

COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

RUDDER PEDAL STEERING ASSEMBLY

PART NUMBER 275A5104-1, -2

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32-51-33



Revision No. 10 Jul 01/2009

To: All holders of RUDDER PEDAL STEERING ASSEMBLY 32-51-33.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

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Location of Change Description of Change

32-51-33

FRONTMATTER Changed the part number information on the title page.

REPAIR 1-1 Added clarifications and updated callouts.

Changed the data in the Consumable Materials list.

ILLUSTRATED PARTS LIST Added the Vendor Codes list.

Changed the data in the NUMERICAL INDEX list.

Added clarifications.

Added clarifications and updated callouts.

Added new Illustrated Parts List.

Added rudder pedal steering assembly 275A5104-2 with changed screws

for easier installation.

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A = Added, R = Revised, D = Deleted, O = Overflow

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TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR 38065	DEC 01/97

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All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Rev	vision	Fi	iled	Rev	rision	Filed		
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All temporary revisions to this manual will be accompanied by a cover sheet bearing the temporary revision number. Enter the temporary revision number in numerical order, together with the temporary revision date, the date the temporary revision is inserted and the initials of the person filing.

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RECORD OF TEMPORARY REVISION



INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alphavariant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.



RUDDER PEDAL STEERING ASSEMBLY - DESCRIPTION AND OPERATION

1. Description

A. The rudder pedal steering assembly consists of a steering crank, clutch arm, cable drum, an eccentric mounted in a clutch crank and a quadrant. These parts are made of aluminum alloys.

2. Operation

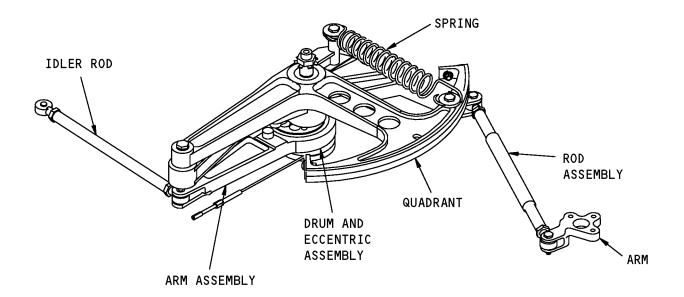
- A. The rudder pedal steering assembly connects the rudder pedals to the nose wheel steering and operates when the nose gear is compressed by the airplane weight. This compression movement of the nose gear is transmitted by linkage and cables to move the eccentric and reposition the clutch crank to prevent stops on the clutch crank from touching the clutch arm. Instead, the clutch arm touches the stops on the steering crank and allows any movement of the rudder pedals to be transmitted from the steering arm to the quadrant. This quadrant connects to the nose wheel steering cables and is free to move with the cables whenever nose wheel steering is used, or to drive the cables when positioned by the rudder pedal system.
- B. When the nose gear shock strut is not compressed, the action of the eccentric moves the stops on the clutch crank into contact with the clutch arm. This prevents any movement of the rudder pedals from reaching the clutch arm and moving the quadrant.

3. <u>Leading Particulars (Approximate)</u>

- A. Length 17 in.
- B. Width 10.5 in.
- C. Height 6.5 in.
- D. Weight 5.6 lbs

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Rudder Pedal Steering Assembly Figure 1

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DESCRIPTION AND OPERATION
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TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

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DISASSEMBLY

(NOT APPLICABLE)

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DISASSEMBLY
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CLEANING

1. General

- A. This procedure has the data necessary to clean the rudder pedal steering assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Cleaning

A. References

Reference	Title
SOPM 20-30-01	CLEANING AND RELUBRICATING BEARINGS
SOPM 20-30-03	GENERAL CLEANING PROCEDURES

B. Procedure

- (1) Clean the bearings (30, 130, 150, 175, 235, 240, 280, 345, 360, 370) as specified in SOPM 20-30-01.
- (2) Clean the other parts by standard industry procedures and the instructions in SOPM 20-30-03.



CHECK

1. General

- A. This procedure has the data necessary to find defects in the material of the specified parts.
- B. Refer to FITS AND CLEARANCES for the design dimension and wear limits.
- C. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- D. Refer to IPL Figure 1 for item numbers.

2. Check

A. References

Reference	Title
SOPM 20-20-02	PENETRANT METHODS OF INSPECTION

B. Procedure

- (1) Use standard industry procedures to do a visual check of all the parts for defects. Do the penetrant or magnetic particle check if the visual check shows possible damage or if you suspect possible damage on the parts listed below:
- (2) Do a penetrant check (SOPM 20-20-02) of these parts:
 - (a) Arm (20)
 - (b) Steering Crank (250)
 - (c) Quadrant (186)
- (3) Do a check of the spring as shown below. The spring must come back to the free length with no permanent deformation.

Table 501:

IPL No. Fig. 1	Approximate Free Length (Inches) (Reference only)	_	Allowable Load Limits (Pounds)
115	115 3.47 10.00 7.50		20.43 - 22.57 13.30 - 14.70



REPAIR

1. General

A. Instructions for repair, refinish, and replacement of the specified subassembly parts are included in each REPAIR when applicable:

Table 601:

PART NUMBER	NAME	REPAIR
_	REFINISH OF OTHER PARTS	1-1
275A5103	DRUM AND ECCENTRIC ASSEMBLY	2-1
65-25843	QUADRANT ASSEMBLY	3-1
69-20495	STEERING ROD ASSEMBLY	4-1

2. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.

