assembly Repair (60) Figure 601

21-40-01 **REPAIR 3-1** Page 602 Mar 01/2006



VALVE BODY ASSEMBLY - REPAIR 4-1

69-42775-41, -42, -45, -46

1. General

- A. This procedure has the data to repair the initial finish of the valve body assembly (5, 10, 15, 20).
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 2 for the item numbers.

2. Refinish

A. Refer to REPAIR 4-1, Figure 601 for valve body assembly repair.









1 BOND FAYING SURFACES AS SHOWN IN SOPM 20-50-12, TYPE 70.

Valve Body Assembly Repair Figure 601

21-40-01 **REPAIR 4-1** Page 602 Mar 01/2006



MISCELLANEOUS PARTS REFINISH - REPAIR 5-1

1. General

- A. This procedure has the data necessary to refinish the parts which are not given in other repairs.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 1 and IPL Figure 2 for the item numbers.

2. Refinish

A. Repair of parts listed in REPAIR 5-1, Table 601 consists of restoration of the original finish.

IPL FIG. & ITEM MATERIAL		FINISH
IPL Fig. 1		
Retainer (150)	AISI 302 CRES	No finish required (F-25.01).
Lever Assembly (30)	AISI 321 CRES Annealed	No finish required (F-25.01).
Valve Body (80,85)	Thermosetting resin impregnated aramid fiber fabric. Polycarbonate.	No finish required (F-25.01).
Bracket (130,135)	Al alloy	Alodize (SRF 2.31)
IPL Fig. 2		
Bracket (50)	Al alloy	Chemical treat and apply primer, C00259 (F-18.06).
Plate (85)	Al alloy	Anodize (F-17.31) and apply primer, C00259 (F-20.02).
Spud (65)	Al alloy	Chemical treat (F-17.08).

Table 601: Refinish Details





ASSEMBLY





FITS AND CLEARANCES





SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

21-40-01 SPECIAL TOOLS, FIXTURES, AND EQUIPMENT Page 901 Mar 01/2006



ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1 2 3 4 5 0	1	2	3	4	5	6	7
-------------	---	---	---	---	---	---	---

- . Assembly
- . Attaching parts for assembly
- . Detail parts for assembly
- . . Subassembly
- . . Attaching parts for subassembly
- Detail parts for subassembly
- . . . Sub-subassembly
- . . . Attaching parts for subassembly
 - Details parts for sub-subassembly

Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
 - (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
 - (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

21-40-01 ILLUSTRATED PARTS LIST Page 1001 Nov 01/2008



Optional (OPT)

Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)

Replaces, Replaced by (REPLACES, REPLACED BY)

The part is optional to and interchangeable with other parts that have the same item number.

The part replaces and is not interchangeable with the initial part.

The part replaces and is interchangeable with, or is an alternative to, the initial part.





NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
214A1502-3		2	85	1
214T3488-265		2	65	1
65-55300-10		1	25	1
65-55300-11		1	1	RF
65-55300-12		1	5	RF
65-55300-13		1	1A	RF
65-55300-14		1	5A	RF
65-55300-15		1	10A	1
65-55300-16		1	15A	1
65-55300-17		1	20A	1
65-55300-18		1	25A	1
65-55300-21		1	1B	RF
65-55300-22		1	5B	RF
65-55300-23		1	1C	RF
65-55300-24		1	5C	RF
65-55300-29		1	1D	RF
65-55300-3		1	10	1
65-55300-30		1	5D	RF
65-55300-35		1	1E	RF
65-55300-36		1	5E	RF
65-55300-4		1	15	1
65-55300-9		1	20	1
69-42774-1		1	30	1
69-42774-13		1	45A	1
69-42774-17		1	45	1
69-42774-3		1	35	1
69-42774-9		1	75	2
69-42775-10		1	95	1
69-42775-11		1	90A	1
69-42775-12		1	95A	1
69-42775-14		1	100	2
69-42775-15		1	80	1
69-42775-16		1	85	1
69-42775-18		1	105	1

