



COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

ELEVATOR ASSEMBLY AND TAB ASSEMBLY

PART NUMBER

**65C25746–119SP, –120SP, –123SP, –124SP, –129SP,
–130SP, –131SP, –132SP, –141SP, –142SP, –153SP,
–154SP, –19SP, –1SP, –20SP, –21SP, –22SP, –27SP,**

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-81SP, -82SP, -83SP, -84SP, -95SP, -96SP,
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COMPONENT MAINTENANCE MANUAL

Revision No. 37
Jul 01/2009

To: All holders of ELEVATOR ASSEMBLY AND TAB ASSEMBLY 55-20-16.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

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Location of Change

Description of Change

NO HIGHLIGHTS

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HIGHLIGHTS

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A = Added, R = Revised, D = Deleted, O = Overflow

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1079	Mar 01/2006	1098.19	Mar 01/2008		
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1082	Mar 01/2006	1098.22	Mar 01/2008		
1083	Mar 01/2006	1098.23	Mar 01/2008		
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1085	Mar 01/2006	1098.25	Mar 01/2008		
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1098.11	Mar 01/2008				
1098.12	Mar 01/2008				
1098.13	Mar 01/2008				
1098.14	Mar 01/2008				
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TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR 33200-2	MAR 05/85
		PRR 33200-6	MAR 05/85
		PRR 33200-7	MAR 05/85
		PRR 33200-8	MAR 05/85
		PRR 33200-9	MAR 05/85
55-1035		PRR 33200-11	SEP 05/85
		PRR 33180-72	MAR 05/85
		PRR 33180-83	MAR 05/85
		PRR 33200-13	JUN 05/85
		PRR 33200-14	JUN 05/85
55-1035			SEP 05/85
		PRR 33600-18	SEP 05/85
55-1036			DEC 05/87
55-1040		PRR 34019	DEC 05/87
55-1040R4			MAR 05/92
		PRR 33976-1	DEC 05/87
51-1011			DEC 05/87
		PRR 33600-37	DEC 05/87
		PRR 33750-13	DEC 05/87
		PRR 33750-32	DEC 05/87
		PRR 33600-59	DEC 05/87
		PRR 34121	DEC 05/87
55-1043			JUN 05/89
		PRR 34502	DEC 05/89
		PRR 34620	DEC 05/89
		PRR 34808	MAR 05/92
		PRR 34535	MAR 05/92
		PRR 34792	SEP 05/92
55-1049		PRR 35005-16	JUN 01/95
55-1054			JUN 01/95

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TR AND SB RECORD

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BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
55-1054R1			SEP 01/95
		PRR 35266	JUN 01/96
55-1049R1			JUL 01/01
55-1054R2			JUL 01/01
55A1070			NOV 01/02
55A1070R2	55-4		MAR 01/08

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TR AND SB RECORD

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COMPONENT MAINTENANCE MANUAL

INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

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INTRODUCTION

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COMPONENT MAINTENANCE MANUAL

ELEVATOR AND TAB ASSEMBLY - DESCRIPTION AND OPERATION

1. Description

A. The elevator and tab assembly has graphite/epoxy front and rear spars, upper and lower panels, and includes access doors, balance weights, ribs and six stabilizer attach points.

2. Operation

A. The elevator is operated hydraulically to control pitch of the airplane.

3. Leading Particulars (Approximate)

- A. Length – 212 inches
- B. Width – 40 inches
- C. Thickness – 10 inches
- D. Weight – 165 pounds

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DESCRIPTION AND OPERATION

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TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

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TESTING AND FAULT ISOLATION

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DISASSEMBLY

1. General

- A. Use this procedure to disassemble the elevator and tab assemblies.
- B. Disassemble this component only as necessary to complete fault isolation, to find the serviceability of parts, to do the repairs, and to put the unit back in serviceable condition
- C. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- D. Refer to IPL Figure 1 thru IPL Figure 7 for the item numbers.

2. Disassembly procedures

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95

- B. References

Reference	Title
SOPM 20-10-08	REMOVAL OF FAYING SURFACE SEALED METAL FITTINGS FROM COMPOSITE STRUCTURES

- C. Procedure

- (1) Use standard industry practices and these steps.

CAUTION: COMPONENTS OF BEARING PLATE ASSEMBLY (40, 130, IPL FIG. 2), OTHER THAN FASTENERS, ARE A MATCHED SET. CAP (120, 200) AND PILLOW BLOCK (125, 205) OR CAP (120, 200) AND PILLOW BLOCK (127, 206) ARE A MATCHED SET.

- (2) Make a note of the thickness of shims (20, 110, IPL Figure 3; 150, 220, IPL Figure 4) when you remove fitting assemblies (30, 120, IPL Figure 3; 105, 107, 165, 235, IPL Figure 4) to help during assembly.

CAUTION: IF YOU WILL ADD OR REPLACE HINGE FITTING (165D, 165E) 65C25798-11 OR -12, REFER TO SB 737-55-1049 FOR DISASSEMBLY PROCEDURES TO KEEP HOLE LOCATIONS AND BEARING ALIGNMENT.

- (3) Remove bearing plate assembly (40, 130, IPL Figure 2) and disassemble bolts (80, 82, 165), washers (85, 90, 170, 175), and nuts (100, 102, 180) to separate matched set cap (120, 200) and pillow block (125, 127, 205, 206).

NOTE: See REPAIR 13-1 for assembly of matched-set bearing plate assembly (40, 130, IPL Figure 2).

- (4) Refer to SOPM 20-10-08 to remove fittings (540, 705, 768, 820, 858A, 896, IPL Figure 2; 30, 120, IPL Figure 3; 165, 135, 136, 137, 138, 235, IPL Figure 4), bracket (585, 645, IPL Figure 2) because these parts are sealed with sealant, A00247.

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DISASSEMBLY

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CLEANING

1. General

- A. This procedure has the data necessary to clean the elevator assembly and tab assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 thru IPL Figure 7 for the item numbers.

2. Cleaning Procedures

A. References

Reference	Title
SOPM 20-30-01	CLEANING AND RELUBRICATING BEARINGS
SOPM 20-30-03	GENERAL CLEANING PROCEDURES

B. Procedure

CAUTION: DO NOT VAPOR DEGREASE EPOXY BONDED STRUCTURES WITH CHLORINATED CLEANING AGENTS SUCH AS METHYLENE CHLORIDE, TRICHLOROETHYLENE, AND TRICHLOROETHANE. CHLORINATED CLEANING AGENTS WILL CAUSE DAMAGE TO EPOXY BONDED STRUCTURES. 1,1,1-TRICHLOROETHANE IS ONE OF THE SOLVENTS ALLOWED FOR CLEANING COMPOSITE COMPONENTS. DO NOT SUBMERGE PARTS IN THE SOLVENT OR ALLOW STANDING SOLVENT ON THE PARTS OR DAMAGE MAY OCCUR. USE 1,1, 1-TRICHLOROETHANE ONLY AS A WIPE SOLVENT.

- (1) Clean all parts by standard industry practices and the instructions in SOPM 20-30-03.
- (2) Use only the special procedure shown to clean bearings (105, 185, 250, 295, 300, 345, 350, 400, 405, IPL Figure 2; 120, IPL Figure 4).
 - (a) Clean the bearings by the special procedure for teflon-lined bearings specified in SOPM 20-30-01.
 - (b) The liners in teflon-lined self-lubricating bearings, and bushings can be damaged if not cleaned correctly.

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CLEANING

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CHECK

1. General

- A. This procedure has the data necessary to check the elevator assembly and tab assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 thru IPL Figure 7 for the item numbers.

2. Check Procedures

A. References

Reference	Title
SOPM 20-20-01	MAGNETIC PARTICLE INSPECTION
SOPM 20-20-02	PENETRANT METHODS OF INSPECTION
737 SRM Chapter 55	Structural Repair Manual

B. Procedures

- (1) Examine all parts for defects by standard industry practices. Do the magnetic particle check or penetrant check if the visual check shows possible damage on the parts listed.
- (2) Magnetic particle check (SOPM 20-20-01) – pillow block (125, 127, 205, 206, IPL Figure 2) cap (120, 200, IPL Figure 2).
- (3) Penetrant Check (SOPM 20-20-02) – fitting (570, 750, 795, 838, 878, 922, IPL Figure 2), (235, IPL Figure 4), failsafe strap (515, IPL Figure 2), bearing plate (255, 305, 355, 410, 411J, 411K, 412J, 412K, 413J, 413K, 414J, 414K, 416J, 416T, 417J, 417T, 418J, 418T, 419J, 419T, IPL Figure 2).
- (4) Examine honeycomb and bonded parts for signs of delamination, internal water, scratches, and contour defect.
- (5) If you see delamination or impact damage, do an ultrasonic inspection to find all of the damage.
- (6) If you think there is delamination, do an ultrasonic inspection to find the possible damage.
- (7) Radiographically examine areas that could contain water to see how much damage there is.
- (8) Examine the edges of panels carefully for cuts and abrasions. Delamination starts very easily from damage to an edge member of the honeycomb panel.
- (9) Refer to the applicable 737 SRM Chapter 55 for damage limits.

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CHECK

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REPAIR

1. Content

A. Repair, refinish, and replacement procedures are included in separate repair sections as follows:

Table 601:

P/N	NAME	REPAIR
65C25759	HINGE FITTING ASSEMBLY	1-1
65C25761	HINGE FITTING ASSEMBLY	2-1
65C25763	HINGE FITTING ASSEMBLY	3-1
65C25765	HINGE FITTING ASSEMBLY	4-1
65C25767	HINGE FITTING ASSEMBLY	5-1
65C31307	HINGE FITTING ASSEMBLY	5-1
65C25769	HINGE FITTING ASSEMBLY	6-1
65C25773	HINGE FITTING ASSEMBLY	7-1
65C25798	HINGE FITTING ASSEMBLY	8-1
65C25799	HINGE FITTING ASSEMBLY	9-1
65C26385	BEARING PLATE ASSEMBLY	10-1
65C26386	BEARING PLATE ASSEMBLY	11-1
65C26481	BEARING PLATE ASSEMBLY	12-1
65C26482	BEARING PLATE ASSEMBLY	12-1
65C26529	BEARING PLATE ASSEMBLY	13-1
65C27301	BEARING PLATE ASSEMBLY	13-1
65C27351	BRACKET ASSEMBLY	14-1
65C27376	MAST ARM FITTING ASSEMBLY	15-1
65C28142	HINGE FITTING ASSEMBLY	16-1
- - -	MISCELLANEOUS PARTS REFINISH	17-1
69-44682	BOLT	18-1
69-63391	BOLT	19-1
65C32050	BEARING PLATE ASSEMBLY	20-1
65C32051	HINGE FITTING ASSEMBLY	21-1
65C34548 65C34549	BEARING PLATE ASSEMBLY	22-1

2. Standard Practices

- A. Refer to these standard practices, as applicable, for details of procedures in the individual repairs.
- SOPM 20-00-00 Introduction
 - SOPM 20-10-03 Shot Peening
 - SOPM 20-10-04 Grinding of Chrome Plated Parts
 - SOPM 20-30-02 Stripping of Protective Finishes

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REPAIR - GENERAL

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- SOPM 20-30-03 General Cleaning Procedures
- SOPM 20-41-01 Decoding Table for Boeing Finish Codes
- SOPM 20-41-02 Application of Chemical and Solvent Resistant Finishes
- SOPM 20-42-03 Hard Chrome Plating
- SOPM 20-42-05 Bright Cadmium Plating
- SOPM 20-43-01 Chromic Acid Anodizing
- SOPM 20-43-03 Chemical Conversion Coating for Aluminum
- SOPM 20-50-03 Bearing and Bushing Replacement
- SOPM 20-50-12 Application of Adhesives
- SOPM 20-60-02 Finishing Materials
- SOPM 20-60-03 Lubricants
- SOPM 20-60-04 Miscellaneous Materials

3. Materials

NOTE: Equivalent substitutes can be used.

- A. adhesive, A00028 Adhesive – Type 70 (SOPM 20-50-12)
- B. coating, C00700 Enamel – BMS 10-60, Type 1 BAC707 gray gloss (SOPM 20-60-02)
- C. Grease – (SOPM 20-60-03)
 - (1) grease, D00015 BMS 3-24
 - (2) grease, D00633 BMS 3-33
 - (3) grease, D00013 MIL-G-23827
- D. Primers (SOPM 20-60-02)
 - (1) primer, C00259 BMS 10-11, Type 1
 - (2) urethane-compatible, corrosion-resistant primer, C50080 BMS 10-79, Type 1
 - (3) primer, C00175 BMS 10-79, Type 3
- E. Sealant – sealant, A00247 BMS 5-95 (SOPM 20-60-04)

4. Standard True Position Dimensioning Symbols

- A. Standard True Position Dimensioning Symbols used in applicable repair procedures are shown in SOPM 20-00-00.

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REPAIR - GENERAL

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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 1-1

65C25759-1

1. General

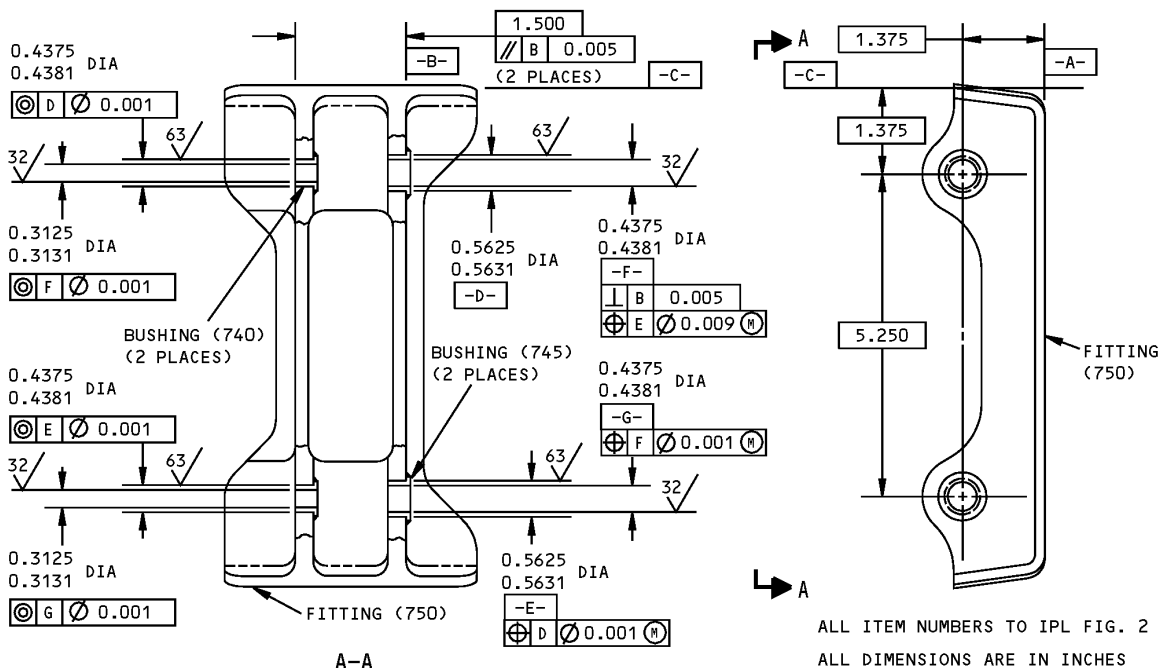
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to(IPL Figure 2) for the item numbers.

2. Bushing (740, 745) Replacement

- A. Remove bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 1-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (750) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (705) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



Bushing Replacement
Figure 601

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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 2-1

65C25761-1

1. General

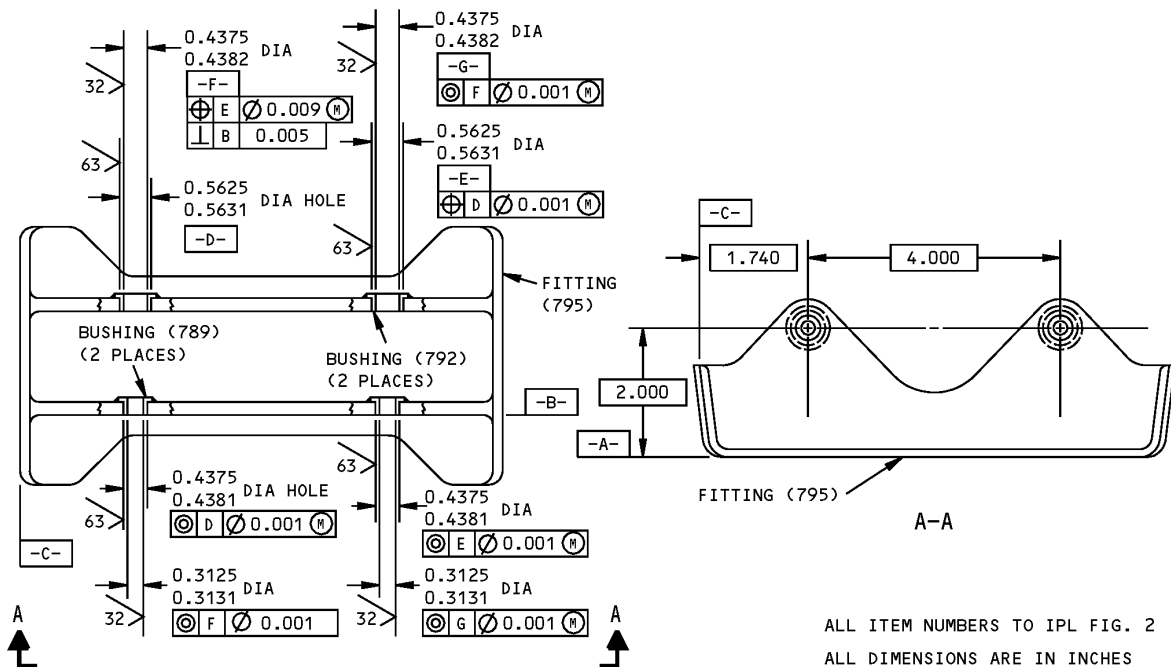
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 2) for the item numbers.

2. Bushing (789, 792) Replacement

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 2-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (795) – Chromic acid anodize, Type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (768) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



Bushing Replacement
Figure 601

ALL ITEM NUMBERS TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES

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REPAIR 2-1
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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 3-1

65C25763-1

1. General

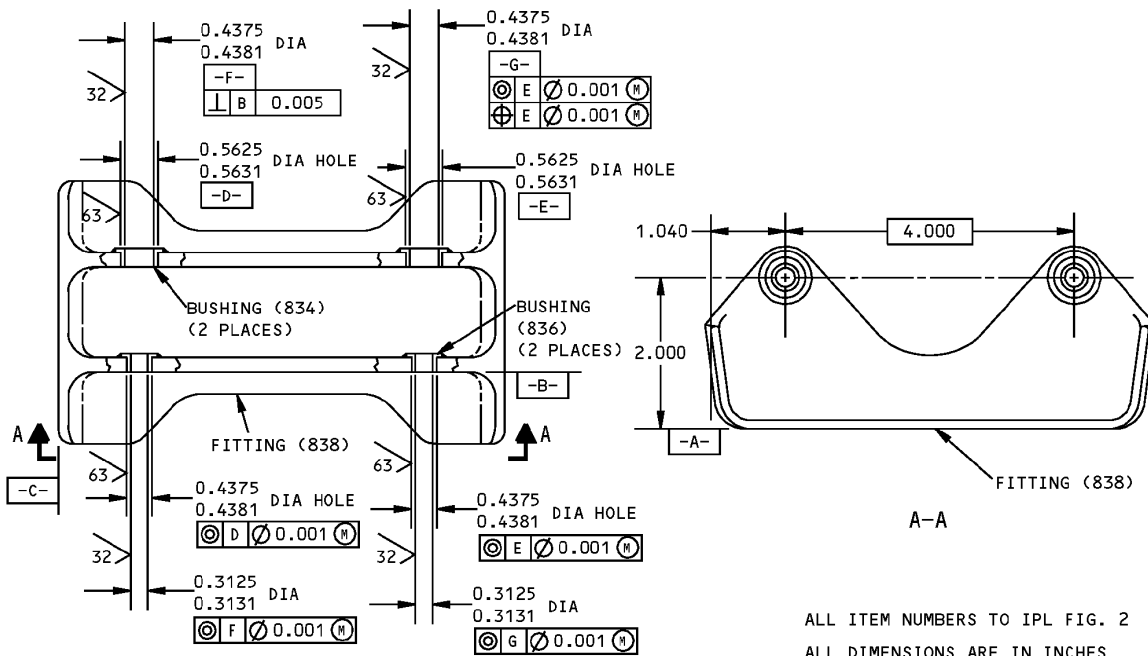
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to IPL Figure 2 for the item numbers.

2. Bushing (834, 836) Replacement

- A. Remove bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 1-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (838) – Chromic acid anodize, Type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (820) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



Bushing Replacement
Figure 601

55-20-16



COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 4-1

65C25765-1

1. General

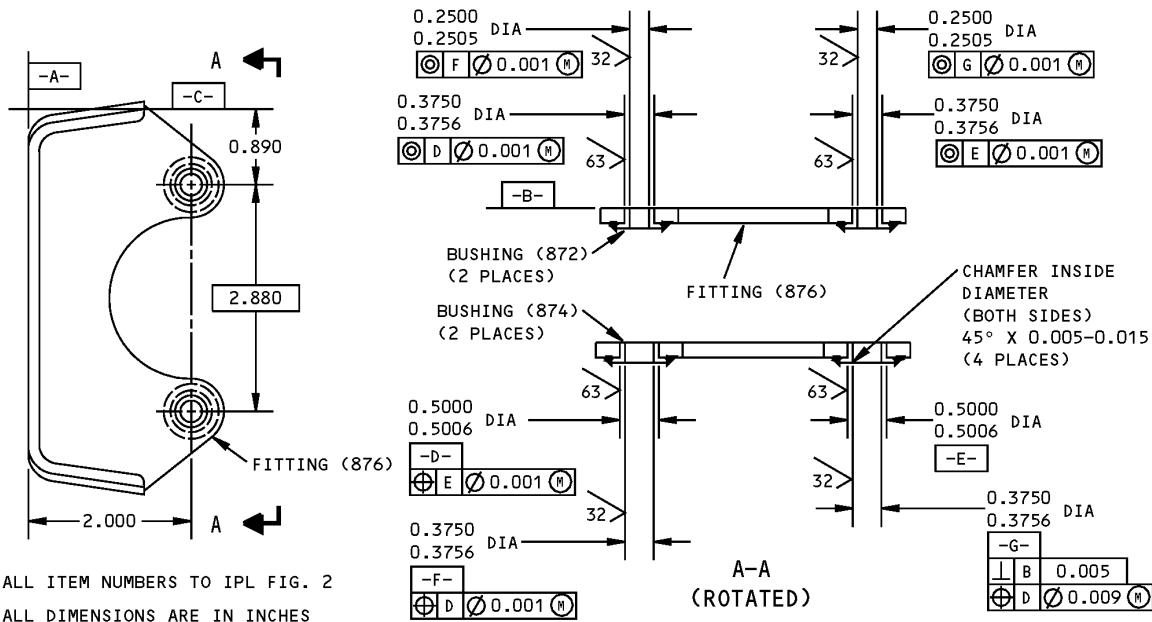
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to(IPL Figure 2) for the item numbers.

2. Bushing (872, 874) Replacement

- A. Remove the bushings.
- B. Install the new bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 1-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (876) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (858) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



ALL ITEM NUMBERS TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES

Bushing Replacement
Figure 601

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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 5-1

65C25767-1, 65C31307-7, -9, -11

1. General

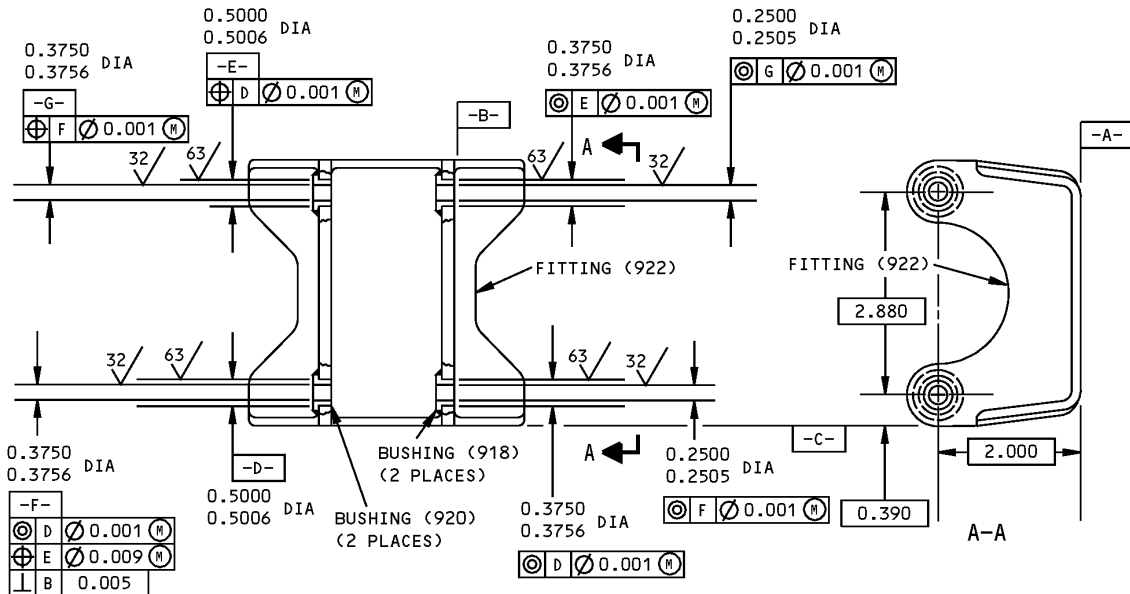
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to(IPL Figure 2) for the item numbers.

2. Bushing (918, 920) Replacement

- A. Remove the bushings.
- B. Install the new bushings using wet sealant, A00247 as specified in SOPM 20-50-03.
- C. Machine bushings per REPAIR 1-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (922) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (896) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



ALL ITEM NUMBERS TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES

Bushing Replacement
Figure 601

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REPAIR 5-1
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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 6-1

65C25769-1, -2, -7, -8, -13, -15

1. General

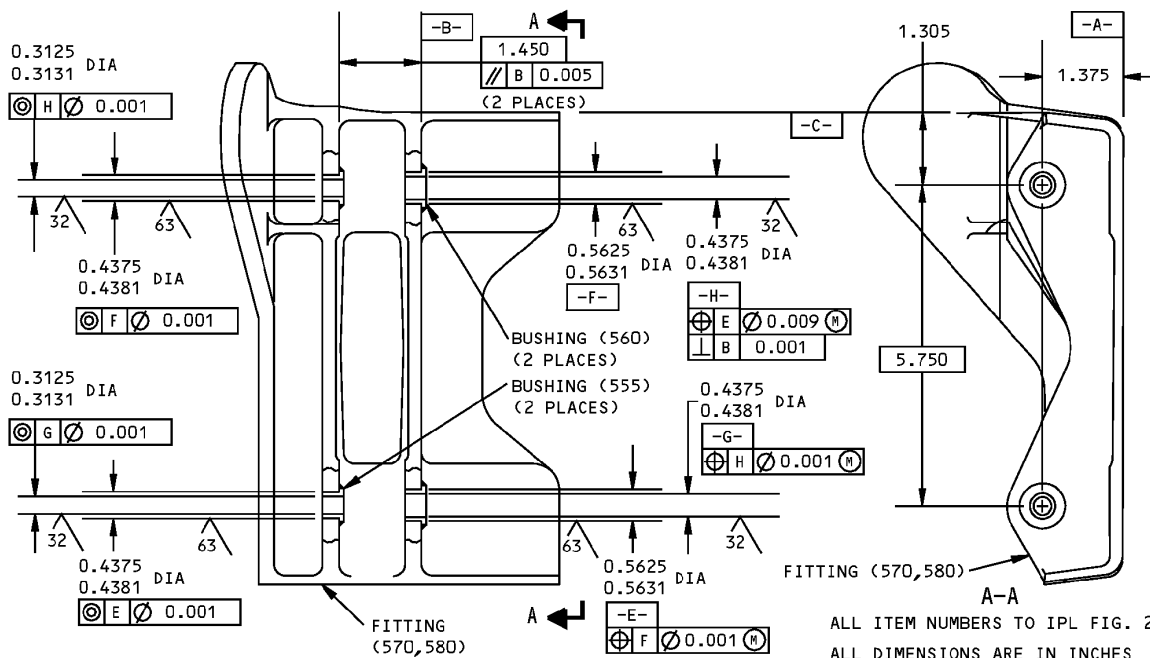
- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to(IPL Figure 2) for the item numbers.

2. Bushing (555, 560) Replacement (IPL Figure 2)

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 1-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Refinish

- A. Fitting (570, 580) – Chromic acid anodize, Type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.
- B. Fitting assembly (540, 550) – Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813).



Bushing Replacement
Figure 601

55-20-16



COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 7-1

65C25773-11, -12, -15, -16

1. General

- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 3) for the item numbers.

2. Bushing Replacement (95, 100, 185)

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 as specified in SOPM 20-50-03 and in REPAIR 7-1, Figure 601.
- C. Machine and chamfer bushings per REPAIR 7-1, Figure 601.
- D. Fillet seal bushing flanges with sealant, A00247 per REPAIR 7-1, Figure 601.

3. Refinish

- A. Fitting (105, 190) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Topcoat fillet seal using coating, C50075 (SRF-14.9813). Material: Aluminum alloy.

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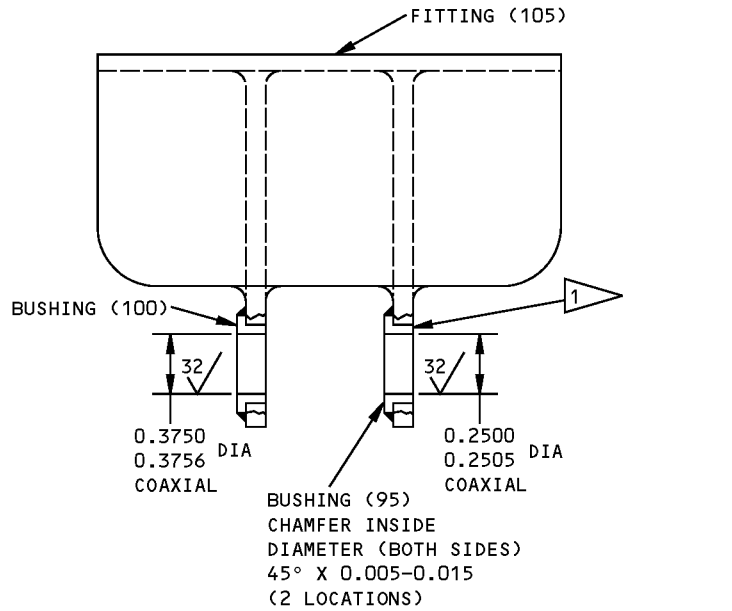
REPAIR 7-1

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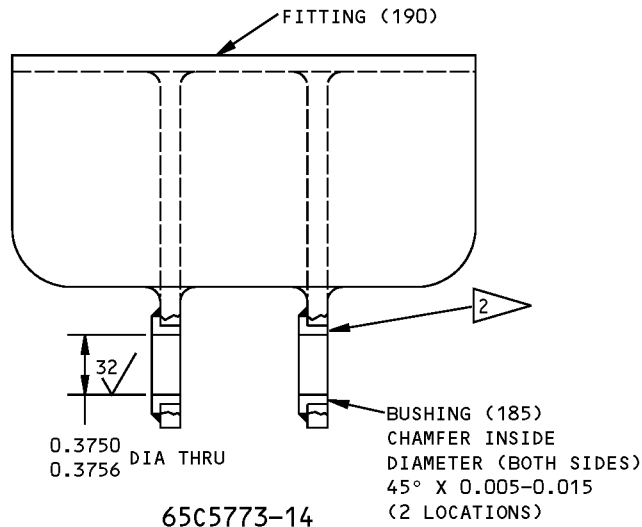
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COMPONENT MAINTENANCE MANUAL



65C5773-13



65C5773-14

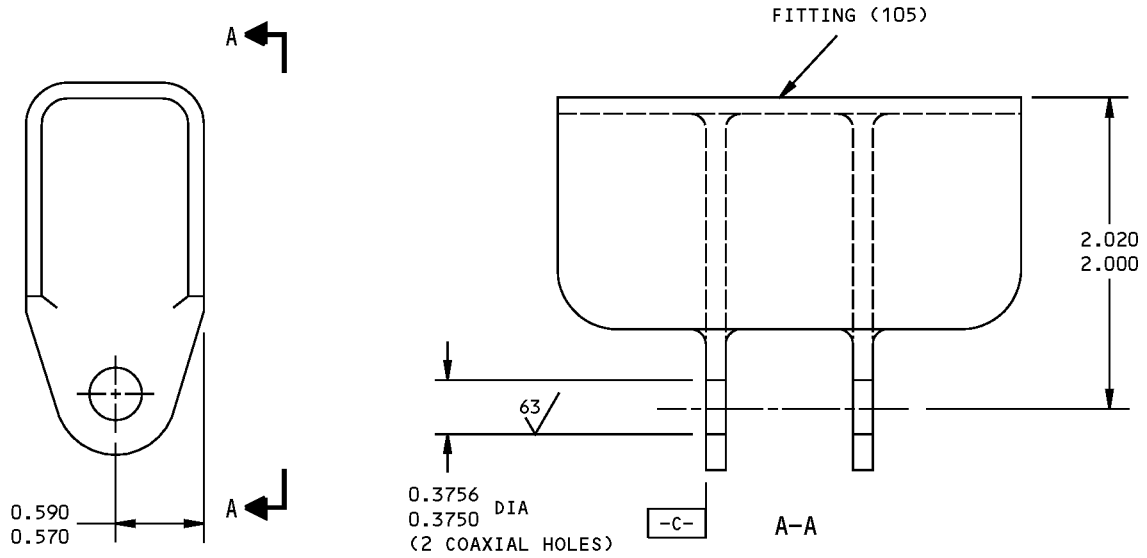
- 1 BUSHING 95C IS INSTALLED FROM THIS SIDE
- 2 BUSHING 185 A IS INSTALLED FROM THIS SIDE

65C25773-11,-12 Bushing Replacement
Figure 601 (Sheet 1 of 2)

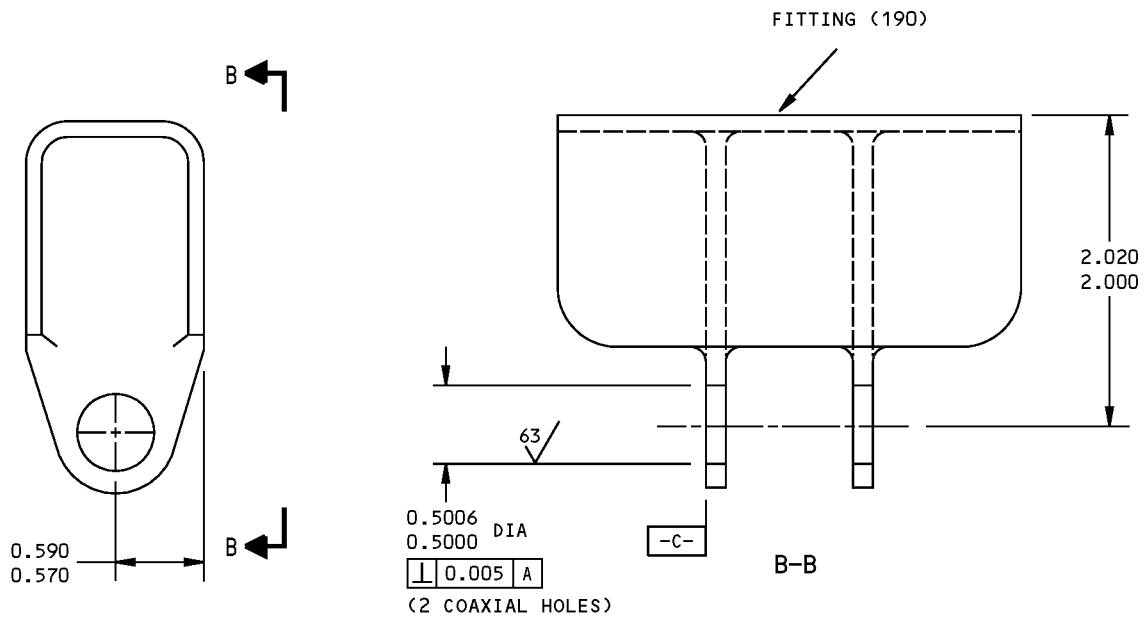
55-20-16

REPAIR 7-1
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65C25773-13



65C25773-14

ITEM NUMBERS REFER TO IPL FIG. 3
ALL DIMENSIONS ARE IN INCHES

65C25773-11,-12 Bushing Replacement
Figure 601 (Sheet 2 of 2)

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REPAIR 7-1
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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 8-1

65C25798-4, -6, -8, -11, -12, -14

1. General

- A. Use this procedure to repair the fitting assembly (165).
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to IPL Figure 4 for the item numbers.

2. Lube Fitting (171) Replacement

- A. Remove the old lube fitting.
- B. Install a replacement lube fitting. Apply grease to the lube fitting until the grease appears at the ID of the bearing. Use a preset grease gun with a nozzle pressure that is not more than 1,000 psi and a flow rate that is not more than 0.50 pint per minute. On fitting assemblies 65C25798-4, 6, -8, -14, use grease, D00633 or grease, D00013. On fitting assemblies 65C25798-11, -12, use grease, D00633 or grease, D00015.

3. Bearing Replacement (REPAIR 8-1, Figure 601)

CAUTION: BEARING REPLACEMENT IS PERMITTED A MAXIMUM OF THREE TIMES. AFTER THE BEARING HAS BEEN REPLACED THREE TIMES, THE HINGE FITTING MUST BE REMOVED AND REPLACED.

- A. Carefully press the worn or damaged bearing out of the hinge fitting as specified in SOPM 20-50-03.

NOTE: If you cannot remove the bearing from the hinge fitting without doing damage to the hinge fitting, then you will have to replace the hinge fitting assembly.

- B. Examine the hinge fitting bore to make sure the diameter is still within drawing tolerances.
- C. Install the replacement bearing (170) with grease, D00633 as specified in SOPM 20-50-03 and REPAIR 8-1, Figure 601.
- D. Roller swage the housing over the bearing as specified in SOPM 20-50-03 and REPAIR 8-1, Figure 601.

4. Refinish

- A. Fitting (175, 176, 177, 177A) – Chromic acid anodize and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813). Do not apply the primer or enamel in the holes for the bearings. Material: Al alloy.
- B. Fitting (177B, 177C) – Boric acid-sulfuric acid anodize or chromic acid anodize (F-17.31). Apply BMS 10-11, Type 1 primer, C00259 (F-20.03). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813). Do not apply the primer or enamel in the holes for the bearings. Material: Al alloy.

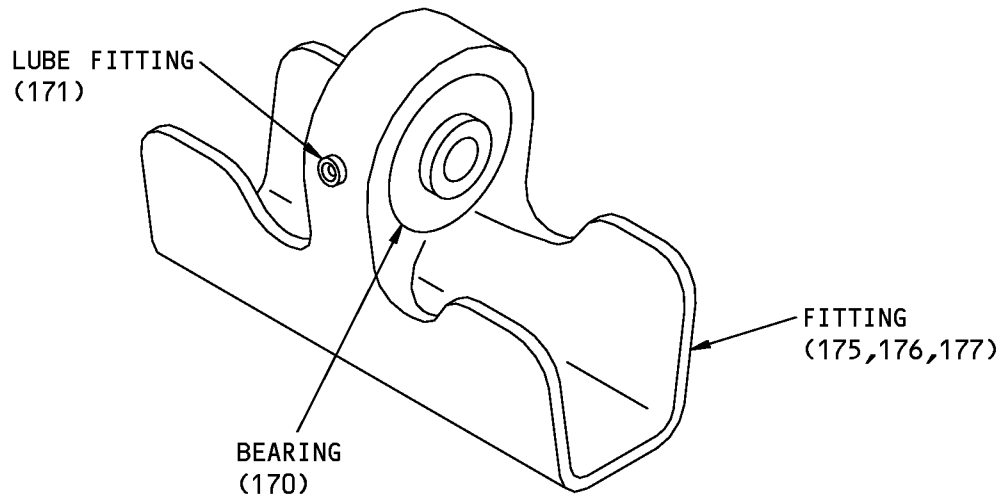
55-20-16

REPAIR 8-1

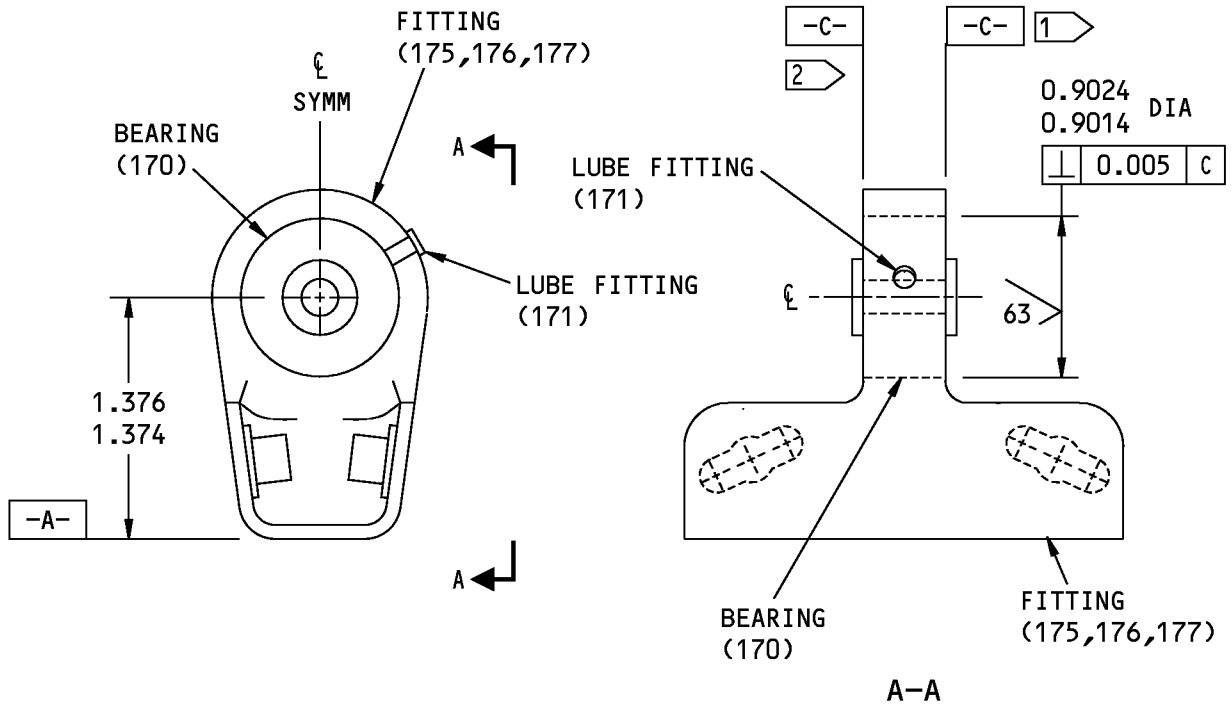
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65C25798-4,-6,-8,-14



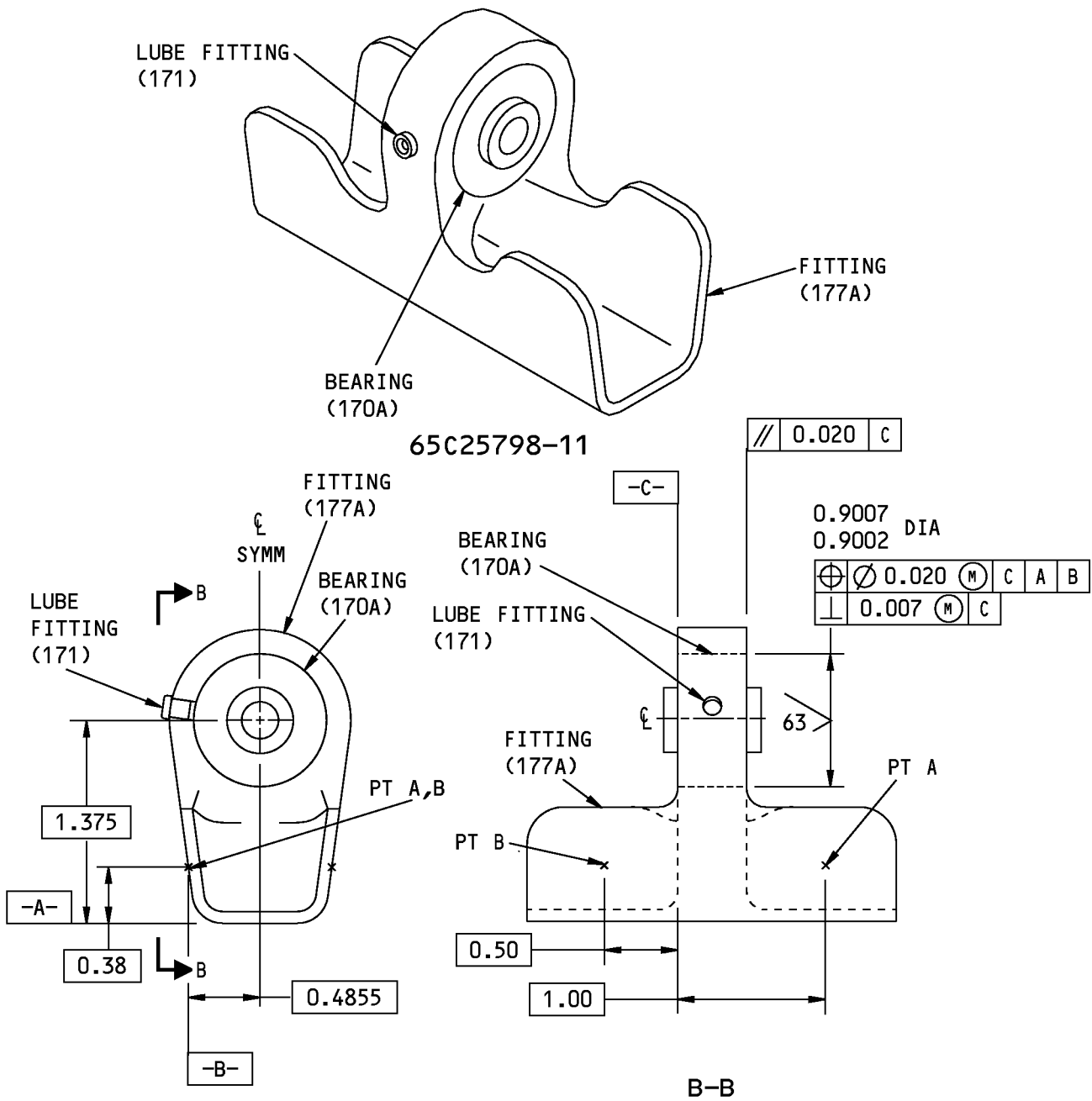
1732863 S0000284455_V1

Bearing Replacement
Figure 601 (Sheet 1 of 3)

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REPAIR 8-1
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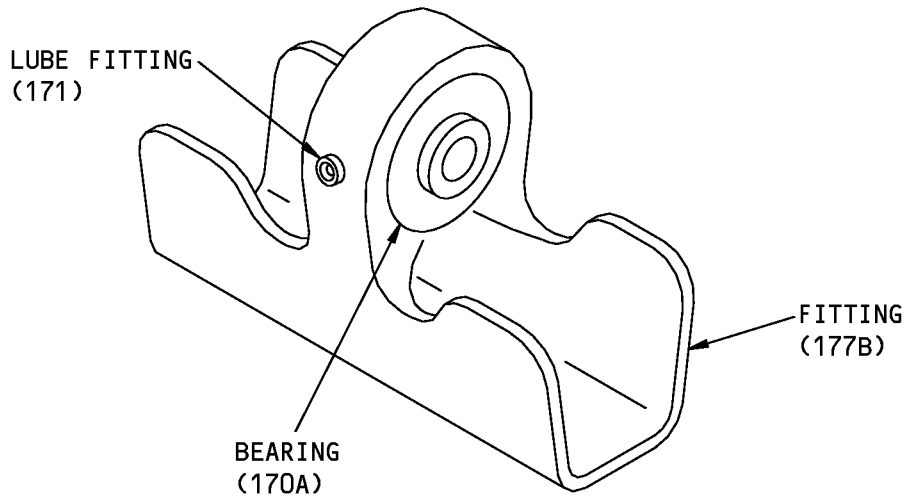
1732884 S0000312848_V1

Bearing Replacement
Figure 601 (Sheet 2 of 3)

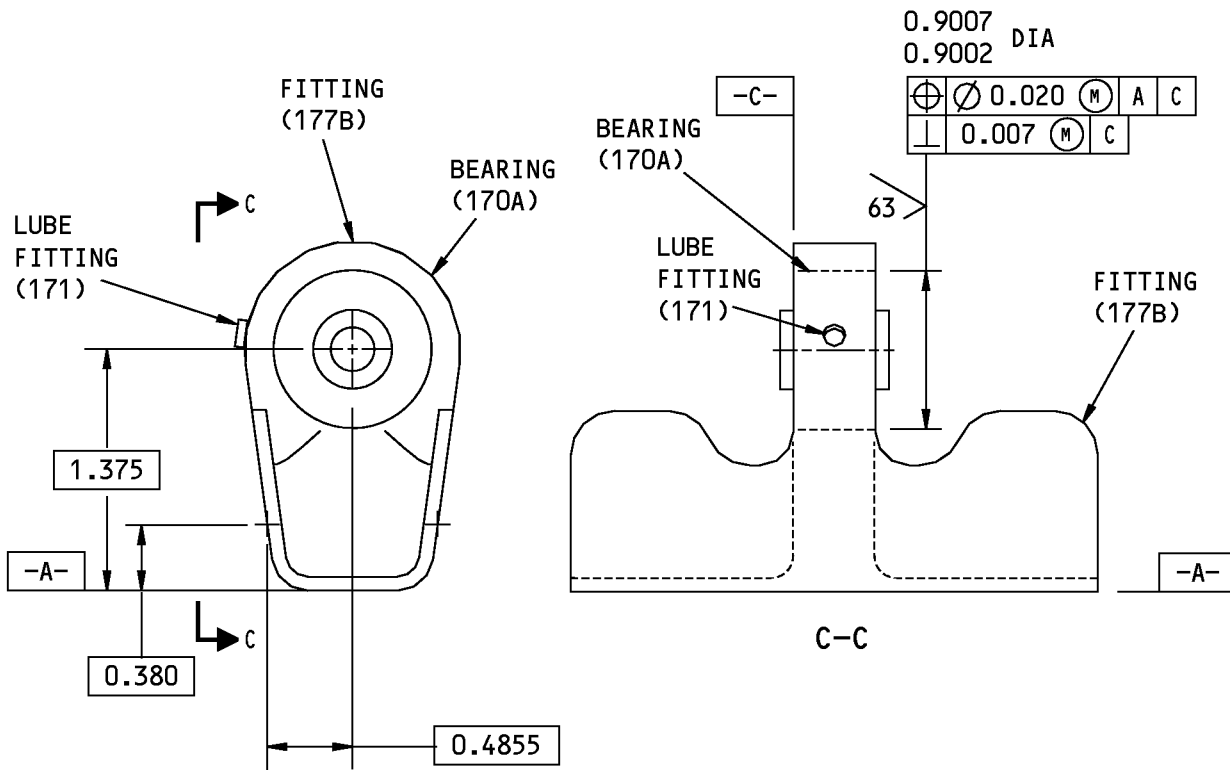
55-20-16

REPAIR 8-1
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65C25798-12



- 1 DATUM -C- FOR 65C25798-4,-6
- 2 DATUM -C- FOR 65C25798-8,-14

ITEM NUMBERS REFER TO IPL FIG. 4
ALL DIMENSIONS ARE IN INCHES

1732898 S0000312929_V1

Bearing Replacement
Figure 601 (Sheet 3 of 3)

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REPAIR 8-1
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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 9-1

65C25799-1, -3, -5, -7, -9

1. General

- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 4) for the item numbers.