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— STRAIGHTNESS □ FLATNESS	Ø \$ Ø	DIAMETER SPHERICAL DIAMETER
<pre></pre>	R SR	RADIUS SPHERICAL RADIUS
O ROUNDNESS O CYLINDRICITY O PROFILE OF A LINE	() BASIC (BSC) OR	REFERENCE A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE. FROM THIS FEATURE PERMIS—
 ○ PROFILE OF A SURFACE ○ CONCENTRICITY = SYMMETRY ∠ ANGULARITY 	DIM	SIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
RUNOUT TOTAL RUNOUT COUNTERBORE OR SPOTFACE COUNTERSINK THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)	—A— (M) (L) (S) (P) FIM	DATUM MAXIMUM MATERIAL CONDITION (MMC) LEAST MATERIAL CONDITION (LMC) REGARDLESS OF FEATURE SIZE (RFS) PROJECTED TOLERANCE ZONE FULL INDICATOR MOVEMENT

EXAMPLES

<u>=770</u>	71111
- 0.002 STRAIGHT WITHIN 0.002	◎ Ø 0.0005 C CONCENTRIC TO DATUM C WITHIN 0.0005 DIAMETER
<u> 0.002 B </u> PERPENDICULAR TO DATUM B WITHIN 0.002	= 0.010 A SYMMETRICAL WITH DATUM A
// 0.002 A PARALLEL TO DATUM A WITHIN 0.002	WITHIN 0.010 0.005 A ANGULAR TOLERANCE 0.005
0.002 ROUND WITHIN 0.002	WITH DATUM A
0.010 CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	□ Ø 0.002 ③ B LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
O.006 A EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES O.006 INCH APART RELATIVE TO DATUM A	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010 INCH DIAMETER, PERPENDICULAR TO DATUM A, AND EXTENDING 0.510 INCH ABOVE DATUM A, MAXIMUM MATERIAL CONDITION
O.020 A SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.020 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE	OR DIMENSION IS 2.000 2.000 BSC

True Position Dimensioning Symbols Figure 601

32-51-33REPAIR - GENERAL
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I

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I



COMPONENT MAINTENANCE MANUAL

REFINISH OF OTHER PARTS - REPAIR 1-1

1. General

- A. Use this procedure to refinish the parts which are not in the other repairs.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the True Position Dimensioning Symbols used in the repair figures.
- D. Refer to IPL Figure 1 for item numbers.

2. Refinish of Other Parts

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
B50080	Compound - Corrosion Preventive, Solvent Cutback Cold-Application (Grade 2 - Soft Film)	s, MIL-PRF-16173, Grade 2 (Supersedes MIL-C-16173, Grade 2)
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Instructions for the repair of the parts in REPAIR 1-1, Table 601 is only replacement of the original finish.
- (2) Refer to REPAIR 1-1, Table 601 for refinish details.

Table 601: Refinish Details

	IPL FIG. & ITEM	MATERIAL	FINISH
ı	IPL Fig. 1		
	Arm (20)	Al alloy	Boric acid-sulfuric acid anodize (F-17.31) and apply primer, C00259 (F-20.03) unless noted by flagnote 1 in REPAIR 1-1, Figure 601.

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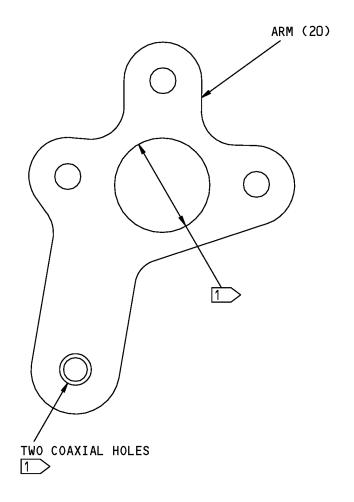
Table 601: Refinish Details (Continued)

IPL FIG. & ITEM	MATERIAL	FINISH
, , ,		Chemical treat or chromic acid anodize and apply primer, C00259 (SRF-2.901).
Plugs (55, 65)	Plastic	No finish (F-25.01).
Spring (115)	Music wire, ASTM A228	Cadmium plate and apply primer, C00259 (F-16.03).
Bolt (120)	4340 steel, no heat treat	Cadmium plate (F-15.02).
Spacers (135, 170, 180, 210, 255, 300, 355A)	4340 or 4330M steel, 200-220 ksi	Cadmium plate (F-1.1913).
Quadrant (186)	Mg alloy	Dow 17 anodize and apply primer, C00259 (F-3.30) but not in grooves.
Plate (195)	4130 steel, no heat treat	Cadmium plate (F-1.1913).
Plugs (245, 340)	Monel	No finish (F-25.01).
Crank (250)	Al alloy	Chemical treat or chromic acid anodize and apply primer, C00259 (SRF-2.30), but no primer on surfaces shown by flagnote 1 in REPAIR 1-1, Figure 602.
Shaft (275)	4340 steel, 200-220 ksi	Cadmium plate (F-15.02) but not on threads. Apply corrosion inhibiting compound, B50080 (F-19.05) on threads and interior surfaces.
Plate (305)	Al alloy	Chemical treat or chromic acid anodize and apply primer, C00259 (F-18.05).
Shim (310)	Al alloy	No finish (F-25.01).
Eccentric (375)	Al alloy	Chemical treat or chromic acid anodize and apply primer, C00259 (SRF-2.30), but no finish on 2.3105-2.3115-inch outside diameter.

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REPAIR 1-1 Page 602 Jul 01/2009





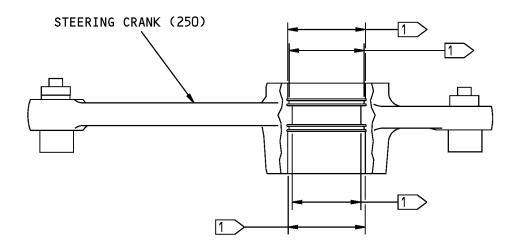
1 DO NOT APPLY PRIMER OR PAINT ON THIS SURFACE.

ITEM NUMBERS REFER TO IPL FIG. 1

275A5105-1 Arm Refinish Figure 601

> 32-51-33 REPAIR 1-1 Page 603 Jul 01/2009





1 DO NOT APPLY PRIMER ON THIS SURFACE. ITEM NUMBERS REFER TO IPL FIG. 1

65-25820-10, -13 Steering Crank Refinish Figure 602

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DRUM AND ECCENTRIC ASSEMBLY - REPAIR 2-1

275A5103-1

1. General

- A. This procedure has the data necessary to repair the drum and eccentric assembly (260).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the True Position Dimensioning Symbols shown in
- D. Refer to IPL Figure 1 for item numbers.

