



# **COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST**

## **LE AUXILIARY SLAT TRACK ASSEMBLY**

### **PART NUMBER**

**65-52144-2, 65-52145-2, 65-52146-2, 65-67173-1,  
65-67174-1, -3, -5, -6, 65-67175-1, -3, -5, -6,  
65-68243-1, 65-68275-1, 65-68276-1, 65-80984-1, -3,  
-5, -6, 65-80990-1, -2**

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## COMPONENT MAINTENANCE MANUAL

Revision No. 8  
Jul 01/2009

To: All holders of LE AUXILIARY SLAT TRACK ASSEMBLY 27-81-52.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

### ATTENTION

IF YOU RECEIVE PRINTED REVISIONS, PLEASE VERIFY THAT YOU HAVE RECEIVED AND FILED THE PREVIOUS REVISION. BOEING MUST BE NOTIFIED WITHIN 30 DAYS IF YOU HAVE NOT RECEIVED THE PREVIOUS REVISION. REQUESTS FOR REVISIONS OTHER THAN THE PREVIOUS REVISION WILL REQUIRE A COMPLETE MANUAL REPRINT SUBJECT TO REPRINT CHARGES SHOWN IN THE DATA AND SERVICES CATALOG.

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TRANSMITTAL LETTER  
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Location of Change

Description of Change

NO HIGHLIGHTS

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A = Added, R = Revised, D = Deleted, O = Overflow

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**TEMPORARY REVISION AND SERVICE BULLETIN RECORD**

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737-57-1243			NOV 01/98

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TR AND SB RECORD

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**COMPONENT MAINTENANCE MANUAL**

All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed		Revision		Filed	
Number	Date	Date	Initials	Number	Date	Date	Initials







**COMPONENT MAINTENANCE MANUAL**

All temporary revisions to this manual will be accompanied by a cover sheet bearing the temporary revision number. Enter the temporary revision number in numerical order, together with the temporary revision date, the date the temporary revision is inserted and the initials of the person filing. When the temporary revision is incorporated or cancelled, and the pages are removed, enter the date the pages are removed and the initials of the person who removed the temporary revision.

Temporary Revision		Inserted		Removed		Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials





## COMPONENT MAINTENANCE MANUAL

### INTRODUCTION

#### 1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
  - (1) Title Page
  - (2) Transmittal Letter
  - (3) Highlights
  - (4) List of Effective Pages
  - (5) Table of Contents
  - (6) Temporary Revision & Service Bulletin Record
  - (7) Record of Revisions
  - (8) Record of Temporary Revisions
  - (9) Introduction
  - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

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INTRODUCTION

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## COMPONENT MAINTENANCE MANUAL

### LEADING EDGE AUXILIARY SLAT TRACK ASSEMBLIES - DESCRIPTION AND OPERATION

1. \_

- A. The leading edge auxiliary slat track assemblies are the tracks used to support and guide the LE slat assemblies during its extension and the retraction. The track assembly of the track forgings and the bushings.

2. Leading Particulars (Approximate)

- A. Auxiliary Track Assembly - Slat No. 1 (65-52144-X, 65-67173-X)
- (1) Length – 11.2 inches
  - (2) Width – 0.34 inches
  - (3) Height – 5.69 inches
  - (4) Weight – Not Available
- B. Auxiliary Track Assembly - Slat No. 2 (65-52145-X, 65-67174-X)
- (1) Length – 11.42 inches
  - (2) Width – 0.34 inches
  - (3) Height – 6.67 inches
  - (4) Weight – Not Available
- C. Auxiliary Track Assembly - Slat No. 3 (65-52146-X, 65-67175-X)
- (1) Length – 11.35 inches
  - (2) Width – 0.34 inches
  - (3) Height – 7.53 inches
  - (4) Weight – Not Available
- D. Auxiliary Track - Slat No. 3 (65-68275-X)
- (1) Length – 8.72 inches
  - (2) Width – 0.38 inches
  - (3) Height – 7.53 inches
  - (4) Weight – Not Available
- E. Auxiliary Track - Slat No. 2 (65-68276-X)
- (1) Length – 8.71 inches
  - (2) Width – 0.34 inches
  - (3) Height – 6.67 inches
  - (4) Weight – Not Available
- F. Auxiliary Track - Slat No. 1 (65-68243-X)
- (1) Length – 11.33 inches
  - (2) Width – 0.34 inches
  - (3) Height – 5.4 inches
  - (4) Weight – Not Available
- G. Auxiliary Track Assembly - Slat No. 1 (65-80984-X)

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- (1) Length – 10.5 inches
  - (2) Width – 0.34 inches
  - (3) Height – 5.69 inches
  - (4) Weight – Not Available
- H. Auxiliary Track - Slat No. 1 (65-80990-X)
- (1) Length – 10.15 inches
  - (2) Width – 0.34 inches
  - (3) Height – 5.41 inches
  - (4) Weight – Not Available

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**TESTING AND FAULT ISOLATION**

**(NOT APPLICABLE)**

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TESTING AND FAULT ISOLATION

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**DISASSEMBLY**

**(NOT APPLICABLE)**

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DISASSEMBLY

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## COMPONENT MAINTENANCE MANUAL

### CLEANING

**(NOT APPLICABLE)**

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CLEANING

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## COMPONENT MAINTENANCE MANUAL

### CHECK

#### 1. General

- A. Check all parts for obvious defects in accordance with standard industry practices.
- B. Refer to FITS AND CLEARANCES for design dimensions and wear limits.

#### 2. Check

- A. Magnetic particle check the following parts per SOPM 20-20-01.
  - (1) Track (5B, 5G, 5H, 10E, 15D, IPL Figure 1)
  - (2) Track (10, IPL Figure 2)
  - (3) Track (10, IPL Figure 3)
  - (4) Track (10, 10A, IPL Figure 4)
  - (5) Track (10, IPL Figure 5)
  - (6) Track (10, 10A, IPL Figure 6)
  - (7) Track (10, 10A, IPL Figure 7)
  - (8) Track (10, IPL Figure 8)

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CHECK  
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## COMPONENT MAINTENANCE MANUAL

### REPAIR

#### 1. Content

A. Repair, refinish and replacement procedures are included in separate sections as follows:

**Table 601:**

<b>P/N</b>	<b>NAME</b>	<b>REPAIR</b>
65-52144	AUX TRACK ASSY - SLAT NO. 1	1-1
65-52145	AUX TRACK ASSY - SLAT NO. 2	2-1
65-67173	AUX TRACK ASSY - SLAT NO. 1	3-1
65-67174	AUX TRACK ASSY - SLAT NO. 2	4-1
65-67175	AUX TRACK ASSY - SLAT NO. 3	5-1
65-68243	AUX TRACK - SLAT NO. 1	6-1
65-68275	AUX TRACK - SLAT NO. 3	7-1
65-68276	AUX TRACK - SLAT NO. 2	8-1
65-80984	AUX TRACK ASSY - SLAT NO. 1	9-1
65-80990	AUX TRACK - SLAT NO. 1	10-1
65-52146	AUX TRACK ASSY - SLAT NO.3	11-1

#### 2. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in SOPM 20-00-00.

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REPAIR - GENERAL

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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 1 - AUXILIARY TRACK ASSEMBLY - REPAIR 1-1

65-52144-2

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 2 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must not protrude more than 0.05 inch on either side of the track (5).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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REPAIR 1-1

