



**COMPONENT MAINTENANCE
MANUAL
WITH
ILLUSTRATED PARTS LIST**

**OVERHEAD STOWAGE COMPARTMENT
ASSEMBLY**

**PART NUMBER
65C27005-76**

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COMPONENT MAINTENANCE MANUAL

Revision No. 8
Jul 01/2009

To: All holders of OVERHEAD STOWAGE COMPARTMENT ASSEMBLY 25-28-43.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

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TRANSMITTAL LETTER

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Location of Change

Description of Change

NO HIGHLIGHTS

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HIGHLIGHTS

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A = Added, R = Revised, D = Deleted, O = Overflow

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TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR 33180-48	JUN 05/85
25A1283		PRR 35041-2R	SEP 05/92
25-1294		PRR 35041-2R	SEP 05/92

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TR AND SB RECORD

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All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed	
Number	Date	Date	Initials

Revision		Filed	
Number	Date	Date	Initials



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All temporary revisions to this manual will be accompanied by a cover sheet bearing the temporary revision number. Enter the temporary revision number in numerical order, together with the temporary revision date, the date the temporary revision is inserted and the initials of the person filing. When the temporary revision is incorporated or cancelled, and the pages are removed, enter the date the pages are removed and the initials of the person who removed the temporary revision.

Temporary Revision		Inserted		Removed		Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials



COMPONENT MAINTENANCE MANUAL

INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

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OVERHEAD STOWAGE COMPARTMENT ASSEMBLY - DESCRIPTION AND OPERATION

1. Description

A. The overhead stowage compartment assembly is composed of a nomex honeycomb core body which supports a light, bullnose and PSU supports, fittings and brackets, a latch which mounts on the bullnose extrusion of the body, and a door which connects to the body by two hinges with mechanical actuators.

2. Operation

A. The door is restrained in the closed position by engagement of pawls on the latch with striker rollers located on each end of the door. When force is applied to the latch handle, the pawls disengage from striker rollers and allow the door, reacting to actuator force, to open and rotate upwards.

3. Leading Particulars (approximate)

- A. Width – 28 inches
- B. Length – 30-60 inches
- C. Height – 14 inches
- D. Weight – 20-27 pounds

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TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

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TESTING AND FAULT ISOLATION

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DISASSEMBLY

1. General

- A. This procedure has the data necessary to disassemble the overhead stowage compartment assembly.
- B. Disassemble this component only as necessary to complete fault isolation, determine part serviceability, perform required repairs, and restore unit to serviceable condition.
- C. Refer to IPL Figure 1 for item numbers.

2. Disassembly

CAUTION: THE STOWAGE ASSEMBLY CONSISTS OF APPEARANCE ITEMS. SPECIAL CARE MUST BE TAKEN TO PREVENT MARRING VISIBLE SURFACES.

- A. No special instructions required. Use standard industry practices.

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DISASSEMBLY

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CLEANING

(NOT APPLICABLE)

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CHECK

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CHECK
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REPAIR

1. Contents

A. Repair, refinish, and replacement procedures are included in separate repair sections as follows:

Table 601:

P/N	NAME	REPAIR
412N1119-1	FITTING ASSY	1-1
412N1220	COMPARTMENT ASSEMBLY	2-1
412N1221	MODULE ASSEMBLY	3-1
- - -	MISC. PARTS REFINISH	4-1

2. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.

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REPAIR - GENERAL

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—	STRAIGHTNESS	∅	DIAMETER
□	FLATNESS	S ∅	SPHERICAL DIAMETER
⊥	PERPENDICULARITY (OR SQUARENESS)	R	RADIUS
//	PARALLELISM	SR	SPHERICAL RADIUS
○	ROUNDNESS	()	REFERENCE
⊙	CYLINDRICITY	BASIC	A THEORETICALLY EXACT DIMENSION USED
⌒	PROFILE OF A LINE	(BSC)	TO DESCRIBE SIZE, SHAPE OR LOCATION OF
⌒	PROFILE OF A SURFACE	OR	A FEATURE. FROM THIS FEATURE PERMISSIBLE
◎	CONCENTRICITY	DIM	VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
≡	SYMMETRY	-A-	DATUM
∠	ANGULARITY	Ⓜ	MAXIMUM MATERIAL CONDITION (MMC)
↗	RUNOUT	Ⓛ	LEAST MATERIAL CONDITION (LMC)
↗	TOTAL RUNOUT	Ⓢ	REGARDLESS OF FEATURE SIZE (RFS)
□	COUNTERBORE OR SPOTFACE	Ⓟ	PROJECTED TOLERANCE ZONE
∇	COUNTERSINK	FIM	FULL INDICATOR MOVEMENT
⊕	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)		

EXAMPLES

— 0.002	STRAIGHT WITHIN 0.002	◎ ∅ 0.0005 C	CONCENTRIC TO DATUM C WITHIN 0.0005 DIAMETER
⊥ 0.002 B	PERPENDICULAR TO DATUM B WITHIN 0.002	≡ 0.010 A	SYMMETRICAL WITH DATUM A WITHIN 0.010
// 0.002 A	PARALLEL TO DATUM A WITHIN 0.002	∠ 0.005 A	ANGULAR TOLERANCE 0.005 WITH DATUM A
○ 0.002	ROUND WITHIN 0.002	⊕ ∅ 0.002 Ⓢ B	LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
⊙ 0.010	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	⊥ ∅ 0.010 Ⓜ A	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010 INCH DIAMETER, PERPENDICULAR TO DATUM A, AND EXTENDING 0.510 INCH ABOVE DATUM A, MAXIMUM MATERIAL CONDITION
⌒ 0.006 A	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM A	0.510 Ⓟ	
⌒ 0.020 A	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.020 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE	2.000	THEORETICALLY EXACT DIMENSION IS 2.000
		OR	
		2.000	
		BSC	

True Position Dimensioning Symbols
Figure 601

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REPAIR - GENERAL

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FITTING ASSEMBLY - REPAIR 1-1

412N1119-1

1. General

- A. This procedure has the data necessary to repair and refinish the fitting assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to the REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing replacement (93, IPL Figure 1) (REPAIR 1-1, Figure 601)

A. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT

B. Procedure

- (1) Remove bushing.
- (2) Install new bushing per SOPM 20-50-03.
- (3) Machine 0.375-0.380 inch hole through bushing per REPAIR 1-1, Figure 601.

3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For finishing materials, refer to SOPM 20-60-02.

- (1) Fitting (99 IPL Figure 1) – Chromic acid anodize and apply primer, C00259 (F-18.13). Material: Al alloy.

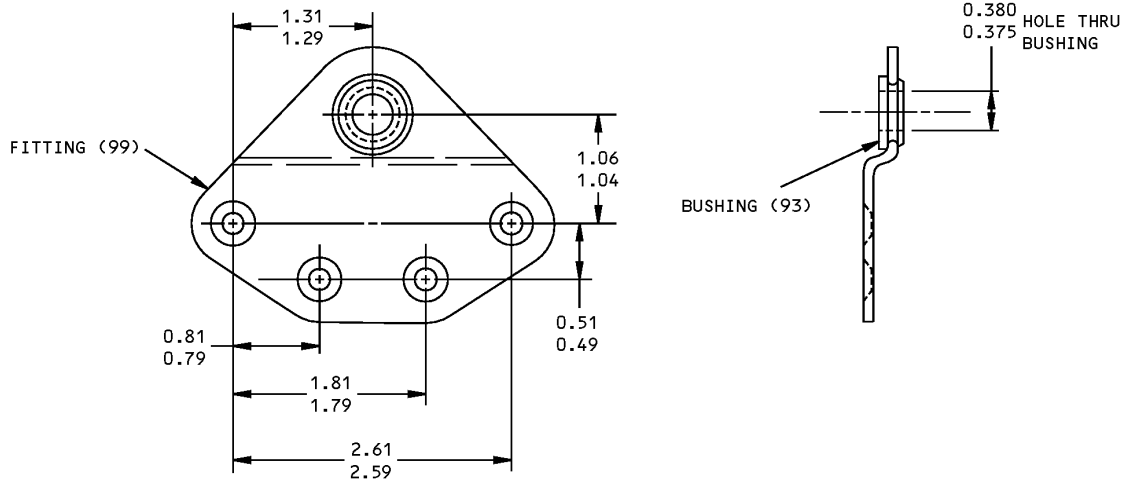
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ALL DIMENSIONS ARE IN INCHES

Bushing Replacement
Figure 601

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REPAIR 1-1

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COMPARTMENT ASSEMBLY - REPAIR 2-1

65C27005-76

1. General

- A. This procedure has the data necessary to repair and refinish the compartment assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Fitting Assembly Replacement (81, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00028	Adhesive - Modified Epoxy For Rigid PVC, Foam Cored Sandwiches	BAC5010, Type 70 (BMS5-92, Type 1)

- B. References

Reference	Title
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-50-12	APPLICATION OF ADHESIVES
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure

NOTE: For general cleaning procedure, refer to SOPM 20-30-03. For Miscellaneous materials, refer to SOPM 20-60-04.

- (1) Remove fitting assembly from compartment.
- (2) Bond fitting assembly to compartment using adhesive, A00028 per SOPM 20-50-12.

3. Fitting Replacement (9, 105, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00028	Adhesive - Modified Epoxy For Rigid PVC, Foam Cored Sandwiches	BAC5010, Type 70 (BMS5-92, Type 1)

- B. References

Reference	Title
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-50-12	APPLICATION OF ADHESIVES
SOPM 20-60-04	MISCELLANEOUS MATERIALS

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C. Procedure

NOTE: For general cleaning procedure, refer to SOPM 20-30-03. For Miscellaneous materials, refer to SOPM 20-60-04.

- (1) Remove fitting from compartment.
- (2) Bond fitting to compartment using adhesive, A00028 per SOPM 20-50-12.

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REPAIR 2-1

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MODULE ASSEMBLY - REPAIR 3-1

65C27008-74

1. General

- A. This procedure has the data necessary to repair and refinish the module assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Bracket replacement (383, 389, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00028	Adhesive - Modified Epoxy For Rigid PVC, Foam Cored Sandwiches	BAC5010, Type 70 (BMS5-92, Type 1)

- B. References

Reference	Title
SOPM 20-50-12	APPLICATION OF ADHESIVES

- C. Procedure

- (1) Removal bracket from module.
- (2) Bond fitting on module using adhesive, A00028 per SOPM 20-50-12.

3. Support replacement (395, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00028	Adhesive - Modified Epoxy For Rigid PVC, Foam Cored Sandwiches	BAC5010, Type 70 (BMS5-92, Type 1)

- B. References

Reference	Title
SOPM 20-50-12	APPLICATION OF ADHESIVES

- C. Procedure

- (1) Remove support from module.
- (2) Bond support to module using adhesive, A00028 per SOPM 20-50-12.

4. Retainer replacement (255, 258 IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

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Reference	Description	Specification
A00028	Adhesive - Modified Epoxy For Rigid PVC, Foam Cored Sandwiches	BAC5010, Type 70 (BMS5-92, Type 1)
A50055	Adhesive - Two-Part, RT Cure, Urethane	BAC5010, Type 89 (BMS5-105)

B. References

Reference	Title
SOPM 20-50-12	APPLICATION OF ADHESIVES

C. Procedure

- (1) Remove retainer from module.
- (2) Bond retainer to module using adhesive, A00028 or urethane adhesive, A50055, per SOPM 20-50-12.

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REPAIR 3-1
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MISCELLANEOUS PARTS REFINISH - REPAIR 4-1

1. General

- A. This procedure has the data necessary to refinish the parts which are not given in the other repairs.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Refinish details

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedures, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Repair of parts listed in REPAIR 4-1, Table 601 consists of restoration of original finish.