2. Repair Procedures

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
References		

B. F

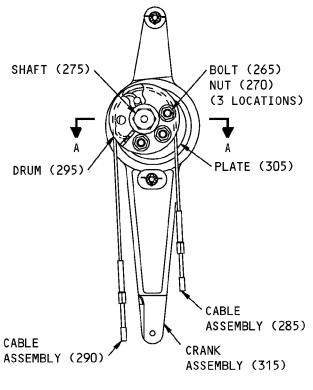
Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-04	MISCELLANEOUS MATERIALS

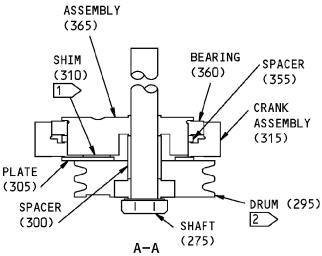
C. Bearing Replacement (REPAIR 2-1, Figure 601)

NOTE: For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Remove the bearing (360) from the drum and eccentric assembly (SOPM 20-50-03).
- (2) Install a replacement bearing (360) in the drum and eccentric assembly with sealant, A00247 per SOPM 20-50-03.
- D. Assembly (REPAIR 2-1, Figure 601)
 - (1) Use standard industry procedures and REPAIR 2-1, Paragraph 2.D.(2).
 - (2) Use the shim noted by flagnote 1 if necessary. Remove 0.003-inch laminations to get a 0.002-0.030 inch gap between the plate and eccentric assembly.







ECCENTRIC

- USE IF AND AS REQUIRED. REMOVE
 0.003 INCH LAMINATIONS TO MAINTAIN
 A 0.002-0.030 GAP BETWEEN THE PLATE
 AND THE ECCENTRIC ASSEMBLY
- 2 INSTALL THE BEARING WITH BMS 5-95 SEALANT PER SOPM 20-50-03

ITEM NUMBERS REFER TO IPL FIG. 1
ALL DIMENSIONS ARE IN INCHES

275A5103-1 Drum and Eccentric Assembly Repair Figure 601

32-51-33

REPAIR 2-1 Page 602 Mar 01/2006



QUADRANT ASSEMBLY - REPAIR 3-1

65-25843-1

1. General

- A. This procedure has the data necessary to repair the quadrant assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Repair Procedures

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-04	MISCELLANEOUS MATERIALS

C. Bearing Replacement

NOTE: For miscellaneous materials, refer to SOPM 20-60-04.

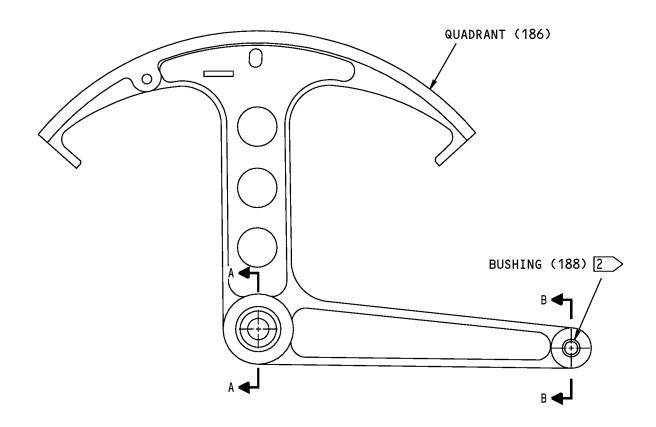
- (1) Remove the bearing (187) from the quadrant assembly (SOPM 20-50-03).
- (2) Install a replacement bearing by the roller-swage method of SOPM 20-50-03 using sealant, A00247.

D. Bushing Replacement

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Remove the bushing (188) from the quadrant assembly (SOPM 20-50-03).
- (2) Install a replacement bushing by the press-fit method of SOPM 20-50-03 using sealant, A00247.
- (3) Make a check of the bore dimension and machine if necessary to design dimensions and finish.







1 ROLLER SWAGE THE BEARING PER SOPM 20-50-03

2 USE PRESS FIT METHOD TO INSTALL THE BUSHING

ITEM NUMBERS REFER TO IPL FIG. 1
ALL DIMENSIONS ARE IN INCHES

65-25843-1 Quadrant Assembly Repair Figure 601

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REPAIR 3-1 Page 602 Mar 01/2006



STEERING ROD ASSEMBLY - REPAIR 4-1

69-20495-1

1. General

- A. This procedure has the data necessary to repair the steering rod assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Parts Replacement

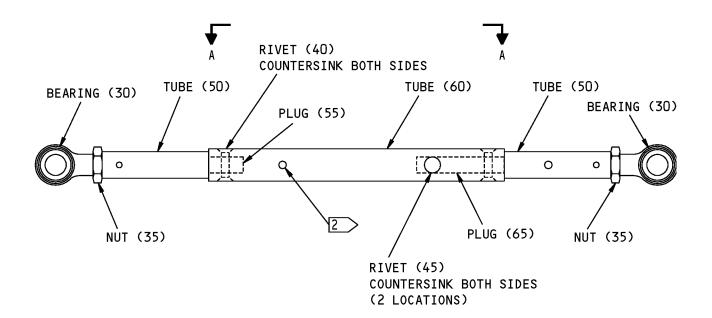
A. References

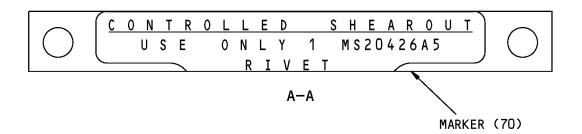
Reference	Title
SOPM 20-50-05	APPLICATION OF ALUMINUM FOIL AND OTHER MARKERS

B. Procedure

- (1) Remove bad bearings (30) from the steering rod assembly (25).
- (2) Use standard industry procedures to install replacement bearings (30) on the tubes (50).
- (3) Assemble the tubes (50, 60, 50) and plugs (55, 65) together with rivets (40, 45). Do not use a different rivet material.
- (4) Install a replacement marker (70), if necessary per SOPM 20-50-05.
- (5) Obey all other flagnotes in REPAIR 4-1, Figure 601.

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- 1 THIS IS A CONTROLLED SHEAROUT RIVET.
 DO NOT SUBSTITUTE THE MATERIAL.
- ITEM NUMBERS REFER TO IPL FIG. 1
- 2 KEEP THIS INSPECTION HOLE OPEN.
- 3 INSTALL THE MARKER PER SOPM 20-50-05.
- 4 APPLY CLEAR SKYDROL RESISTANT EDGE SEAL PER D6-5005.