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

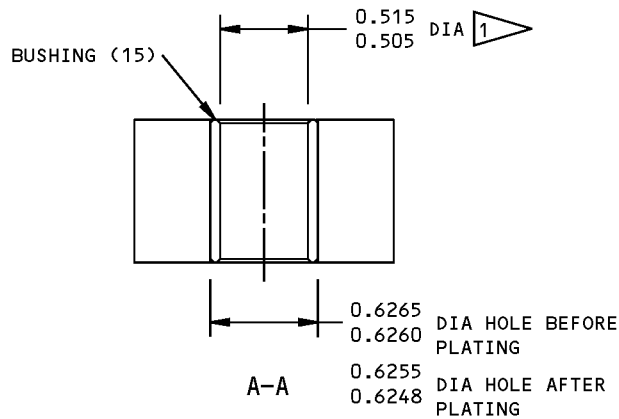
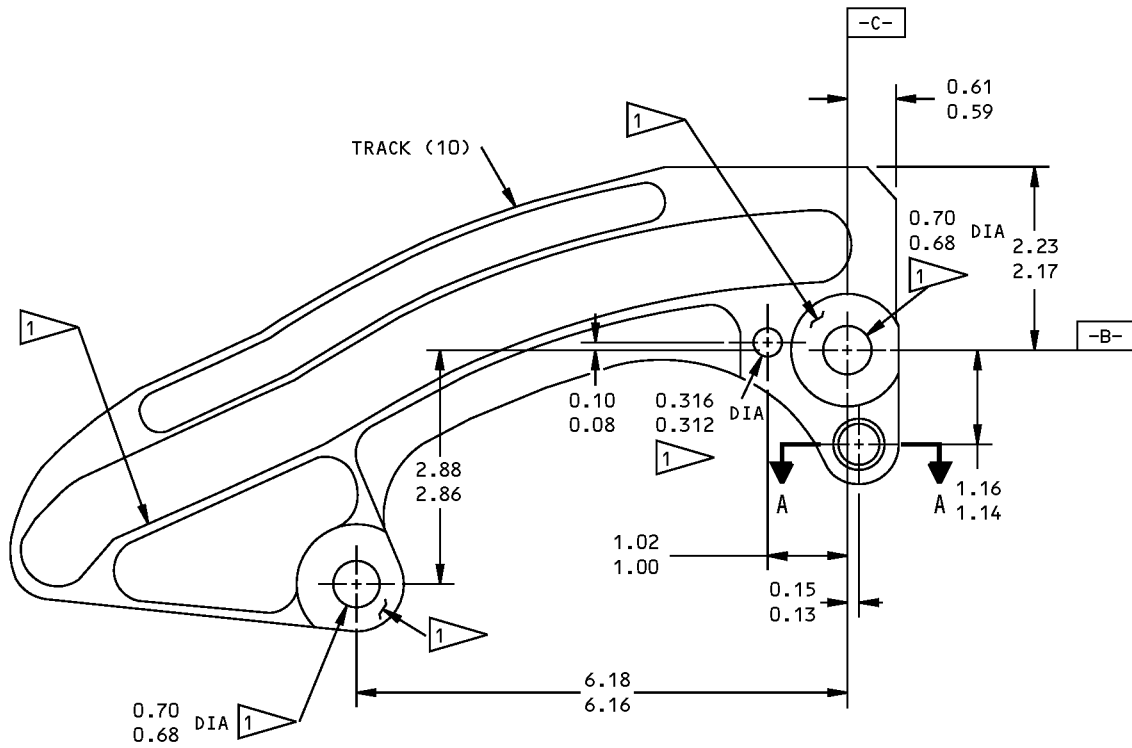
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REPAIR 1-1

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COMPONENT MAINTENANCE MANUAL



1 OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 2  
ALL DIMENSIONS ARE IN INCHES

65-52144-2 Slat No. 1 Auxiliary Track Assembly Repair  
Figure 601

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REPAIR 1-1  
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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 2 - AUXILIARY TRACK ASSEMBLY - REPAIR 2-1

65-52145-2

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Track to 220-240 ksi
- D. Refer to IPL Figure 5 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the strink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must not protrude more than 0.05 inch on either side of the track (5).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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REPAIR 2-1

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

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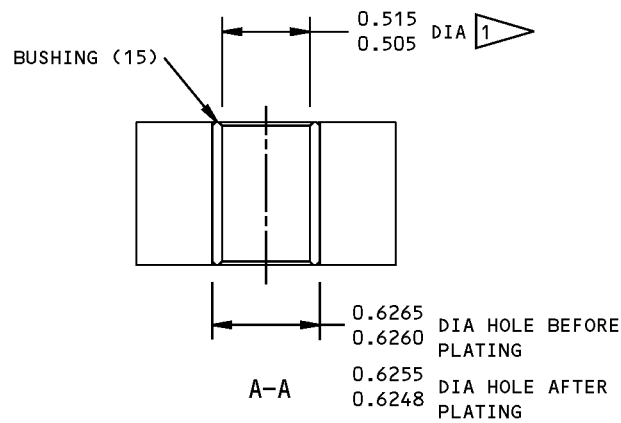
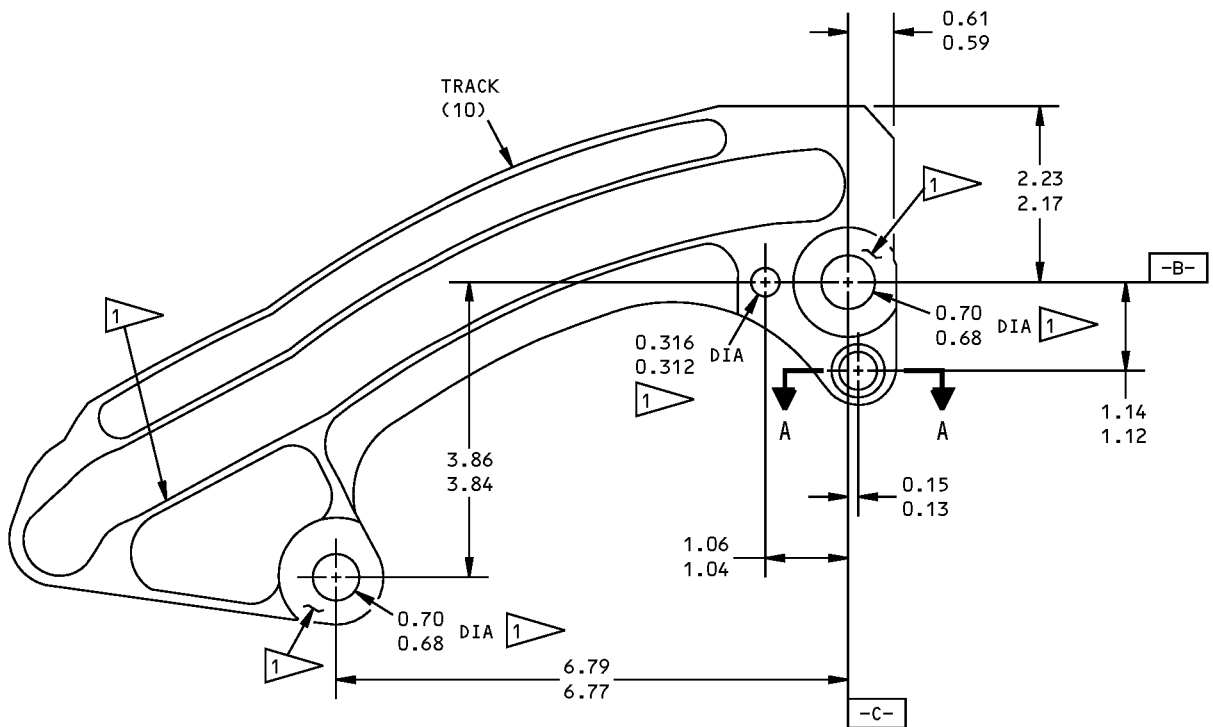
REPAIR 2-1

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1 OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 5  
ALL DIMENSIONS ARE IN INCHES

