



COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

INBOARD FLAP OUTBOARD TRACK FAIRING AND TAILCONE ASSEMBLY

PART NUMBER

**65C26300–100SP, –105SP, –106SP, –107SP, –108SP,
–111SP, –112SP, –3SP, –4SP, –57SP, –58SP, –59SP,
–5SP, –60SP, –63SP, –64SP, –69SP, –6SP, –70SP,**

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COMPONENT MAINTENANCE MANUAL

Revision No. 19
Jul 01/2009

To: All holders of INBOARD FLAP OUTBOARD TRACK FAIRING AND TAILCONE ASSEMBLY 57-26-45.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

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TRANSMITTAL LETTER

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Location of Change

Description of Change

NO HIGHLIGHTS

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		PRR 33555-1	JUN 05/84
		PRR 33761	MAR 05/85
		PRR 33432	MAR 05/85
		PRR 33617	MAR 05/85
		PRR 33750-5	MAR 05/86
		PRR 33750-15	MAR 05/86
		PRR 33885	MAR 05/86
SB 57-1150		PRR 33890-6	MAR 05/86
		PRR 33750-11	DEC 05/86
		PRR 33875-3	DEC 05/86
		PRR 33943	DEC 05/86
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All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed	
Number	Date	Date	Initials

Revision		Filed	
Number	Date	Date	Initials

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Temporary Revision		Inserted		Removed		Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials

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COMPONENT MAINTENANCE MANUAL

INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

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INTRODUCTION

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COMPONENT MAINTENANCE MANUAL

INBOARD FLAP OUTBOARD TRACK FAIRING AND TAILCONE ASSEMBLY - DESCRIPTION AND OPERATION

1. Description

- A. The inboard flap outboard track fairing and tailcone assemblies consist of aluminum frames enclosed by a honeycomb/fiberglass shell that provides smooth airflow around the inboard flap outboard track.

2. Leading Particulars (approximate)

- A. Length – 99 inches
- B. Width – 16 inches
- C. Thickness – 24 inches
- D. Weight – 83 pounds

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DESCRIPTION AND OPERATION

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TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

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TESTING AND FAULT ISOLATION

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DISASSEMBLY

1. General

- A. This procedure has the data to disassemble the inboard flap outboard track tailcone assembly.
- B. Disassemble this component only as necessary to complete fault isolation, determine serviceability of parts, do the necessary repairs and return the unit to a serviceable condition.
- C. Refer to IPL Figure 1 and IPL Figure 2 for item numbers.

2. Disassembly

- A. Procedure
 - (1) Remove springs (500, 501, IPL Figure 1) before attempting removal of door assy (525, 530, IPL Figure 1), hinge fittings (708, 725, 750, 755, IPL Figure 1) and bolt (475, 476, IPL Figure 1).

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DISASSEMBLY

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CLEANING

1. General

- A. This procedure has the data to clean the inboard flap outboard track tailcone assembly (IPL Figure 1 and IPL Figure 2).
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.

2. CLEANING

A. References

Reference	Title
SOPM 20-30-03	GENERAL CLEANING PROCEDURES

B. Procedure

CAUTION: DO NOT VAPOR DEGREASE EPOXY BONDED STRUCTURES WITH CHLORINATED CLEANING AGENTS SUCH AS METHYLENE CHLORIDE, TRICHLOROETHYLENE, AND TRICHLOROETHANE. CHLORINATED CLEANING AGENTS WILL CAUSE DAMAGE TO EPOXY BONDED STRUCTURES. 1,1,1-TRICHLOROETHANE IS ONE OF THE SOLVENTS ALLOWED FOR CLEANING COMPOSITE COMPONENTS. DO NOT SUBMERGE PARTS IN THE SOLVENT OR ALLOW STANDING SOLVENT ON THE PARTS OR DAMAGE MAY OCCUR. USE 1,1, 1-TRICHLOROETHANE ONLY AS A WIPE SOLVENT.

CAUTION: 1, 1, 1 - TRICHLOROETHANE IS ONE OF THE SOLVENTS ALLOWED FOR CLEANING COMPOSITE COMPONENTS. DO NOT SUBMERGE PARTS IN THE SOLVENT OR ALLOW STANDING SOLVENT ON THE PARTS OR DAMAGE MAY OCCUR. USE 1, 1, 1 - TRICHLOROETHANE ONLY AS A WIPE SOLVENT.

- (1) Clean all parts in accordance with standard industry practices (SOPM 20-30-03) except for advanced composite structures.

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CLEANING

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CHECK

1. General

- A. This procedure has the data to find defects in the specific parts of the inboard flap outboard track fairing and the tailcone assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 and IPL Figure 2 for the item numbers.

2. Check

A. References

Reference	Title
SOPM 20-20-01	MAGNETIC PARTICLE INSPECTION
SOPM 20-20-02	PENETRANT METHODS OF INSPECTION

B. Procedure

- (1) Check all parts for obvious defects in accordance with standard industry practices.
- (2) Penetrant check per SOPM 20-20-02 - Forward Bulkheads (260, 265, IPL Figure 1), Supports (220, 225, IPL Figure 1), Fittings (290, 295, 340, 345, IPL Figure 1) and Spring (500, 501, IPL Figure 1).
- (3) Magnetic particle check per SOPM 20-20-01 - Struts (65, 135, IPL Figure 1).
- (4) Door Assy (525, 525A, 525C, 525E, 530, 530A, 530C, 530E, IPL Figure 1) must rotate freely when the pressure is released from springs (500, 501, IPL Figure 1)

C. Spring Check (500, IPL Figure 1)

- (1) Expand spring to 5.59 inches and check that load is 24.9 - 30.3 pounds.
- (2) Expand spring to 8.26 inches and check that load is 70.7 - 80.5 pounds.

D. Spring Check (501, IPL Figure 1)

- (1) Expand spring to 5.66 inches and check that load is 42.3 - 51.7 pounds.
- (2) Expand spring to 8.26 inches and check that load is 119.3 - 134.7 pounds.

3. Visual Check

A. References

Reference	Title
737 NDT Part 1, 51-05-01	Tap Test Inspection of Honeycomb Sandwich Structure
737 NDT Part 4, 51-00-02	Full Depth Honeycomb and Laminate Structure Inspection
737 NDT Part 9, 51-00-01	Non-Destructive Testing
737 SRM Chapter 57	Structural Repair Manual

B. Procedure

- (1) Do a check of the honeycomb structure and bonded parts for evidence of delamination, internal water, scratches, and countour defects.

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- (a) If you see delamination or impact damage when you do a visual check, do an ultrasonic inspection or a tap test to find all damage. Refer to 737 NDT Part 4, 51-00-02 or 737 NDT Part 1, 51-05-01.

NOTE: For the tap test, use a solid metal disk and tap the surface area lightly but firmly. The void areas will produce a dull sound as opposed to a sharp sound on a solid bonded area. Refer to 737 NDT Part 1, 51-05-01.
 - (b) Do a check on areas you suspect of containing water with the radiographic or thermographic method. Refer to 737 NDT Part 9, 51-00-01.
 - (c) Do a check on the edges of the panel carefully for cuts and abrasions. Delamination starts very easily from damage to an edgemember of honeycomb panel.
- (2) Check assembly for nicks, scratches and corrosion and refer to 737 SRM Chapter 57 for allowable damage data and repair procedures.

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REPAIR

1. General

- A. Instructions for repair, refinish and replacement procedures are included in separate repair procedures as follows:

Table 601:

P/N	NAME	REPAIR
65C26300 69-73102 69-77062	STRUT ASSY	1-1
65C26302 65C30937	FITTING ASSY	2-1
65C26304 65C30918	BULKHEAD ASSY	3-1
65C26306	FITTING ASSY	4-1
65-70947	DOOR ASSY	5-1
65C30479 69-73736	SUPPORT ASSY	6-1
- - -	MISCELLANEOUS PARTS REFINISH	7-1
65C26303 65C30938	FITTING ASSY	8-1

2. Dimensioning Symbols

- A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.

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REPAIR - GENERAL

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—	STRAIGHTNESS	\oplus	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)
\square	FLATNESS	\varnothing	DIAMETER
\perp	PERPENDICULARITY (OR SQUARENESS)	$s \varnothing$	SPHERICAL DIAMETER
//	PARALLELISM	R	RADIUS
\bigcirc	ROUNDNESS	SR	SPHERICAL RADIUS
\bigcirc	CYLINDRICITY	()	REFERENCE
\frown	PROFILE OF A LINE	BASIC (BSC) OR	A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE FROM WHICH PERMISSIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
\triangle	PROFILE OF A SURFACE	DIM	
\odot	CONCENTRICITY	-A-	DATUM
\equiv	SYMMETRY	\textcircled{M}	MAXIMUM MATERIAL CONDITION (MMC)
\sphericalangle	ANGULARITY	\textcircled{L}	LEAST MATERIAL CONDITION (LMC)
\nearrow	RUNOUT	\textcircled{S}	REGARDLESS OF FEATURE SIZE (RFS)
\nearrow	TOTAL RUNOUT	\textcircled{P}	PROJECTED TOLERANCE ZONE
\sqcup	COUNTERBORE OR SPOTFACE	FIM	FULL INDICATOR MOVEMENT
\sphericalangle	COUNTERSINK	TIR	TOTAL INDICATOR READING

EXAMPLES

$\boxed{-0.002}$	STRAIGHT WITHIN 0.002	$\boxed{\textcircled{\varnothing}0.0005 C}$	CONCENTRIC TO C WITHIN 0.0005 DIAMETER
$\boxed{\perp 0.002 B}$	PERPENDICULAR TO B WITHIN 0.002	$\boxed{\equiv 0.010 A}$	SYMMETRICAL WITH A WITHIN 0.010
$\boxed{\parallel 0.002 A}$	PARALLEL TO A WITHIN 0.002	$\boxed{\sphericalangle 0.005 A}$	ANGULAR TOLERANCE 0.005 WITH A
$\boxed{\bigcirc 0.002}$	ROUND WITHIN 0.002	$\boxed{\oplus \varnothing 0.002 \textcircled{S} B}$	LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
$\boxed{\bigcirc 0.010}$	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	$\boxed{\perp \varnothing 0.010 \textcircled{M} A}$	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010-INCH DIAMETER, PERPENDICULAR TO, AND EXTENDING 0.510-INCH ABOVE, DATUM A, MAXIMUM MATERIAL CONDITION
$\boxed{\frown 0.006 A}$	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM PLANE A	$\boxed{0.510 \textcircled{P}}$	
$\boxed{\triangle 0.020 A}$	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.02 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE	$\boxed{2.000}$	THEORETICALLY EXACT DIMENSION IS 2.000
		OR	
		2.000	
		BSC	
		$\boxed{0.020 A}$	
		$\boxed{A 0.020}$	

NOTE: DATUM MAY APPEAR AT EITHER SIDE OF TOLERANCE FRAME

True Position Dimensioning Symbols
Figure 601

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REPAIR - GENERAL

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STRUT ASSEMBLY-OUTBOARD - REPAIR 1-1

65C26300-45, 69-73102-5, 69-77062-1

1. General

- A. This procedure has the data necessary to replace the rod end assembly (115, IPL Figure 1).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Replacement of Rod End Assy (115, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
D50080	Lubricant - Solid Film	MIL-PRF-46010, Type 1 (Supersedes MIL-L-8937)

- B. References

Reference	Title
SOPM 20-50-08	APPLICATION OF BONDED SOLID FILM LUBRICANTS

- C. Procedure

- (1) Remove rod end assy.
- (2) Apply solid film lubricant, D50080 to threads of rod end assy, per SOPM 20-50-08.
- (3) Install rod end assy with lock rod end (105) and nut (110).

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REPAIR 1-1

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FITTING ASSEMBLY - REPAIR 2-1

65C26302-7, -8, 65C30937-1, -2

1. General

- A. This procedure has the data necessary to replace the bushing in the fitting assembly (270, 275).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to the REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (280, IPL Figure 1

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For finishing materials, refer to SOPM 20-60-02. For miscellaneous materials, refer to SOPM 20-60-04

- (1) Remove bushings (280) and sleeve (285).
- (2) Install sleeve (285) per SOPM 20-50-03 with wet sealant, A00247.
- (3) Machine sleeve (285) per REPAIR 2-1, Figure 601.
- (4) Install bushing per SOPM 20-50-03 with wet sealant, A00247.
- (5) Machine bushing bores per REPAIR 2-1, Figure 601.
- (6) Fillet seal bushing flanges with sealant, A00247.
- (7) Top coat fillet seals with gloss enamel coating, C50050.

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REPAIR 2-1

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3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-43-01	CHROMIC ACID ANODIZING
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For application of chromic acid anodizing, refer to SOPM 20-43-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Fitting (290A, 290B, 295A, 295B) - Chromic acid anodize and apply primer, C00259 (F-18.13); omit primer from bores. Apply enamel coating, C50050 (SRF-14.9812). Material: Aluminum alloy.

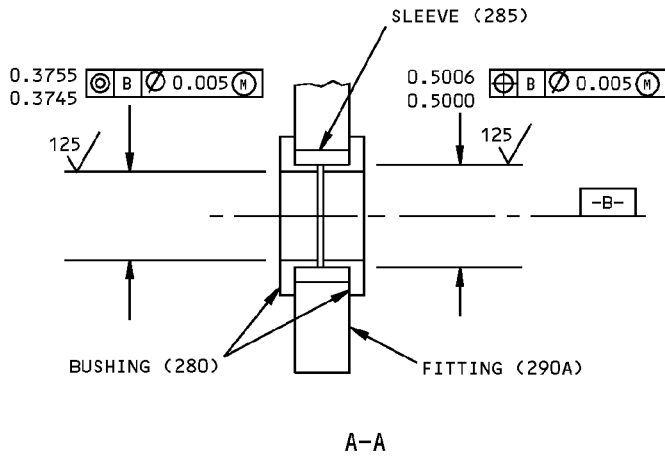
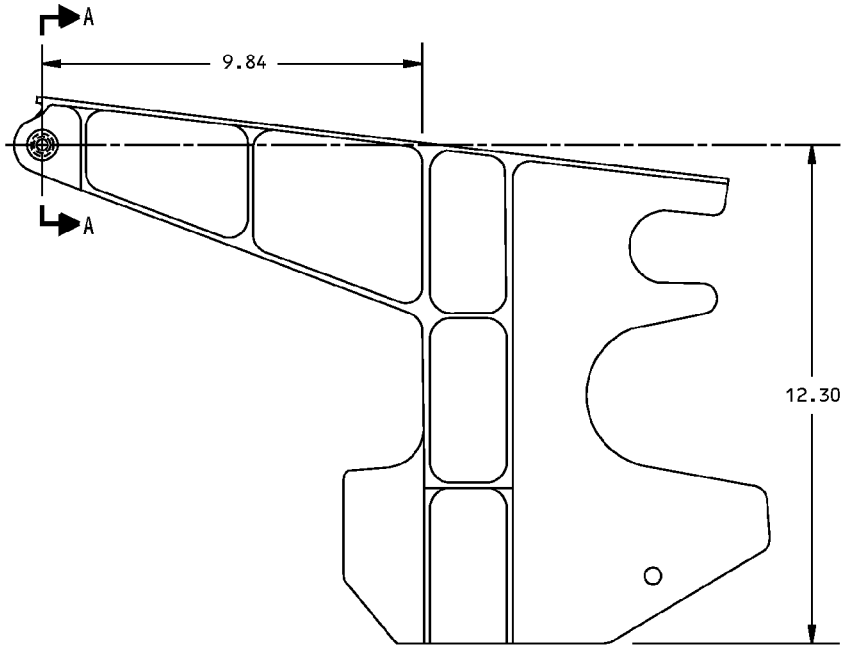
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REPAIR 2-1

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ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

65C26302-7,-8
68C30937-1,-2

Bushing Replacement
Figure 601

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REPAIR 2-1
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COMPONENT MAINTENANCE MANUAL

BULKHEAD ASSEMBLY - REPAIR 3-1

65C26304-1, -2, -7, -8, -11, -12, -15, -16, 65C30918-1, -2

1. General

- A. This procedure has the data necessary to replace the bushings in the bulkhead assembly (190, 195).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (240, 245, 250, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02. For miscellaneous materials, refer to SOPM 20-60-04

- (1) Remove bushings (240, 245, 250) and sleeve (255).
- (2) Install sleeve (255) per SOPM 20-50-03, except install with wet sealant, A00247.
- (3) Install bushings (240, 245, 250) per SOPM 20-50-03, except install with wet sealant, A00247.
- (4) Machine bushings (240, 245, 250) per REPAIR 3-1, Figure 601.
- (5) Fillet seal bushing flanges with sealant, A00247.
- (6) Top coat fillet seal with enamel coating, C50050.