69-20495-1 Steering Rod Assembly Repair Figure 601

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REPAIR 4-1 Page 602 Mar 01/2006



ASSEMBLY

1. General

- A. This procedure has the data necessary to assemble the rudder pedal steering assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Assembly

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95

B. References

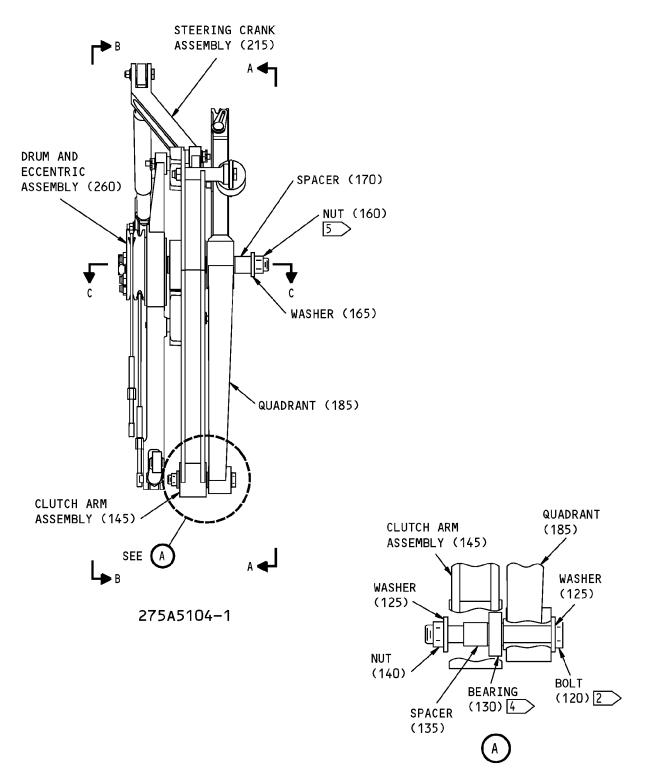
Reference	Title
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-50-01	BOLT AND NUT INSTALLATION
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-04	MISCELLANEOUS MATERIALS

C. Procedure

NOTE: For decoding table for Boeing finish codes, refer to SOPM 20-41-01. For bolt and nut installation, refer to SOPM 20-50-01. For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Use standard industry procedures and ASSEMBLY, Paragraph 2.C.(2).
- (2) Obey the flagnotes in ASSEMBLY, Figure 701 use sealant, A00247 where required (SOPM 20-50-03).



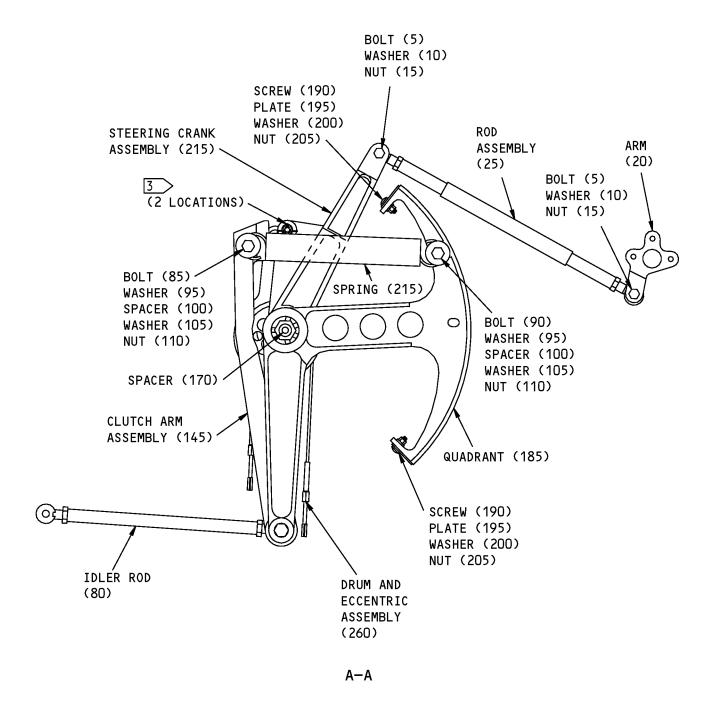


Rudder Pedal Steering Assembly Figure 701 (Sheet 1 of 4)

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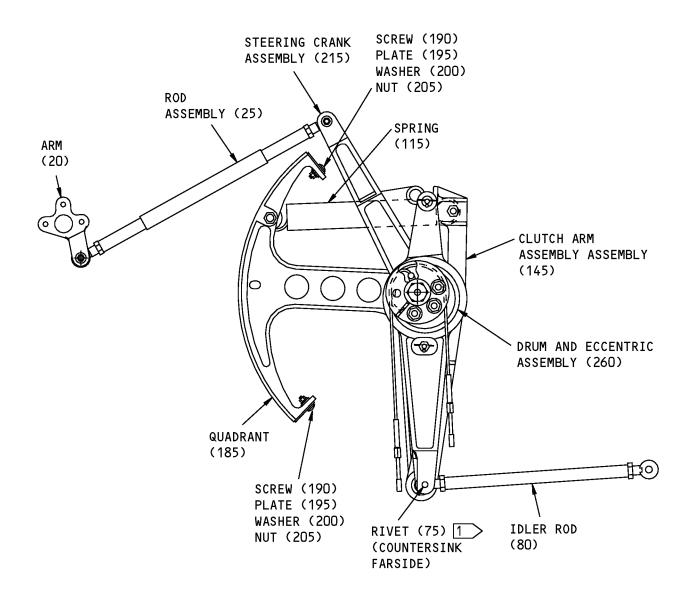


Rudder Pedal Steering Assembly Figure 701 (Sheet 2 of 4)

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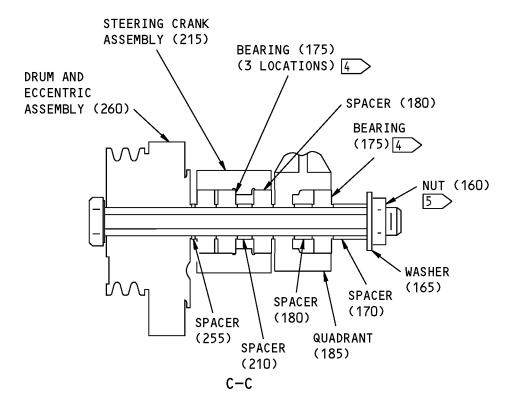
B-B