Table 601: Refinish Details

IPL FIG. & ITEM	MATERIAL	FINISH
Fig. 1		
Fitting (9,105, 117) Bracket (383,389)	Al alloy	Chromic acid anodize and apply (F-18.13).
Bracket (129,132, 135)	Al alloy	Chemical treat and apply primer, C00259 (F-18.06).
Support (416)	Al alloy	Sulfuric acid anodize (F-14.2992).
Fitting (347)	Al alloy	Anodize type I or type II (F-17.05)
Support (395)	Al alloy	Apply colorless alodine coating (F-14.05) all over.
Support (177)	Al alloy	Chemical treat and apply primer, C00259 (F-18.05).

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REPAIR 4-1

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ASSEMBLY

1. General

- A. This procedure has the data necessary to assemble the overhead stowage compartment assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Assembly

A. References

Reference	Title
SOPM 20-50-01	BOLT AND NUT INSTALLATION
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-03	LUBRICANTS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

B. Procedure

NOTE: For bolt and nut installation, refer to SOPM 20-50-01. For finishing materials, refer to SOPM 20-60-02. For lubricants, refer to SOPM 20-60-03. For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Use standard industrial practices and the following adjustment procedures.
- (2) Latch and door adjustment:
 - (a) Loosen slotted shank bolts (273) to allow movement of hinge assemblies (267, 268, 270, 271) on stowage compartment.
 - (b) Verify that striker plates (204) are located on door (210) to the dimensions shown on ASSEMBLY, Figure 701 and that attaching screws (207) are not tightened.
 - (c) Close door and check that strikers are aligned centrally in latch covers (365, 368). Adjust door on hinge quadrants as necessary to properly align strikers.
 - (d) With door closed and latched, pull down on center of door to seat striker as far downward and inward as possible in the latch covers. While holding door in that position, tighten slotted shank bolts (273) which protrude from side panels of compartment.
 - (e) Tighten striker plate attach screws (455).
 - (f) Ensure proper adjustment of latch. Close door and check that:
 - 1) Door closes with minimal force.
 - 2) Latching pawls engage with striker rollers on door.
 - 3) Latch actuating PUSH handle protrude from latch housing per ASSEMBLY, Figure 702. This condition will only exist when the latching pawls are fully engaged and door fully closed. Any reading beyond the tolerance given in ASSEMBLY, Figure 702 will indicate that door is not latched and ASSEMBLY, Paragraph 2.B.(2) must be repeated.
 - (g) After adjustment of latch assembly, open door and tighten hinge assembly attach bolts from inside of stowage compartment.

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- (3) Perform latching and loading test as follows:
 - (a) With door closed and latched, position bin at 60 degrees (ASSEMBLY, Figure 703) using an appropriate fixture.
 - (b) Open door 1 inch and hold open using an appropriate spacer block.
 - (c) Apply seven (7) pound load at position one (1), pull spacer out, and check that door closes and latches at both latch locations.
 - (d) Repeat test at position two (2) on door.
 - (e) If door does not properly close and latch, readjust door and latches per ASSEMBLY, Paragraph 2.B.(2).

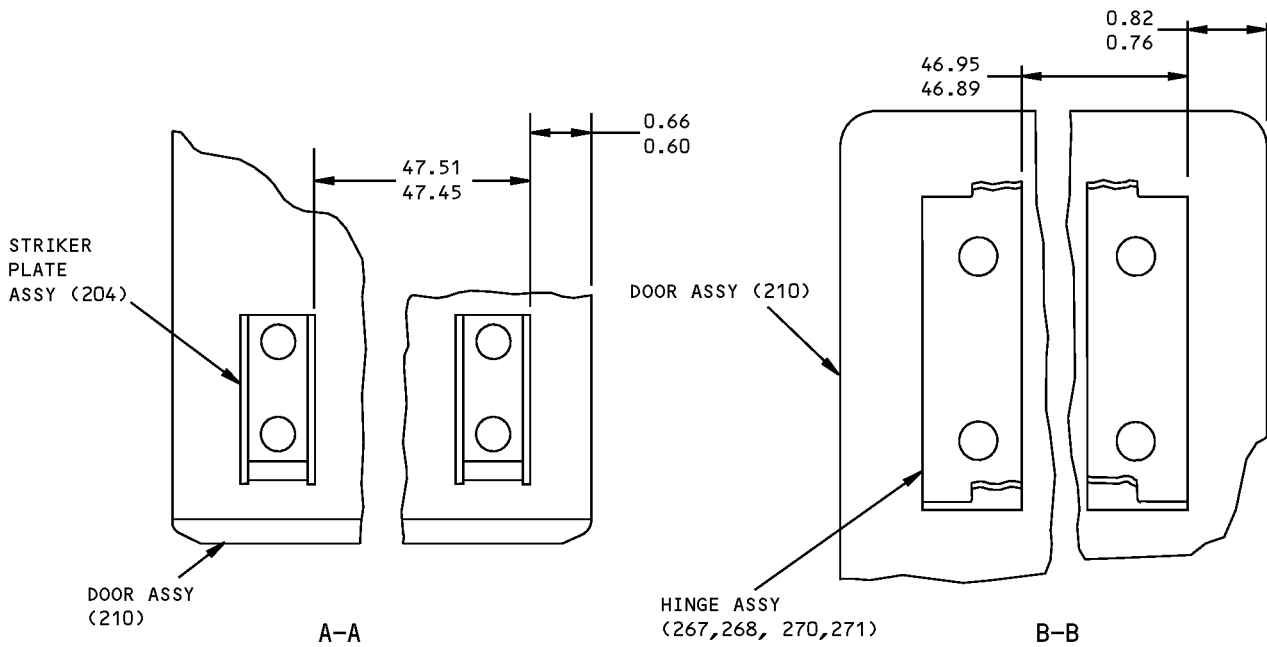
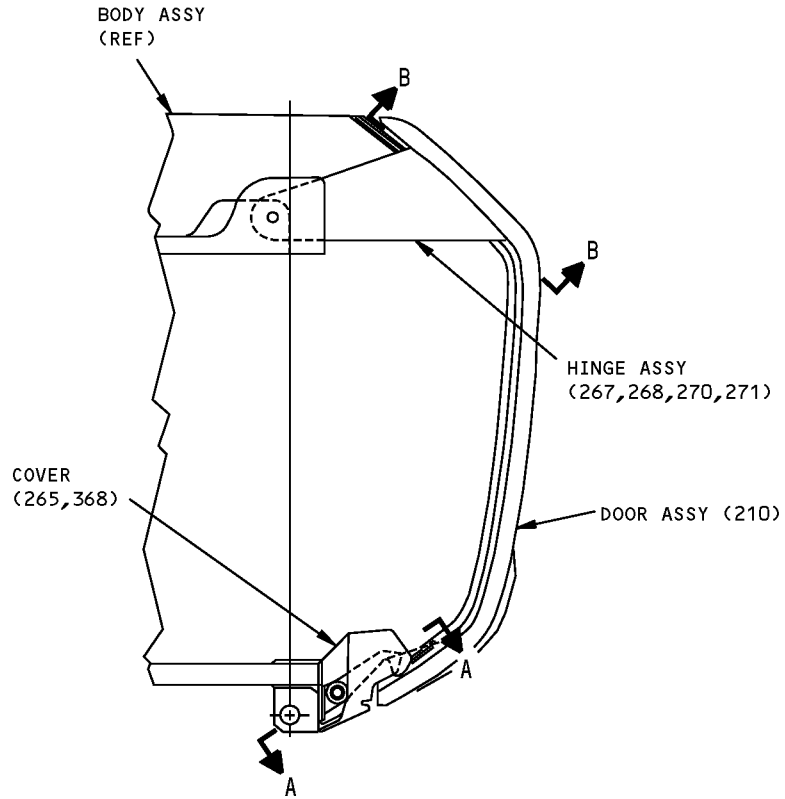
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ASSEMBLY

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ALL DIMENSIONS ARE IN INCHES

Module Assembly
Figure 701

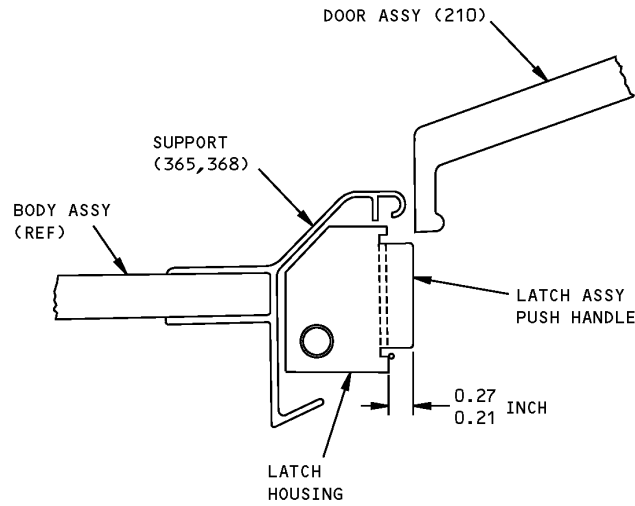
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ASSEMBLY

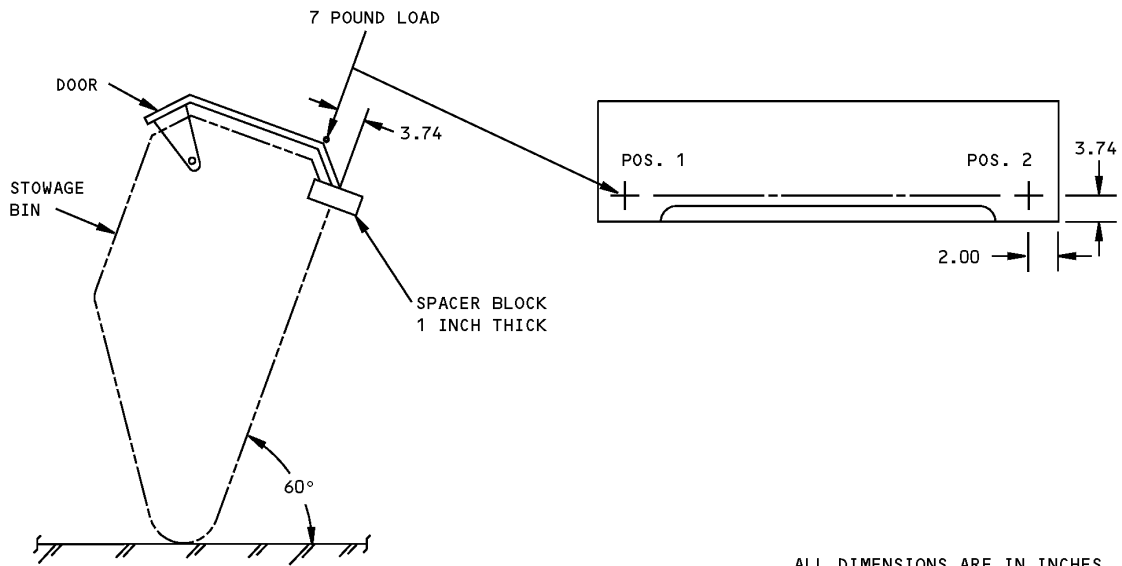
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Door and Latch Assembly Adjustment
Figure 702



ALL DIMENSIONS ARE IN INCHES

Door Latching and Loading Test
Figure 703

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ASSEMBLY

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FITS AND CLEARANCES

(NOT APPLICABLE)

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FITS AND CLEARANCES

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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

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COMPONENT MAINTENANCE MANUAL

ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
.	Assembly					
.	Attaching parts for assembly					
.	.	Detail parts for assembly				
.	.	Subassembly				
.	.	Attaching parts for subassembly				
.	.	.	Detail parts for subassembly			
.	.	.	Sub-subassembly			
.	.	.	Attaching parts for subassembly			
.	.	.	.	Details parts for sub-subassembly		
						Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
- (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
- (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

VENDOR CODES

Code	Name
01556	EMHART FASTENING SYSTEMS GROUP HELI-COIL DIV SHELTER ROCK LANE DANBURY, CONNECTICUT 06810 FORMERLY HELI-COIL DIV OF BLACK AND DECKER
04169	WESTERN SKY INDUSTRIES A DIVISION OF ATLAS CORPORATION 1280 SAN LUIS OBISPO STREET HAYWARD, CALIFORNIA 94544-7916 FORMERLY WESTERN SKY IND VB0008
05617	IDD AEROSPACE CORP 18225 NORTHEAST 76TH STREET PO BOX 97056 REDMOND, WASHINGTON 98073-9756 FORMERLY FARWEST ELEC INC; FORMERLY IN BELLEVUE, WA; FORMERLY BELL IND FARWEST MFG DIV; FORMERLY BELL IND ILLUMINATED DISPLAYS DIV
06710	LAMSON AND SESSIONS CO THE VALLEY-TODECO 12975 BRADLEY AVENUE SYLMAR, CALIFORNIA 91342-3830 FORMERLY VALLEY BOLT CORP VB0097 IN NORTH HOLLYWOOD, CA
06725	AIR INDUSTRIES CORPORATION 12570 KNOTT STREET GARDEN GROVE, CALIFORNIA 92641-3932 FORMERLY AIR INDUSTRIES OF CALIF IN GARDENA, CALIF.