65-52145-2 Slat No. 2 Auxiliary Track Assembly Repair  
Figure 601

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REPAIR 2-1  
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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 1 - AUXILIARY TRACK ASSEMBLY - REPAIR 3-1

65-67173-1

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 3 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must not protrude more than 0.05 inch on either side of the track (5).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

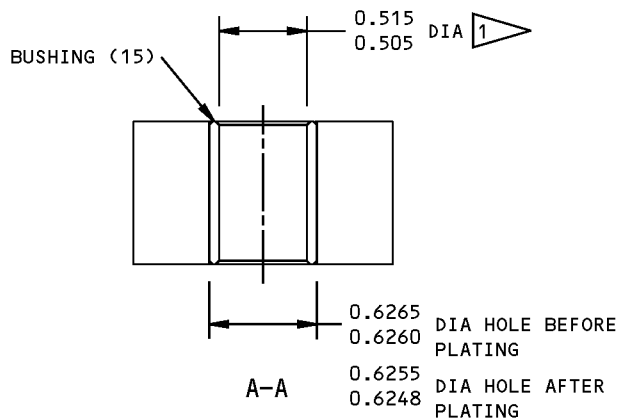
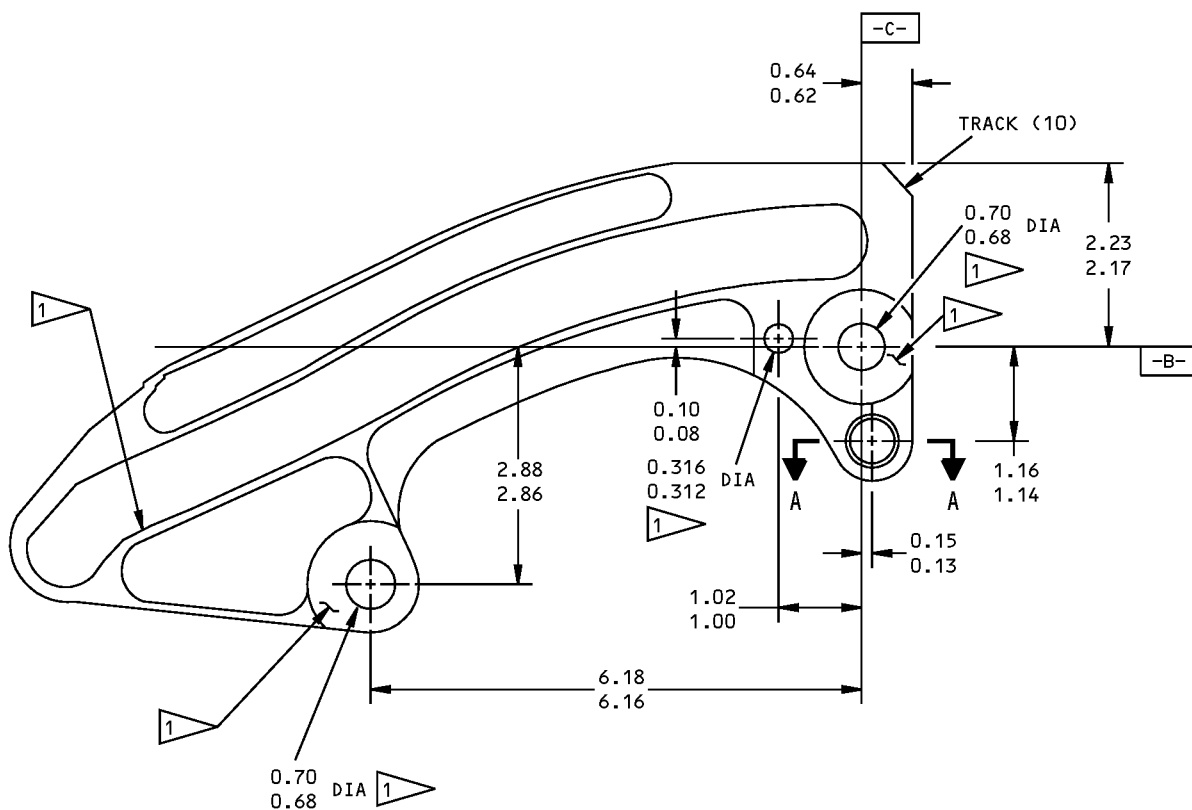
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REPAIR 3-1

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1 OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 3  
ALL DIMENSIONS ARE IN INCHES

65-67173-1 Slat No. 1 Auxiliary Track Assembly Repair  
Figure 601

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REPAIR 3-1  
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**COMPONENT MAINTENANCE MANUAL**

**SLAT NO. 2 - AUXILIARY TRACK ASSEMBLY - REPAIR 4-1**

**65-67174-1, -3, -5, -6**

**1. General**

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 6 for the item numbers.

**2. Bushing Replacement**

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track assembly (1).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must be protrude more than 0.05 inch on either side of the track(5).

**NOTE:** The bushing (10A) flange must be on the far side for the auxiliary track assembly (1B) and near side for the auxiliary track assembly (1C).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

**3. Auxiliary Track Refinish**

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

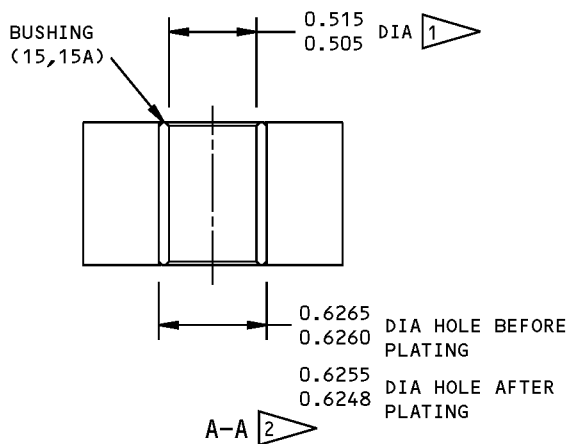
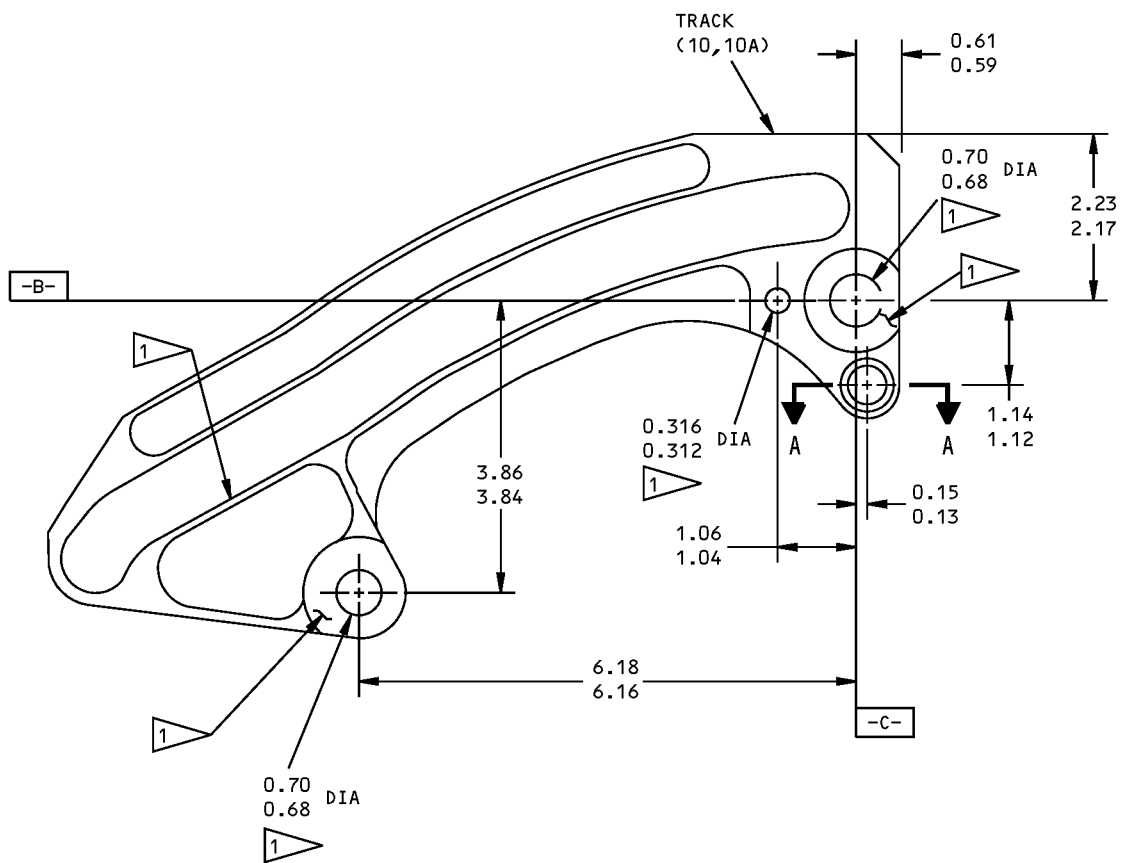
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- 1 OMIT BMS 10-11 PRIMER IN THIS AREA
- 2 BUSHING FLANGES NOT SHOWN

