COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

CARGO COMPARTMENT PROTECTIVE GRILL ASSEMBLY

PART NUMBER 65C21468–1, –10, –13, –15, –2, –9

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COMPONENT MAINTENANCE MANUAL

Revision No. 9
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To: All holders of CARGO COMPARTMENT PROTECTIVE GRILL ASSEMBLY 25-50-04.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

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A = Added, R = Revised, D = Deleted, O = Overflow

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TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL

All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed	
Number	Date	Date	Initials

Revision		Filed	
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Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials

Temporary Revision		Inserted		Removed	
Date	Initials	Number	Date	Date	Initials

Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials

Temporary Revision		Inserted		Removed	
Date	Initials	Number	Date	Date	Initials

INTRODUCTION

1. General

A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.

B. This manual is divided into different parts:

(1) Title Page

(2) Transmittal Letter

(3) Highlights

(4) List of Effective Pages

(5) Table of Contents

(6) Temporary Revision & Service Bulletin Record

(7) Record of Revisions

(8) Record of Temporary Revisions

(9) Introduction

(10) Procedures & IPL Sections

C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.

D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.

E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.

F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.

G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

CARGO COMPARTMENT PROTECTIVE GRILL ASSEMBLY - DESCRIPTION AND OPERATION

1. Description

A. The Cargo Compartment Protective Grill Assembly can be a welded tube frame with a welded wire grid. The Grill Assembly can also be a welded angle frame with a welded tube grid.

2. Operation

A. The Protective Grill is a part of the blowout panel assembly. The Grill prevents damage to the blowout panel caused by cargo movement when the airplane is in flight. The Grill also prevents damage to the blowout panel during the loading of the cargo. There is one grill assembly in the aft bulkhead of the forward cargo compartment. On some airplanes, there is also one grill assembly in the aft bulkhead of the aft cargo compartment.

3. Leading Particulars (approximate)

A. Length – 18 to 19 inches

B. Height – 13 to 15 inches

C. Width – 0.75 inch

D. Weight – 2 pounds

COMPONENT MAINTENANCE MANUAL

TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

(NOT APPLICABLE)

(NOT APPLICABLE)

CHECK

1. General

A. This procedure has the data necessary to find defects in the material of specified parts.

2. Check

A. Tools/Equipment

NOTE: Equivalent substitutes may be used.

Reference	Description
STD-1070	Lens - Magnifying, 10X, Hand Held

B. Procedure

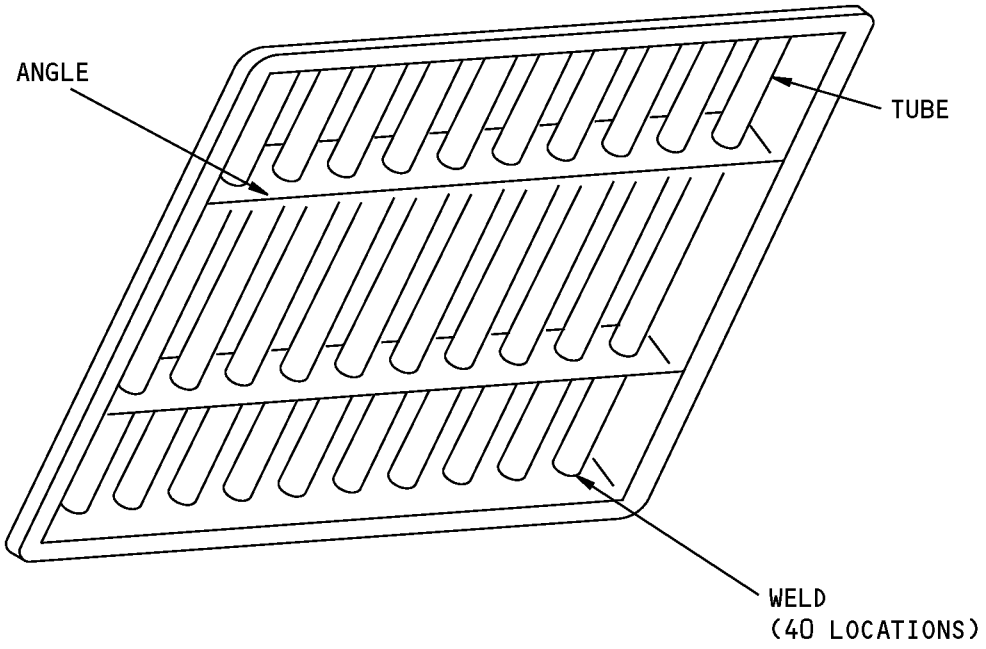
(1) Use standard industry practices to do a visual check of all the visible parts for defects.

NOTE: Refer to CHECK, Figure 501 for grill assembly check.

(a) Examine the grill assembly for missing tubes or wires.

(b) Examine the tubes for creases or buckling. Small bends in the tubes are permitted.

(c) Do a visual check for cracked or broken welds. If you think there is a cracked or broken weld, examine the area with a 10x hand held magnifying lens, STD-1070 to make sure.



Grill Assembly Check
Figure 501

REPAIR

1. General

A. Instructions for repair, refinish and replacement of the specified sub-assembly parts are included in each REPAIR when applicable:

Table 601:

PART NUMBER	NAME	REPAIR
———	REFINISH OF OTHER PARTS	1-1
65C21468-1,-2	GRILL ASSEMBLY	2-1
65C21468-9,-10,-13,-15	GRILL ASSEMBLY	3-1

2. Dimensional Symbols

A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.