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REPAIR 3-1

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COMPONENT MAINTENANCE MANUAL

3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-43-01	CHROMIC ACID ANODIZING
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For application of chromic acid anodizing, refer to SOPM 20-43-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Forward Bulkhead (260, 265) – Chromic acid anodize and apply primer, C00259 (F-18.13); omit primer from bushing bores. Apply enamel coating, C50050 (SRF-14.9812); omit enamel from bores. Material: Aluminum alloy

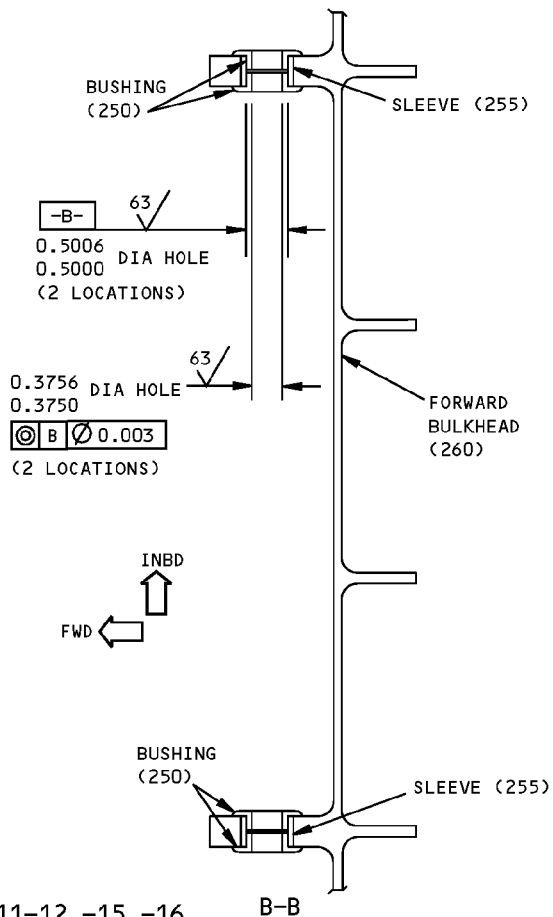
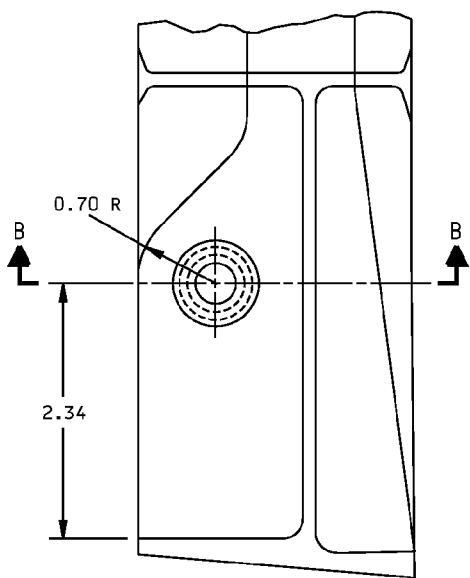
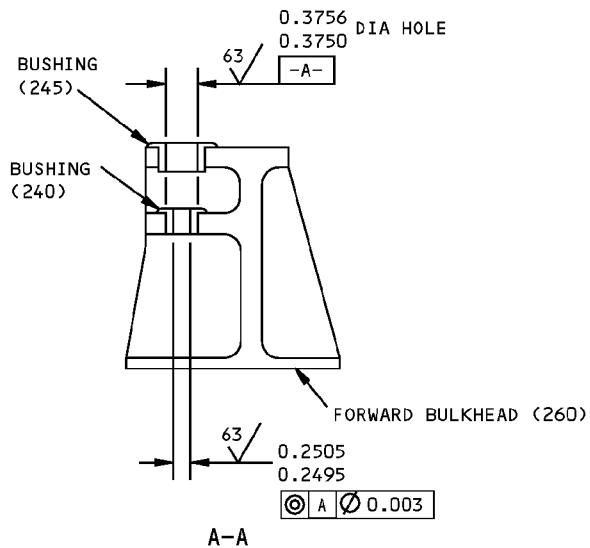
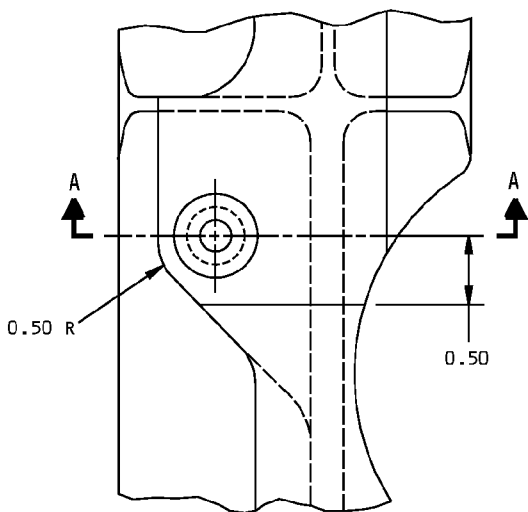
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ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

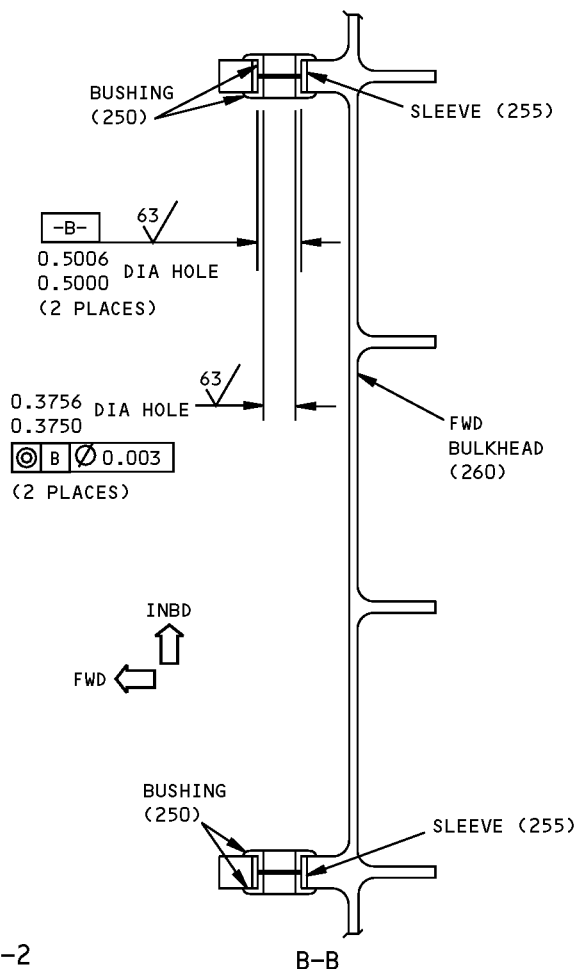
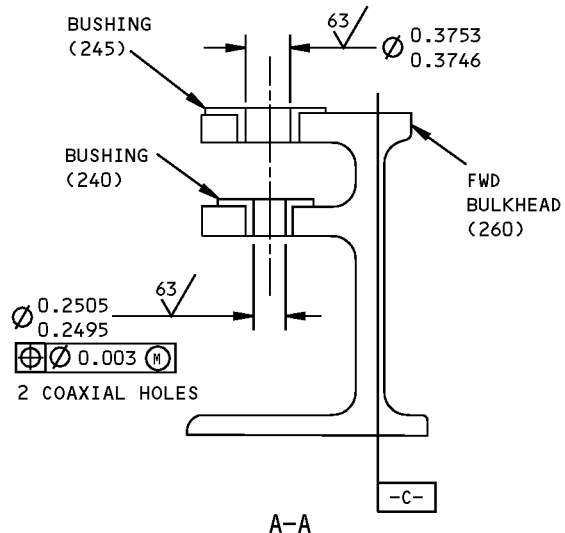
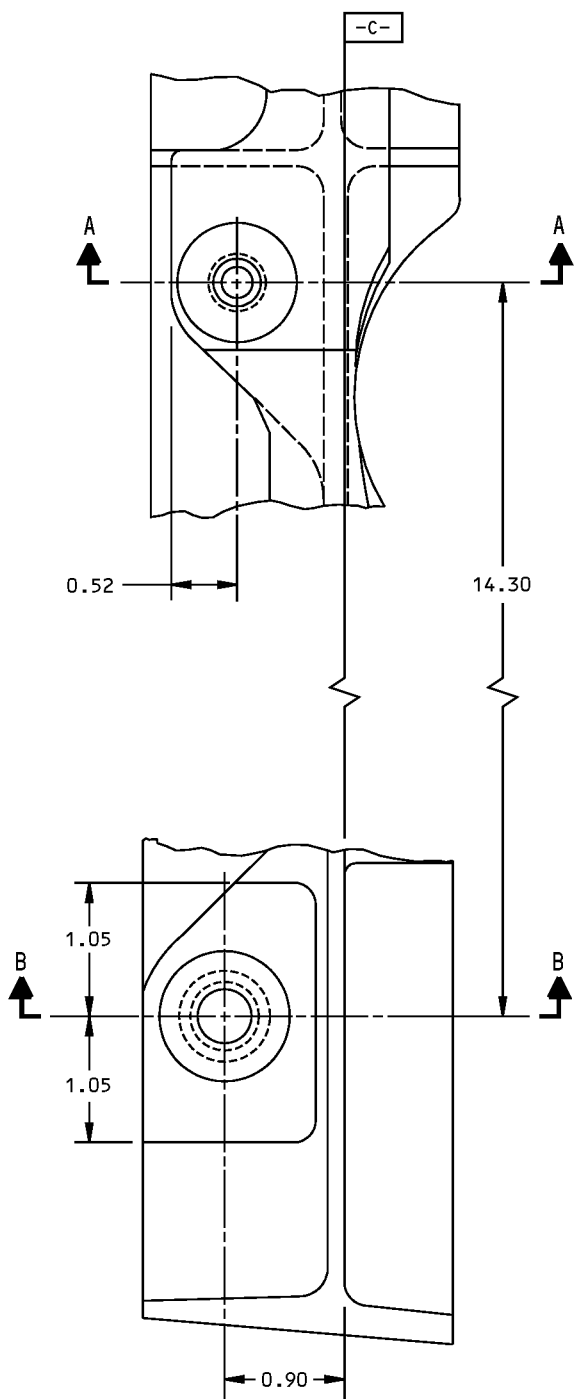
65C26304-1,-2,-7,-8,-11-12,-15,-16

Bushing Replacement
Figure 601 (Sheet 1 of 2)

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REPAIR 3-1
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ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

65C30918-1,-2

Bushing Replacement
Figure 601 (Sheet 2 of 2)

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REPAIR 3-1
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COMPONENT MAINTENANCE MANUAL

FITTING ASSEMBLY - REPAIR 4-1

65C26306-1, -2, -7, -8, -15, -16, -19, -20

1. General

- A. This procedure has the data necessary to replace the bushings in the fitting assembly (150, 155).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (160, 165, 170 IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02. For miscellaneous materials, refer to SOPM 20-60-04

- (1) Remove bushings (160, 165, 170) and sleeve (175).
- (2) Install sleeve (175) per SOPM 20-50-03, except install with wet sealant, A00247.
- (3) Machine sleeve (175) per REPAIR 4-1, Figure 601.
- (4) Install bushings (160, 165, 170) per SOPM 20-50-03, except install with wet sealant, A00247.
- (5) Machine bushings (160, 165, 170) per REPAIR 4-1, Figure 601.
- (6) Fillet seal bushings with sealant, A00247.
- (7) Top coat fillet seal with enamel coating, C50050.

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REPAIR 4-1

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COMPONENT MAINTENANCE MANUAL

3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-43-01	CHROMIC ACID ANODIZING
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For application of chromic acid anodizing, refer to SOPM 20-43-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Fitting (180, 185) – Chromic acid anodize and apply primer, C00259 (F-18.13); omit primer from bushing bores. Apply enamel coating, C50050 (SRF-14.9812); omit enamel from bushing bores. Material: Aluminum alloy

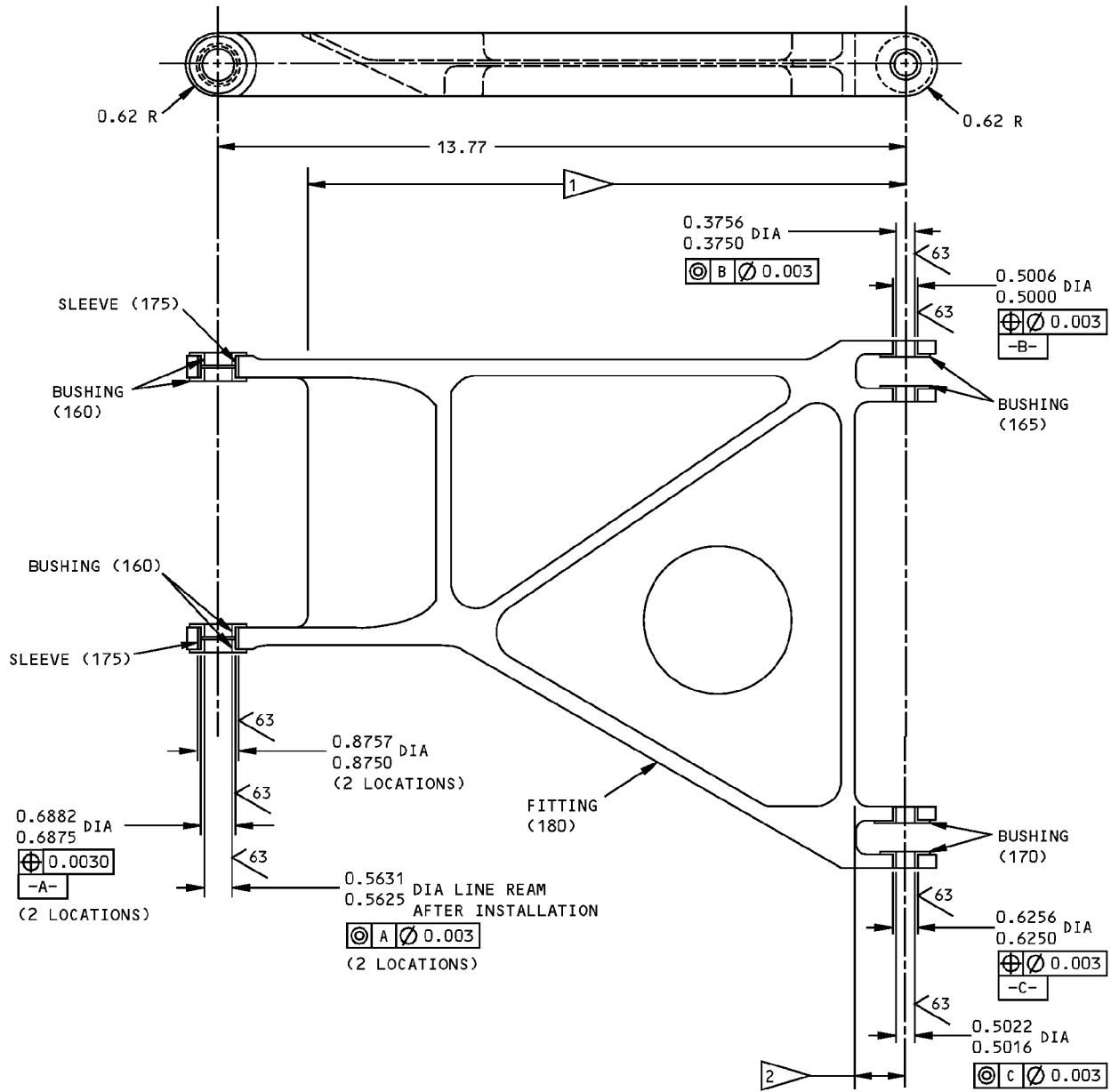
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- 1 11.95 FOR 65C26306-3,-4
12.01 FOR 65C26306-9,-10,-17,-18
- 2 1.00 FOR 65C26306-3,-4
1.06 FOR 65C26306-9,-10,-17,-18

ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

65C26306-1,-2,-7,-8,-15,-16,-19,-20

Bushing Replacement
Figure 601

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REPAIR 4-1
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COMPONENT MAINTENANCE MANUAL

DOOR ASSEMBLY - REPAIR 5-1

65-70947-27, -28, -287, -288, -293, -294, -295, -296

1. General

- A. This procedure has the data necessary to replace the bushings in the door assembly (680, 685).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (690, 695, IPL Figure 1) and (REPAIR 5-1, Figure 601)

A. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT

B. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03.

- (1) Remove the old bushings (690, 695).
- (2) Install replacement bushings (690, 695) by the press-fit or shrink-fit method of SOPM 20-50-03.
- (3) Machine only the BACB28-series bushings to design dimensions and finish. Do not machine the KJB-series bushings, because they are supplied adjusted to final dimensions.

3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

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REPAIR 5-1

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C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

- (1) Door (700, 701, 702, 705, 706, 707) - Sulfuric acid anodize (F-2.201). Apply primer, C00259 (SRF-12.205) to interior surfaces. Material: Al alloy

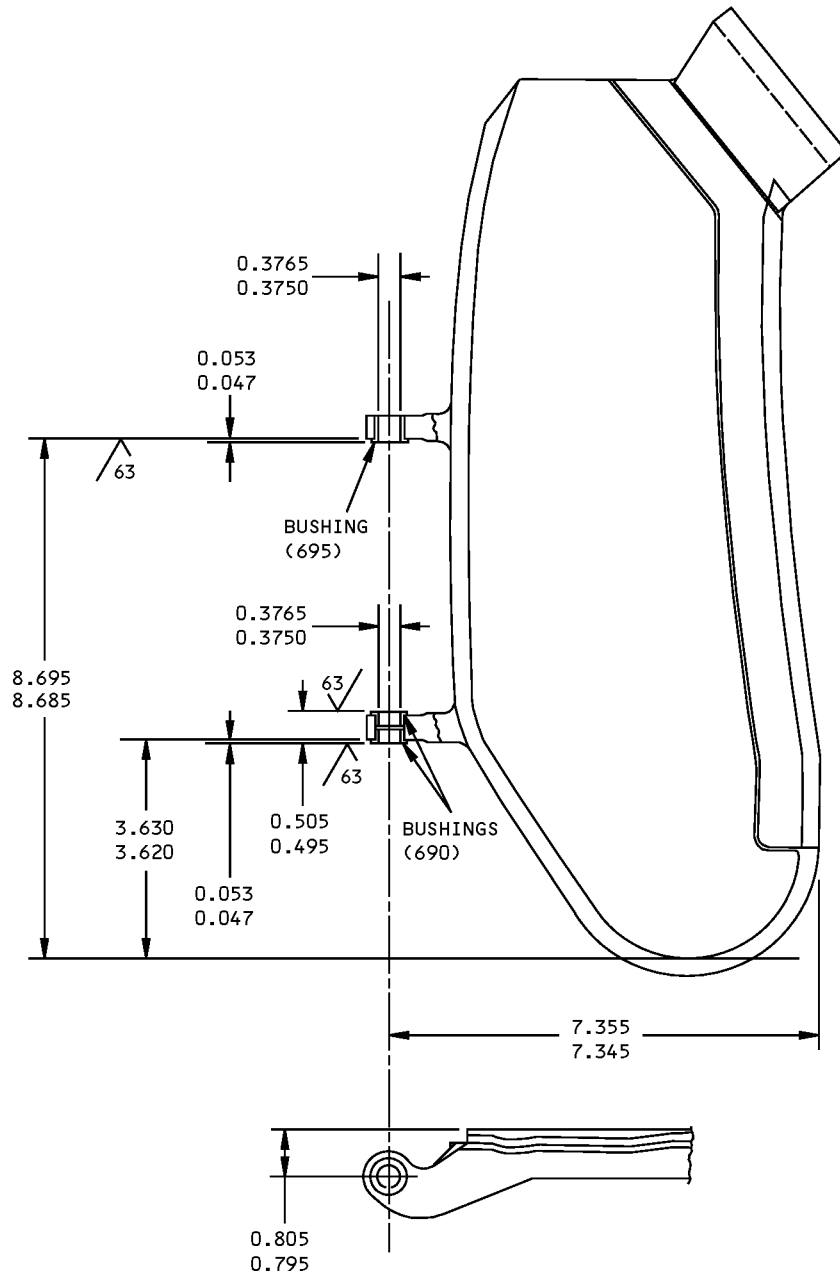
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REPAIR 5-1

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ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

65-70947-27,-28,-287,-288,-293,-294 Bushing Replacement
Figure 601

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REPAIR 5-1
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COMPONENT MAINTENANCE MANUAL

SUPPORT ASSEMBLY - REPAIR 6-1

69-73736-1, -2, 65C30479-1, -2, -7, -8

1. General

- A. This procedure has the data necessary to replace the bushings in the support assembly (200, 205).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to the REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (230, 235, IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For miscellaneous materials, refer to SOPM 20-60-04

- (1) Remove bushings (230, 235).
- (2) Install bushings (230, 235) per SOPM 20-50-03, except install with wet sealant, A00247.
- (3) Machine bushings (230, 235) per REPAIR 6-1, Figure 601.
- (4) Fillet seal bushings flanges with sealant, A00247.