2. Pin Replacement (110, 111A)

- A. Remove the pin.

CAUTION: IF THE BEARING IS LOOSE, DAMAGED, CORRODED, OR EXCEEDS THE SERVICE WEAR LIMITS AS SHOWN IN THE FITS AND CLEARANCES, THEN BEARING REPLACEMENT IS NOT ALLOWED.

- B. Install the pin using grease, D00015 to assemble hinge fitting assembly (145, 146) and bearing plate assembly (115, 116).

3. Bearing Replacement

- A. See REPAIR 10-1 for bearing plate assembly, P/N 65C26305-1,-3, and REPAIR 20-1 for bearing plate assembly, P/N 65C32050-1, -3.

55-20-16

REPAIR 9-1

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COMPONENT MAINTENANCE MANUAL

BEARING PLATE ASSEMBLY - REPAIR 10-1

65C26385-1, -3

1. General

- A. This procedure has the data necessary to repair the bearing plate assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 4) for the item numbers.

2. Bearing Replacement (120)

CAUTION: IF THE BEARING IS LOOSE, DAMAGED, CORRODED, OR EXCEEDS THE SERVICE WEAR LIMITS AS SHOWN IN THE FITS AND CLEARANCES, THEN BEARING REPLACEMENT IS NOT ALLOWED.

- A. See SB 737-55-1049 for the replacement of the hinge fitting assembly.

3. Bushing Replacement (125, 126)

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03 and REPAIR 10-1, Figure 601.
- C. Line ream and chamfer bushings inside diameters per REPAIR 10-1, Figure 601.
- D. Fillet seal bushing flanges with sealant, A00247 per REPAIR 10-1, Figure 601.
- E. Topcoat seal using enamel coating, C50075 (SRF-14.9813).

4. Refinish

- A. Plate (130) – Chromic acid anodize, Type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bushing holes. Material: Aluminum alloy.

55-20-16

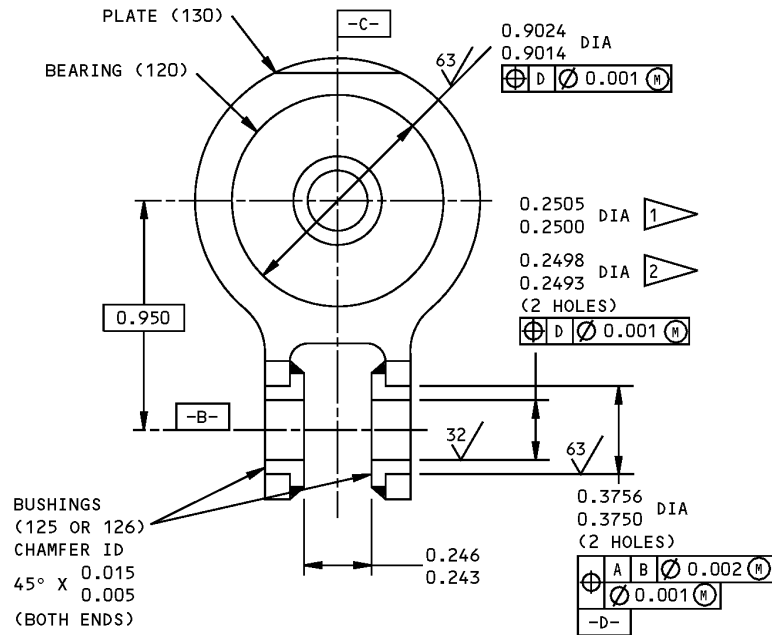
REPAIR 10-1

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- 1 65C26385-1
- 2 65C26385-3

ITEM NUMBERS REFER TO IPL FIG. 4
ALL DIMENSIONS ARE IN INCHES

65C26385-1,-3 Bushing Replacement
Figure 601

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REPAIR 10-1
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FITTING ASSEMBLY - REPAIR 11-1

65C26386-8, -4, -1, 65C32051-1

1. General

- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 4) for the item numbers.

2. Bushing Replacement (140, 141, 142)

- A. Remove the bushing.
- B. Install the new bushing using wet sealant, A00247 as specified in SOPM 20-50-03.

3. Refinish

- A. Fitting (145, 146, 147) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in bushing holes. Material: Aluminum alloy.

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REPAIR 11-1

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PLATE ASSEMBLY - REPAIR 12-1

65C26481-1, 65C26482-1, 69-76409-1, -5, -7, 69-76419-1, -5, -7

1. General

- A. This procedure has the data necessary to repair the plate assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 2) for the item numbers.

2. Bushing Replacement (245, 295, 345, 400, 411F, 412F, 413F, 414F, 416F, 416Q, 417F, 417Q, 418F, 418Q, 419F, 419Q)

- A. Remove bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03 press fit method.
- C. Roller swage bushings over 0.020-0.030 x 45 degree chamfer in bearing plate (255, 305, 355, 410, 411J, 412J, 413J, 414J) holes per REPAIR 12-1, Figure 601.

3. Bearing Replacement (250, 300, 350, 405, 411G, 412G, 413G, 414G, 416G, 416R,

CAUTION: THE BEARING REPLACEMENT IS A ONE TIME REPLACEMENT. RETENTION OF THE BEARING MAY NOT OCCUR, IF A BEARING IS REPLACED MORE THAN ONE TIME IN THE BEARING PLATE.

- A. Carefully press the worn or damaged bearing out of the bearing plate (255, 305, 355, 410) as specified by SOPM 20-50-03.

NOTE: If you cannot remove the bearing from the bearing plate without doing damage to the bearing plate, then you will have to replace the bearing plate assembly .

- B. Install bearing using grease, D00015 per SOPM 20-50-03 and REPAIR 12-1, Figure 601.
- C. Roller swage bearing plate (255, 305, 355, 410) over bearing per REPAIR 12-1, Figure 601.

4. Refinish

- A. Plate (255, 305, 355, 410, 411J, 411K, 412J, 412K, 413J, 413K, 414J, 414K, 416J, 416T, 417J, 417T, 418J, 418T, 419J, 419T) – Chromic acid anodize, Type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit the primer and enamel in the bearing and bushing holes.
Material: Aluminum alloy.

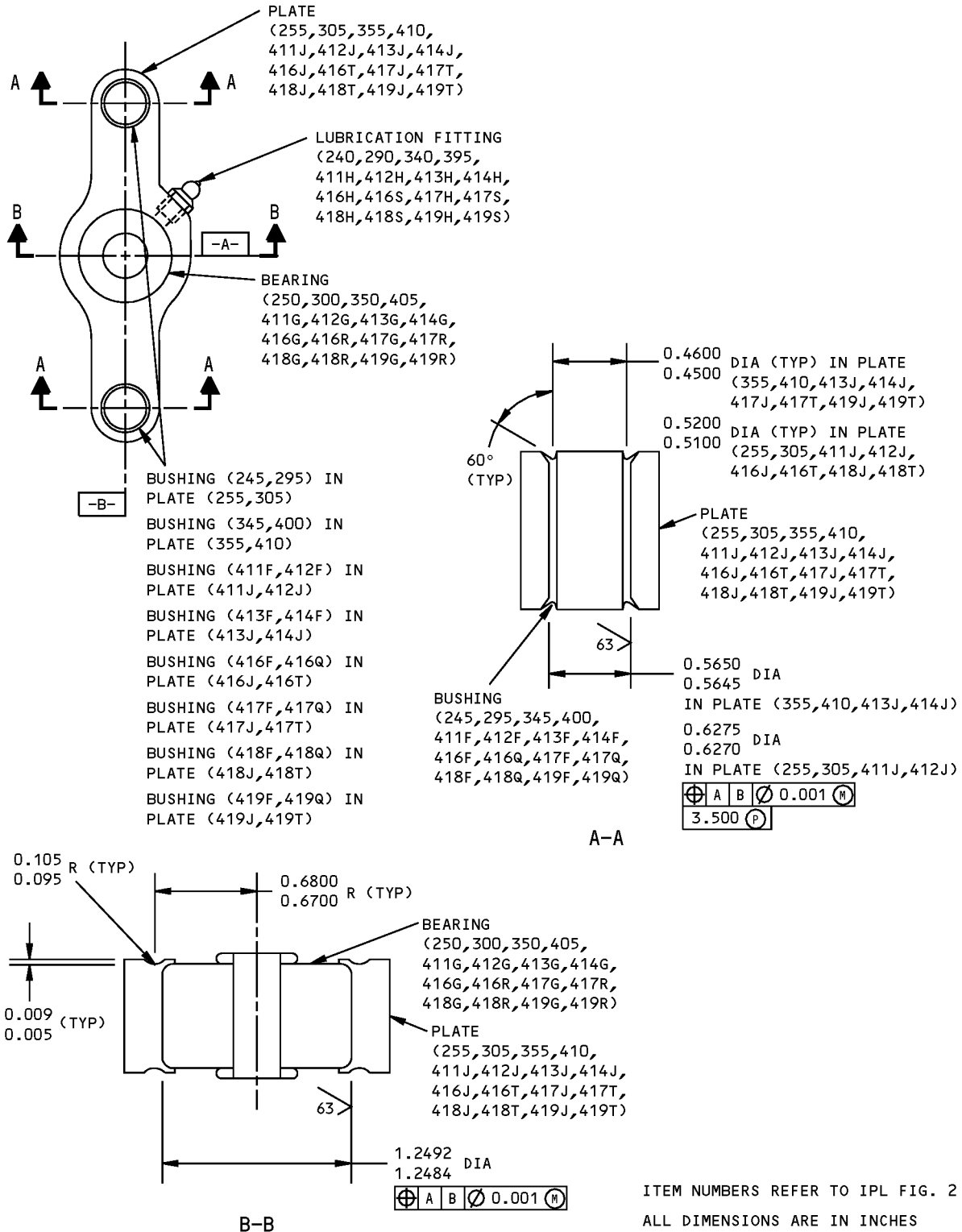
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REPAIR 12-1

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Bearing and Bushing Replacement Figure 601

ITEM NUMBERS REFER TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES



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PLATE ASSEMBLY - REPAIR 13-1

65C26529-1, -4, 65C27301-1, -4, -5

1. General

- A. This procedure has the data necessary to repair the plate assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 2) for the item numbers.

2. Bushing (110, 115, 116, 117, 190, 195, 196, 197) Replacement

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 13-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Bearing Replacement (105, 185)

- A. Remove the bearings.
- B. Install bearing using grease, D00015 per SOPM 20-50-03.

4. Refinish

- A. Bearing plate assembly (40, 40A, 40B, 40C, 130, 130A, 130B) – Cadmium plate 0.0002-0.0004-inch thick per QQ-P-416, type 2, class 3 (F-15.02) per 20-42-05. Apply primer, C00259 (F-20-02) per 20-41-02. Apply enamel coating, C50075 (SRF.14.9813) all over and topcoat fillet seals. Omit primer and enamel in the bushing and bearing holes, keeper and relief bores, and faying surfaces. Material: 15-5 PH CRES, 180-200 KSI

5. Repair Assembly (REPAIR 13-1, Figure 601)

- A. Assemble cap (120, 200) and pillow block (125, 205) or (127, 206), using bolts (80, 82, 165) with countersunk washers (85, 170) under boltheads and plain washers (90, 175) under nuts (100, 102, 180).
- B. Tighten the nuts (100, 102, 180) within 220-360 pound-inches torque range.
- C. Install lubrication fittings (70, 160).
- D. Rubber stamp on each part, paper tag, and lead seal each set, "DO NOT ISSUE SEPARATELY". Cap (120, 200) and pillow block (125, 205) or cap (120, 200) and pillow block (127, 206) comprise a matched set.

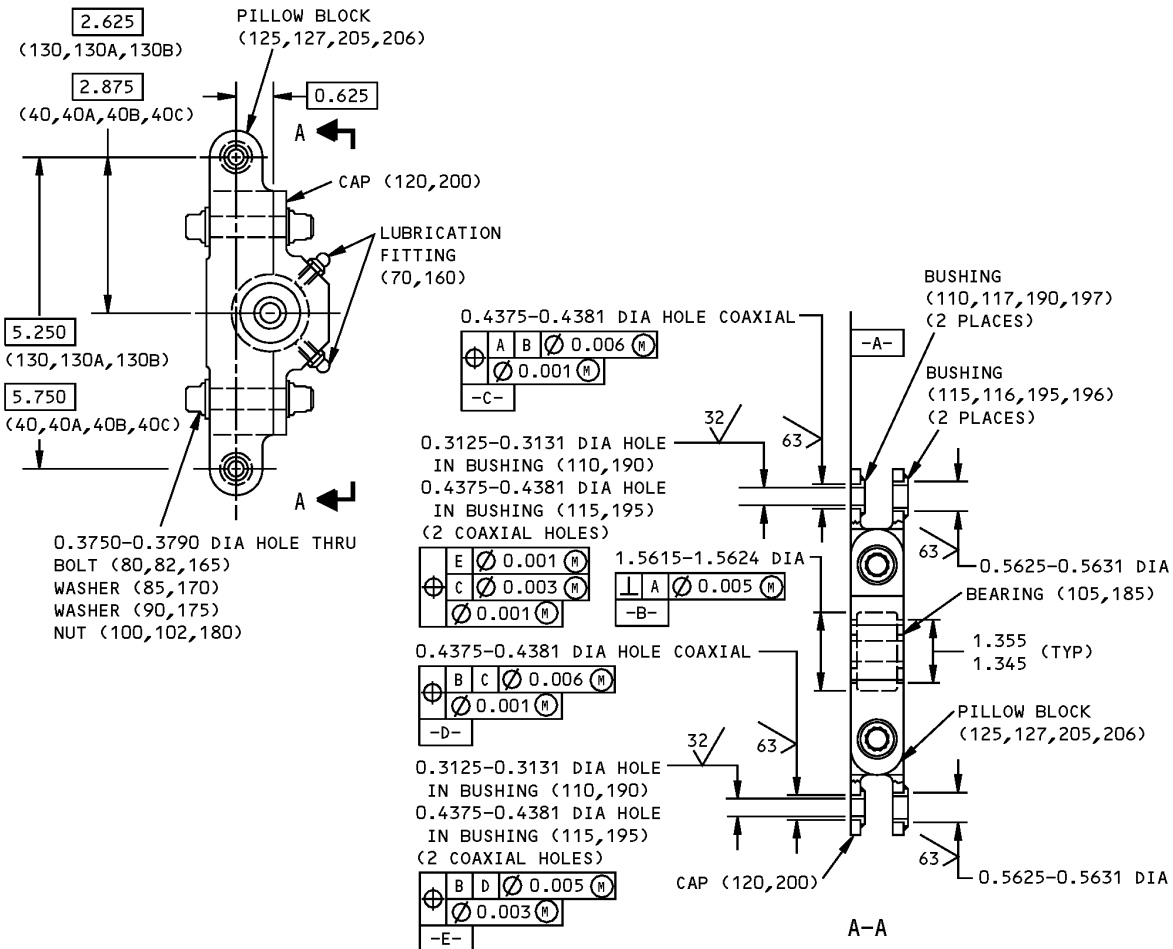
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REPAIR 13-1

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ALL ITEM NUMBERS REFER TO IPL FIG. 2
 ALL DIMENSIONS ARE IN INCHES

Bearing Plate Repair
 Figure 601



COMPONENT MAINTENANCE MANUAL

BRACKET ASSEMBLY - REPAIR 14-1

65C27351-1, -2, -5, -6, -9

1. General

- A. This procedure has the data necessary to repair the bracket assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 2) for the item numbers.

2. Bushing Replacement (REPAIR 14-1, Figure 601)

- A. Remove the old bushings (613, 614, 615, 620, 675, 680).
- B. Install replacement bushings by the shrink-fit method (SOPM 20-50-03).
- C. Fillet seal the bushing flanges with sealant, A00247.

3. Refinish

- A. Angle (630A, 640, 690A, 700) – Chromic acid anodize and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813). No primer, C00259 or enamel in the holes for the bushings. Material: Al alloy.
- B. Angle (630, 635, 690, 695) – Chromic acid anodize and apply primer, C00259 (F-18.13). Material: Al alloy.
- C. Bracket assembly (585, 585A, 645, 645A) – On spotface surfaces, chemical treat (F-17.10) and apply primer, C00259 (F-20.02). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813). No primer or enamel in bushing bores or on flange faces. Material: Al alloy.

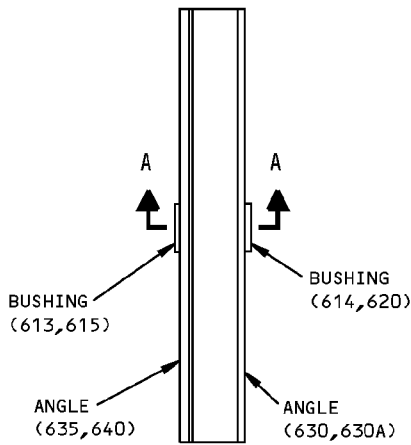
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REPAIR 14-1

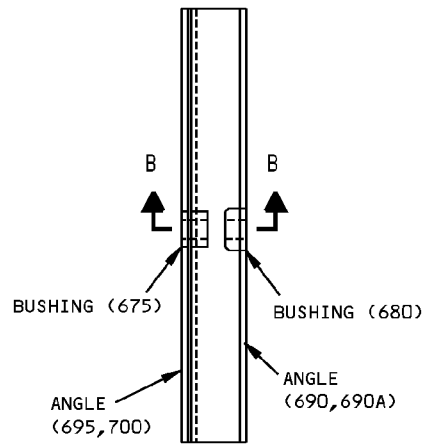
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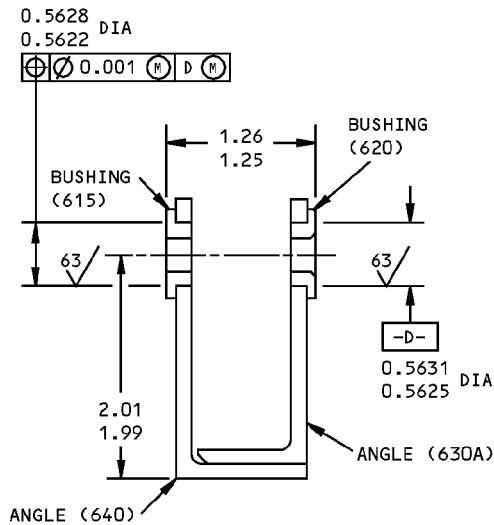
COMPONENT MAINTENANCE MANUAL



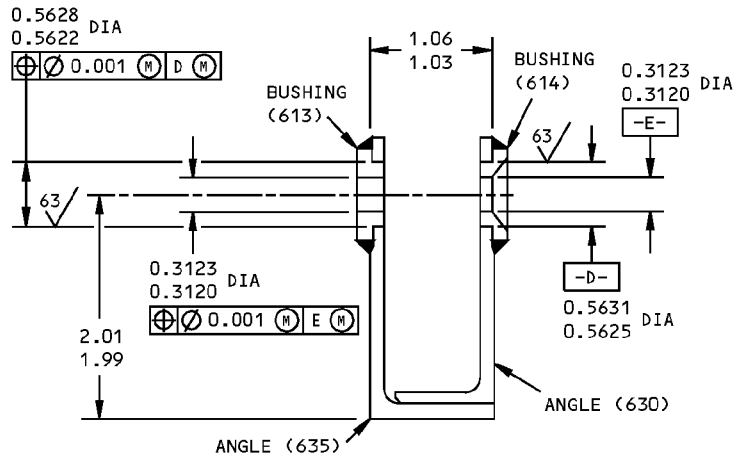
65C27351-1,-5



65C27351-2,-6,-9



65C27351-1
A-A



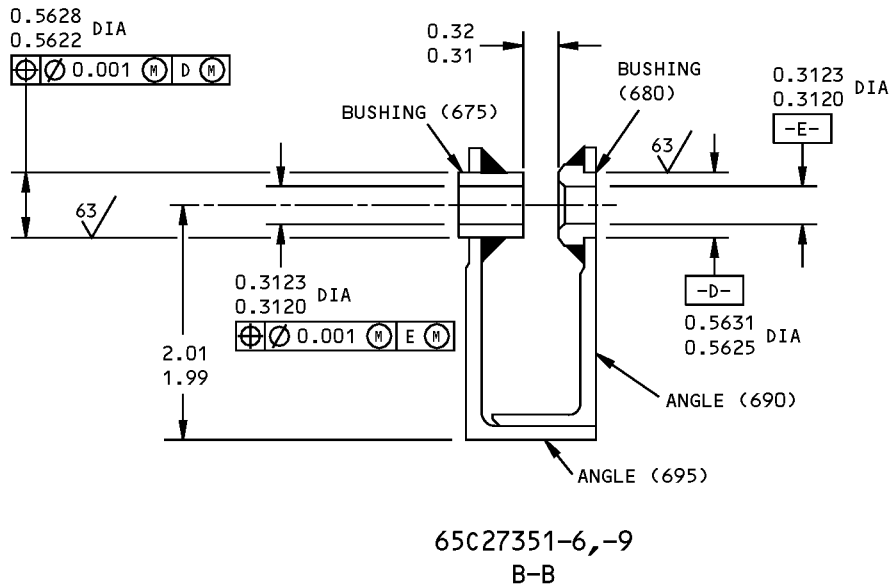
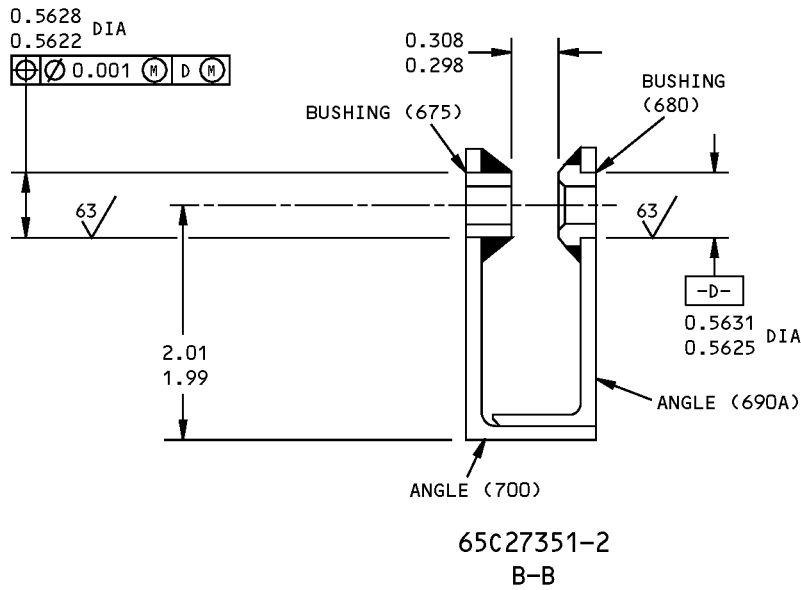
65C27351-5
A-A

65C27351-1,-2,-5,-6,-9 Bushing Replacement
Figure 601 (Sheet 1 of 2)

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REPAIR 14-1
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ITEM NUMBERS REFER TO IPL FIG. 2

ALL DIMENSIONS ARE IN INCHES

65C27351-1,-2,-5,-6,-9 Bushing Replacement
Figure 601 (Sheet 2 of 2)

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REPAIR 14-1

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COMPONENT MAINTENANCE MANUAL

MAST ARM FITTING ASSEMBLY - REPAIR 15-1

65C27376-1, -12

1. General

- A. This procedure has the data necessary to repair the mast arm fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 2) for the item numbers.

2. Bearing and Sleeve Replacement (525, 530)

- A. Remove bearing and sleeve.
- B. Install sleeve dry.
- C. Install bearing using grease, D00015.
- D. Roller stake sleeve over both housing and bearing both sides per SOPM 20-50-03.

3. Refinish

- A. Mast Arm fitting assembly (415, 420) – Apply enamel coating, C50075 (SRF-14.9813). On bearing hole apply Alodine 1200 coating, C50033 (F-17.10) and primer, C00259 per (F-20.02). Omit primer, C00259 and enamel coating, C50075 in bearing hole.

4. Assembly

NOTE: See REPAIR 6-1 for bushing replacement

- A. Bond two mating surfaces of mast arm fitting assembly (335, 540, 545, 550) failsafe strap (515, 520) using SOPM 20-50-12, Type 70.
- B. Install bolts (505), nuts (506) and washers (507) using wet sealant, A00247.

CAUTION: DAMAGE TO BEARING SEALS MAY RESULT FROM EXCESSIVE GREASE GUN NOZZLE PRESSURE AND FLOW RATE.

- C. Apply grease, D00015 to lube fitting (500) until grease appears at ID of bearing (525). Use a pre-set grease gun with nozzle pressure not to exceed 1000 psi and flow rate not to exceed 0.50 pint per minute.

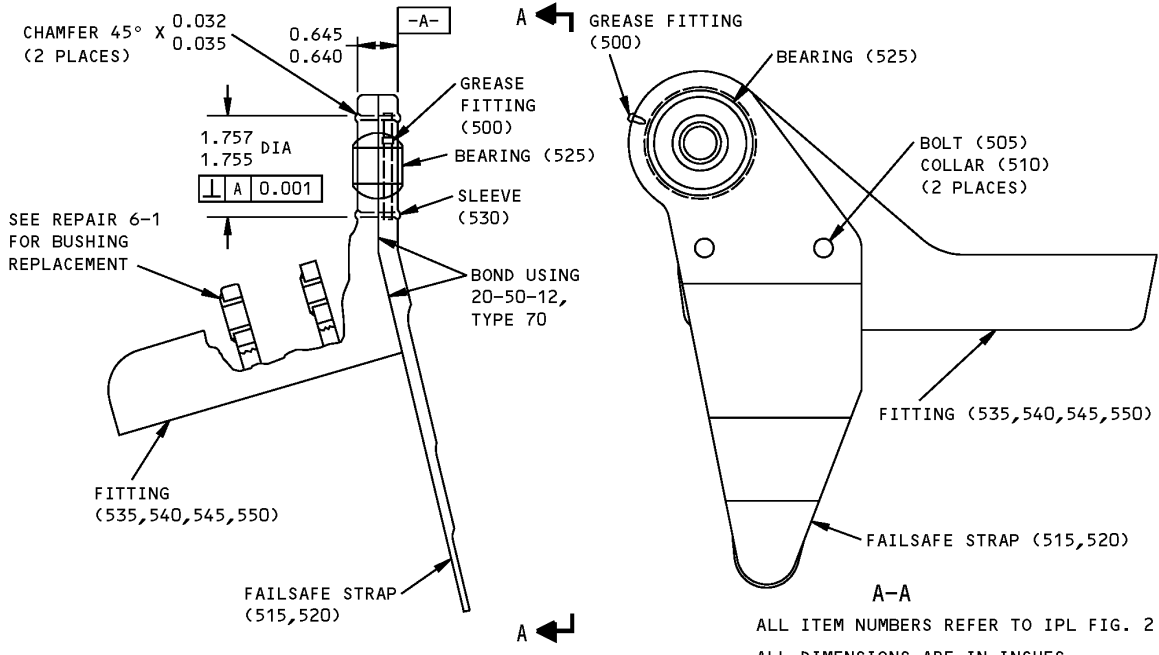
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REPAIR 15-1

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Bearing Replacement
Figure 601



COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 16-1

65C28142-1, -2

1. General

- A. This procedure has the data necessary to repair the fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to (IPL Figure 4) for the item numbers.

2. Bushing (250) Replacement

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03 the shrink fit method.
- C. Machine bushings per REPAIR 16-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.002-0.005-inch or radius.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Bushing (245) Replacement

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03 shrink fit method and REPAIR 16-1, Figure 601.
- C. Machine bushings per REPAIR 16-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.002-0.005-inch or radius.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

4. Refinish

- A. Fitting (255, 260) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit primer and enamel in bushing holes and under flanges.
Material: Aluminum alloy.

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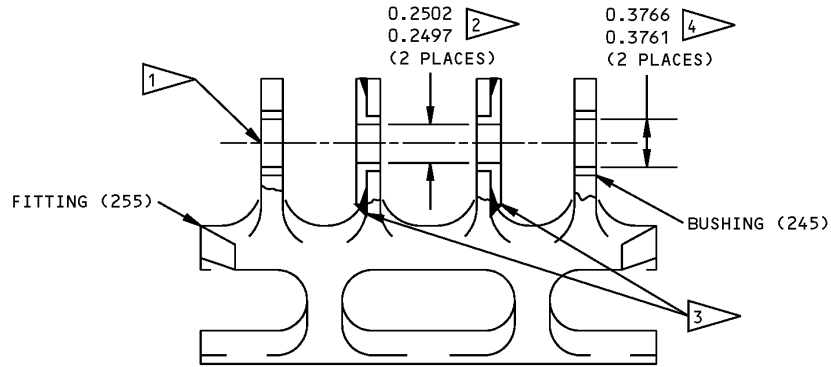
REPAIR 16-1

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- 1 HOLES TO BE CONCENTRIC WITHIN 0.002 T.I.R.
- 2 0.002-0.005 CHAMFER OR RADIUS
- 3 INSTALL BUSHING WITHIN PLUS 0.002 OR MINUS 0.000 AGAINST FITTING
- 4 INSTALL BUSHING WITHIN PLUS 0.000 OR MINUS 0.005 TO INSIDE FACE OF CLEVIS

ALL ITEM NUMBERS REFER TO IPL FIG. 4
 ALL DIMENSIONS ARE IN INCHES

Bushing Replacement
 Figure 601

55-20-16

REPAIR 16-1
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MISCELLANEOUS PARTS REFINISH - REPAIR 17-1

1. General

- A. This procedure has the data necessary to refinish the miscellaneous parts.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.

2. Procedure

- A. Refer to REPAIR 17-1, Table 601 for the refinish of miscellaneous parts.

Table 601: Refinish Details

IPL FIG. & ITEM	MATERIAL	FINISH
IPL Fig. 1		
Weight (75)	Tungsten alloy	Apply primer, C00259 (F-20.03).
Weight (80)	4130 Steel	Cadmium plate and apply primer, C00259 (F-16.01) Apply enamel coating, C00033 (F-14.9815-201, which replaces SRF-14.9815-201).
Conductive frame (295,300,305,310)	Al alloy	Chemical treat (F-17.07). Apply primer, C00175 all over (F-19.47) but not on area under static discharger base.
Tip fitting assembly (315,320)	Al alloy	Chromic acid anodize and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813).
Tip fitting assembly (325,330)	Glass-filled nylon molding	Apply primer, C00175 (F-19.52)
Elevator assembly (1-series,5-series)		Prepare surface and apply enamel coating, C50075 and enamel coating, C00033 per operator's standard color scheme (SRF-14.9863-xxx) per Fig. 602.
IPL Fig. 2		
Failsafe strap (515,520)	Al alloy	Chromic acid anodize and apply primer, C00259 (F-18.13).
Bonded assembly, front spar (30,30A, 30B)	Bonded glass fabric/graphite epoxy	On exterior and trimmed edges prepare surface and apply Dexter 28-C-1 Static Conditioner (F-14.67) and apply primer, C00319 and enamel coating, C50075 (F-19.40). Overspray permitted.
Cover (10)	Graphite/epoxy	Prepare exterior surface (SRF-14.672). Apply primer, C00319 (F-19.46). Apply enamel coating, C50075 (F-14.9813, which replaces SRF-14.9813), including holes.
IPL Fig. 3		
Bond assembly, rear spar (10,15)	Bonded glass fabric/graphite epoxy	On exterior and trimmed edges, prepare surface (F-14.67). Apply primer, C00319 and enamel coating, C50075 (F-19.40).

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REPAIR 17-1

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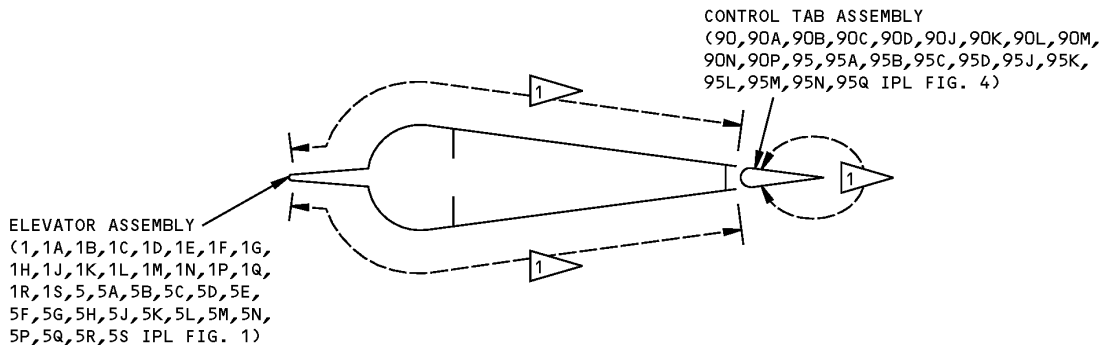
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Table 601: Refinish Details (Continued)

IPL FIG. & ITEM	MATERIAL	FINISH
IPL Fig. 4		
Mast fitting fairing (210,215)	Bonded aramid/glass fabric	Prepare exterior surface (SRF-14.672). Apply coating, C00841 (F-14.685, which replaces SRF-14.68). Apply primer, C00319 (F-19.46).
Control tab assembly (90,90A, 90B,90C,90D,90J, 90K,90L,90M,95,95A, 95B,95C,95D,95J, 95K,95L,95M,95N)	Nomex H/C core/ phenolic sheet	Same as elevator assembly above, but overspray is permitted on interior surface and finish the edges (SRF-14.9863-xxx) per REPAIR 17-1, Figure 601.
IPL Fig. 6		
Seal retainer (100,105,110)	Al alloy	Chemical treat and apply primer, C00259 (F-18.06) all over.
IPL Fig. 7		
Hinge half (30,35, IPL Fig. 5; 45,50, 55, IPL Fig. 6; 30, 35, IPL Fig. 7)	Al-Ni-Bronze	Cadmium plate and apply primer, C00259 (F-16.01), but no primer in 0.183-inch pin holes. Primer permitted on remaining surfaces.
Fitting fork (65, IPL Fig. 5; 125, IPL Fig. 6; 65, IPL Fig. 7)		Chemical treat or chromic acid anodize and apply primer, C00259 (SRF-2.30). On three inside surfaces and two edges, apply primer, C00259 (SRF-12.206). Overspray permitted.



1 SRF 14.9863-XXX ENAMEL

TYPICAL ELEVATOR AND TAB CROSS SECTION

Elevator and Tab Refinish
Figure 601

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REPAIR 17-1
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BOLT - REPAIR 18-1

69-44682-2

1. General

- A. This procedure has the data necessary to repair the bolt.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.

2. Shank Repair (REPAIR 18-1, Figure 601)

NOTE: For repair which is only replacement of the original finish, refer to Refinish instructions.

- A. Machine as required, within repair limits, to remove defects.
- B. Magnetic particle examine (SOPM 20-20-01, Class B).
- C. Build up with chrome plate (SOPM 20-42-03). Grind (SOPM 20-10-04) to design dimensions and finish.

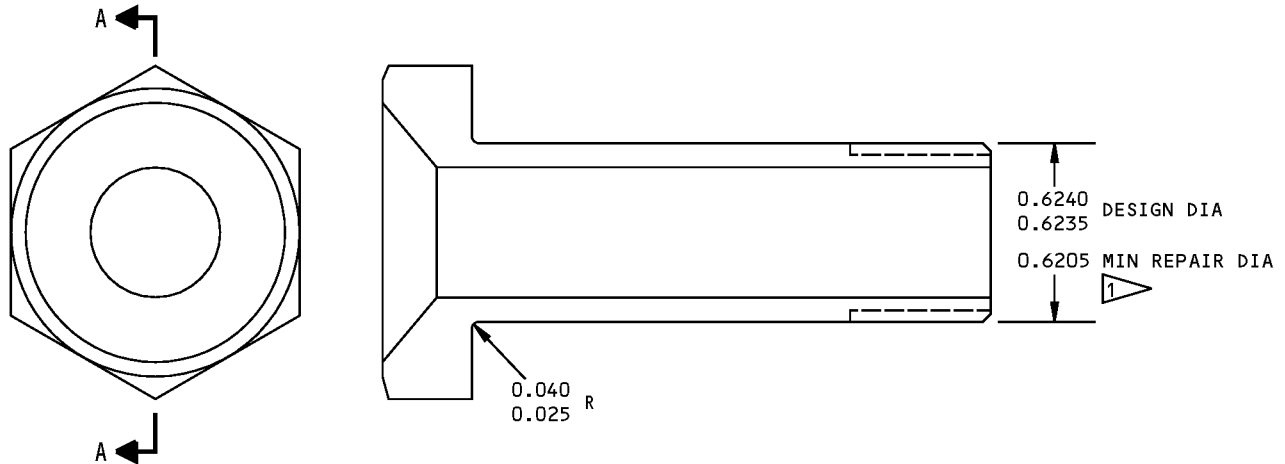
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REPAIR 18-1

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FINISH

CADMIUM-TITANIUM PLATE (F-15.01), 0.0002-0.0003 THICK. NO PLATING IN BORE.

LIMIT FOR CHROME PLATE BUILDUP (SOPM 20-42-03). GRIND TO DESIGN DIMENSIONS AND FINISH

REPAIR

REF 125 ALL MACHINED SURFACES UNLESS SHOWN DIFFERENTLY

MATERIAL: 4340 STEEL, 160-180 KSI
 DIMENSIONS APPLY AFTER PLATING UNLESS SHOWN DIFFERENTLY
 ALL DIMENSIONS ARE IN INCHES

69-44682-2 Bolt Repair and Refinish
 Figure 601

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REPAIR 18-1
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COMPONENT MAINTENANCE MANUAL

BOLT - REPAIR 19-1

69-63391-5, -6, -7, -8

1. General

- A. This procedure has the data necessary to repair the bolt.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.

2. Shank Repair

NOTE: For repair which is only replacement of the original finish, refer to Refinish instructions in REPAIR 19-1, Figure 601.

- A. Machine as required, within repair limits, to remove defects.
- B. Magnetic particle examine (SOPM 20-20-01, Class B).
- C. Shot peen (SOPM 20-10-03) 0.017-0.039 shot size, 0.016A2 intensity.
- D. Build up with chrome plate (SOPM 20-42-03) and grind (SOPM 20-10-04) to design dimensions and finish.

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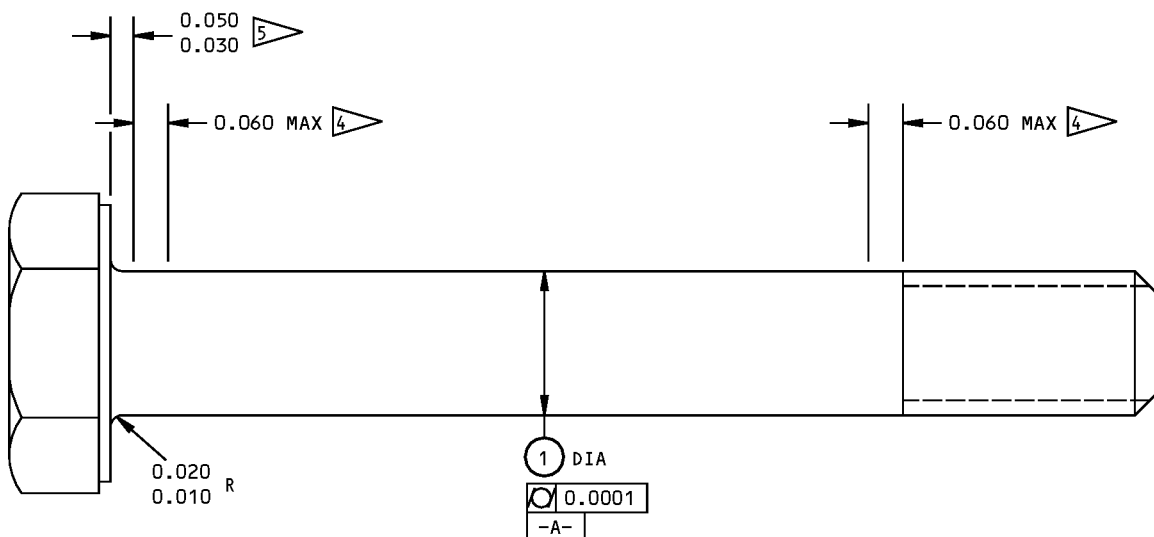
REPAIR 19-1

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	① ①	① ②
DESIGN DIM	0.2497 0.2494	0.2494 0.2492
REPAIR LIMIT	0.2445 ③	0.2445 ③

REFINISH

CHROME PLATE (F-1.90, WHICH REPLACES F-1.842) DIA -A-, WITH 0.060 PLATING RUNOUT AT EDGES. PASSIVATE (F-17.25, WHICH REPLACES F-17.09) OTHER SURFACES.

- ① 69-63391-5,-6
- ② 69-63391-7,-8
- ③ LIMIT FOR CHROME PLATE BUILDUP (SOPM 20-42-03) AND GRIND TO DESIGN DIMENSIONS AND FINISH. PUT A PLATING RUNOUT AS SHOWN
- ④ CHROME PLATE RUNOUT
- ⑤ NO CHROME PLATE

REPAIR

REF ③
 125/ ALL MACHINED SURFACES UNLESS SHOWN DIFFERENTLY
 MATERIAL: A286 CORROSION RESISTANT STEEL, 95 KSI
 DIMENSIONS APPLY AFTER PLATING
 ALL DIMENSIONS ARE IN INCHES

69-63391-5 THRU -8 Bolt Repair and Refinish
 Figure 601

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REPAIR 19-1
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BEARING PLATE ASSEMBLY - REPAIR 20-1

65C32050-1, -3

1. General

- A. This procedure has the data necessary to bearing plate assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to IPL Figure 4 for the item numbers.