Rudder Pedal Steering Assembly Figure 701 (Sheet 3 of 4)

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- 1 APPLY BMS 5-95 SEALANT (F-19.48)
 IN THE HOLE BEFORE YOU INSTALL
 THE RIVET
- 2 TIGHTEN TO 100-140 INCH-LBS
- THESE WASHERS (225) CAN BE INSTALLED UNDER THE BEARING FOR BETTER CAM ENGAGEMENT, BUT AT LEAST ONE WASHER MUST BE UNDER THE NUT
- 4 INSTALL THE BEARING WITH BMS 5-95 SEALANT PER SOPM 20-50-03
- TIGHTEN TO 100-125 INCH-LBS ABOVE THE RUNNING TORQUE

ITEM NUMBERS REFER TO IPL FIG. 1
ALL DIMENSIONS ARE IN INCHES

Rudder Pedal Steering Assembly Figure 701 (Sheet 4 of 4)

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FITS AND CLEARANCES

(NOT APPLICABLE)



SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

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ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7

- . Assembly
- . Attaching parts for assembly
- . Detail parts for assembly
- . . Subassembly
- . Attaching parts for subassembly
- . Detail parts for subassembly
- . . . Sub-subassembly
- . . . Attaching parts for subassembly
- . Details parts for sub-subassembly

Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
 - (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
 - (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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Optional The part is optional to and interchangeable with other parts

(OPT) that have the same item number.

Replaces, Replaced by and not

interchangeable with

(REPLACES, REPLACED BY AND

NOT INTCHG/W)

Replaces, Replaced by (REPLACES, REPLACED BY)

The part replaces and is not interchangeable with the initial

part.

The part replaces and is interchangeable with, or is an

alternative to, the initial part.

VENDOR CODES

Code	Name
06144	INDUSTRIAL TECTONICS BEARING CORP 18301 SOUTH SANTA FE AVENUE RANCHO DOMINGUEZ, CALIFORNIA 90221 FORMERLY IN COMPTON, CALIFORNIA
11815	CHERRY AEROSPACE FASTENERS DIV OF TEXTRON 1224 EAST WARNER AVENUE PO BOX 2157 SANTA ANA, CALIFORNIA 92707-0157 FORMERLY IN LOS ANGELES, CALIF, FORMERLY CHERRY FASTENERS TOWNSEND DIV OF TEXTRON INC V71087
15653	ALCOA GLOBAL FASTENERS INC DIV KAYNAR PRODUCTS 800 S STATE COLLEGE BLVD FULLERTON, CALIFORNIA 92831-3001 FORMERLY VK6405 MICRODOT AEROSP LTD; FORMERLY KAYNAR TECH FORMERLY FAIRCHILD FASTENERS KAYNAR DIV
21335	TIMKEN US CORPORATION DIV FAFNIR 336 MECHANIC STREET LEBANON, NH 03766-0267 FORMERLY FAFNIR BRG AND TEXTRON INC FAFNIR DIV IN NEW BRITAIN, CONNECTICUT; FORMERLY TORRINGTON CO THE SPECIAL PRODUCTS DIV SUB OF THE INGERSOLL-RAND CO V8D210 FORMERLY TORRINGTON CO FAFNIR BEARING DIV IN TORRINGTON, CT
21760	SCHATZ BEARING CORP 10 FAIRVIEW AVENUE PO BOX 1191 POUGHKEEPSIE, NEW YORK 12601-1312 FORMERLY FEDERAL BRG CO AND SCHATZ MFG CO V53268 FORMERLY SCHATZ MFG CO

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Code	Name
30163	VALENTEC DAYRON INC 333 MAGUIRE BLVD PO BOX 140394 ORLANDO, FLORIDA 32814-0394
38443	MRC BEARINGS 402 CHANDLER STREET JAMESTOWN, NEW YORK 14701-3802 FORMERLY MARLIN-ROCKWELL CORP DIV TRW AND TRW INC
40920	MPB MINIATURE PRECISION BEARING DIV PRECISION PARK PO BOX 547 KEENE, NEW HAMPSHIRE 03431 FORMERLY MPB CORP AND MINIATURE BRG DIV MPB CORP
43991	FAG BEARING INCORPORATED 118 HAMILTON AVENUE STAMFORD, CONNECTICUT 06904 FORMERLY NORMA-HOFFMAN BEARING CORPORATION FORMERLY NORMA FAG BEARINGS CORPORATION
52828	REPUBLIC FASTENER MFG CORP 1300 RANCHO CONEJO BLVD NEWBURY PARK, CALIFORNIA 91320-1405 FORMERLY IN SYLMAR, CALIFORNIA
62554	SIMMONDS MECAERO FASTENERS INC 1734 SEQUOIA AVENUE ORANGE, CALIFORNIA 92668
72962	HARVARD INDUSTRIES INC 3 WERNER WAY SUITE 210 LEBANON, NEW JERSEY 08833 FORMERLY ESNA V7A079 FORMERLY ELASTIC STOP NUT IN UNION, NJ
80539	SPS TECHNOLOGIES INC DIV AERPSOACE - SANTA ANA 2701 SOUTH HARBOR BOULEVARD SANTA ANA, CALIFORNIA 92704-5803 FORMERLY NUTT-SHEL DIV OF SPC WESTERN CO V80539 AND STANDARD PRESSED STEEL WESTERN DIV V17279

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Code	Name
83086	NEW HAMPSHIRE BALL BEARING, INC HITECH DIVISION 172 JAFFREY ROAD PETERBOROUGH, NEW HAMPSHIRE 03458
F0224	SIMMONDS SA FAIRCHILD FASTENERS ST COSME ST COSME EN VAIRAIS F-72580, FRANCE
K8455	RHP BEARINGS PLC RHP AEROSPACE OLDENDS LANE STONEHOUSE GL10 3RM UK



NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
102LH9075-4W		1	110	2
		1	270	3
102LH9075-5W		1	140	1
275A5103-1		1	260	1
275A5104-1		1	1A	RF
275A5104-2		1	1B	RF
275A5105-1		1	20	1
65-25820-10		1	250	1
65-25820-11		1	215	1
65-25820-13		1	250A	1
65-25843-1		1	185	1
65-25843-2		1	186	1
65-25844-8		1	315	1
65-25844-9		1	350	1
65-25861-6		1	145	1
65-25861-7		1	155	1
66-1354-2		1	245	AR
		1	340	AR
66-15669-1		1	275	1
66-15707-1		1	195	2
66-15736-1		1	170	1
66-15736-2		1	180	1
66-15736-3		1	210	1
66-15736-4		1	300	1
66-15736-5		1	255	1
66-15736-6		1	355A	1
66-15736-7		1	135	1
66-25793-1		1	120	1
69-20285-2		1	295	1
69-20309-1		1	365	1
69-20309-2		1	375	1
69-20427-1		1	305	1
69-20495-1		1	25	1
69-20495-2		1	50	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-20495-3		1	60	1
69-20495-4		1	55	1
69-20495-5		1	65	1
69-20539-2		1	115	1
69-58176-1		1	310	AR
AN201KP5A		1	150	1
AN201KP8A		1	187	1
		1	370	1
AN316-6R		1	35	2
AN316-8R		1	160	1
AN320-3		1	230	2
		1	330	2
AN380-2-2		1	220	2
		1	320	2
AN960D10L		1	225	6
AN960PD10L		1	325	2
B546-2TS		1	360	1
B546DD		1	345	1
		1	345	1
		1	360	1
B546DDFS101		1	360	1
B546DDFS428		1	360	1
B546DDNJC		1	360	1
B546DDP		1	360	1
B546FS101		1	360	1
B546ZZ		1	345	1
BACB10A32DD		1	345	1
BACB10A435		1	30	2
BACB10AF3F5H		1	240	1
BACB10AF3F7H		1	235	1
		1	335	2
BACB10BX5		1	130	1
BACB10BX8		1	175	4
		1	280	1
BACB10CF37PP		1	360	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB30LU4-21		1	265	3
BACB30NF4-13		1	90	1
BACB30NF4-30		1	85	1
BACB30NM3K10		1	5	2
BACC2A3A00097CH		1	290	1
BACC2A3A00110DH		1	285	1
BACM10L1CWM		1	70	1
BACN10JC4CD		1	110	2
		1	270	3
BACN10JC5CD		1	140	1
BACN10YR3CD		1	15	2
		1	205	2
BACR15BA6D		1	75	1
BACR24N4B95		1	80	1
BACS12GU3K9		1	190A	2
BACW10P128A		1	95	4
BRH10C4D		1	110	2
		1	270	3
BRH10C5D		1	140	1
CS205E		1	130	1
CS208E		1	175	4
		1	280	1
H51650-4BAC		1	110	2
		1	270	3
H51650-5BAC		1	140	1
KP5A		1	130	1
		1	130	1
KP5A2TS		1	130	1
KP5AFS428		1	130	1
KP5AG27		1	130	1
KP5ALY196		1	130	1
KP5ANJC		1	130	1
KP5ASD610		1	130	1
KP8A		1	175	4
		1	175	4

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	280	1
		1	280	1
KP8A2TS		1	175	4
		1	280	1
KP8AFS428		1	175	4
		1	175	4
		1	280	1
		1	280	1
KP8AG27		1	175	4
		1	280	1
KP8ANJC		1	175	4
		1	280	1
KP8BSD610		1	175	4
		1	280	1
LLKP5A		1	130	1
LLKP8A		1	175	4
		1	280	1
MS20426A5		1	40	1
MS20426B5		1	45	2
MS27111-5		1	165	1
NAS1149C0532R		1	125	2
NAS1149D0316J		1	10	2
NAS1149D0363J		1	200	2
NAS1149D0416J		1	105	2
NAS43DD4-7FC		1	100	2
NAS538-5P78		1	188	1
NAS623-3-4		1	190	2
NS202486-048		1	110	2
		1	270	3
PLH53CD		1	15	2
		1	15	2
		1	205	2
		1	205	2
REP3M6-2N1		1	30	2
REP3M62NFS210		1	30	2

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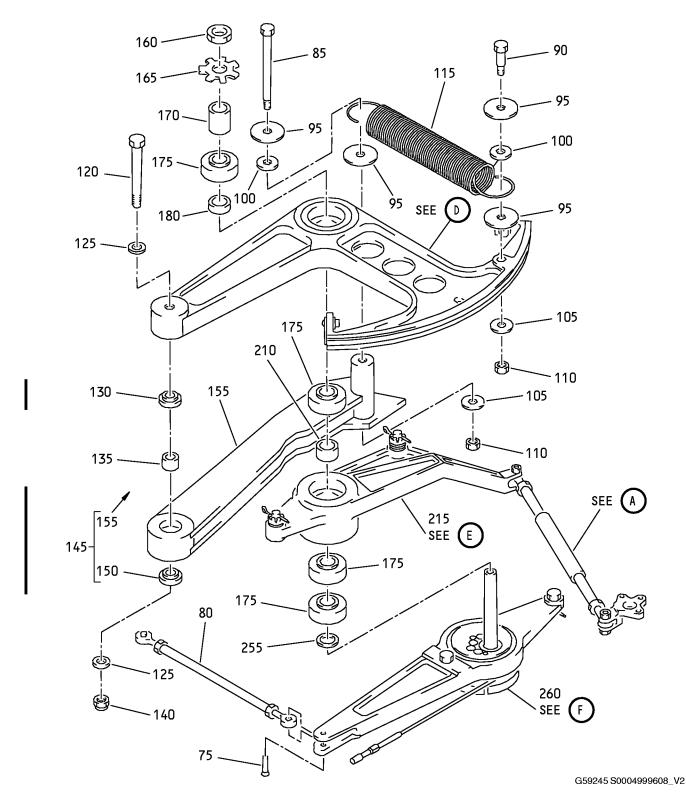
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I	PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
	T346E		1	360	1
	T6C428JCD		1	110	2
			1	270	3
	T6C524JCD		1	140	1