3. Refinish

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

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B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02.

- (1) Support (220, 225) - Chromic acid anodize and apply primer, C00259 (F-18.13); except omit primer from bores. Apply enamel coating, C50050 (SRF-14.9812). Material: Aluminum alloy

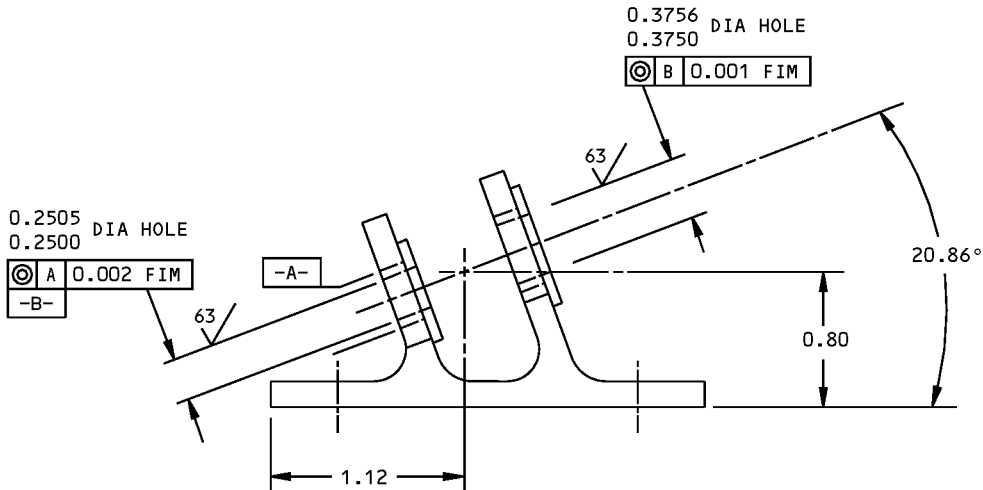
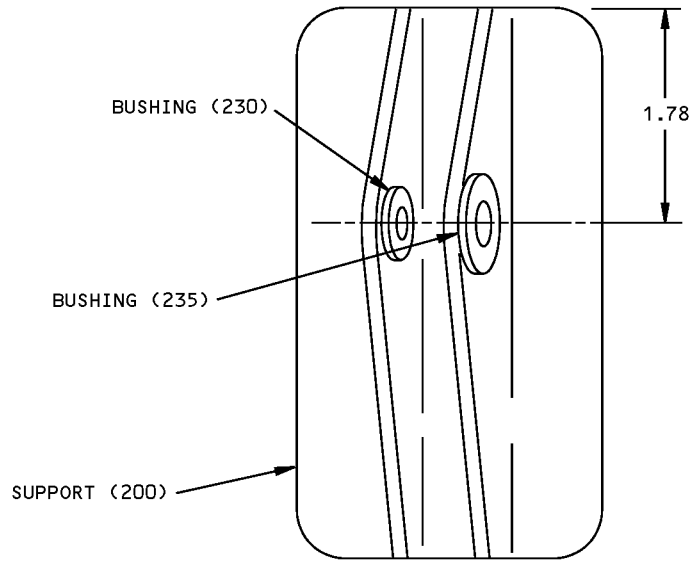
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REPAIR 6-1

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ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

69-73736-1,-2 65C30479-1,-2,-7,-8 Bushing Replacement
Figure 601

57-26-45

REPAIR 6-1
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MISCELLANEOUS PARTS REFINISH - REPAIR 7-1

1. General

- A. This procedure has the data necessary to refinish the parts, which are not given in the specified repairs.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in the procedure.
- C. Refer to IPL Figure 1 and IPL Figure 2 for the item numbers.

2. Refinish of Other Parts

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00195	Adhesive - Corrosion Inhibiting Coating, Adhesive Primer	BMS5-89
C00033	Coating - Exterior Protective Enamel, Flexibility Use	BMS10-60, Type II
C00175	Primer - Urethane Compatible, Corrosion Resistant (Less Than 1% Aromatic Amines)	BMS10-79, Type III
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C00319	Primer - Urethane Compatible, Corrosion Resistant	BMS10-79, Type II
C00767	Coating - Anti-Static Coating	BMS10-21, Type III
C00803	Primer - Polyurethane, High Temperature, Green	BAC5710, Type 51
C50021	Enamel - High Temperature, Epoxy, Max Operating Temperature 400F, Intermittent 550F	BAC5710, Type 53
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedures, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For finishing materials, refer to SOPM 20-60-02.

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REPAIR 7-1

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- (1) Repair of these parts is only replacement of the original finish. Refer to REPAIR 7-1, Table 601 for repair of the initial finish.

Table 601: Refinish details

IPL FIG. & ITEM	MATERIAL	FINISH
Fig. 1		
Inbd flap outbd track fairing assy (1,5)	Fiberglass/Epoxy	Prepare the surface (F-14.672). Apply coating, C00767 (F-14.685, which replaces SRF-14.68). Apply primer, C00319(F-19.46) on exterior surfaces and trimmed edges. Apply primer, C00319 and enamel coating, C00033 (SRF-14.9863-xxxx) on exterior and edges in the color of the airline.
Support Fitting (945)	Al Alloy	Hard anodize (F-2.204) the 1.20-inch radius area. On the other surfaces, chromic acid anodize (F-2.26) and apply primer, C00259 (SRF-12.205).
Angles (575,580), Channel (615), Hinge Fitting (725)	Al Alloy	Chemical treat or chromic acid anodize and apply primer, C00259 (F-18.05).
Spring (500, 501)	Chrome-Silicon Alloy Steel Wire	Apply primer, C00259 (SRF 12.206), with the spring extended to be sure of complete coverage.
Hinge Fitting (750,755)	Al Alloy	Chromic acid anodize and apply primer, C00259 (F-18.13).
Fairing (970)	Al Alloy	Apply adhesive, A00195 primer to inside surface (F-20.26).
Fig. 2		
End Cap (10)	Acetal Resin	Apply primer, C00319 (F-19.46).
End Cap (10A,10B)	Acetal Resin	Apply no finish (F-25.01).
End Cap (10C,10D)	Al Alloy	Chromic acid anodize (F-17.04). Apply primer, C00803 (F-19.43) and enamel, C50021 (F-21.20) on exterior surfaces and edges.
End Cap (10H)	Al Alloy	Chromic acid anodize (F-17.19). Apply primer, C00175 (F-19.47) all over. Apply enamel coating, C50050 (F-19.39-707) on exterior surfaces only.
Tailcone (15,20)	Fiberglass/Epoxy	Prepare the surface (F-14.672). Apply coating, C00767 (F-14.685, which replaces SRF-14.68). Apply primer, C00319 (F-19.46) on exterior surfaces and edges.
Tailcone (15A,15B, 20A, 20B)	Fiberglass/Epoxy	Prepare the surface (F-14.672). Apply coating, C00767 (F-14.685, which replaces SRF-14.68). Apply primer, C00803 (F-19.43) on exterior surfaces and edges.

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REPAIR 7-1
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FITTING ASSEMBLY - REPAIR 8-1

65C26303-7, -8, -9, -10, 65C30938-1, -2

1. General

- A. This procedure has the data necessary to replace the bushings in the fitting assembly (320, 325).
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- D. Refer to IPL Figure 1 for item numbers.

2. Bushing Replacement (330 IPL Figure 1)

- A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

- B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-60-02	FINISHING MATERIALS

- C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02.

- (1) Remove bushings (330) and sleeve (335).
- (2) Install sleeve (335) per SOPM 20-50-03 with wet sealant, A00247.
- (3) Machine sleeve (335) per REPAIR 8-1, Figure 601.
- (4) Install bushing (330) per SOPM 20-50-03 with wet sealant, A00247.
- (5) Machine bushing bores per REPAIR 8-1, Figure 601.
- (6) Fillet seal bushing flanges with wet sealant, A00247.
- (7) Top coat fillet seals with enamel coating, C50050.

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REPAIR 8-1
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3. Refinish

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C50050	Coating - Exterior Protective Enamel, White Gloss Color (BAC 702)	BMS10-60, Type I

B. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-60-02	FINISHING MATERIALS

C. Procedure

NOTE: For stripping of protective finishes, refer to SOPM 20-30-02. For general cleaning procedure, refer to SOPM 20-30-03. For the decoding table for Boeing finish codes, refer to SOPM 20-41-01. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02.

- (1) Fitting (340A, 340B, 345A, 345B) – Chromic acid anodize and apply primer, C00259 (F-18.13); omit primer from bores. Apply enamel coating, C50050 (SRF-14.9812). Material: Aluminum alloy.

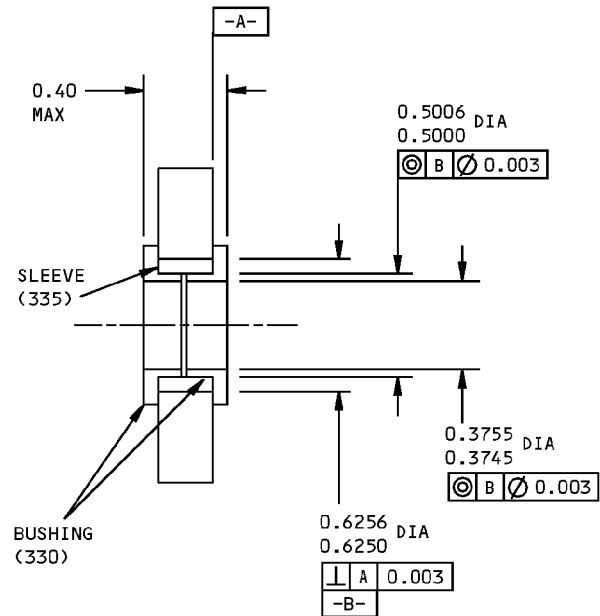
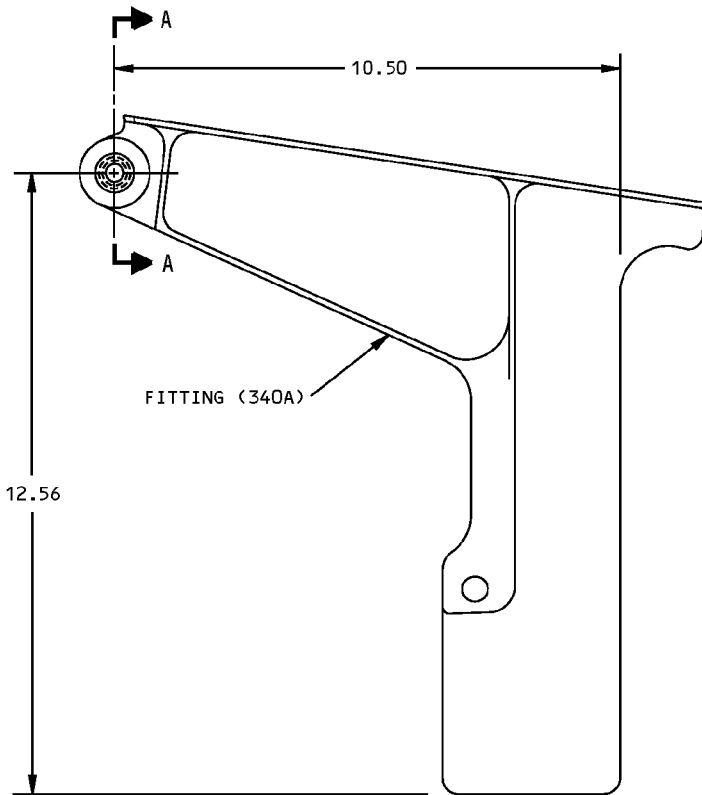
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REPAIR 8-1

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ALL DIMENSIONS ARE IN INCHES

ITEM NUMBERS REFER TO IPL FIG. 1

65C26303-7, -8, -9, -10
65C30938-1, -2

A-A

Bushing Replacement
Figure 601

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REPAIR 8-1

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ASSEMBLY

1. General

- A. This procedure has the data necessary to assemble the inboard flap outboard track tailcone assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 and IPL Figure 2 for the item numbers.

2. Assembly (IPL Figure 1)

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
A00247	Sealant - Pressure And Environmental - Chromate Type	BMS 5-95
A00900	Sealant - Silicone, RTV - Dow Corning 93-006-1	
A01076	Adhesive - Synthetic Rubber	BAC5010, Type 93 (BMS5-95, Class B)
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
C00580	Primer - Adhesive Bonding, RTV - DC 1200	
D00015	Grease - Aircraft Bearing (Use BMS 3-24 until existing stocks are depleted, BMS 3-33 supersedes BMS 3-24)	BMS3-24 (Superseded by BMS 3-33)

B. References

Reference	Title
SOPM 20-10-06	REPAIR OF CONDUCTIVE COATINGS
SOPM 20-11-03	REPAIR OF ELECTRICAL TERMINATIONS AND ELECTRICAL BONDING AREAS
SOPM 20-30-03	GENERAL CLEANING PROCEDURES
SOPM 20-41-02	APPLICATION OF CHEMICAL AND SOLVENT RESISTANT FINISHES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-50-11	APPLICATION OF AERODYNAMIC SMOOTHING SEALANT
SOPM 20-50-12	APPLICATION OF ADHESIVES
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

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ASSEMBLY

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COMPONENT MAINTENANCE MANUAL

C. Procedure

NOTE: For general cleaning procedure, refer to SOPM 20-30-03. For application of chemical and solvent-resistant finishes, refer to SOPM 20-41-02. For finishing materials, refer to SOPM 20-60-02. For miscellaneous materials, refer to SOPM 20-60-04

CAUTION: ONLY TITANIUM OR CORROSION RESISTANT STEEL FASTENERS SHALL BE USED WITH FIBERGLASS/EPOXY COMPOSITES. SUBSTITUTION WITH ALUMINUM OR PLATED ALLOY STEEL FASTENERS IS NOT ALLOWED AS GALVANIC CORROSION WILL SHORTEN COMPONENT SERVICE LIFE.

- (1) Use standard industry practices to assemble this component plus the following additional steps.
- (2) Gaps between adjacent panels and skins shall be 0.08 inches nominal, plus 0.02 inch, minus 0.04 inch.
- (3) Bolts (210, 620, 645, 710, 730, 735, 930) - Install with wet sealant, A00247.
- (4) Door Assy (525, 525A, 525C, 525E, 530, 530A, 530C, 530E).
 - (a) Apply parting agent to door (680, 681, 682, 685, 686, 687) per ASSEMBLY, Figure 701.
 - (b) With door held open apply a "FORM IN PLACE" gasket using DC 1200 primer, C00580 plus Dow Corning 93-006-1 sealant, A00900 sealant to inside of aft and lower structure surrounding door over length corresponding to ASSEMBLY, Figure 701. Close door slowly until door is fully closed and support by rub strip (660, 665), to form gasket. Remove excess Dow Corning 93-006-1 sealant, A00900 from exterior. Squeeze out on interior surfaces is acceptable. After cure trim Dow Corning 93-006-1 sealant, A00900 flush with edge of door aperture.
- (5) Fasteners (370, 905) – Install and check conductance per SOPM 20-10-06, check resistance level per SOPM 20-11-03.
- (6) Shim (670) - Use if and as required for mismatch per ASSEMBLY, Figure 702.
- (7) Shims (395A, 400, 405, 410A, 415A, 420, 425)
 - (a) Bond shim to metallic structure per SOPM 20-50-12, adhesive, A01076.
 - (b) Delaminate as required.
 - (c) Coat exposed surfaces with primer, C00259.
- (8) Shim-washer (145) - Remove shim laminations as required to achieve full clamp-up (clearance - 0.000 inches nominal, plus 0.002 inches).
- (9) Shim-Washer (137)
 - (a) Remove lamination as required.
 - (b) Install with primer, C00259 wet or dry.
- (10) Bolts (505) - install with grease, D00015 on all surfaces.
- (11) Bushings (485, 490, 491, 555, 560, 561, 600, 600A, 605, 606)
 - (a) Install per SOPM 20-50-03.
 - (b) Assemble with primer, C00259.
- (12) Install tailcone assembly (1D, 1E, 1F, 1I, 1O, 5D, 5E, 5F, 5I, 5O; IPL Figure 1).
 - (a) Bolts (910, 910A, 915, 915A, 920, 920A) - Install with sealant, A00247.