2. Bearing Replacement (120)

CAUTION: IF THE BEARING IS LOOSE, DAMAGED, CORRODED, OR EXCEEDS THE SERVICE WEAR LIMITS AS SHOWN IN THE FITS AND CLEARANCES, THEN BEARING REPLACEMENT IS NOT ALLOWED.

- A. See SB 737-55-1049 for the replacement of the hinge fitting assembly.

3. Bushing Replacement (127)

- A. Remove the bushing(s).
- B. Check hole for damage or corrosion and repair if necessary per REPAIR 20-1, Paragraph 4.A..
- C. Install bushings using wet sealant, A00247 per SOPM 20-50-03 and REPAIR 20-1, Figure 601.
- D. Machine inside diameter of bushings and chamfer inside diameter (both sides) per REPAIR 20-1, Figure 601.
- E. Fillet seal bushing flanges with sealant, A00247 per REPAIR 20-1, Figure 601.

4. Repair

- A. Repair hole for bushing (127)
 - (1) Machine bearing plate (131) as required to remove discrepancy, per REPAIR 20-1, Figure 601.
 - (2) Penetrant check per SOPM 20-20-02.
 - (3) Refinish per REPAIR 20-1, Paragraph 5.A..
 - (4) Fabricate oversize bushing(s) per REPAIR 20-1, Figure 602.

5. Refinish

- A. Plate (131) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply enamel coating, C50075 (SRF-14.9813). Omit primer and enamel in bushing holes. Topcoat fillet seal using enamel coating, C50075 (SRF-14.9813). Material: Aluminum alloy.

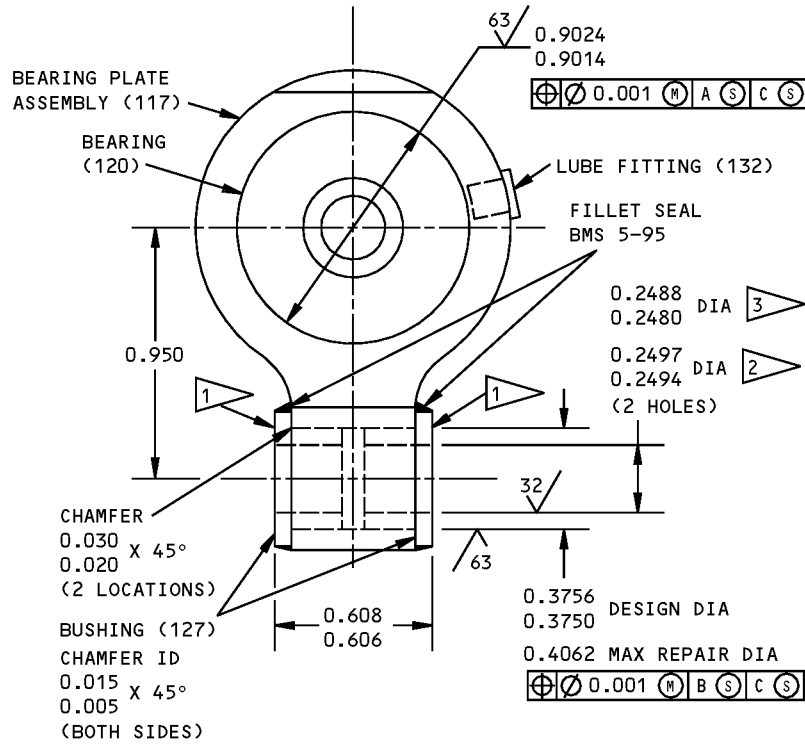
55-20-16

REPAIR 20-1

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- 1 OMIT PRIMER AND ENAMEL THIS SURFACE
- 2 65C32050-1
- 3 65C32050-3

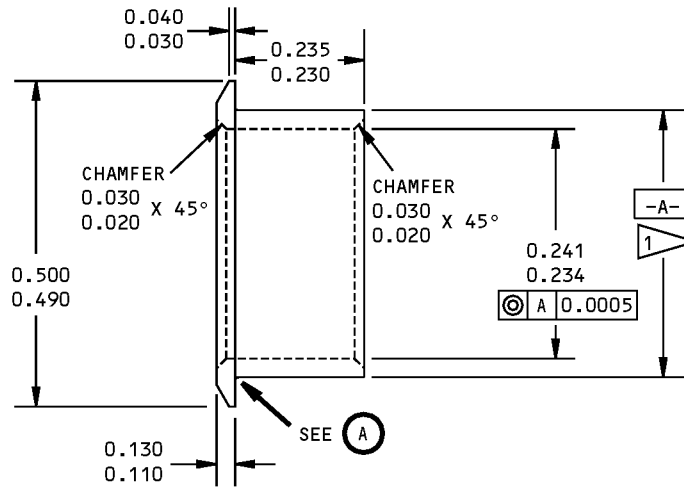
ALL ITEM NUMBERS REFER TO IPL FIG. 4
 ALL DIMENSIONS ARE IN INCHES

Bushing Replacement
 Figure 601

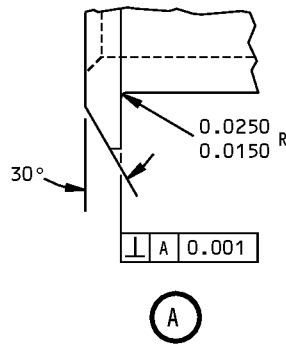
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REPAIR 20-1
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OVERSIZE PLACEMENT BUSHING FOR BUSHING (127)



63/ ALL MACHINED SURFACES, EXCEPT AS NOTED

MATERIAL: AL-NI-BR PER AMS4640 OR 4880

FINISH: CADMIUM PLATE TO MEET THE REQUIREMENTS OF QQ-P-416, TYPE 2, CLASS 2 (PLATING IN BORE OPTIONAL)

ITEM NUMBERS REFER TO IPL FIG. 1

ALL DIMENSIONS ARE IN INCHES

DIMENSIONS AFTER PLATING

1 FINAL BUSHING OUTSIDE DIA EQUALS REPAIR DIA OF FITTING PLUS 0.0003-0.0015 INCH INTERFERENCE

Oversize Bushing Details
Figure 602

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REPAIR 20-1

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COMPONENT MAINTENANCE MANUAL

HINGE FITTING ASSEMBLY - REPAIR 21-1

65C32051-1, -4, -6

1. General

- A. This procedure has the data necessary to repair the hinge fitting assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to IPL Figure 4 for the item numbers.

2. Bushing Replacement (142)

- A. Remove the bushing(s).
- B. Check hole for damage or corrosion and repair if necessary per REPAIR 21-1, Paragraph 3.A..
- C. Install bushing using wet sealant, A00247 per SOPM 20-50-03 and REPAIR 21-1, Figure 601.
- D. Machine inside diameter of bushings and chamfer inside diameter on flanged ends of both bushings per REPAIR 21-1, Figure 601.
- E. Fillet seal bushing flanges with sealant, A00247 per REPAIR 21-1, Figure 601.

3. Repair

- A. Repair hole for bushing (142).
 - (1) Machine hinge fitting (147) as required to remove discrepancy, per REPAIR 21-1, Figure 601.
 - (2) Penetrant check per SOPM 20-20-02.
 - (3) Refinish per REPAIR 21-1, Paragraph 4.A..
 - (4) Fabricate oversize bushing per REPAIR 21-1, Figure 602.

4. Refinish

- A. Fitting (147) – Chromic acid anodize, type 1 and apply primer, C00259 (F-18.13). Apply coating, C50075 (SRF-14.9813). Omit primer and enamel in bushing holes. Topcoat fillet seal using coating, C50075 (SRF-14.9813). Material: Aluminum alloy.

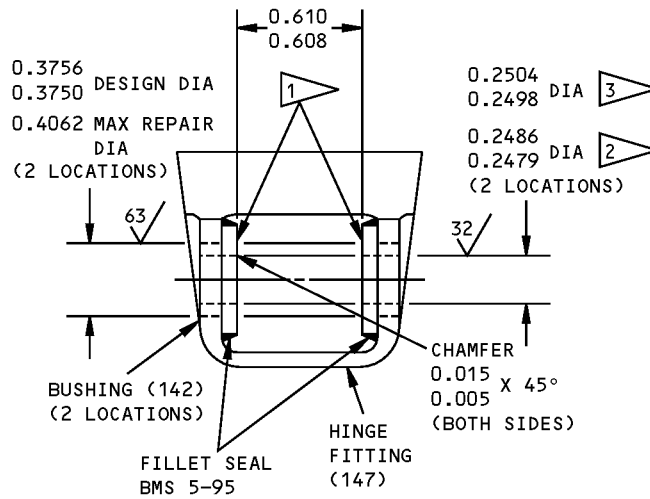
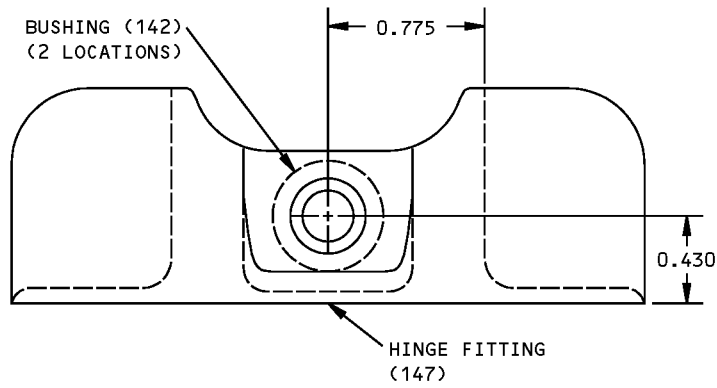
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REPAIR 21-1

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- 1 OMIT PRIMER AND ENAMEL THIS SURFACE
- 2 65C32051-1,-4
- 3 65C32051-6

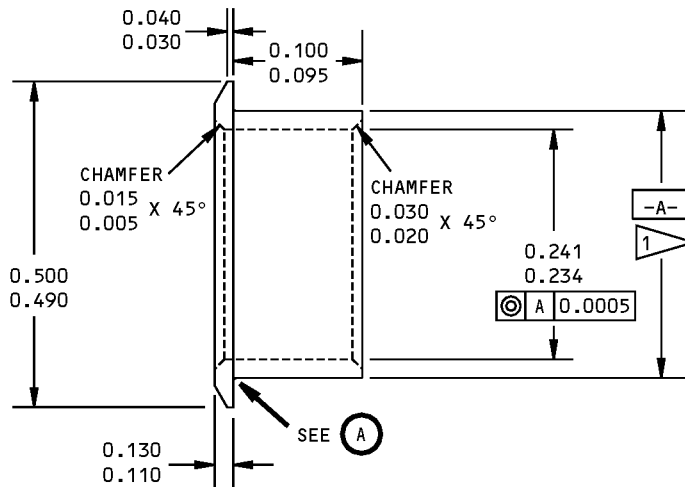
ALL ITEM NUMBERS REFER TO IPL FIG. 4
ALL DIMENSIONS ARE IN INCHES

Bearing Replacement
Figure 601

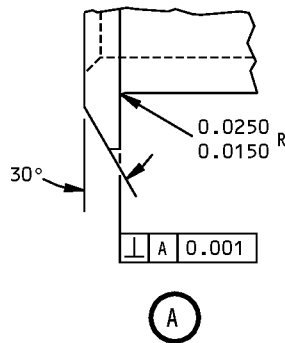
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REPAIR 21-1
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OVERSIZE REPLACEMENT BUSHING FOR BUSHING (142)



1 FINAL BUSHING OUTSIDE DIA EQUALS REPAIR DIA OF FITTING PLUS 0.0003-0.0015 INCH INTERFERENCE

REPAIR

63/ ALL MACHINED SURFACES, EXCEPT AS NOTED

MATERIAL: AL-NI-BR PER AMS4640 OR 4880

FINISH: CADMIUM PLATE TO MEET THE REQUIREMENTS OF QQ-P-416, TYPE 2, CLASS 2 (PLATING IN BORE OPTIONAL)

ITEM NUMBERS REFER TO IPL FIG. 1

ALL DIMENSIONS ARE IN INCHES

DIMENSIONS AFTER PLATING

Oversize Bushing Details
Figure 602

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REPAIR 21-1

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BEARING PLATE ASSEMBLY - REPAIR 22-1

65C34548-1, 65C34549-1

1. General

- A. This procedure has the data necessary to repair the bearing plate assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for list of applicable standard practices identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for list of applicable consumable codes identified in this procedure.
- D. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- E. Refer to IPL Figure 2 for the item numbers.

2. Bushing (118, 198) Replacement

- A. Remove the bushings.
- B. Install bushings using wet sealant, A00247 per SOPM 20-50-03.
- C. Machine bushings per REPAIR 22-1, Figure 601.
- D. Chamfer inside diameter both sides of bushings 0.005-0.015 x 45 degrees.
- E. Fillet seal periphery of bushing flanges using sealant, A00247.

3. Bearing Replacement (105, 185)

- A. Remove the bearing.
- B. Install bearing using grease, D00015 per SOPM 20-50-03.

4. Laminated Shim Replacement (122, 202)

- A. Remove the shim.
- B. Install laminated shim.
- C. Peel shim as required for fit. No gap allowed between retainer (121, 201) and bearing (105, 185).
- D. Maximum fastener pull up to be 0.005 inches.

5. Refinish

- A. Laminated shim (122, 202) – Apply primer, C00259 (F-20.02) after peeling.
- B. Bearing plate assembly (40E, 40G, 130C, 130D) – Chromic acid anodize per coating, C00851 and apply primer, C00259 per SOPM 20-41-02. Apply enamel coating, C50075 (SRF.14.9813) all over and topcoat fillet seals. Omit primer and enamel in bushing and bearing holes, keeper and relief bores, and faying surfaces. Material: Retainer plate (121, 202) - Bearing support (128, 207) - 7075-T7351 Al alloy 15-5 PH CRES, 180-200 KSI.

6. Repair Assembly REPAIR 22-1, Figure 601

- A. Assemble retainer plate (121, 201), laminated shim (122, 202) and bearing support (128, 207) using bolts (83, 166), plain washer (91, 176) under nut (103, 181).
- B. Tighten nuts (103, 181) within 50-75 pound-inches torque range.
- C. Install lubrication fittings (70, 160). After bearing assembly is complete, apply grease, D00015 to lube fitting until grease appears at inside diameter of bearing (105, 185). Use a preset grease gun with nozzle pressure not to exceed 1,000 psi and flow rate not to exceed 0.50 pints per minute.

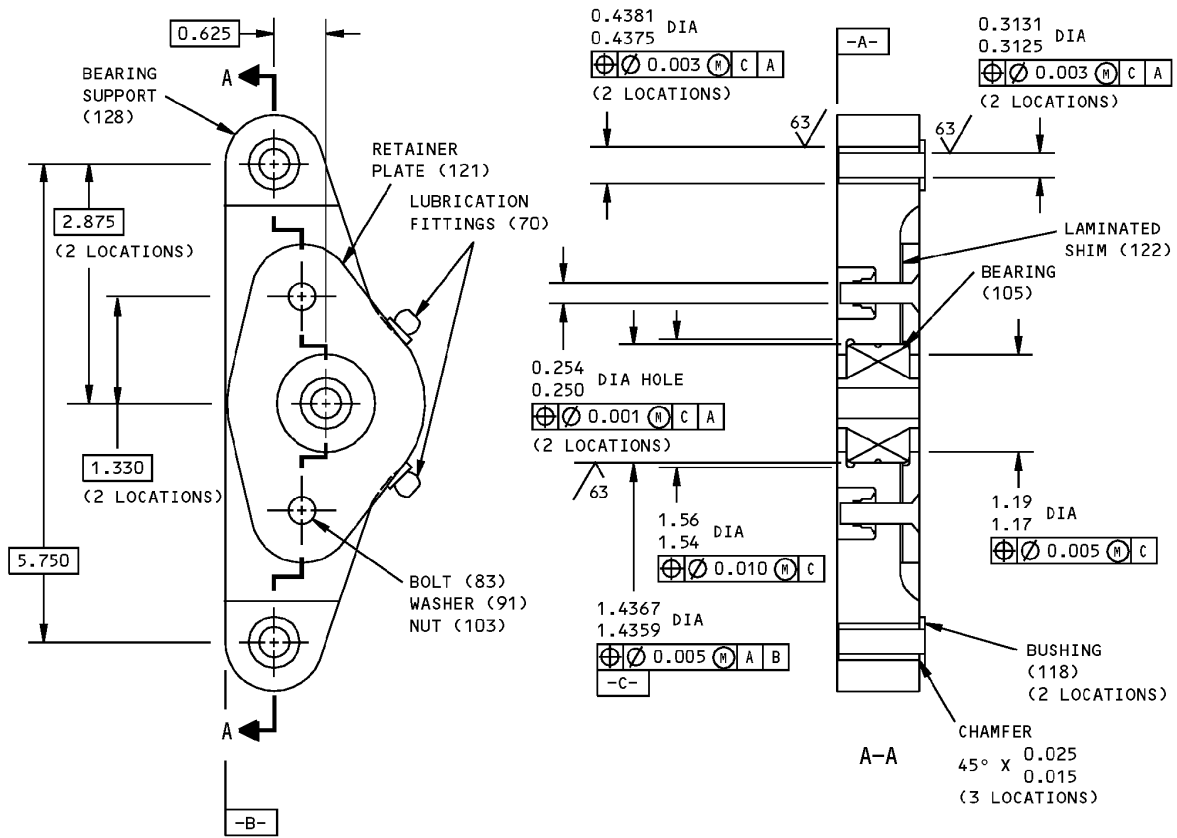
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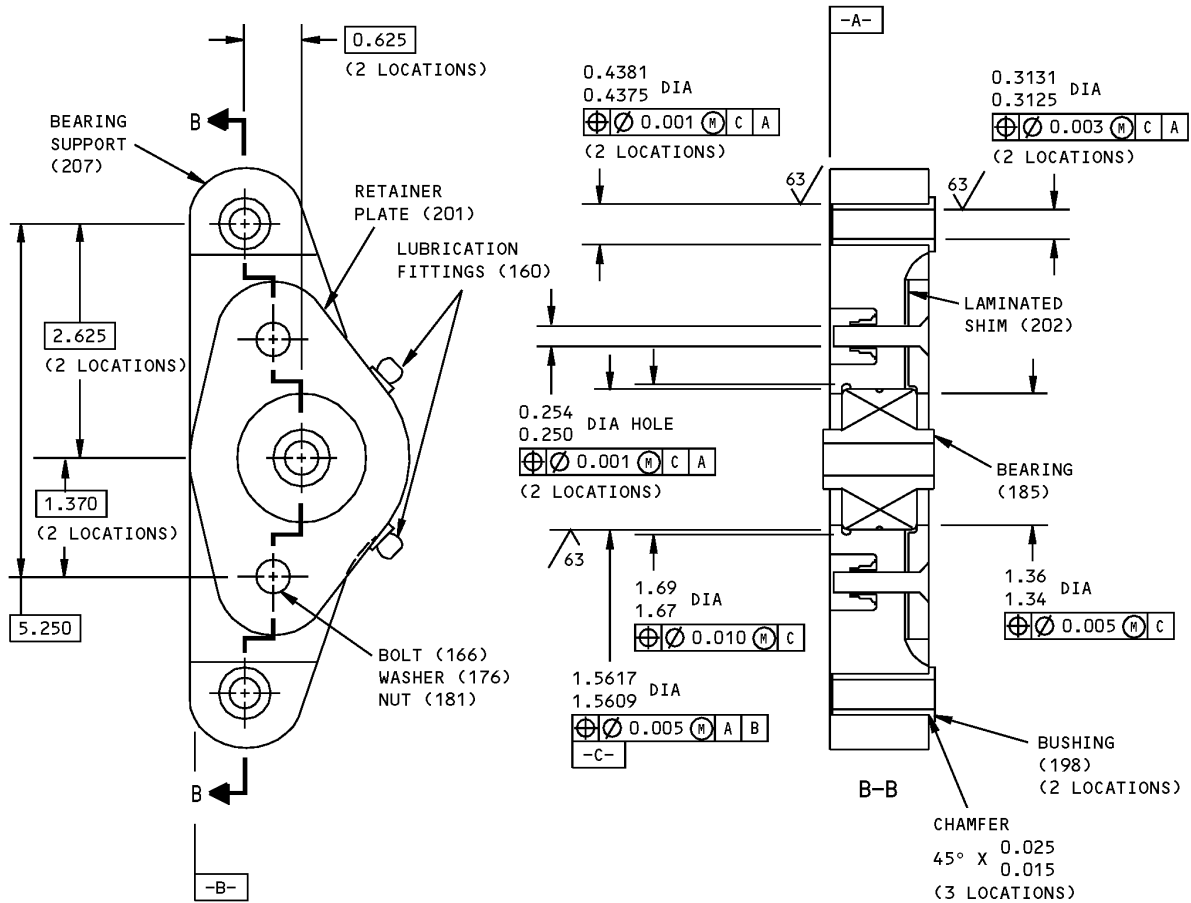
ITEM NUMBERS REFER TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES

65C34549-1 Bearing Plate Repair
Figure 601 (Sheet 1 of 2)

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REPAIR 22-1
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ITEM NUMBERS REFER TO IPL FIG. 2
ALL DIMENSIONS ARE IN INCHES

65C34549-1 Bearing Plate Repair
Figure 601 (Sheet 2 of 2)

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REPAIR 22-1
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COMPONENT MAINTENANCE MANUAL

ASSEMBLY

1. General

- A. This procedure has the data necessary to assemble elevator assembly and tab assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.

2. Equipment

NOTE: Equivalent substitutes can be used.

- A. Tab Hinge Center Line Alignment Tool – no part number assigned. To make this tool, refer to SB 737-55-1049.

3. Assembly

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
C50075	Coating - Exterior Protective Enamel, Gray	BMS10-60, Type II, BAC707 Gray

B. References

Reference	Title
SOPM 20-11-03	REPAIR OF ELECTRICAL TERMINATIONS AND ELECTRICAL BONDING AREAS
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-50-01	BOLT AND NUT INSTALLATION
SOPM 20-50-02	INSTALLATION OF SAFETYING DEVICES
SOPM 20-50-10	APPLICATION OF STENCILS, INSIGNIA, SILK SCREEN, PART NUMBERING AND IDENTIFICATION MARKINGS
737 SRM 51-60-00	Structural Repair Manual
737 SRM 51-80-6	Structural Repair Manual

C. Assembly procedures

CAUTION: USE ONLY TITANIUM OR CORROSION RESISTANT STEEL FASTENERS WITH GRAPHITE COMPONENTS. DO NOT SUBSTITUTE ALUMINUM OR PLATED ALLOY STEEL BECAUSE GALVANIC CORROSION WILL OCCUR AND SHORTEN THE SERVICE LIFE.

- (1) Use standard industry practices and these steps. Unless specified otherwise, tighten all fasteners to standard torques (SOPM 20-50-01). For fastener flushness requirements see ASSEMBLY, Figure 701.
- (2) Elevator and Tab Assembly (IPL Figure 1)

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ASSEMBLY
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- (a) Install weights (75, 80) with bolts (85), washers (90), and nuts (95).
 - (b) Check balance of elevator assembly per 737 SRM 51-60-00 or 737 SRM 51-80-6.
 - NOTE:** Grip length of bolts (85) depends on number of weights used. Actual quantity of weights (75, 80) will be determined by static balance of elevator. All bolt holes are to be used.
 - (c) Install hinge cover panels (145, 150, 155, 160, 165, 170, 175, 180) with bolts (190).
 - (d) Install trailing discharger retainer (40) with rivets (45) in base and adapter assembly (50) and rivets (55) in inserts (55).
 - (e) Apply fillet seal using sealant, A00247 around edge of base and adapter assembly (50) and reference structure. If equipped with conductive frame (295, 300, 305, 310), fillet seal with sealant, A00247 around edge of base and adapter assembly (50) after installation.
 - (f) Check electrical resistance between base and adapter assembly (50) and reference structure per SOPM 20-11-03. Resistance shall not exceed 0.01 ohm.
 - (g) Check conductivity between base and adapter assembly (50) and conductive frame (295, 300, 305, 310) if so equipped. Resistance shall not exceed 0.01 ohm. Check conductivity between discharger base (50) and jumper cable (335). Resistance shall not exceed 0.03 ohm.
 - (h) Install inboard tip fairing (265) with bolts (270).
 - (i) Install tip weight (120) with bolts (125), washers (130), and nuts (135).
 - (j) Pack bolt (125) counterbores in weight (120) using sealant, A00247.
 - (k) Install seal (195), angles (230, 235) with bolts (200, 205), fillers (210), washers (215, 220), and nuts (225).
 - (l) Install identification plates (240, 245) with rivets (250) using sealant, A00247 on fay surfaces and rivets.
- (3) Front Spar Installation (IPL Figure 2)
- (a) Install covers (10) with bolts (15) and washers (20, 25).
 - (b) Use sealant, A00247 at stations 24.9, 31.0, 36.0, 39.02, 66.54, 121.59, 176.64, and 213.32 as follows:
 - 1) Install all fitting/spar attach fasteners with wet sealant, A00247.
 - 2) Fillet seal all intersecting surfaces and all adjacent edges common to metal parts, shims, and graphite/epoxy surfaces and edges with sealant, A00247
 - 3) Fay surface seal and fill all voids and cavities between fittings/spar surfaces with sealant, A00247.
 - (c) At station 24.90 install mastarm fitting assemblies (415, 420) with bolts (425, 430, 435, 440, 445, 450, 455, 460), washers (465, 470, 475), collars (480, 485), bonding jumper (495), and nuts (490).
 - (d) At station 39.02 install hinge fitting assembly (705) with bolts (710, 725), washers (715, 730), collars (720), and nuts (735).

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ASSEMBLY

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CAUTION: COMPONENTS OF BEARING PLATE ASSEMBLY (40, 40A, 40B, 40C, 130, 130A, 130B), OTHER THAN FASTENERS ARE A MATCHED SET. CAP (120, 200) AND PILLOW BLOCK (125, 205) OR CAP (120, 200) AND PILLOW BLOCK (127, 206) ARE A MATCHED SET.

- (e) At stations 24.90 and 39.02 install bearing plate assemblies (40, 40A, 40B, 40C, 130, 130A, 130B) with bolts (45, 135), washers (50, 60, 140, 150), bushings (55, 145), and nuts (65, 155).
 - (f) At station 31.0 install bracket assembly (585) with bolts (590, 595, 600), washers (605), and collars (610) or bolts (591, 596, 601), washers (606, 607, 608) and nuts (611).
 - (g) At station 36.0 install bracket assembly (645) with bolts (650, 655, 660), washers (665), and collars (670) or bolts (651, 656, 661), washers (666, 667, 668) and nuts (671).
 - (h) At station 66.54 install hinge fitting assembly (768) with bolts (771, 780), washers (774, 783), collars (777), and nuts (786).
 - (i) At station 66.54 install bearing plate assembly (210, 410R, 410S) with bolts (215, 411A, 416A, 418A), washers (220, 230, 411B, 411D, 416B, 416D, 418B, 418D), bushings (225, 411C, 416C, 418C), and nuts (235, 411E, 416E, 418E).
 - (j) At station 121.59 install hinge fitting assembly (820) with bolts (822, 828), washers (824, 830), collars (826), and nuts (832).
 - (k) At station 121.59 install bearing plate assembly (260, 411R, 411S) with bolts (265, 412A, 416L, 418L), washers (270, 280, 412B, 412D, 416M, 416O, 418M, 418O), bushing (275, 412C, 416N, 418N), and nuts (285, 412E, 416P, 418P).
 - (l) At station 176.64 install hinge fitting assembly (858) with bolts (860, 866), washers (862, 868), collars (864), and nuts (870).
 - (m) At station 176.64 install bearing plate assembly (310, 412R, 413S) with bolts (315, 413A, 417A, 419A), washers (320, 330, 413B, 413D, 417B, 417D, 419B, 419D), bushing (325, 413C, 417C, 419C), and nuts (335, 413E, 417E, 419E).
 - (n) At station 213.32 install hinge fitting assembly (896), with bolts (898, 904, 910), washers (900, 906, 912), collars (902), bonding jumper (916), and nuts (914).
 - (o) At station 213.32 install bearing plate assembly (360, 413R, 413S) with bolts (365, 414A, 417L, 419L), washers (370, 380, 414B, 414D, 417M, 417O, 419M, 419O), bushing (375, 414C, 417N, 419N), and nuts (385, 414E, 417P, 419P).
 - (p) Touchup hinge side of graphite/epoxy spar and assembled hinge fittings using enamel coating, C50075 per SOPM 20-41-02 at eight station locations.
- (4) Rear Spar Installation (IPL Figure 3)
- (a) Install hinge fitting assemblies (30, 120) with shims (20, 110) and fillers (25, 115) using bolts (35, 75, 125, 165), washers (40, 85, 90, 130, 175, 180), and nuts (80, 170) as follows:
 - 1) Install laminated shims as required by delaminating 0.003-inch laminations to fill gap between fillers (25, 115) and fitting assemblies (30, 120). Maximum allowable pullup after shimming is 0.005-inch.
 - 2) Apply sealant, A00247 to fay surfaces of fillers (25, 115) and hinge fitting assemblies (30, 120) and fill all cavities.
 - 3) Install bolts (35, 75, 125, 165) with wet sealant, A00247.
 - 4) Cap seal installed nutplates (50, 140) with sealant, A00247 after installation of hinge fitting assemblies (30, 120).

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ASSEMBLY

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(5) Tab Installation (IPL Figure 4)

- (a) Install shims (220) and mast fitting assembly (235) on reference front spar structure with bolts (180, 185, 190), washers (200, 205), and nuts (195).

CAUTION: IF YOU INSTALL A NEW HINGE FITTING (165D, 165E) 65C25798-11, -12, REFER TO SB 55-1049 FOR PROCEDURES TO DRILL HOLES AND ALIGN THE HINGES.

- (b) Install hinge fitting assemblies (105, 106, 107, 165) and shims (150) with bolts (155, 155A), washers (160, 160A, 161) and nuts (162).

NOTE: Hinge fitting assemblies should be installed with the zerk fitting located toward the lower surface of the tab.

- (c) If necessary, install a replacement decal (100) per SOPM 20-50-10.
- (d) Make a check of the balance of the elevator assembly per 737 SRM 51-60-00 or 737 SRM 51-80-6.
- (e) Install the elevator tab assembly (90) onto the elevator assembly as follows:
- 1) Position the elevator tab assembly (90) onto the rear spar of the elevator assembly.
 - 2) For the elevator control tab installations, 65C25796-1, -2, install as follows:
 - a) For hinges 1, 3, and 4, install bolt (10A), washer (15), bushings (26 and 21), washer (35), and nut (40) as shown in ASSEMBLY, Figure 702.

CAUTION: DO NOT BOTTOM THE NUT ON THE UNTHREADED PART OF THE BOLT WHEN THE NUT IS TIGHTENED. A LOOSE CLAMPUP CAN RESULT IN EXCESSIVE WEAR TO THE PARTS WHICH CAN CAUSE AIRPLANE VIBRATION.

CAUTION: DO NOT EXTEND THE COTTER PIN IN THE CASTELLATED NUT MORE THAN 50% OF THE COTTER PIN DIAMETER ABOVE THE TOP OF THE SLOT.

< 1 > Use washers (38, 38A), as required, to make sure that nut (40) does not touch the unthreaded part of the bolt when tightened.

NOTE: Washer (38) is 0.063 inch thick and washer (38A) is 0.032 inch thick.

< 2 > Use washers (38, 38A), as required, to align the cotter pin hole with the slot on the castellated nut (40) when tightened. See ASSEMBLY, Figure 705.

< 3 > Tighten the nut (40) to 50-80 pound-inches.

< 4 > Install the cotter pin (42) as shown in SOPM 20-50-02.

- b) For hinge 2, install the bolt (10A), washer (15), bushings (26, 30A), washer (35), and nut (40) as shown in ASSEMBLY, Figure 702.

- c) Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a) < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a) < 4 > .

NOTE: See SB 737-55A1070 for controlled application of tamper-proof putty on the head of the hinge bolt at the clamped hinge location.

- 3) For the elevator control tab installations, 65C25796-19, -20, install as follows:

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ASSEMBLY

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- a) For hinges 2, 3, and 4, install the bolt (10), washer (15), bushing (20), washer (35A) and nut (40) as shown in ASSEMBLY, Figure 703.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- b) For hinge 1, install the bolt (10), washer (15), bushings (25, 30), washer (35A), and nut (40) as shown in ASSEMBLY, Figure 703.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- NOTE:** See SB 737-55A1070 for controlled application of tamper-proof putty on the head of the hinge bolt at the clamped hinge location.
- 4) For the elevator control tab installations, 65C25796-27, -28, install as follows:
- a) For hinges 2, 3 and 4, install the bolt (10G), washer (15A), bushings (20A), washer (35B) and nut (40) as shown in ASSEMBLY, Figure 704.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- b) For hinge 1, install the bolt (10F), washer (15A), bushing (25A), washer (21F), washer (36A), and nut (40) as shown in ASSEMBLY, Figure 704.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- NOTE:** See SB 737-55A1070 for controlled application of tamper-proof putty on the head of the hinge bolt at the clamped hinge location.
- 5) For the elevator control tab installations, 65C25796-29, -30, install as follows:
- a) For hinges 2, 3, and 4, install the bolt (10I), washer (15A), bushing (20A), washer (35B), and nut (40) as shown in ASSEMBLY, Figure 704.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- b) For hinge 1, install the bolt (10H), washer (15A), bushing (25A), washer (21F), washer (36A), and nut (40) as shown in ASSEMBLY, Figure 704.
- < 1 > Tighten the nut (40) and install the cotter pin (42) as shown in ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 1 > thru ASSEMBLY, Paragraph 3.C.(5)(e)2)a < 4 > .
- NOTE:** See SB 737-55A1070 for controlled application of tamper-proof putty on the head of the hinge bolt at the clamped hinge location.
- (f) Make sure that the dimensions between the elevator tab assembly (90) and the elevator assembly are as specified in ASSEMBLY, Figure 706.
- (g) Attach bonding jumpers (80, 81) with bolts (50, 55), washers (60, 65, 70), and nuts (75).
- NOTE:** Bonding jumpers (80, 81) to be installed between washer (65) and washer (70).

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ASSEMBLY

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- (6) Bay No. 2 Balance Panel Assembly (IPL Figure 5)
 - (a) Install hinge half (50) and fitting (65) using bolts (70), washers (75), and nuts (80).
 - (b) Install seals (95, 110, 125, 140) using bolts (100, 115, 130, 145) and washers (105, 120, 135, 150).
 - (c) Temporarily attach remaining parts to be assembled on airplane installation as follows:
 - 1) Pins (10, 20, 25)
 - 2) Washers (15)
 - 3) Hinge half (30)
 - 4) Bolts (40, 45)
 - 5) Seal (60)
- (7) Bay No. 3 Balance Panel Assembly (IPL Figure 6)
 - (a) Install hinge half (55) and fitting (125) with shims (130, 135) using bolts (15).
 - (b) Install seals (65, 70, 95) with retainers (75, 80, 100, 105, 110) using bolts (85) and washers (90, 120).
 - (c) Temporarily attach remaining parts to be assembled on airplane installation as follows:
 - 1) Pins (20, 30, 35, 40)
 - 2) Washer (25)
 - 3) Hinge half (45)
 - 4) Bolts (10, 15)
 - 5) Seal (60)
- (8) Bay No. 4 Balance Panel Assembly (IPL Figure 7)
 - (a) Install hinge half (50) and fitting (65) with shim (85) using bolts (70), washers (75), and nuts (80).
 - (b) Install seals (90, 105, 120, 135) using bolts (95, 110, 125, 140) and washers (100, 115, 130, 145).
 - (c) Temporarily attach remaining parts to be assembled on airplane installation as follows:
 - 1) Pins (10, 20, 25)
 - 2) Washer (15)
 - 3) Hinge half (30, 35)
 - 4) Bolts (40, 45)
 - 5) Seal (60)

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ASSEMBLY

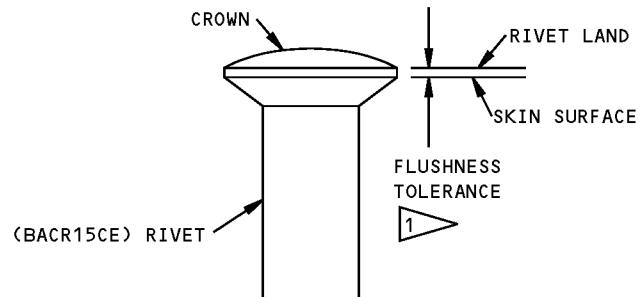
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1. ALL STEEL LOCKBOLTS AND HI-LOKS SHALL BE FLUSH WITHIN THE TOLERANCE 0.003 LOW TO 0.005 HIGH. STEEL FASTENER HEADS ARE NOT TO BE SHAVED.
2. ALL SCREWS AND BOLTS SHALL BE INSTALLED FLUSH WITHIN THE TOLERANCE 0.010 LOW TO 0.005 HIGH.
3. ALL BLIND FLUSH RIVETS SHALL BE DRIVEN WITH FLUSHNESS TOLERANCE OF THE HEAD -0.002 TO $+0.007$. THE LOCKING RING AND PULL STEM MAY PROTRUDE 0.017 MAXIMUM ABOVE SKIN SURFACE.
4. ALL ALUMINUM MS20426 RIVETS SHALL BE FLUSH WITHIN THE TOLERANCE 0.002 LOW TO 0.007 HIGH.
5. ALL ALUMINUM BACR15CE RIVETS SHALL BE FLUSH WITHIN THE TOLERANCE 0.000 LOW TO 0.005 HIGH EXCEPT A MAXIMUM OF 10% MAY BE FLUSH WITHIN THE TOLERANCE OF 0.0005 LOW TO 0.0070 HIGH WITH NO TWO ADJACENT. SEE DETAIL I.



DETAIL I

1 BACR15CE RIVET FLUSHNESS IS TO BE DETERMINED BY THE HEIGHT OF THE RIVET LAND (NOT THE CROWN) ABOVE THE SKIN SURFACE

ALL DIMENSIONS ARE IN INCHES

Fastener Flushness Requirements
Figure 701

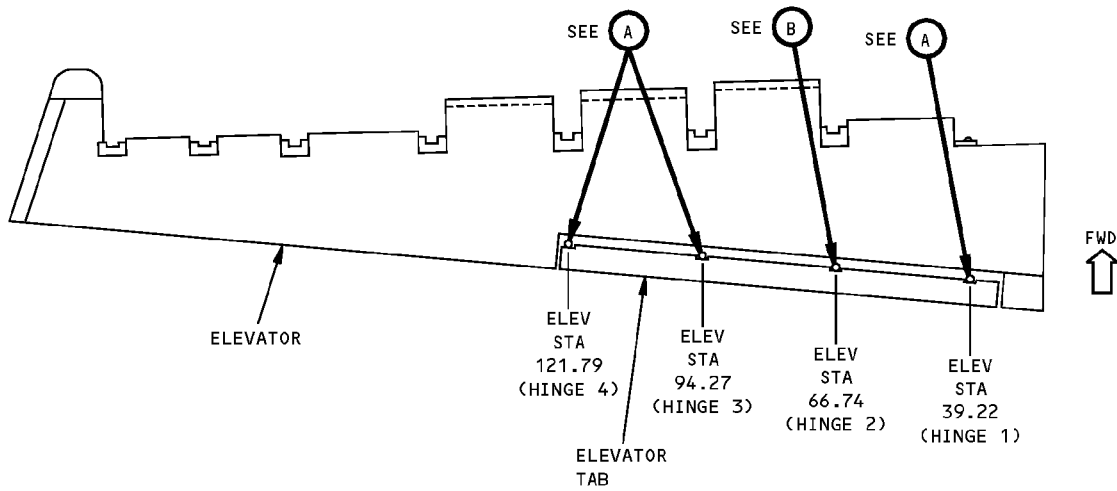
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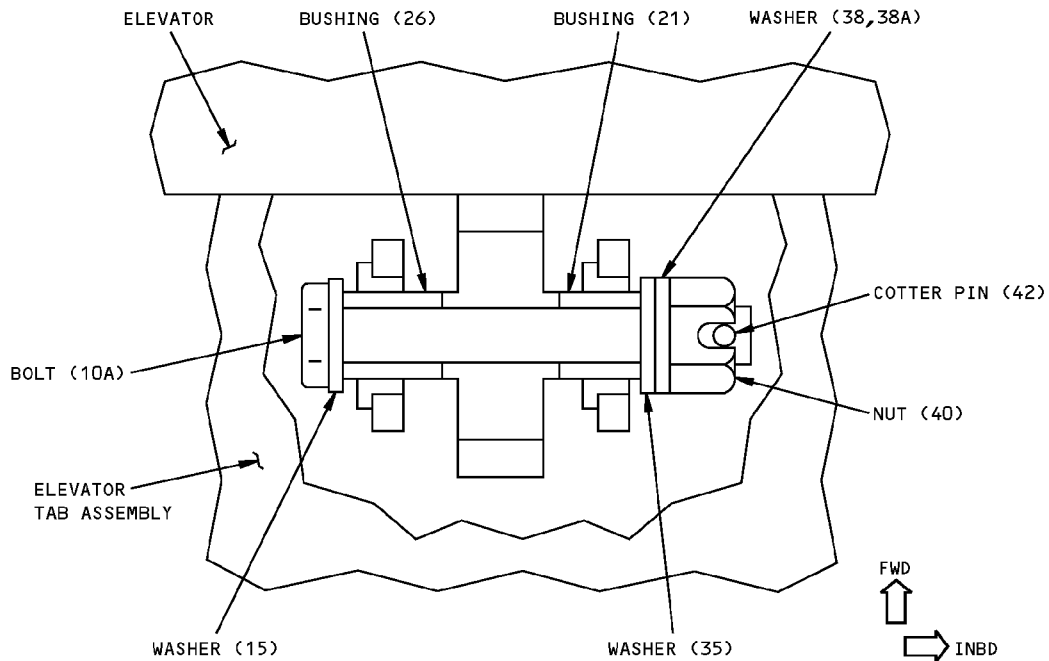
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ELEVATOR CONTROL TAB INSTALLATION 65C25796-1,-2



TYPICAL FOR ELEVATOR STATIONS
39.22, 94.27, AND 121.79

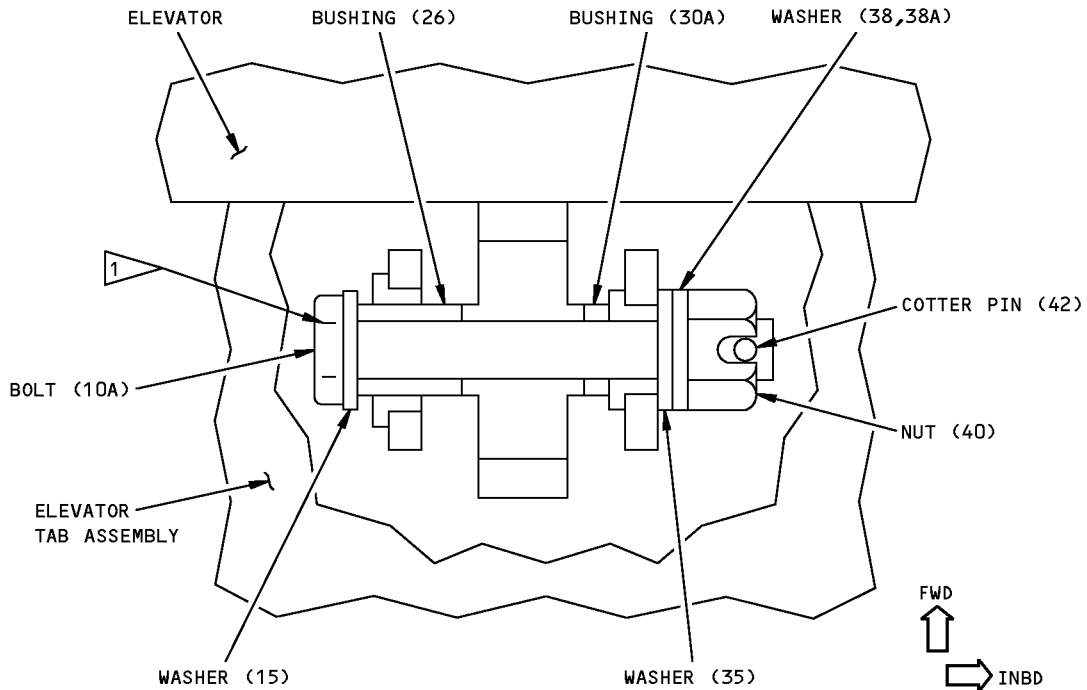
(A)

Elevator Tab Assembly
Figure 702 (Sheet 1 of 2)

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ASSEMBLY
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ELEVATOR STATION 66.74

(B)

1 SEE SB 737-55A1070 FOR CONTROLLED APPLICATION OF TAMPER-PROOF PUTTY ON THE HEAD OF THE HINGE BOLT AT THE CLAMPED HINGE LOCATION

ITEM NUMBERS REFER TO IPL FIG. 4

L06153 S00041003186_V2

Elevator Tab Assembly
Figure 702 (Sheet 2 of 2)