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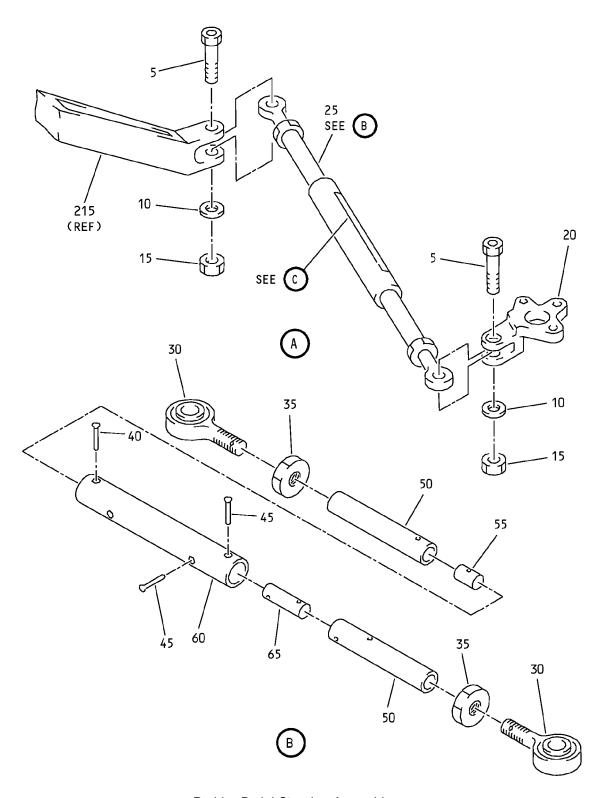




Rudder Pedal Steering Assembly IPL Figure 1 (Sheet 1 of 5)

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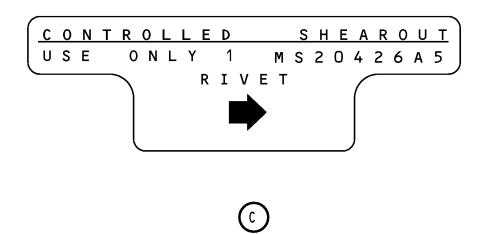


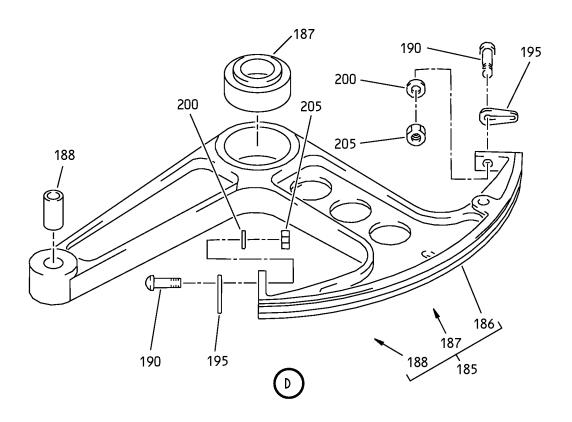
Rudder Pedal Steering Assembly IPL Figure 1 (Sheet 2 of 5)

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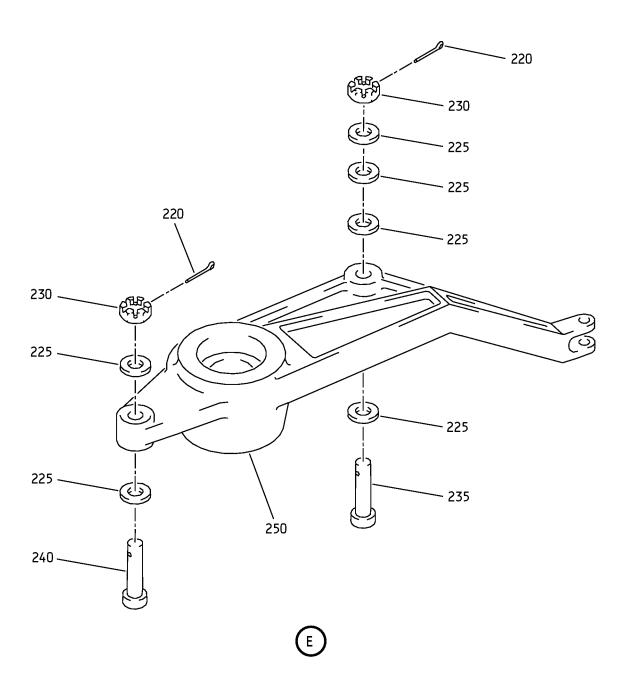




Rudder Pedal Steering Assembly IPL Figure 1 (Sheet 3 of 5)

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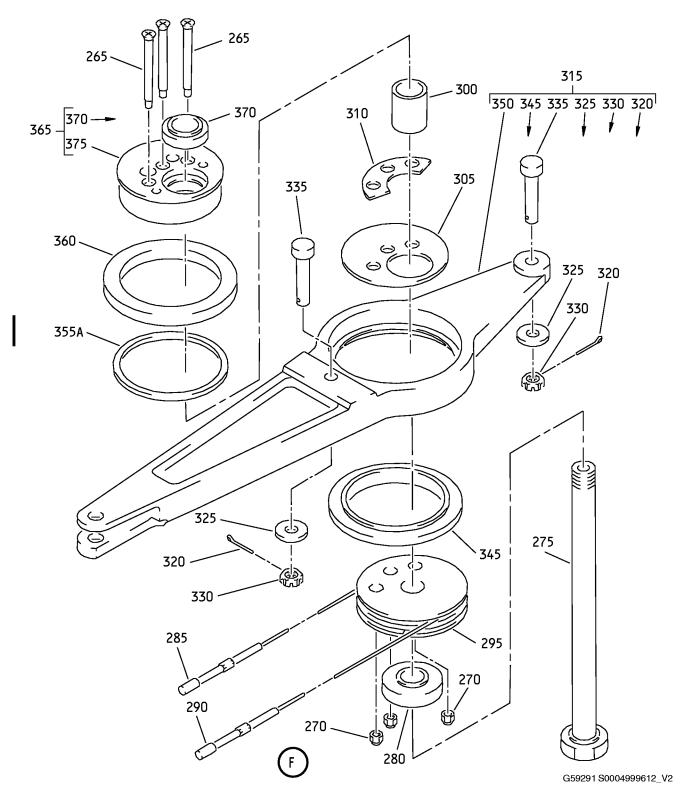