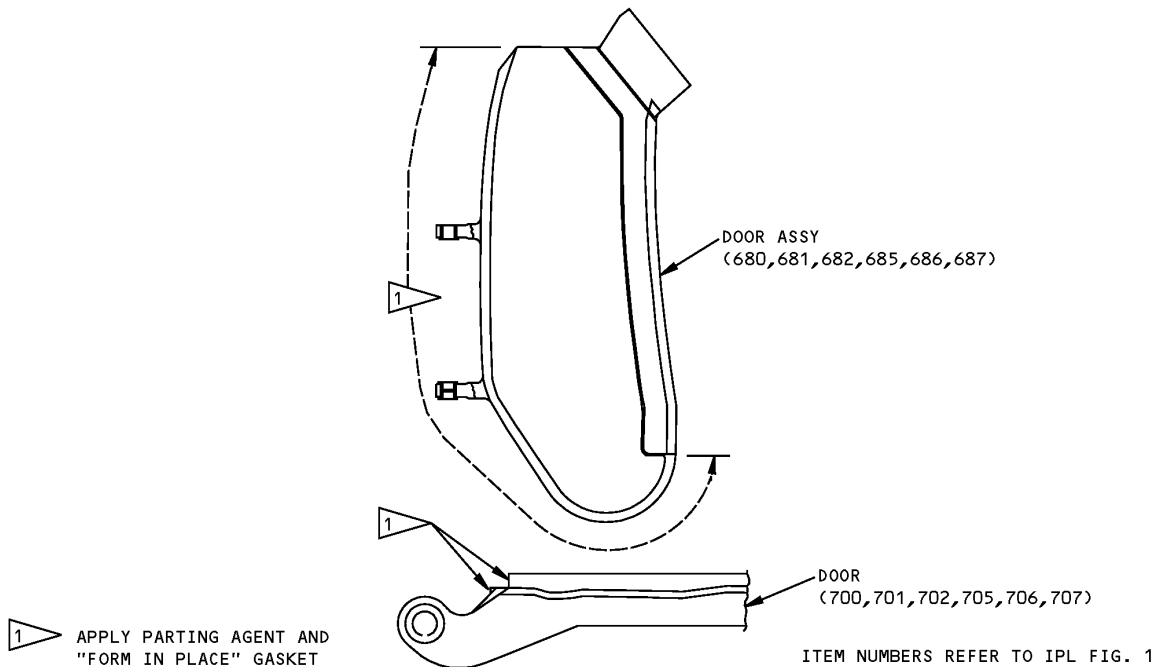
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ASSEMBLY
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- (b) Fill joint with aerodynamic filler sealant, A00247 per SOPM 20-50-11 to within plus or minus 0.02 inches of fairing surface.
- (13) Strip-rub (805, 810, 835, 840, 841, 842)
 - (a) Bond strip-rub per SOPM 20-50-12, with adhesive, A01076.
 - (b) Install fasteners (760, 765, 770, 775, 780, 795, 800, 820, 825, 830) with wet sealant, A00247 as follows:
 - 1) Coat fastener heads and shank; on hi-loks remove sealant, A00247 from groove.
 - 2) Install fastener within application time of sealant, A00247.
 - 3) Clean off excess sealant, A00247.
- (14) Fairing (970)
 - (a) Bond to fairing assembly (1, 1B, 1C, 1G, 1H, 1J, 5, 5B, 5C, 5G, 5H, 5J) per SOPM 20-50-12, with adhesive, A01076.
 - (b) Install Hi-Loks (950, 955, 960) with wet sealant, A00247, (See ASSEMBLY, Paragraph 2.C.(13)(b)).



Location of "Form in Place" Gasket
Figure 701

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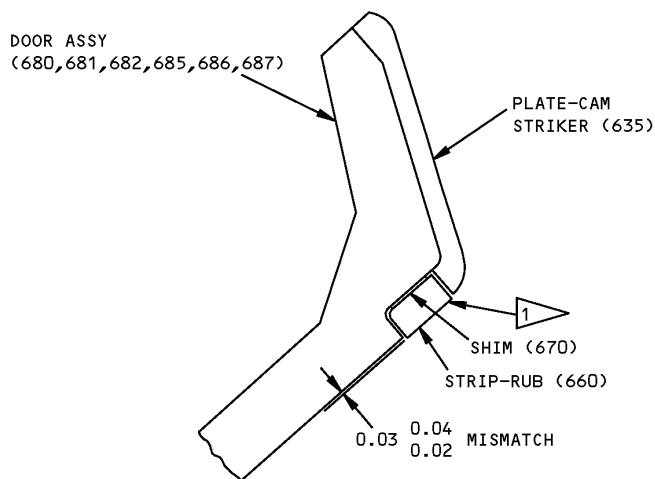
ASSEMBLY

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1 BLEND INTO PROFILE WITH
1.00-2.00 RADIUS

ALL DIMENSIONS ARE IN INCHES
ITEM NUMBERS REFER TO IPL FIG. 1

Shim Installation
Figure 702

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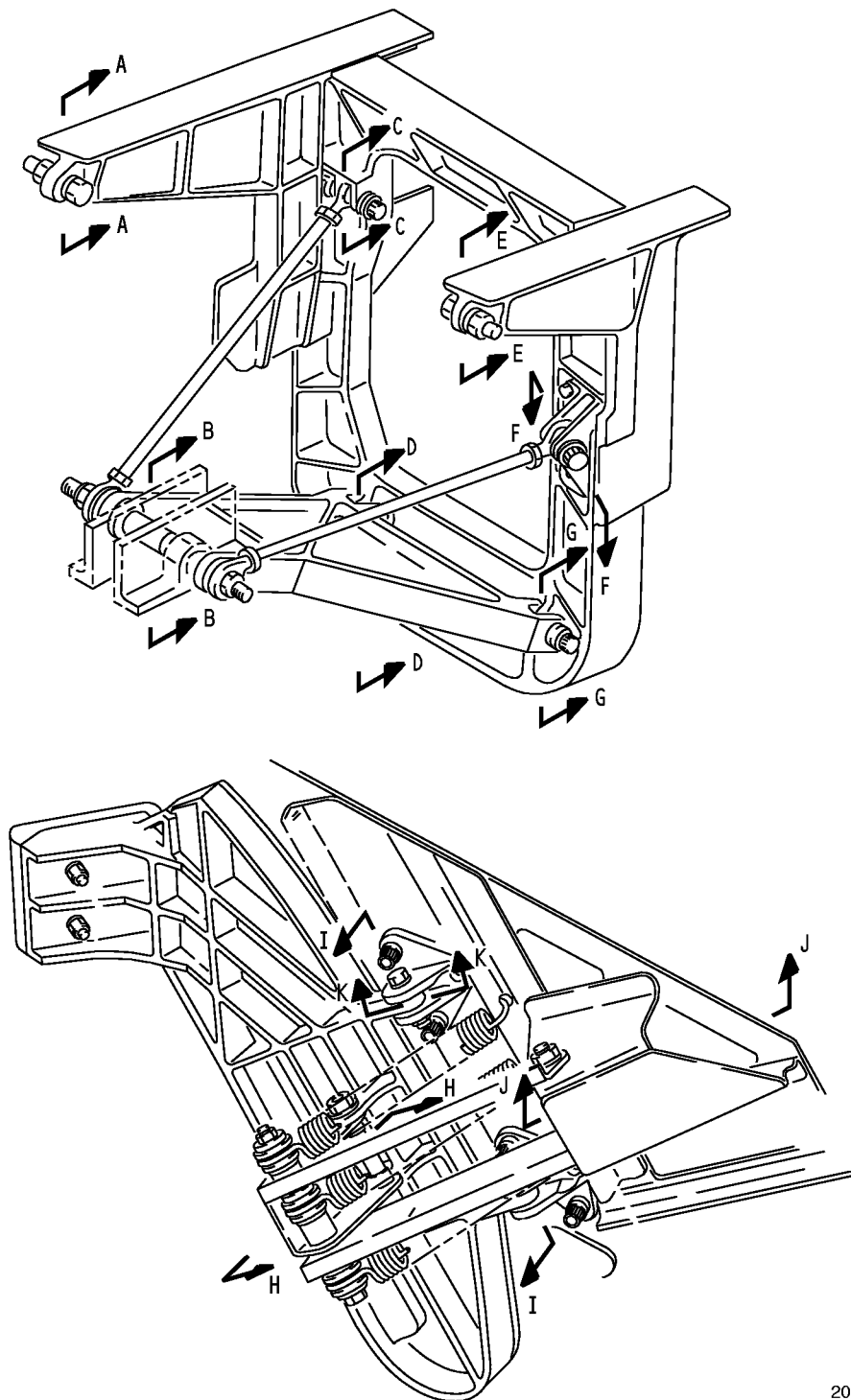
ASSEMBLY

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FITS AND CLEARANCES

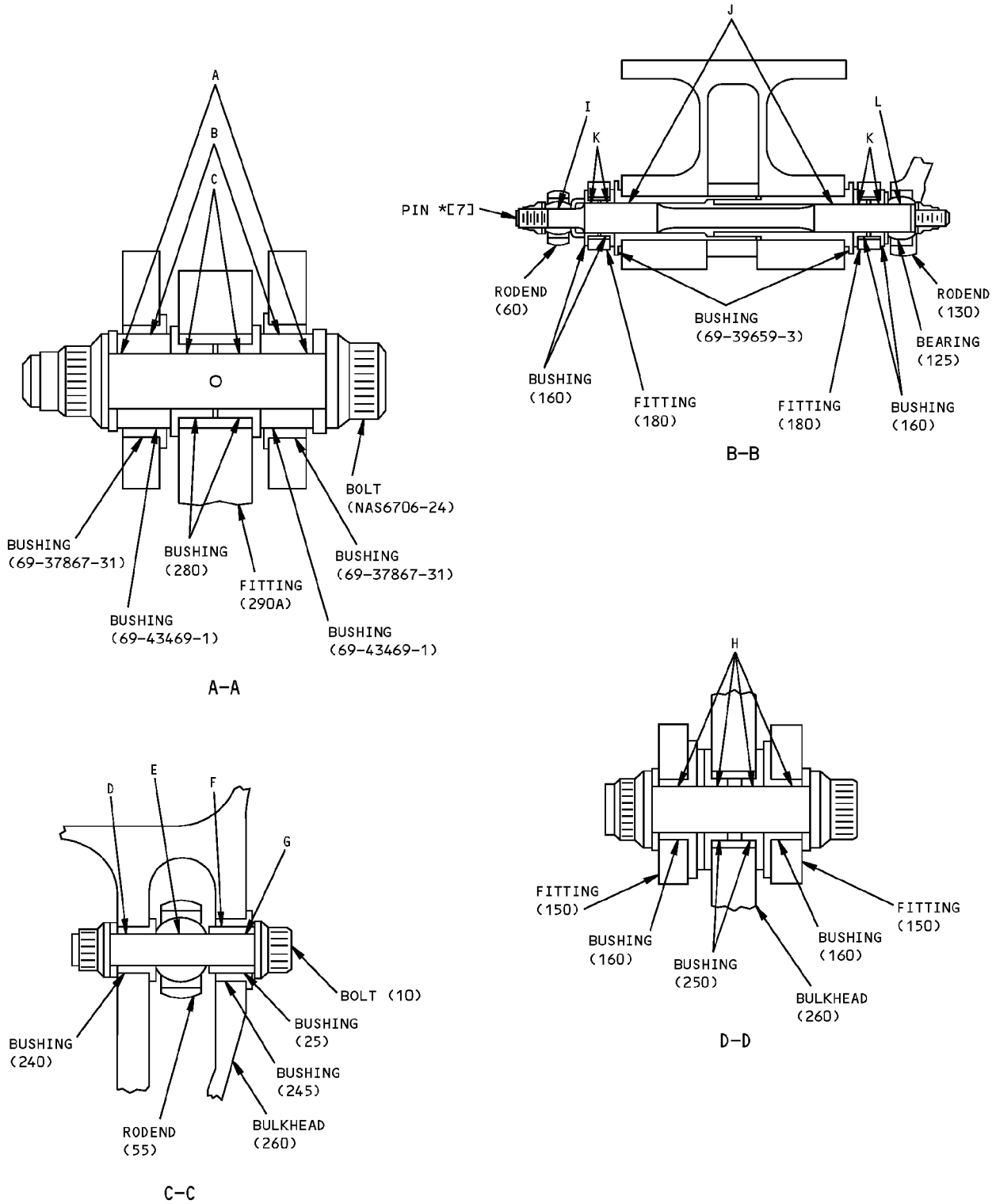


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Fits and Clearances
Figure 801 (Sheet 1 of 7)

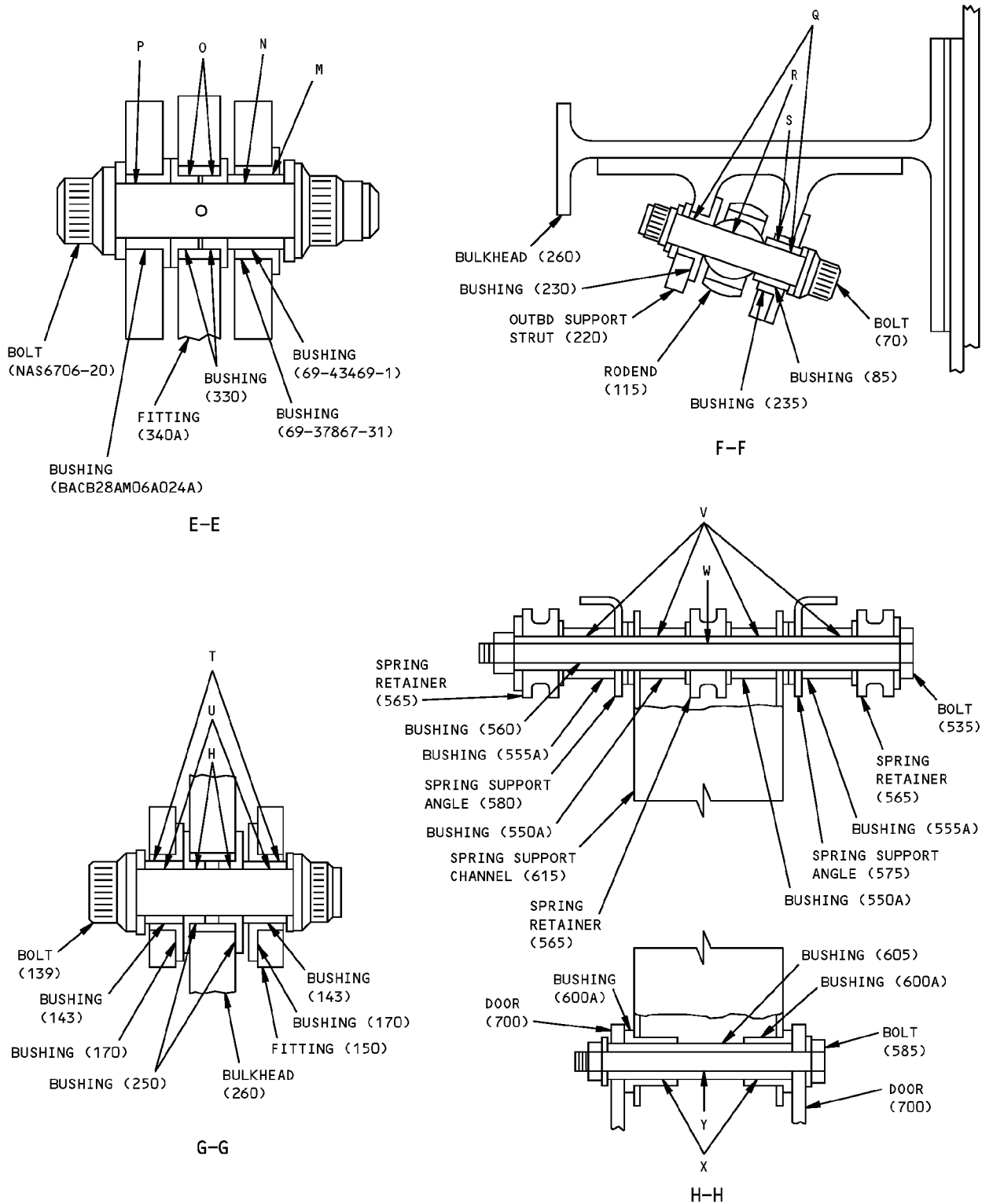
57-26-45
FITS AND CLEARANCES
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COMPONENT MAINTENANCE MANUAL



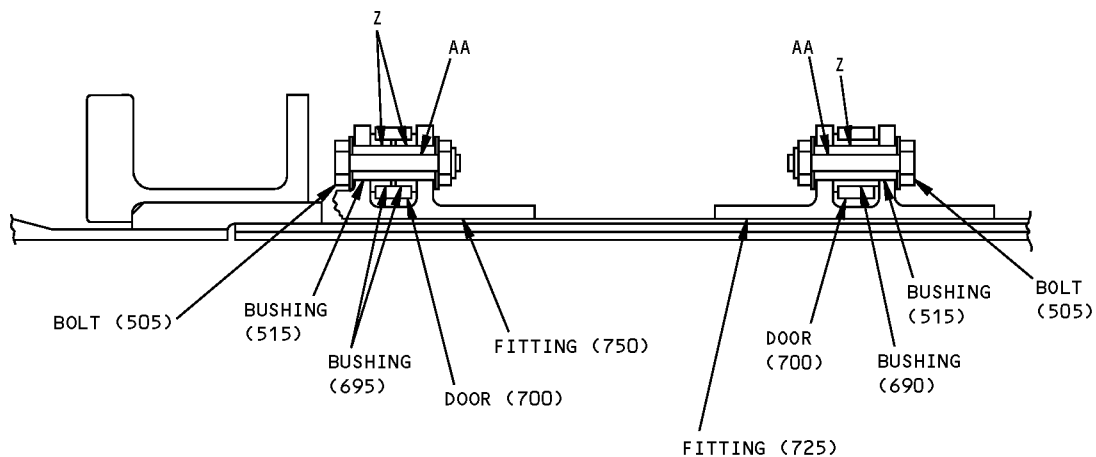
Fits and Clearances
Figure 801 (Sheet 2 of 7)