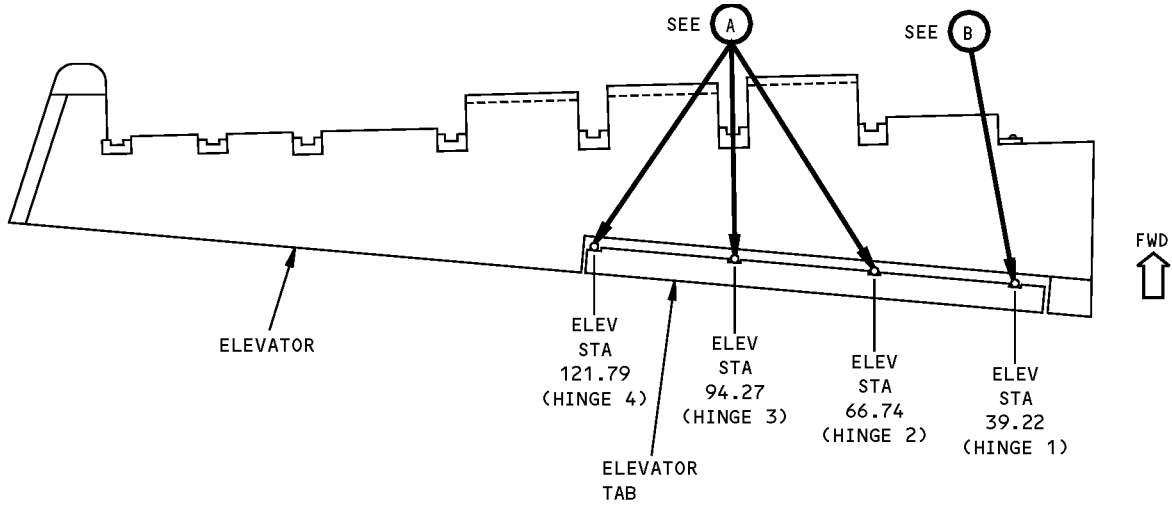
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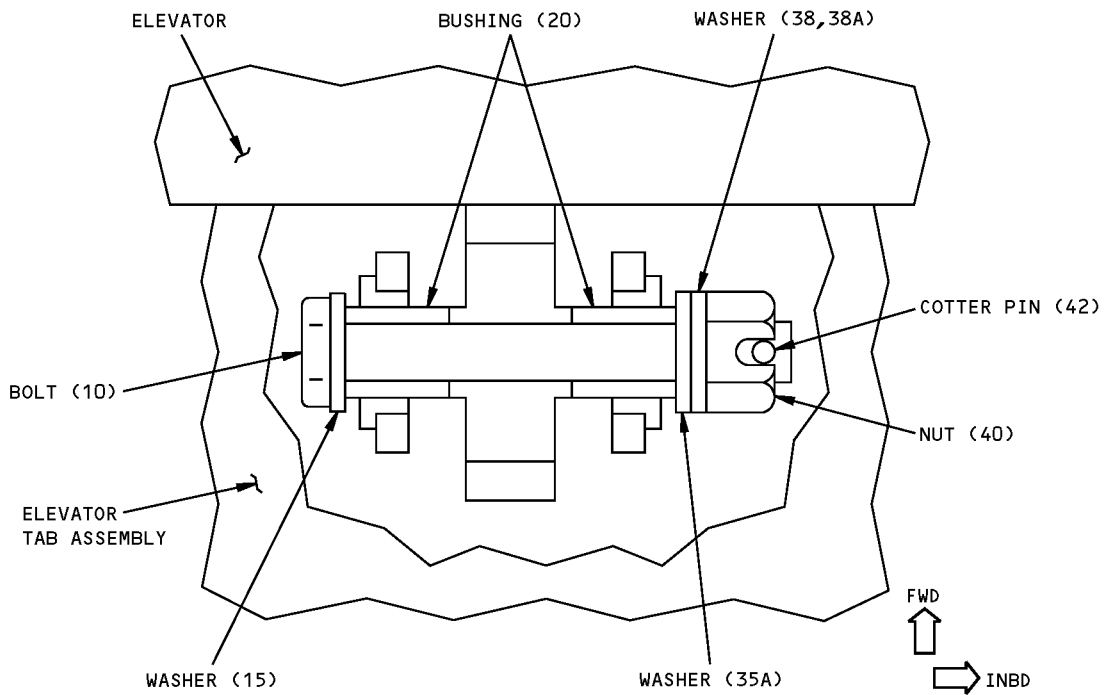
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COMPONENT MAINTENANCE MANUAL



ELEVATOR CONTROL TAB INSTALLATION 65C25796-19,-20



TYPICAL FOR ELEVATOR STATIONS
66.74, 94.27, AND 121.79

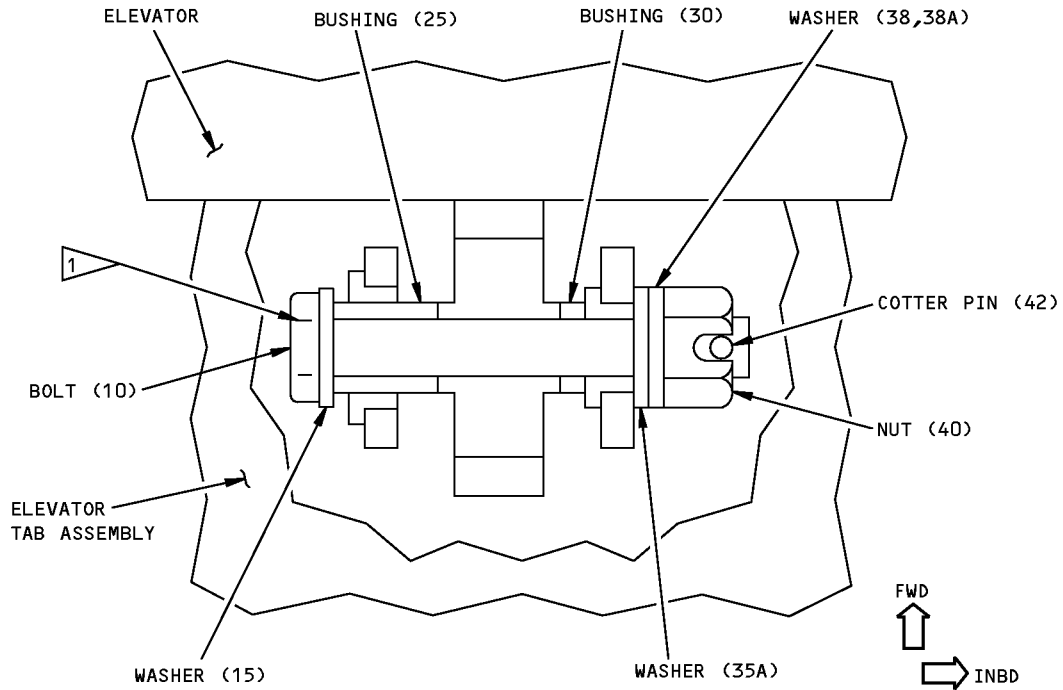
(A)

Elevator Tab Assembly
Figure 703 (Sheet 1 of 2)

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ASSEMBLY
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ELEVATOR STATION 39.22

(B)

1 SEE SB 737-55A1070 FOR CONTROLLED APPLICATION OF TAMPER-PROOF PUTTY ON THE HEAD OF THE HINGE BOLT AT THE CLAMPED HINGE LOCATION

ITEM NUMBERS REFER TO IPL FIG. 4

L06155 S00041003188_V2

Elevator Tab Assembly
Figure 703 (Sheet 2 of 2)

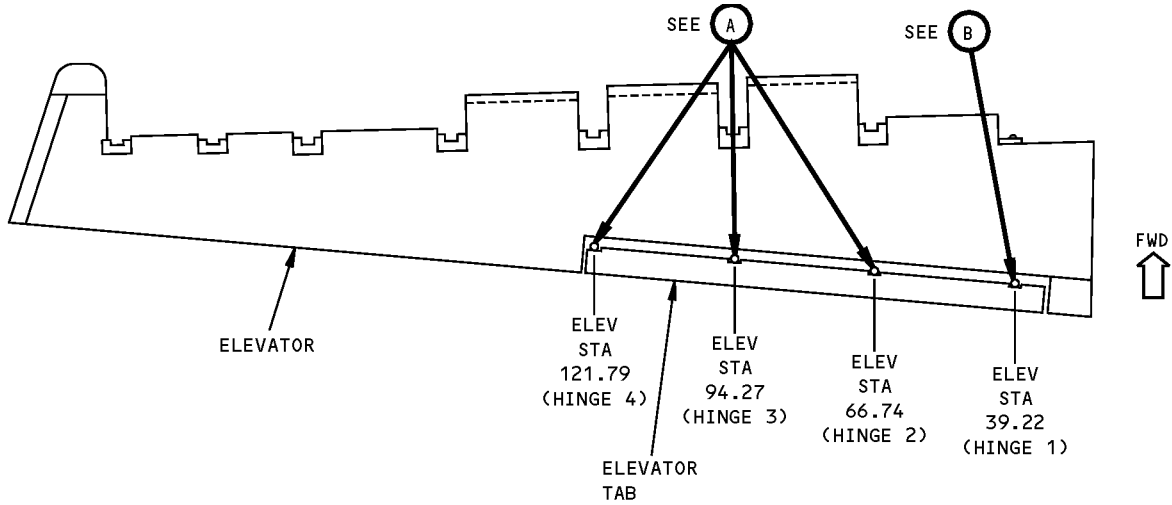
55-20-16

ASSEMBLY

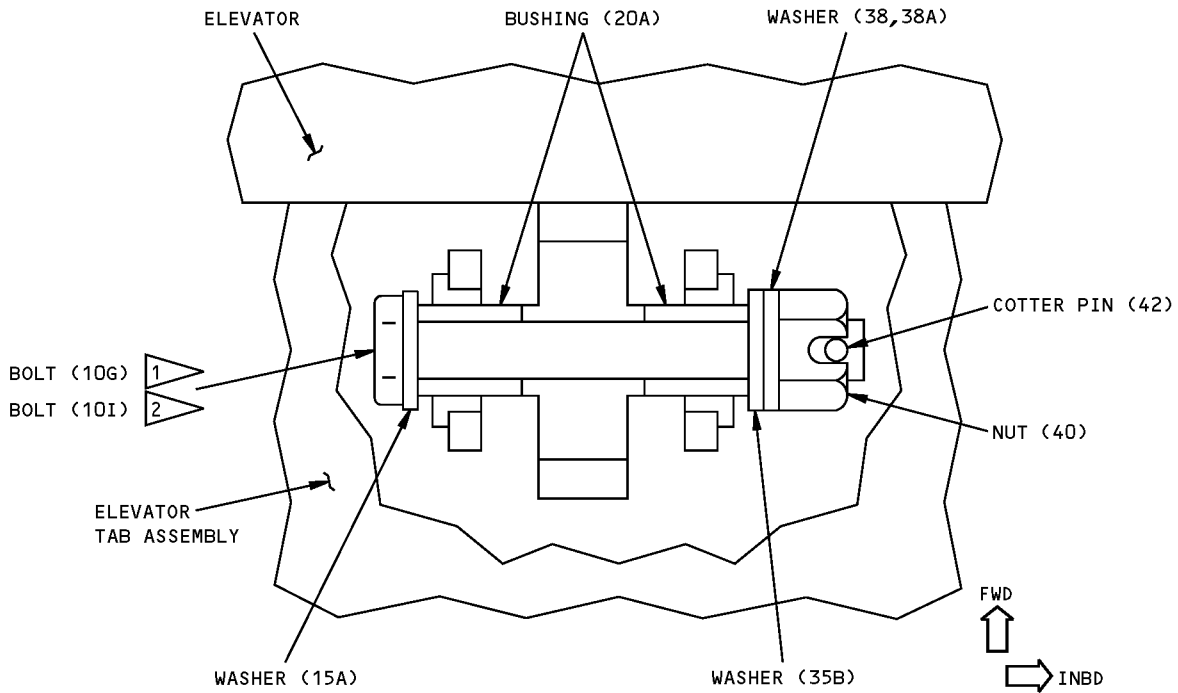
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COMPONENT MAINTENANCE MANUAL



ELEVATOR CONTROL TAB INSTALLATION 65C25796-27 THRU -30



TYPICAL FOR ELEVATOR STATIONS
66.74, 94.27, AND 121.79

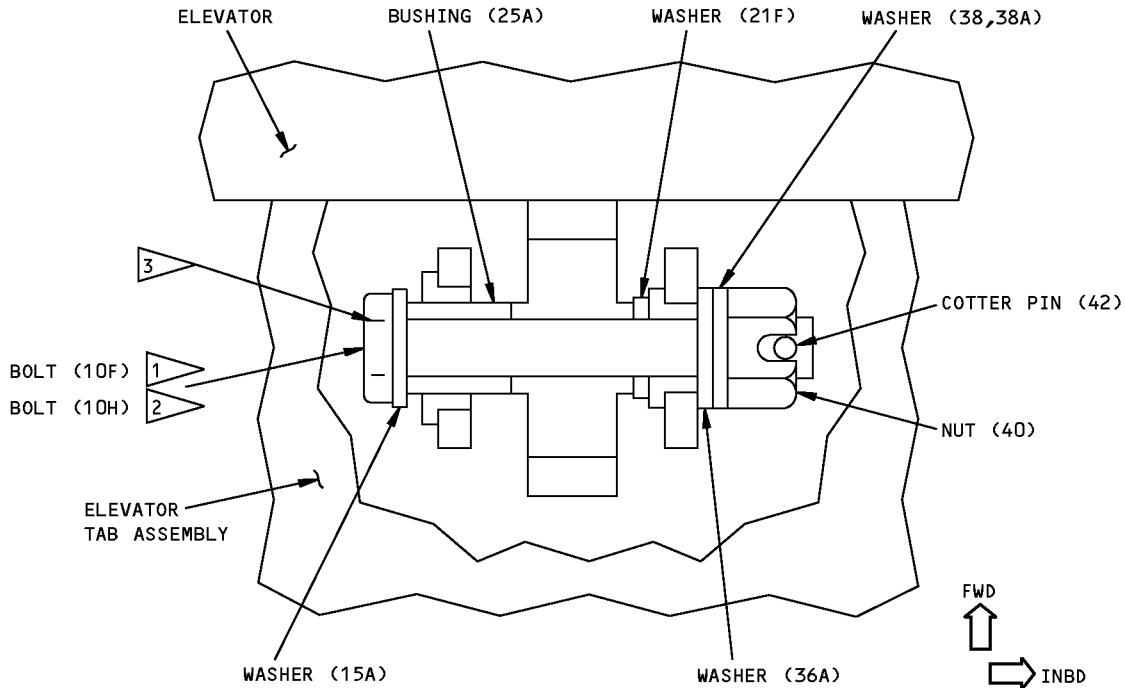
A

Elevator Tab Assembly
Figure 704 (Sheet 1 of 2)

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ASSEMBLY
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COMPONENT MAINTENANCE MANUAL



ELEVATOR STATION 39.22

(B)

- 1 FOR 65C25796-27,-28 INSTALLATION
- 2 FOR 65C25796-29,-30 INSTALLATION
- 3 SEE SB 737-55A1070 FOR CONTROLLED APPLICATION OF TAMPER-PROOF PUTTY ON THE HEAD OF THE HINGE BOLT AT THE CLAMPED HINGE LOCATION

ITEM NUMBERS REFER TO IPL FIG. 4

L06162 S00041003190_V2

Elevator Tab Assembly
Figure 704 (Sheet 2 of 2)

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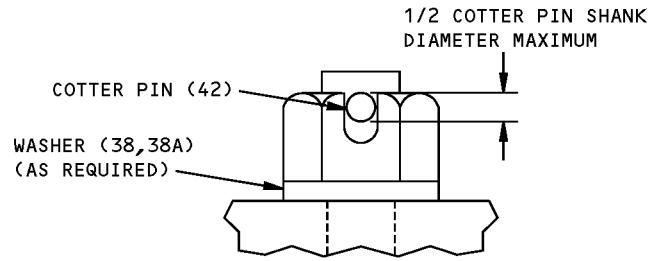
ASSEMBLY

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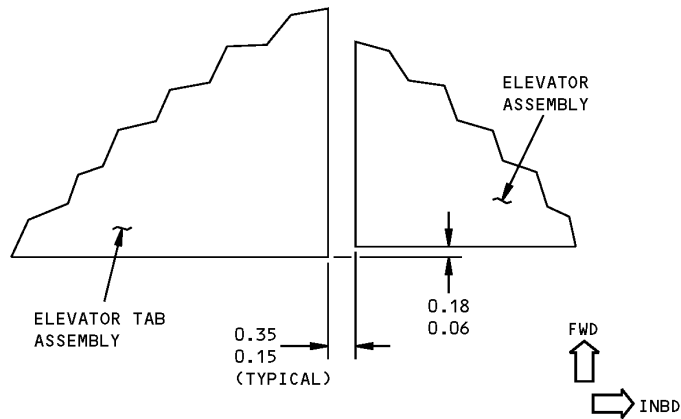


COMPONENT MAINTENANCE MANUAL



ITEM NUMBERS REFER TO IPL FIG. 4

Cotter Pin Installation
Figure 705



ALL DIMENSIONS ARE IN INCHES

Elevator Tab Assembly and Elevator Assembly Installation
Figure 706

55-20-16

ASSEMBLY

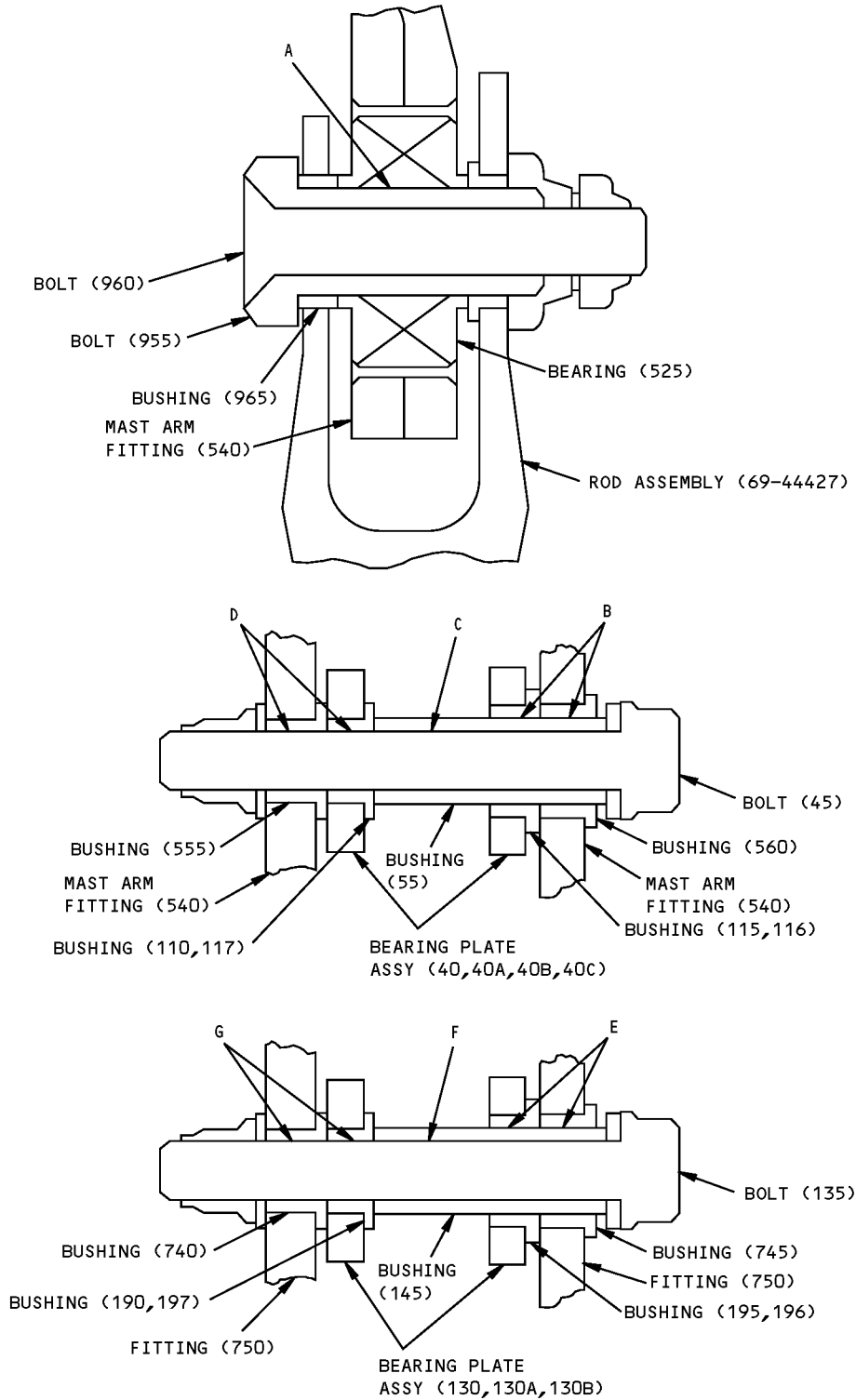
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COMPONENT MAINTENANCE MANUAL

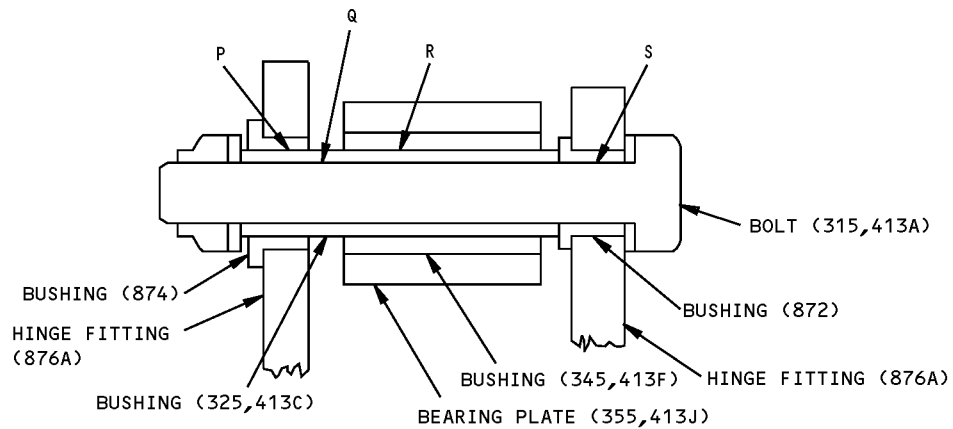
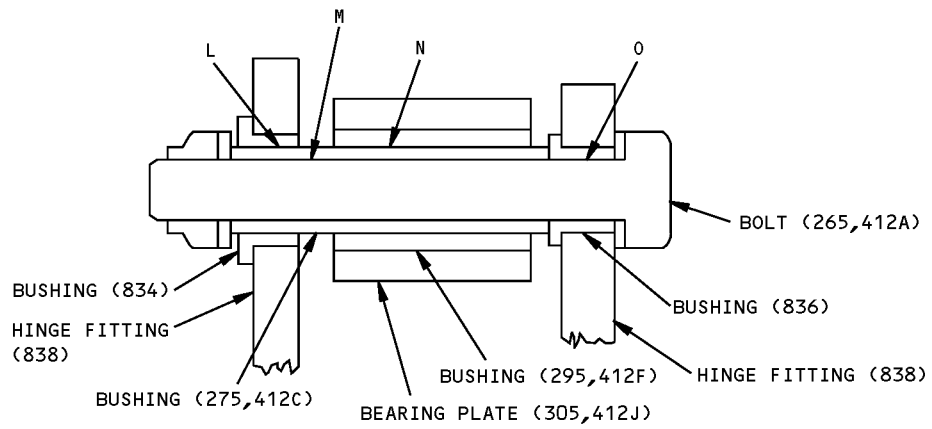
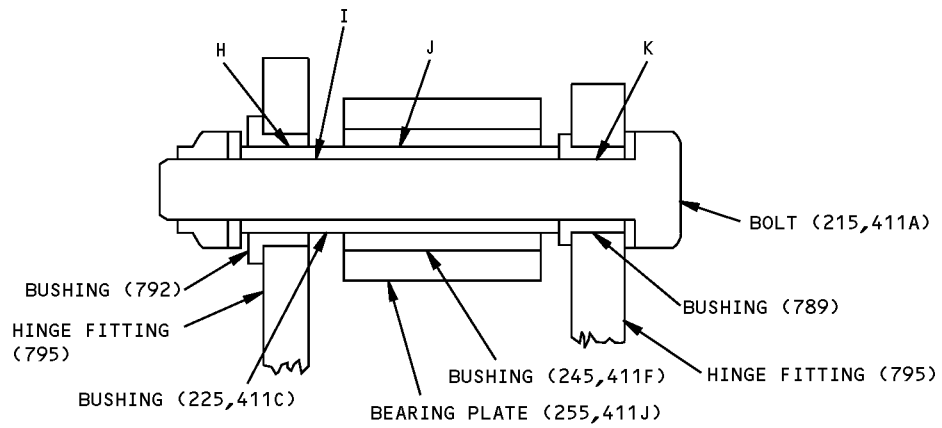
FITS AND CLEARANCES



ALL ITEM NUMBERS REFER TO IPL FIG. 2

Fits and Clearances
Figure 801 (Sheet 1 of 5)

COMPONENT MAINTENANCE MANUAL

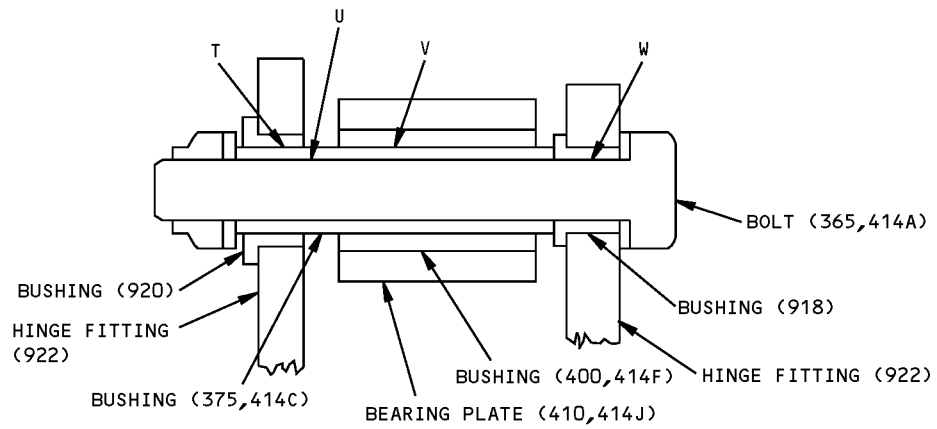


ALL ITEM NUMBERS REFER TO IPL FIG. 2

Fits and Clearances
Figure 801 (Sheet 2 of 5)



COMPONENT MAINTENANCE MANUAL



ALL ITEM NUMBERS REFER TO IPL FIG. 2

Fits and Clearances
Figure 801 (Sheet 3 of 5)



COMPONENT MAINTENANCE MANUAL

Ref Letter Fig.801	Mating Item No. IPL Fig.2	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
A	ID 525	0.6245	0.6250	0.0005	0.0015		0.6271	0.0035
	OD *E1	0.6235	0.6240			0.6225		
B	ID 115,116 560	0.4375	0.4381	0.0005	0.0016		0.4400	0.0035
	OD 55	0.4365	0.4370			0.4353		
C	ID 55	0.3125	0.3130	0.0005	0.0020		0.3149	0.0040
	OD 45	0.3110	0.3120			0.3070		
D	ID 110,117 555	0.3125	0.3131	0.0005	0.0021		0.3149	0.0040
	OD 45	0.3110	0.3120			0.3070		
E	ID 195,196 745	0.4375	0.4381	0.0005	0.0016		0.4400	0.0035
	OD 145	0.4365	0.4370			0.4353		
F	ID 145	0.3125	0.3130	0.0005	0.0020		0.3149	0.0040
	OD 135	0.3110	0.3120			0.3070		
G	ID 190,197 740	0.3125	0.3131	0.0005	0.0021		0.3149	0.0032
	OD 135	0.3110	0.3120			0.3070		
H	ID 792	0.4375	0.4382	0.0005	0.0017		0.4410	0.0045
	OD 225,411C	0.4365	0.4370			0.4353		
I	ID 225,411C	0.3125	0.3130	0.0005	0.0020		0.3149	0.0045
	OD 215,411A	0.3110	0.3120			0.3070		
J	ID 245,411F	0.4380	0.4385	0.0010	0.0020		0.4411	0.0045
	OD 225,411J	0.4365	0.4370			0.4360		
K	ID 798	0.3125	0.3131	0.0005	0.0021		0.3149	0.0045
	OD 215,411A	0.3110	0.3120			0.3070		
L	ID 834	0.4375	0.4381	0.0005	0.0016		0.4408	0.0045
	OD 275,412C	0.4365	0.4370			0.4353		

Fits and Clearances
Figure 801 (Sheet 4 of 5)

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FITS AND CLEARANCES
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COMPONENT MAINTENANCE MANUAL

Ref Letter Fig.801	Mating Item No. IPL Fig.2	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
M	ID 275,412C	0.3125	0.3130	0.0005	0.0020		0.3149	0.0045
	OD 265,412A	0.3110	0.3120			0.3070		
N	ID 295,412F	0.4380	0.4385	0.0010	0.0020		0.4411	0.0045
	OD 275,412C	0.4365	0.4370			0.4360		
O	ID 836	0.3125	0.3131	0.0005	0.0021		0.3149	0.0045
	OD 265,412A	0.3110	0.3120			0.3070		
P	ID 874	0.3750	0.3756	0.0005	0.0016		0.3808	0.0060
	OD 325,413C	0.3740	0.3745			0.3729		
Q	ID 325,413C	0.2500	0.2505	0.0005	0.0020		0.2525	0.0040
	OD 315,413A	0.2485	0.2495			0.2480		
R	ID 345,413F	0.3755	0.3760	0.0010	0.0020		0.3784	0.0045
	OD 325,413C	0.3740	0.3745			0.3735		
S	ID 872	0.2500	0.2505	0.0005	0.0020		0.2560	0.0080
	OD 315,413A	0.2485	0.2495			0.2480		
T	ID 920	0.3750	0.3756	0.0005	0.0016		0.3790	0.0050
	OD 375,414C	0.3740	0.3745			0.3729		
U	ID 375,414C	0.2500	0.2505	0.0005	0.0020		0.2525	0.0040
	OD 365,414A	0.2485	0.2495			0.2480		
V	ID 400,414F	0.3755	0.3760	0.0010	0.0020		0.3784	0.0045
	OD 375,414C	0.3740	0.3745			0.3735		
W	ID 918	0.2500	0.2505	0.0005	0.0020		0.2540	0.0050
	OD 365,414A	0.2485	0.2495			0.2480		

* [1] BOLT (69-44683-2)

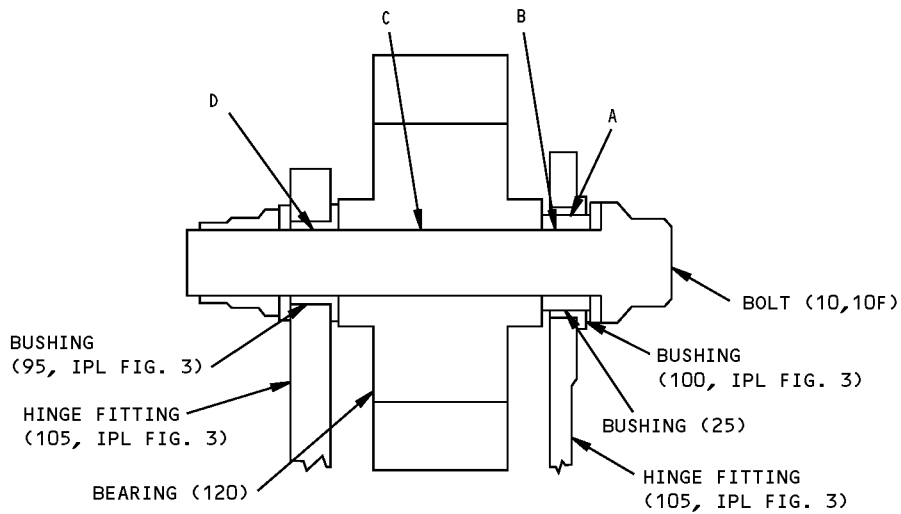
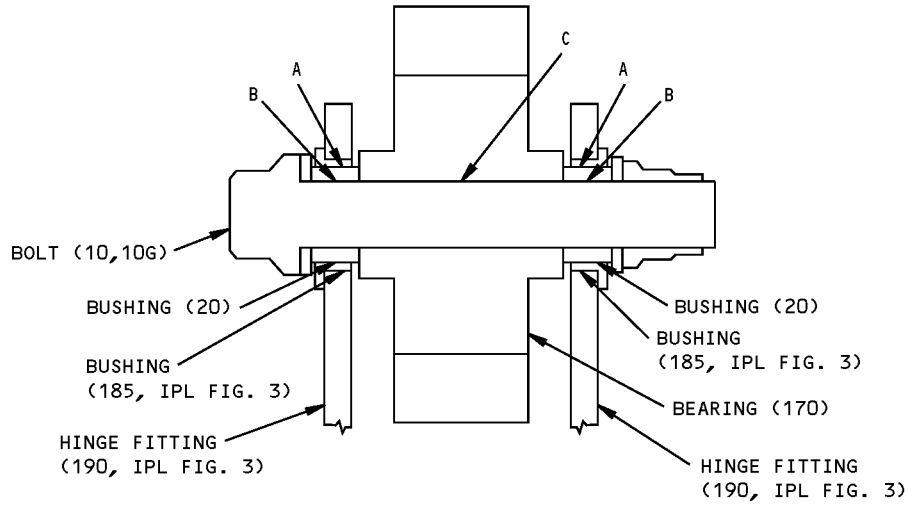
ALL DIMENSIONS ARE IN INCHES

Fits and Clearances
Figure 801 (Sheet 5 of 5)

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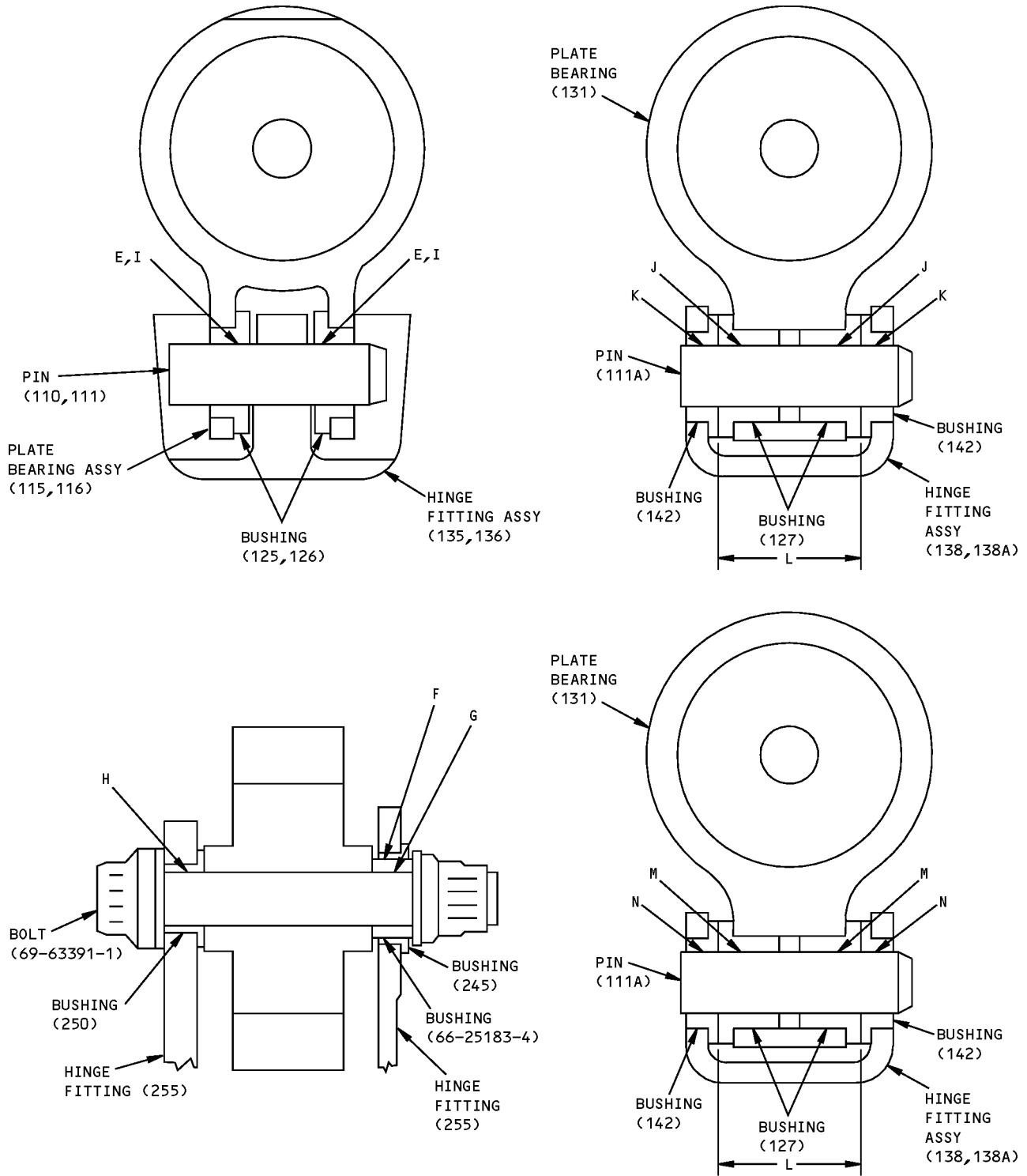
COMPONENT MAINTENANCE MANUAL



ALL ITEM NUMBERS REFER TO IPL FIG. 4
UNLESS OTHERWISE SPECIFIED

Fits and Clearances
Figure 802 (Sheet 1 of 4)

COMPONENT MAINTENANCE MANUAL











ALL ITEM NUMBERS REFER TO IPL FIG. 4

Fits and Clearances
Figure 802 (Sheet 2 of 4)








COMPONENT MAINTENANCE MANUAL

Ref Letter Fig.802	Mating Item No. IPL Fig.4 	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance 		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
B	ID 100, IPL FIG. 3 185, IPL FIG. 3	0.3750	0.3756	0.0005	0.0016		0.3800	0.0050
	OD 20 25	0.3740	0.3745			0.3728		
B	ID 20 25	0.2500	0.2505	0.0003	0.0011		0.2518	0.0050
	OD 10	0.2494	0.2497			0.2445		
C	ID 120 170	0.2495	0.2500	-0.0002	0.0006		0.2513	0.0050
	OD 10	0.2494	0.2497			0.2445		
D	ID 95, IPL FIG. 3	0.2500	0.2505	0.0003	0.0011		0.2518	0.0050
	OD 10	0.2494	0.2497			0.2445		
E	ID 125	0.2493	0.2498	0.0000	0.0008		0.2511	0.0025
	OD 110	0.2490	0.2493			0.2475		
F	ID 245	0.3761	0.3766	0.0000	0.0010		0.3783	0.0030
	OD 	0.3756	0.3761			0.3730		
G	ID 	0.2497	0.2502	0.0000	0.0008		0.2528	0.0030
	OD 	0.2494	0.2497			0.2482		
H	ID 250	0.2497	0.2502	0.0000	0.0008		0.2520	0.0018
	OD 	0.2494	0.2497			0.2482		
I	ID 126	0.2500	0.2505	0.0005	0.0025		0.2545	0.0050
	OD 111	0.2480	0.2495			0.2450		
J 	ID 127	0.2494	0.2497	0.0001	0.0007		0.2520	0.0030
	OD 111A	0.2490	0.2493			0.2467		
K 	ID 142	0.2479	0.2486	-0.0014	-0.0004		0.2488	-0.0002
	OD 111A	0.2490	0.2493			0.2488		









Fits and Clearances
Figure 802 (Sheet 3 of 4)



COMPONENT MAINTENANCE MANUAL

Ref Letter Fig.802	Mating Item No. IPL Fig.4	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance 		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
L	ID 	0.6080	0.6100	0.0000	0.0040	--	0.6120	0.0060
	OD 	0.6060	0.6080			0.6040	--	
 M	ID 127	0.2480	0.2488	-0.0013	-0.0002	--	0.2489	-0.0001
	OD 111	0.2490	0.2493			0.2489	--	
 N	ID 142	0.2498	0.2504	0.0005	0.0014	--	0.2520	0.0030
	OD 111	0.2490	0.2493			0.2474	--	

ALL DIMENSIONS ARE IN INCHES



-  1 NEGATIVE VALUE DENOTES INTERFERENCE FIT
-  2 BUSHING (66-25183-4)
-  3 BOLT (69-63391-1)
-  4 UNLESS OTHERWISE SPECIFIED
-  5 DIMENSION BETWEEN FLANGE FACES OF BUSHINGS IN 65C32051-1 FITTING ASSEMBLY (138)
-  6 DIMENSION ACROSS FLANGE FACES OF BUSHINGS IN 65C32050-1 BEARING PLATE ASSEMBLY (117)
-  7 FOR THE 65C25799-5 AND -7 HINGE ASSEMBLY ONLY
-  8 FOR THE 65C25799-9 HINGE ASSEMBLY ONLY

Fits and Clearances
Figure 802 (Sheet 4 of 4)


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REF IPL		NAME	TORQUE*	
FIG. NO.	ITEM NO.		POUND-INCHES	POUND-FEET
2	100, 102, 180	Nut	220-360	
2	411A, 412A, 416A, 416L	Nut	90-125	
2	413A, 414A, 417A, 417L	Nut	50-75	
2	970	Nut 	150-220	
2	975	Nut 	500-600	
4	40	Nut	15	

* REFER TO SOPM 20-50-01 FOR TORQUE VALUES OF STANDARD FASTENERS.

 **CAUTION:** DO NOT ATTEMPT TO RESET TORQUE VALVE OF NUT (975). FIRST TORQUE NUT (975) AND THEN NUT (970).

Torque Table
Figure 803



COMPONENT MAINTENANCE MANUAL

SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

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COMPONENT MAINTENANCE MANUAL

ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
.	Assembly					
.	Attaching parts for assembly					
.	.	Detail parts for assembly				
.	.	Subassembly				
.	.	Attaching parts for subassembly				
.	.	.	Detail parts for subassembly			
.	.	.	Sub-subassembly			
.	.	.	Attaching parts for subassembly			
.	.	.	.	Details parts for sub-subassembly		
						Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
- (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
- (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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ILLUSTRATED PARTS LIST

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Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

VENDOR CODES

Code	Name
55635	DAYTON-GRANGER INC 3299 SW 9TH AVENUE PO BOX 350550 FT LAUDERDALE, FLORIDA 33335 FORMERLY DAYTON-GRAINGER V1056B; GRANGER ASSOCIATES V08935 FORMERLY IN CALIFORNIA
77876	
77896	REXNORD INC BEARING OPERATION 2400 CURTIS STREET DOWNERS GROVE, ILLINOIS 60515-4005 FORMERLY SHAEFER BEARING DIV REX CHAINBELT FORMERLY REX CHAINBELT INC BEARING DIV.