Rudder Pedal Steering Assembly IPL Figure 1 (Sheet 4 of 5)

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Rudder Pedal Steering Assembly IPL Figure 1 (Sheet 5 of 5)

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	FIG/	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
	1–					
ı	-1A	275A5104-1		RUDDER PEDAL ASSY-STEERING	Α	RF
ı	–1B	275A5104-2		RUDDER PEDAL ASSY-STEERING	В	RF
	5	BACB30NM3K10		. BOLT		2
	10	NAS1149D0316J		. WASHER		2
	15	PLH53CD		. NUT (VF0224) (SPEC BACN10YR3CD) (OPT PLH53CD (V62554))		2
	20	275A5105-1		. ARM		1
	25	69-20495-1		. ROD ASSY		1
	30	REP3M62NFS210		BEARING-ROD END (V21335) (SPEC BACB10A435) (OPT REP3M6-2N1 (V38443))		2
	35	AN316-6R		NUT		2
	40	MS20426A5		RIVET (SIZE DETERMINED ON INST)		1
	45	MS20426B5		RIVET (SIZE DETERMINED ON INST)		2
	50	69-20495-2		TUBE		2
	55	69-20495-4		PLUG		1
	60	69-20495-3		TUBE-LINK		1
	65	69-20495-5		PLUG		1
	70	BACM10L1CWM		MARKER-ALUMINUM FOIL		1
	75	BACR15BA6D		. RIVET (SIZE DETERMINED ON INST)		1
	80	BACR24N4B95		. ROD-IDLER		1
	85	BACB30NF4-30		. BOLT		1
	90	BACB30NF4-13		. BOLT		1
	95	BACW10P128A		. WASHER		4
	100	NAS43DD4-7FC		. SPACER		2
	105	NAS1149D0416J		. WASHER		2

-Item not Illustrated

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
110	BRH10C4D		. NUT		2
115	69-20539-2		. SPRING		1
120	66-25793-1		. BOLT		1
125	NAS1149C0532R		. WASHER		2
130	KP5ANJC		. BEARING		1
135	66-15736-7		. SPACER		1
140	BRH10C5D		. NUT (V52828) (SPEC BACN10JC5CD) (OPT T6C524JCD (V11815)) (OPT 102LH9075-5W (V72962)) (OPT H51650-5BAC (V15653))		1
145	65-25861-6		. ARM ASSY		1
150	AN201KP5A		BEARING		1
155	65-25861-7		ARM		1
160	AN316-8R		. NUT		1
165	MS27111-5		. WASHER		1
170	66-15736-1		. SPACER		1



	FIG/	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
	1–					
	175	KP8AFS428		. BEARING		4
	180	66-15736-2		. SPACER		1
	185	65-25843-1		. QUADRANT ASSY-RUDDER PEDAL STEERING		1
	186	65-25843-2		QUADRANT		1
	187	AN201KP8A		BEARING		1
	188	NAS538-5P78		BUSHING		1
I	190	NAS623-3-4		. SCREW	Α	2
I	-190A	BACS12GU3K9		. SCREW	В	2
	195	66-15707-1		. PLATE-CABLE END		2
	200	NAS1149D0363J		. WASHER		2
	205	PLH53CD		. NUT (VF0224) (SPEC BACN10YR3CD) (OPT PLH53CD (V62554))		2
	210	66-15736-3		. SPACER		1
	215	65-25820-11		. CRANK ASSY		1
	220	AN380-2-2		PIN		2
	225	AN960D10L		WASHER		6
	230	AN320-3		NUT		2
	235	BACB10AF3F7H		BEARING		1
	240	BACB10AF3F5H		BEARING		1
	-245	66-1354-2		PLUG		AR
	250	65-25820-10		CRANK (OPT ITEM 250A)		1

-Item not Illustrated

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FIG/	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1-					
-250A	65-25820-13		CRANK (OPT ITEM 250)		1
255	66-15736-5		. SPACER		1
260	275A5103-1		. DRUM AND ECCENTRIC ASSY		1
265	BACB30LU4-21		BOLT		3
270	BRH10C4D		NUT (V52828) (SPEC BACN10JC4CD) (OPT T6C428JCD (V11815)) (OPT NS202486-048 (V80539)) (OPT 102LH9075-4W (V72962)) (OPT H51650-4BAC (V15653))		3
275	66-15669-1		SHAFT		1
280	KP8AFS428		BEARING (V06144) (SPEC BACB10BX8) (OPT KP8AFS428 (V21335)) (OPT KP8A2TS (V43991)) (OPT LLKP8A (V38443)) (OPT KP8AG27 (V30163)) (OPT KP8A (V38443)) (OPT KP8BSD610 (V83086)) (OPT CS208E (VK8455)) (OPT KP8A (V21760)) (OPT KP8ANJC (V06144))		1
285	BACC2A3A00110DH		CABLE ASSY		1
290	BACC2A3A00097CH		CABLE ASSY		1
295	69-20285-2		DRUM		1
300	66-15736-4		SPACER		1
305	69-20427-1		PLATE		1
310	69-58176-1		SHIM		AR
315	65-25844-8		CRANK ASSY-CLUTCH		1
320	AN380-2-2		PIN		2
325	AN960PD10L		WASHER		2
330	AN320-3		NUT		2
335	BACB10AF3F7H		BEARING		2
-340	66-1354-2		PLUG		AR

-Item not Illustrated

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ILLUSTRATED PARTS LIST
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	FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
	1–					
	345	B546DD		BEARING (V38443) (SPEC BACB10A32DD) (OPT B546DD (V21335)) (OPT B546ZZ (V38443))		1
	350	65-25844-9		CRANK		1
I	-355	66-15836-6		DELETED		
I	355A	66-15736-6		SPACER		1
	360	B546DDNJC		BEARING (V06144) (SPEC BACB10CF37PP) (OPT B546-2TS (V43991)) (OPT B546DDFS428 (V21335)) (OPT T346E (VK8455)) (OPT B546DDFS101 (V06144)) (OPT B546DD (V38443)) (OPT B546FS101 (V06144)) (OPT B546DDP (V21760))		1
	365	69-20309-1		ECCENTRIC ASSY		1
	370	AN201KP8A		BEARING		1
	375	69-20309-2		ECCENTRIC		1