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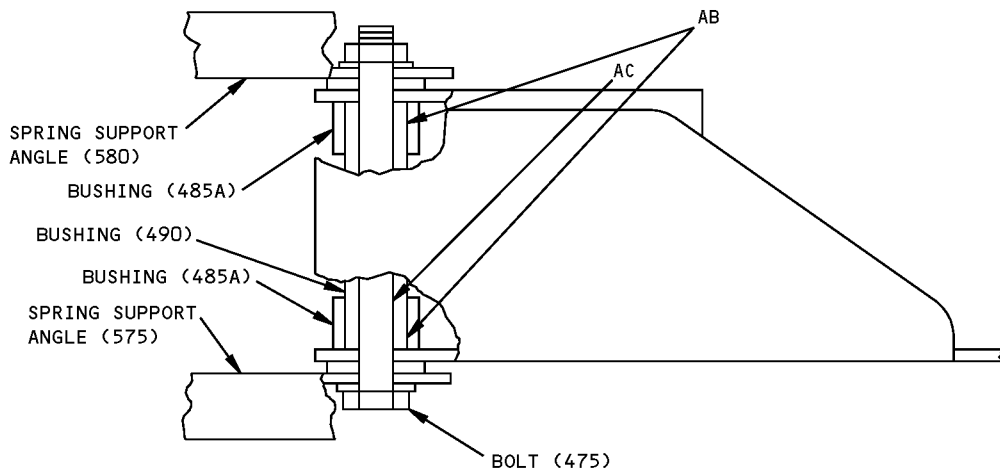


Fits and Clearances
Figure 801 (Sheet 3 of 7)

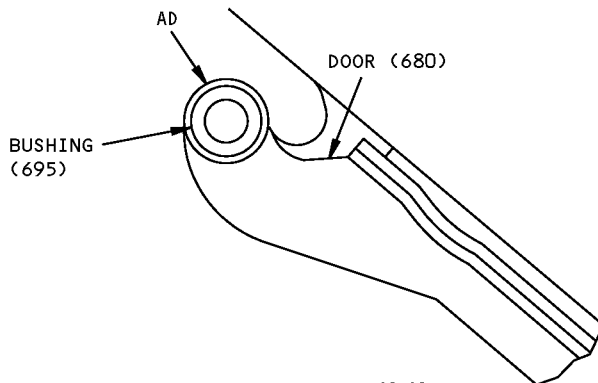
COMPONENT MAINTENANCE MANUAL



I-I

















J-J



K-K

Fits and Clearances
Figure 801 (Sheet 4 of 7)

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Ref Letter Fig.801	Mating Item No. IPL Fig.1	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
A	ID 	0.3775	0.3790	0.003	0.0055	0.3675	0.3845	0.0100
	OD 	0.3735	0.3745					
B	ID 	0.6225	0.6240	0.001	0.004	0.6145	0.6295	0.008
	OD 	0.6200	0.6215					
C	ID 280	0.3745	0.3755	0.000	0.002	0.3705	0.3785	0.004
	OD 	0.3735	0.3745					
D	ID 240	0.2495	0.2505	0.000	0.002	0.2455	0.2535	0.004
	OD 10	0.2485	0.2495					
E	ID 55	0.2495	0.2500	0.000	0.0015	0.2465	0.2525	0.003
	OD 10	0.2485	0.2495					
F	ID 245	0.3750	0.3756	0.0005	0.0016	0.3718	0.3777	0.0032
	OD 25	0.3740	0.3745					
G	ID 25	0.2500	0.2505	0.0005	0.002	0.2460	0.2535	0.004
	OD 10	0.2485	0.2495					
H	ID 160, 250	0.3750	0.3756	0.0005	0.0021	0.3708	0.3787	0.0042
	OD 139	0.3735	0.3745					
I	ID 60	0.3745	0.3750	0.000	0.001	0.3725	0.3765	0.002
	OD 	0.3740	0.3745					
J	ID 	0.5620	0.5630	0.0005	0.002	0.5580	0.5655	0.004
	OD 	0.5610	0.5615					
K	ID 160	0.5625	0.5631	0.001	0.0021	0.5583	0.5657	0.0042
	OD 	0.5610	0.5615					
L	ID 125	0.5620	0.5625	0.0005	0.0015	0.5590	0.5645	0.003
	OD 	0.5610	0.5615					
M	ID 	0.6225	0.6240	0.001	0.004	0.6145	0.6295	0.008
	OD 	0.6200	0.6215					
N	ID 	0.3775	0.3790	0.003	0.0055	0.3675	0.3845	0.100
	OD 	0.3735	0.3745					

Fits and Clearances
Figure 801 (Sheet 5 of 7)



COMPONENT MAINTENANCE MANUAL

Ref Letter Fig.801	Mating Item No. IPL Fig.1	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
O	ID 330	0.3745	0.3755	0.000	0.002		0.3785	0.004
	OD	0.3735	0.3745			0.3705		
P	ID	0.3750	0.3756	0.0005	0.0021		0.3787	0.0042
	OD	0.3735	0.3745			0.3708		
Q	ID 85,230	0.2500	0.2505	0.0005	0.002		0.2535	0.004
	OD 70	0.2485	0.2495			0.2460		
R	ID 115	0.2495	0.2500	0.000	0.0015		0.2525	0.003
	OD 70	0.2485	0.2495			0.2465		
S	ID 235	0.3750	0.3756	0.0005	0.0016		0.3777	0.0032
	OD 85	0.3740	0.3745			0.3718		
T	ID 170	0.5016	0.5022	0.0003	0.0016		0.5045	0.0032
	OD 143	0.5006	0.5013			0.4984		
U	ID 143	0.3750	0.3765	0.0005	0.003		0.3805	0.006
	OD 139	0.3735	0.3745			0.3690		
V	ID 550A, 555A	0.3750	0.3765	0.002	0.0055		0.3830	0.0100
	OD 560	0.3710	0.3730			0.3650		
V	ID 617	0.3765	0.3775	0.0020	0.0035		0.3845	0.0100
	OD 561	0.3740	0.3745			0.3665		
W	ID 560	0.1895	0.1910	0.000	0.0025		0.1945	0.005
	OD 535	0.1885	0.1895			0.1845		
W	ID 561	0.2500	0.2505	0.0005	0.0020		0.2545	0.0050
	OD 536	0.2485	0.2495			0.2450		
X	ID 600A	0.3750	0.3765	0.002	0.0055		0.3830	0.0100
	OD 605	0.3710	0.3730			0.3650		
X	ID 617	0.3765	0.3775	0.0020	0.0035		0.3845	0.0100
	OD 606	0.3740	0.3745			0.3665		



ALL DIMENSIONS ARE IN INCHES

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



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


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Ref Letter Fig.801	Mating Item No. IPL Fig.1	Design Dimension				Service Wear Limit			
		Dimension		Assembly Clearance		Dimension		Maximum Clearance	
		Min	Max	Min	Max	Min	Max		
Y	ID 605	0.1895	0.1910	0.000	0.0025	0.1845	0.1945	0.005	
	OD 585	0.1885	0.1895						
Y	ID 606	0.2500	0.2505	0.0005	0.0020	0.2450	0.2545	0.0050	
	OD 586	0.2485	0.2495						
Z	ID 690,695	0.3750	0.3765	0.002	0.0055	0.3650	0.3830	0.0100	
	OD 515	0.3710	0.3730						
Z	ID 695A	0.3755	0.3765	0.0010	0.0025	0.3655	0.3845	0.0100	
	OD 515A	0.3740	0.3745						
AA	ID 515	0.1895	0.1910	0.000	0.0025	0.1845	0.1945	0.005	
	OD 505	0.1885	0.1895						
AA	ID 515A	0.2500	0.2505	0.0005	0.0020	0.2450	0.2545	0.0050	
	OD 505C	0.2485	0.2495						
AB	ID 485A	0.3750	0.3765	0.002	0.0055	0.3650	0.3830	0.0100	
	OD 490	0.3710	0.3730						
AB	ID 482C	0.3765	0.3775	0.0020	0.0035	0.3665	0.3845	0.0100	
	OD 491	0.3740	0.3745						
AC	ID 490	0.1895	0.1910	0.000	0.0025	0.1845	0.1945	0.005	
	OD 475	0.1885	0.1895						
AC	ID 491	0.2500	0.2505	0.0005	0.0020	0.2450	0.2545	0.0050	
	OD 476	0.2485	0.2495						
AD	ID			---	---				
	OD 680 	0.36	0.38			0.34	---	---	
AD	ID			---	---				
	OD 683 	0.3600	0.3800			0.3400	---	---	

ALL DIMENSIONS ARE IN INCHES

-  BUSHING (69-43469-1)
-  BOLT (NAS6706-24)
-  BUSHING (69-37867-31)
-  BOLT (NAS6706-20)

-  BUSHING (BACB28AM06A024A)
-  BUSHING (69-39659-3)
-  PIN (69-73103-2)
-  MINIMUM WALL THICKNESS IS 0.09

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Fits and Clearances
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REF IPL		NAME	TORQUE*	
FIG. NO.	ITEM NO.		POUND-INCHES	POUND-FEET
1	20	Nut	65-90	
1	80	Nut	65-90	
1	147	Nut	220-360	

* REFER TO SOPM 20-50-01 FOR TORQUE VALUES OF STANDARD FASTENERS.

Torque Table
Figure 802



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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

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ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
.	Assembly					
.	Attaching parts for assembly					
.	.	Detail parts for assembly				
.	.	Subassembly				
.	.	Attaching parts for subassembly				
.	.	.	Detail parts for subassembly			
.	.	.	Sub-subassembly			
.	.	.	Attaching parts for subassembly			
.	.	.	.	Details parts for sub-subassembly		
						Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
- (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
- (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

VENDOR CODES

Code	Name
06710	LAMSON AND SESSIONS CO THE VALLEY-TODECO 12975 BRADLEY AVENUE SYLMAR, CALIFORNIA 91342-3830 FORMERLY VALLEY BOLT CORP VB0097 IN NORTH HOLLYWOOD, CA
50632	KAMATICS CORP SUB OF KAMAN CORP 1335 BLUE HILLS ROAD BLOOMFIELD, CONNECTICUT 06002-1304
57606	REXNORD CORP PSI BEARINGS DIV 2175 UNION PL SIMI VALLEY, CALIFORNIA 93065-1661 FORMERLY PSI BEARINGS

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NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-26300-112		2	5D	RF
65-26300-78		2	5C	RF
65-70930-22		1	615	1
65-70930-27		1	575	1
65-70930-28		1	580	1
65-70938-3		1	970	1
65-70947-149		1	700	1
65-70947-150		1	705	1
65-70947-27		1	680	1
65-70947-28		1	685	1
65-70947-287		1	681	1
65-70947-288		1	686	1
65-70947-289		1	701	1
		1	702	1
65-70947-290		1	706	1
		1	707	1
65-70947-293		1	682	1
65-70947-294		1	687	1
65-70947-295		1	683	1
65-70947-296		1	688	1
65C25300-111SP		1	1P	RF
65C26300-100SP		1	5L	RF
65C26300-101		1	525H	1
65C26300-102		1	530H	1
65C26300-105SP		1	1M	RF
65C26300-106SP		1	5M	RF
65C26300-107SP		1	1N	RF
65C26300-108SP		1	5N	RF
65C26300-111		2	1D	RF
65C26300-112SP		1	5P	RF
65C26300-15		1	137	2
		1	137A	2
65C26300-21		1	400	AR
65C26300-22		1	405	AR

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C26300-30		1	885	4
65C26300-39		1	525	1
65C26300-3SP		1	1	RF
65C26300-40		1	530	1
65C26300-42		1	395A	1
		1	410A	1
65C26300-43		1	415A	1
65C26300-45		1	90A	1
		1	90D	1
65C26300-47		1	525A	1
65C26300-48		1	530A	1
65C26300-4SP		1	5	RF
65C26300-5		2	1	RF
65C26300-57		2	1A	RF
65C26300-57SP		1	1E	RF
65C26300-58		2	5A	RF
65C26300-58SP		1	5E	RF
65C26300-59SP		1	1B	RF
65C26300-5SP		1	1D	RF
65C26300-6		2	5	RF
65C26300-60SP		1	5B	RF
65C26300-63SP		1	1C	RF
65C26300-64SP		1	5C	RF
65C26300-69		2	1B	RF
65C26300-69SP		1	1F	RF
65C26300-6SP		1	5D	RF
65C26300-7		1	30	1
		1	30A	1
65C26300-70		2	5B	RF
65C26300-70SP		1	5F	RF
65C26300-73SP		1	1G	RF
65C26300-74SP		1	5G	RF
65C26300-75SP		1	1H	RF
65C26300-76SP		1	5H	RF
65C26300-77		2	1C	RF

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C26300-77SP		1	1I	RF
65C26300-78SP		1	5I	RF
65C26300-79		1	525C	1
		1	525E	1
65C26300-80		1	530C	1
		1	530E	1
65C26300-85		1	525D	1
65C26300-86		1	530D	1
65C26300-87SP		1	1J	RF
65C26300-88SP		1	5J	RF
65C26300-89		1	525F	1
		1	525G	1
65C26300-90		1	530F	1
		1	530G	1
65C26300-93SP		1	1K	RF
65C26300-94SP		1	5K	RF
65C26300-99SP		1	1L	RF
65C26301-43		1	375	1
65C26301-44		1	380	1
65C26301-45		1	385	1
65C26301-46		1	390	1
65C26301-51		2	15	1
65C26301-52		2	20	1
65C26301-63		2	15A	1
65C26301-64		2	20A	1
65C26301-65		1	980	1
		1	980A	1
		2	7	RF
65C26301-66		1	985	1
		1	985A	1
		2	8	RF
65C26301-69		2	15B	1
65C26301-70		2	20B	1
65C26301-71		1	975	1
		1	975A	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	6	RF
65C26301-75		1	391	1
65C26301-76		1	392	1
65C26301-77		1	381	1
65C26301-78		1	382	1
65C26301-87		1	980B	1
		1	980C	1
		2	7A	RF
65C26301-88		1	985B	1
		1	985C	1
		2	8A	RF
65C26301-91		1	975B	1
		1	975C	1
		2	6A	RF
65C26302-10		1	295A	1
65C26302-7		1	270A	1
		1	270B	1
		1	270C	1
		1	270G	1
65C26302-8		1	275A	1
		1	275B	1
		1	275C	1
		1	275G	1
65C26302-9		1	290A	1
65C26303-10		1	345A	1
65C26303-7		1	320A	1
		1	320B	1
		1	320C	1
65C26303-8		1	325A	1
		1	325B	1
		1	325C	1
65C26303-9		1	340A	1
65C26304-1		1	190A	1
65C26304-10		1	265	1
65C26304-11		1	190B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	190C	1
65C26304-12		1	195A	1
		1	195C	1
65C26304-13		1	260B	1
65C26304-15		1	190E	1
		1	190F	1
65C26304-16		1	195E	1
		1	195F	1
65C26304-17		1	260D	1
65C26304-18		1	265D	1
65C26304-2		1	194	1
65C26304-5		1	260A	1
65C26304-6		1	265A	1
65C26304-7		1	190	1
65C26304-8		1	195	1
65C26304-9		1	260	1
65C26306-1		1	150A	1
		1	150C	1
65C26306-10		1	185A	1
65C26306-15		1	150D	1
		1	150E	1
		1	150F	1
65C26306-16		1	155D	1
		1	155E	1
		1	155F	1
65C26306-17		1	180B	1
65C26306-18		1	185B	1
65C26306-19		1	150G	1
		1	150H	1
		1	150I	1
		1	150J	1
65C26306-2		1	155A	1
		1	155C	1
65C26306-20		1	155G	1
		1	155H	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	155I	1
		1	155J	1
65C26306-21		1	180C	1
65C26306-22		1	185C	1
65C26306-3		1	180	1
65C26306-4		1	185	1
65C26306-7		1	150	1
		1	150B	1
65C26306-8		1	155	1
		1	155B	1
65C26306-9		1	180A	1
65C26307-13		1	845B	1
65C26307-14		1	850B	1
65C26307-15		1	845D	1
		1	845E	1
65C26307-16		1	850D	1
		1	850E	1
65C26307-17		1	845F	1
65C26307-18		1	850F	1
65C26307-19		1	845G	1
65C26307-20		1	850G	1
65C26307-5		1	845C	1
65C26307-6		1	850C	1
65C30443-1		1	725B	1
		1	725D	1
		1	725E	1
65C30448-5		1	750E	1
		1	750F	1
		1	750G	1
65C30448-6		1	755E	1
		1	755F	1
		1	755H	1
65C30479-1		1	200A	1
		1	200C	1
65C30479-10		1	225B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65C30479-2		1	205A	1
		1	205C	1
		1	205E	1
65C30479-3		1	220A	1
65C30479-4		1	225A	1
65C30479-8		1	205F	1
65C30479-9		1	220B	1
65C30937-1		1	270D	1
		1	270E	1
65C30937-2		1	275D	1
		1	275E	1
65C30937-3		1	290B	1
65C30937-4		1	295B	1
65C30938-1		1	320D	1
		1	320E	1
65C30938-10		1	325G	1
		1	325H	1
		1	325I	1
65C30938-2		1	325D	1
		1	325E	1
65C30938-3		1	340B	1
65C30938-4		1	345B	1
65C30938-7		1	340C	1
65C30938-8		1	345C	1
65C30938-9		1	320G	1
		1	320H	1
		1	320I	1
65C31457-1		2	10C	1
		2	10G	1
65C31457-2		1	990	1
		1	990A	1
		2	9	1
		2	10D	1
		2	10E	1
		2	10H	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		2	10J	1
65C31457-3		2	10K	1
65C33034-1		1	616	1
		1	616A	1
65C33034-2		1	618	2
		1	618A	2
65C33034-3		1	616L	1
		1	618L	1
65C33034-5		1	616B	1
65C33034-6		1	618B	2
65C33034-7		1	616N	1
		1	618N	1
65C33043-1		1	481	1
65C33043-2		1	482	1
65C33123-10		1	725I	1
65C33123-11		1	725H	1
65C33123-12		1	755G	1
65C33123-9		1	750J	1
65C33609-1		1	708	1
		1	725F	1
		1	725J	1
65C33609-2		1	708A	1
		1	725G	1
		1	725K	1
65C33609-5		1	708B	1
		1	750H	1
		1	750K	1
65C33609-6		1	708C	1
		1	750I	1
		1	750L	1
65C3518-1		2	10F	1
65C35318-1		2	10I	1
69-30479-1		1	200E	1
69-30479-7		1	200F	1
69-37234-28		1	485A	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	550A	2
		1	600A	4
69-37234-29		1	555A	2
69-37234-30		1	485B	2
		1	550B	2
		1	600B	4
69-37234-31		1	555B	2
69-57701-2		1	945	1
69-57703-1		1	725	1
		1	725A	1
69-57704-1		1	635	1
69-57704-2		1	640	1
69-57704-3		1	565	3
69-57704-5		1	635A	1
69-57704-6		1	640A	1
69-57705-3		1	660	1
69-57705-4		1	665	1
69-57705-5		1	660A	1
69-57705-6		1	665A	1
69-57705-7		1	660B	1
69-57705-8		1	665B	1
69-57706-2		1	500	3
69-67237-1		1	670	AR
69-73101-1		2	10	1
69-73101-2		2	10A	1
		2	10B	1
69-73102-1		1	120	1
69-73102-2		1	130	1
69-73102-3		1	65	1
69-73102-5		1	135A	1
69-73712-1		1	750	1
		1	750C	1
69-73712-2		1	755	1
		1	755C	1
69-73736-1		1	200	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	200B	1
		1	200D	1
69-73736-2		1	205	1
		1	205B	1
		1	205D	1
69-73736-3		1	220	1
69-73736-4		1	225	1
69-74570-1		1	805	1
69-74570-13		1	843	1
		1	843A	1
69-74570-14		1	844	1
		1	844A	1
69-74570-15		1	805A	1
69-74570-16		1	810A	1
69-74570-17		1	835A	1
		1	835C	1
69-74570-18		1	840A	1
		1	840C	1
69-74570-2		1	810	1
69-74570-3		1	835	1
		1	835B	1
69-74570-4		1	840	1
		1	840B	1
69-74570-5		1	841	1
69-74570-6		1	842	1
69-77062-1		1	90C	1
		1	90E	1
		1	90F	1
69-77100-1		1	501	2
		1	501A	2
69-77184-1		1	566	2
		1	566A	3
69C26304-14		1	265B	1
AN960-10L		1	715	3
		1	740	3