—	STRAIGHTNESS
▱	FLATNESS
⊥	PERPENDICULARITY (OR SQUARENESS)
//	PARALLELISM
○	ROUNDNESS
⌭	CYLINDRICITY
⌒	PROFILE OF A LINE
⌓	PROFILE OF A SURFACE
◎	CONCENTRICITY
⌯	SYMMETRY
∠	ANGULARITY
↗	RUNOUT
↗↗	TOTAL RUNOUT
⌴	COUNTERBORE OR SPOTFACE
∨	COUNTERSINK
⌖	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)
∅	DIAMETER
S ∅	SPHERICAL DIAMETER
R	RADIUS
SR	SPHERICAL RADIUS
()	REFERENCE
BASIC (BSC) OR [DIM]	A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE. FROM THIS FEATURE PERMISSIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
[-A-]	DATUM
Ⓜ	MAXIMUM MATERIAL CONDITION (MMC)
Ⓛ	LEAST MATERIAL CONDITION (LMC)
Ⓢ	REGARDLESS OF FEATURE SIZE (RFS)
Ⓟ	PROJECTED TOLERANCE ZONE
FIM	FULL INDICATOR MOVEMENT

EXAMPLES

— | 0.002	STRAIGHT WITHIN 0.002
⊥ | 0.002 | B	PERPENDICULAR TO DATUM B WITHIN 0.002
// | 0.002 | A	PARALLEL TO DATUM A WITHIN 0.002
○ | 0.002	ROUND WITHIN 0.002
⌭ | 0.010	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER
⌒ | 0.006 | A	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM A
⌓ | 0.020 | A	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.020 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE
◎ | ∅ 0.0005 | C	CONCENTRIC TO DATUM C WITHIN 0.0005 DIAMETER
⌯ | 0.010 | A	SYMMETRICAL WITH DATUM A WITHIN 0.010
∠ | 0.005 | A	ANGULAR TOLERANCE 0.005 WITH DATUM A
⌖ | ∅ 0.002 Ⓢ | B	LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
⊥ | ∅ 0.010 Ⓜ | A 0.510 Ⓟ	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010 INCH DIAMETER, PERPENDICULAR TO DATUM A, AND EXTENDING 0.510 INCH ABOVE DATUM A, MAXIMUM MATERIAL CONDITION
[2.000] OR 2.000 BSC	THEORETICALLY EXACT DIMENSION IS 2.000

True Position Dimensioning Symbols
Figure 601

REFINISH OF OTHER PARTS - REPAIR 1-1

1. General

A. This procedure has the data necessary to refinish the parts, which are not given in the specified repairs.

B. Refer to the Standard Overhaul Practices Manual (SOPM) for details of the SOPM subjects identified in the procedure.

C. Refer to IPL Figure 1, IPL Figure 2, IPL Figure 3 and IPL Figure 4 for the item numbers.

2. Refinish of Other Parts

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C00260	Coating - Chemical And Solvent Resistant Finish, Epoxy Resin Enamel	BMS10-11, Type II
C00304	Coating - Teflon Filled, Non Decorative, Sprayable Material	BMS 10-86 Type I
C50033	Chromated Conversion Coating for Aluminum - Alodine 1200	

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

C. General

(1) Instructions for the repair of the parts listed in REPAIR 1-1, Table 601 are for repair of the initial finish.

D. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

(1) Refer to REPAIR 1-1, Table 601 for refinish details.

Table 601: Refinish Details

IPL FIG. & ITEM	MATERIAL	FINISH
IPL Fig. 1		
Grill Assembly (1A)	321/347 CRES	Apply primer, C00259(F-20.02). Apply abrasion resistant finish coating, C00304 (SRF-14.9624).
IPL Fig. 2		
Grill Assembly (1A)	321/347 CRES	Apply primer, C00259 (F-20.02). Apply abrasion resistant finish coating, C00304(SRF-14.9624).
IPL Fig. 3		
Grill Assembly (1A)	Al alloy	Chemical treat and apply primer, C00259 (F-18.06). Apply gloss enamel coating, C00260 (F-21.03).
Grill Assembly (1B)	Al alloy	Apply Alodine 1200 coating, C50033 to bare aluminum surfaces. Apply primer, C00259 (F-20.02). Apply gloss enamel coating, C00260 (F-21.25).
Grill Assembly (1C)	Al alloy	Apply chemical conversion coating to all surfaces (F-30.006). Apply primer, C00259 (F-20.02). Apply gloss enamel coating, C00260 (F-21.17).
IPL Fig. 4		
Grill Assembly (1A)	Al alloy	Apply Alodine 1200 coating, C50033 to bare aluminum surfaces. Apply primer, C00259 (F-20.02). Apply gloss enamel coating, C00260 (F-21.25).

GRILL ASSEMBLY - REPAIR 2-1

65C21468–1, –2

1. General

A. This procedure has the data necessary to repair the grill assembly (1A, IPL Figure 1; 1A, IPL Figure 2).

B. Refer to the Standard Overhaul Practices Manual (SOPM) for details of the SOPM subjects identified in the procedure.

C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.

D. Refer to IPL Figure 1 and IPL Figure 2 for the item numbers.

2. Wire (5, 10) Replacement

A. Tools/Equipment

NOTE: Equivalent substitutes may be used.