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ILLUSTRATED PARTS LIST

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NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
10-60754-689		7	120	1
10-61975-2		2	245	2
		2	295	2
		2	411F	2
		2	412F	2
		2	416F	2
		2	416Q	2
		2	418F	2
		2	418Q	2
10-61975-3		2	345	2
		2	400	2
		2	413F	2
		2	414F	2
		2	417F	2
		2	417Q	2
		2	419F	2
		2	419Q	2
16335		1	40	1
173N2024-1		2	10	3
65-53708-1		5	65	1
		6	125	1
		7	65	1
65-53720-10		4	215A	1
65-53720-11		4	210	1
		4	210C	1
65-53720-12		4	215	1
		4	215C	1
65-53720-13		1	255	1
65-53720-14		1	260	1
65-53720-15		4	210B	1
		4	210D	1
65-53720-16		4	215B	1
		4	215D	1
65-53720-5		1	255B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	255C	1
65-53720-6		1	260B	1
		1	260C	1
65-53720-9		4	210A	1
65-53728-4		6	95	1
		7	135	1
65-53728-6		5	125	1
65-58720-13		1	255A	1
65-58720-14		1	260A	1
65-60535-13		1	70B	1
65-60535-14		1	70C	1
65-60535-15		1	70D	1
65-60535-16		1	70E	1
65-60535-17		1	70	1
65-60535-18		1	70A	1
65C25746-11		1	280C	1
65C25746-113		1	275H	1
65C25746-114		1	285H	1
65C25746-115		1	280H	1
65C25746-116		1	290H	1
65C25746-119SP		1	1N	RF
65C25746-12		1	290B	1
65C25746-120SP		1	5N	RF
65C25746-123SP		1	1P	RF
65C25746-124SP		1	5P	RF
65C25746-125		1	275F	1
65C25746-126		1	285F	1
65C25746-127		1	280F	1
65C25746-128		1	290F	1
65C25746-129SP		1	1Q	RF
65C25746-13		1	275C	1
65C25746-130SP		1	5Q	RF
65C25746-131SP		1	1R	RF
65C25746-132SP		1	5R	RF
65C25746-133		1	195B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	195C	1
65C25746-134		1	198B	1
		1	198C	1
65C25746-135		1	195D	1
		1	195E	1
65C25746-136		1	198D	1
		1	198E	1
65C25746-14		1	285C	1
65C25746-141SP		1	1S	RF
65C25746-142SP		1	5S	RF
65C25746-143		1	275I	1
65C25746-144		1	285I	1
65C25746-145		1	280I	1
65C25746-146		1	290I	1
65C25746-147		1	280J	1
65C25746-148		1	290J	1
65C25746-149		1	280K	1
65C25746-15		1	280D	1
65C25746-150		1	290K	1
65C25746-151		1	280L	1
65C25746-152		1	290L	1
65C25746-153SP		1	1T	RF
65C25746-154SP		1	5T	RF
65C25746-16		1	290D	1
65C25746-17		1	275E	1
65C25746-18		1	285E	1
65C25746-19SP		1	1E	RF
65C25746-1SP		1	1D	RF
65C25746-20SP		1	5E	RF
65C25746-21SP		1	1F	RF
65C25746-22SP		1	5F	RF
65C25746-23		1	280E	1
65C25746-24		1	290C	1
65C25746-25		1	275D	1
65C25746-26		1	285D	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25746-27SP		1	1	RF
65C25746-28SP		1	5	RF
65C25746-2SP		1	5D	RF
65C25746-3SP		1	1G	RF
65C25746-47		1	275	1
65C25746-48		1	285	1
65C25746-49		1	280	1
65C25746-4SP		1	5G	RF
65C25746-50		1	290	1
65C25746-51		1	195	1
		1	195A	1
65C25746-52		1	198	1
		1	198A	1
65C25746-53		1	275A	1
65C25746-54		1	285A	1
65C25746-55		1	280A	1
65C25746-56		1	290A	1
65C25746-59SP		1	1H	RF
65C25746-60SP		1	5H	RF
65C25746-61SP		1	1A	RF
65C25746-62SP		1	5A	RF
65C25746-65SP		1	1J	RF
65C25746-66SP		1	5J	RF
65C25746-67SP		1	1K	RF
65C25746-68SP		1	5K	RF
65C25746-69SP		1	1B	RF
65C25746-70SP		1	5B	RF
65C25746-71		1	275G	1
65C25746-72		1	285G	1
65C25746-73		1	280G	1
65C25746-74		1	290G	1
65C25746-75		1	275B	1
65C25746-76		1	285B	1
65C25746-77		1	280B	1
65C25746-78		1	290E	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25746-81SP		1	1L	RF
65C25746-82SP		1	5L	RF
65C25746-83SP		1	1M	RF
65C25746-84SP		1	5M	RF
65C25746-95SP		1	1C	RF
65C25746-96SP		1	5C	RF
65C25755-1		2	30A	1
65C25755-19		2	30B	1
65C25755-20		2	30	1
65C25755-24		2	30C	1
65C25756-1		1	15D	1
		2	1B	RF
65C25756-10		1	15G	1
		2	5C	RF
65C25756-11		1	15H	1
		2	1D	RF
65C25756-12		1	15J	1
		2	5D	RF
65C25756-13		1	15	1
		2	1	RF
65C25756-14		1	15A	1
		2	5	RF
65C25756-19		1	15B	1
		2	1A	RF
65C25756-2		1	15E	1
		2	5B	RF
65C25756-20		1	15C	1
		2	5A	RF
65C25756-25		1	15K	1
		2	1E	RF
65C25756-26		1	15L	1
		2	5E	RF
65C25756-27		1	15M	1
		2	1F	RF
65C25756-28		1	15N	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	5F	RF
65C25756-29		1	15P	1
		2	1G	RF
65C25756-30		1	15Q	1
		2	5G	RF
65C25756-31		1	15R	1
		2	1H	RF
65C25756-32		1	15S	1
		2	5H	RF
65C25756-33		1	15T	1
		2	1J	RF
65C25756-34		1	15U	1
		2	5J	RF
65C25756-41		1	15V	1
		2	1K	RF
65C25756-42		1	15W	1
		2	5K	RF
65C25756-9		1	15F	1
		2	1C	RF
65C25759-1		2	705	1
65C25759-2		2	750	1
65C25761-1		2	768	1
65C25761-2		2	795	1
65C25763-1		2	820	1
65C25763-2		2	838	1
65C25765-4		2	876A	1
65C25765-6		2	858A	1
65C25767-1		2	896	1
65C25767-2		2	922	1
65C25769-1		2	540A	1
65C25769-10		2	580	1
65C25769-11		2	570B	1
65C25769-12		2	580B	1
65C25769-13		2	540B	1
65C25769-14		2	550B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25769-15		2	540C	1
65C25769-16		2	550C	1
65C25769-19		2	570C	1
65C25769-2		2	550A	1
65C25769-20		2	580C	1
65C25769-3		2	570A	1
65C25769-4		2	580A	1
65C25769-7		2	540	1
65C25769-8		2	550	1
65C25769-9		2	570	1
65C25771-1		1	20B	1
		3	1A	RF
65C25771-2		1	20C	1
		3	5A	RF
65C25771-3		3	20	1
		3	110	3
65C25771-5		1	20	1
		3	1	RF
65C25771-6		1	20A	1
		3	5	RF
65C25772-1		3	10	1
65C25772-2		3	15	1
65C25773-11		3	30	1
65C25773-12		3	120A	3
		3	120B	1
65C25773-13		3	105	1
65C25773-14		3	190	1
65C25773-15		3	120	3
65C25773-16		3	30A	1
65C25783-11		1	25D	1
		5	1B	RF
65C25783-12		1	25E	1
		5	5B	RF
65C25783-15		1	25	1
		5	1	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25783-16		1	25A	1
		5	5	RF
65C25783-19		5	140A	1
65C25783-20		5	95A	1
65C25783-21		5	110A	1
65C25783-3		1	25B	1
		5	1A	RF
65C25783-30		5	110B	1
65C25783-31		1	25F	1
		5	1C	RF
65C25783-32		1	25G	1
		5	5C	RF
65C25783-35		5	95B	1
65C25783-36		5	110C	1
65C25783-37		5	95C	1
65C25783-38		5	110D	1
65C25783-39		5	140B	1
65C25783-4		1	25C	1
		5	5A	RF
65C25783-8		5	85	2
		5	92	2
65C25783-9		5	90	1
		5	92A	2
65C25784-11		1	30D	1
		6	1B	RF
65C25784-12		1	30E	1
		6	5B	RF
65C25784-15		1	30	1
		6	1	RF
65C25784-16		1	30A	1
		6	5	RF
65C25784-20		6	65A	1
65C25784-21		6	70A	1
65C25784-29		1	30F	1
		6	1C	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25784-3		1	30B	1
		6	1A	RF
65C25784-30		1	30G	1
		6	5C	RF
65C25784-31		6	70B	1
65C25784-32		6	65B	1
65C25784-33		6	70C	1
65C25784-34		6	65C	1
		1	30C	1
65C25784-4		6	5A	RF
		6	80	1
65C25784-6		6	75	1
65C25784-7		6	135	1
65C25784-8		6	130	2
65C25784-9		6	140	2
		1	35D	1
65C25785-11		7	1B	RF
		1	35E	1
65C25785-12		7	5B	RF
		1	35	1
65C25785-15		7	1	RF
		1	35A	1
65C25785-16		7	5	RF
		7	90A	1
65C25785-19		7	90C	1
		7	105A	1
65C25785-20		7	105C	1
		7	90B	1
65C25785-26		7	90D	1
		7	105B	1
65C25785-27		7	105B	1
65C25785-28		1	35G	1
		7	5C	RF
65C25785-29		1	35F	1
		7	1C	RF
65C25785-3		1	35B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		7	1A	RF
65C25785-32		7	105D	1
65C25785-33		7	90E	1
65C25785-34		7	105E	1
65C25785-4		1	35C	1
		7	5A	RF
65C25785-8		7	87A	3
65C25785-9		7	85	2
		7	87	3
65C25796-1		4	1A	RF
65C25796-10		4	87A	1
		4	87B	1
65C25796-11		4	86B	1
65C25796-12		4	95E	1
65C25796-13		4	86C	1
65C25796-14		4	95F	1
65C25796-15		4	86D	1
65C25796-16		4	95G	1
65C25796-17		4	86E	1
65C25796-18		4	95H	1
65C25796-19		4	1	RF
65C25796-19SP		1	10	RF
65C25796-1SP		1	10B	RF
65C25796-2		4	5A	RF
65C25796-20		4	5	RF
65C25796-20SP		1	10A	RF
65C25796-21		4	88A	1
65C25796-22		4	88B	1
65C25796-23		4	36	1
		4	89B	1
65C25796-27		4	1B	RF
65C25796-27SP		1	10D	RF
65C25796-28		4	5B	RF
65C25796-28SP		1	10E	RF
65C25796-29		4	1C	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C25796-29SP		1	10F	RF
65C25796-2SP		1	10C	RF
65C25796-30		4	5C	RF
65C25796-30SP		1	10G	RF
65C25796-9		4	86	1
		4	86A	1
65C25797-1		4	225A	1
65C25797-15		4	225	1
65C25797-16		4	230	1
65C25797-17		4	225B	1
65C25797-18		4	230B	1
65C25797-2		4	230A	1
65C25798-10		4	177A	1
65C25798-11		4	165D	4
65C25798-12		4	165E	8
65C25798-13		4	177B	1
65C25798-14		4	165F	2
65C25798-15		4	177C	1
65C25798-4		4	165	2
		4	165A	2
65C25798-5		4	175	1
65C25798-6		4	165B	2
65C25798-7		4	176	1
65C25798-8		4	165C	2
65C25798-9		4	177	1
65C25799-1		4	106	2
65C25799-3		4	105	2
65C25799-5		4	107	2
65C25799-7		4	107A	2
65C25799-9		4	107B	2
65C25809-1		1	140	1
65C25809-2		1	140B	1
65C25809-3		1	140A	1
65C25809-4		1	140C	1
65C26384-1		4	90A	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C26384-11		4	90	1
		4	90J	1
65C26384-12		4	95	1
		4	95J	1
65C26384-13		4	90B	1
65C26384-14		4	95B	1
65C26384-15		4	90K	1
65C26384-16		4	95K	1
65C26384-19		4	90L	1
65C26384-2		4	95A	1
65C26384-20		4	95L	1
65C26384-23		4	90M	1
65C26384-24		4	95M	1
65C26384-25		4	90N	1
65C26384-26		4	95N	1
65C26384-27		4	90P	1
65C26384-28		4	95Q	1
65C26384-3		4	220	2
65C26384-4		4	150	8
		4	150A	8
65C26384-5		4	90C	1
65C26384-6		4	95C	1
65C26384-7		4	90D	1
65C26384-8		4	95D	1
65C26385-1		4	116	1
65C26385-2		4	130	1
65C26385-3		4	115	1
65C26386-1		4	137	1
65C26386-2		4	146	1
65C26386-4		4	136	1
65C26386-6		4	145	1
65C26386-8		4	135	1
65C26388-1		1	180	2
65C26388-2		1	175	2
65C26388-23		1	180A	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	180C	1
65C26388-24		1	180B	1
		1	180D	1
65C26388-25		1	175A	1
		1	175C	1
65C26388-26		1	175B	1
		1	175D	1
65C26388-27		1	170A	1
		1	170C	1
65C26388-28		1	170B	1
		1	170D	1
65C26388-29		1	165A	1
		1	165C	1
65C26388-3		1	170	2
65C26388-30		1	165B	1
		1	165D	1
65C26388-31		1	160A	1
		1	160C	1
65C26388-32		1	160B	1
		1	160D	1
65C26388-33		1	155A	1
		1	155C	1
65C26388-34		1	155B	1
		1	155D	1
65C26388-35		1	145D	1
		1	145F	1
65C26388-36		1	145E	1
		1	145G	1
65C26388-4		1	165	2
65C26388-5		1	160	2
65C26388-6		1	155	2
65C26388-7		1	145	1
		1	145B	1
65C26388-8		1	145A	1
		1	145C	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C26390-1		1	265	1
65C26390-2		1	265B	1
65C26390-3		1	265A	1
65C26390-4		1	265C	1
65C26393-3		5	50	1
		6	55	1
		7	50	1
65C26393-5		5	30	1
		6	45	5
		7	30	1
65C26393-6		5	35	1
		6	50	1
		7	35	1
65C26396-1		1	120	1
65C26396-2		1	120A	1
65C26481-1		2	210	1
		2	260	1
65C26481-2		2	255	1
		2	305	1
65C26482-1		2	310	1
		2	360	1
65C26482-2		2	355	1
		2	410	1
65C26528-1		2	515	1
65C26528-2		2	520	1
65C26529-1		2	130	1
		2	130A	1
65C26529-4		2	130B	1
65C26530-1		2	200	1
65C26531-1		2	205	1
65C27274-1		5	65A	1
		6	125A	1
65C27280-1		1	120B	1
65C27280-2		1	120C	1
65C27281-1		1	140D	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C27282-1		1	265D	1
65C27282-2		1	265E	1
65C27284-8		6	140A	2
65C27301-1		2	40A	1
65C27301-4		2	40	1
		2	40B	1
65C27301-5		2	40C	1
		2	40D	1
65C27302-1		2	120	1
65C27303-1		2	125	1
65C27351-1		2	585A	1
65C27351-2		2	645A	1
65C27351-3		2	630A	1
		2	690A	1
65C27351-4		2	640	1
		2	700	1
65C27351-5		2	585	1
65C27351-6		2	645	1
65C27351-7		2	630	1
		2	690	1
65C27351-8		2	635	1
		2	695	1
65C27351-9		2	645B	1
		2	645C	1
65C27376-1		2	415B	1
65C27376-10		2	420E	1
65C27376-11		2	415F	1
65C27376-12		2	420F	1
65C27376-2		2	420B	1
65C27376-3		2	415	1
		2	415A	1
65C27376-4		2	420	1
		2	420A	1
65C27376-5		2	415C	1
65C27376-6		2	420C	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C27376-7		2	415D	1
65C27376-8		2	420D	1
65C27376-9		2	415E	1
65C27460-1		6	105	1
65C27460-2		6	100	1
65C27460-5		6	110	1
65C27564-13		1	100	1
		1	100A	1
65C27564-14		1	105	1
		1	105A	1
65C27564-15		1	110	1
		1	110A	1
65C27564-16		1	115	1
		1	115A	1
65C27564-25		1	100B	1
		1	100C	1
65C27564-26		1	105B	1
		1	105C	1
65C27564-27		1	110B	1
		1	110C	1
65C27564-28		1	115B	1
		1	115C	1
65C27589-1		1	140E	1
65C28035-1		1	230	1
65C28035-2		1	230A	1
65C28035-3		1	235	1
65C28035-4		1	235A	1
65C28142-1		4	235	1
65C28142-2		4	240	1
65C28142-3		4	255	1
65C28142-4		4	260	1
65C28166-1		1	315	1
65C28166-2		1	320	1
65C28166-3		1	335	1
65C28166-5		1	325	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C28166-6		1	330	1
65C29213-1		1	340	1
65C31307-10		2	922D	1
65C31307-11		2	896C	1
65C31307-12		2	922C	1
65C31307-4		2	896A	1
65C31307-6		2	922A	1
65C31307-7		2	896E	1
65C31307-8		2	922E	1
65C31307-9		2	896D	1
65C31311-2		1	120D	1
65C31488-1		1	295	1
65C31488-2		1	305	1
65C31488-3		1	310	1
65C31488-4		1	300	1
65C31695-1		1	345	1
65C32050-1		4	117	1
65C32050-2		4	131	1
65C32050-3		4	117A	1
65C32051-1		4	138	1
65C32051-2		4	147	1
65C32051-4		4	138A	1
65C32051-5		4	147A	1
65C32051-6		4	138B	1
65C34066-1		4	235B	1
		4	235C	1
65C34066-2		4	240B	1
		4	240C	1
65C34066-3		4	255B	1
65C34066-4		4	260B	1
65C34548-1		2	40E	1
		2	40G	1
65C34548-2		2	128	1
65C34549-1		2	130C	1
		2	130D	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C34549-2		2	207	1
65C36621-1		5	110E	1
65C36621-2		5	95D	1
65C36621-3		6	70D	1
65C36621-4		6	65D	1
65C36621-5		7	105F	1
65C36621-6		7	90F	1
65C36621-7		5	140C	1
66-25183-3		2	675	1
66-25183-6		2	675A	1
69-40240-3		4	255A	1
69-40240-4		4	260A	1
69-40240-503		4	235A	1
69-40240-504		4	240A	1
69-44494-4		2	680B	1
69-44494-502		2	680	1
		2	680A	1
69-44682-2		2	955	1
69-44683-2		2	950	1
69-50565-3		2	615	1
69-50565-4		2	620	1
69-50565-5		2	613	1
69-50565-6		2	614	1
69-53211-2		1	75	22
69-53223-2		1	80	6
69-56340-2		4	250A	2
69-56340-501		4	250	2
69-59227-3		4	245	2
69-59227-4		4	245A	2
69-63391-5		4	10	4
		4	10F	1
		4	89	4
69-63391-6		4	10G	3
69-63391-7		4	10H	1
69-63391-8		4	10I	3

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-63523-1		2	530	1
69-63523-2		2	530A	1
69-63523-3		2	530B	1
69-72039-37		4	125	2
		4	142	2
69-72039-38		3	185A	1
69-72039-39		4	127	2
69-72899-1		5	60	1
		6	60	1
		7	60	1
69-73579-1		3	25	2
		3	115	6
69-73701-1		2	10A	3
69-74355-1		1	50	1
69-74355-2		1	50A	1
69-74582-1		4	110	1
69-74582-2		4	111A	1
69-74582-3		4	111B	1
69-76409-1		2	410R	1
		2	411R	1
69-76409-2		2	411J	1
		2	412J	1
69-76409-5		2	410S	1
		2	411S	1
		2	416	1
		2	416K	1
69-76409-6		2	411K	1
		2	412K	1
		2	416J	1
		2	416T	1
69-76409-7		2	418	1
		2	418K	1
69-76409-8		2	418J	1
		2	418T	1
69-76419-1		2	412R	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	413R	1
69-76419-2		2	413J	1
		2	414J	1
69-76419-5		2	412S	1
		2	413S	1
		2	417	1
		2	417K	1
69-76419-6		2	413K	1
		2	414K	1
		2	417J	1
		2	417T	1
69-76419-7		2	419	1
		2	419K	1
69-76419-8		2	419J	1
		2	419T	1
69-76940-2		2	206	1
69-76941-2		2	127	1
69-77838-1		2	121	1
69-77838-2		2	201	1
69-77839-1		2	122	1
69-77839-2		2	202	1
69-78806-1		1	166	1
69-78806-2		1	162	1
69-78806-3		1	161	1
69-78806-4		1	157	1
69-78806-5		1	156	1
69-78806-6		1	146	1
AN960-8		4	161D	16
AN960-8L		4	161E	16
AN960C10L		1	90	AR
AN960C416		4	38	AR
AN960C416L		4	38A	AR
AN960C8		4	161A	8
AN960C8L		4	161B	8
AN960JD416		2	507	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY		
AN960KD416		2	330A	2		
		2	380	2		
		2	413D	2		
		2	414D	2		
		2	417D	2		
		2	417O	2		
		2	419D	2		
		2	419O	2		
		2	880A	RF		
		2	926	RF		
		AN960KD516		2	230	2
				2	280	2
				2	411D	2
2	412D			2		
2	416D			2		
2	416O			2		
2	418D			2		
2	418O			2		
2	812			RF		
2	842			RF		
AN960KD516L				1	130	8
AN960PD10L				5	75	16
				7	75	10
		7	75A	10		
AN960XC416L		4	112	2		
BAC27DWG26		4	100	1		
BAC27DWG27		1	240	1		
BACB10CH50		2	250	1		
		2	300	1		
		2	350	1		
		2	405	1		
		2	411G	1		
		2	412G	1		
		2	413G	1		
		2	414G	1		

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	416G	1
		2	416R	1
		2	417G	1
		2	417R	1
		2	418G	1
		2	418R	1
		2	419G	1
		2	419R	1
BACB10CH65		2	185	1
BACB10L22A		4	141	1
BACB28AK04-012		4	30	1
		4	30A	1
BACB28AK04-033		4	21	3
BACB28AK04-037		4	25A	1
BACB28AK04-040		4	20	6
		4	89A	6
BACB28AK04-044		4	20A	6
BACB28AK04-045		4	25	1
		4	26	4
BACB28AK04-187		2	325	2
		2	413C	2
		2	417C	2
		2	419C	2
		2	882	RF
BACB28AK04-190		2	375	2
		2	414C	2
		2	417N	2
		2	419N	2
		2	928	RF
BACB28AK05-042		2	145A	2
BACB28AK05-046		2	55A	1
BACB28AK05-125		2	55	2
BACB28AK05-130		2	145	2
		2	759A	RF
BACB28AK05-168		2	225	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	411C	2
		2	416C	2
		2	418C	2
		2	814	RF
BACB28AK05-187		2	275	2
		2	412C	2
		2	416N	2
		2	844	RF
BACB28AK05-190		2	418N	2
BACB28AM04B010A		4	126	2
BACB28AM06B013A		3	95C	1
		3	100	1
		3	185	2
BACB28AM06B015A		2	920	2
BACB28AM06B018A		2	874	2
BACB28AM07B018A		2	834	2
BACB28AM07B019		2	115	2
		2	195	2
BACB28AM07B020A		2	792	2
BACB28AM07B025A		2	745	2
BACB28AM07B029A		2	116	2
		2	560	2
BACB28AM07B031A		2	196	2
BACB28AP04P011		3	95A	1
BACB28AP04P013		3	95	1
BACB28AP04P015		2	918	2
BACB28AP04P018		2	872	2
BACB28AP05A023		2	740	2
BACB28AP05F029		2	560A	2
BACB28AP05P018		2	836	2
BACB28AP05P019		2	110	2
		2	190	2
BACB28AP05P020		2	789	2
BACB28AP05P029		2	117	2
BACB28AP05P030		2	555	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB28AP05P031		2	197	2
BACB28AP05P098		2	118	2
BACB28AP05P108		2	198	2
BACB28AT07B028C		2	555A	2
BACB28U4B022		4	140	1
BACB30LE5-13		1	125	4
BACB30LE6U24		2	80	2
		2	165	2
BACB30LH3-2		1	190	70
BACB30LH3-5		1	270	58
BACB30LH3-6		1	205	1
BACB30LJ3-5		1	200	9
BACB30LJ4DSU21		4	10A	4
BACB30LL3-3		4	50A	2
BACB30LL3-4		4	55A	2
BACB30LM3-10		1	85E	AR
BACB30LM3-11		1	85F	AR
BACB30LM3-3		1	85	AR
BACB30LM3-5		1	85A	AR
BACB30LM3-6		1	85B	AR
BACB30LM3-7		1	85C	AR
BACB30LM3-8		1	85D	AR
BACB30LU3-4		7	70	10
BACB30LU3-5		7	70A	10
BACB30LU3-5X		5	70	16
BACB30MY6K4		2	898	3
BACB30MY8C9		2	505	2
BACB30MY8K3		2	430	1
BACB30MY8K4		2	435	1
BACB30MY8K5		2	440	1
BACB30MY8K6		2	445	7
BACB30NM3C5		3	35	2
		3	125	2
BACB30NM3K12		2	596	2
		2	656	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB30NM3K3		2	15	6
BACB30NM3K6		2	601	3
		2	661	3
BACB30NM3K8		2	591	2
		2	651	2
		4	190	1
BACB30NM5K32		2	45	2
		2	135	2
		2	765A	RF
BACB30NM6K24		2	82	2
BACB30NN3K3		4	180	4
		4	180A	3
BACB30NN3K6		4	185	2
BACB30NN4K8		2	83	2
BACB30NN4K9		2	166	2
BACB30NR4K36		2	315	2
		2	365	2
		2	413A	2
		2	414A	2
		2	417A	2
		2	417L	2
		2	419A	2
		2	419L	2
		2	886	RF
		2	932	RF
BACB30NR5K33		2	215	2
		2	411A	2
		2	416A	2
		2	418A	2
		2	818	RF
BACB30NR5K36		2	265	2
		2	412A	2
		2	416L	2
		2	418L	2
		2	848	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB30NT3K3		4	180F	1
BACB30NT3K8		4	190A	1
BACB30NW6K5		2	866	8
		2	904	7
BACB30NW6K7		2	910	1
BACB30NW8K6		2	780	6
		2	828	8
BACB30NX10K5		2	771	4
BACB30NX6K11		2	595	2
		2	655	2
BACB30NX6K4		2	860	3
BACB30NX6K6		2	600	3
		2	660	3
BACB30NX6K7		2	590	2
		2	650	2
BACB30NX8K12		2	460	1
BACB30NX8K6		2	822	4
BACB30NX8K8		2	710	4
BACB30NY8K9		2	455	1
BACB30NZ8K8		2	450	6
		2	725	6
BACB30VF08K3		4	155	16
BACB30VF3R3		4	50	2
BACB30VF3R4		4	55	2
BACB30VF4K8		2	425	1
BACB30VT8K10		2	505A	2
BACC30AB6C		2	902	3
BACC30AB8C		2	480	10
BACC30X10		2	777	4
BACC30X6		2	610	7
		2	670	7
		2	864	3
BACC30X8		2	485	1
		2	720	4
		2	826	4

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACF33C101-0750		1	210	10
BACJ40A20-5		4	80	1
BACJ40A20-6		4	81	1
BACJ40AB23-8		2	495	1
BACJ40AB26-5		2	916	1
BACN10BY510		2	975	1
BACN10HR4CD		2	506	2
BACN10HR5CD		1	135	4
BACN10HR6CS		2	100	2
		2	180	2
BACN10JC3		1	95	AR
		4	195	6
		5	80	16
		7	80	10
		7	80A	10
BACN10JC3C		4	75	4
BACN10JC3CD		1	225	10
		2	832	8
		2	870	8
		2	908	7
		2	914	1
BACN10JC4CD		2	335	2
		2	385	2
		2	413E	2
		2	414E	2
		2	417E	2
		2	417P	2
		2	419E	2
		2	419P	2
		2	490	8
		2	735	6
		2	786	6
		2	878	RF
		2	924	RF
BACN10JC5CD		2	235	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	285	2
		2	411E	2
		2	412E	2
		2	416E	2
		2	416P	2
		2	418E	2
		2	418P	2
		2	810	RF
		2	840	RF
BACN10JD4ASU		4	40	4
BACN10JP08ACM		4	178	4
BACN10JP08BCD		4	114A	4
BACN10JR3CFD		1	350	6
BACN10JR3CFDR		2	27	6
BACN10JR3FDR		3	50	2
		3	140	2
BACN10YR08CD		4	162	8
		4	162A	16
		4	162B	16
BACN10YR3CD		4	195A	6
BACN10YR4CD		2	506A	2
BACN11N4CS		4	40A	AR
BACNJP08BCM		4	114	4
BACP18BC02A06P		4	42A	AR
BACP18T4-21		4	111	1
BACR15BA5A6		1	45	4
BACR15BA6D		2	625A	2
		2	685A	2
BACR15DR3A2		2	28	12
		3	45	4
		3	135	4
BACR15DR3AC2		4	118	8
		4	179	8
BACR15DR3PA2		1	250	6
BACR15GA5-11		1	55	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACS12ER08K05		1	60	2
BACS40V010C030		4	150B	8
		4	150C	8
		4	150D	8
BACS40V011D040		4	220A	2
BACW10B3NDP		2	607	2
BACW10BN3UP		4	205	4
BACW10BN4AC		2	320	2
		2	370	2
		2	413B	2
		2	414B	2
		2	417B	2
		2	417M	2
		2	419B	2
		2	419M	2
		2	884	RF
		2	930	RF
BACW10BN5AC		2	220	2
		2	270	2
		2	411B	2
		2	412B	2
		2	416B	2
		2	416M	2
		2	418B	2
		2	418M	2
		2	816	RF
		2	846	RF
BACW10BN6SC		2	85	2
		2	170	2
BACW10BN6UP		2	90	2
		2	175	2
BACW10BP3APU		4	65	4
		7	100	3
		7	115	4
BACW10BP3CD		2	608	7

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	668	7
BACW10BP3CT		2	20	4
BACW10BP3CTU		2	25	2
		3	40	2
		3	130	2
BACW10BP3NAPU		2	605	2
		2	606	5
		2	665	2
		2	666	5
		2	862	3
		2	900	3
		4	70	4
BACW10BP3NDP		2	667	2
		2	830	8
		2	868	8
		2	906	7
		2	912	1
		4	205A	4
BACW10BP4CT		4	15	4
BACW10BP4CTU		4	15A	4
BACW10BP4NAPU		2	465	10
		2	715	4
		2	824	4
BACW10BP4NDP		2	91	2
		2	176	2
		2	470	6
		2	730	6
		2	783	6
BACW10BP4PT		4	35	4
		4	35A	3
BACW10BP4PTU		4	21F	1
		4	35B	3
		4	36A	1
BACW10BP5APU		2	774	4
BACW10BP5CTU		2	50	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	140	2
		2	762	RF
BACW10BP5DP		2	60	2
		2	150	2
		2	756	RF
BACW10CA3CCU		1	220	13
BACW10CA3CVU		1	215	13
BACW10DM5P		4	160	16
BACW10DM6		4	200	3
BACW10DM6P		4	60	4
		4	200A	3
BACW10P43AL		5	135A	3
		6	120A	2
		7	130A	3
BACW10P61S		2	475	2
BACW10UC10P		4	200B	3
DAS10-26A1-4		2	525	1
DAS10-26A1-501		2	525A	1
DAS4-14A1-508		4	170A	1
DAS6-23B		2	105	1
MS15001-1		2	70	2
		2	160	2
		2	240	1
		2	290	1
		2	340	1
		2	395	1
		2	411H	1
		2	412H	1
		2	413H	1
		2	414H	1
		2	416H	1
		2	416S	1
		2	417H	1
		2	417S	1
		2	418H	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	418S	1
		2	419H	1
		2	419S	1
MS20253-4-2122		5	20	1
		6	35	1
		7	20	1
MS20392-2C11		5	10	1
		6	30	1
		7	10	1
MS20392-2C7		5	25	1
		6	40	1
		7	25	1
MS24665-132		6	20	1
MS24665-153		4	42	4
		4	113	1
MS27253C2		1	245	1
MS27253F2		1	245A	1
NAS1149C0432R		4	38B	AR
NAS1149FN832P		4	161	8
		4	161C	16
NAS1804-5		2	753	RF
NAS1804-5L		2	65	2
		2	155	2
NAS1805-3L		2	611	7
		2	671	7
NAS1805-4N		2	103	2
		2	181	2
NAS1805-6		2	102	2
NAS1836C08-08M		1	65	2
NAS516-1		2	500	1
NAS516-1A		4	132	1
		4	171	1
NAS587-29		2	960	1
NAS620-10L		5	15	1
		6	25	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		7	15	1
NAS6603-2		5	145A	4
NAS6603-3		5	130	3
NAS6603-4		5	115	4
		5	145	4
		6	115	2
		7	125	3
		7	140	2
NAS6603-5		5	100A	4
		5	115A	4
		6	85A	8
		7	95A	3
		7	110A	7
NAS6603-6		5	40	1
		6	10	1
		7	40	1
NAS6604-6		5	45	5
		5	55	6
		6	15	11
		7	45	5
		7	55	6
NAS679A7		2	970	1
NAS76A10-11		2	965	1
SA4-14A		4	120	1
		4	170	1

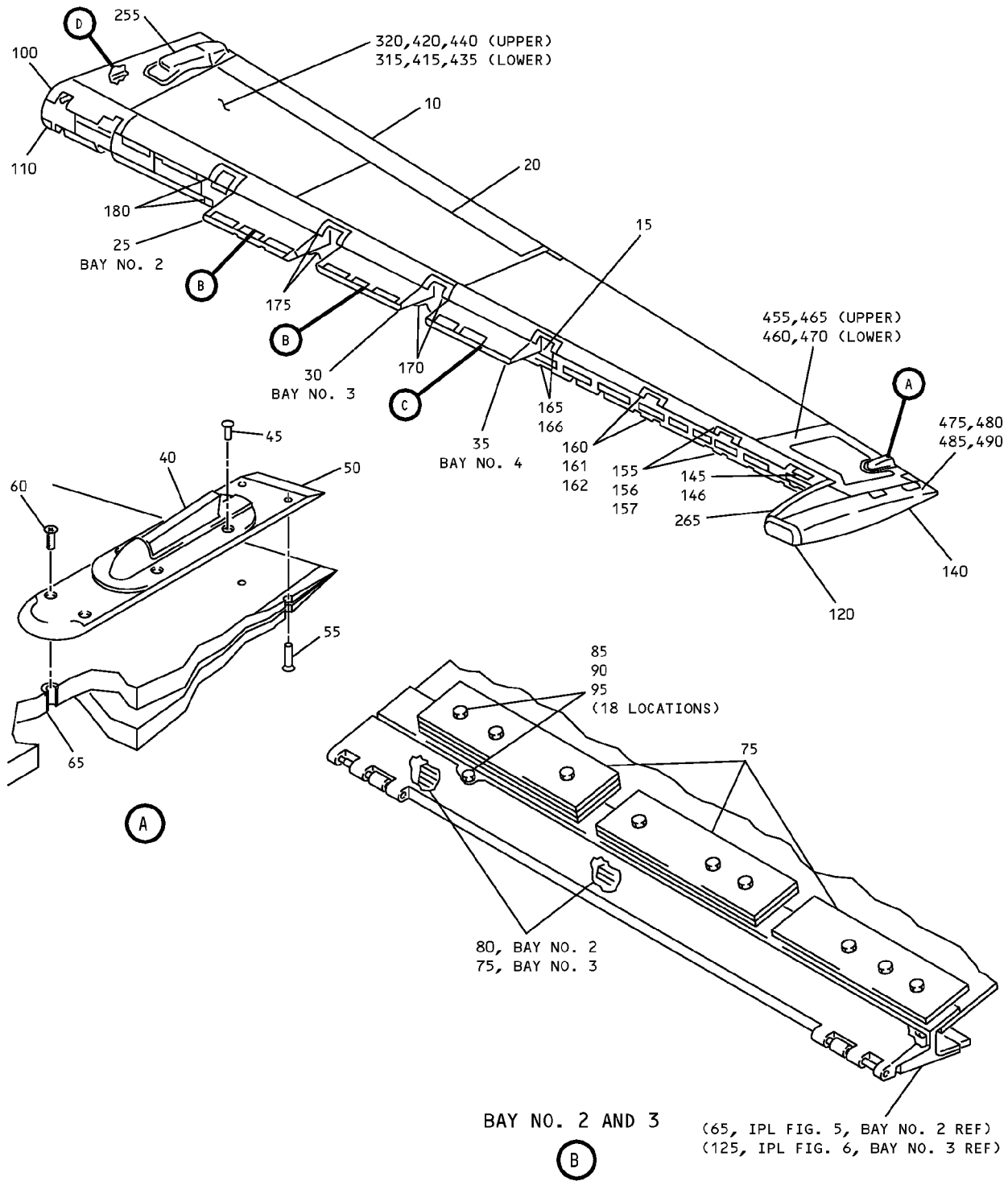
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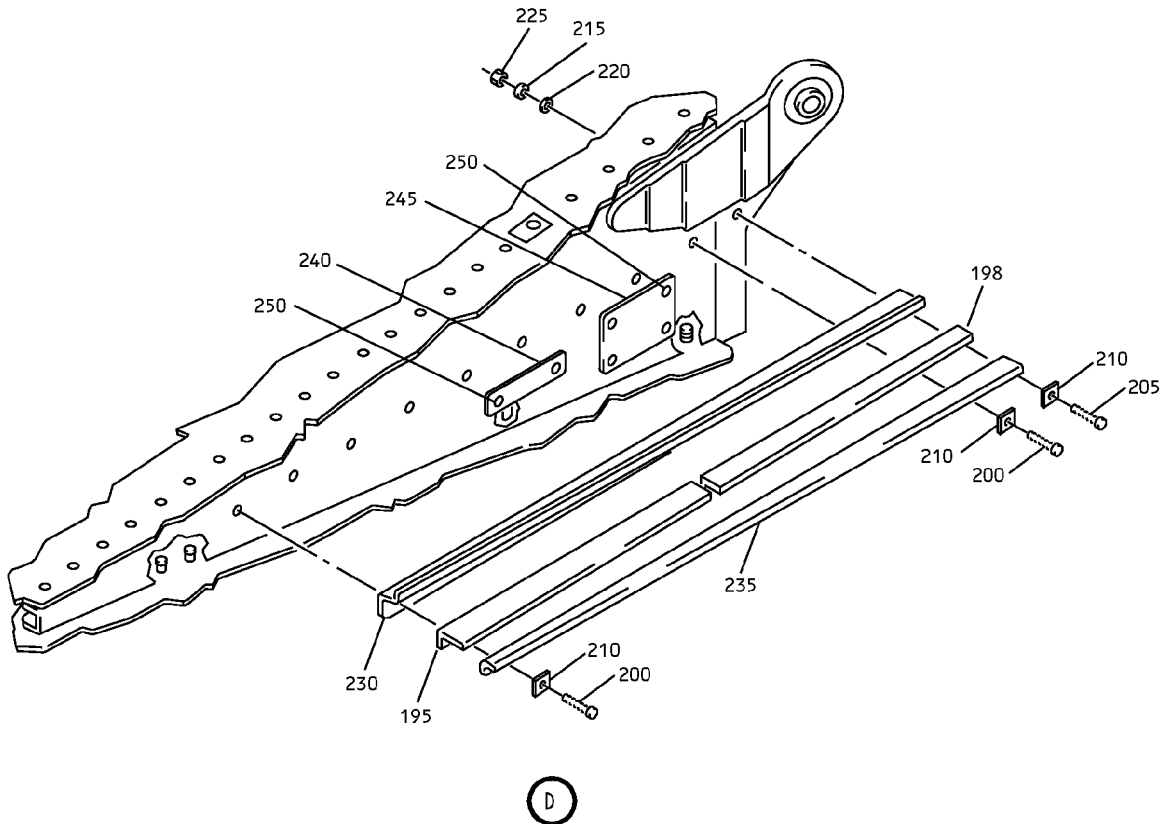
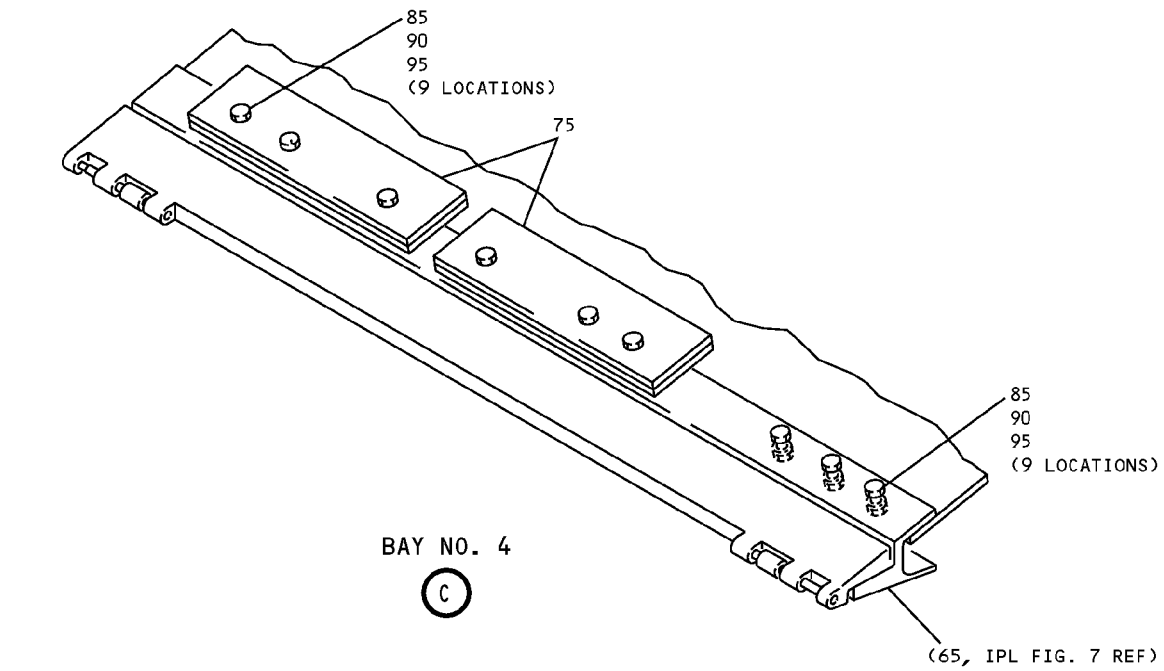
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Elevator and Tab Assembly
IPL Figure 1 (Sheet 1 of 2)

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Elevator and Tab Assembly
IPL Figure 1 (Sheet 2 of 2)



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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1	65C25746-27SP									A	RF
-1A	65C25746-61SP									C	RF
-1B	65C25746-69SP									E	RF
-1C	65C25746-95SP									G	RF
-1D	65C25746-1SP									J	RF
-1E	65C25746-19SP									L	RF
-1F	65C25746-21SP									N	RF
-1G	65C25746-3SP									Q	RF
-1H	65C25746-59SP									S	RF
-1J	65C25746-65SP									U	RF
-1K	65C25746-67SP									W	RF
-1L	65C25746-81SP									Y	RF
-1M	65C25746-83SP									BA	RF
-1N	65C25746-119SP									DA	RF
-1P	65C25746-123SP									FA	RF
-1Q	65C25746-129SP									HA	RF
-1R	65C25746-131SP									JA	RF
-1S	65C25746-141SP									LA	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1T	65C25746-153SP									NA	RF
-5	65C25746-28SP									B	RF
-5A	65C25746-62SP									D	RF
-5B	65C25746-70SP									F	RF
-5C	65C25746-96SP									H	RF
-5D	65C25746-2SP									K	RF
-5E	65C25746-20SP									M	RF
-5F	65C25746-22SP									P	RF
-5G	65C25746-4SP									R	RF
-5H	65C25746-60SP									T	RF
-5J	65C25746-66SP									V	RF
-5K	65C25746-68SP									X	RF
-5L	65C25746-82SP									Z	RF
-5M	65C25746-84SP									CA	RF
-5N	65C25746-120SP									EA	RF
-5P	65C25746-124SP									GA	RF
-5Q	65C25746-130SP									IA	RF
-5R	65C25746-132SP									KA	RF
-5S	65C25746-142SP									MA	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-5T	65C25746-154SP									PA	RF
10	65C25796-19SP										RF
-10A	65C25796-20SP										RF
-10B	65C25796-1SP										RF
-10C	65C25796-2SP										RF
-10D	65C25796-27SP										RF
-10E	65C25796-28SP										RF
-10F	65C25796-29SP										RF
-10G	65C25796-30SP										RF
15	65C25756-13									A, C, E, S, W	1
-15A	65C25756-14									B, D, F, T, X	1
-15B	65C25756-19									G	1
-15C	65C25756-20									H	1
-15D	65C25756-1									J, L, Q	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-15E	65C25756-2		.	SPAR	INSTL-FRONT					K, M, R	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15F	65C25756-9		.	SPAR	INSTL-FRONT					N	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15G	65C25756-10		.	SPAR	INSTL-FRONT					P	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15H	65C25756-11		.	SPAR	INSTL-FRONT					J, L	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15J	65C25756-12		.	SPAR	INSTL-FRONT					K, M	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15K	65C25756-25		.	SPAR	INSTL-FRONT					Y, DA	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15L	65C25756-26		.	SPAR	INSTL-FRONT					Z, EA	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15M	65C25756-27		.	SPAR	INSTL-FRONT					BA	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15N	65C25756-28		.	SPAR	INSTL-FRONT					CA	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15P	65C25756-29		.	SPAR	INSTL-FRONT					FA	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15Q	65C25756-30		.	SPAR	INSTL-FRONT					GA	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							
-15R	65C25756-31		.	SPAR	INSTL-FRONT					U, HA	1
				(LH)							
				(FOR DETAILS SEE FIG. 2)							
-15S	65C25756-32		.	SPAR	INSTL-FRONT					V, IA	1
				(RH)							
				(FOR DETAILS SEE FIG. 2)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-15T	65C25756-33		. SPAR INSTL-FRONT (LH) (FOR DETAILS SEE FIG. 2)	JA, LA	1
-15U	65C25756-34		. SPAR INSTL-FRONT (RH) (FOR DETAILS SEE FIG. 2)	KA, MA	1
-15V	65C25756-41		. SPAR INSTL-FRONT (LH) (FOR DETAILS SEE FIG. 2)	LA, NA	1
-15W	65C25756-42		. SPAR INSTL-FRONT (RH) (FOR DETAILS SEE FIG. 2)	MA, PA	1
20	65C25771-5		. SPAR INSTL-REAR (LH) (FOR DETAILS SEE FIG. 3)	A, C, E, G, S, U, W, Y, BA, DA, FA, HA, JA, NA	1
-20A	65C25771-6		. SPAR INSTL-REAR (RH) (FOR DETAILS SEE FIG. 3)	B, D, F, H, T, V, X, Z, CA, EA, GA, IA, KA, PA	1
-20B	65C25771-1		. SPAR INSTL-REAR (LH) (FOR DETAILS SEE FIG. 3)	J, L, N, Q	1
-20C	65C25771-2		. SPAR INSTL-REAR (RH) (FOR DETAILS SEE FIG. 3)	K, M, P, R	1
25	65C25783-15		. PANEL INST-BAY NO. 2, BALANCE (LH) (FOR DETAILS SEE FIG. 5)	A, C, E, G, S, W	1
-25A	65C25783-16		. PANEL INST-BAY NO. 2, BALANCE (RH) (FOR DETAILS SEE FIG. 5)	B, D, F, H, T, X	1
-25B	65C25783-3		. PANEL INST-BAY NO. 2, BALANCE (LH) (FOR DETAILS SEE FIG. 5)	J, L, N	1
-25C	65C25783-4		. PANEL INST-BAY NO. 2, BALANCE (RH) (FOR DETAILS SEE FIG. 5)	K, M, P	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-25D	65C25783-11		. PANEL INST-BAY NO. 2, BALANCE (RH) (FOR DETAILS SEE FIG. 5)	Q	1
-25E	65C25783-12		. PANEL INST-BAY NO. 2, BALANCE (RH) (FOR DETAILS SEE FIG. 5)	R	1
-25F	65C25783-31		. PANEL INST-BAY NO. 2, BALANCE (LH) (FOR DETAILS SEE FIG. 5)	U, Y, BA, DA, FA, HA, JA, NA	1
-25G	65C25783-32		. PANEL INST-BAY NO. 2, BALANCE (RH) (FOR DETAILS SEE FIG. 5)	V, Z, CA, EA, GA, IA, KA, PA	1
30	65C25784-15		. PANEL INST-BAY NO. 3, BALANCE (LH) (FOR DETAILS SEE FIG. 6)	A, C, E, G, S, W	1
-30A	65C25784-16		. PANEL INST-BAY NO. 3, BALANCE (RH) (FOR DETAILS SEE FIG. 6)	B, D, F, H, T, X	1
30B	65C25784-3		. PANEL INST-BAY NO. 3, BALANCE (LH) (FOR DETAILS SEE FIG. 6)	J, L, N	1
-30C	65C25784-4		. PANEL INST-BAY NO. 3, BALANCE (RH) (FOR DETAILS SEE FIG. 6)	K, M, P	1
-30D	65C25784-11		. PANEL INST-BAY NO. 3, BALANCE (LH) (FOR DETAILS SEE FIG. 6)	Q	1
-30E	65C25784-12		. PANEL INST-BAY NO. 3, BALANCE (RH) (FOR DETAILS SEE FIG. 6)	R	1
-30F	65C25784-29		. PANEL INST-BAY NO. 3, BALANCE (LH) (FOR DETAILS SEE FIG. 6)	U, Y, BA, DA, FA, HA, JA, NA	1
-30G	65C25784-30		. PANEL INST-BAY NO. 3, BALANCE (RH) (FOR DETAILS SEE FIG. 6)	V, Z, CA, EA, GA, IA, KA, PA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1-												
35	65C25785-15									. PANEL INST-BAY NO. 4, BALANCE (LH) (FOR DETAILS SEE FIG. 7)	A, C, E, G, S, W	1
-35A	65C25785-16									. PANEL INST-BAY NO. 4, BALANCE (RH) (FOR DETAILS SEE FIG. 7)	B, D, F, H, T, X	1
-35B	65C25785-3									. PANEL INST-BAY NO. 4, BALANCE (LH) (FOR DETAILS SEE FIG. 7)	J, L, N	1
-35C	65C25785-4									. PANEL INST-BAY NO. 4, BALANCE (RH) (FOR DETAILS SEE FIG. 7)	K, M, P	1
-35D	65C25785-11									. PANEL INST-BAY NO. 4, BALANCE (LH) (FOR DETAILS SEE FIG. 7)	Q	1
-35E	65C25785-12									. PANEL INST-BAY NO. 4, BALANCE (RH) (FOR DETAILS SEE FIG. 7)	R	1
-35F	65C25785-29									. PANEL INST-BAY NO. 4, BALANCE (LH) (FOR DETAILS SEE FIG. 7)	U, Y, BA, DA, FA, HA, JA, NA	1
-35G	65C25785-28									. PANEL INST-BAY NO. 4, BALANCE (RH) (FOR DETAILS SEE FIG. 7)	V, Z, CA, EA, GA, IA, KA, PA	1
40	16335									. RETAINER-TRAINING DISCHARGER (V55635)	A, B, J- M, S-Z, DA-IA	1
										ATTACHING PARTS		
45	BACR15BA5A6									. RIVET	A, B	4
										----- * -----		
50	69-74355-1									. BASE AND ADAPTER ASSY-STATIC DISCHARGER (LH)	A, J, L, S, U, W, Y, DA, FA, HA	1
-50A	69-74355-2									. BASE AND ADAPTER ASSY-STATIC DISCHARGER (RH)	B, K, M, T, V, X, Z, EA, GA, IA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-			ATTACHING PARTS								
55	BACR15GA5-11		.	R	I	V	E	T		A, B, J- M, S, T, W, X, Y, DA, EA	2
60	BACS12ER08K05		.	S	C	R	E	W		A, B, J- M, T-Z, EA-IA	2
65	NAS1836C08-08M		.	I	N	S	E	R	T	A, B, J- M, T-Z, EA-IA	2
-70	65-60535-17		.	W	E	I	G	H	T	A	1
										(LH)	
-70A	65-60535-18		.	W	E	I	G	H	T	A	1
										(RH)	
-70B	65-60535-13		.	W	E	I	G	H	T	A	1
										(LH)	
-70C	65-60535-14		.	W	E	I	G	H	T	A	1
										(RH)	
-70D	65-60535-15		.	W	E	I	G	H	T	A	1
										(LH)	
-70E	65-60535-16		.	W	E	I	G	H	T	A	1
										(RH)	
75	69-53211-2		.	.	W	E	I	G	H	T	22
80	69-53223-2		.	.	W	E	I	G	H	T	6
85	BACB30LM3-3		.	.	B	O	L	T			AR
-85A	BACB30LM3-5		.	.	B	O	L	T			AR
-85B	BACB30LM3-6		.	.	B	O	L	T			AR
-85C	BACB30LM3-7		.	.	B	O	L	T			AR
-85D	BACB30LM3-8		.	.	B	O	L	T			AR
-85E	BACB30LM3-10		.	.	B	O	L	T			AR