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	740A	4
AN960-516L		1	927	4
AN960XC9		1	935	2
BACB10CL9		1	125	1
BACB10Y4T		1	55	1
		1	115	1
BACB10Y6M		1	60A	1
BACB10Y6T		1	60	1
BACB28AA4B013		1	230	1
BACB28AA4B024		1	240	1
BACB28AA6B016		1	235	1
BACB28AA6B018		1	250	4
BACB28AA6B022		1	280	2
BACB28AA6B026		1	165	2
		1	245	1
BACB28AA8B027		1	170	2
BACB28AA9B020		1	160	4
BACB28AAB012		1	330	2
BACB28AK04-030		1	85	1
		1	85A	1
BACB28AK04-040		1	25	1
		1	25A	1
BACB28AK04-070		1	87	2
		1	87A	2
		1	515A	2
		1	515C	2
		1	515D	2
BACB28AK04-200		1	606	1
BACB28AK04-285		1	491	1
BACB28AK04-420		1	561	1
BACB28U11C043		1	175	2
BACB28U8C027		1	335	1
BACB28U8C037		1	255	2
BACB28U8C045		1	285	1
BACB28W6D018		1	690	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB28W6D040		1	695	1
BACB28Y6C040		1	143	2
		1	143A	2
BACB30LH3-6		1	620A	4
BACB30LR3-6		1	365	56
		1	370	4
		1	900	28
		1	905	4
		1	910	6
		1	915	2
		1	920	1
BACB30MY6K5		1	210	4
BACB30NN3K5		1	915A	2
BACB30NN3K6		1	620	4
		1	645A	6
BACB30NN3K7		1	710	3
		1	730	2
		1	730A	3
		1	910A	6
		1	920A	1
BACB30NN3K8		1	645	7
BACB30NN3K9		1	735	1
BACB30NN5K11		1	903	1
BACB30NN5K8		1	902	3
BACB30NW5K		1	780A	4
		1	785A	2
BACB30NW5K7		1	775	4
		1	825	1
BACB30NW6K2		1	950	1
BACB30NW6K3		1	960	1
BACB30NW6K4		1	770	1
BACB30NW6K5		1	765	14
		1	907	6
BACB30NW6K6		1	760	2
		1	955	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB30NW6K7		1	825A	1
BACB30VF3K3		1	930	2
BACC30AB5C		1	800	4
		1	830	1
BACC30AB6C		1	830A	1
BACC30M		1	215	4
		1	795	17
BACC30M6		1	929	6
		1	965	3
BACF3T03M6G06		1	790	4
BACN10JC3CD		1	495	1
		1	520	2
		1	570	1
		1	595	1
		1	610	1
		1	630	4
		1	655	7
		1	655A	6
		1	940	2
BACN10JS3A2CD		1	455	2
BACN10JS3A3CD		1	440	12
BACN10JS3A4CD		1	435	36
		1	875	10
BACN10KE3DCD		1	445	4
		1	870	2
BACN10KE3E2CD		1	865	2
BACN10KE3E3CD		1	880	1
BACN10YF32CD		1	860	26
BACN10YF33CD		1	450	5
BACN10YF34CD		1	460	1
BACN10YR4CD		1	496	2
		1	520A	2
		1	611	1
BACR15BA3AD		1	430	120
		1	855	82

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACR15BA6AD7		1	709	1
BACR15CE6AD8		1	644	1
BACR15GA5-4		1	820	3
BACR15GA6-5		1	820A	3
BACS40R020C128F		1	425	AR
		1	425A	AR
BACS40R025E130F		1	420	AR
BACS40R08F120F		1	415B	1
		1	415C	1
		1	415D	1
BACS40R08F155F		1	400L	2
		1	400M	2
		1	400N	2
		1	405L	2
		1	405M	2
		1	405N	2
BACS40R08F175F		1	395B	1
		1	395C	1
		1	395D	1
		1	410B	1
		1	410C	1
		1	410D	1
BACS40R08F88F		1	400F	2
		1	400G	2
		1	400H	2
		1	405F	2
		1	405G	2
		1	405H	2
BACS40U6L3		1	145	1
		1	145A	1
BACW10BP4DP		1	15	1
		1	15A	1
		1	75	1
		1	75A	1
		1	510A	4

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACW10BP6DP		1	141	2
		1	141A	2
BACW10P183AL		1	480	2
		1	510	4
		1	590	2
		1	625	4
		1	650	7
		1	650A	6
		1	546	1
BACW10P186C		1	546	1
BACW10P33AL		1	480A	2
		1	591	2
BACW10P35AL		1	545	4
BACW10P44AL		1	540	2
KJB270306B		1	482C	2
		1	617	4
		1	619	4
KJB291906B		1	695A	2
NAS1804-3		1	720	3
		1	745	3
		1	745A	4
NAS1804-5		1	928	4
NAS1805-4		1	571	1
NAS1805-4N		1	20	1
		1	20A	1
		1	80	1
NAS1805-6N		1	80A	1
		1	147	2
NAS1805-6N		1	147A	2
		1	147A	2
NAS500FC10		1	465	4
		1	925	1
NAS509-5		1	40	1
		1	100	1
NAS509-6		1	50	1
		1	110	1
NAS559-1		1	35	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	95	1
NAS559-2		1	45	1
		1	105	1
NAS6703-14		1	505	2
NAS6703-30		1	475	1
NAS6703-35		1	585	1
NAS6703-65		1	535	1
NAS6704-13		1	505A	2
		1	505C	2
		1	505D	2
NAS6704-14		1	505B	2
NAS6704-15		1	70	1
		1	70A	1
NAS6704-18		1	10	1
		1	10A	1
NAS6704-35		1	586	1
NAS6704-46		1	476	1
NAS6704-66		1	536	1
NAS6706-21		1	139	2
		1	139A	2
NAS74A3E011P		1	515	2
		1	515B	2
NAS74A3E113P		1	490	1
		1	605	1
NAS74A3E315P		1	560	1
P35940		1	135B	1
VTA06430		1	135C	1

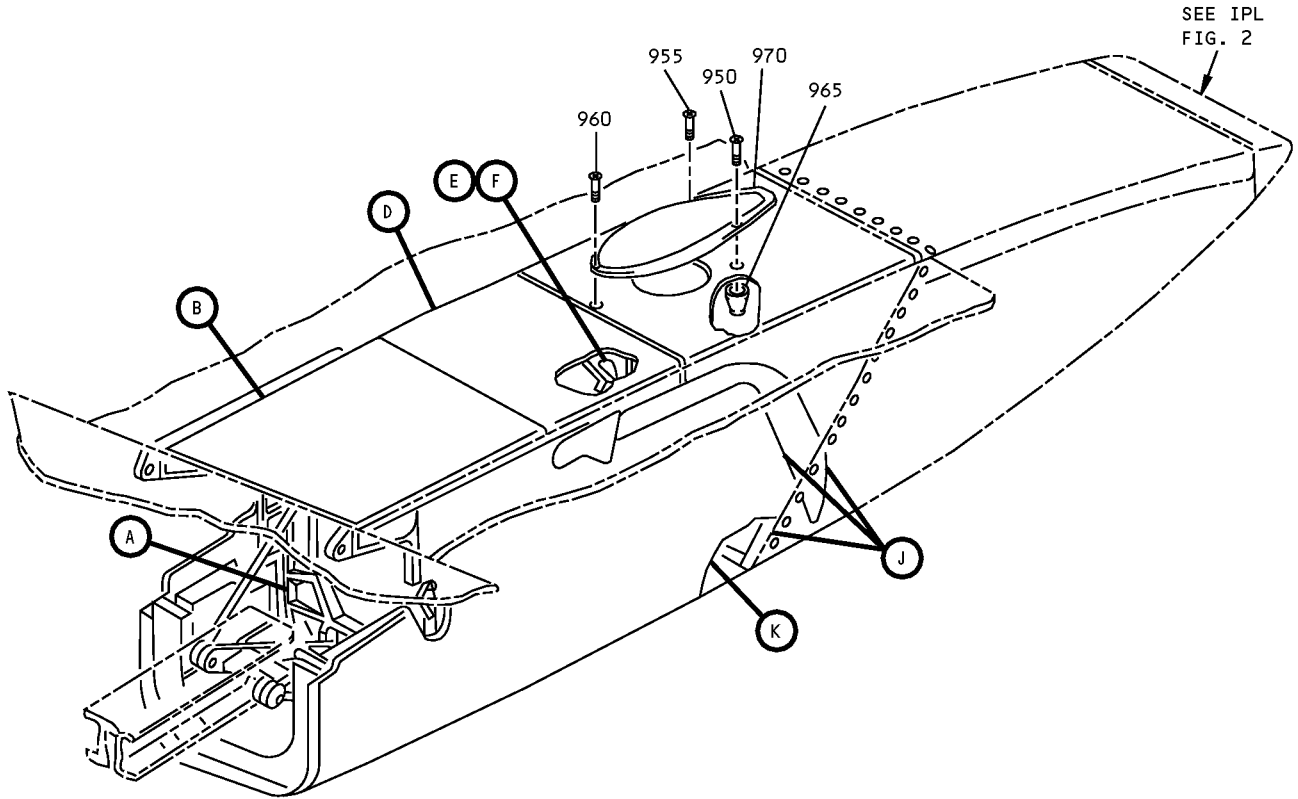
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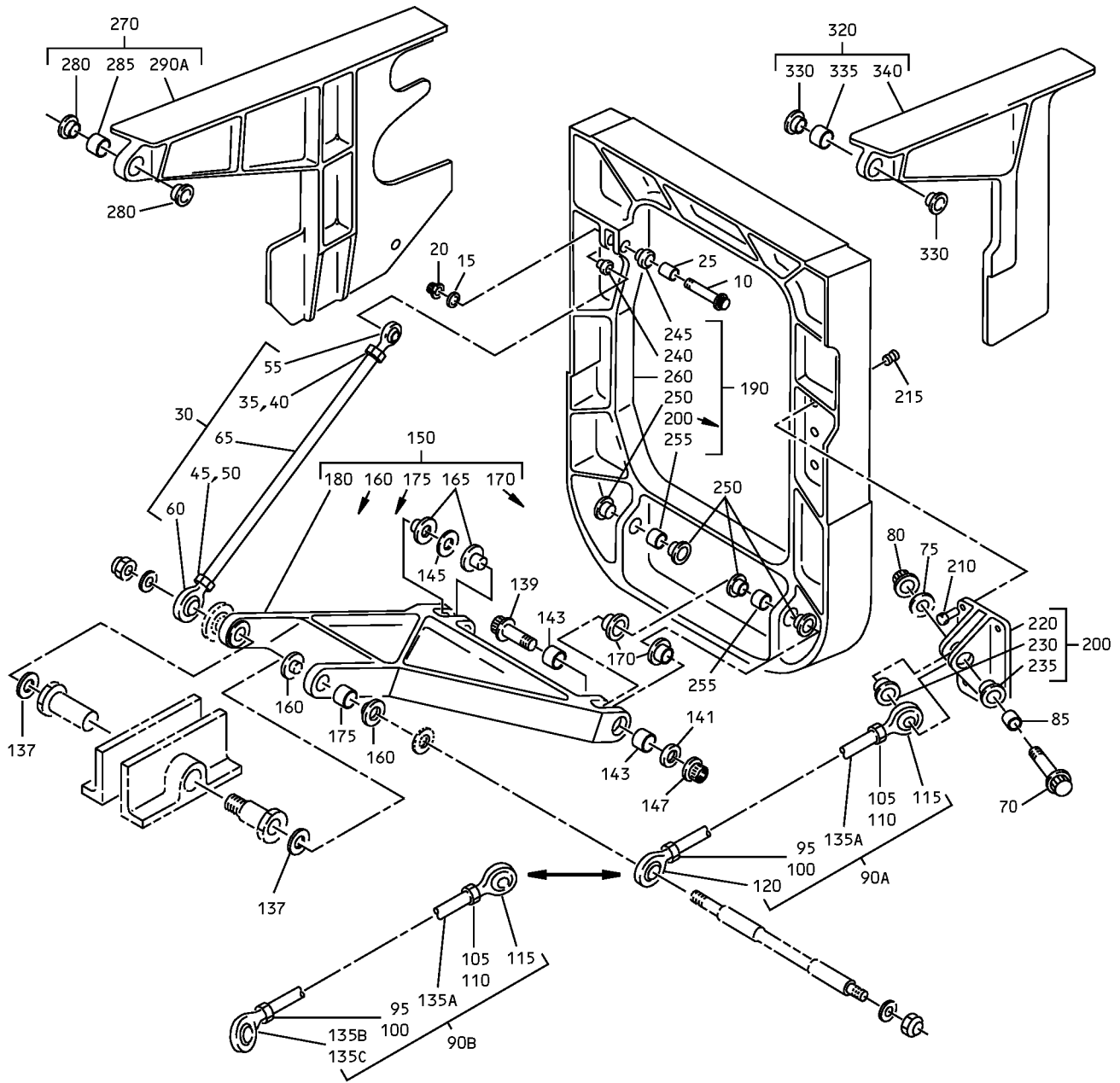
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Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 1 of 10)

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A

Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 2 of 10)