ITEM NUMBERS REFER TO IPL FIG. 6  
ALL DIMENSIONS ARE IN INCHES

65-67174-1,-3,-5,-6 Slat No. 2 Auxiliary Track Assembly Repair  
Figure 601

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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 3 - AUXILIARY TRACK ASSEMBLY - REPAIR 5-1

65-67175-1, -3, -5, -6

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 7 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (910) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (1) must not protrude more than 0.05 inch on either side of the track (5).

**NOTE:** The bushing (10A) flange must be on the far side for the auxiliary track assembly (1B) and near side for the auxiliary track assembly (1c).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

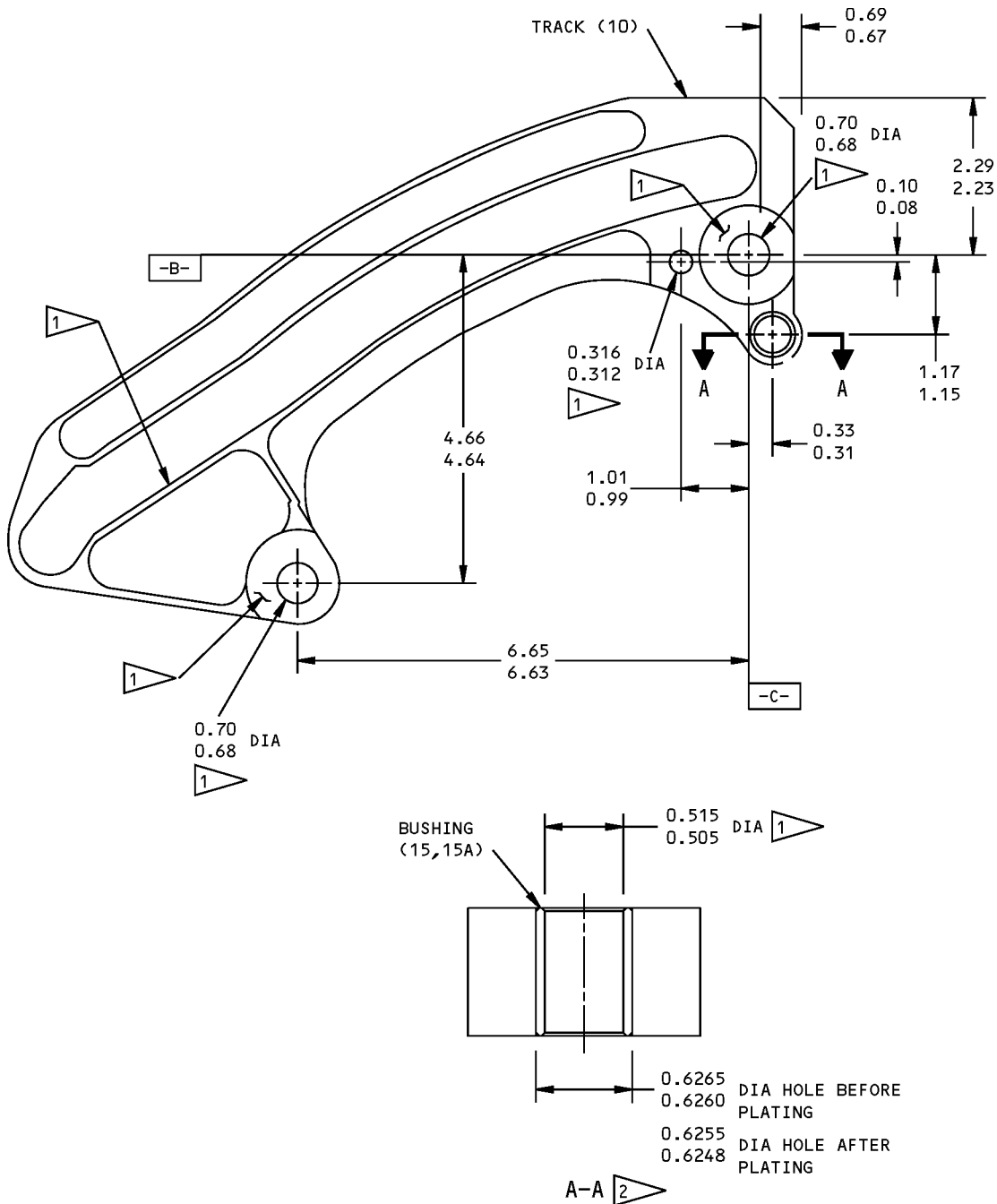
# 27-81-52

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- 1 OMIT BMS 10-11 PRIMER IN THIS AREA
- 2 BUSHING FLANGE NOT SHOWN FOR CLARITY

ITEM NUMBERS REFER TO IPL FIG. 7  
ALL DIMENSIONS ARE IN INCHES

65-67175-1,-3,-5,-6 Slat No. 3 Auxiliary Track Assembly Repair  
Figure 601

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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 1 - AUXILIARY TRACK - REPAIR 6-1

65-68243-1

#### 1. General

- A. This repair gives the data that is necessary to refinish the auxiliary track (1B).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 1 for the item numbers.