Reference	Description
STD-1070	Lens - Magnifying, 10X, Hand Held

B. References

Reference	Title
CMM 25-50-04	CARGO COMPARTMENT PROTECTIVE GRILL ASSEMBLY
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedures

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

(1) Remove the damaged wire as shown in REPAIR 2-1, Figure 601. Make sure the surface roughness is 63 microinch or smoother.

(2) Visually examine the grill assembly (1A, IPL Figure 1; 1A, IPL Figure 2) to make sure the damage was completely removed.

(3) Make a repair wire from CRES 321-347 and equal in length to the removed wire.

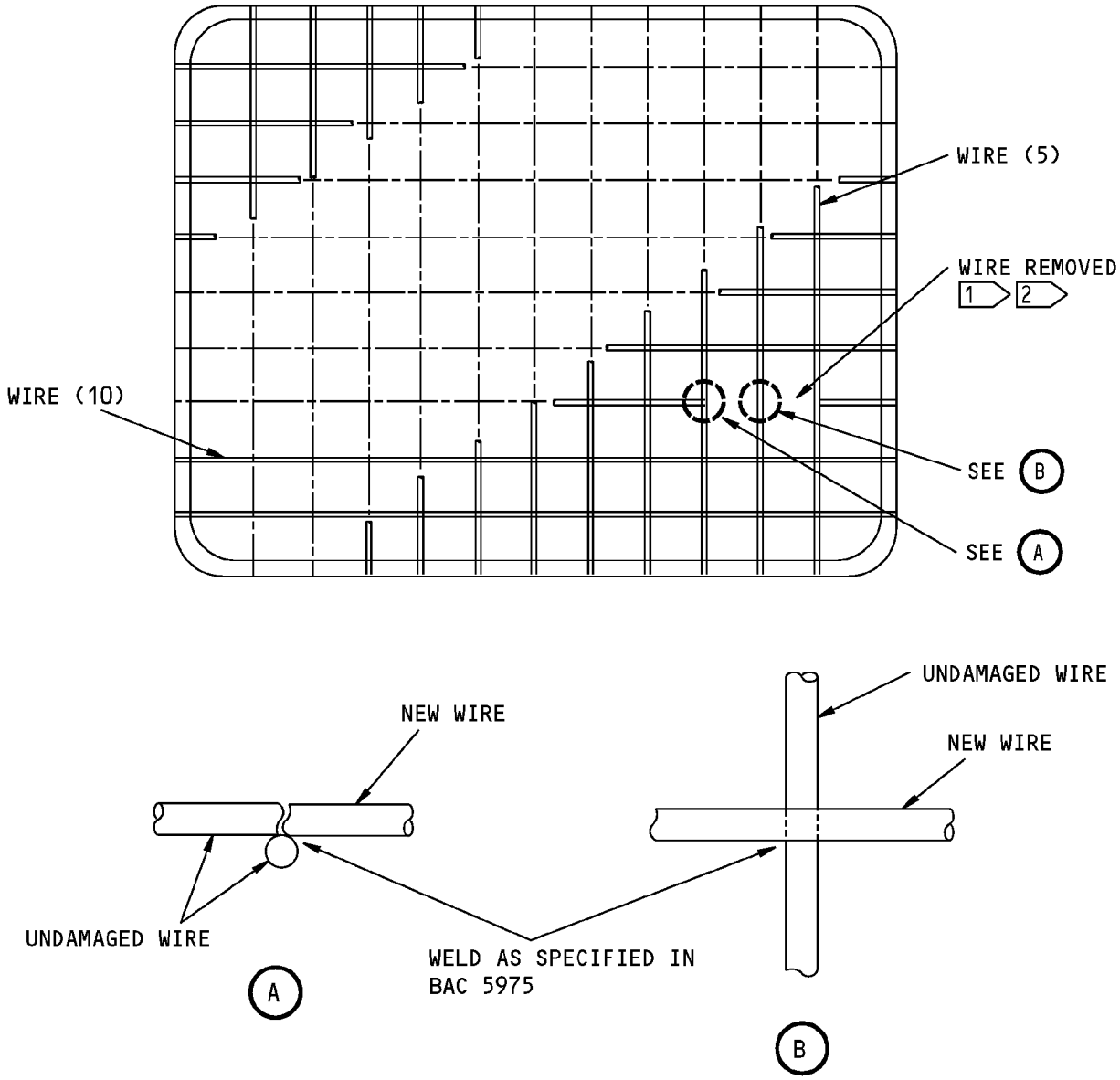
(4) Weld the repair wire in its position as specified in BAC5975, Class 5A or 6. Use MIL-R-5031 filler rod. Material: CRES 321-347.

(5) Do a visual inspection of the welded area with a 10x hand held magnifying lens, STD-1070 to make sure the weld is satisfactory.

D. Refinish

(1) Break sharp edges 0.020/0.030 R.

(2) Apply a finish as specified in CMM 25-50-04, Repair 1-1.



1 REMOVE DAMAGED WIRE COMPLETELY. MAKE SURE THE ENDS ARE REMOVED TO APPROXIMATELY HALF THE DIAMETER OF THE UNDAMAGED WIRE

2 MAKE SURE THE SURFACE ROUGHNESS IS 63 MICROINCH OR SMOOTHER

ITEM NUMBERS REFER TO IPL FIG. 1 AND 2

65C21468-1,-2 Wire Replacement
Figure 601

GRILL ASSEMBLY - REPAIR 3-1

65C21468–9, –10, –13, –15

1. General

A. This procedure has the data necessary to repair the grill assembly (1A, 1B, 1C, IPL Figure 3; 1A, IPL Figure 4).

B. Refer to the Standard Overhaul Practices Manual (SOPM) for details of the SOPM subjects identified in the procedure.

C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.