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COMPONENT MAINTENANCE MANUAL

Code	Name
06950	SCREWCORP VSI AEROSPACE PRODUCTS DIV FAIRCHILD IND DIV 13001 EAST TEMPLE AVENUE PO BOX 730 CITY OF INDUSTRY, CALIFORNIA 91746-1417 FORMERLY VB0096 AND VSI CORP SCREWCORP DIV FORMERLY IN CULVER CITY, CALIFORNIA SCREW CORP SEE V.S.I. CORP SCREWCORP DIVISION
08524	Replaced: [V08524] DEUTSCH FASTENER CORP SEE CODE V97928 Replaced: [V97928] SEE V17446 HUCK INTL by Code: Name and Address below 17446: HUCK INTL INC AEROSPACE FASTENER DIV 900 WATSON CENTER ROAD CARSON, CALIFORNIA 90745-4201 FORMERLY V32134 REXNORD INC; FORMERLY V97928 HUCK INTL Referenced in FORMERLY line below [17419] DEUTSCH COMPANY THE WELLS FARGO BANK BLDG 2444 WILSHIRE BLVD #600 SANTA MONICA, CALIFORNIA 90403 FORMERLY DEUTSCH FASTENER CORP V08524 FORMERLY IN LOS ANGELES
08805	GENERAL ELECTRIC 1975 NOBLE RD CLEVELAND, OHIO 44112-1719
10630	ANILLO INDUSTRIES, INCORPORATED 2090 NORTH GLASSELL ORANGE, CALIFORNIA 92667 FORMERLY WESTERN WASHER DIV OF SENG CO V87487
11815	CHERRY AEROSPACE FASTENERS DIV OF TEXTRON 1224 EAST WARNER AVENUE PO BOX 2157 SANTA ANA, CALIFORNIA 92707-0157 FORMERLY IN LOS ANGELES, CALIF , FORMERLY CHERRY FASTENERS TOWNSEND DIV OF TEXTRON INC V71087
15653	ALCOA GLOBAL FASTENERS INC DIV KAYNAR PRODUCTS 800 S STATE COLLEGE BLVD FULLERTON, CALIFORNIA 92831-3001 FORMERLY VK6405 MICRODOT AEROSP LTD; FORMERLY KAYNAR TECH FORMERLY FAIRCHILD FASTENERS KAYNAR DIV

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COMPONENT MAINTENANCE MANUAL

Code	Name
17023	BRUCE INDUSTRIES, INCORPORATED 101 EVANS AVENUE PO BOX 1700 DAYTON, NEVADA 89403 FORMERLY IN GARDENA, CALIFORNIA
24984	HYDRAFLOW 1881 W MALVERN FULLERTON, CALIFORNIA 92833-2403 FORMERLY IN CERRITOS, CA
27624	PB FASTENERS DIV OF BRILES PAUL R 1700 WEST 132ND STREET GARDENA, CALIFORNIA 90249 FORMERLY PAUL R BRILES INC P.B. FASTENER DIV
34252	GULF OIL CO USA DIV OF GULF OIL CORP PO BOX 1519 HOUSTON, TEXAS 77001
34941	HARPER ENGINEERING COMPANY 200 SOUTH TOBIN STREET RENTON, WASHINGTON 98055-1307
52828	REPUBLIC FASTENER MFG CORP 1300 RANCHO CONEJO BLVD NEWBURY PARK, CALIFORNIA 91320-1405 FORMERLY IN SYLMAR, CALIFORNIA
56878	SPS TECHNOLOGIES INC AEROSPACE AND INDUSTRIAL PRODUCTS DIV 301 HIGHLAND AVE JENKINTOWN, PENNSYLVANIA 19046 FORMERLY STANDARD PRESSED STEEL FORMERLY IN SALT LAKE, UTAH
60116	PSI WEST 255 SOUTH 7TH AVENUE CITY OF INDUSTRY, CALIFORNIA 91744-3079 FORMERLY V0812B

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COMPONENT MAINTENANCE MANUAL

Code	Name
71087	<p>Replaced: [V71087] BOOTS ACFT NUT DIV TOWNSEND CO SEE TEXTRON INC CHERRY FASTENER TOWNSEND DIV V11815 by Code: Name and Address below 11815: CHERRY AEROSPACE FASTENERS DIV OF TEXTRON 1224 EAST WARNER AVENUE PO BOX 2157 SANTA ANA, CALIFORNIA 92707-0157 FORMERLY IN LOS ANGELES, CALIF , FORMERLY CHERRY FASTENERS TOWNSEND DIV OF TEXTRON INC V71087</p>
72962	<p>HARVARD INDUSTRIES INC 3 WERNER WAY SUITE 210 LEBANON, NEW JERSEY 08833 FORMERLY ESNA V7A079 FORMERLY ELASTIC STOP NUT IN UNION, NJ</p>
73197	<p>HI-SHEAR TECHNOLOGY CORP 2600 SKYPARK DRIVE TORRANCE, CALIFORNIA 90509</p>
80539	<p>SPS TECHNOLOGIES INC DIV AERPSOACE - SANTA ANA 2701 SOUTH HARBOR BOULEVARD SANTA ANA, CALIFORNIA 92704-5803 FORMERLY NUTT-SHEL DIV OF SPC WESTERN CO V80539 AND STANDARD PRESSED STEEL WESTERN DIV V17279</p>
83014	<p>HARTWELL CORPORATION 900 SOUTH RICHFIELD ROAD PLACENTIA, CALIFORNIA 92670-6732 FORMERLY V0532B IN LOS ANGELES, CALIFORNIA</p>
92215	<p>FAIRCHILD IND INC FAIRCHILD AEROSPACE FASTENER DIV 3010 W LOMITA BLVD TORRANCE, CALIFORNIA 90505-5102 FORMERLY VOI-SHAN IN CULVER CITY, CALIF</p>
93907	<p>TEXTRON INC CAMCAR DIV 600 18TH AVENUE ROCKFORD, ILLINOIS 61101</p>

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**COMPONENT MAINTENANCE MANUAL**

Code	Name
97393	SHUR-LOK CORPORATION 2541 WHITE ROAD PO BOX 19584 IRVINE, CALIFORNIA 92623-9584 FORMERLY SHUR LOK CORP VB0060 FORMERLY IN SANTA ANA, CALIFORNIA 92714
97928	Replaced: [V97928] SEE V17446 HUCK INTL by Code: Name and Address below 17446: HUCK INTL INC AEROSPACE FASTENER DIV 900 WATSON CENTER ROAD CARSON, CALIFORNIA 90745-4201 FORMERLY V32134 REXNORD INC; FORMERLY V97928 HUCK INTL

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NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
05939-7		1	36	1
2055132		1	42	2
22K1-02		1	33	3
26CL02		1	33	3
411N3142-28		1	69	1
412N1113-7		1	365	1
412N1113-8		1	368	1
412N1117-1		1	347	1
412N1118-1		1	255	1
412N1118-2		1	258	1
412N1119-1		1	81	1
412N1119-12		1	117	1
412N1119-14		1	9	1
412N1119-2		1	99	1
412N1119-62		1	105	1
412N1130-5		1	126	1
412N1140-13		1	247	1
412N1140-42		1	210	1
412N1140-59		1	246	1
412N1141-2		1	248	1
412N1141-8		1	249	1
412N1142-16		1	250	1
412N1143-13		1	240	1
412N1143-14		1	244	1
412N1145-21		1	129	1
412N1145-27		1	383	1
412N1145-28		1	389	1
412N1145-29		1	132	1
412N1145-30		1	135	2
412N1145-69		1	390	1
412N1146-2		1	398	1
412N1146-22		1	416	1
412N1153-102		1	353	1
412N1154-1		1	261	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
412N1154-2		1	264	1
412N1155-1		1	267	1
412N1155-2		1	270	1
412N1155-7		1	268	1
412N1155-8		1	271	1
412N1156-1		1	314	1
412N1156-13		1	315	1
412N1156-14		1	318	1
412N1156-15		1	324	1
412N1156-16		1	327	1
412N1156-2		1	317	1
412N1156-3		1	323	1
412N1156-4		1	326	1
412N1215-71		1	525	1
412N1227-1		1	273	4
412N1245-1		1	232	2
417N5120-12		1	39	1
417N5120-20		1	12	2
417N5120-7		1	15	1
6041-3BRX225		1	320	2
65C27005-76		1	1	RF
65C27008-74		1	192	1
65C27011-1008		1	435	1
65C27011-1207		1	445	1
65C27011-1210		1	460	1
65C27011-1274		1	480	1
65C27011-1322		1	495	1
65C27011-2007		1	505	1
65C27011-2107		1	515	1
65C27011-30		1	430	1
65C27023-16		1	395	1
65C27027-14		1	177	1
65C27027-17		1	159	1
65C27027-18		1	156	1
65C27027-19		1	162	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C27027-20		1	180	2
65C27027-21		1	168	1
65C27027-22		1	171	1
65C27027-8		1	174	1
65C28117-3		1	51	1
800000255-17		1	54	1
		80	1A	RF
800000255-7		1	54A	1
		80	1B	RF
96-02		1	344	4
96-82		1	338	4
		1	407	11
AIC1321-3-8		1	273A	4
AN960PD10		1	6	8
		1	78	4
		1	123	8
		1	144	8
AN960PD10L		1	24	4
		1	218	4
		1	287	2
		1	296	2
AN960PD8		1	335	4
		1	404	11
B30MY6K2		1	341	4
BACB28B6-220P		1	93	1
BACB30LK3-18		1	276	2
BACB30MY6K2		1	341	4
BACB30NM3K1		1	3	8
		1	3	8
		1	3	8
		1	3	8
		1	3	8
		1	3	8
		1	3	8
		1	3	8
		1	3	8

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACS12ER08K12		1	359	2
		1	359	2
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
		1	401	11
BACS12ER08K14		1	332	4
BACS12ER3K1		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
		1	120	8
BACS12ER3K6		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3
		1	27	3