#### 2. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

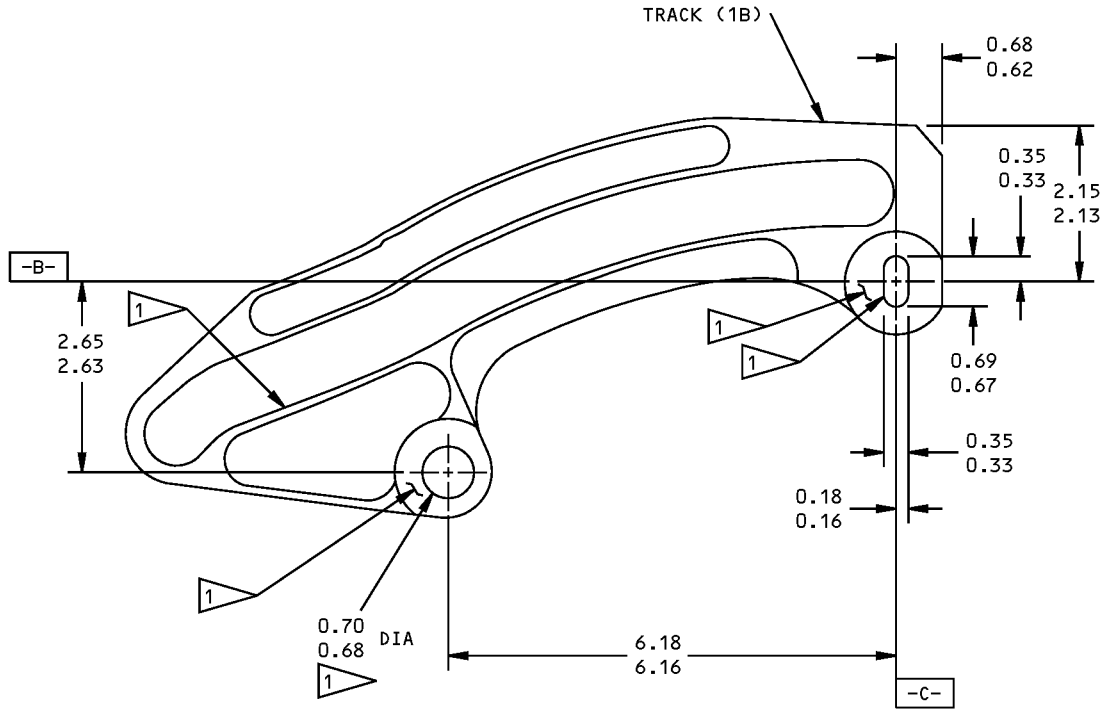
**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-15.01) and apply primer, C00259 (F-20.02) to Auxiliary Track (1B).

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COMPONENT MAINTENANCE MANUAL



OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 1  
ALL DIMENSIONS ARE IN INCHES

65-68243-1 Slat No. 1 Auxiliary Track Repair  
Figure 601

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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 3 - AUXILIARY TRACK - REPAIR 7-1

65-68275-1

#### 1. General

- A. This repair gives the data that is necessary to refinish the auxiliary track (10D).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 1 for the item numbers.

#### 2. Auxiliary Track Refinish

##### A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

<u>Reference</u>	<u>Description</u>	<u>Specification</u>
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

##### B. References

<u>Reference</u>	<u>Title</u>
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

##### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-15.01) and apply primer, C00259 (F-20.02) to Auxiliary Track (10D).

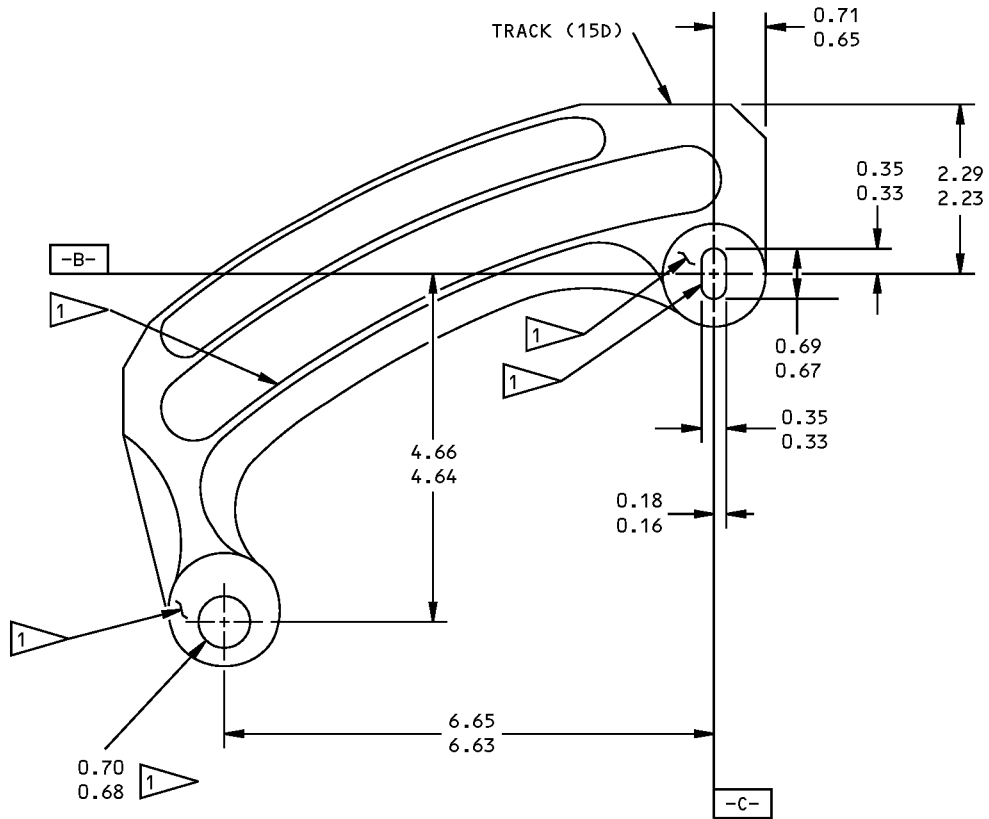
# 27-81-52

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1 OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 1  
ALL DIMENSIONS ARE IN INCHES

65-68275-1 Slat No. 3 Auxiliary Track Repair  
Figure 601

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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 2 - AUXILIARY TRACK - REPAIR 8-1

65-68276-1

#### 1. General

- A. This repair gives the data that is necessary to refinish the auxiliary track (15E).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 1 for item numbers.

#### 2. Auxiliary Track Refinish

##### A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

<u>Reference</u>	<u>Description</u>	<u>Specification</u>
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

##### B. References

<u>Reference</u>	<u>Title</u>
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

##### C. Procedure

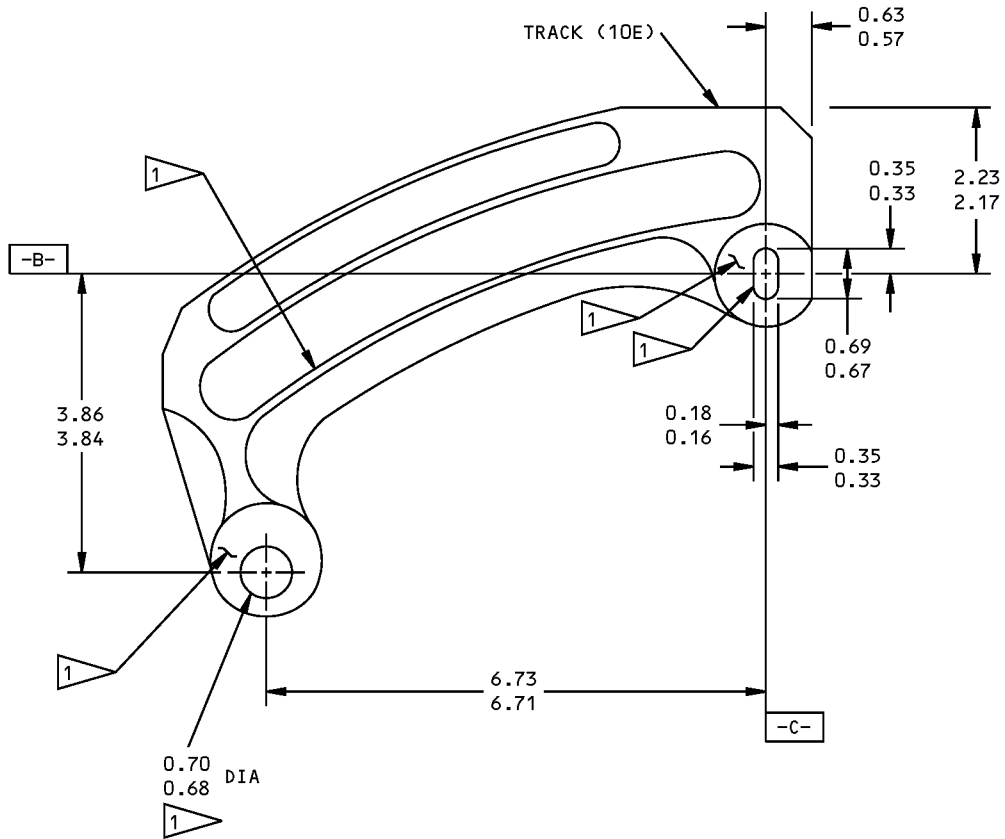
**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-15.01) and apply primer, C00259 (F-20.02) to Auxiliary Track (5E).