D. Refer to IPL Figure 3 and IPL Figure 4 for the item numbers.

2. Tube (40) Replacement

A. Tools/Equipment

NOTE: Equivalent substitutes may be used.

Reference	Description
STD-1070	Lens - Magnifying, 10X, Hand Held

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

(1) Remove the damaged tube as shown in REPAIR 3-1, Figure 601. Make sure the surface roughness is 63 microinch or smoother.

(2) Visually examine the grill assembly (1A, 1B, 1C, IPL Figure 3; 1A, IPL Figure 4) to make sure the damage was completely removed.

(3) Make a repair tube or rod from one of the materials that follow and equal in length to the removed tube.

(a) 0.25 inch OD X 0.028 inch wall 6061-T6 aluminum tubing as specified in WW-T-700/6.

(b) Solid 6061-T6 Aluminum, machined to 0.24 to 0.26 inch diameter.

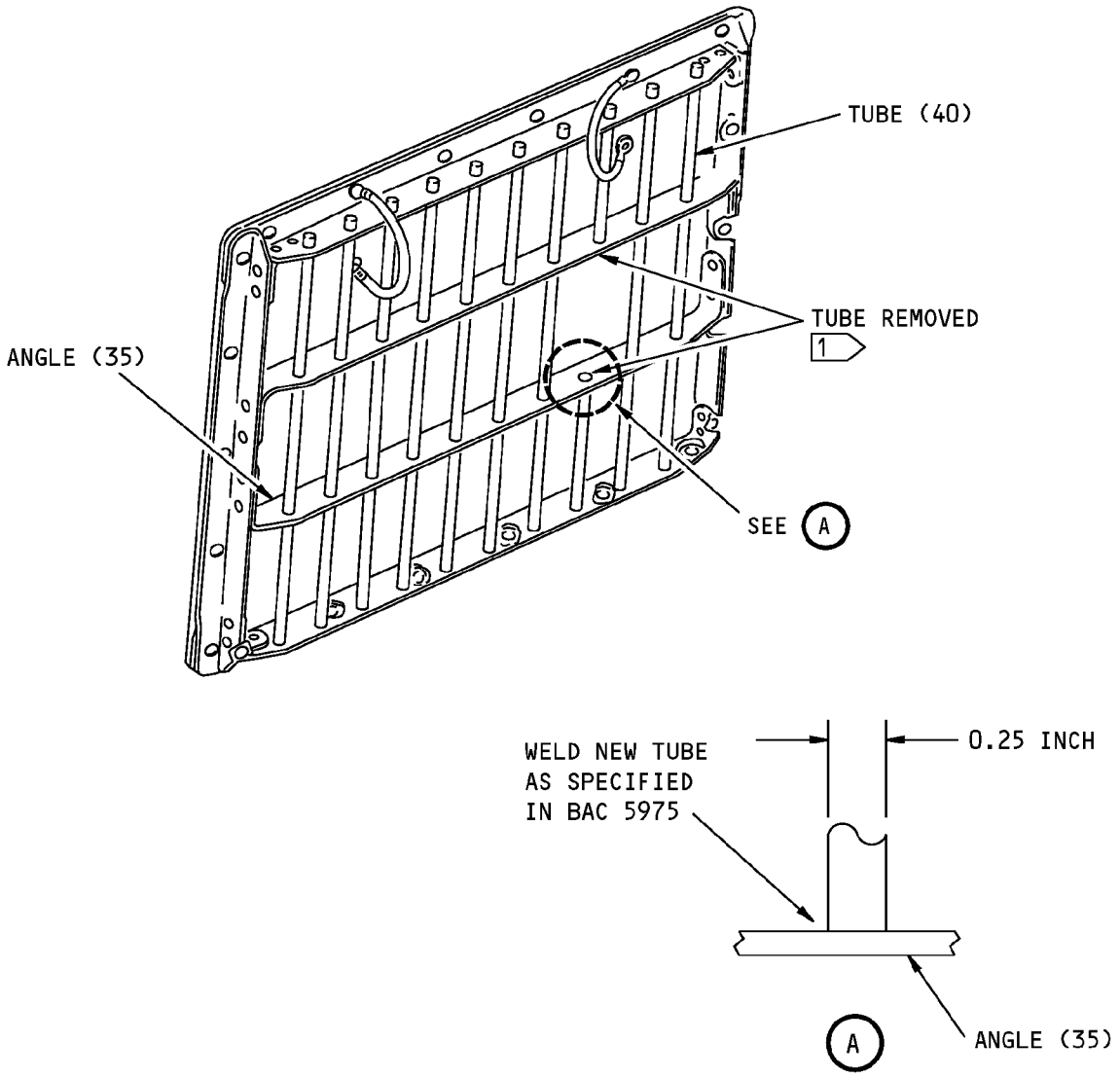
(4) Weld the repair tube or rod in its position as specified in BAC5975, Class B. Use AMS 4190 filler rod. Material: 6061-T6 Aluminum.

(5) Do a visual inspection of the welded area with a 10x hand held magnifying lens, STD-1070 to make sure the weld is satisfactory.