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACW10P136AL		1	279	6
		1	279	6
BACW10UD10		1	30	3
BR1000A08		1	413	2
BRH10A08		1	338	4
		1	407	11
BRH10A3		1	344	4
BV03112-03-33		80	10	2
BV03112-33		80	10A	2
F15T8WW		1	45	1
F22K1-02		1	33	3
H10-08BAC		1	338	4
		1	407	11
H10-3BAC		1	344	4
H2948-1		1	308	1
H2948-2		1	305	1
H3-1145-1-4760		1	371	1
H3-1152-1		1	204	2
HL10VAZ6-2		1	341	4
		1	341	4
		1	341	4
		1	341	4
K2000-08		1	413	2
		1	413	2
L8006K2		1	341	4
LP1610-18		1	48	1
MS20392-2C13		1	284	1
MS24585-1230		1	66	1
MS24665-132		1	281	1
MS35206-323		80	5	4
MS35333-39		1	208	4
MS51862-4		1	236	2
NA43DD3-8		1	303	1
NAS1801-3-10		1	276A	2
NAS1801-3-11		1	60	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
NAS1801-3-4		1	214	4
NAS1801-3-5		1	21	4
NAS1833-3-400		1	329	6
NAS1834-4-400		1	465	2
NAS1836-3-11		1	440	19
		1	450	4
		1	470	4
		1	485	8
		1	500	8
		1	510	4
		1	520	6
NAS1836-4-11		1	490	2
NAS43DD3-3		1	300	1
NAS43DD3-5		1	299	1
NAS43DD3-6		1	302	1
NAS514P832-5		1	362	4
NAS620A8L		1	230	2
NAS6703PU11		1	293	1
NAS6703PU15		1	290	1
NS202101-02		1	344	4
NS202101-82		1	338	4
		1	407	11
RMA9207-82		1	413	2
RMLH9075-3W		1	344	4
RMLH9075-82W		1	338	4
		1	407	11
S283N702-7		1	36	1
S412N101-1		1	204	2
S412N101-1050		1	371	1
S412N102-1		1	308	1
S412N102-2		1	305	1
S412N104-1		1	311	1
S417N507-17		1	54	1
		80	1A	RF
S417N507-7		1	54A	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		80	1B	RF
SB113		1	311	1
SL210-06-2		1	374	4
SL210-08-1		1	377	4
SL2334-3A2		1	165	6
		1	455	4
		1	475	4
SL618-3A		1	222	8
T6S1032J		1	344	4
T6S832J		1	338	4
		1	407	11
T8109S832		1	413	2
VN101A1-82		1	413	2
VN303A02		1	344	4
VN303A82		1	338	4
		1	407	11

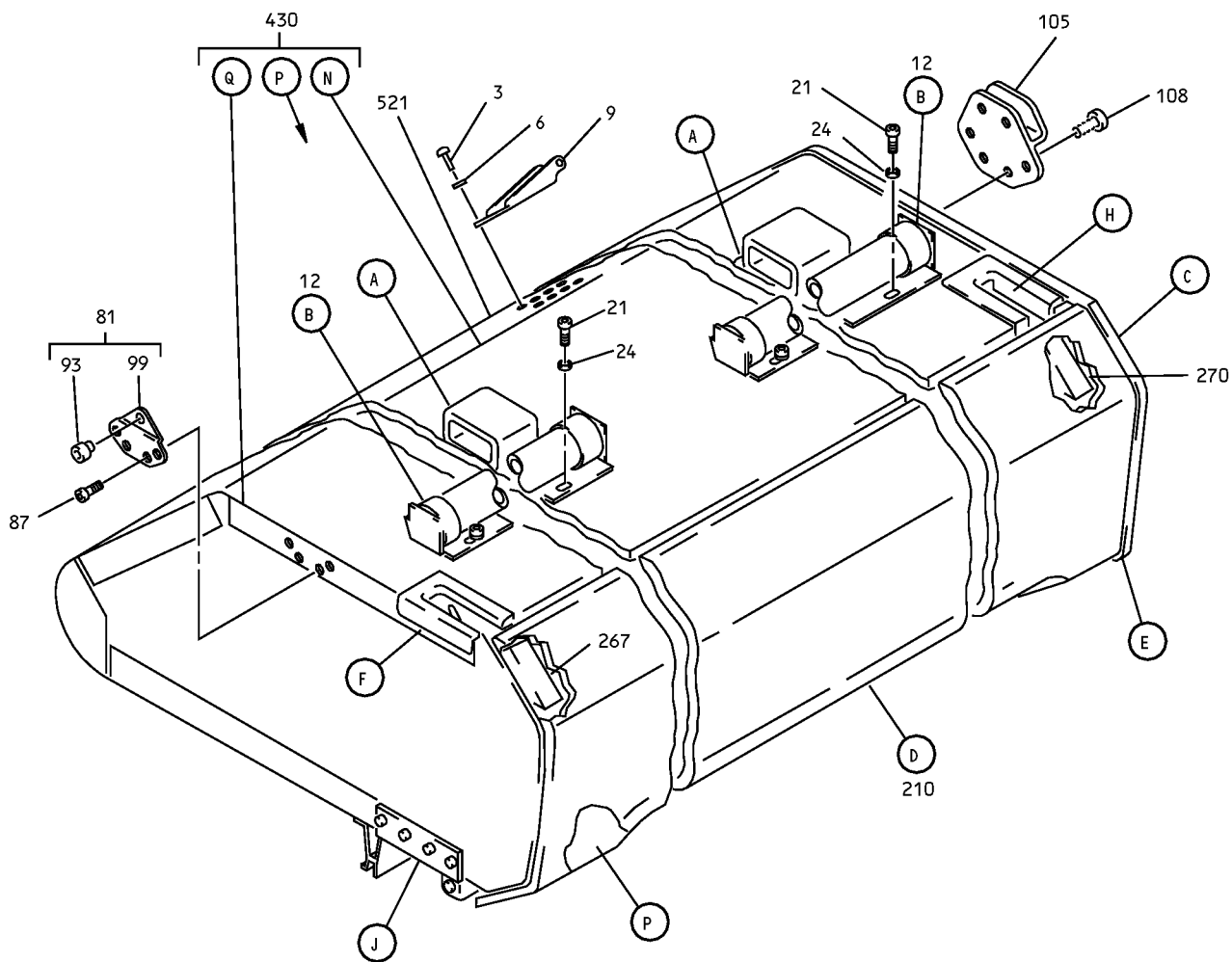
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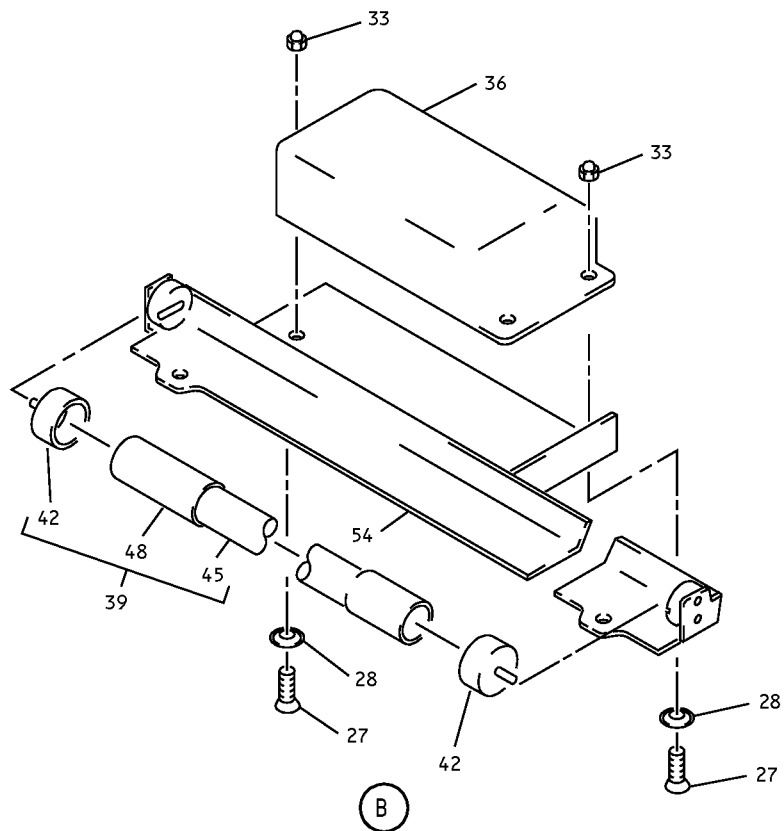
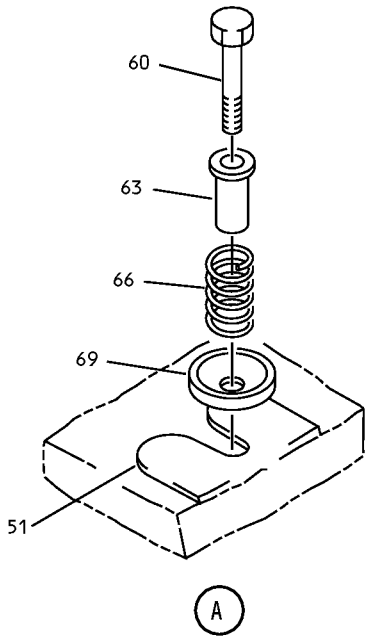
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Overhead Stowage Compartment Assembly
IPL Figure 1 (Sheet 1 of 10)

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Overhead Stowage Compartment Assembly
IPL Figure 1 (Sheet 2 of 10)

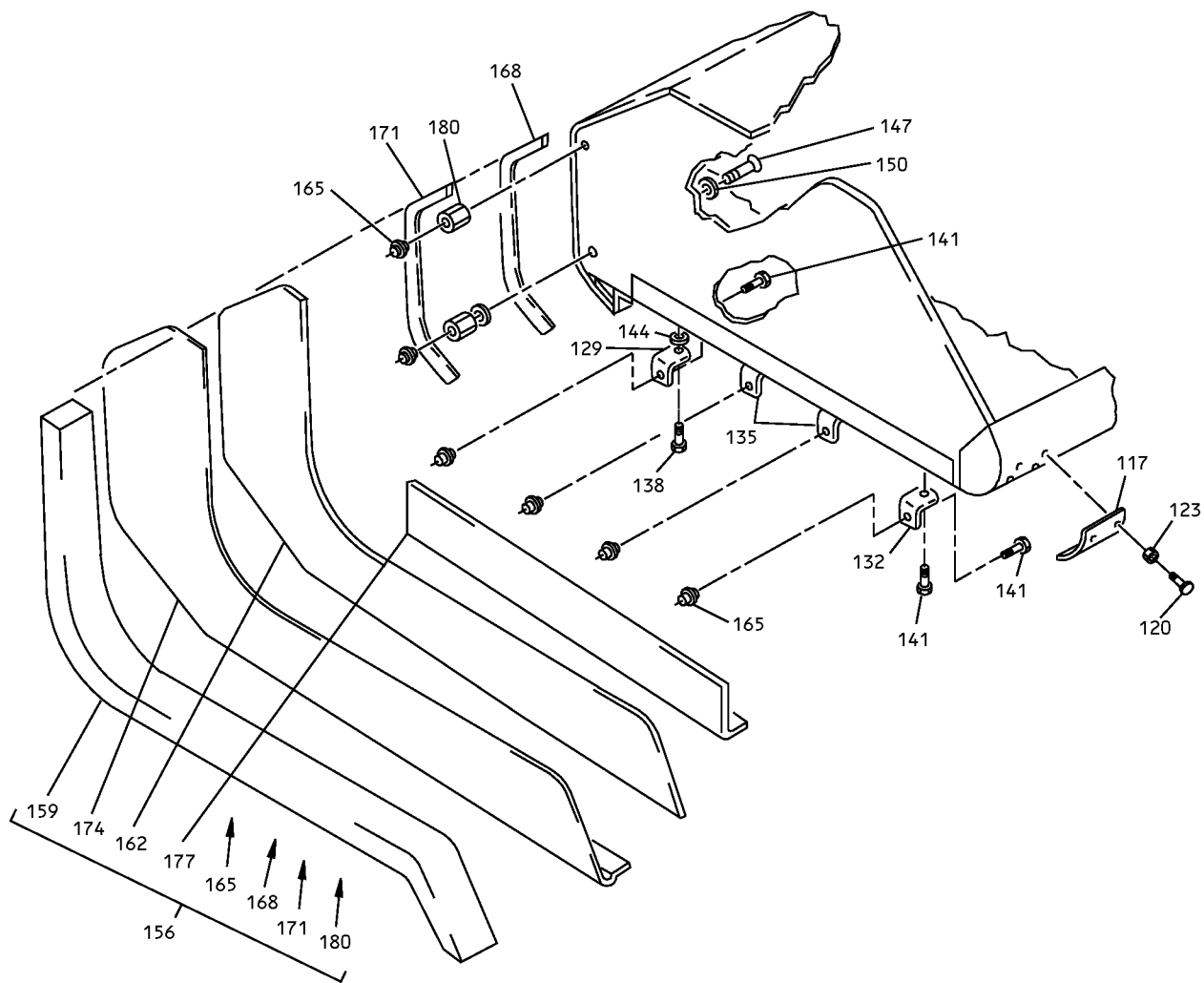
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Overhead Stowage Compartment Assembly
IPL Figure 1 (Sheet 3 of 10)