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OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 1  
ALL DIMENSIONS ARE IN INCHES

65-68276-1 Slat No. 2 Auxiliary Track Repair  
Figure 601

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REPAIR 8-1  
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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 1 - AUXILIARY TRACK ASSEMBLY - REPAIR 9-1

65-80984-1, -3, -5, -6

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 4 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must not protrude more than 0.05 inch on either side of the track (5).

**NOTE:** The bushing (10A) flange must be on the far side for the auxiliary track assembly (1B) and near side for the auxiliary track assembly (1C).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

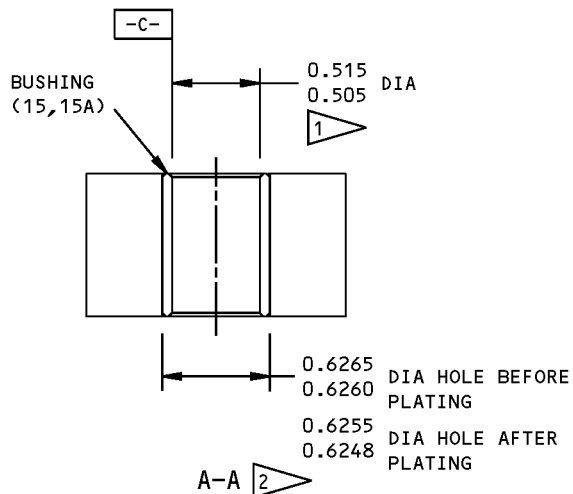
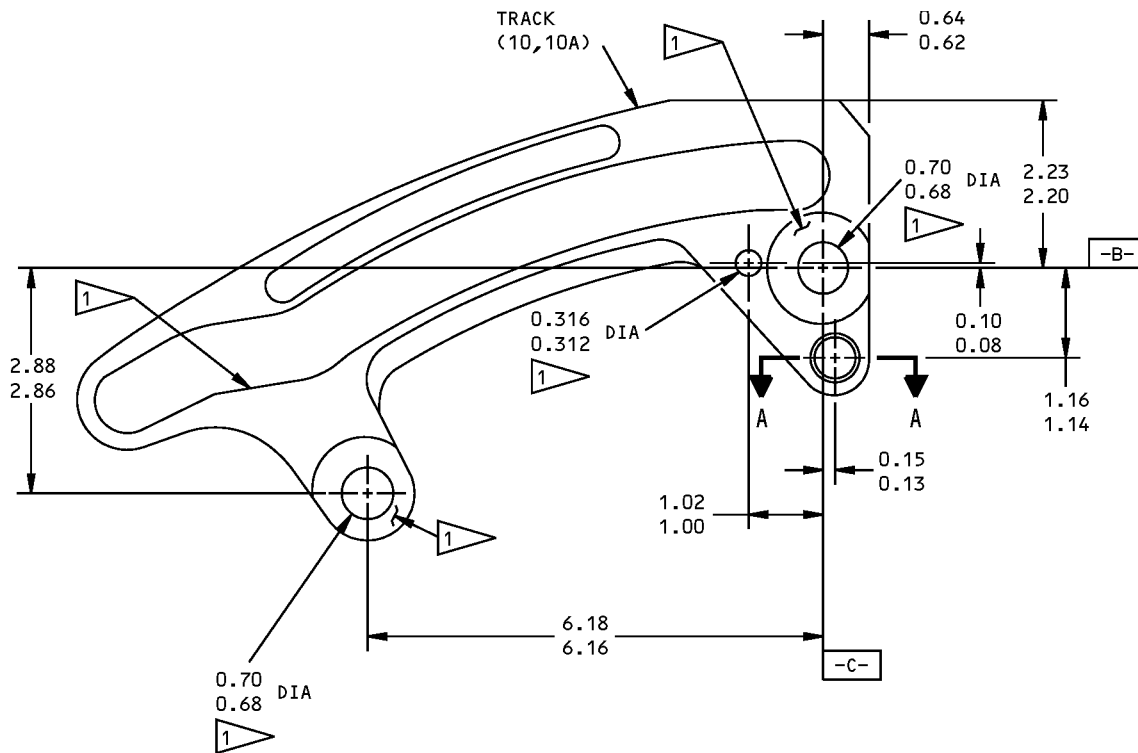
**27-81-52**

REPAIR 9-1

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- 1 OMIT BMS 10-11 PRIMER IN THIS AREA
- 2 BUSHING FLANGES NOT SHOWN FOR CLARITY

ITEM NUMBERS REFER TO IPL FIG. 4  
ALL DIMENSIONS ARE IN INCHES

65-80984-1,-3,-5,-6 Slat No. 1 Auxiliary Track Assembly Repair  
Figure 601

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REPAIR 9-1  
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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 1 - AUXILIARY TRACK - REPAIR 10-1

65-80990-1, -2

#### 1. General

- A. This repair gives the data that is necessary to refinish the auxiliary track (1G, 1H).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4330 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 1 for the item numbers.

#### 2. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

<u>Reference</u>	<u>Description</u>	<u>Specification</u>
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

<u>Reference</u>	<u>Title</u>
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-15.01) and apply primer, C00259 (F-20.02) to Auxiliary Track (1G, 1H).