D. Refinish

(1) Break sharp edges 0.020/0.030 R.

(2) Apply a finish as specified in REPAIR 1-1.



1 REMOVE DAMAGED TUBE COMPLETELY. MAKE SURE THE SURFACE ROUGHNESS IS 63 MICROINCH OR SMOOTHER

ITEM NUMBERS REFER TO IPL FIG. 3 AND 4

65C21468-9,-10,-13,-15 Tube Replacement
Figure 601

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ASSEMBLY

(NOT APPLICABLE)

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FITS AND CLEARANCES

(NOT APPLICABLE)

(NOT APPLICABLE)

ILLUSTRATED PARTS LIST

1. Introduction

A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.

B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1 2 3 4 5 6 7

. Assembly

. Attaching parts for assembly

. . Detail parts for assembly

. . Subassembly

. . Attaching parts for subassembly

. . . Detail parts for subassembly

. . . Sub-subassembly

. . . Attaching parts for subassembly

. . . . . Details parts for sub-subassembly

Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.

D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.

E. Color-coded parts are identified with a single digit alpha following the dash number or with ''SP'' suffix. If the ''SP'' suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.

F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a ''~'' at the end of the line and will be continued on the next line. The ''~'' denotes that the part number continues on the next line.

G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.

(1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.

(2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.