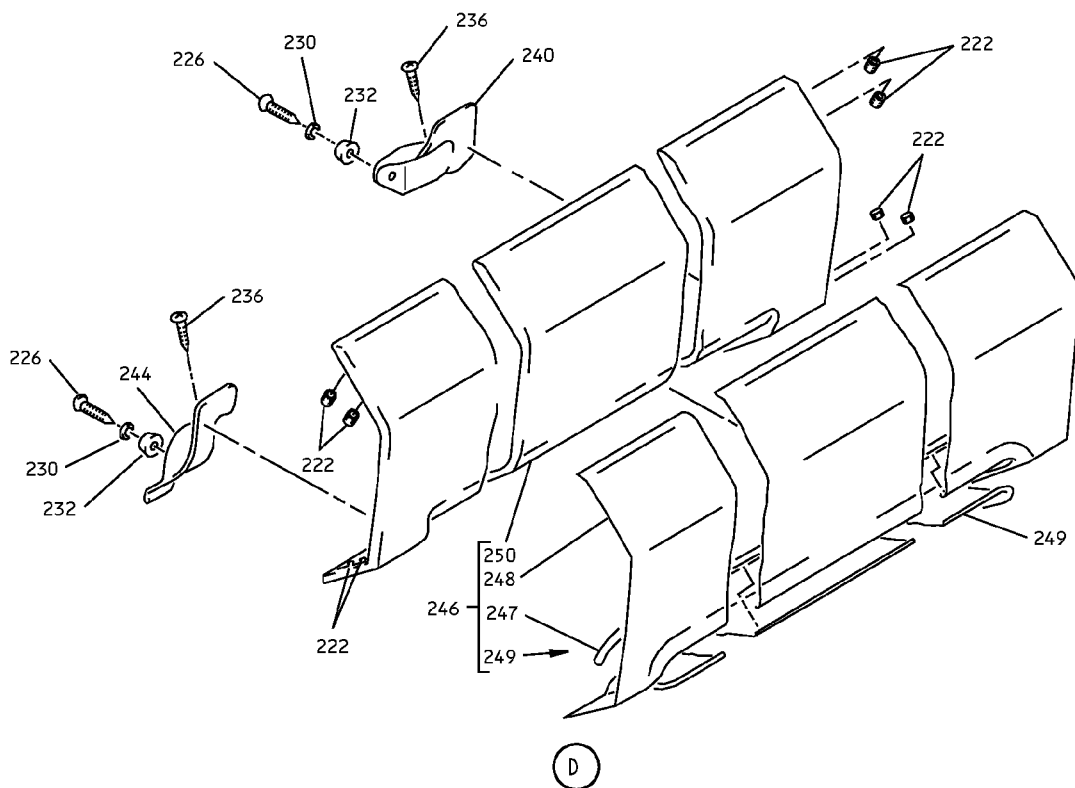
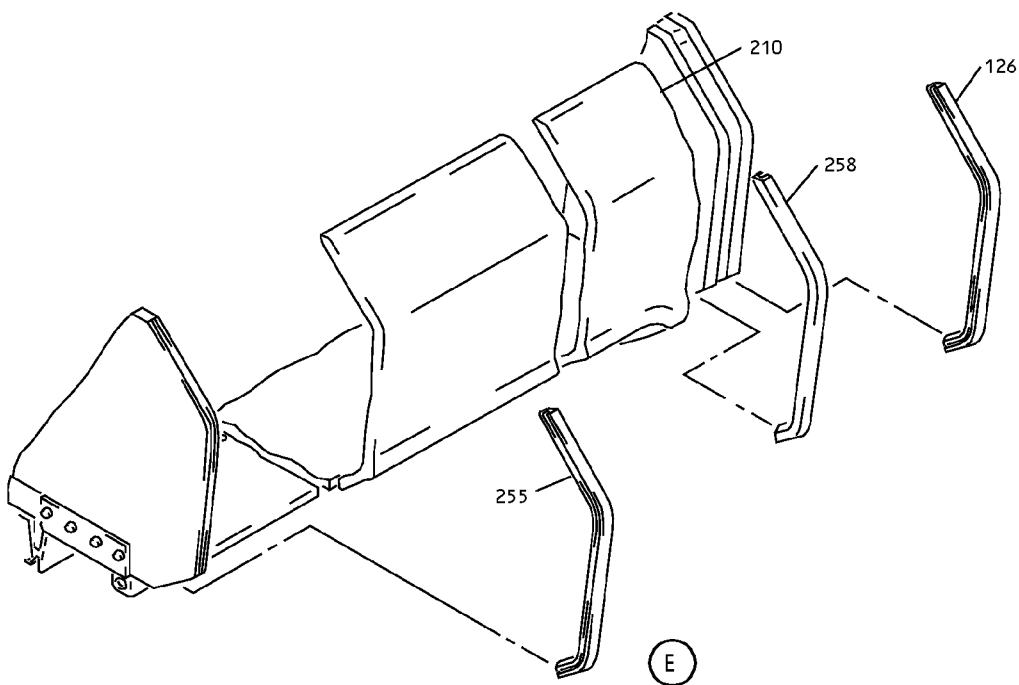
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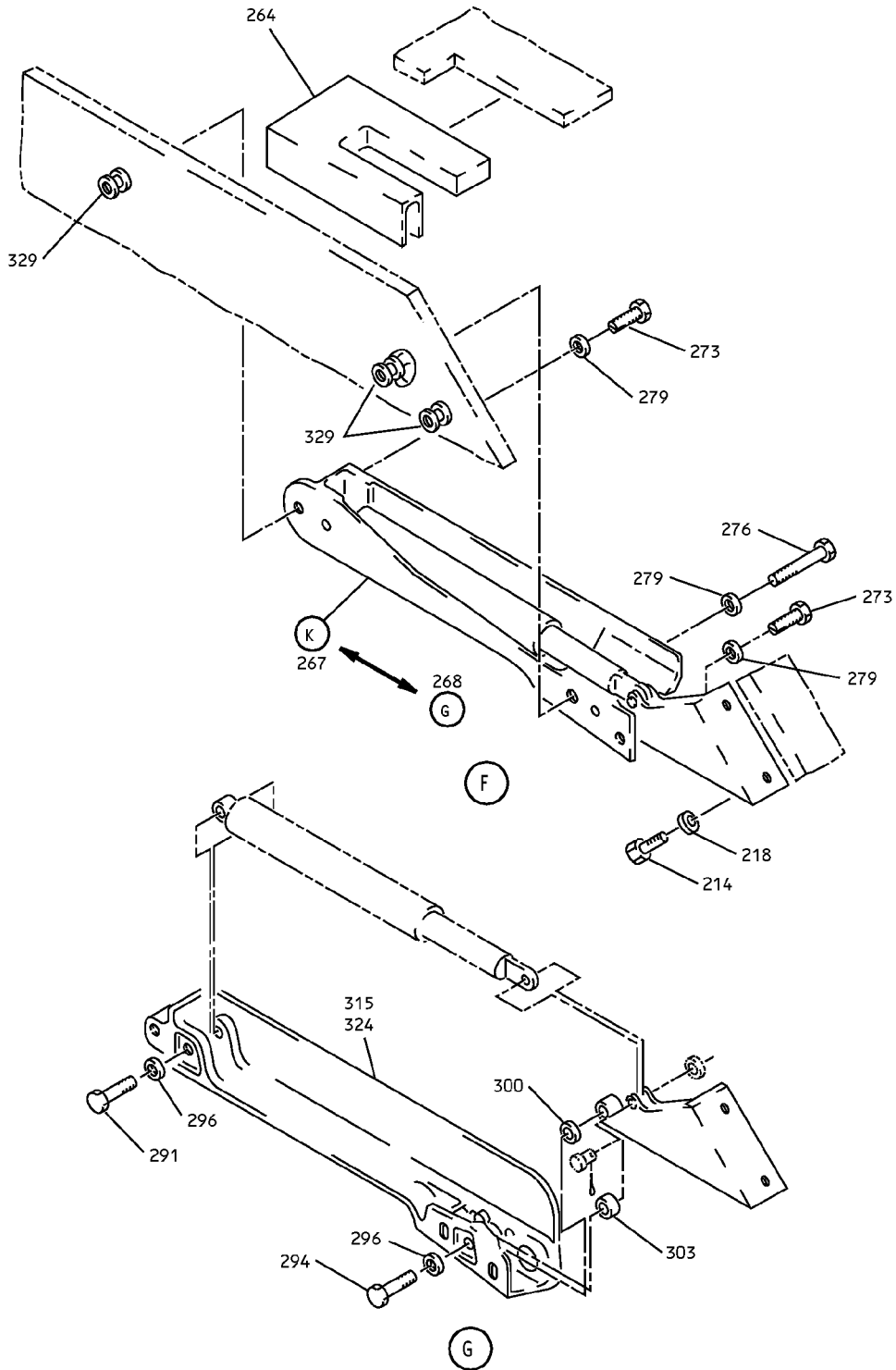
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Overhead Storage Compartment Assembly
IPL Figure 1 (Sheet 4 of 10)

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Overhead Stowage Compartment Assembly
IPL Figure 1 (Sheet 5 of 10)