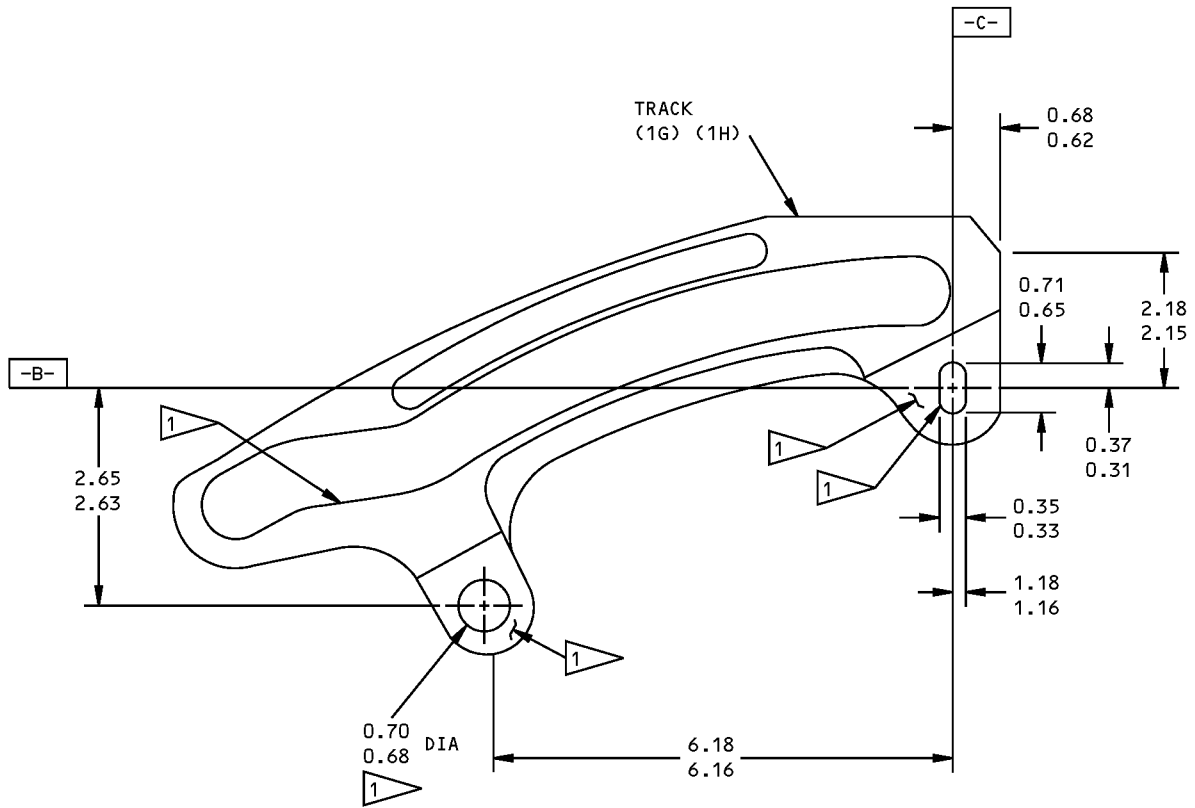
# 27-81-52

REPAIR 10-1

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OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 1  
ALL DIMENSIONS ARE IN INCHES

65-80990-1,-2 Slat No. 1 Auxiliary Track Repair  
Figure 601

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REPAIR 10-1  
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## COMPONENT MAINTENANCE MANUAL

### SLAT NO. 3 - AUXILIARY TRACK ASSEMBLY - REPAIR 11-1

65-52146-2

#### 1. General

- A. This repair gives the data that is necessary to repair and refinish the auxiliary track assembly (1).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. General Repair Details
  - (1) Track Material - 4370 bar as shown in BMS 7-122, Heat Treat to 220-240 ksi
- D. Refer to IPL Figure 8 for the item numbers.

#### 2. Bushing Replacement

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

- B. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

**NOTE:** For finishing materials, refer to SOPM 20-60-02.

- (1) Remove the bushing (10) from the auxiliary track (5).
- (2) Install the bushing (10) into the auxiliary track (5) with wet primer, C00259. Use the shrink-fit procedure (SOPM 20-50-03).

**NOTE:** The bushing (10) must not protrude more than 0.05 inch on either side of the track (5).

- (3) Machine the bushing (10) inner diameter to 0.505-0.515 inch.

#### 3. Auxiliary Track Refinish

- A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

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## COMPONENT MAINTENANCE MANUAL

### B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

### C. Procedure

**NOTE:** For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Apply cadmium-titanium plate (F-1.308) and apply primer, C00259 (SRF-12.205) to Auxiliary Track (5).

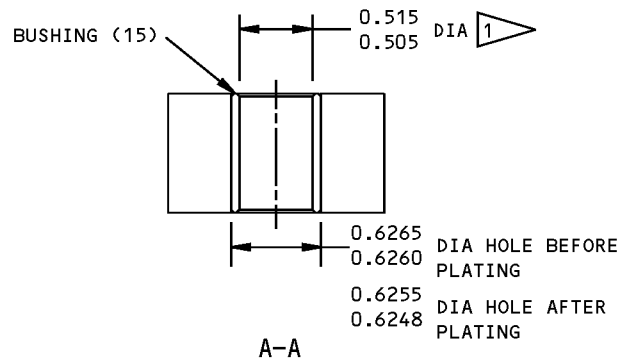
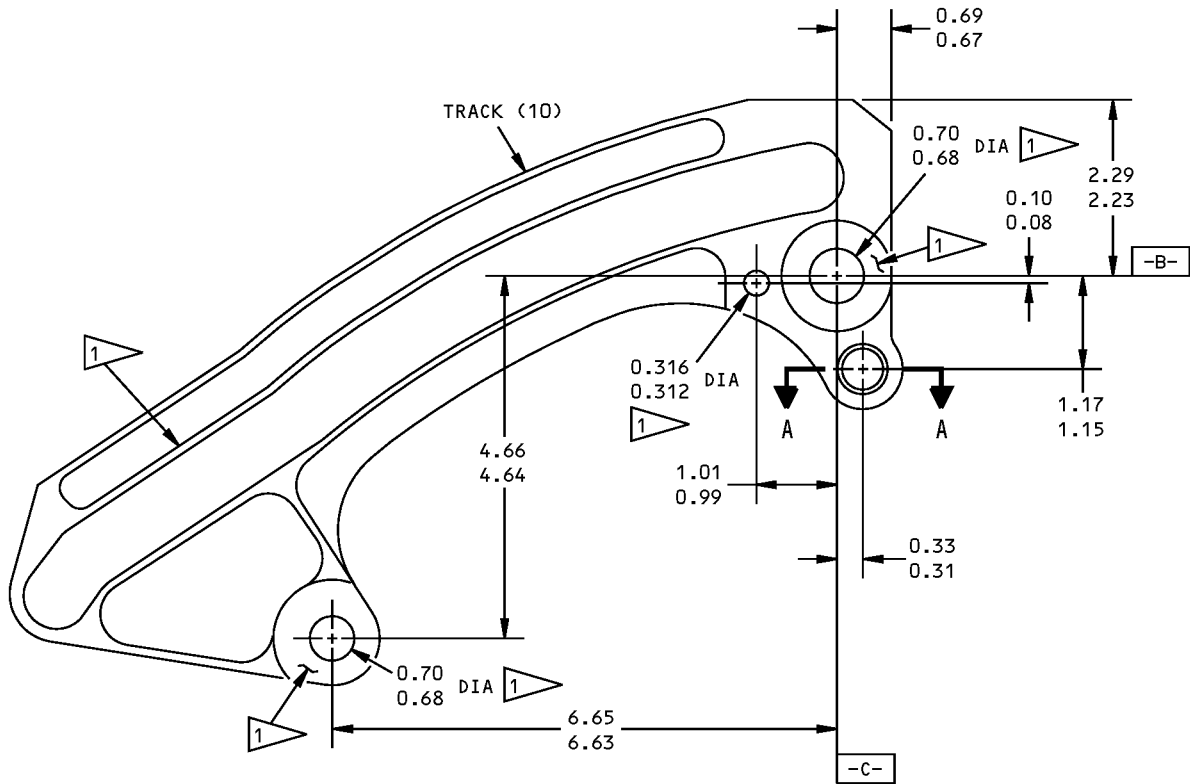
**27-81-52**

REPAIR 11-1

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COMPONENT MAINTENANCE MANUAL



OMIT BMS 10-11 PRIMER IN THIS AREA

ITEM NUMBERS REFER TO IPL FIG. 8

ALL DIMENSIONS ARE IN INCHES

65-52146-2 Slat No. 3 Auxiliary Track Assembly Repair  
Figure 601

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SEE TITLE PAGE FOR  
LIST OF PART NUMBERS



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**ASSEMBLY**

**(NOT APPLICABLE)**

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ASSEMBLY

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SEE TITLE PAGE FOR  
LIST OF PART NUMBERS



## COMPONENT MAINTENANCE MANUAL

### FITS AND CLEARANCES

**(NOT APPLICABLE)**

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FITS AND CLEARANCES

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SEE TITLE PAGE FOR  
LIST OF PART NUMBERS



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**SPECIAL TOOLS, FIXTURES, AND EQUIPMENT**

**(NOT APPLICABLE)**

**27-81-52**

SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

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## COMPONENT MAINTENANCE MANUAL