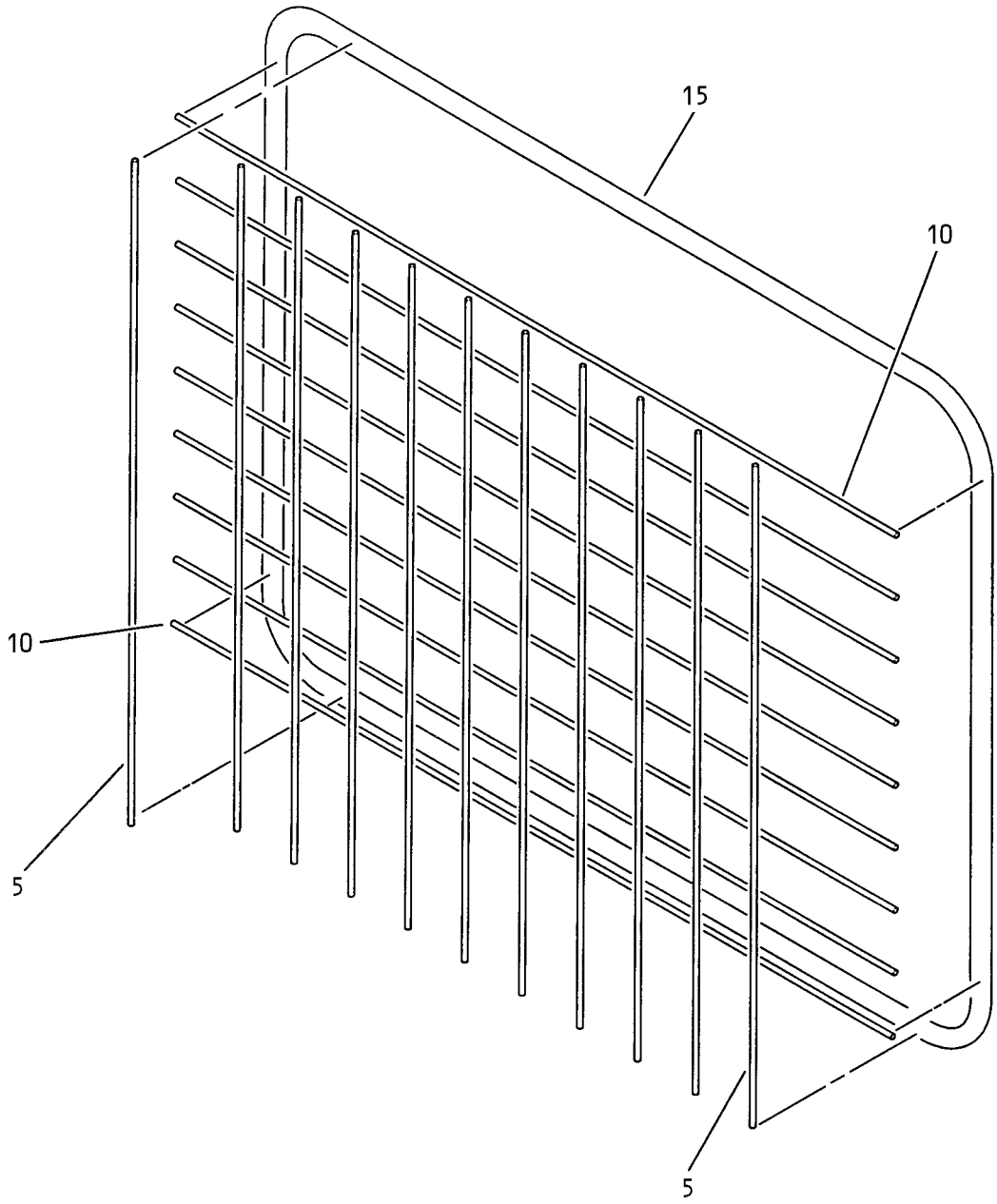
H. Interchangeable Parts

Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C21468-1		1	1A	RF
65C21468-10		1	1D	RF
		4	1A	RF
65C21468-11		3	25	1
65C21468-12		4	25	1
65C21468-13		1	1E	RF
		3	1B	RF
65C21468-14		3	25A	1
65C21468-15		1	1F	RF
		3	1C	RF
65C21468-2		1	1B	RF
		2	1A	RF
65C21468-3		1	15	1
65C21468-4		2	15	1
65C21468-5		1	5	11
65C21468-6		1	10	9
65C21468-7		2	5	11
65C21468-8		2	10	8
65C21468-9		1	1C	RF
		3	1A	RF
65C27524-1		3	15	2
		3	15B	2
65C27524-10		3	20	4
		3	20B	4
		4	20	4
65C27524-11		4	40	10
65C27524-12		3	15A	2
		3	15C	2
65C27524-13		3	20A	4
		3	20C	4
65C27524-2		3	30	2
65C27524-3		3	35	2
65C27524-4		3	40	10
65C27524-7		4	30	2

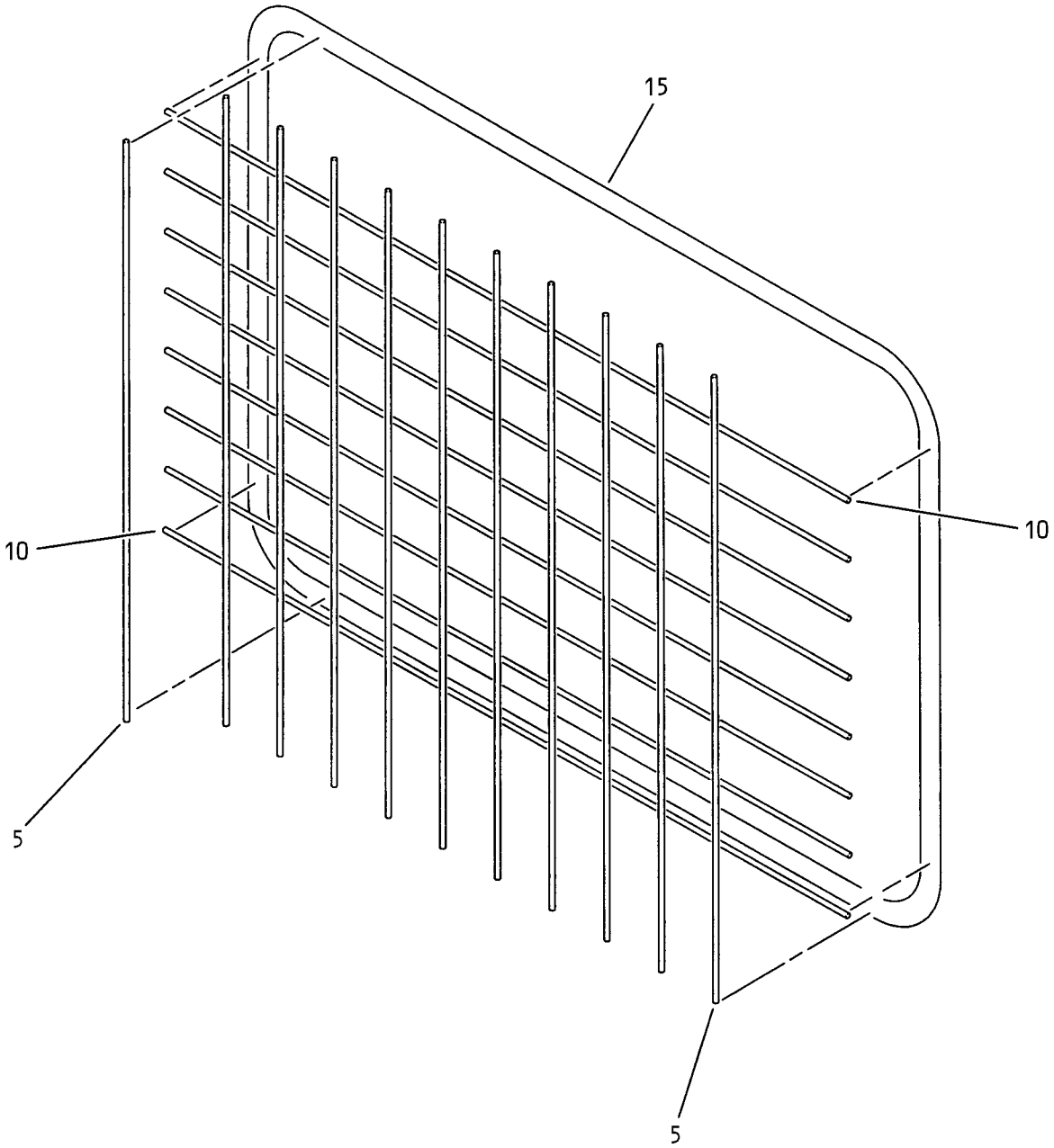
PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C27524-8		4	15	2
65C27524-9		4	35	2
BACR15FTD		3	10	24
		4	10	8
BACR15GF5		3	5	8
		4	5	24