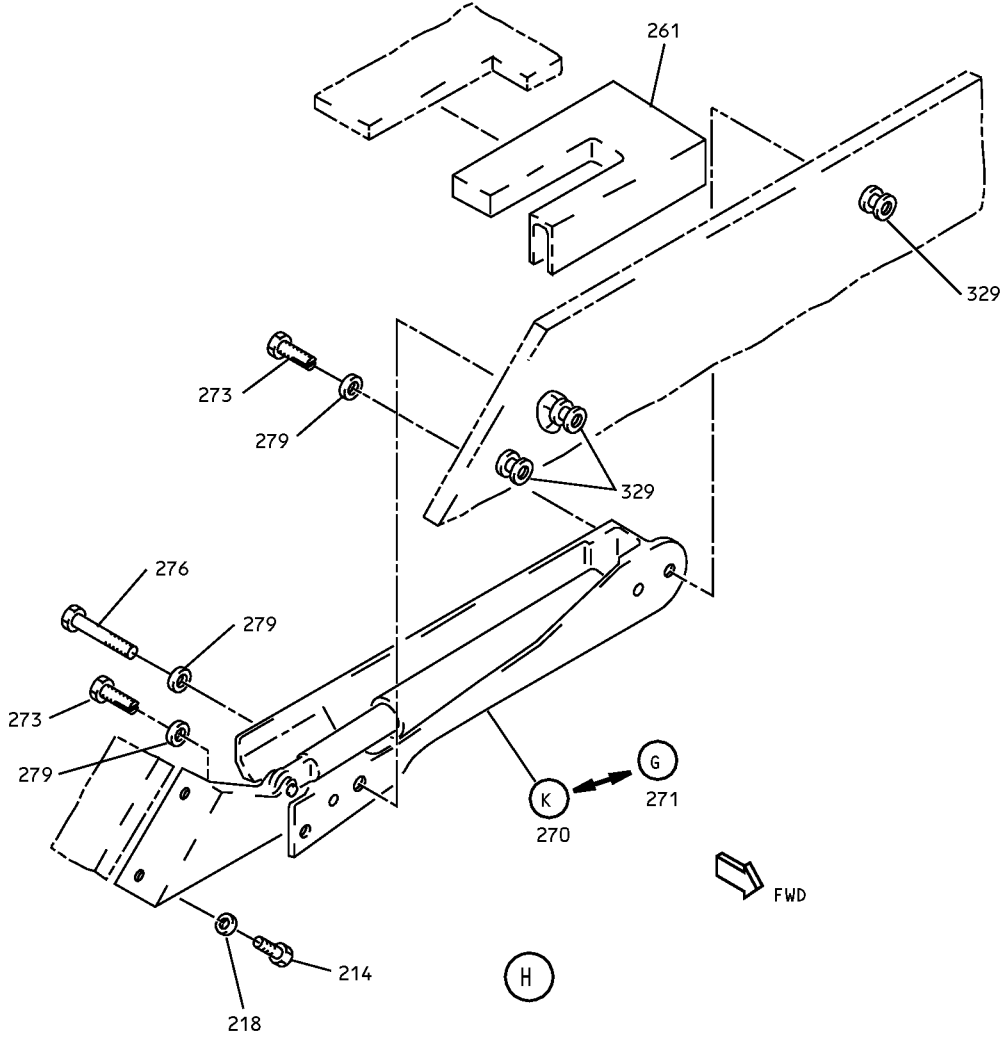
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Overhead Stowage Compartment Assembly
IPL Figure 1 (Sheet 6 of 10)

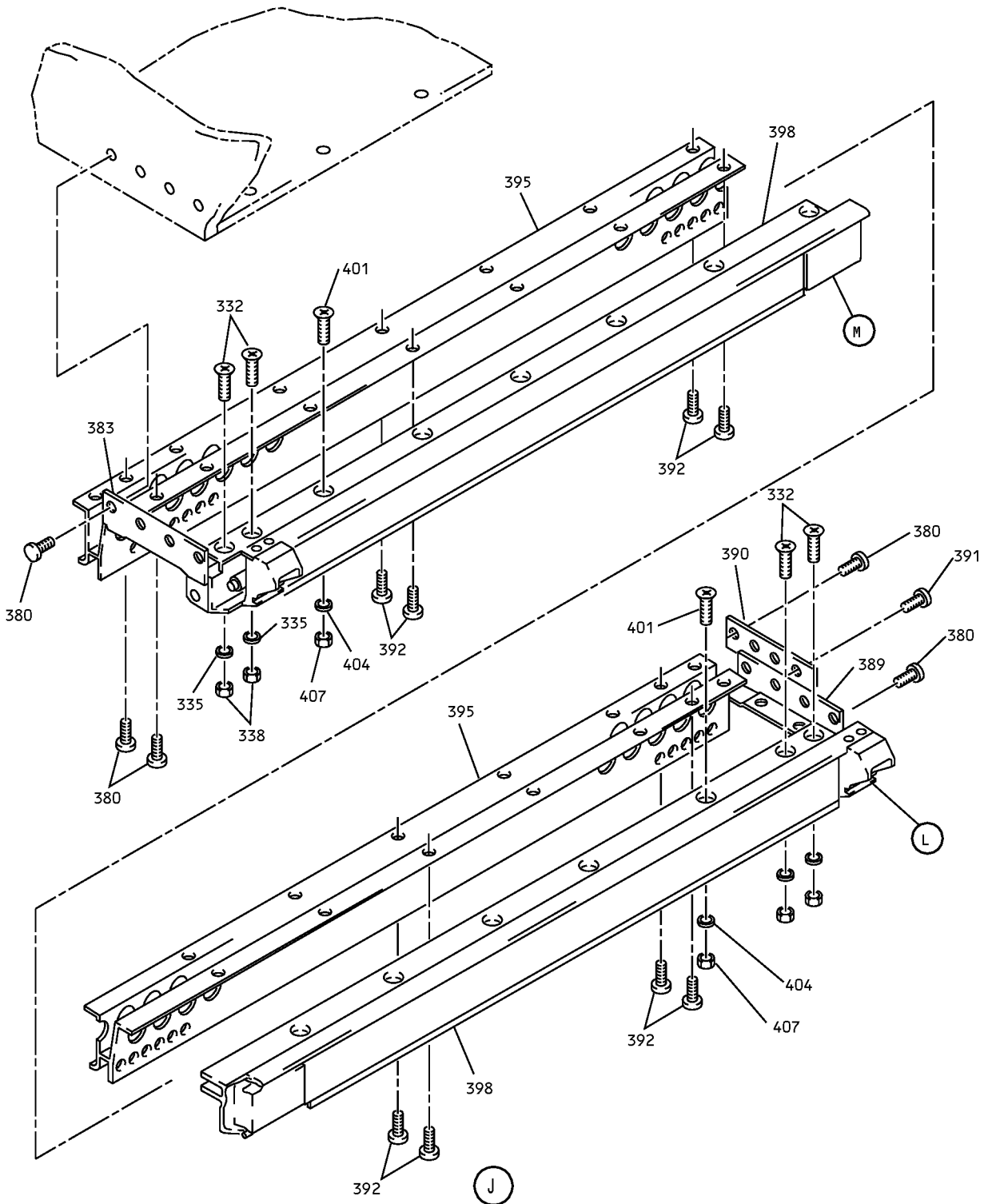
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Overhead Storage Compartment Assembly
IPL Figure 1 (Sheet 7 of 10)

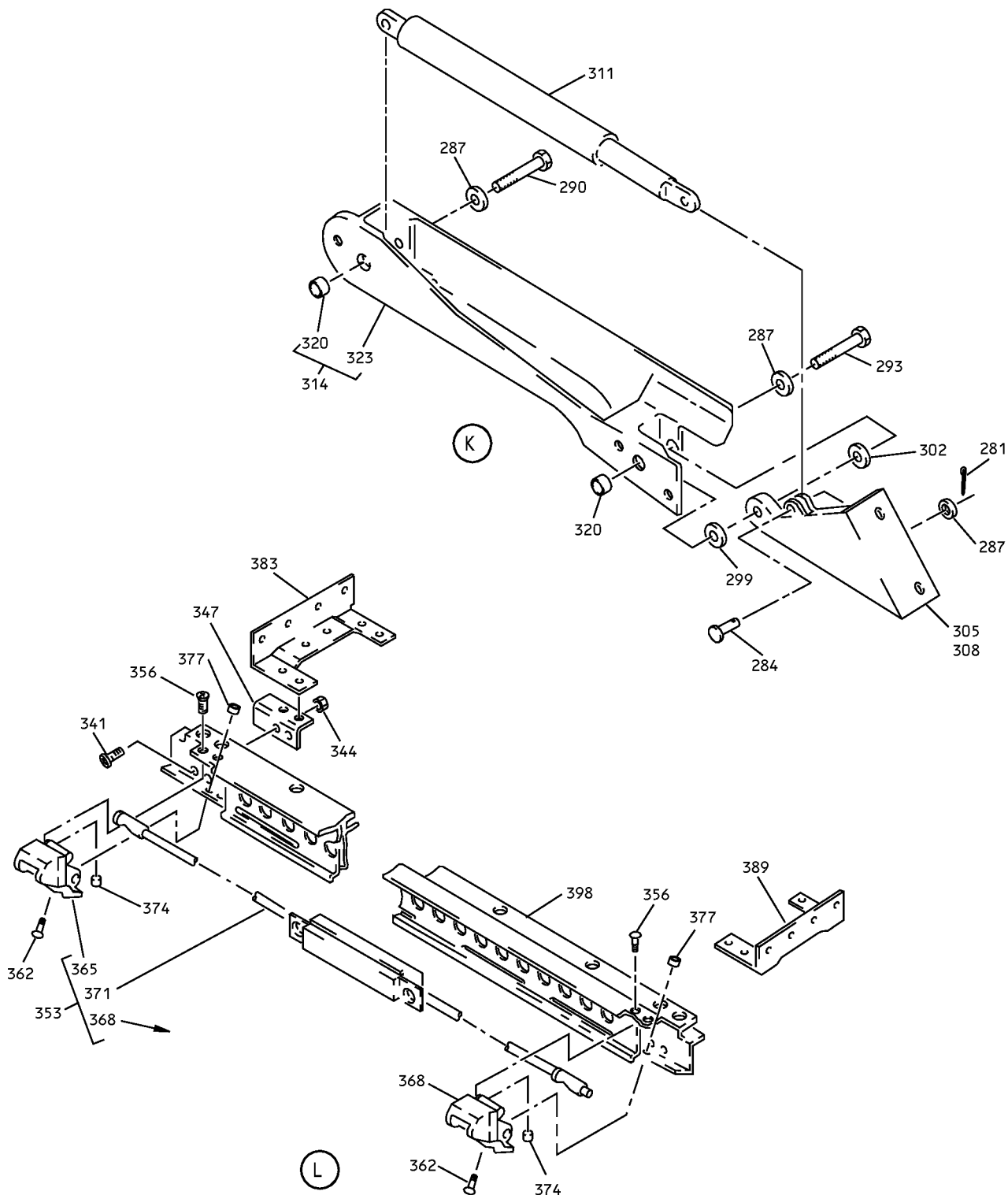
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IPL Figure 1 (Sheet 8 of 10)