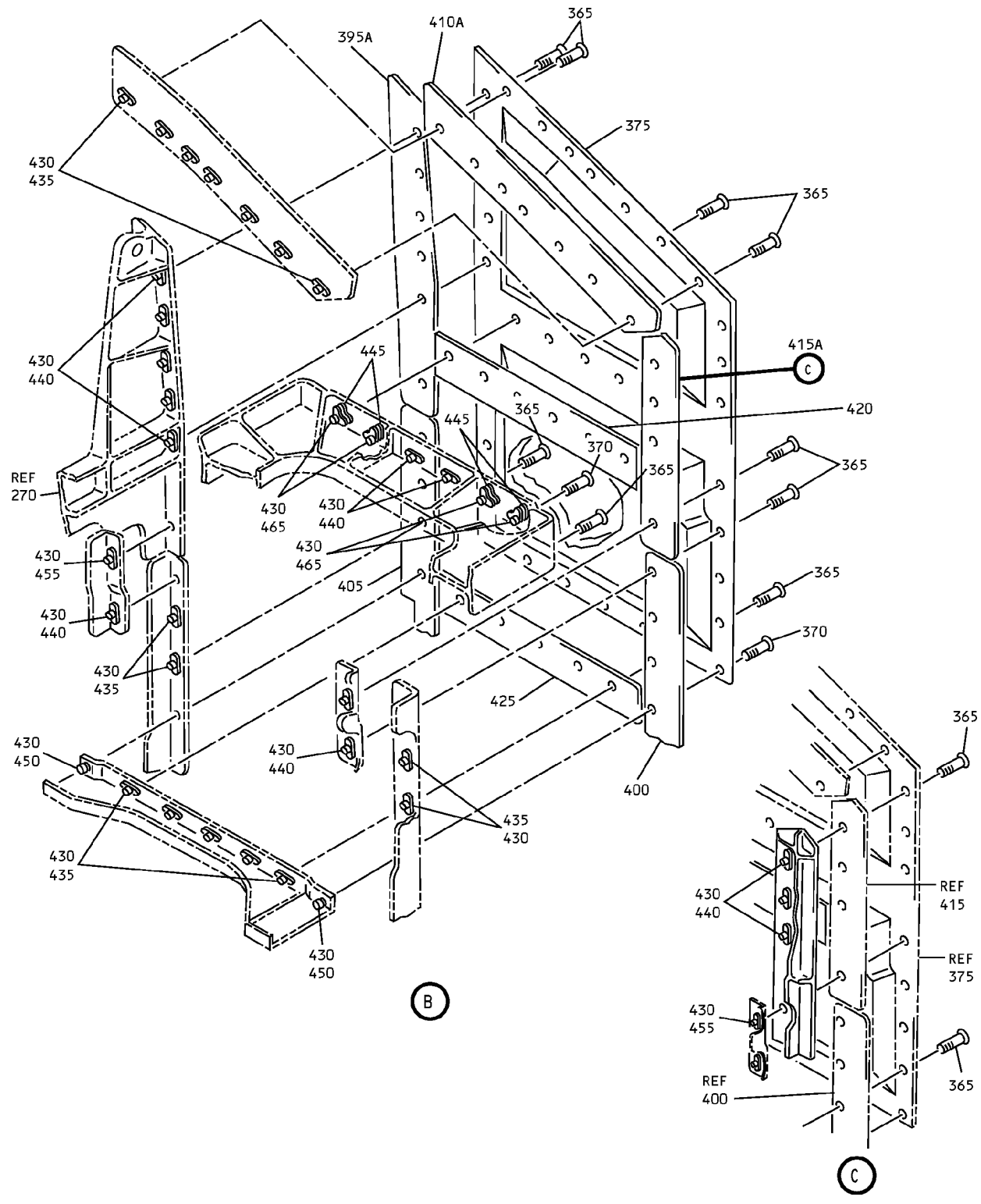
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Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 3 of 10)

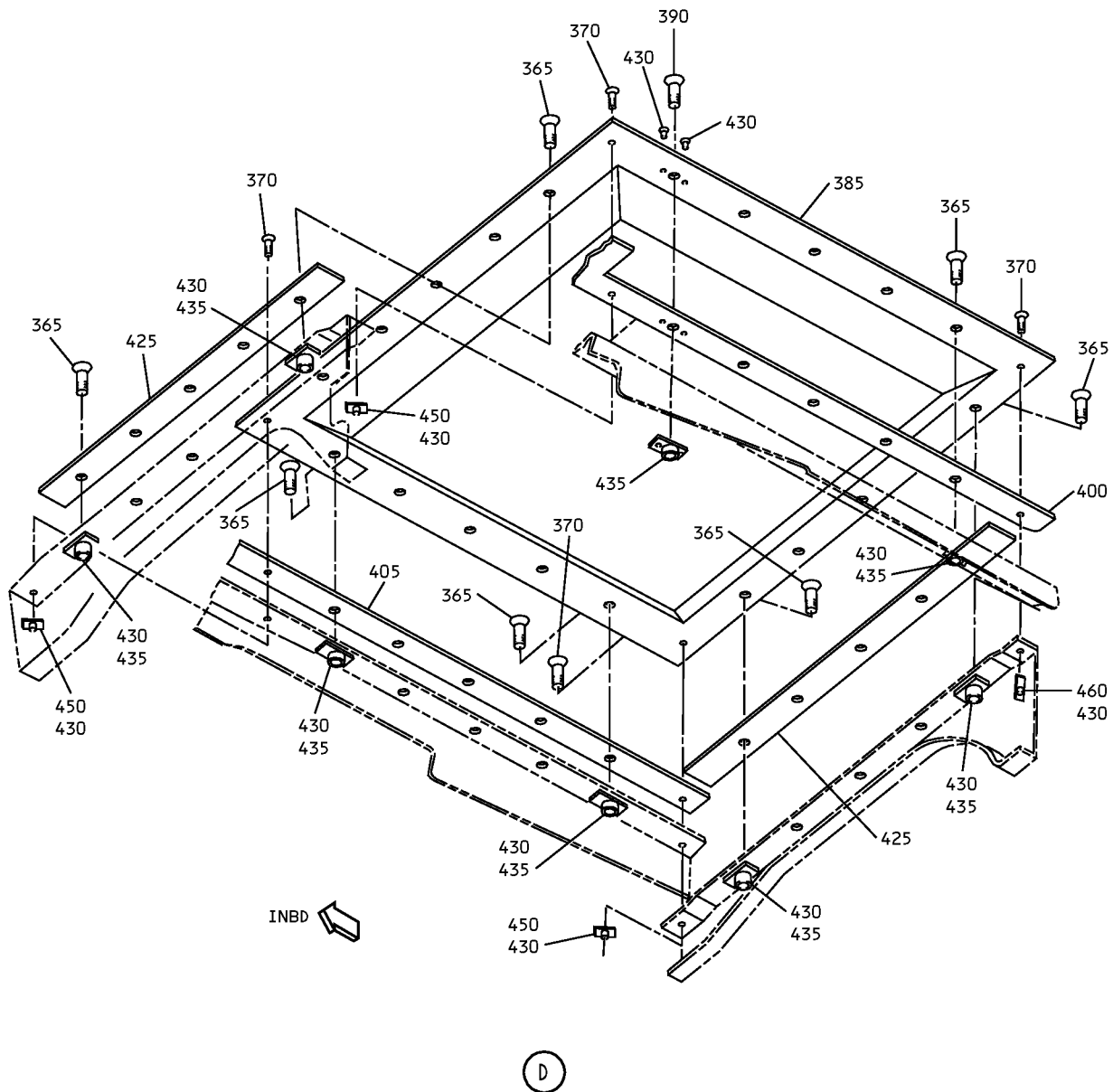
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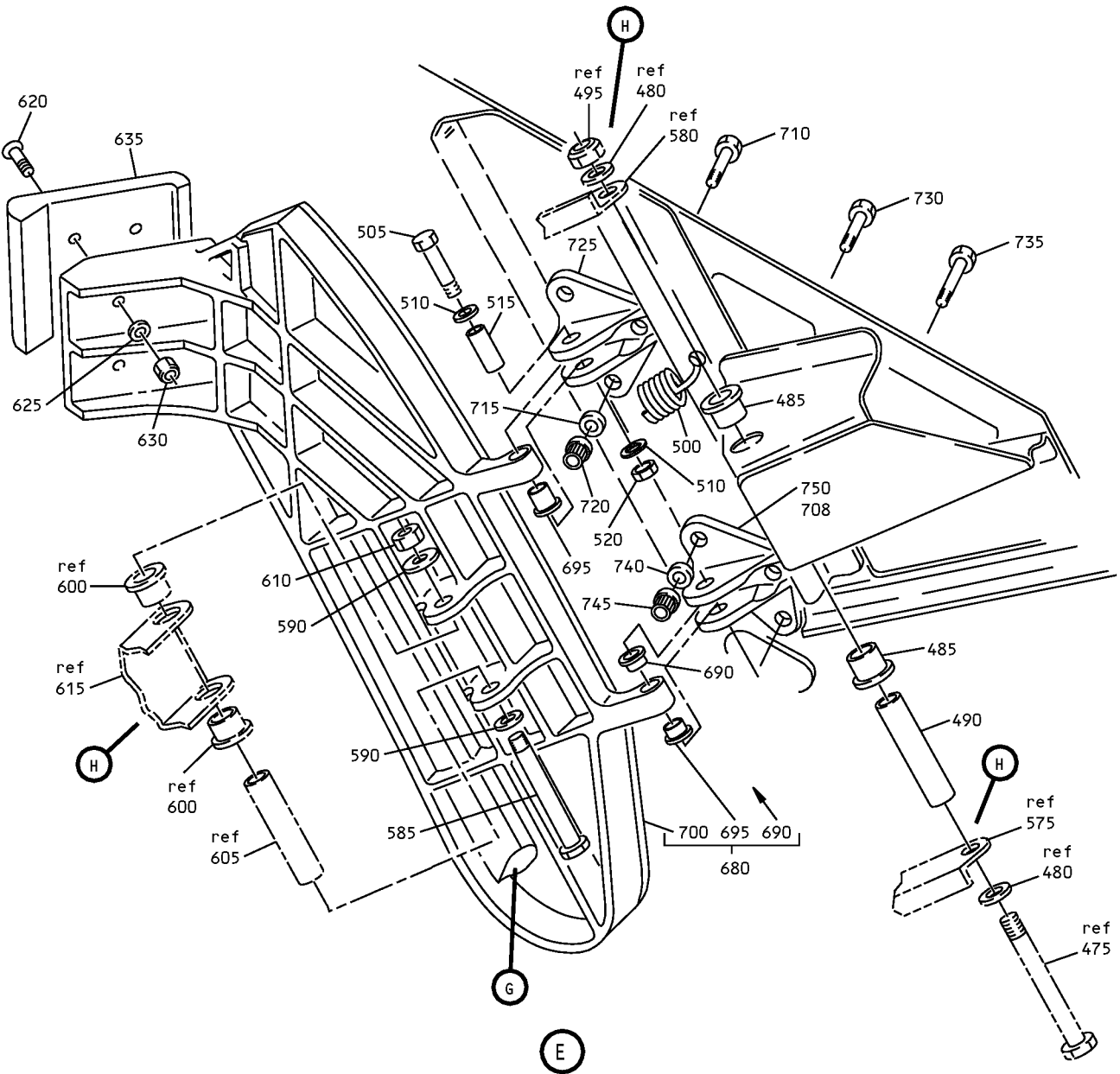
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IPL Figure 1 (Sheet 4 of 10)

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Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 5 of 10)

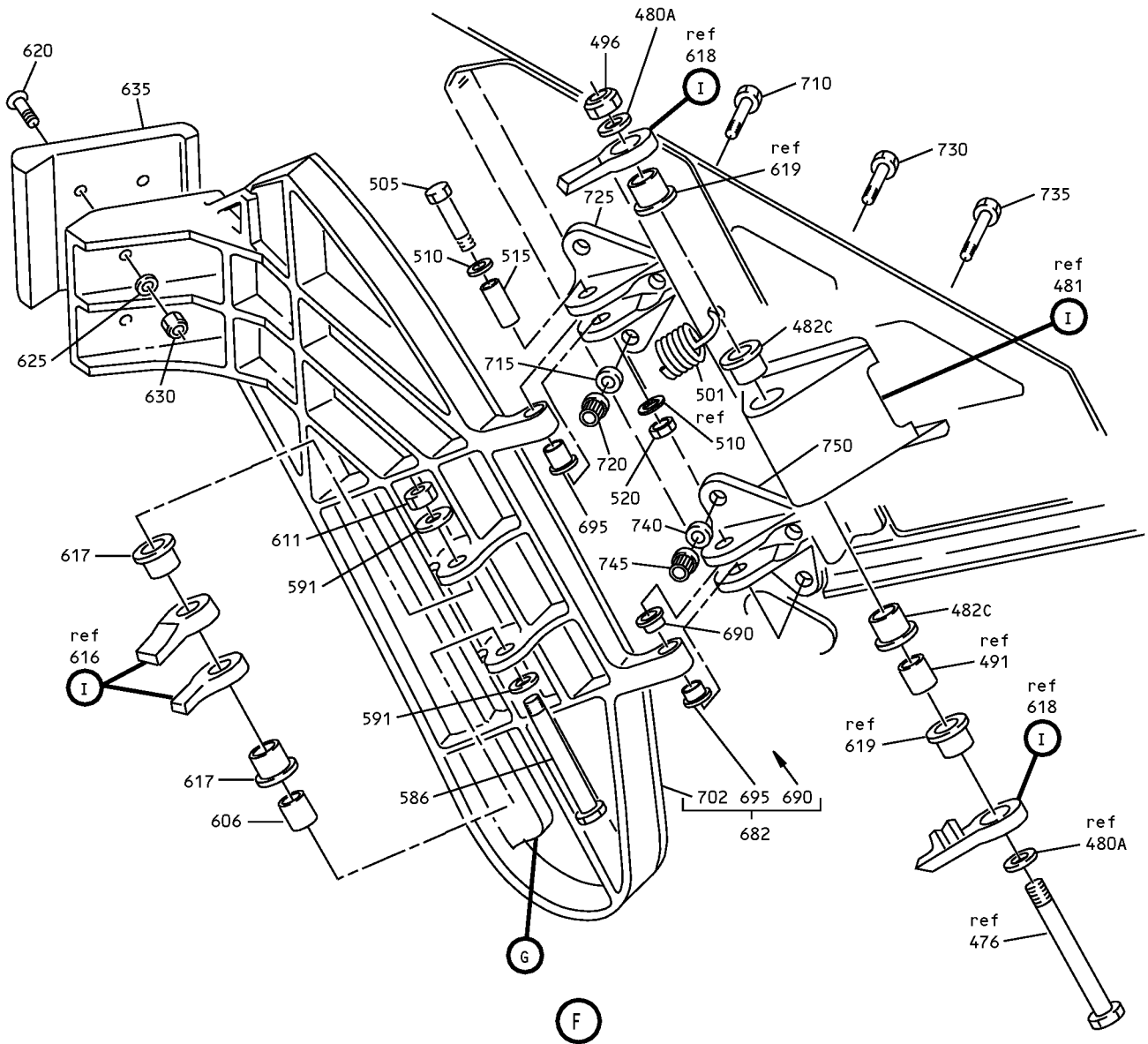
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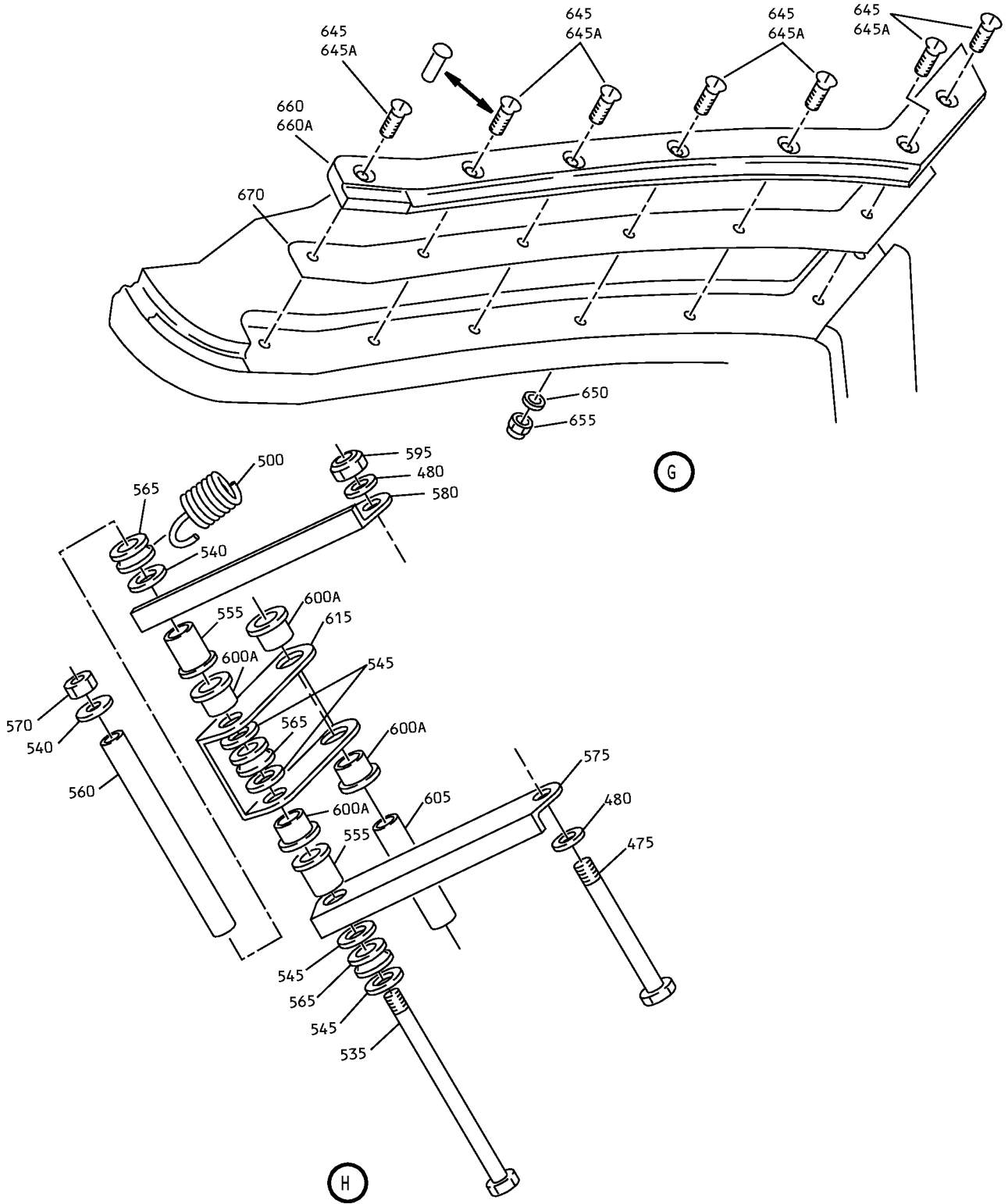
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IPL Figure 1 (Sheet 6 of 10)

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Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 7 of 10)