Protective Grill Assembly
IPL Figure 1

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
–1A	65C21468-1		GRILL ASSY-PROTECTIVE, CARGO COMPT	A	RF
–1B	65C21468-2		GRILL ASSY-PROTECTIVE, CARGO COMPT (FOR DETAILS SEE FIG. 2)	B	RF
–1C	65C21468-9		GRILL ASSY-PROTECTIVE, CARGO COMPT (FOR DETAILS SEE FIG. 3)	C	RF
–1D	65C21468-10		GRILL ASSY-PROTECTIVE, CARGO COMPT (FOR DETAILS SEE FIG. 4)	D	RF
–1E	65C21468-13		GRILL ASSY-PROTECTIVE, CARGO COMPT (FOR DETAILS SEE FIG. 3)	E	RF
–1F	65C21468-15		GRILL ASSY-PROTECTIVE, CARGO COMPT (FOR DETAILS SEE FIG. 3)	F	RF
5	65C21468-5		. WIRE	A	11
10	65C21468-6		. WIRE	A	9
15	65C21468-3		. TUBE	A	1

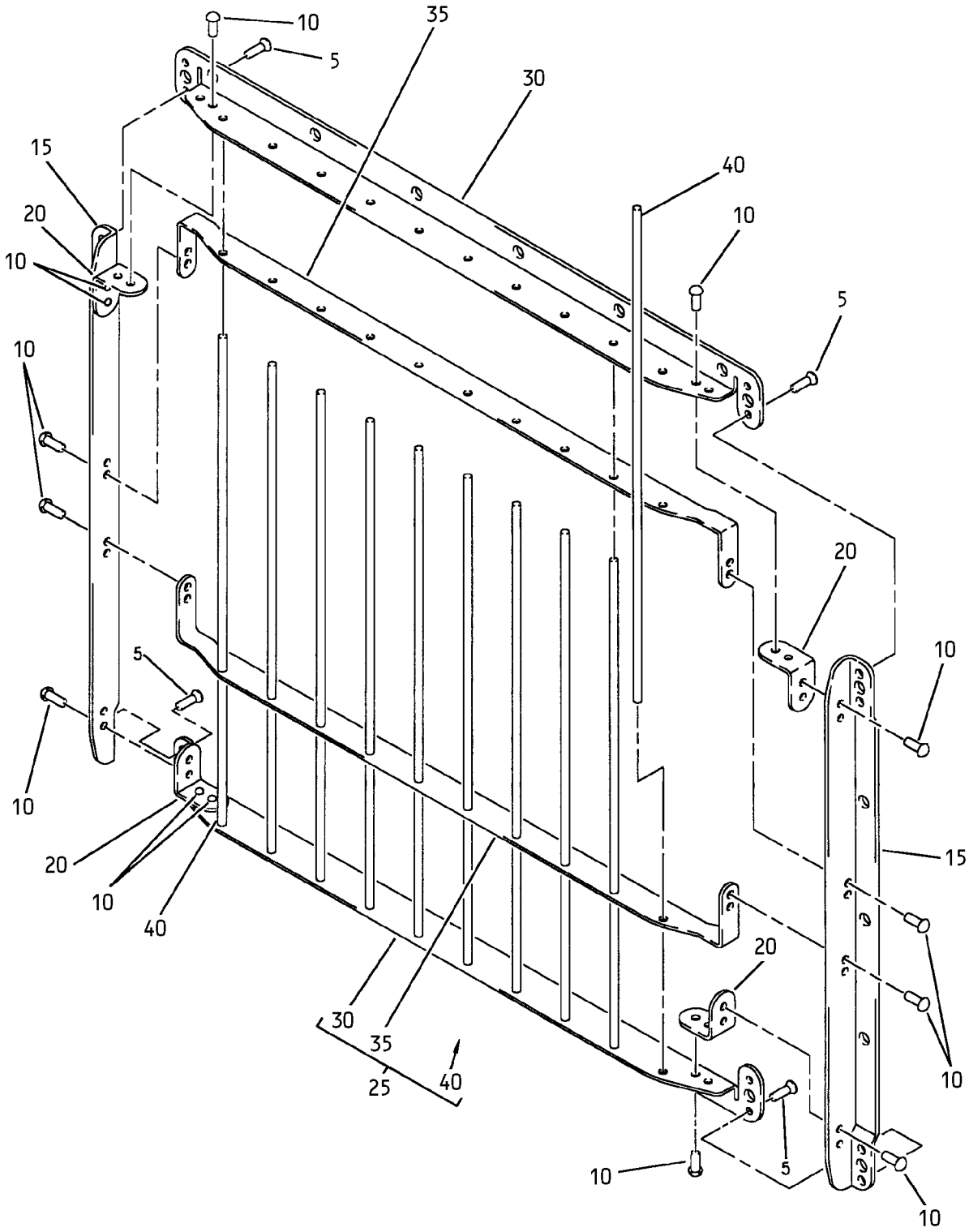
–Item not Illustrated



Protective Grill Assembly
IPL Figure 2

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
2–					
–1A	65C21468-2		GRILL ASSY-PROTECTIVE, CARGO COMPT	B	RF
5	65C21468-7		. WIRE	B	11
10	65C21468-8		. WIRE	B	8
15	65C21468-4		. TUBE	B	1