### ILLUSTRATED PARTS LIST

#### 1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
.	Assembly					
.	Attaching parts for assembly					
.	.	Detail parts for assembly				
.	.	Subassembly				
.	.	Attaching parts for subassembly				
.	.	.	Detail parts for subassembly			
.	.	.	Sub-subassembly			
.	.	.	Attaching parts for subassembly			
.	.	.	.	Details parts for sub-subassembly		
						Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
  - (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
  - (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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ILLUSTRATED PARTS LIST

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Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

### VENDOR CODES

<b>Code</b>	<b>Name</b>
23294	AVALON MACHINE PRODUCTS INC 15337 ALLEN STREET PARAMOUNT, CALIFORNIA 90723-4011
70265	ALL POWER MANUFACTURING COMPANY 13141 MOLETTE STREET SANTA FE SPRINGS, CALIFORNIA 90670-5500 FORMERLY IN MONTEBELLO, CALIFORNIA
94892	MASTER MACHINE PRODUCTS CORPORATION 1551 SOUTH PRIMROSE AVE MONROVIA, CALIFORNIA 91016-4542 FORMERLY IN HUNTINGTON PARK, CALIFORNIA

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NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-52144-1		2	5	1
65-52144-2		1	1	RF
		2	1	RF
65-52145-1		5	5	1
65-52145-2		1	5	RF
		5	1	RF
65-52146-1		8	10	1
65-52146-2		1	10E	RF
		8	1	RF
65-67173-1		1	1A	RF
		3	1	RF
65-67173-2		3	5	1
65-67174-1		1	5A	RF
		6	1	RF
65-67174-2		6	5	1
65-67174-3		1	5B	RF
		6	1A	RF
65-67174-4		6	5A	1
65-67174-5		1	5C	RF
		6	1B	RF
65-67174-6		1	5D	RF
		6	1C	RF
65-67175-1		1	10	RF
		7	1	RF
65-67175-2		7	5	1
65-67175-3		1	10A	RF
		7	1A	RF
65-67175-4		7	5A	1
65-67175-5		1	10B	RF
		7	1B	RF
65-67175-6		1	10C	RF
		7	1C	RF
65-68243-1		1	1B	RF
65-68275-1		1	10D	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-68276-1		1	5E	RF
65-80984-1		1	1C	RF
		4	1	RF
65-80984-2		4	5	1
65-80984-3		1	1D	RF
		4	1A	RF
65-80984-4		4	5A	1
65-80984-5		1	1E	RF
		4	1B	RF
65-80984-6		1	1F	RF
		4	1C	RF
65-80990-1		1	1G	RF
65-80990-2		1	1H	RF
BACB28W8B033		4	10A	1
		4	10A	1
		4	10A	1
		6	10A	1
		6	10A	1
		6	10A	1
		7	10A	1
		7	10A	1
		7	10A	1
BACB28Y8B35		2	10	1
		2	10	1
		2	10	1
		3	10	1
		3	10	1
		3	10	1
		4	10	1
		4	10	1
		4	10	1
		5	10	1
		5	10	1
		5	10	1
		6	10	1

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<b>PART NUMBER</b>	<b>AIRLINE PART NUMBER</b>	<b>FIGURE</b>	<b>ITEM</b>	<b>UNITS PER ASSEMBLY</b>
BACB28YB38		6	10	1
		6	10	1
		7	10	1
		7	10	1
		7	10	1
		8	15	1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Slat Track Assemblies - Leading Edge  
IPL Figure 1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1	65-52144-2									A	RF
-1A	65-67173-1									B	RF
-1B	65-68243-1									C	RF
-1C	65-80984-1									D	RF
-1D	65-80984-3									E	RF
-1E	65-80984-5									F	RF
-1F	65-80984-6									G	RF
-1G	65-80990-1									H	RF
-1H	65-80990-2									I	RF
-5	65-52145-2									J	RF
-5A	65-67174-1									K	RF
-5B	65-67174-3									L	RF
-5C	65-67174-5									M	RF
-5D	65-67174-6									N	RF
-5E	65-68276-1									O	RF
-10	65-67175-1									P	RF
-10A	65-67175-3									Q	RF

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**COMPONENT MAINTENANCE MANUAL**

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1- -10B	65-67175-5		AUX TRACK ASSY-SLAT NO. 3 (FOR DETAILS SEE FIG. 7) (POST SB 737-57-1243)							R	RF
-10C	65-67175-6		AUX TRACK ASSY-SLAT NO. 3 (FOR DETAILS SEE FIG. 7) (POST SB 737-57-1243)							S	RF
-10D	65-68275-1		AUX TRACK-SLAT NO. 3 (POST SB 737-57-1243)							T	RF
-10E	65-52146-2		AUX TRACK ASSY-SLAT NO. 3 (FOR DETAILS SEE FIG. 8) (PRE SB 737-57-1243)							V	RF

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly- Slat No. 1, 65-52144-2  
IPL Figure 2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2- -1	65-52144-2		AUX TRACK ASSY-SLAT NO. 1 (PRE SB 737-57-1243) (FOR POST SERVICE BULLETIN CONFIGURATION SEE FIG. 1 AND 3)							A	RF
-5	65-52144-1		. TRACK-AUX								1
-10	BACB28Y8B35		. BUSHING (V23294) (OPT BACB28Y8B35 (V70265)) (OPT BACB28Y8B35 (V94892))								1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly - Slat No. 1, 65-67173-1  
IPL Figure 3

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
3-											
-1	65-67173-1									B	RF
-5	65-67173-2										1
-10	BACB28Y8B35										1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly - Slat No. 1, 65-80984-1,-3,-5,-6  
IPL Figure 4

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COMPONENT MAINTENANCE MANUAL

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-1	65-80984-1									D	RF
-1A	65-80984-3									E	RF
-1B	65-80984-5									F	RF
-1C	65-80984-6									G	RF
-5	65-80984-2									D	1
-5A	65-80984-4									E, F, G	1
-10	BACB28Y8B35									D, E	1
-10A	BACB28W8B033									F, G	1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly - Slat No. 2, 65-52145-2  
IPL Figure 5

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
5- -1	65-52145-2		AUX TRACK ASSY-SLAT NO. 2 (PRE SB 737-57-1243) (FOR POST SERVICE BULLETIN CONFIGURATION SEE FIG. 1 AND 6)							J	RF
-5	65-52145-1		. ASSY-TRACK								1
-10	BACB28Y8B35		. BUSHING (V23294) (OPT BACB28Y8B35 (V70265)) (OPT BACB28Y8B35 (V94892))								1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track assembly - Slat No. 2, 65-67174-1,-3,-5,-6  
IPL Figure 6

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
6-											
-1	65-67174-1										
-1A	65-67174-3										
-1B	65-67174-5										
-1C	65-67174-6										
-5	65-67174-2										
-5A	65-67174-4										
-10	BACB28Y8B35										
-10A	BACB28W8B033										

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly - Slat No. 3, 65-67174-1,-3,-5,-6  
IPL Figure 7

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
7-											
-1	65-67175-1									P	RF
-1A	65-67175-3									Q	RF
-1B	65-67175-5									R	RF
-1C	65-67175-6									S	RF
5	65-67175-2									P	1
5A	65-67175-4									Q, R, S	1
10	BACB28Y8B35									P, Q	1
10A	BACB28W8B033									R, S	1

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## COMPONENT MAINTENANCE MANUAL

FIGURE NOT ILLUSTRATED

Auxiliary Track Assembly - Slat No. 3, 65-52146  
IPL Figure 8

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
8- 1	65-52146-2		AUX TRACK ASSY-SCAT NO. 3 (PRE SB 737-57-1243) (FOR POST SERVICE BULLETIN CONFIGURATION SEE FIG. 1 AND 7)							V	RF
10	65-52146-1		. TRACK-AUX								1
15	BACB28YB38		. BUSHING (V23294)								1

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