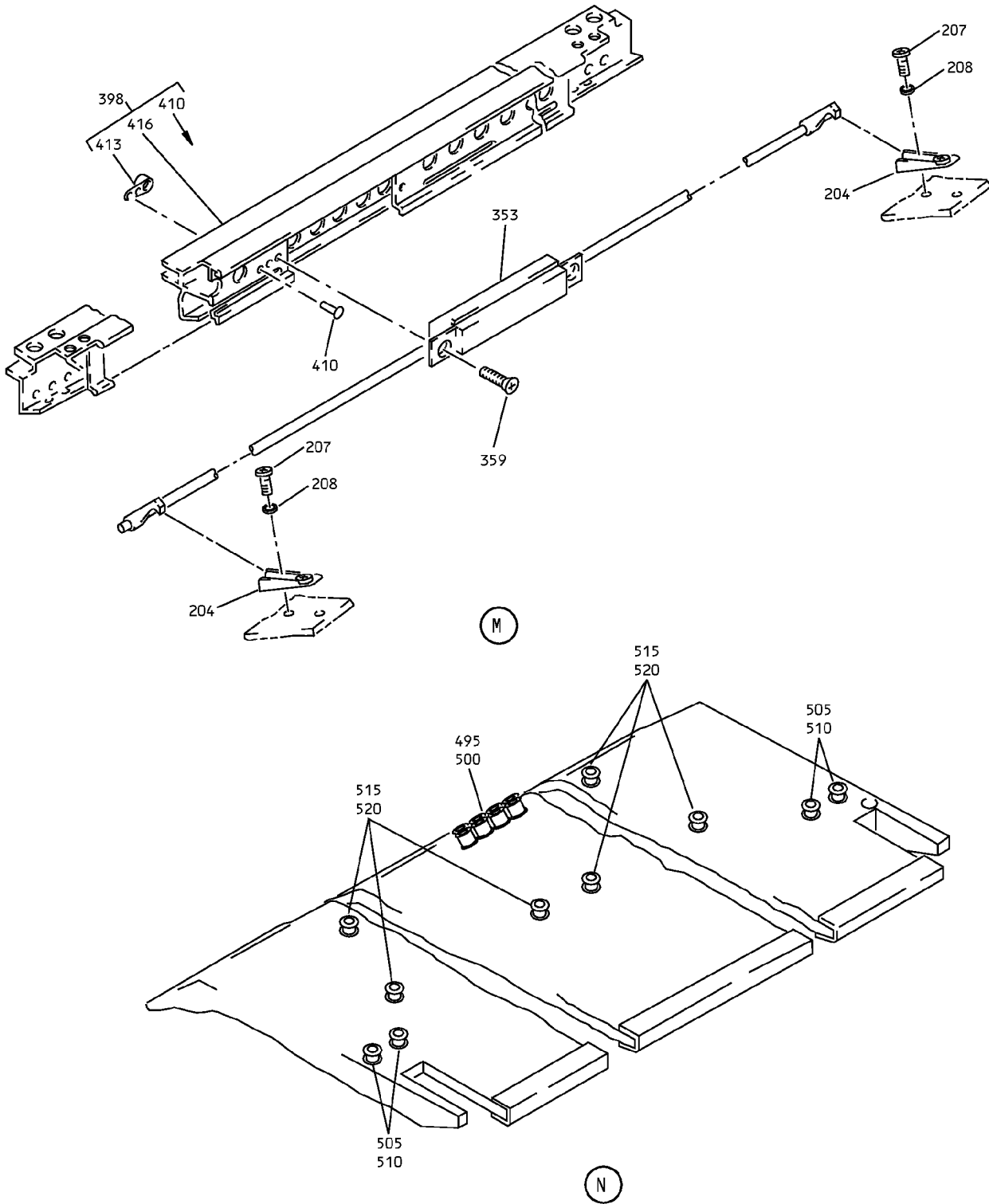
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IPL Figure 1 (Sheet 9 of 10)

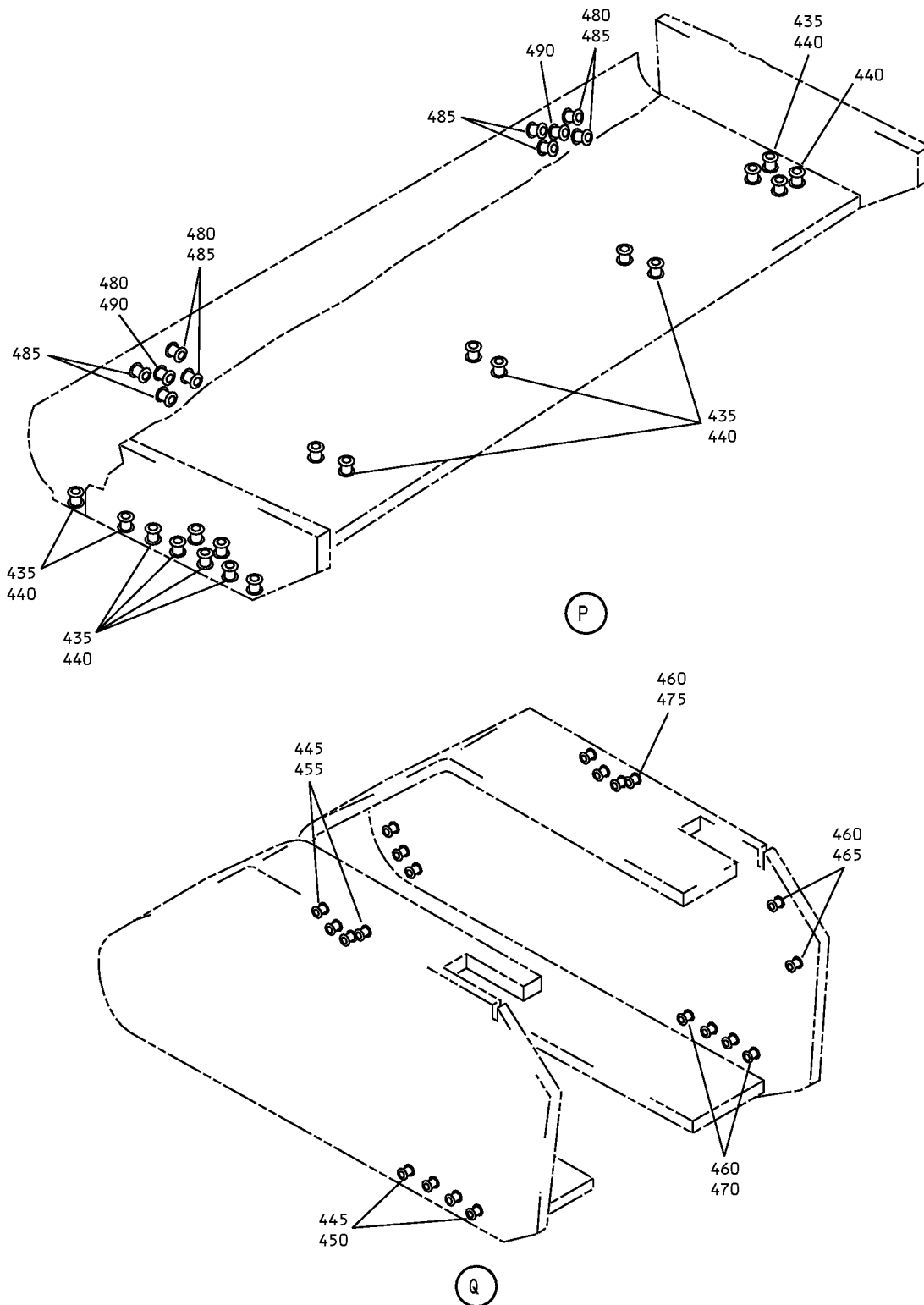
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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1	65C27005-76										RF
3	BACB30NM3K1										8
6	AN960PD10										8
9	412N1119-14										1
12	417N5120-20										2
-15	417N5120-7										1
21	NAS1801-3-5										4
24	AN960PD10L										4
27	BACS12ER3K6										3
30	BACW10UD10										3