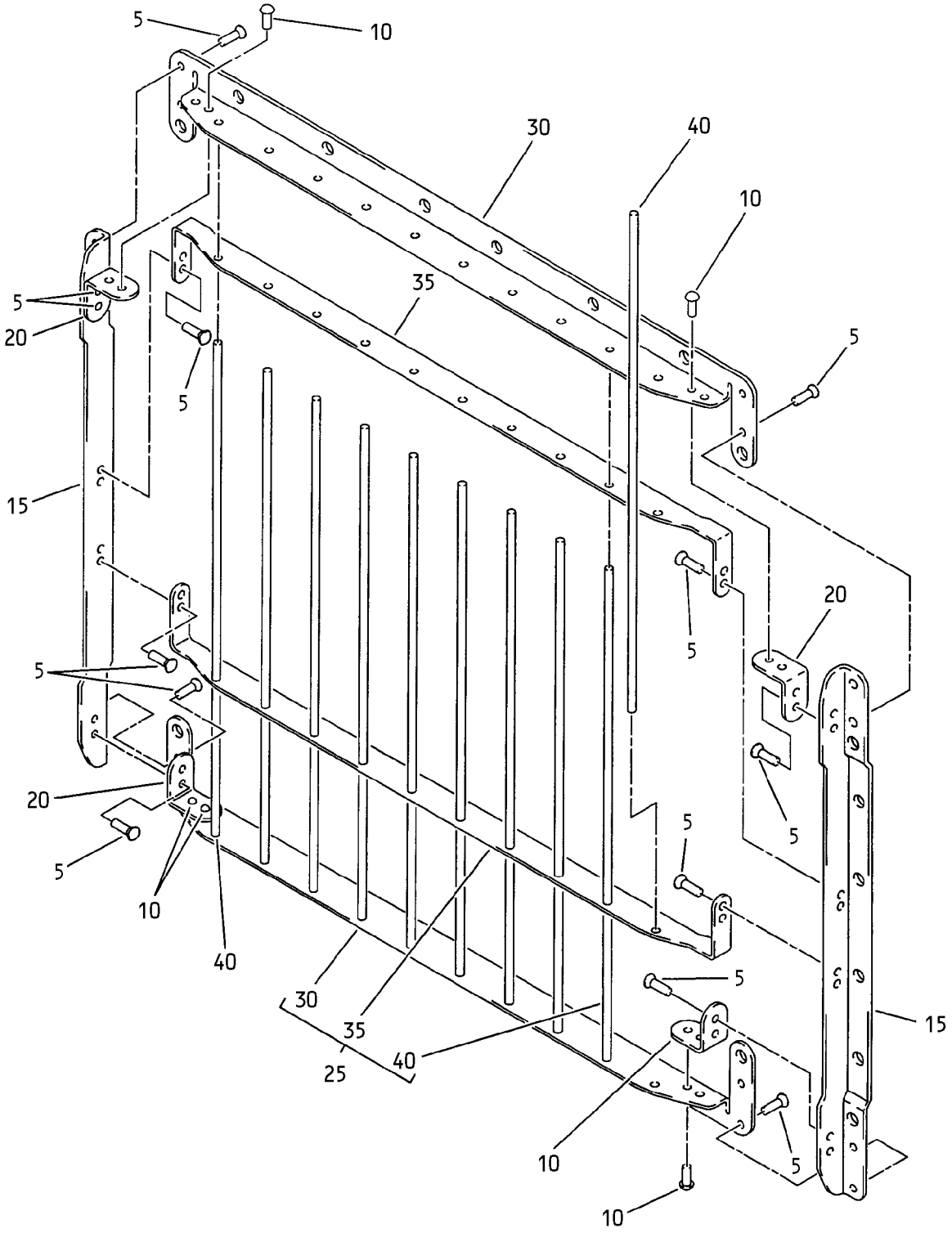
–Item not Illustrated



Protective Grill Assembly
IPL Figure 3

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
3–					
–1A	65C21468-9		GRILL ASSY-PROTECTIVE, CARGO COMPT	C	RF
–1B	65C21468-13		GRILL ASSY-PROTECTIVE, CARGO COMPT	E	RF
–1C	65C21468-15		GRILL ASSY-PROTECTIVE, CARGO COMPT	F	RF
5	BACR15GF5		. RIVET (SIZE DETERMINED ON INST)	C, E, F	8
10	BACR15FTD		. RIVET (SIZE DETERMINED ON INST)	C, E, F	24
15	65C27524-1		. ANGLE	C	2
–15A	65C27524-12		. ANGLE	E	2
–15B	65C27524-1		. ANGLE (OPT ITEM 015C)	F	2
–15C	65C27524-12		. ANGLE (OPT ITEM 015B)	F	2
20	65C27524-10		. ANGLE	C	4
–20A	65C27524-13		. ANGLE	E	4
–20B	65C27524-10		. ANGLE (OPT ITEM 020C)	F	4
–20C	65C27524-13		. ANGLE (OPT ITEM 020B)	F	4
25	65C21468-11		. TUBE ASSY	C, F	1
–25A	65C21468-14		. TUBE ASSY	E	1
30	65C27524-2		. . ANGLE	C, E, F	2
35	65C27524-3		. . ANGLE	C, E, F	2
40	65C27524-4		. . TUBE	C, E, F	10

–Item not Illustrated



Protective Grill Assembly
IPL Figure 4

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
4–					
–1A	65C21468-10		GRILL ASSY-PROTECTIVE, CARGO COMPT	D	RF
5	BACR15GF5		. RIVET (SIZE DETERMINED ON INST)	D	24
10	BACR15FTD		. RIVET (SIZE DETERMINED ON INST)	D	8
15	65C27524-8		. ANGLE	D	2
20	65C27524-10		. ANGLE	D	4
25	65C21468-12		. TUBE ASSY	D	1
30	65C27524-7		. . ANGLE	D	2
35	65C27524-9		. . ANGLE	D	2
40	65C27524-11		. . TUBE	D	10

–Item not Illustrated