21-40-01 ILLUSTRATED PARTS LIST Page 1003 Mar 01/2007



PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-42775-19		1	130A	1
69-42775-20		1	135A	1
69-42775-31		1	80B	1
69-42775-32		1	85B	1
69-42775-35		2	25	2
69-42775-36		2	30	2
69-42775-38		2	35	1
69-42775-39		2	40	1
69-42775-4		1	75A	2
		1	130	1
69-42775-41		2	5	RF
69-42775-42		2	5A	RF
		2	10	RF
69-42775-43		2	45	1
69-42775-44		2	50	1
69-42775-45		2	5B	RF
		2	15	RF
69-42775-46		2	5C	RF
		2	20	RF
69-42775-47		2	35A	1
		2	55	1
69-42775-48		2	30A	2
		2	60	1
69-42775-49		2	5D	RF
69-42775-5		1	40	1
		1	135	1
69-42775-50		2	5E	RF
69-42775-53		2	5F	RF
69-42775-54		2	5G	RF
69-42775-55		2	5H	RF
69-42775-56		2	5J	RF
69-42775-7		1	80A	1
69-42775-8		1	85A	1
69-42775-9		1	90	1
69-42780-3		1	150	1

21-40-01 ILLUSTRATED PARTS LIST Page 1004 Mar 01/2007



PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-42780-4		1	150A	1
69-47862-7		1	60	1
69-47862-8		1	65	1
AN737TW48		1	70	1
AN960PD3		1	125	K 4
AN960PD4		1	145	AR
BACR15BA5AD(-)C		2	90	2
BACR15BB4A		1	140	2
BACS12GU3K7		2	80	1
MS20470A		1	120	4
MS20615-2M3		1	50	2
MS20615-2M5		1	55	2
MS21042L3		2	75	1
MS24665-151		1	155	2
NAS1149DD332J		2	70	2
NAS557-3A		1	165	4
NAS620C10		1	160	AR







Foot Outlet Assembly IPL Figure 1

> **21-40-01** ILLUSTRATED PARTS LIST Page 1006 Mar 01/2006



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
1	65-55300-11		OUTLET ASSY-FOOT-CABIN CONTROL	А	RF
-1A	65-55300-13		OUTLET ASSY-FOOT-CABIN CONTROL	С	RF
–1B	65-55300-21		OUTLET ASSY-FOOT-CABIN CONTROL	E	RF
-1C	65-55300-23		OUTLET ASSY-FOOT-CABIN CONTROL	G	RF
–1D	65-55300-29		OUTLET ASSY-FOOT-CABIN CONTROL	I	RF
–1E	65-55300-35		OUTLET ASSY-FOOT-CABIN CONTROL	К	RF
5	65-55300-12		OUTLET ASSY-FOOT-CABIN CONTROL	В	RF
–5A	65-55300-14		OUTLET ASSY-FOOT-CABIN CONTROL	D	RF
–5B	65-55300-22		OUTLET ASSY-FOOT-CABIN CONTROL	F	RF
–5C	65-55300-24		OUTLET ASSY-FOOT-CABIN CONTROL	н	RF
–5D	65-55300-30		OUTLET ASSY-FOOT-CABIN CONTROL	J	RF
–5E	65-55300-36		OUTLET ASSY-FOOT-CABIN CONTROL	L	RF
10	65-55300-3		. VALVE ASSY (OPPOSITE ITEM 15)	A, I, K	1
–10A	65-55300-15		. VALVE ASSY (OPPOSITE ITEM 15A)	C, E, G	1
-15	65-55300-4		. VALVE ASSY (OPPOSITE ITEM 10)	B, J, L	1
–15A	65-55300-16		. VALVE ASSY (OPPOSITE ITEM 10A)	D, F, H	1
20	65-55300-9		. VALVE ASSY (OPPOSITE ITEM 25)	A, I, K	1
–20A	65-55300-17		. VALVE ASSY (OPPOSITE ITEM 25A)	C, E, G	1
-25	65-55300-10		. VALVE ASSY (OPPOSITE ITEM 20)	B, J, L	1
–25A	65-55300-18		. VALVE ASSY (OPPOSITE ITEM 20A)	D, F, H	1
30	69-42774-1		LEVER ASSY		1
35	69-42774-3		PLATE		1
40	69-42775-5		ARM-VALVE		1
45	69-42774-17		CAP (USED ON ITEMS 10, 15, 20, 25)		1