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-85F	BACB30LM3-11		.	.	BOLT					Q, R	AR
90	AN960C10L		.	.	WASHER						AR
95	BACN10JC3		.	.	NUT						AR
100	65C27564-13		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (LH)					A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA	1
-100A	65C27564-13		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (LH)					JA	1
-100B	65C27564-25		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (LH)					JA	1
-100C	65C27564-25		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (LH)					LA, NA	1
-105	65C27564-14		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (RH)					B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA	1
-105A	65C27564-14		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (RH) (LIMITED USAGE)					KA	1
-105B	65C27564-26		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (RH) (LIMITED USAGE)					KA	1
105C	65C27564-26		.		PANEL BOND ASSY-UPPER INBD HINGE COVER (RH)					MA, PA	1
110	65C27564-15		.		PANEL BOND ASSY-LOWER INBD HINGE COVER (LH)					A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-110A	65C27564-15		. PANEL BOND ASSY-LOWER INBD HINGE COVER (LH) (LIMITED USAGE)	JA	1
-110B	65C27564-27		. PANEL BOND ASSY-LOWER INBD HINGE COVER (LH) (LIMITED USAGE)	JA, LA	1
-110C	65C27564-27		. PANEL BOND ASSY-LOWER INBD HINGE COVER (LH)	NA	1
-115	65C27564-16		. PANEL BOND ASSY-LOWER INBD HINGE COVER (RH)	B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA, KA	1
-115A	65C27564-16		. PANEL BOND ASSY-LOWER INBD HINGE COVER (RH) (LIMITED USAGE)	KA	1
-115B	65C27564-28		. PANEL BOND ASSY-LOWER INBD HINGE COVER (RH) (LIMITED USAGE)	KA, MA	1
-115C	65C27564-28		. PANEL BOND ASSY-LOWER INBD HINGE COVER (RH)	PA	1
120	65C26396-1		. WEIGHT-ELEV TIP,BALANCE	A, B, J-L, S-Z, DA- IA	1
-120A	65C26396-2		. WEIGHT-ELEV TIP,BALANCE (OPT ITEM 120)	B, K, M, T, V, X, Z, EA, GA, IA	1
-120B	65C27280-1		. WEIGHT-ELEV TIP,BALANCE	C, D, E, F, N, P	1
-120C	65C27280-2		. WEIGHT-ELEV TIP,BALANCE (OPT ITEM 120D)	G, H, BA, CA, JA, KA, NA, PA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1- -120D	65C31311-2		. WEIGHT-ELEV TIP,BALANCE (OPT ITEM 120D)	G, H, BA, CA, JA, KA, NA, PA	1
-125	BACB30LE5-13		. BOLT	A, B, J- M, S-Z, DA-IA	4
-130	AN960KD516L		. WASHER	A, B, S- Z, DA-IA	8
-135	BACN10HR5CD		. NUT		4
			----- * -----		
140	65C25809-1		. FAIRING-OUTBD TIP (LH) (OPT ITEM 140A)	A, L, S, U, W, Y, DA, FA, HA	1
-140A	65C25809-3		. FAIRING-OUTBD TIP (LH) (OPT ITEM 140)	A, L, S, U, W, Y, DA, FA, HA	1
-140B	65C25809-2		. FAIRING-OUTBD TIP (RH) (OPT ITEM 140C)	B, K, M, T, V, X, Z, EA, GA, IA	1
-140C	65C25809-4		. FAIRING-OUTBD TIP (RH) (OPT ITEM 140B)	B, K, M, T, V, X, Z, EA, GA, IA	1
-140D	65C27281-1		. FAIRING-OUTBD TIP	C-H, N, P, BA, CA, JA, KA, NA, PA	1
-140E	65C27589-1		. FAIRING-OUTBD TIP	Q, R	1
145	65C26388-7		. PANEL-HINGE COVER (LH-UPR)	A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-145A	65C26388-8		. PANEL-HINGE COVER (LH-LWR)	A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA	1
-145B	65C26388-7		. PANEL-HINGE COVER (RH-LWR)	B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA	1
-145C	65C26388-8		. PANEL-HINGE COVER (RH-LWR)	B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA	1
-145D	65C26388-35		. PANEL-HINGE COVER (LH-UPR)	JA, LA, NA	1
-145E	65C26388-36		. PANEL-HINGE COVER (LH-LWR)	JA, LA, NA	1
-145F	65C26388-35		. PANEL-HINGE COVER (RH-LWR)	KA, MA, PA	1
-145G	65C26388-36		. PANEL-HINGE COVER (RH-UPR)	KA, MA, PA	1
146	69-78806-6		. FILLER-HINGE COVER	NA, PA	1
155	65C26388-6		. PANEL-HINGE COVER		2
-155A	65C26388-33		. PANEL-HINGE COVER (LH-UPR)	JA, LA, NA	1
-155B	65C26388-34		. PANEL-HINGE COVER (LH-LWR)	JA, LA, NA	1
-155C	65C26388-33		. PANEL-HINGE COVER (RH-LWR)	KA, MA, PA	1
-155D	65C26388-34		. PANEL-HINGE COVER (RH-UPR)	KA, MA, PA	1
156	69-78806-5		. FILLER-HINGE COVER	NA, PA	1
157	69-78806-4		. FILLER-HINGE COVER	NA, PA	1
160	65C26388-5		. PANEL-HINGE COVER		2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-160A	65C26388-31		.							JA, LA, NA	1
-160B	65C26388-32		.							JA, LA, NA	1
-160C	65C26388-31		.							KA, MA, PA	1
-160D	65C26388-32		.							KA, MA, PA	1
161	69-78806-3		.							NA, PA	1
162	69-78806-2		.							NA, PA	1
165	65C26388-4		.								2
-165A	65C26388-29		.							JA, LA, NA	1
-165B	65C26388-30		.							JA, LA, NA	1
-165C	65C26388-29		.							KA, MA, PA	1
-165D	65C26388-30		.							KA, MA, PA	1
166	69-78806-1		.							NA, PA	1
170	65C26388-3		.								2
-170A	65C26388-27		.							JA, LA, NA	1
-170B	65C26388-28		.							JA, LA, NA	1
-170C	65C26388-27		.							KA, MA, PA	1
-170D	65C26388-28		.							KA, MA, PA	1
175	65C26388-2		.								2
-175A	65C26388-25		.							JA, LA, NA	1
-175B	65C26388-26		.							JA, LA, NA	1
-175C	65C26388-25		.							KA, MA, PA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-175D	65C26388-26		.	PANEL-HINGE COVER (RH-UPR)						KA, MA, PA	1
180	65C26388-1		.	PANEL-HINGE COVER							2
-180A	65C26388-23		.	PANEL-HINGE COVER (LH-UPR)						JA, LA, NA	1
-180B	65C26388-24		.	PANEL-HINGE COVER (LH-LWR)						JA, LA, NA	1
-180C	65C26388-23		.	PANEL-HINGE COVER (RH-LWR)						KA, MA, PA	1
-180D	65C26388-24		.	PANEL-HINGE COVER (RH-UPR)						KA, MA, PA	1
				ATTACHING PARTS							
-190	BACB30LH3-2		.	BOLT							70
				-----*-----							
195	65C25746-51		.	SEAL (MAKE FROM 10-60754-722)						A-R	1
-195A	65C25746-51		.	SEAL (MAKE FROM 10-60754-722) (LIMITED USAGE)						Y, Z, BA, CA	1
-195B	65C25746-133		.	SEAL (MAKE FROM 10-60754-801)						FA, GA	1
-195C	65C25746-133		.	SEAL (MAKE FROM 10-60754-801) (LIMITED USAGE)						Y, Z, HA- KA, BA, CA	1
-195D	65C25746-135		.	SEAL (MAKE FROM 10-60754-835)						LA-PA	1
-195E	65C25746-135		.	SEAL (MAKE FROM 10-60754-835) (LIMITED USAGE)						HA-KA	1
-198	65C25746-52		.	SEAL (MAKE FROM 10-60754-722)						A-R	1
-198A	65C25746-52		.	SEAL (MAKE FROM 10-60754-722) (LIMITED USAGE)						Y, Z, BA, CA	1
-198B	65C25746-134		.	SEAL (MAKE FROM 10-60754-801)						FA, GA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-198C	65C25746-134		.	SEAL						HA-KA, Y, Z, BA, CA	1
				(MAKE FROM 10-60754-801)							
				(LIMITED USAGE)							
-198D	65C25746-136		.	SEAL						LA-PA	1
				(MAKE FROM 10-60754-835)							
-198E	65C25746-136		.	SEAL						HA-KA	1
				(MAKE FROM 10-60754-835)							
				(LIMITED USAGE)							
				ATTACHING PARTS							
200	BACB30LJ3-5		.	BOLT							9
205	BACB30LH3-6		.	BOLT							1
210	BACF33C101-0750		.	FILLER							10
215	BACW10CA3CVU		.	WASHER							13
220	BACW10CA3CCU		.	WASHER							13
225	BACN10JC3CD		.	NUT							10
				-----*							
230	65C28035-1		.	ANGLE (LH)						A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA, JA, NA	1
-230A	65C28035-2		.	ANGLE (RH)						B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA, KA, PA	1
235	65C28035-3		.	ANGLE (LH)						A, C, E, G, J, L, N, Q, S, U, W, Y, BA, DA, FA, HA, JA, NA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-235A	65C28035-4		.	ANGLE						B, D, F, H, K, M, P, R, T, V, X, Z, CA, EA, GA, IA, KA, PA	1
240	BAC27DWG27		.	PLATE-IDENTIFICATION							1
245	MS27253C2		.	PLATE-IDENTIFICATION						J-R, BA, CA, JA- PA	1
-245A	MS27253F2		.	PLATE-IDENTIFICATION						A-H, S-Z, DA, EA- IA	1
				ATTACHING PARTS							
250	BACR15DR3PA2		.	RIVET							6
				-----*-----							
255	65-53720-13		.	FAIRING-MAST FITTING (LH)						A, C, E, G, S, U, W, Y, BA, DA, FA, HA, JA, NA	1
-255A	65-58720-13		.	FAIRING-MAST FITTING (LH) (LIMITED USAGE)						J, L, N	1
-255B	65-53720-5		.	FAIRING-MAST FITTING (LH) (LIMITED USAGE)						J, L, N	1
-255C	65-53720-5		.	FAIRING-MAST FITTING (LH)						Q	1
-260	65-53720-14		.	FAIRING-MAST FITTING (RH)						B, D, F, H, T, V, X, Z, CA, EA, GA, IA, KA, PA	1
-260A	65-58720-14		.	FAIRING-MAST FITTING (RH) (LIMITED USAGE)						K, M, P	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-260B	65-53720-6		. FAIRING-MAST FITTING (RH) (LIMITED USAGE)	K, M, P	1
-260C	65-53720-6		. FAIRING-MAST FITTING (RH)	R	1
265	65C26390-1		. FAIRING-INBD TIP (LH) (OPT ITEM 265A)	A, J, L, S, U, W, Y, FA, HA	1
-265A	65C26390-3		. FAIRING-INBD TIP (LH) (OPT ITEM 265)	A, J, L, S, U, W, Y, FA, HA	1
-265B	65C26390-2		. FAIRING-INBD TIP (RH) (OPT ITEM 265C)	B, K, M, T, V, X, Z, IA, GA	1
-265C	65C26390-4		. FAIRING-INBD TIP (RH) (OPT ITEM 265B)	B, K, M, T, V, X, Z, IA, GA	1
-265D	65C27282-1		. FAIRING-INBD TIP	C, D, E, F, N, P	1
-265E	65C27282-2		. FAIRING-INBD TIP	G, H, BA, CA, JA, KA, NA, PA	1
			ATTACHING PARTS		
-270	BACB30LH3-5		. BOLT ----- * -----		58
275	65C25746-47		. PANEL BOND ASSY (LH-LWR)	A, S	1
-275A	65C25746-53		. PANEL BOND ASSY (LH-LWR)	C	1
-275B	65C25746-75		. PANEL BOND ASSY (LH-LWR)	E, G, BA, JA	1
-275C	65C25746-13		. PANEL BOND ASSY (LH-LWR)	J, L	1
-275D	65C25746-25		. PANEL BOND ASSY (LH-LWR)	N	1
-275E	65C25746-17		. PANEL BOND ASSY (LH-LWR)	Q	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-275F	65C25746-125		. PANEL BOND ASSY (LH-LWR)	U, FA, HA	1
-275G	65C25746-71		. PANEL BOND ASSY (LH-LWR)	W, Y, DA	1
-275H	65C25746-113		. PANEL BOND ASSY (LH-LWR)	Y, DA	1
-275I	65C25746-143		. PANEL BOND ASSY (LH-LWR)	LA, NA	1
280	65C25746-49		. PANEL BOND ASSY (LH-UPR)	A, S	1
-280A	65C25746-55		. PANEL BOND ASSY (LH-UPR)	C	1
-280B	65C25746-77		. PANEL BOND ASSY (LH-UPR)	E, G, BA, JA	1
-280C	65C25746-11		. PANEL BOND ASSY (LH-UPR)	J, L	1
-280D	65C25746-15		. PANEL BOND ASSY (LH-UPR)	Q	1
-280E	65C25746-23		. PANEL BOND ASSY (LH-UPR)	N	1
-280F	65C25746-127		. PANEL BOND ASSY (LH-UPR)	U, FA, HA	1
-280G	65C25746-73		. PANEL BOND ASSY (LH-UPR)	W, Y, DA	1
-280H	65C25746-115		. PANEL BOND ASSY (LH-UPR)	Y, DA	1
-280I	65C25746-145		. PANEL BOND ASSY (LH-UPR) (LIMITED USAGE)	LA, NA	1
-280J	65C25746-147		. PANEL BOND ASSY (LH-UPR) (LIMITED USAGE)	LA, NA	1
-280K	65C25746-149		. PANEL BOND ASSY (LH-UPR)	LA, NA	1
-280L	65C25746-151		. PANEL BOND ASSY (LH-UPR)	LA, NA	1
-285	65C25746-48		. PANEL BOND ASSY (RH-LWR)	B, T	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-285A	65C25746-54		. PANEL BOND ASSY (RH-LWR)	D	1
-285B	65C25746-76		. PANEL BOND ASSY (RH-LWR)	F, H, CA, KA	1
-285C	65C25746-14		. PANEL BOND ASSY (RH-LWR)	K, M	1
-285D	65C25746-26		. PANEL BOND ASSY (RH-LWR)	P	1
-285E	65C25746-18		. PANEL BOND ASSY (RH-LWR)	R	1
-285F	65C25746-126		. PANEL BOND ASSY (RH-LWR)	V, GA, IA	1
-285G	65C25746-72		. PANEL BOND ASSY (RH-LWR)	X, Z, EA	1
-285H	65C25746-114		. PANEL BOND ASSY (RH-LWR)	Z, EA	1
-285I	65C25746-144		. PANEL BOND ASSY (RH-LWR)	MA, PA	1
-290	65C25746-50		. PANEL BOND ASSY (RH-UPR)	B, T	1
-290A	65C25746-56		. PANEL BOND ASSY (RH-UPR)	D	1
-290B	65C25746-12		. PANEL BOND ASSY (RH-UPR)	K, M	1
-290C	65C25746-24		. PANEL BOND ASSY (RH-UPR)	P	1
-290D	65C25746-16		. PANEL BOND ASSY (RH-UPR)	R	1
-290E	65C25746-78		. PANEL BOND ASSY (RH-UPR)	F, H, CA, KA	1
-290F	65C25746-128		. PANEL BOND ASSY (RH-UPR)	V, GA, IA	1
-290G	65C25746-74		. PANEL BOND ASSY (RH-UPR)	X, Z, EA	1
-290H	65C25746-116		. PANEL BOND ASSY (RH-UPR)	Z, EA	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-290I	65C25746-146		. PANEL BOND ASSY (RH-UPR) (LIMITED USAGE)	MA, PA	1
-290J	65C25746-148		. PANEL BOND ASSY (RH-UPR) (LIMITED USAGE)	MA, PA	1
-290K	65C25746-150		. PANEL BOND ASSY (RH-UPR)	MA, PA	1
-290L	65C25746-152		. PANEL BOND ASSY (RH-UPR)	MA, PA	1
295	65C31488-1		. CONDUCTIVE FRAME (LH-UPR)	U, FA, HA	1
300	65C31488-4		. CONDUCTIVE FRAME (LH-LWR)	U, FA, HA	1
305	65C31488-2		. CONDUCTIVE FRAME (RH-UPR)	V, GA, IA	1
310	65C31488-3		. CONDUCTIVE FRAME (RH-LWR)	V, GA, IA	1
315	65C28166-1		. FITTING-TIP ASSY (LH)	C, E, G, N, BA, JA, NA	1
320	65C28166-2		. FITTING-TIP ASSY (RH)	D, F, H, P, CA, KA, PA	1
325	65C28166-5		. FITTING-TIP ASSY (LH)	G, BA, JA, NA	1
330	65C28166-6		. FITTING-TIP ASSY (RH)	H, CA, KA, PA	1
-335	65C28166-3		. . FITTING-TIP (USED ON ITEMS 475,480)		1
-340	65C29213-1		. . FITTING-TIP (OPT ITEM 495)		1
-345	65C31695-1		. . FITTING (USED ON ITEMS 485,490)		1
-350	BACN10JR3CFD		. . NUTPLATE		6

-Item not Illustrated

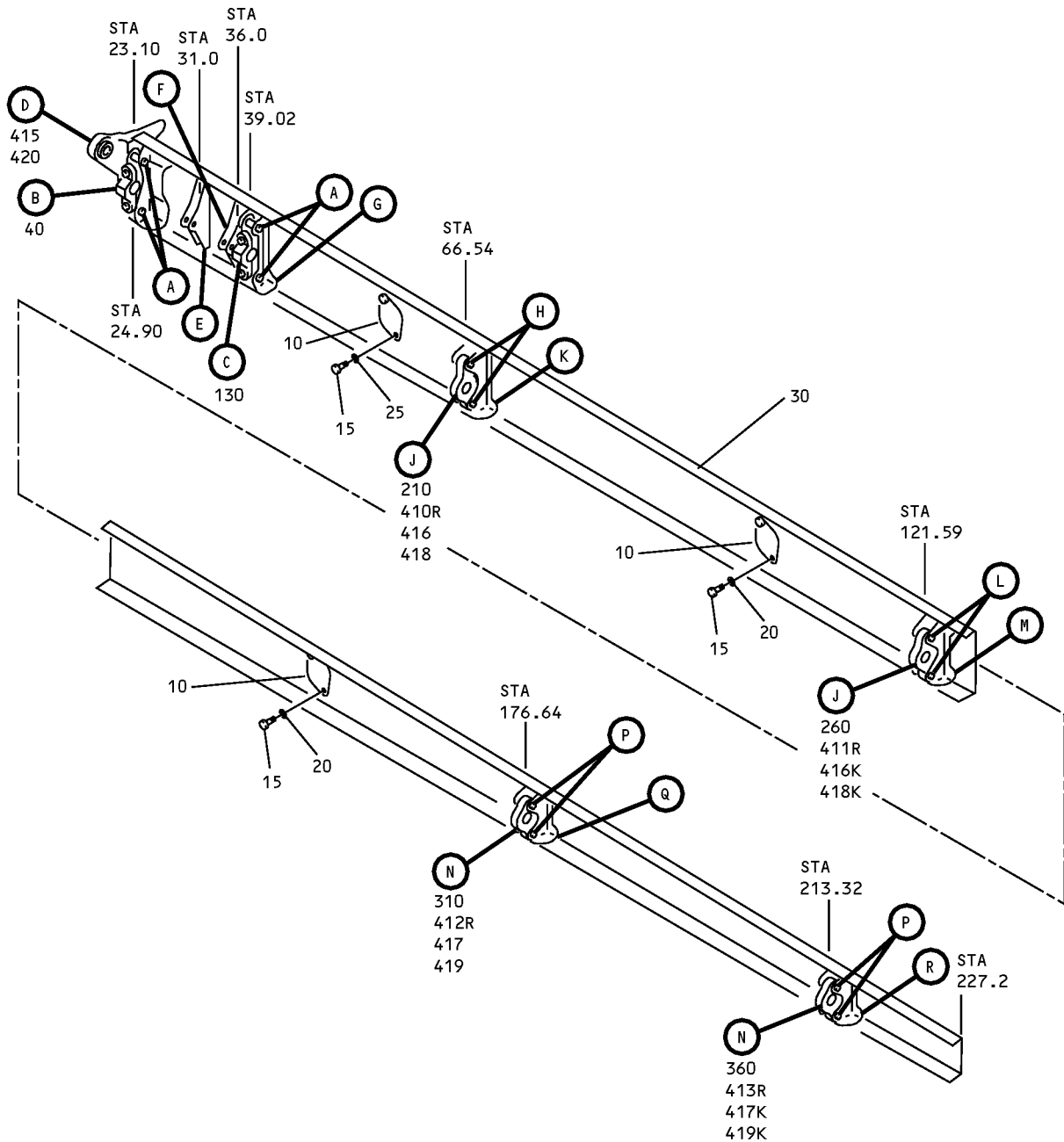
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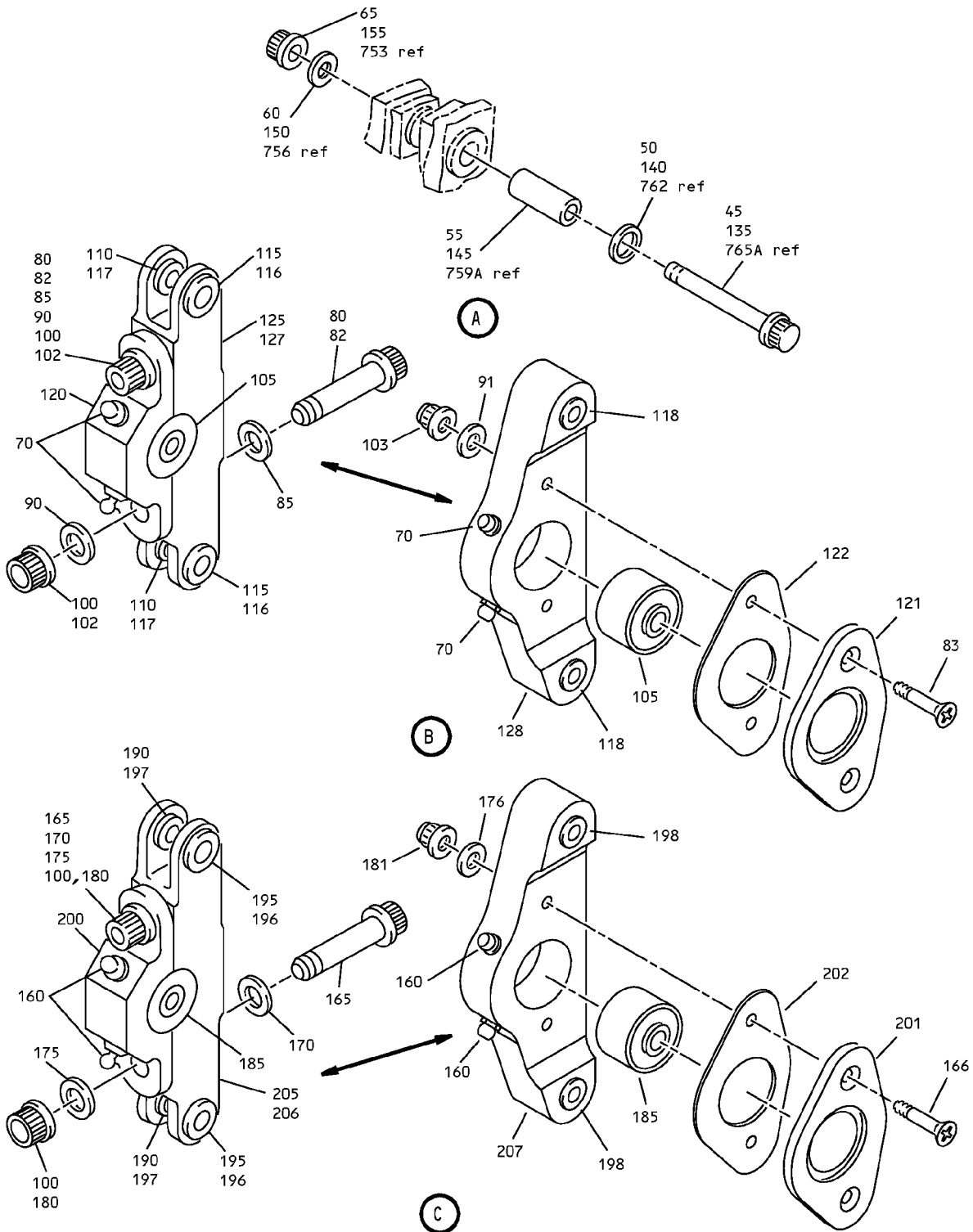
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Front Spar Installation
IPL Figure 2 (Sheet 1 of 6)

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Front Spar Installation
IPL Figure 2 (Sheet 2 of 6)

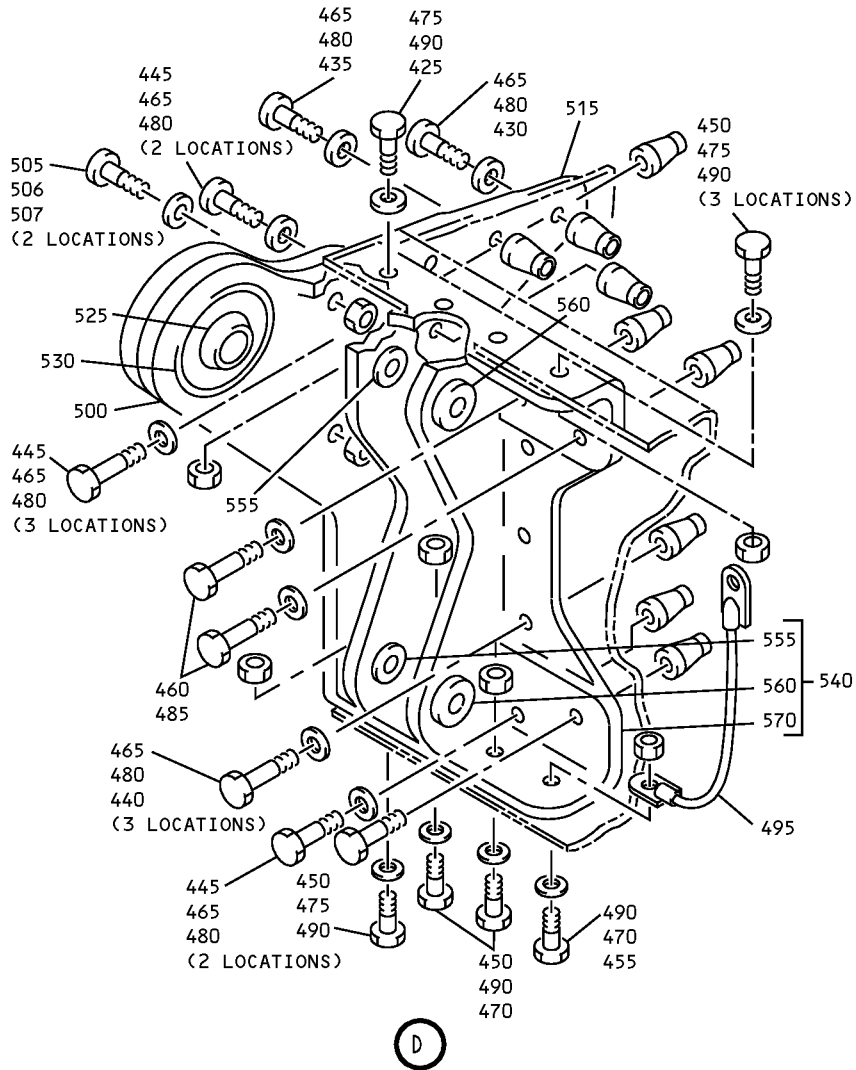
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IPL Figure 2 (Sheet 3 of 6)

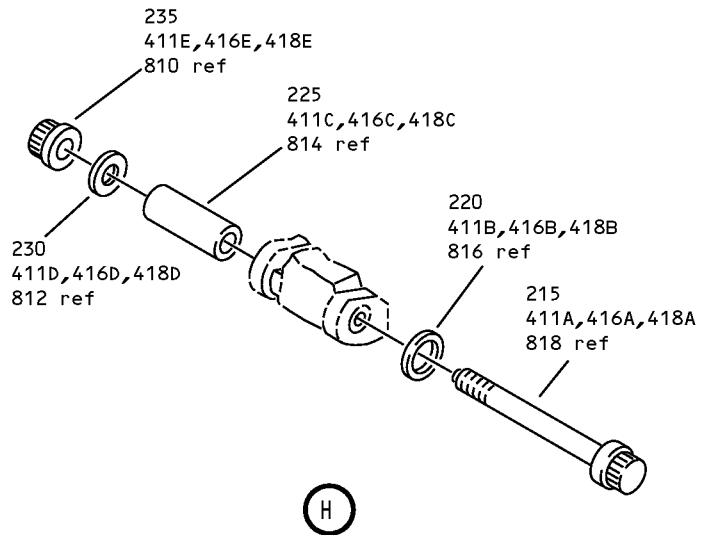
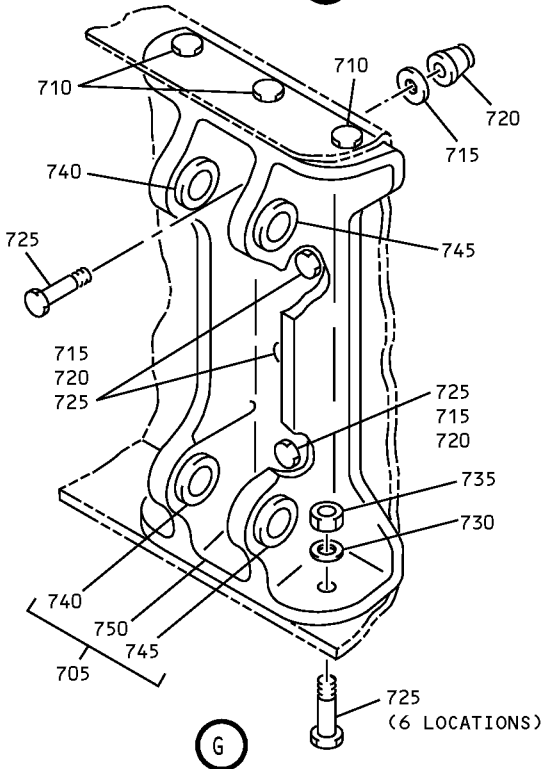
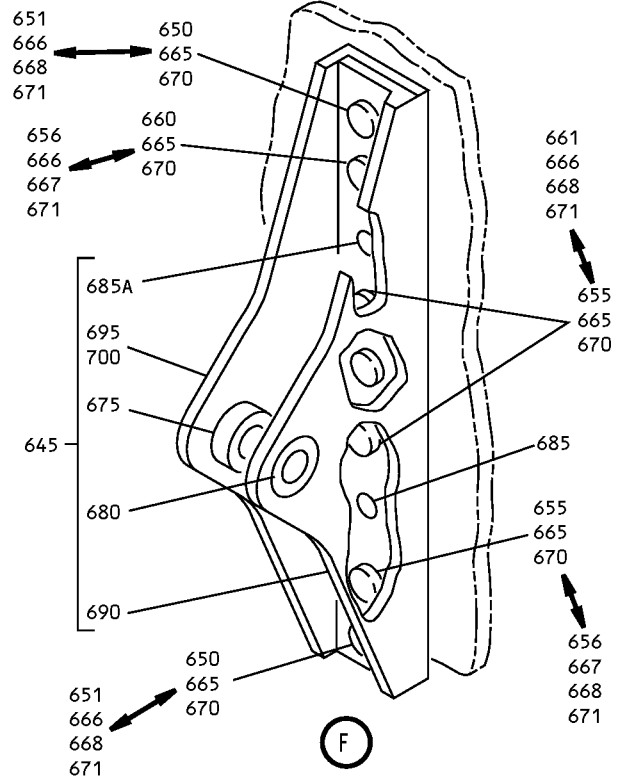
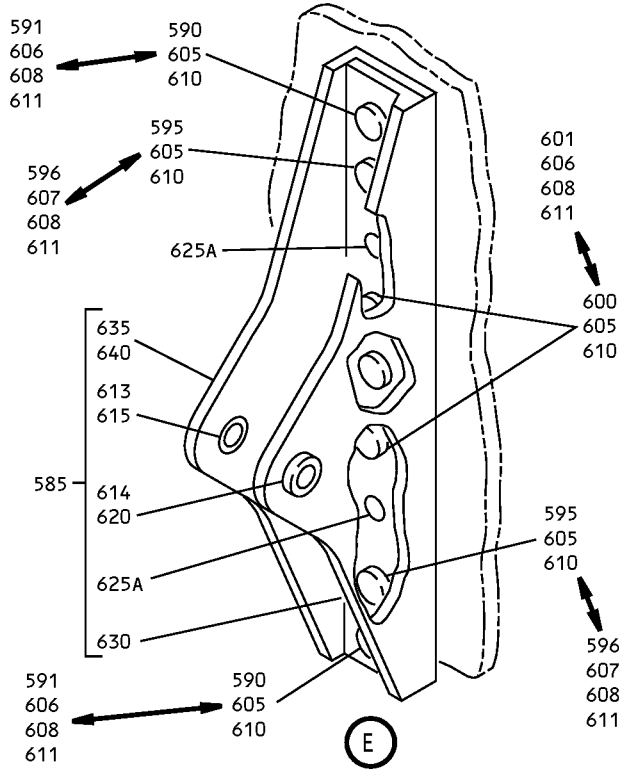
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IPL Figure 2 (Sheet 4 of 6)

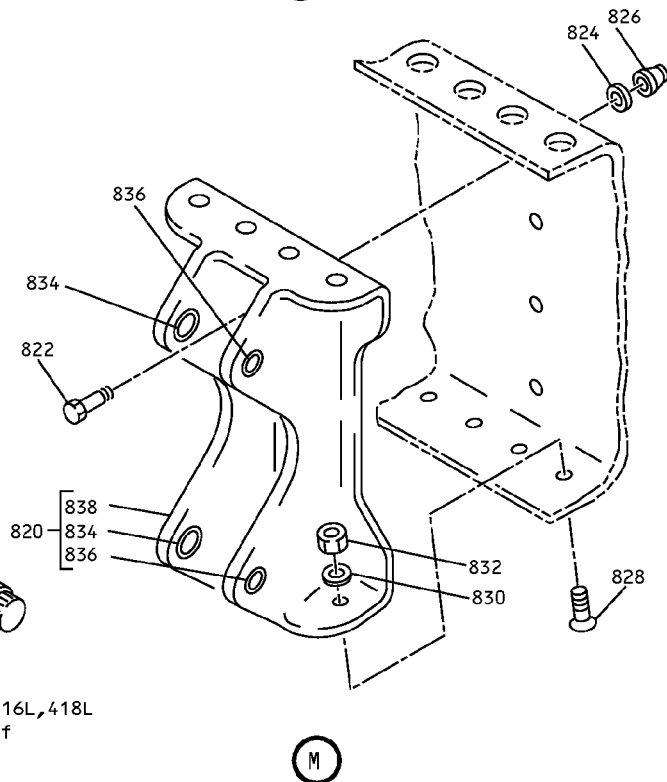
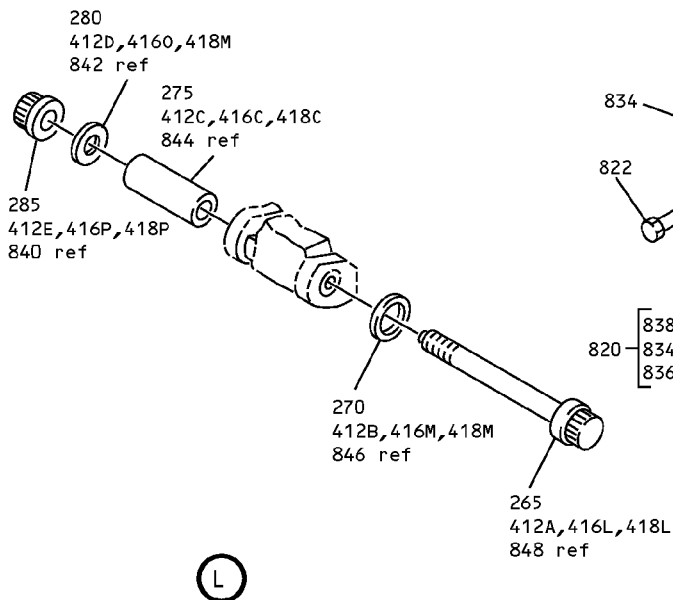
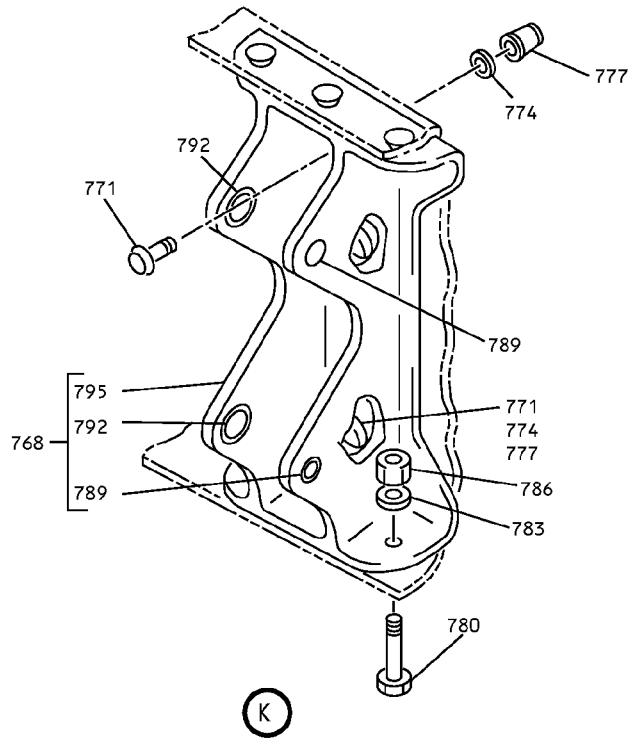
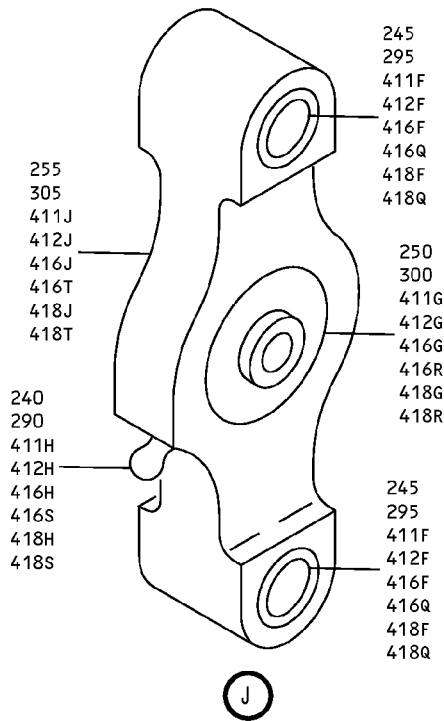
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Front Spar Installation
IPL Figure 2 (Sheet 5 of 6)

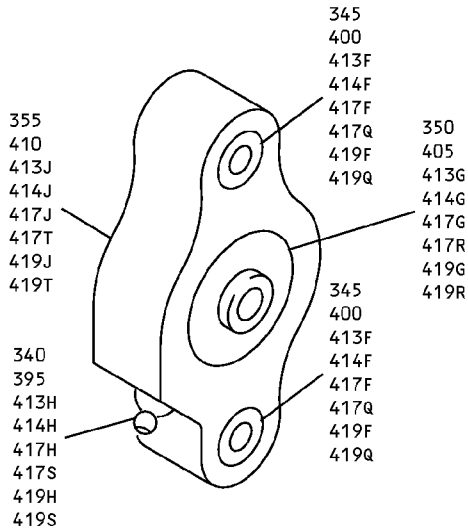
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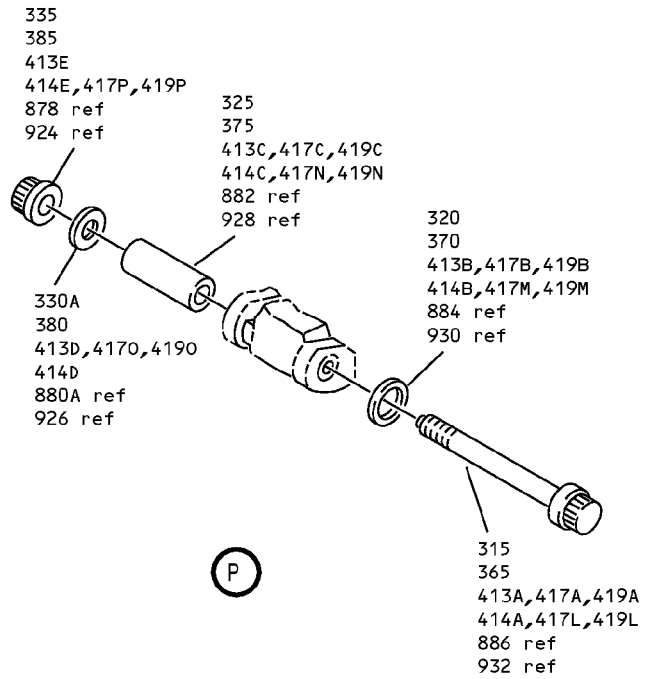
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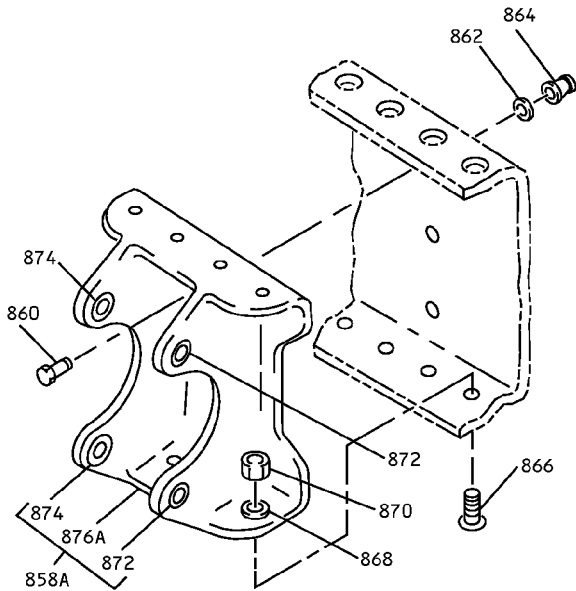
COMPONENT MAINTENANCE MANUAL



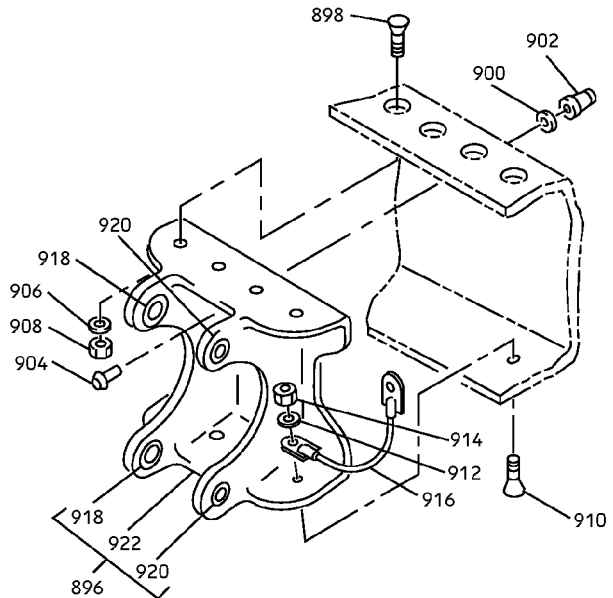
(N)



(P)



(Q)



(R)