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
33	F22K1-02		.	.	NUT						3
					(V72962)						
					(SPEC BACN10R10L)						
					(OPT 22K1-02 (V72962))						
					(OPT 26CL02 (V80539))						
36	05939-7		.	.	BALLAST						1
					(V17023)						
					(SPEC S283N702-7)						
39	417N5120-12		.	.	LAMP ASSY						1
42	2055132		.	.	CAP-END						2
					(V04169)						
45	F15T8WW		.	.	LAMP-FLOUR.						1
					(V08805)						
48	LP1610-18		.	.	TUBE-COVER						1
					(V60116)						
51	65C28117-3		.	.	PAD						1
54	800000255-17		.	.	LIGHT ASSY						1
					(V05617)						
					(SPEC S417N507-17)						
					(USED ON ITEM 12)						
					(FOR DETAILS SEE FIG. 80)						
-54A	800000255-7		.	.	LIGHT ASSY						1
					(V05617)						
					(SPEC S417N507-7)						
					(USED ON ITEM 15)						
					(FOR DETAILS SEE FIG. 80)						
60	NAS1801-3-11		.	.	SCREW						1
63	BACS18C8-6		.	.	SPACER						1
66	MS24585-1230		.	.	SPRING						1
69	411N3142-28		.	.	WASHER-CUPPED						1
78	AN960PD10		.	.	WASHER						4
81	412N1119-1		.	.	FITTING ASSY						1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-			ATTACHING PARTS								
87	BACS12ER3K6		.								4
93	BACB28B6-220P		.	.							1
99	412N1119-2		.	.							1
105	412N1119-62		.								
108	BACS12ER3K6		.								4
117	412N1119-12		.								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1-												
204	H3-1152-1											2
207	BACS12FA3K4											4
208	MS35333-39											4
210	412N1140-42											1
214	NAS1801-3-4											4
218	AN960PD10L											4
222	SL618-3A											8
226	BACS12H4-8											2
230	NAS620A8L											2
232	412N1245-1											2
236	MS51862-4											2
240	412N1143-13											1
244	412N1143-14											1
246	412N1140-59											1
247	412N1140-13											1
248	412N1141-2											1
249	412N1141-8											1
250	412N1142-16											1
255	412N1118-1											1
258	412N1118-2											1
261	412N1154-1											1
264	412N1154-2											1
267	412N1155-1											1
268	412N1155-7											1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1- 270	412N1155-2		. .									1
271	412N1155-8		. .									1
273	412N1227-1		. .									4
-273A	AIC1321-3-8		. .									4
276	BACB30LK3-18		. .)		2
-276A	NAS1801-3-10		. .)		2
279	BACW10P136AL		. .									6
281	MS24665-132		. . .									1
284	MS20392-2C13		. . .									1
287	AN960PD10L		. . .									2
290	NAS6703PU15		. . .									1
291	BACB30NM3K12		. . .									1
293	NAS6703PU11		. . .									1
290	BACB30NM3K8		. . .									1
296	AN960PD10L		. . .									2
299	NAS43DD3-5		. . .									1
300	NAS43DD3-3		. . .									1
302	NAS43DD3-6		. . .									1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
303	NA43DD3-8		. . .	SPACER							1
				(USED ON ITEM 268, 271)							
305	H2948-2		. . .	HINGE							1
				(V83014)							
				(SPEC S412N102-2)							
				(USED ON ITEM 267)							
-308	H2948-1		. . .	HINGE							1
				(V83014)							
				(SPEC S412N102-1)							
				(USED ON ITEM 270)							
311	SB113		. . .	ACTUATOR							1
				(V24984)							
				(SPEC S412N104-1)							
314	412N1156-1		. . .	CHASSIS ASSY							1
				(USED ON ITEM 267)							
315	412N1156-13		. . .	CHASSIS ASSY							1
				(USED ON ITEM 268)							
-317	412N1156-2		. . .	CHASSIS ASSY							1
				(USED ON ITEM 270)							
-318	412N1156-14		. . .	CHASSIS ASSY							1
				(USED ON ITEM 271)							
320	6041-3BRX225		INSERT-DODGE							2
				(V01556)							
323	412N1156-3		CHASSIS							1
				(USED ON ITEM 314)							
324	412N1156-15		CHASSIS							1
				(USED ON ITEM 315)							
-326	412N1156-4		CHASSIS							1
				(USED ON ITEM 317)							
-327	412N1156-16		CHASSIS							1
				(USED ON ITEM 318)							
329	NAS1833-3-400		. .	INSERT							6
332	BACS12ER08K14		. .	SCREW							4
335	AN960PD8		. .	WASHER							4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1- 338	H10-08BAC		. . NUT (V15653) (SPEC BACN10JC08) (OPT NS202101-82 (V80539)) (OPT RMLH9075-82W (V72962)) (OPT T6S832J (V71087)) (OPT VN303A82 (V92215)) (OPT 96-82 (V80539)) (OPT BRH10A08 (V52828))								4
341	HL10VAZ6-2		. . BOLT (V56878) (SPEC BACB30MY6K2) (OPT B30MY6K2 (V97928)) (OPT HL10VAZ6-2 (V73197)) (OPT HL10VAZ6-2 (V92215)) (OPT HL10VAZ6-2 (V97928)) (OPT L8006K2 (V06725))								4
344	H10-3BAC		. . NUT (V15653) (SPEC BACN10JC3) (OPT NS202101-02 (V80539)) (OPT RMLH9075-3W (V72962)) (OPT T6S1032J (V71087)) (OPT VN303A02 (V92215)) (OPT 96-02 (V80539)) (OPT BRH10A3 (V52828))								4
347	412N1117-1		. . FITTING								1
353	412N1153-102		. . LATCH ASSY								1
			ATTACHING PARTS								
356	BACS12ER06K8		. . SCREW (V06710) (SPEC BACS12ER06K8) (OPT BACS12ER06K8 (V06725)) (OPT BACS12ER06K8 (V06950)) (OPT BACS12ER06K8 (V27624)) (OPT BACS12ER06K8 (V34252)) (OPT BACS12ER06K8 (V56878)) (OPT BACS12ER06K8 (V73197)) (OPT BACS12ER06K8 (V80539)) (OPT BACS12ER06K8 (V92215)) (OPT BACS12ER06K8 (V93907)) (OPT BACS12ER06K8 (V97928))								4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
359	BACS12ER06K8		. . SCREW (V06710) (SPEC BACS12ER06K8) (OPT BACS12ER06K8 (V06725)) (OPT BACS12ER06K8 (V06950)) (OPT BACS12ER06K8 (V27624)) (OPT BACS12ER06K8 (V34252)) (OPT BACS12ER06K8 (V56878)) (OPT BACS12ER06K8 (V73197)) (OPT BACS12ER06K8 (V80539)) (OPT BACS12ER06K8 (V92215)) (OPT BACS12ER06K8 (V93907)) (OPT BACS12ER06K8 (V97928))								2
362	NAS514P832-5		. . SCREW -----*-----								4
365	412N1113-7		. . . COVER								1
368	412N1113-8		. . . COVER								1
371	H3-1145-1-4760		. . . LATCH ASSY (V34941) (SPEC S412N101-1050)								1
374	SL210-06-2		. . INSERT (V97393)								4
377	SL210-08-1		. . INSERT (V97393)								4
380	BACS12FA3K5		. . SCREW (V06710) (SPEC BACS12FA3K5) (OPT BACS12FA3K5 (V06725)) (OPT BACS12FA3K5 (V06950)) (OPT BACS12FA3K5 (V27624)) (OPT BACS12FA3K5 (V34252)) (OPT BACS12FA3K5 (V56878)) (OPT BACS12FA3K5 (V73197)) (OPT BACS12FA3K5 (V80539)) (OPT BACS12FA3K5 (V92215)) (OPT BACS12FA3K5 (V93907)) (OPT BACS12FA3K5 (V97928))								16
383	412N1145-27		. . BRACKET								1
389	412N1145-28		. . BRACKET								1
390	412N1145-69		. . STRAP								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1- 391	BACS12ER3K6		. . SCREW (V06710) (SPEC BACS12ER3K6) (OPT BACS12ER3K6 (V06725)) (OPT BACS12ER3K6 (V06950)) (OPT BACS12ER3K6 (V27624)) (OPT BACS12ER3K6 (V34252)) (OPT BACS12ER3K6 (V56878)) (OPT BACS12ER3K6 (V73197)) (OPT BACS12ER3K6 (V80539)) (OPT BACS12ER3K6 (V92215)) (OPT BACS12ER3K6 (V93907)) (OPT BACS12ER3K6 (V97928))								2
392	BACS12FA3K5		. . SCREW (V06710) (SPEC BACS12FA3K5) (OPT BACS12FA3K5 (V06725)) (OPT BACS12FA3K5 (V06950)) (OPT BACS12FA3K5 (V27624)) (OPT BACS12FA3K5 (V34252)) (OPT BACS12FA3K5 (V56878)) (OPT BACS12FA3K5 (V73197)) (OPT BACS12FA3K5 (V80539)) (OPT BACS12FA3K5 (V92215)) (OPT BACS12FA3K5 (V93907)) (OPT BACS12FA3K5 (V97928))								26
395	65C27023-16		. . SUPPORT-PASS. SVCE UNIT								1
398	412N1146-2		. . SUPPORT ASSY-BULLNOSE								
			ATTACHING PARTS								
401	BACS12ER08K12		. . SCREW (V06710) (SPEC BACS12ER08K12) (OPT BACS12ER08K12 (V06725)) (OPT BACS12ER08K12 (V06950)) (OPT BACS12ER08K12 (V27624)) (OPT BACS12ER08K12 (V34252)) (OPT BACS12ER08K12 (V56878)) (OPT BACS12ER08K12 (V73197)) (OPT BACS12ER08K12 (V80539)) (OPT BACS12ER08K12 (V92215)) (OPT BACS12ER08K12 (V93907)) (OPT BACS12ER08K12 (V97928))								11
404	AN960PD8		. . WASHER								11

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1- 407	H10-08BAC		. . NUT (V15653) (SPEC BACN10JC08) (OPT NS202101-82 (V80539)) (OPT RMLH9075-82W (V72962)) (OPT T6S832J (V71087)) (OPT VN303A82 (V92215)) (OPT 96-82 (V80539)) (OPT BRH10A08 (V52828))								11
410	BACR15DR3		----- * ----- . . . RIVET (SIZE DETERMINED ON INST)								4
413	BR1000A08		. . . NUTPLATE (V52828) (SPEC BACN10KB08) (OPT K2000-08 (V15653)) (OPT RMA9207-82 (V72962)) (OPT T8109S832 (V11815)) (OPT VN101A1-82 (V92215)) (OPT K2000-08 (V15653))								2
416	412N1146-22		. . . SUPPORT								1
430	65C27011-30		. . BODY ASSY								1
435	65C27011-1008		. . . PATTERN ASSY-INSERT								1
440	NAS1836-3-11	 INSERT								19
445	65C27011-1207		. . . PATTERN ASSY-INSERT								1
450	NAS1836-3-11	 INSERT								4
455	SL2334-3A2	 INSERT (V97393)								4
460	65C27011-1210		. . . PATTERN ASSY-INSERT								1
465	NAS1834-4-400	 INSERT								2
470	NAS1836-3-11	 INSERT								4
475	SL2334-3A2	 INSERT (V97393)								4
480	65C27011-1274		. . . PATTERN ASSY-INSERT								1
485	NAS1836-3-11	 INSERT								8
490	NAS1836-4-11	 INSERT								2
495	65C27011-1322		. . . PATTERN ASSY-INSERT								1
500	NAS1836-3-11	 INSERT								8

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
505	65C27011-2007										1
510	NAS1836-3-11										4
515	65C27011-2107										1
520	NAS1836-3-11										6
525	412N1215-71										1

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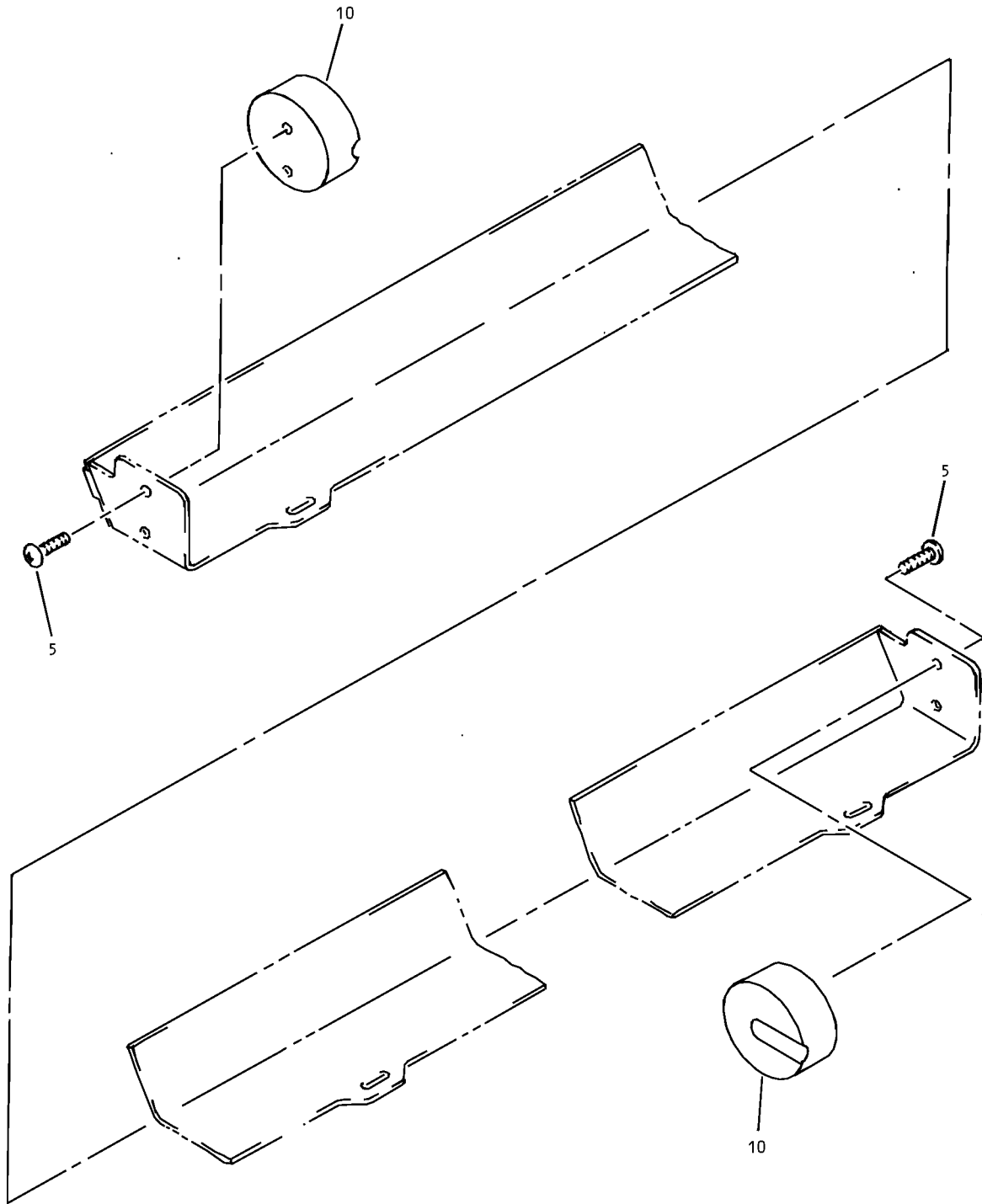
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Ceiling Light Assembly
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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
80- -1A	800000255-17		LIGHT ASSY-CLG (V05617) (SPEC S417N507-17) (THE ABOVE ASSEMBLY IS NOT BOEING MANUFACTURED. PARTS BREAKDOWN FOR REFERENCE ONLY. CONSULT VENDOR FOR ANY ADDITIONAL INFORMATION)							A	RF
-1B	800000255-7		LIGHT ASSY-CLG (V05617) (SPEC S417N507-7) (THE ABOVE ASSEMBLY IS NOT BOEING MANUFACTURED. PARTS BREAKDOWN FOR REFERENCE ONLY. CONSULT VENDOR FOR ANY ADDITIONAL INFORMATION)							B	RF
5	MS35206-323		. SCREW								4
10	BV03112-03-33		. HOLDER-BRUCE LAMP (V17023) (USED ON ITEM 1)								2
-10A	BV03112-33		. HOLDER-LAMP (V17023) (USED ON ITEM 1B)								2

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