-Item not Illustrated

21-40-01 ILLUSTRATED PARTS LIST Page 1007 Mar 01/2006



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
-45A	69-42774-13		CAP (USED ON ITEMS 10A, 15A, 20A, 25A)		1
50	MS20615-2M3		RIVET		2
55	MS20615-2M5		RIVET		2
-60	69-47862-7		. DIFFUSER ASSY	A, C	1
65	69-47862-8		. DIFFUSER ASSY	B, D	1
-70	AN737TW48		. CLAMP	A-D	1
75	69-42774-9		. SEAL		2
–75A	69-42775-4		. SEAL (OPT ITEM 60)		2
80	69-42775-15		. BODY-VALVE	А	1
–80A	69-42775-7		. BODY ASSY-VALVE	C, E, G	1
-80B	69-42775-31		. BODY-VALVE	I, K	1
85	69-42775-16		. BODY-VALVE	В	1
–85A	69-42775-8		. BODY ASSY-VALVE	D, F, H	1
–85B	69-42775-32		. BODY-VALVE	J, L	1
-90	69-42775-9		BODY HALF-VALVE (USED ON ITEM 80A) (OPPOSITE ITEM 100)		1
-90A	69-42775-11		BODY HALF-VALVE USED ON ITEM 80A) (OPPOSITE ITEM 105)		1
-95	69-42775-10		BODY HALF-VALVE (USED ON ITEM 85A) (OPPOSITE ITEM 90)		1
-95A	69-42775-12		BODY HALF-VALVE (USED ON ITEM 85A) (OPPOSITE ITEM 95)		1
-100	69-42775-14		PLATE-SPLICE (USED ON ITEMS 80A, 85A)		2
-105	69-42775-18		SPUD (USED ON ITEMS 80A, 85A)		1
120	MS20470A		. RIVET		4
125	AN960PD3		. WASHER	I, J, L	K 4
130	69-42775-4		. BRACKET	A, C	1

-Item not Illustrated

21-40-01 ILLUSTRATED PARTS LIST Page 1008 Mar 01/2006



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
–130A	69-42775-19		. BRACKET	E, G, I, K	1
-135	69-42775-5		. BRACKET	B, D	1
–135A	69-42775-20		. BRACKET	F, H, J, L	1
140	BACR15BB4A		. RIVET	E-L	2
145	AN960PD4		. WASHER	E-L	AR
150	69-42780-3		. RETAINER	A-D	1
-150A	69-42780-4		. RETAINER	E-L	1
155	MS24665-151		. PIN-COTTER		2
160	NAS620C10		. WASHER	E-L	AR
165	NAS557-3A		. GROMMET		4



-Item not Illustrated





Valve Body Assembly IPL Figure 2

21-40-01 ILLUSTRATED PARTS LIST Page 1010 Mar 01/2007



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
2–					
5	69-42775-41		BODY ASSY-VALVE	А	RF
–5A	69-42775-42		BODY ASSY-VALVE	В	RF
–5B	69-42775-45		BODY ASSY-VALVE	С	RF
–5C	69-42775-46		BODY ASSY-VALVE	D	RF
–5D	69-42775-49		BODY ASSY-VALVE	E	RF
–5E	69-42775-50		BODY ASSY-VALVE	F	RF
–5F	69-42775-53		BODY ASSY-VALVE	G	RF
–5G	69-42775-54		BODY ASSY-VALVE	н	RF
–5H	69-42775-55		BODY ASSY-VALVE	J	RF
–5J	69-42775-56		BODY ASSY-VALVE	К	RF
-10	69-42775-42		DELETED		
-15	69-42775-45		DELETED		
-20	69-42775-46		DELETED		
25	69-42775-35		. VALVE-MOVING		2
30	69-42775-36		. VALVE-STATIONARY	А, В	2
30A	69-42775-48		. VALVE-STATIONARY	C-K	2
35	69-42775-38		. VALVE HALF-INDEXED	А, В	1
35A	69-42775-47		. VALVE HALF-INDEXED	C-K	1
40	69-42775-39		. VALVE-HALF		1
45	69-42775-43		. DUCT-ELBOW		1
50	69-42775-44		. BRACKET		1
55	69-42775-47		DELETED		
-60	69-42775-48		DELETED		
65	214T3488-265		. SPUD	G-K	1
70	NAS1149DD332J		. WASHER		2
75	MS21042L3		. NUT		1
80	BACS12GU3K7		. SCREW		1
85	214A1502-3		. PLATE	E-K	1
			INSTALLATION PARTS		
90	BACR15BA5AD(-)C		. RIVET (SIZE DETERMINED ON INST)	E-K	2

-Item not Illustrated