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-1	65C25756-13									A	RF
-1A	65C25756-19									C	RF
-1B	65C25756-1									E	RF
-1C	65C25756-9									G	RF
-1D	65C25756-11									J	RF
-1E	65C25756-25									L	RF
-1F	65C25756-27									N	RF
-1G	65C25756-29									Q	RF
-1H	65C25756-31									S	RF
-1J	65C25756-33									U	RF
-1K	65C25756-41									W	RF
-5	65C25756-14									B	RF
-5A	65C25756-20									D	RF
-5B	65C25756-2									F	RF
-5C	65C25756-10									H	RF
-5D	65C25756-12									K	RF
-5E	65C25756-26									M	RF
-5F	65C25756-28									P	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-5G	65C25756-30									R	RF
-5H	65C25756-32									T	RF
-5J	65C25756-34									V	RF
-5K	65C25756-42									X	RF
10	173N2024-1										3
-10A	69-73701-1									A, B, L, M, Q, R- X	3
15	BACB30NM3K3										6
20	BACW10BP3CT										4
25	BACW10BP3CTU										2
27	BACN10JR3CFDR										6
28	BACR15DR3A2										12
30	65C25755-20									A-D, L-P, Q, R	1
-30A	65C25755-1									E, F, J, K	1
-30B	65C25755-19									G, H	1
-30C	65C25755-24									S-X	1
40	65C27301-4									E-H,S-V	1
-40A	65C27301-1									E-H	1
-40B	65C27301-4									A-D,J- N,P-R	1
-40C	65C27301-5									S, T	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
2- -40D	65C27301-5		. PLATE ASSY-BEARING (LIMITED USAGE) (USED WITH ITEM 55) (65C34548-1 AND BACB28K05-046 OPT TO 65C27301-5 AND BACB28AK05-125)	U, V	1
-40E	65C34548-1		. PLATE ASSY-BEARING (LIMITED USAGE) (USED WITH ITEM 55A) (65C34548-1 AND BACB28K05-046 OPT TO 65C27301-5 AND BACB28AK05-125)	U, V	1
-40F	65C34549-1		DELETED ATTACHING PARTS		
-40G	65C34548-1		. PLATE ASSY-BEARING (USED WITH ITEM 55A)	W, X	1
45	BACB30NM5K32		. BOLT		2
50	BACW10BP5CTU		. WASHER (UNDER HEAD)		2
55	BACB28AK05-125		. BUSHING		2
-55A	BACB28AK05-046		. BUSHING (USED WITH ITEM 40E,40G) (65C34548-1 AND BACB28K05-046 OPT TO 65C27301-5 AND BACB28AK05-125)		1
60	BACW10BP5DP		. WASHER (UNDER NUT)		2
65	NAS1804-5L		. NUT		2
			----- * -----		
70	MS15001-1		. . FITTING-LUBE		2
80	BACB30LE6U24		. . BOLT (USED ON ITEM 40A)		2
82	BACB30NM6K24		. . BOLT (USED ON ITEMS 40, 40B, 40C)		2
83	BACB30NN4K8		. . BOLT (USED WITH ITEM 40E, 40G)		2
85	BACW10BN6SC		. . WASHER		2
90	BACW10BN6UP		. . WASHER		2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
91	BACW10BP4NDP		. .	WASHER							2
				(USED WITH ITEM 40E, 40G)							
100	BACN10HR6CS		. .	NUT							2
				(USED ON ITEM 40A)							
102	NAS1805-6		. .	NUT							2
				(USED ON ITEMS 40, 40B, 40C)							
103	NAS1805-4N		. .	NUT							2
				(USED WITH ITEM 40E, 40G)							
105	DAS6-23B		. .	BEARING							1
				(V77876)							
110	BACB28AP05P019		. .	BUSHING							2
115	BACB28AM07B019		. .	BUSHING							2
116	BACB28AM07~ B029A		. .	BUSHING							2
				(USED ON ITEM 40C)							
117	BACB28AP05P029		. .	BUSHING							2
				(USED ON ITEM 40C)							
118	BACB28AP05P098		. .	BUSHING							2
				(USED WITH ITEM 40E, 40G)							
120	65C27302-1		. .	CAP							1
121	69-77838-1		. .	RETAINER PLATE							1
				(USED WITH ITEM 40E, 40G)							
122	69-77839-1		. .	LAMINATED SHIM							1
				(USED WITH ITEM 40E, 40G)							
125	65C27303-1		. .	BLOCK-PILLOW							1
127	69-76941-2		. .	BLOCK-PILLOW							1
				(USED ON 40C)							
128	65C34548-2		. .	BUSHING SUPPORT							1
				(USED WITH ITEM 40E, 40G)							
130	65C26529-1		. .	PLATE ASSY-BEARING					A-R		1
				(ELEV STA 39.02)							
-130A	65C26529-1		. .	PLATE ASSY-BEARING					S-V		1
				(LIMITED USAGE)							
-130B	65C26529-4		. .	PLATE ASSY-BEARING					S-V		1
				(LIMITED USAGE)							
-130C	65C34549-1		. .	PLATE ASSY-BEARING					U, V		1
				(LIMITED USAGE)							
				(USED WITH ITEM 145A)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
2- -130D	65C34549-1		. PLATE ASSY-BEARING (USED WITH ITEM 145A) ATTACHING PARTS	W, X	1
135	BACB30NM5K32		. BOLT		2
140	BACW10BP5CTU		. WASHER (UNDER HEAD)		2
145	BACB28AK05-130		. BUSHING		2
-145A	BACB28AK05-042		. BUSHING (USED WITH ITEM 130C, 130D)		2
150	BACW10BP5DP		. WASHER (UNDER NUT)		2
155	NAS1804-5L		. NUT -----*		2
160	MS15001-1		. . FITTING-LUBE		2
165	BACB30LE6U24		. . BOLT		2
166	BACB30NN4K9		. . BOLT (USED WITH ITEM 130C, 130D)		2
170	BACW10BN6SC		. . WASHER		2
175	BACW10BN6UP		. . WASHER		2
176	BACW10BP4NDP		. . WASHER (USED WITH ITEM 130C, 130D)		2
180	BACN10HR6CS		. . NUT		2
181	NAS1805-4N		. . NUT (USED WITH ITEM 130C, 130D)		2
185	BACB10CH65		. . BEARING		1
190	BACB28AP05P019		. . BUSHING		2
195	BACB28AM07B019		. . BUSHING		2
196	BACB28AM07~ B031A		. . BUSHING (USED ON ITEM 130A)		2
197	BACB28AP05P031		. . BUSHING (USED ON ITEM 130A)		2
198	BACB28AP05P108		. . BUSHING (USED WITH ITEM 130C, 130D)		2
200	65C26530-1		. . CAP		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
201	69-77838-2		. .								1
202	69-77839-2		. .								1
205	65C26531-1		. .								1
206	69-76940-2		. .								1
207	65C34549-2		. .								1
210	65C26481-1		. PLATE ASSY-BEARING						A-P		1
			(ELEV STA 66.54)								
			ATTACHING PARTS								
215	BACB30NR5K33		. BOLT								2
			(HEAD DIRECTION OPT)								
220	BACW10BN5AC		. WASHER								2
			(UNDER HEAD)								
225	BACB28AK05-168		. BUSHING								2
230	AN960KD516		. WASHER								2
			(UNDER NUT)								
235	BACN10JC5CD		. NUT								2
			----- * -----								
240	MS15001-1		. . FITTING-LUBE								1
245	10-61975-2		. . BUSHING								2
250	BACB10CH50		. . BEARING								1
255	65C26481-2		. . PLATE-BEARING								1
260	65C26481-1		. PLATE ASSY-BEARING						A-P		1
			(ELEV STA 121.59)								
			ATTACHING PARTS								
265	BACB30NR5K36		. BOLT								2
270	BACW10BN5AC		. WASHER								2
			(UNDER HEAD)								
275	BACB28AK05-187		. BUSHING								2
280	AN960KD516		. WASHER								2
			(UNDER NUT)								

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
285	BACN10JC5CD										
290	MS15001-1										
295	10-61975-2										
300	BACB10CH50										
305	65C26481-2										
310	65C26482-1								A-P		
315	BACB30NR4K36										
320	BACW10BN4AC										
325	BACB28AK04-187										
330	BACW10BN4AC										
-330A	AN960KD416										
335	BACN10JC4CD										
340	MS15001-1										
345	10-61975-3										
350	BACB10CH50										
355	65C26482-2										
360	65C26482-1										
365	BACB30NR4K36										
370	BACW10BN4AC										
375	BACB28AK04-190										
380	AN960KD416										
385	BACN10JC4CD										

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-395	MS15001-1										1
400	10-61975-3										2
405	BACB10CH50										1
410	65C26482-2										1
410R	69-76409-1								L-V		1
-410S	69-76409-5								L-V		1
411	69-76409-1										
411A	BACB30NR5K33										2
411B	BACW10BN5AC										2
411C	BACB28AK05-168										2
411D	AN960KD516										2
411E	BACN10JC5CD										2
411F	10-61975-2										2
411G	BACB10CH50										1
411H	MS15001-1										1
411J	69-76409-2										1
-411K	69-76409-6										1
411R	69-76409-1										1
-411S	69-76409-5										1
412	69-76409-1										

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-			ATTACHING PARTS								
412A	BACB30NR5K36		.	BOLT						2	
412B	BACW10BN5AC		.	WASHER (UNDER HEAD)						2	
412C	BACB28AK05-187		.	BUSHING						2	
412D	AN960KD516		.	WASHER (UNDER NUT)						2	
412E	BACN10JC5CD		.	NUT						2	
			----- * -----								
-412F	10-61975-2		..	BUSHING						2	
-412G	BACB10CH50		..	BEARING						1	
-412H	MS15001-1		..	FITTING-LUBE						1	
-412J	69-76409-2		..	PLATE-BEARING						1	
-412K	69-76409-6		..	PLATE-BEARING						1	
412R	69-76419-1		.	PLATE ASSY-BEARING (ELEV STA 176.54) (PRE SB 737-55-1054) (PRE SB 737-55-1054R2)					L-V	1	
-412S	69-76419-5		.	PLATE ASSY-BEARING (ELEV STA 176.54) (POST SB 737-55-1054) (POST SB 737-55-1054R2)					L-V	1	
413	69-76419-1		DELETED								
			ATTACHING PARTS								
413A	BACB30NR4K36		.	BOLT						2	
413B	BACW10BN4AC		.	WASHER (UNDER HEAD)						2	
413C	BACB28AK04-187		.	BUSHING						2	
413D	AN960KD416		.	WASHER (UNDER NUT)						2	
413E	BACN10JC4CD		.	NUT						2	
			----- * -----								
-413F	10-61975-3		..	BUSHING						2	
-413G	BACB10CH50		..	BEARING						1	
-413H	MS15001-1		..	FITTING-LUBE						1	

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-413J	69-76419-2		.	.							1
-413K	69-76419-6		.	.							1
413R	69-76419-1		.						L-V		1
-413S	69-76419-5		.						L-V		1
414	69-76419-1										
414A	BACB30NR4K36		.								2
414B	BACW10BN4AC		.								2
414C	BACB28AK04-190		.								2
414D	AN960KD416		.								2
414E	BACN10JC4CD		.								2
-414F	10-61975-3		.	.							2
-414G	BACB10CH50		.	.							1
-414H	MS15001-1		.	.							1
-414J	69-76419-2		.	.							1
-414K	69-76419-6		.	.							1
415	65C27376-3		.						A, C, J, L, N, Q		1
-415A	65C27376-3		.						E, G, S, U		1
-415B	65C27376-1		.						E, G		1
-415C	65C27376-5		.						S, U		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-415D	65C27376-7		.	FITTING ASSY-MASTARM						U	1
				(LH)							
				(LIMITED USAGE)							
				(OPT ITEM 415C)							
-415E	65C27376-9		.	FITTING ASSY-MASTARM						U, W	1
				(LH)							
				(LIMITED USAGE)							
-415F	65C27376-11		.	FITTING ASSY-MASTARM						W	1
				(LH)							
				(LIMITED USAGE)							
416	69-76409-5		.	PLATE ASSY-BEARING						W, X	1
				(ELEV STA 66.54)							
				(OPT ITEM 418)							
				ATTACHING PARTS							
-416A	BACB30NR5K33		.	BOLT							2
416B	BACW10BN5AC		.	WASHER							2
				(UNDER HEAD)							
416C	BACB28AK05-168		.	BUSHING							2
416D	AN960KD516		.	WASHER							2
				(UNDER NUT)							
416E	BACN10JC5CD		.	NUT							2
				----- * -----							
416F	10-61975-2		. .	BUSHING							2
416G	BACB10CH50		. .	BEARING							1
416H	MS15001-1		. .	FITTING-LUBE							1
416J	69-76409-6		. .	PLATE-BEARING							1
416K	69-76409-5		.	PLATE ASSY-BEARING						W, X	1
				(ELEV STA 121.59)							
				(OPT ITEM 418K)							
				ATTACHING PARTS							
416L	BACB30NR5K36		.	BOLT							2
416M	BACW10BN5AC		.	WASHER							2
				(UNDER HEAD)							
416N	BACB28AK05-187		.	BUSHING							2
416O	AN960KD516		.	WASHER							2
				(UNDER NUT)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
416P	BACN10JC5CD							. NUT			2
								-----*			
416Q	10-61975-2							. . BUSHING			2
416R	BACB10CH50							. . BEARING			1
416S	MS15001-1							. . FITTING-LUBE			1
416T	69-76409-6							. . PLATE-BEARING			1
417	69-76419-5							. PLATE ASSY-BEARING (ELEV STA 176.54) (OPT ITEM 419)	W, X		1
								ATTACHING PARTS			
417A	BACB30NR4K36							. BOLT			2
417B	BACW10BN4AC							. WASHER (UNDER HEAD)			2
417C	BACB28AK04-187							. BUSHING			2
417D	AN960KD416							. WASHER (UNDER NUT)			2
417E	BACN10JC4CD							. NUT			2
								-----*			
417F	10-61975-3							. . BUSHING			2
417G	BACB10CH50							. . BEARING			1
417H	MS15001-1							. . FITTING-LUBE			1
417J	69-76419-6							. . PLATE-BEARING			1
417K	69-76419-5							. PLATE ASSY-BEARING (ELEV STA 213.32) (OPT ITEM 419K)	W, X		1
								ATTACHING PARTS			
417L	BACB30NR4K36							. BOLT			2
417M	BACW10BN4AC							. WASHER (UNDER HEAD)			2
417N	BACB28AK04-190							. BUSHING			2
417O	AN960KD416							. WASHER (UNDER NUT)			2
417P	BACN10JC4CD							. NUT			2
								-----*			

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-417Q	10-61975-3										2
417R	BACB10CH50										1
417S	MS15001-1										1
417T	69-76419-6										1
418	69-76409-7									W, X	1
418A	BACB30NR5K33										2
418B	BACW10BN5AC										2
418C	BACB28AK05-168										2
418D	AN960KD516										2
418E	BACN10JC5CD										2
-418F	10-61975-2										2
-418G	BACB10CH50										1
-418H	MS15001-1										1
-418J	69-76409-8										1
-418K	69-76409-7									W, X	1
418L	BACB30NR5K36										2
418M	BACW10BN5AC										2
418N	BACB28AK05-190										2
418O	AN960KD516										2
418P	BACN10JC5CD										2
418Q	10-61975-2										2
418R	BACB10CH50										1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
418S	MS15001-1							. . FITTING-LUBE			1
418T	69-76409-8							. . PLATE-BEARING			1
419	69-76419-7							. PLATE ASSY-BEARING (ELEV STA 176.54) (OPT ITEM 417)	W, X		1
								ATTACHING PARTS			
419A	BACB30NR4K36							. BOLT			2
419B	BACW10BN4AC							. WASHER (UNDER HEAD)			2
419C	BACB28AK04-187							. BUSHING			2
419D	AN960KD416							. WASHER (UNDER NUT)			2
419E	BACN10JC4CD							. NUT			2
								-----*			
419F	10-61975-3							. . BUSHING			2
419G	BACB10CH50							. . BEARING			1
419H	MS15001-1							. . FITTING-LUBE			1
419J	69-76419-8							. . PLATE-BEARING			1
419K	69-76419-7							. PLATE ASSY-BEARING (ELEV STA 213.32) (OPT ITEM 417K)	W, X		1
								ATTACHING PARTS			
419L	BACB30NR4K36							. BOLT			2
419M	BACW10BN4AC							. WASHER (UNDER HEAD)			2
419N	BACB28AK04-190							. BUSHING			2
419O	AN960KD416							. WASHER (UNDER NUT)			2
419P	BACN10JC4CD							. NUT			2
								-----*			
419Q	10-61975-3							. . BUSHING			2
419R	BACB10CH50							. . BEARING			1
419S	MS15001-1							. . FITTING-LUBE			1
419T	69-76419-8							. . PLATE-BEARING			1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
2-					
420	65C27376-4		. FITTING ASSY-MASTARM (RH)	B, D, K, M, P, R	1
-420A	65C27376-4		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	F, H, T, V	1
-420B	65C27376-2		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	F, H	1
-420C	65C27376-6		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	T, V	1
-420D	65C27376-8		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	V	1
-420E	65C27376-10		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	V, X	1
-420F	65C27376-12		. FITTING ASSY-MASTARM (RH) (LIMITED USAGE)	X	1
			ATTACHING PARTS		
425	BACB30VF4K8		. BOLT		1
430	BACB30MY8K3		. BOLT		1
435	BACB30MY8K4		. BOLT		1
440	BACB30MY8K5		. BOLT	A-K	1
445	BACB30MY8K6		. BOLT		7
450	BACB30NZ8K8		. BOLT		6
455	BACB30NY8K9		. BOLT		1
460	BACB30NX8K12		. BOLT		1
465	BACW10BP4NAPU		. WASHER		10
470	BACW10BP4NDP		. WASHER	S, T, U, V	6
475	BACW10P61S		. WASHER		2
480	BACC30AB8C		. COLLAR	E-K	10
485	BACC30X8		. COLLAR	A-K	1
490	BACN10JC4CD		. NUT		8

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
495	BACJ40AB23-8								. JUMPER-BONDING		1
									-----*-----		
500	NAS516-1								. . FITTING-LUBE		1
505	BACB30MY8C9								. . BOLT (USED ON ITEMS 415, 415A, 415B, 420, 420A, 420B)		2
-505A	BACB30VT8K10								. . BOLT (USED ON ITEMS 415C, 415D, 415E, 415F, 420C, 420D, 420E, 420F)		2
506	BACN10HR4CD								. . NUT (USED ON ITEMS 415, 415A, 415B, 420, 420A, 420B)		2
-506A	BACN10YR4CD								. . NUT (USED ON ITEMS 415C, 415D, 415E, 415F, 420C, 420D, 420E, 420F)		2
507	AN960JD416								. . WASHER (USED ON ITEMS 415C, 415D, 415E, 415F, 420C, 420D, 420E, 420F)		2
510	BACC30AG8								DELETED		
515	65C26528-1								. . STRAP-FAIL SAFE (LH) (USED ON ITEMS 415, 415A, 415B, 415C, 415D, 415E, 415F)		1
-520	65C26528-2								. . STRAP-FAIL SAFE (RH) (USED ON ITEMS 420, 420A, 420B, 420C, 420D, 420E, 420F)		1
525	DAS10-26A1-4								. . BEARING (REPLACED BY ITEM 525A) (V77896) (USED ON ITEMS 415, 415A, 415B, 420, 420A, 420B)		1
-525A	DAS10-26A1-501								. . BEARING (REPLACES ITEM 525) (V77896) (USED ON ITEMS 415C, 415D, 415E, 415F, 420C, 420D, 420E, 420F)		1
530	69-63523-1								. . SLEEVE (REPLACED BY ITEM 530B) (USED ON ITEMS 415, 415A, 415B, 420, 420A, 420B)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-530A	69-63523-2		. .	SLEEVE							1
				(REPLACED BY ITEM 530B)							
				(USED ON ITEMS 415C, 415D, 415E, 415F, 420C, 420D, 420E, 420F)							
-530B	69-63523-3		. .	SLEEVE							1
				(REPLACES ITEMS 530, 530A)							
				(USED ON ITEMS 415F, 420F)							
540	65C25769-7		. .	FITTING-MAST							1
				(LH)							
				(USED ON ITEMS 415, 415A, 415C)							
540A	65C25769-1		. .	FITTING-MAST							1
				(LH)							
				(USED ON ITEM 415B)							
-540B	65C25769-13		. .	FITTING-MAST							1
				(USED ON ITEM 415D)							
-540C	65C25769-15		. .	FITTING-ASSY MAST							1
				(USED ON ITEM 415E)							
550	65C25769-8		. .	FITTING-MAST							1
				(RH)							
				(USED ON ITEMS 420, 420A, 420C)							
-550A	65C25769-2		. .	FITTING-MAST							1
				(RH)							
				(USED ON ITEM 420B)							
-550B	65C25769-14		. .	FITTING-MAST							1
				(USED ON ITEM 420D)							
-550C	65C25769-16		. .	FITTING-MAST							1
				(USED ON ITEMS 420E, 420F)							
555	BACB28AP05P030		. . .	BUSHING							2
				(USED ON ITEMS 540, 540A, 540B, 550, 550A, 550B)							
-555A	BACB28AT07B028C		. . .	BUSHING							2
				(USED ON ITEMS 540C, 550C)							
560	BACB28AM07~ B029A		. . .	BUSHING							2
				(USED ON ITEMS 540, 540A, 540B, 550, 550A, 550B)							
-560A	BACB28AP05F029		. . .	BUSHING							2
				(USED ON ITEMS 540C, 550C)							
570	65C25769-9		. . .	FITTING							1
				(LH)							
				(USED ON ITEM 540)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
570A	65C25769-3						. . . FITTING (LH) (USED ON ITEM 540A)				1
-570B	65C25769-11						. . . FITTING (USED ON ITEM 540B)				1
-570C	65C25769-19						. . . FITTING (USED ON ITEM 540C)				1
580	65C25769-10						. . . FITTING (RH) (USED ON ITEM 550)				1
-580A	65C25769-4						. . . FITTING (RH) (USED ON ITEM 550A)				1
-580B	65C25769-12						. . . FITTING (RH) (USED ON ITEM 550D)				1
-580C	65C25769-20						. . . FITTING (RH) (USED ON ITEM 550C)				1
585	65C27351-5						. BRACKET ASSY-CONTROL ROD				1
-585A	65C27351-1						. BRACKET ASSY-CONTROL ROD (OPT ITEM 585)				1
							ATTACHING PARTS				
590	BACB30NX6K7						. BOLT	A-K			2
591	BACB30NM3K8						. BOLT	L-X			2
595	BACB30NX6K11						. BOLT	A-K			2
596	BACB30NM3K12						. BOLT	L-X			2
600	BACB30NX6K6						. BOLT	A-K			3
601	BACB30NM3K6						. BOLT	L-X			3
605	BACW10BP3NAPU						. WASHER	A-K			2
606	BACW10BP3NAPU						. WASHER (UNDER NUT)	L-X			5
607	BACW10B3NDP						. WASHER (UNDER NUT)	L-X			2
608	BACW10BP3CD						. WASHER (UNDER HEAD)	L-X			7
610	BACC30X6						. COLLAR	A-K			7

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
611	NAS1805-3L		.	NUT						L-X	7
				-----*							
613	69-50565-5		..	BUSHING (USED ON ITEM 585)							1
614	69-50565-6		..	BUSHING (USED ON ITEM 585)							1
615	69-50565-3		..	BUSHING (USED ON ITEM 585A)							1
620	69-50565-4		..	BUSHING (USED ON ITEM 585A)							1
625	BACR15DA6D			DELETED							
-625A	BACR15BA6D		..	RIVET							2
630	65C27351-7		..	ANGLE (USED ON ITEM 585)							1
-630A	65C27351-3		..	ANGLE (USED ON ITEM 585A)							1
635	65C27351-8		..	ANGLE (USED ON ITEM 585)							1
640	65C27351-4		..	ANGLE (USED ON ITEM 585A)							1
645	65C27351-6		.	BRACKET ASSY-CONTROL ROD (LIMITED USAGE)						A-T	1
-645A	65C27351-2		.	BRACKET ASSY-CONTROL ROD (OPT ITEM 645) (LIMITED USAGE)						A-T	1
-645B	65C27351-9		.	BRACKET ASSY-CONTROL ROD (LIMITED USAGE)						U, V	1
-645C	65C27351-9		.	BRACKET ASSY-CONTROL ROD						W, X	1
				ATTACHING PARTS							
650	BACB30NX6K7		.	BOLT						A-K	2
651	BACB30NM3K8		.	BOLT						L-X	2
655	BACB30NX6K11		.	BOLT						A-K	2
656	BACB30NM3K12		.	BOLT						L-X	2
660	BACB30NX6K6		.	BOLT						A-K	3
661	BACB30NM3K6		.	BOLT						L-X	3

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
665	BACW10BP3NAPU		.	W	A	S	H	E	R		2
666	BACW10BP3NAPU		.	W	A	S	H	E	R	L-X	5
667	BACW10BP3NDP		.	W	A	S	H	E	R	L-X	2
668	BACW10BP3CD		.	W	A	S	H	E	R	L-X	7
670	BACC30X6		.	C	O	L	L	A	R	A-K	7
671	NAS1805-3L		.	N	U	T				L-X	7
675	66-25183-3		.	.	B	U	S	H	I		1
-675A	66-25183-6		.	.	B	U	S	H	I		1
680	69-44494-502		.	.	B	U	S	H	I		1
-680A	69-44494-502		.	.	B	U	S	H	I		1
-680B	69-44494-4		.	.	B	U	S	H	I		1
685	BACR15DA6D										
-685A	BACR15BA6D		.	.	R	I	V	E	T		2
690	65C27351-7		.	.	A	N	G	L	E		1
-690A	65C27351-3		.	.	A	N	G	L	E		1
695	65C27351-8		.	.	A	N	G	L	E		1
700	65C27351-4		.	.	A	N	G	L	E		1
705	65C25759-1		.	F	I	T	T	I	N		1
710	BACB30NX8K8		.	B	O	L	T			A-K	4
715	BACW10BP4NAPU		.	W	A	S	H	E	R		4
720	BACC30X8		.	C	O	L	L	A	R	A-K	4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
725	BACB30NZ8K8		.	B	O	L	T				6
730	BACW10BP4NDP		.	W	A	S	H	E	R	S, T, U, V	6
735	BACN10JC4CD		.	N	U	T					6
			-----*-----								
740	BACB28AP05A023		.	.	B	U	S	H	I	N	2
745	BACB28AM07~ B025A		.	.	B	U	S	H	I	N	2
750	65C25759-2		.	.	F	I	T	T	I	N	1
753	NAS1804-5		.	N	U	T					RF
756	BACW10BP5DP		.	W	A	S	H	E	R	(UNDER NUT)	RF
759	BACB28AK05-125		DELETED								
-759A	BACB28AK05-130		.	B	U	S	H	I	N		RF
762	BACW10BP5CTU		.	W	A	S	H	E	R	(UNDER HEAD)	RF
765	BACB30LE5K32		DELETED								
-765A	BACB30NM5K32		.	B	O	L	T				RF
768	65C25761-1		.	F	I	T	T	I	N	A	1
			ATTACHING PARTS								
771	BACB30NX10K5		.	B	O	L	T			A-K	4
774	BACW10BP5APU		.	W	A	S	H	E	R		4
777	BACC30X10		.	C	O	L	L	A	R	A-K	4
780	BACB30NW8K6		.	B	O	L	T				6
783	BACW10BP4NDP		.	W	A	S	H	E	R	S, T, U, V	6
786	BACN10JC4CD		.	N	U	T					6
			-----*-----								
789	BACB28AP05P020		.	.	B	U	S	H	I	N	2
792	BACB28AM07~ B020A		.	.	B	U	S	H	I	N	2
795	65C25761-2		.	.	F	I	T	T	I	N	1
798	MS15001-1		DELETED								
801	10-61975-2		DELETED								
804	BACB10CH50		DELETED								

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
807	65C26481-2										
810	BACN10JC5CD										RF
812	AN960KD516										RF
814	BACB28AK05-168										RF
816	BACW10BN5AC										RF
818	BACB30NR5K33										RF
820	65C25763-1										1
822	BACB30NX8K6										4
824	BACW10BP4NAPU										4
826	BACC30X8								A-K		4
828	BACB30NW8K6										8
830	BACW10BP3NDP								C, D, N, P, S, T, U, V		8
832	BACN10JC3CD								A-K		8
834	BACB28AM07~ B018A										2
836	BACB28AP05P018										2
838	65C25763-2										1
840	BACN10JC5CD										RF
842	AN960KD516										RF
844	BACB28AK05-187										RF
846	BACW10BN5AC										RF
848	BACB30NR5K36										RF
850	MS15001-1										DELETED
852	10-61975-2										DELETED
854	BACB10CH50										DELETED

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
856	65C26481-2										
858	65C25765-1										
-858A	65C25765-6										1
860	BACB30NX6K4									A-K, N, P, U, V	3
862	BACW10BP3NAPU										3
864	BACC30X6									A-K	3
866	BACB30NW6K5										8
868	BACW10BP3NDP									C, D, N, P, S, T, U, V	8
870	BACN10JC3CD									L-V	8
872	BACB28AP04P018										2
874	BACB28AM06~ B018A										2
876	65C25765-2										
-876A	65C25765-4										1
878	BACN10JC4CD										RF
880	BACW10BN4AC										
-880A	AN960KD416										RF
882	BACB28AK04-187										RF
884	BACW10BN4AC										RF
886	BACB30NR4K36										RF
888	MS15001-1										
890	10-61975-2										
892	BACB10CH50										
894	65C26481-2										
896	65C25767-1									A, B, E- M, Q-T	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-896A	65C31307-4								. FITTING ASSY-HINGE (OPT ITEM 896C)	D, P, V	1
-896B	65C31307-3								DELETED		
-896C	65C31307-11								. FITTING ASSY-HINGE (OPT ITEM 896A)	D, P, V	1
-896D	65C31307-9								. FITTING ASSY-HINGE (OPT ITEM 896E)	C, N, U	1
-896E	65C31307-7								. FITTING ASSY-HINGE (OPT ITEM 896D)	C, N, U	1
									ATTACHING PARTS		
898	BACB30MY6K4								. BOLT		3
900	BACW10BP3NAPU								. WASHER		3
902	BACC30AB6C								. COLLAR	A-K	3
904	BACB30NW6K5								. BOLT		7
906	BACW10BP3NDP								. WASHER	C, D, N, P, S-V	7
908	BACN10JC3CD								. NUT	L-V	7
910	BACB30NW6K7								. BOLT		1
912	BACW10BP3NDP								. WASHER	C, D, N, P, S-V	1
914	BACN10JC3CD								. NUT	L-V	1
916	BACJ40AB26-5								. JUMPER-BONDING		1
									----- * -----		
918	BACB28AP04P015								. . BUSHING		2
920	BACB28AM06~ B015A								. . BUSHING		2
922	65C25767-2								. . FITTING (USED ON ITEM 896)		1
-922A	65C31307-6								. . FITTING (USED ON ITEM 896A)		1
-922B	65C31307-5								DELETED		
-922C	65C31307-12								. . FITTING (USED ON ITEM 896C)		1
-922D	65C31307-10								. . FITTING (USED ON ITEM 896D)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2-											
-922E	65C31307-8		.	.	FITTING (USED ON ITEM 896E)						1
924	BACN10JC4CD		.		NUT						RF
926	AN960KD416		.		WASHER (UNDER NUT)						RF
928	BACB28AK04-190		.		BUSHING				A, B, E- M, Q-T		RF
930	BACW10BN4AC		.		WASHER (UNDER HEAD)						RF
932	BACB30NR4K36		.		BOLT						RF
934	MS15001-1				DELETED						
936	10-61975-2				DELETED						
938	BACB10CH50				DELETED						
940	65C26481-2				DELETED						
					INSTALLATION PARTS						
-950	69-44683-2				BOLT ASSY						1
-955	69-44682-2		.		BOLT-HOLLOW						1
-960	NAS587-29		.		BOLT						1
-965	NAS76A10-11		.		BUSHING						1
-970	NAS679A7				NUT						1
-975	BACN10BY510				NUT						1

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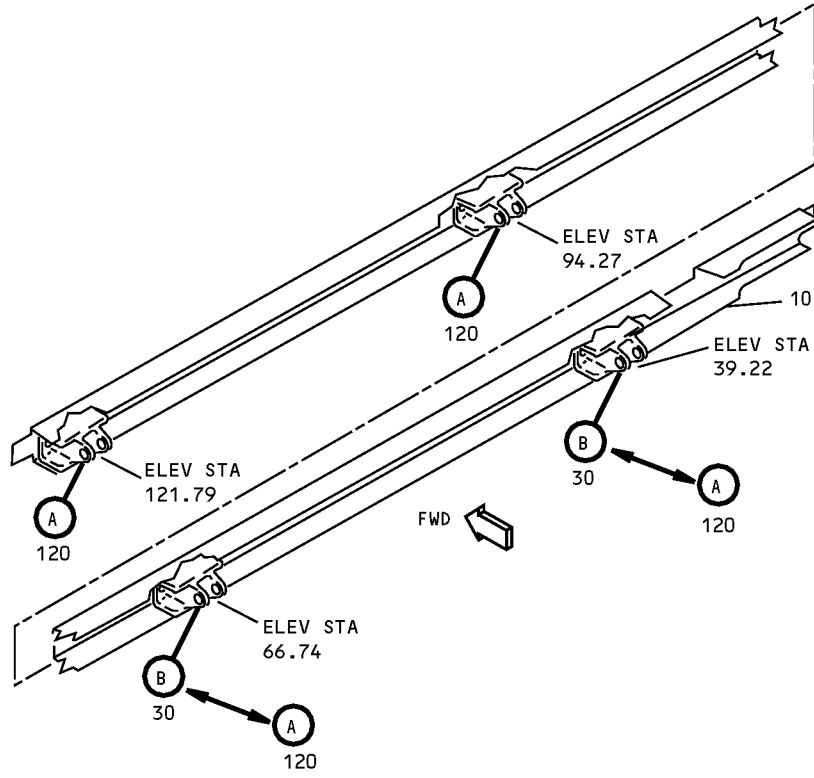
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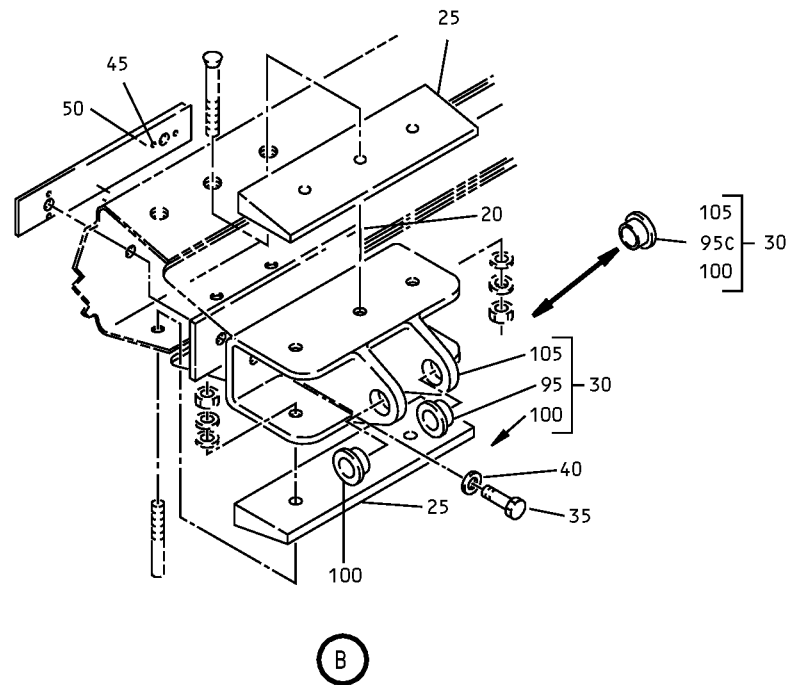
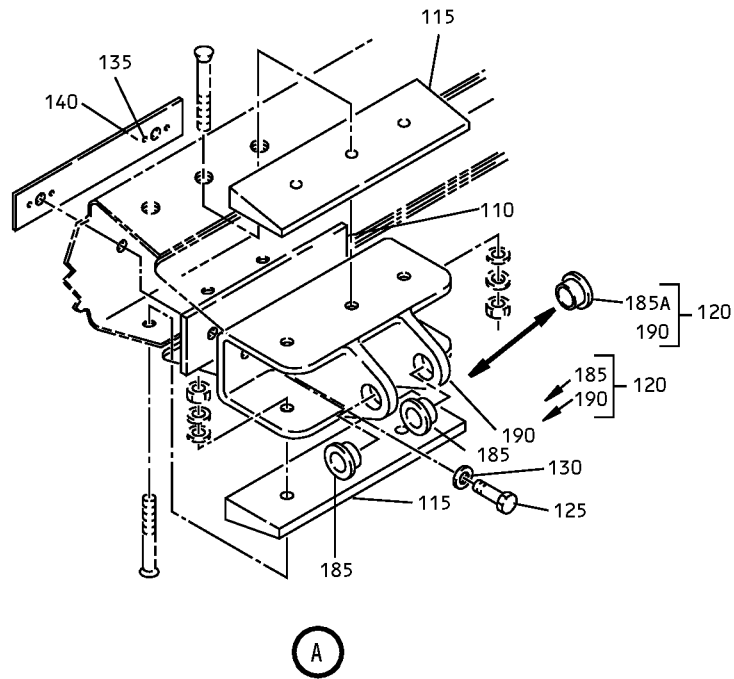
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REAR VIEW
REAR SPAR

Rear Spar Installation
IPL Figure 3 (Sheet 1 of 2)

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Rear Spar Installation
IPL Figure 3 (Sheet 2 of 2)



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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
3-					
-1	65C25771-5		SPAR INSTL-REAR (LH)	A	RF
-1A	65C25771-1		SPAR INSTL-REAR (LH)	C	RF
-5	65C25771-6		SPAR INSTL-REAR (RH)	B	RF
-5A	65C25771-2		SPAR INSTL-REAR (RH)	D	RF
10	65C25772-1		. SPAR (LH)	A, C	1
15	65C25772-2		. SPAR (RH)	B, D	1
20	65C25771-3		. SHIM		1
25	69-73579-1		. FILLER		2
30	65C25773-11		. FITTING ASSY-HINGE (PRE SB 737-55-1043)		1
-30A	65C25773-16		. FITTING ASSY-HINGE ATTACHING PARTS		1
35	BACB30NM3C5		. BOLT		2
40	BACW10BP3CTU		. WASHER		2
45	BACR15DR3A2		. RIVET		4
-50	BACN10JR3FDR		. NUTPLATE -----*		2
-55	65C25746-27		DELETED		
-60	65C25746-28		DELETED		
-65	65C25746-61		DELETED		
-70	65C25746-62		DELETED		
75	BACB30LH3SU8		DELETED		
80	BACN10JC3CD		DELETED		
85	BACWCA3CVU		DELETED		
90	BACW10CA3CCU		DELETED		
95	BACB28AP04P013		. . BUSHING (USED ON ITEM 30) (PRE SB 737-55-1043)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
3-											
-95A	BACB28AP04P011		.	.	BUSHING (USED ON ITEM 30A) (POST SB 737-55-1043)						1
95C	BACB28AM06~ B013A		.	.	BUSHING (ELEVATOR STATION 66.74 ONLY)						1
100	BACB28AM06~ B013A		.	.	BUSHING						1
105	65C25773-13		.	.	FITTING-HINGE						1
110	65C25771-3		.		SHIM						3
115	69-73579-1		.		FILLER						6
120	65C25773-15		.		FITTING ASSY-HINGE (POST SB 737-55-1035)				A, B		3
120A	65C25773-12		.		FITTING ASSY-HINGE (PRE SB 737-55-1035)				A, B		3
-120B	65C25773-12		.		FITTING ASSY-HINGE (PRE SB 737-55-1043)				A, B		1
			ATTACHING PARTS								
125	BACB30NM3C5		.		BOLT						2
130	BACW10BP3CTU		.		WASHER						2
135	BACR15DR3A2		.		RIVET						4
140	BACN10JR3FDR		.		NUTPLATE						2
			----- * -----								
-145	65C25746-27		DELETED								
-150	65C25746-28		DELETED								
-155	65C25746-61		DELETED								
-160	65C25746-62		DELETED								
165	BACB30LH3SU8		DELETED								
170	BACN10JC3CD		DELETED								
175	BACWCA3CVU		DELETED								
180	BACW10CA3CCU		DELETED								
185	BACB28AM06~ B013A		.	.	BUSHING						2

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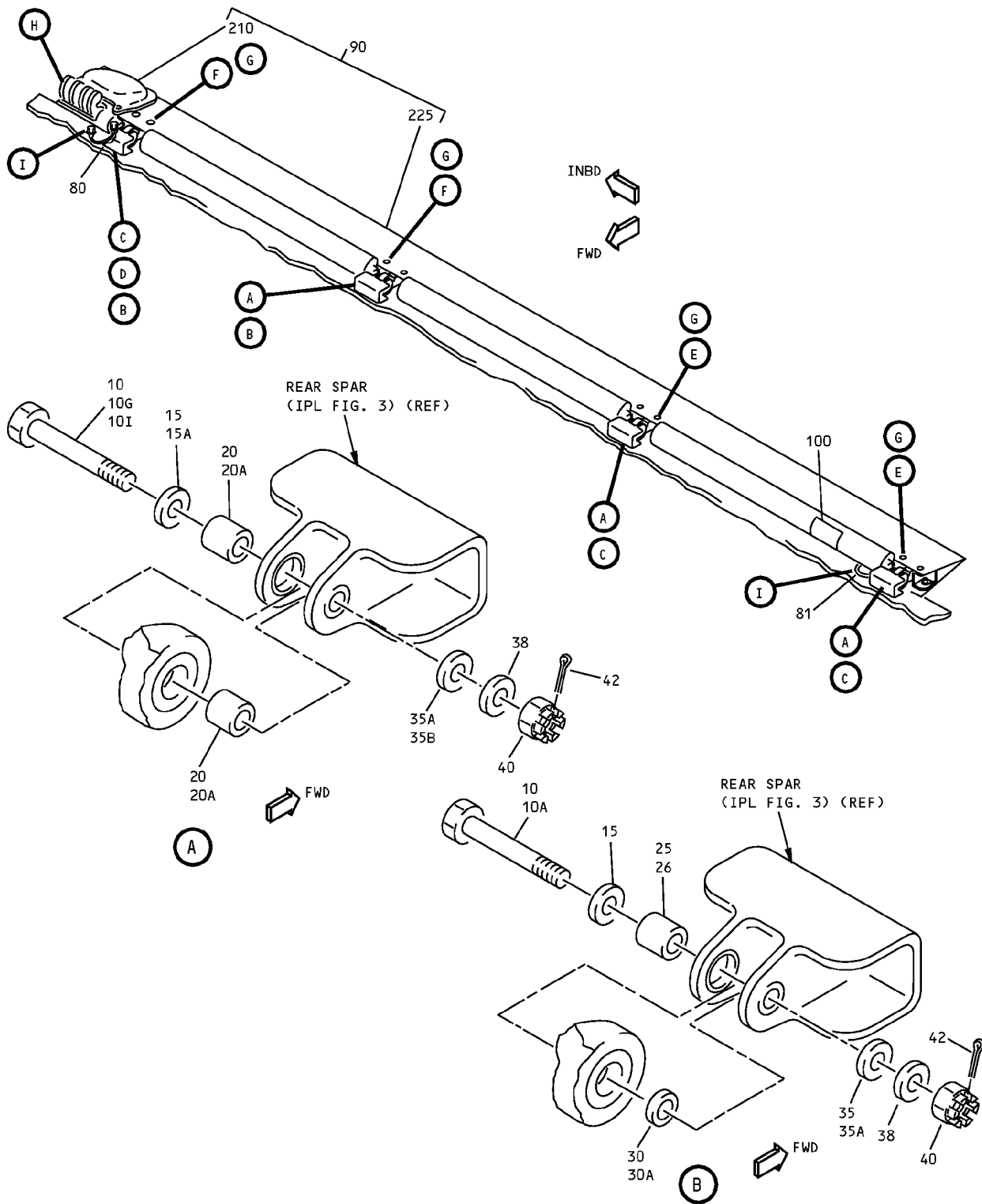


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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
3- 185A	69-72039-38		. . BUSHING (PRE SB 737-55-1043) (POST SB 737-55-1043) (ELEVATOR STATION 39.22 ONLY)								1
190	65C25773-14		. . FITTING-HINGE								1

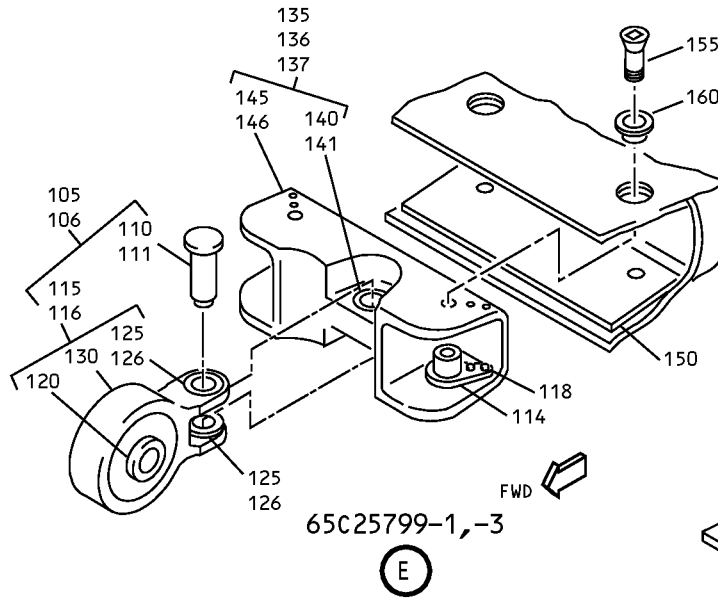
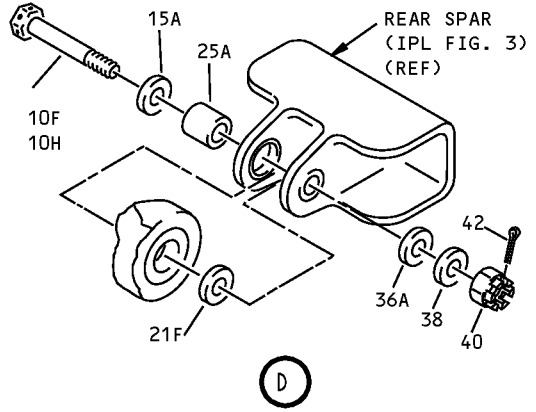
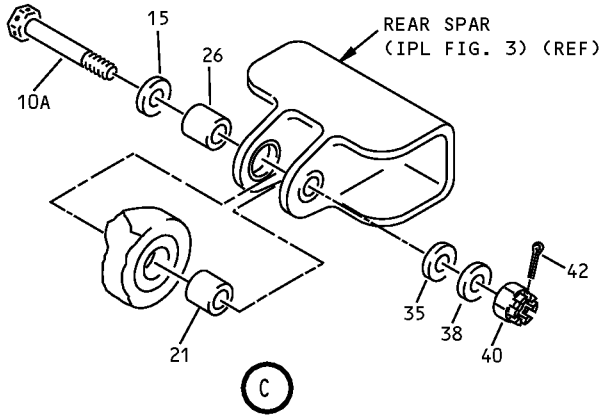
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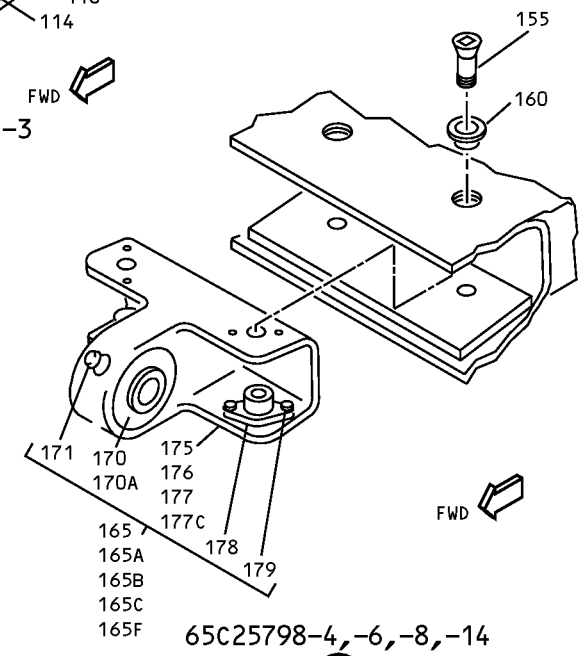
Control Tab Installation
IPL Figure 4 (Sheet 1 of 3)

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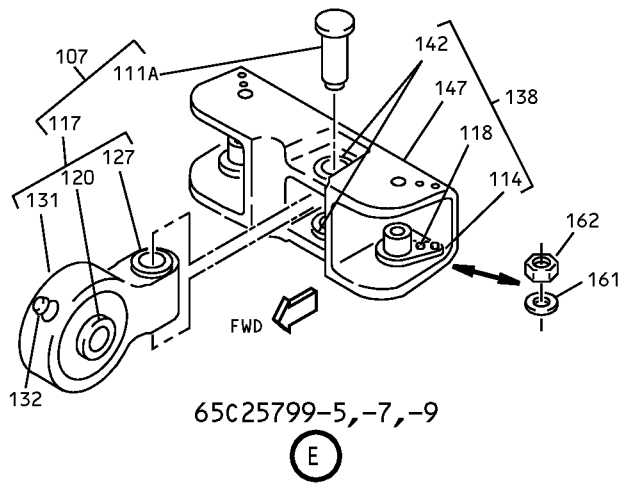
65C25799-1,-3

E



65C25798-4,-6,-8,-14

F



65C25799-5,-7,-9

E

Control Tab Installation
IPL Figure 4 (Sheet 2 of 3)

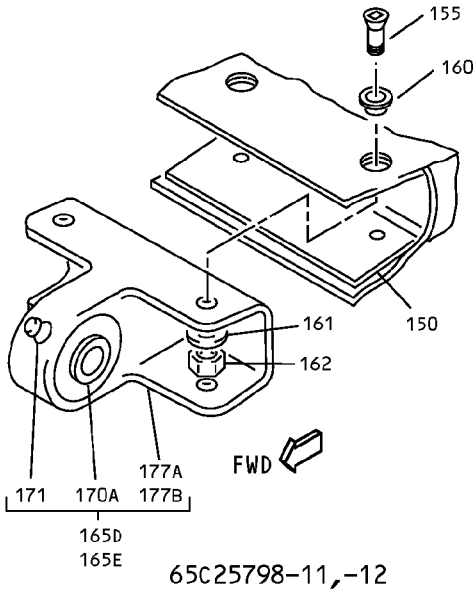
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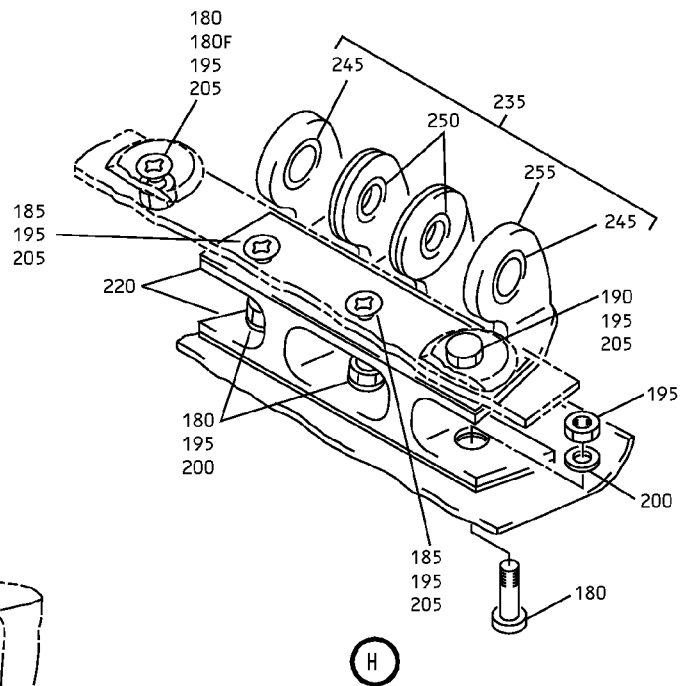
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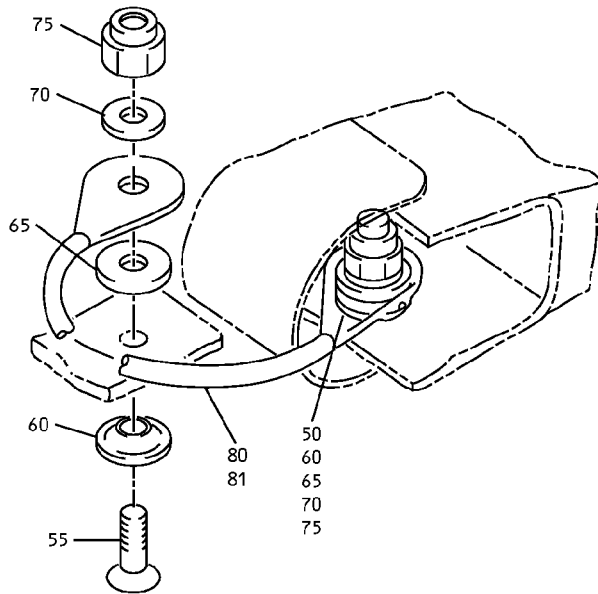
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H



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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-1	65C25796-19									A	RF
-1A	65C25796-1									C	RF
-1B	65C25796-27									E	RF
-1C	65C25796-29									G	RF
-5	65C25796-20									B	RF
-5A	65C25796-2									D	RF
-5B	65C25796-28									F	RF
-5C	65C25796-30									H	RF
10	69-63391-5									A-D	4
-10A	BACB30LJ4DSU21									C, D	4
10F	69-63391-5									E, F	1
10G	69-63391-6									E, F	3
10H	69-63391-7									E-H	1
10I	69-63391-8									E-H	3
15	BACW10BP4CT									A-D	4
-15A	BACW10BP4CTU									E-H	4
20	BACB28AK04-040									A-D	6
-20A	BACB28AK04-044									E-H	6
21	BACB28AK04-033									C, D	3
-21F	BACW10BP4PTU									E-H	1
25	BACB28AK04-045									A-D	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-25A	BACB28AK04-037		.	BUSHING						E-H	1
26	BACB28AK04-045		.	BUSHING (PRE SB 737-55-1035)						C, D	4
30	BACB28AK04-012		.	BUSHING (POST SB 737-55-1035)						A-D	1
-30A	BACB28AK04-012		.	BUSHING (PRE SB 737-55-1035)						C, D	1
35	BACW10BP4PT		.	WASHER (PRE SB 737-55-1035)						C, D	4
35A	BACW10BP4PT		.	WASHER (POST SB 737-55-1035)						A-D	3
35B	BACW10BP4PTU		.	WASHER						E-H	3
36	65C25796-23		.	WASHER (POST SB 737-55-1035)						A-D	1
-36A	BACW10BP4PTU		.	WASHER						E-H	1
38	AN960C416		.	WASHER (PRE ALERT SB 737-55A1070)							AR
-38A	AN960C416L		.	WASHER (OPT ITEM 38) (PRE ALERT SB 737-55A1070)							AR
-38B	NAS1149C0432R		.	WASHER (POST SB 737-55A1070R2)							AR
40	BACN10JD4ASU		.	NUT (PRE SB 737-55A1070)							4
-40A	BACN11N4CS		.	NUT-SELF LOCKING CASTELLATED (POST SB 737-55A1070)							AR
42	MS24665-153		.	PIN-COTTER (PRE SB 737-55A1070R2)							4
-42A	BACP18BC02A06P		.	PIN-COTTER (POST SB 737-55A1070R2)							AR
45	BACNJP08BCM			DELETED							
50	BACB30VF3R3		.	BOLT							2
-50A	BACB30LL3-3		.	BOLT (OPT ITEM 50)							2
55	BACB30VF3R4		.	BOLT							2
-55A	BACB30LL3-4		.	BOLT (OPT ITEM 55)							2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-56	BACB30LL3-3							DELETED			
-58	BACB30LL3-4							DELETED			
60	BACW10DM6P							. WASHER			4
65	BACW10BP3APU							. WASHER			4
70	BACW10BP3NAPU							. WASHER			4
75	BACN10JC3C							. NUT			4
80	BACJ40A20-5							. JUMPER-BONDING			1
81	BACJ40A20-6							. JUMPER-BONDING			1
85	BACJ40A20-6							DELETED			
-86	65C25796-9							. TAB INSTL-ELEV CONT	A		1
-86A	65C25796-9							. TAB INSTL-ELEV CONT (LIMITED USAGE)	C		1
-86B	65C25796-11							. . TAB ASSY-ELEV, LH (LIMITED USAGE) (USED ON ITEMS 86,86A)			1
-86C	65C25796-13							. . TAB ASSY-ELEV, LH (LIMITED USAGE) (USED ON ITEMS 86,86A)			1
-86D	65C25796-15							. . TAB ASSY-ELEV, LH (LIMITED USAGE) (USED ON ITEMS 86,86A)			1
-86E	65C25796-17							. . TAB ASSY-ELEV, LH (LIMITED USAGE) (USED ON ITEMS 86,86A)			1
-86F	65C25796-11							DELETED			
-86G	65C25796-13							DELETED			
-86H	65C25796-15							DELETED			
-87A	65C25796-10							. TAB INSTL-ELEV CONT, RH	B		1
-87B	65C25796-10							. TAB INSTL-ELEV CONT, RH (LIMITED USAGE)	D		1
-88A	65C25796-21							. TAB INSTL-ELEV CONT, LH (LIMITED USAGE)	C		1
-88B	65C25796-22							. TAB INSTL-ELEV CONT, RH (LIMITED USAGE)	D		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-			ATTACHING PARTS								
89	69-63391-5		.	.	BOLT	(USED ON ITEMS 88A,88B)					4
-89A	BACB28AK04-040		.	.	BUSHING	(USED ON ITEMS 88A,88B)					6
-89B	65C25796-23		.	.	WASHER	(USED ON ITEMS 86A,86B)					1
			----- * -----								
90	65C26384-11		.	TAB ASSY-ELEV, LH	(LIMITED USAGE) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049R1)					A	1
-90A	65C26384-1		.	TAB ASSY-ELEV, LH	(LIMITED USAGE) (REPLACED BY ITEM 90J) (PRE SB 737-55-1035) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049R1)					C	1
-90B	65C26384-13		.	TAB ASSY-ELEV, LH	(LIMITED USAGE) (REPLACED BY ITEM 90J) (PRE SB 737-55-1035) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049R1)					C	1
-90C	65C26384-5		.	TAB ASSY-ELEV, LH	(LIMITED USAGE) (REPLACED BY ITEM 90J) (PRE SB 737-55-1035) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049R1)					C	1
-90D	65C26384-7		.	TAB ASSY-ELEV, LH	(LIMITED USAGE) (REPLACED BY ITEM 90J) (PRE SB 737-55-1035) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049R1)					C	1
90J	65C26384-11		.	TAB ASSY-ELEV, LH	(REPLACES ITEMS 90A THRU 90D) (LIMITED USAGE) (POST SB 737-55-1035)					A, C	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
4- 90K	65C26384-15		.	TAB	ASSY-ELEV, LH						A, C	1
					(LIMITED USAGE)							
					(POST SB 737-55-1040R4)							
					(PRE SB 737-55-1049R1)							
-90L	65C26384-19		.	TAB	ASSY-LH						A	1
					(LIMITED USAGE)							
					(PRE SB 737-55-1049R1)							
-90M	65C26384-23		.	TAB	ASSY-LH						A	1
					(LIMITED USAGE)							
					(PRE SB 737-55-1049R1)							
-90N	65C26384-25		.	TAB	ASSY-ELEV, LH						A	1
					(LIMITED USAGE)							
					(PRE SB 737-55-1049R1)							
-90P	65C26384-27		.	TAB	ASSY-ELEV, LH						E, G	1
					(POST SB 737-55-1049R1)							
-95	65C26384-12		.	TAB	ASSY-ELEV, RH						B	1
					(LIMITED USAGE)							
					(REPLACES ITEMS 95A THRU 95D)							
					(POST SB 737-55-1035)							
					(PRE SB 737-55-1040R4)							
					(PRE SB 737-55-1049R1)							
-95A	65C26384-2		.	TAB	ASSY-ELEV, RH						D	1
					(LIMITED USAGE)							
					(REPLACED BY ITEM 95J)							
					(PRE SB 737-55-1035)							
					(PRE SB 737-55-1040R4)							
					(PRE SB 737-55-1049R1)							
-95B	65C26384-14		.	TAB	ASSY-ELEV, RH						D	1
					(LIMITED USAGE)							
					(REPLACED BY ITEM 95J)							
					(PRE SB 737-55-1035)							
					(PRE SB 737-55-1040R4)							
					(PRE SB 737-55-1049R1)							
-95C	65C26384-6		.	TAB	ASSY-ELEV, RH						D	1
					(LIMITED USAGE)							
					(REPLACED BY ITEM 95J)							
					(PRE SB 737-55-1035)							
					(PRE SB 737-55-1040R4)							
					(PRE SB 737-55-1049R1)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
4-					
-95D	65C26384-8		. TAB ASSY-ELEV, RH (LIMITED USAGE) (REPLACED BY ITEM 95J) (PRE SB 737-55-1040R4) (PRE SB 737-55-1035)	D	1
-95E	65C25796-12		. . TAB ASSY-ELEV, RH (USED ON ITEMS 87A,87B) (LIMITED USAGE)		1
-95F	65C25796-14		. . TAB ASSY-ELEV, RH (USED ON ITEMS 87A,87B) (LIMITED USAGE)		1
-95G	65C25796-16		. . TAB ASSY-ELEV, RH (USED ON ITEMS 87A,87B) (LIMITED USAGE)		1
-95H	65C25796-18		. . TAB ASSY-ELEV, RH (USED ON ITEMS 87A,87B) (LIMITED USAGE)		1
-95J	65C26384-12		. TAB ASSY-ELEV, RH (LIMITED USAGE) (REPLACES ITEMS 95A THRU 95D) (POST SB 737-55-1035) (PRE SB 737-55-1049R1)	B, D	1
-95K	65C26384-16		. TAB ASSY-ELEV, RH (LIMITED USAGE) (POST SB 737-55-1040R4)	B, D	1
-95L	65C26384-20		. TAB ASSY-RH (LIMITED USAGE) (PRE SB 737-55-1049R1)	B	1
-95M	65C26384-24		. TAB ASSY-RH (LIMITED USAGE) (PRE SB 737-55-1049R1)	B	1
-95N	65C26384-26		. TAB ASSY-ELEV, RH (LIMITED USAGE)	B	1
-95P	65C25796-18		DELETED		
-95Q	65C26384-28		. TAB ASSY-ELEV, RH (POST SB 737-55-1049R1)	F, H	1
100	BAC27DWG26		. . DECAL		1
105	65C25799-3		. . FITTING ASSY-HINGE (PRE SB 737-55-1040R4) (PRE SB 737-55-1049) (USED ON ITEMS 90,95)		2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
106	65C25799-1							. . FITTING ASSY-HINGE (PRE SB 737-55-1035) (PRE SB 737-55-1040R4) (PRE SB 737-55-1049) (USED ON ITEMS 90A THRU 90D,95A THRU 95D)			2
107	65C25799-5							. . FITTING ASSY-HINGE (OPT ITEM 107A) (PRE SB 737-55-1049) (USED ON ITEMS 90K,95K)			2
-107A	65C25799-7							. . FITTING ASSY-HINGE (POST SB 737-55-1040R4) (PRE SB 737-55-1049) (USED ON ITEMS 90K,90L,90M,95K,95L,95M)			2
-107B	65C25799-9							. . FITTING ASSY-HINGE (PRE SB 737-55-1049) (USED ON ITEMS 90N,95N)			2
110	69-74582-1							. . . PIN (USED ON ITEM 105)			1
111	BACP18T4-21							. . . PIN (USED ON ITEM 106)			1
-111A	69-74582-2							. . . PIN (USED ON ITEM 107)			1
-111B	69-74582-3							. . . PIN (USED ON ITEMS 107A,107B)			1
112	AN960XC416L							. . . WASHER (USED ON ITEM 106)			2
113	MS24665-153							. . . COTTER PIN (USED ON ITEM 106)			1
114	BACNJP08BCM							. . . NUTPLATE (USED ON ITEMS 105,106,107)			4
-114A	BACN10JP08BCD							. . . NUTPLATE (USED ON ITEMS 107A)			4
115	65C26385-3							. . . PLATE ASSY BEARING (USED ON ITEM 105)			1
116	65C26385-1							. . . PLATE ASSY-BEARING (USED ON ITEM 106)			1
117	65C32050-1							. . . PLATE ASSY-BEARING (USED ON ITEM 107,107A)			1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-117A	65C32050-3								... PLATE ASSY-BEARING (USED ON ITEM 107B)		1
118	BACR15DR3AC2								... RIVET (USED ON ITEMS 105,106,107,107A)		8
120	SA4-14A							 BEARING (V77896)		1
125	69-72039-37							 BUSHING (USED ON ITEM 115)		2
126	BACB28AM04~ B010A							 BUSHING (USED ON ITEM 116)		2
127	69-72039-39							 BUSHING (USED ON ITEM 117,117A)		2
130	65C26385-2							 PLATE-BEARING (USED ON ITEMS 115,116)		1
131	65C32050-2							 PLATE-BEARING (USED ON ITEM 117,117A)		1
132	NAS516-1A							 FITTING-LUBE (USED ON ITEM 117,117A)		1
135	65C26386-8								... FITTING ASSY-HINGE (USED ON ITEM 105)		1
136	65C26386-4								... FITTING ASSY-HINGE (OPT ITEM 137) (USED ON ITEM 106)		1
137	65C26386-1								... FITTING ASSY-HINGE (OPT ITEM 136) (USED ON ITEM 106)		1
138	65C32051-1								... FITTING ASSY-HINGE (USED ON ITEM 107)		1
-138A	65C32051-4								... FITTING ASSY-HINGE (USED ON ITEM 107B)		1
-138B	65C32051-6								... FITTING ASSY-HINGE (USED ON ITEM 107B)		1
140	BACB28U4B022							 BUSHING (USED ON ITEMS 135,136)		1
141	BACB10L22A							 BEARING (USED ON ITEM 137)		1
142	69-72039-37							 BUSHING-FLANGED (USED ON ITEM 138,138A,138B)		2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
4-												
145	65C26386-6											1
146	65C26386-2											1
147	65C32051-2											1
-147A	65C32051-5											1
150	65C26384-4											8
-150A	65C26384-4											8
-150B	BACS40V010C030											8
-150C	BACS40V010C030											8
-150D	BACS40V010C030											8
155	BACB30VF08K3											16
155A	BACB30VF08K3											DELETED
160	BACW10DM5P											16
160A	BACW10DM5P											DELETED
161	NAS1149FN832P											8

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4- 161A	AN960C8		.	.	WASHER (USED ON ITEMS 90N,95N) (USED WITH ITEM 107B) (OPT ITEM 161,161B) (PRE SB 737-55-1049) (PRE SB 737-55-1049R1)						8
161B	AN960C8L		.	.	WASHER (USED ON ITEMS 90N,95N) (USED WITH ITEM 107B) (OPT ITEM 161,161A) (PRE SB 737-55-1049) (PRE SB 737-55-1049R1)						8
161C	NAS1149FN832P		.	.	WASHER (USED ON ITEMS 90P,95Q) (USED WITH ITEMS 165D, 165E) (OPT ITEM 161D,161E) (POST SB 737-55-1049) (POST SB 737-55-1049R1)						16
161D	AN960-8		.	.	WASHER (USED ON ITEMS 90P,95Q) (USED WITH ITEMS 165D, 165E) (OPT ITEM 161C,161E) (POST SB 737-55-1049) (POST SB 737-55-1049R1)						16
161E	AN960-8L		.	.	WASHER (USED ON ITEMS 90P,95Q) (USED WITH ITEMS 165D, 165E) (OPT ITEM 161C,161D) (POST SB 737-55-1049) (POST SB 737-55-1049R1)						16
162	BACN10YR08CD		.	.	NUT (USED ON ITEMS 90N,95N) (USED WITH ITEM 107B) (PRE SB 737-55-1049)						8
162A	BACN10YR08CD		.	.	NUT (USED ON ITEMS 90P,95Q) (USED WITH ITEM 165D) (POST SB 737-55-1049) (POST SB 737-55-1049R1)						16
162B	BACN10YR08CD		.	.	NUT (USED ON ITEMS 90P,95Q) (USED WITH ITEM 165D) (POST SB 737-55-1049) (POST SB 737-55-1049R1)						16

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4- 165	65C25798-4		2
165A	65C25798-4		2
165B	65C25798-6		2
165C	65C25798-8		2
165D	65C25798-11		4
165E	65C25798-12		8
165F	65C25798-14		2
170	SA4-14A		1
170A	DAS4-14A1-508		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
171	NAS516-1A								. . . FITTING, LUBE (USED ON ITEMS 165B,165C THRU 165F) (V77896)		1
175	65C25798-5								. . . FITTING (USED ON ITEMS 165,165A)		1
176	65C25798-7								. . . FITTING (USED ON ITEM 165B)		1
177	65C25798-9								. . . FITTING (USED ON ITEM 165C)		1
177A	65C25798-10								. . . FITTING (USED ON ITEM 165D)		1
177B	65C25798-13								. . . FITTING (USED ON ITEM 165E)		1
177C	65C25798-15								. . . FITTING (USED ON ITEM 165F)		1
178	BACN10JP08ACM								. . . NUTPLATE (USED ON ITEMS 165 THRU 165C,165F)		4
179	BACR15DR3AC2								. . . RIVET (USED ON ITEMS 165 THRU 165C,165F)		8
180	BACB30NN3K3								. . BOLT (USED ON ITEMS 90 THRU 90C,95A THRU 95C)		4
-180A	BACB30NN3K3								. . BOLT (USED ON ITEMS 90,90D,90J THRU 90P,95,95D,95E,95J THRU 95N,95Q)		3
-180F	BACB30NT3K3								. . BOLT (USED ON ITEMS 90L THRU 90P,95L THRU 95N,95Q)		1
185	BACB30NN3K6								. . BOLT		2
190	BACB30NM3K8								. . BOLT (USED ON ITEMS 90 THRU 90D,90K,95 THRU 95D,95K)		1
-190A	BACB30NT3K8								. . BOLT (USED ON ITEMS 90L THRU 90P,95L THRU 95N,95Q)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
195	BACN10JC3		. .	NUT							6
				(USED ON ITEMS 90 THRU 90D,90K,95 THRU 95D,95K)							
-195A	BACN10YR3CD		. .	NUT							6
				(USED ON ITEMS 90L THRU 90P,95L THRU 95N,95Q)							
200	BACW10DM6		. .	WASHER							3
				(USED ON ITEMS 90 THRU 90D,90K,95 THRU 95D,95K)							
-200A	BACW10DM6P		. .	WASHER							3
				(USED ON ITEMS 90L,95L)							
-200B	BACW10UC10P		. .	WASHER							3
				(USED ON ITEMS 90M THRU 90P,95M,95N,95Q)							
205	BACW10BN3UP		. .	WASHER							4
				(USED ON ITEMS 90 THRU 90D,90K,95 THRU 95D,95K)							
-205A	BACW10BP3NDP		. .	WASHER							4
				(USED ON ITEMS 90L THRU 90P,95L THRU 95N,95Q)							
210	65-53720-11		. .	FAIRING-MAST FITTING, LH							1
				(USED ON ITEMS 90,90B,90D,90J,90K)							
-210A	65-53720-9		. .	FAIRING-MAST FITTING, LH							1
				(USED ON ITEMS 90A,90C)							
-210B	65-53720-15		. .	FAIRING-MAST FITTING, LH							1
				(USED ON ITEM 90P)							
-210C	65-53720-11		. .	FAIRING							1
				(USED ON ITEMS 90L,90M,90N)							
-210D	65-53720-15		. .	FAIRING							1
				(USED ON ITEMS 90L,90M,90N) (OPT ITEM 210C)							
-215	65-53720-12		. .	FAIRING-MAST FITTING, RH							1
				(USED ON ITEMS 95,95B,95D,95J,95K)							
-215A	65-53720-10		. .	FAIRING-MAST FITTING, RH							1
				(USED ON ITEMS 95A,95C)							
-215B	65-53720-16		. .	FAIRING-MAST FITTING, RH							1
				(USED ON ITEM 95Q)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-215C	65-53720-12							. . FAIRING (USED ON ITEMS 95L,95M,95N)			1
-215D	65-53720-16							. . FAIRING (USED ON ITEMS 95L,95M,95N) (OPT ITEM 215C)			1
220	65C26384-3							. . SHIM (USED ON ITEMS 90 THRU 90D,90K,95 THRU 95D,95K)			2
-220A	BACS40V011D040							. . SHIM (USED ON ITEMS 90L THRU 90P,95L THRU 95N,95Q)			2
225	65C25797-15							. . TAB ASSY-BOND, (LH) (USED ON ITEMS 90,90K)			1
-225A	65C25797-1							. . TAB ASSY-ASSY BOND, (LH) (USED ON ITEMS 90A THRU 90D)			1
-225B	65C25797-17							. . TAB ASSY-ASSY BOND, (LH) (USED ON ITEMS 90L THRU 90P)			1
-230	65C25797-16							. . TAB ASSY-BOND, RH (USED ON ITEMS 95,95K)			1
-230A	65C25797-2							. . TAB ASSY-ASSY BOND, (RH) (USED ON ITEMS 95A THRU 95D)			1
-230B	65C25797-18							. . TAB ASSY-BOND, (RH) (USED ON ITEMS 95L THRU 95N,95Q)			1
235	65C28142-1							. . FITTING ASSY-MAST, (LH) (USED ON ITEMS 90,90B THRU 90D,90K)			1
-235A	69-40240-503							. . FITTING ASSY-MAST, (LH) (USED ON ITEMS 90A)			1
-235B	65C34066-1							. . FITTING ASSY-MAST, (LH) (LIMITED USAGE) (USED ON ITEMS 90,90B,90C,90D,90K)			1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
4-											
-235C	65C34066-1										
-240	65C28142-2										
-240A	69-40240-504										
-240B	65C34066-2										
-240C	65C34066-2										
245	69-59227-3										
-245A	69-59227-4										
250	69-56340-501										
-250A	69-56340-2										
255	65C28142-3								A		
-255A	69-40240-3										
-255B	65C34066-3										
-260	65C28142-4								B		
-260A	69-40240-4										

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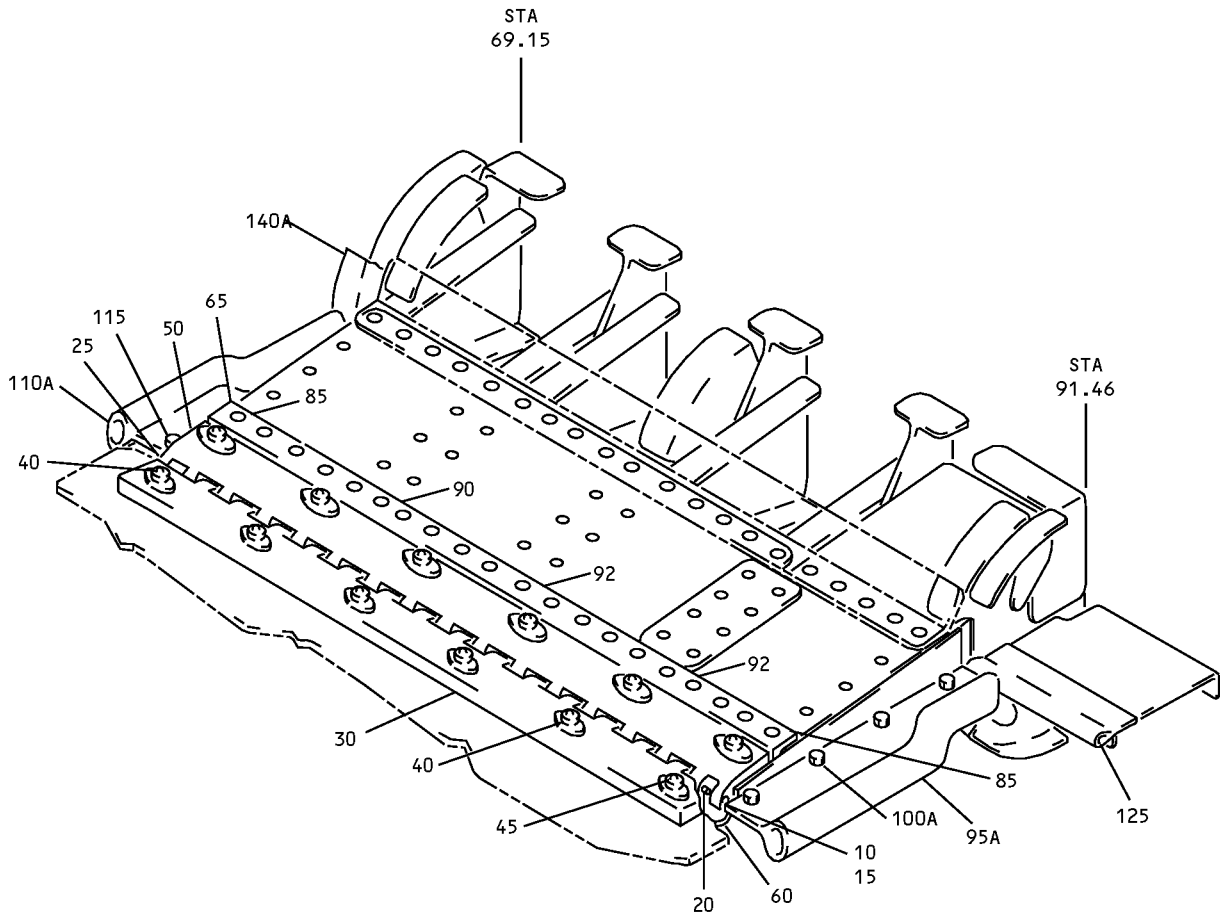
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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
4- -260B	65C34066-4											1

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Bay No. 2 Balance Panel Installation
IPL Figure 5

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
5-											
-1	65C25783-15									A	RF
-1A	65C25783-3									C	RF
-1B	65C25783-11									E	RF
-1C	65C25783-31									G	RF
-5	65C25783-16									B	RF
-5A	65C25783-4									D	RF
-5B	65C25783-12									F	RF
-5C	65C25783-32									H	RF
10	MS20392-2C11										1
15	NAS620-10L										1
20	MS20253-4-2122										1
25	MS20392-2C7										1
30	65C26393-5									A, C, E, G	1
35	65C26393-6									B, D, F, H	1
40	NAS6603-6										1
45	NAS6604-6										5
50	65C26393-3										1
-55	NAS6604-6										6
60	69-72899-1										1
65	65-53708-1									A-D, G, H	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
5-											
65A	65C27274-1									E, F	1
-70	BACB30LU3-5X									E, F	16
-75	AN960PD10L										16
-80	BACN10JC3										16
85	65C25783-8										2
90	65C25783-9										1
92	65C25783-8									C, D	2
92A	65C25783-9									A, B, E-H	2
95	10-60754-695										
95A	65C25783-20									A-F	1
95B	65C25783-35									G, H	1
95C	65C25783-37									G, H	1
95D	65C36621-2									G, H	1
-100	NAS6603-4										
100A	NAS6603-5										4
-105	BACW10BP3APU										
110	10-60754-696										
110A	65C25783-21									A-D	1
110B	65C25783-30									A-F	1
110C	65C25783-36									G, H	1
110D	65C25783-38									G, H	1
110E	65C36621-1									G, H	1

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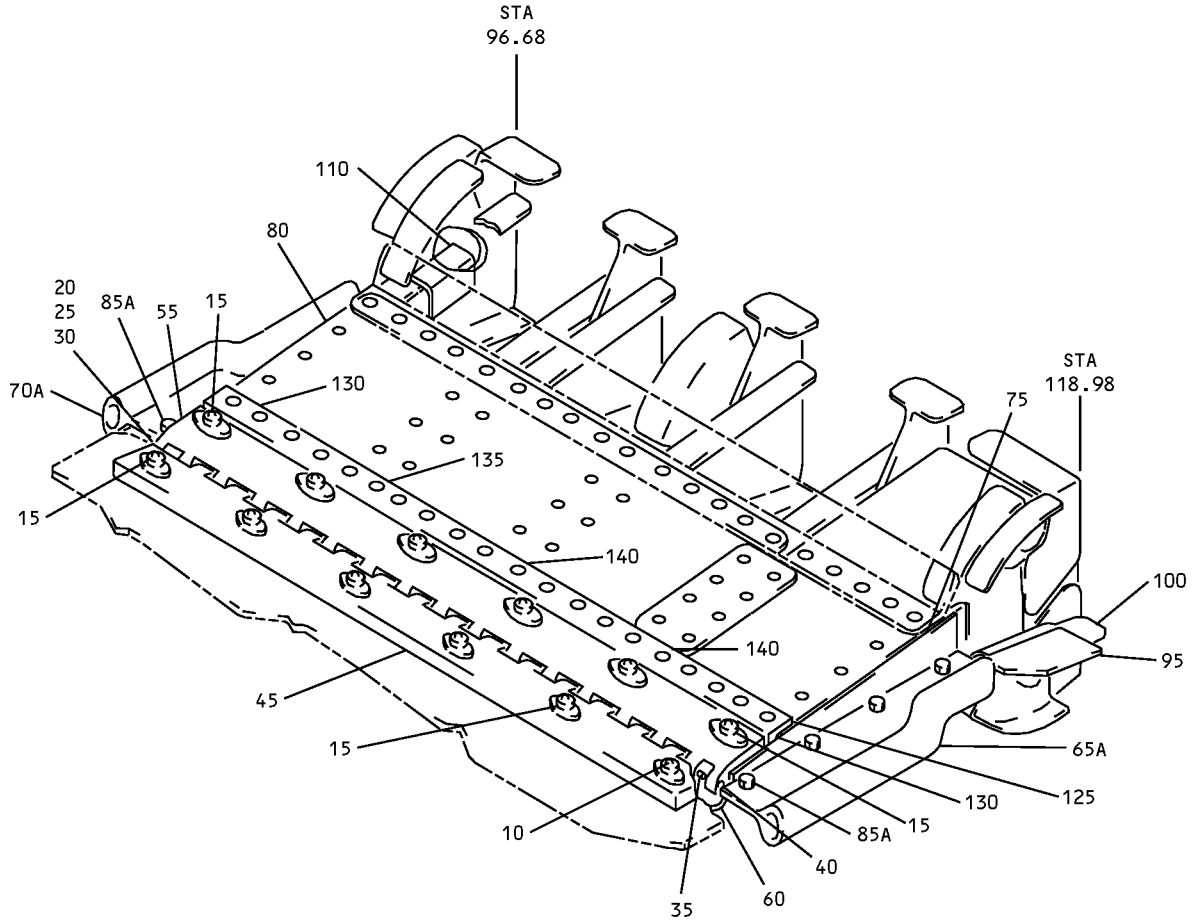


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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
5-			ATTACHING PARTS								
115	NAS6603-4		.	B	O	L	T			G, H	4
115A	NAS6603-5		.	B	O	L	T			A-F	4
120	BACW10BP3APU		DELETED								
			-----*								
125	65-53728-6		.	S	E	A	L				1
			ATTACHING PARTS								
-130	NAS6603-3		.	B	O	L	T				3
-135	BACW10BP3APU		DELETED								
-135A	BACW10P43AL		.	W	A	S	H	E	R		3
			-----*								
140	10-60754-690		DELETED								
140A	65C25783-19		.	S	E	A	L			A-F	1
140B	65C25783-39		.	S	E	A	L			G, H	1
			(LIMITED USAGE)								
140C	65C36621-7		.	S	E	A	L			G, H	1
			(LIMITED USAGE)								
			ATTACHING PARTS								
-145	NAS6603-4		.	B	O	L	T			A-F	4
-145A	NAS6603-2		.	B	O	L	T			G, H	4
-150	BACW10BP3APU		DELETED								
			-----*								

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Bay No. 3 Balance Panel Installation
IPL Figure 6

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
6-											
-1	65C25784-15									A	RF
-1A	65C25784-3									C	RF
-1B	65C25784-11									E	RF
-1C	65C25784-29									G	RF
-5	65C25784-16									B	RF
-5A	65C25784-4									D	RF
-5B	65C25784-12									F	RF
-5C	65C25784-30									H	RF
10	NAS6603-6										1
15	NAS6604-6										11
20	MS24665-132										1
25	NAS620-10L										1
30	MS20392-2C11										1
35	MS20253-4-2122										1
40	MS20392-2C7										1
45	65C26393-5									A, C, E, G	5
-50	65C26393-6									B, D, F, H	1
-55	65C26393-3										1
60	69-72899-1										1
65	10-60754-693										DELETED
65A	65C25784-20									A-F	1
65B	65C25784-32									G, H	1
											(OPT ITEM 65C) (LIMITED USAGE)

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
6-											
65C	65C25784-34							. SEAL-OUTBD (OPT ITEM 65B) (LIMITED USAGE)	G, H	1	
65D	65C36621-4							. SEAL-OUTBD (LIMITED USAGE)	G, H	1	
70	10-60754-694							DELETED			
70A	65C25784-21							. SEAL-INBD	A-F	1	
70B	65C25784-31							. SEAL-INBD (OPT ITEM 70C) (LIMITED USAGE)	G, H	1	
70C	65C25784-33							. SEAL-INBD (OPT ITEM 70B) (LIMITED USAGE)	G, H	1	
70D	65C36621-3							. SEAL-INBD ATTACHING PARTS	G, H	1	
75	65C25784-7							. RETAINER-SEAL, OUTBD	A, C, E, G	1	
80	65C25784-6							. RETAINER-SEAL, INBD	A, C, E, G	1	
85	NAS6603-4							DELETED			
85A	NAS6603-5							. BOLT		8	
90	BACW10BP3APU							DELETED			
								----- * -----			
95	65-53728-4							. SEAL ATTACHING PARTS		1	
100	65C27460-2							. RETAINER-SEAL, OUTBD (LH)	A, C, E, G	1	
-105	65C27460-1							. RETAINER-SEAL, OUTBD (RH)	B, D, F, H	1	
110	65C27460-5							. RETAINER-SEAL, INBD		1	
-115	NAS6603-4							. BOLT		2	
-120	BACW10BP3APU							DELETED			
-120A	BACW10P43AL							. WASHER		2	
								----- * -----			
125	65-53708-1							. FITTING-HINGE	A-D, G, H	1	

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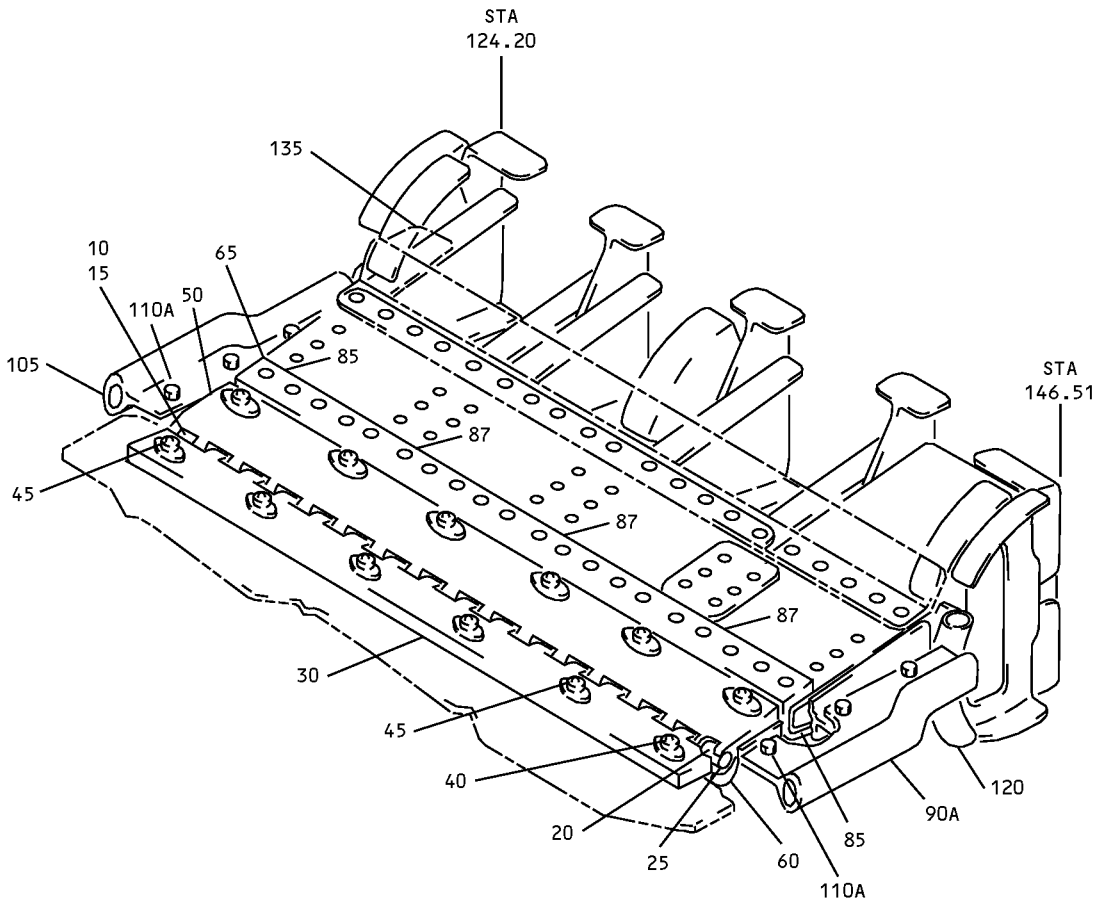
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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
6-											
125A	65C27274-1									E, F	1
130	65C25784-9										2
135	65C25784-8										1
140	65C25784-9									C, D	2
140A	65C27284-8									A, B, E-H	2

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Bay No. 4 Balance Panel Installation
IPL Figure 7

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
7-											
-1	65C25785-15									A	RF
-1A	65C25785-3									C	RF
-1B	65C25785-11									E	RF
-1C	65C25785-29									G	RF
-5	65C25785-16									B	RF
-5A	65C25785-4									D	RF
-5B	65C25785-12									F	RF
-5C	65C25785-28									H	RF
10	MS20392-2C11										1
15	NAS620-10L										1
20	MS20253-4-2122										1
25	MS20392-2C7										1
30	65C26393-5									A, C, E, G	1
35	65C26393-6									B, D, F, H	1
40	NAS6603-6										1
45	NAS6604-6										5
50	65C26393-3										1
-55	NAS6604-6										6
60	69-72899-1										1
65	65-53708-1									A-D, G, H	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
7-			ATTACHING PARTS								
-70	BACB30LU3-4		.	BOLT						C, D	10
-70A	BACB30LU3-5		.	BOLT						A, B, E-H	10
-75	AN960PD10L		.	WASHER						C, D	10
-75A	AN960PD10L		.	WASHER						A, B, E-H	10
-80	BACN10JC3		.	NUT						C, D	10
-80A	BACN10JC3		.	NUT						A, B, E-H	10
			----- * -----								
85	65C25785-9		.	SHIM							2
87	65C25785-9		.	SHIM						C, D	3
87A	65C25785-8		.	SHIM						A, B, E-H	3
90	10-60754-691		DELETED								
90A	65C25785-19		.	SEAL						A, B	1
				(LIMITED USAGE)							
90B	65C25785-26		.	SEAL						A, B, G, H	1
				(LIMITED USAGE)							
90C	65C25785-19		.	SEAL						C-F	1
90D	65C25785-26		.	SEAL						G, H	1
				(LIMITED USAGE)							
				(OPT ITEM 90E)							
90E	65C25785-33		.	SEAL						G, H	1
				(LIMITED USAGE)							
				(OPT ITEM 90D)							
90F	65C36621-6		.	SEAL						G, H	1
				(LIMITED USAGE)							
			ATTACHING PARTS								
-95	NAS6603-4		DELETED								
-95A	NAS6603-5		.	BOLT							3
-100	BACW10BP3APU		.	WASHER						A-F	3
			----- * -----								
105	10-60754-689		DELETED								
105A	65C25785-20		.	SEAL						A, B	1
				(LIMITED USAGE)							

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COMPONENT MAINTENANCE MANUAL

FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
7-											
105B	65C25785-27		.	SEAL						A, B, G, H	1
				(LIMITED USAGE)							
105C	65C25785-20		.	SEAL						C-F	1
105D	65C25785-32		.	SEAL						G, H	1
				(LIMITED USAGE)							
				(OPT ITEM 105E)							
105E	65C25785-34		.	SEAL						G, H	1
				(LIMITED USAGE)							
				(OPT ITEM 105D)							
105F	65C36621-5		.	SEAL						G, H	1
				(LIMITED USAGE)							
				ATTACHING PARTS							
-110	NAS6603-4			DELETED							
110A	NAS6603-5		.	BOLT							7
-115	BACW10BP3APU		.	WASHER						A-F	4
				-----*							
120	10-60754-689		.	SEAL							1
				ATTACHING PARTS							
-125	NAS6603-4		.	BOLT							3
-130	BACW10BP3APU			DELETED							
-130A	BACW10P43AL		.	WASHER							3
				-----*							
135	65-53728-4		.	SEAL							1
				ATTACHING PARTS							
-140	NAS6603-4		.	BOLT							2
-145	BACW10BP3APU			DELETED							
				-----*							

-Item not Illustrated

55-20-16

ILLUSTRATED PARTS LIST

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