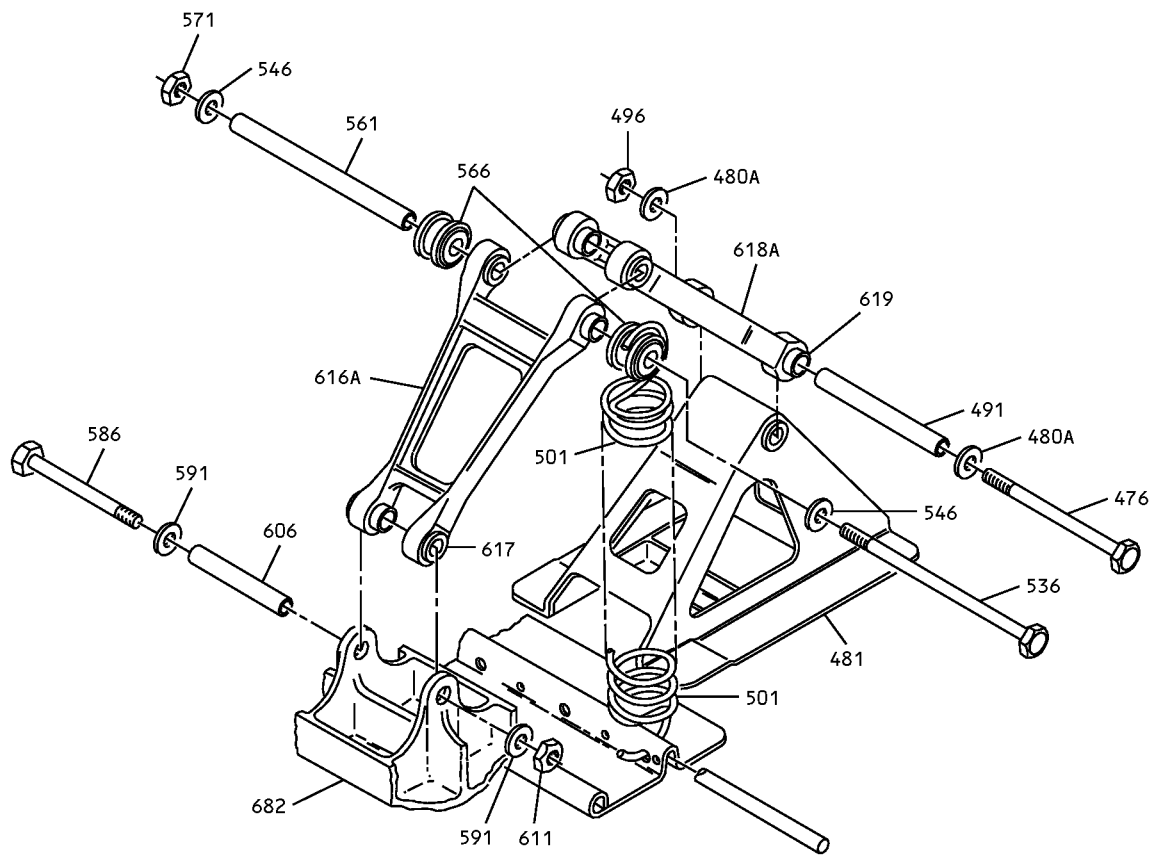
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I

Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 8 of 10)

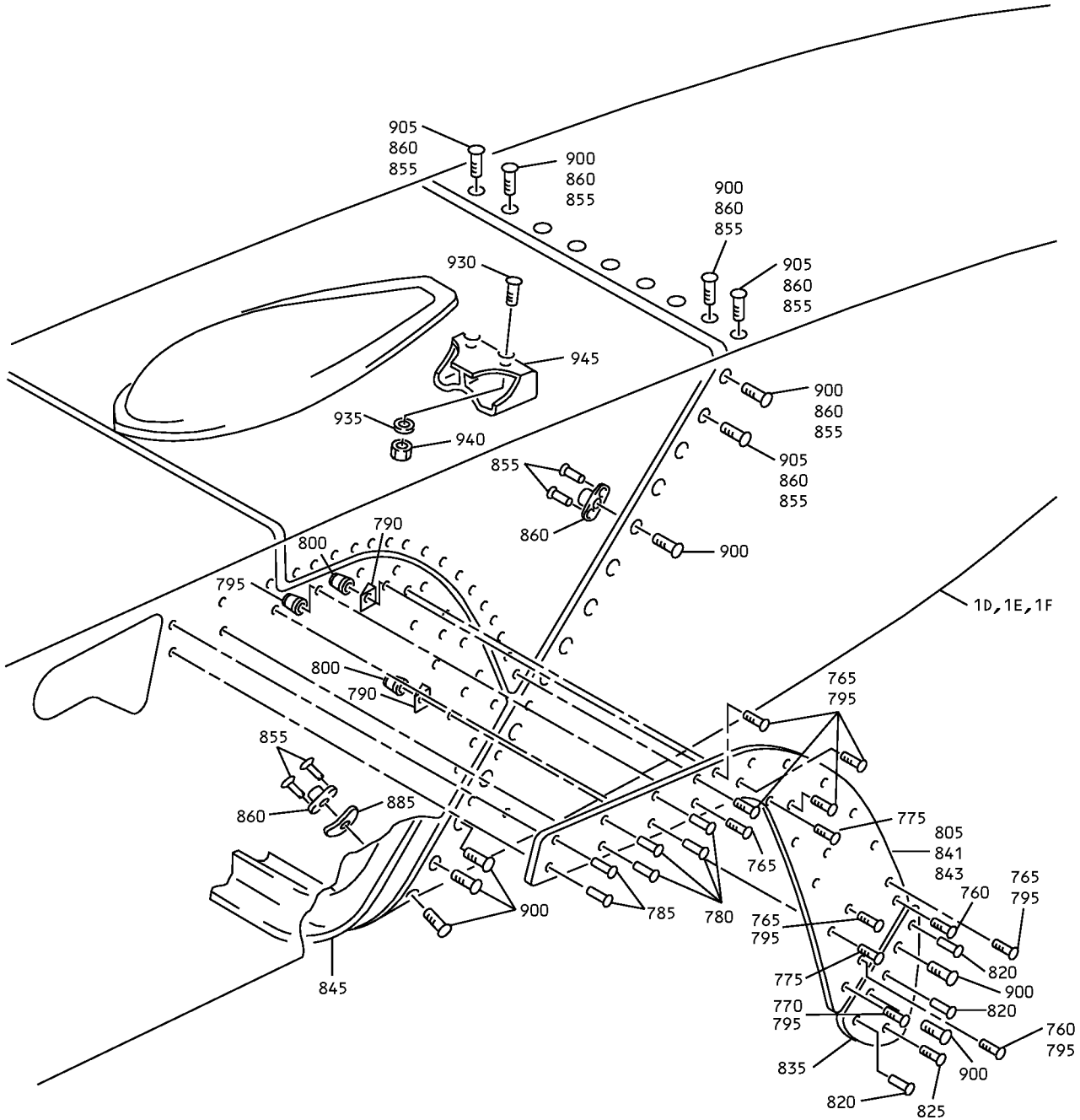
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J

Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 9 of 10)

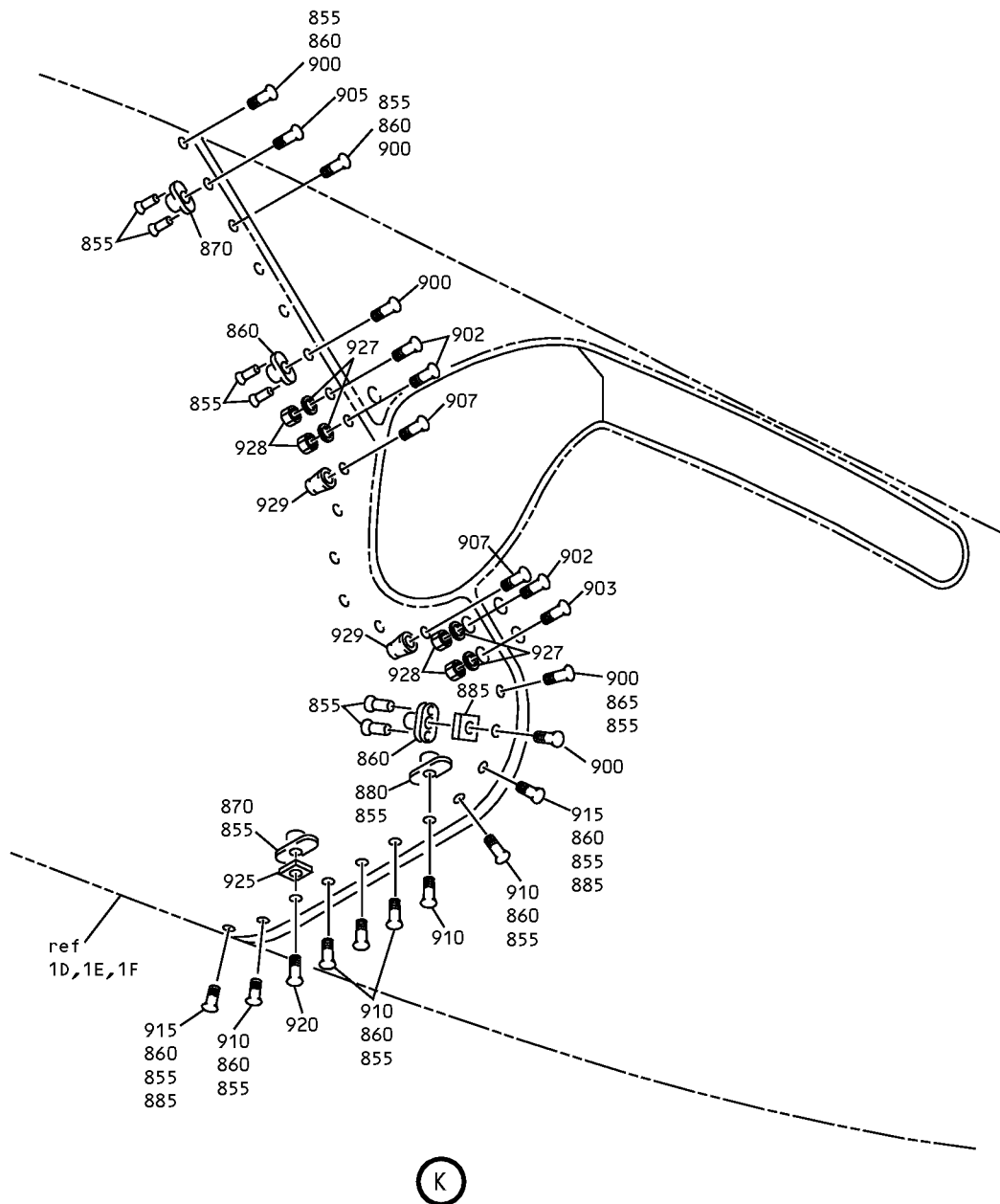
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Inbd Flap Outbd Track Fairing Assembly
IPL Figure 1 (Sheet 10 of 10)



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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1-												
1	65C26300-3SP											
1A	65C26300-55SP											
-1B	65C26300-59SP											
-1C	65C26300-63SP											
-1D	65C26300-5SP											
-1E	65C26300-57SP											
-1F	65C26300-69SP											
-1G	65C26300-73SP											
-1H	65C26300-75SP											
-1I	65C26300-77SP											
-1J	65C26300-87SP											
-1K	65C26300-93SP											

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1L	65C26300-99SP									O	RF
-1M	65C26300-105SP									Q	RF
-1N	65C26300-107SP									S	RF
-1P	65C25300-111SP										RF
-5	65C26300-4SP									B	RF
-5A	65C26300-56SP										
-5B	65C26300-60SP									D	RF
-5C	65C26300-64SP									F	RF
-5D	65C26300-6SP										RF
-5E	65C26300-58SP										RF
-5F	65C26300-70SP										RF
-5G	65C26300-74SP									H	RF
-5H	65C26300-76SP									J	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-5I	65C26300-78SP										RF
-5J	65C26300-88SP									L	RF
-5K	65C26300-94SP									N	RF
-5L	65C26300-100SP									P	RF
-5M	65C26300-106SP									R	RF
-5N	65C26300-108SP									T	RF
-5P	65C26300-112SP										RF
10	NAS6704-18									A-N	1
-10A	NAS6704-18									O-T	1
15	BACW10BP4DP									A-N	1
-15A	BACW10BP4DP									O-T	1
20	NAS1805-4N									A-N	1
-20A	NAS1805-4N									O-T	1
25	BACB28AK04-040									A-N	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-25A	BACB28AK04-040		.							O-T	1
30	65C26300-7		.							A-N	1
-30A	65C26300-7		.							O-T	1
35	NAS559-1		.	.							1
40	NAS509-5		.	.							1
45	NAS559-2		.	.							1
50	NAS509-6		.	.							1
55	BACB10Y4T		.	.							1
60	BACB10Y6T		.	.							1
-60A	BACB10Y6M		.	.							1
65	69-73102-3		.	.							1
70	NAS6704-15		.							A-N	1
-70A	NAS6704-15		.							O-T	1
75	BACW10BP4DP		.							A-N	1
-75A	BACW10BP4DP		.							O-T	1
80	NAS1805-4N		.							A-N	1
-80A	NAS1805-4N		.							O-T	1
85	BACB28AK04-030		.							A-N	1
-85A	BACB28AK04-030		.							O-T	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-87	BACB28AK04-070		.	BUSHING						K, L	2
				(LIMITED USAGE)							
-87A	BACB28AK04-070		.	BUSHING						M, N	2
-90	65C26300-8			DELETED							
90A	65C26300-45		.	STRUT ASSY-OUTBD							1
-90B	69-73102-5			DELETED							
-90C	69-77062-1		.	STRUT ASSY-OUTBD						I, J	1
				(LIMITED USAGE)							
-90D	65C26300-45		.	STRUT ASSY-OUTBD						I, J	1
				(LIMITED USAGE)							
-90E	69-77062-1		.	STRUT ASSY-OUTBD						K-N	1
-90F	69-77062-1		.	STRUT ASSY-OUTBD						O-T	1
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
95	NAS559-1		.	LOCK-ROD END							1
100	NAS509-5		.	NUT							1
105	NAS559-2		.	LOCK-ROD END							1
110	NAS509-6		.	NUT							1
115	BACB10Y4T		.	ROD END							1
120	69-73102-1		.	ROD END ASSY							1
				(USED ON ITEMS 90A, 90D)							
-125	BACB10CL9		.	BEARING-ROD END							1
-130	69-73102-2		.	ROD END							1
-135	69-73102-4			DELETED							
135A	69-73102-5		.	STRUT							1
135B	P35940		.	ROD END ASSY							1
				(V57606)							
				(OPT ITEM 135C)							
				(USED ON ITEMS 90C, 90E, 90F)							
-135C	VTA06430		.	ROD END ASSY							1
				(V06710)							
				(OPT ITEM 135B)							
				(USED ON ITEMS 90C, 90E, 90F)							
137	65C26300-15		.	SHIM-WASHER						A-N	2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-137A	65C26300-15		. SHIM-WASHER (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	2
139	NAS6706-21		. BOLT	A-N	2
-139A	NAS6706-21		. BOLT (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	2
141	BACW10BP6DP		. WASHER	A-N	2
-141A	BACW10BP6DP		. WASHER (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	2
143	BACB28Y6C040		. BUSHING	A-N	2
-143A	BACB28Y6C040		. BUSHING (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	2
145	BACS40U6L3		. SHIM-WASHER	A-N	1
-145A	BACS40U6L3		. SHIM-WASHER (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	1
147	NAS1805-6N		. NUT	A-N	2
-147A	NAS1805-6N		. NUT (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	2
150	65C26306-7		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 150A)	A, C, E, G	1
-150A	65C26306-1		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 150)	A, C, E, G	1
-150B	65C26306-7		. FITTING ASSY-LWR MT ATTACH (LIMITED USAGE)	I	1
-150C	65C26306-1		. FITTING ASSY-LWR MT ATTACH (LIMITED USAGE)	I	1
-150D	65C26306-15		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 150G) (LIMITED USAGE)	I	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-150E	65C26306-15		.							K, M	1
-150F	65C26306-15		.							O-T	1
-150G	65C26306-19		.							I	1
-150H	65C26306-19		.							K	1
-150I	65C26306-19		.							M	1
-150J	65C26306-19		.							O-T	1
155	65C26306-8		.							B, D, F, H	1
-155A	65C26306-2		.							B, D, F, H	1
-155B	65C26306-8		.							J	1
-155C	65C26306-2		.							J	1
-155D	65C26306-16		.							J	1
-155E	65C26306-16		.							L, N	1
-155F	65C26306-16		.							O-T	1
-155G	65C26306-20		.							J	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-155H	65C26306-20		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 155E) (LIMITED USAGE)	L	1
-155I	65C26306-20		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 155E)	N	1
-155J	65C26306-20		. FITTING ASSY-LWR MT ATTACH (OPT ITEM 155F) (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)	O-T	1
160	BACB28AA9B020		. . BUSHING		4
165	BACB28AA6B026		. . BUSHING		2
170	BACB28AA8B027		. . BUSHING		2
175	BACB28U11C043		. . SLEEVE		2
180	65C26306-3		. . FITTING (USED ON ITEMS 150A, 150C)		1
-180A	65C26306-9		. . FITTING (USED ON ITEMS 150, 150B)		1
-180B	65C26306-17		. . FITTING (USED ON ITEMS 150D THRU 150F)		1
-180C	65C26306-21		. . FITTING (USED ON ITEMS 150G THRU 150J)		1
-185	65C26306-4		. . FITTING (USED ON ITEMS 155A, 155C)		1
-185A	65C26306-10		. . FITTING (USED ON ITEMS 155, 155B)		1
-185B	65C26306-18		. . FITTING (USED ON ITEMS 155D THRU 155F)		1
-185C	65C26306-22		. . FITTING (USED ON ITEMS 155G THRU 155T)		1
190	65C26304-7		. BULKHEAD ASSY (OPT ITEM 190A) (LIMITED USAGE)	A	1
-190A	65C26304-1		. BULKHEAD ASSY (OPT ITEM 190) (LIMITED USAGE)	A	1
-190B	65C26304-11		. BULKHEAD ASSY (LIMITED USAGE)	A, E, I	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-190C	65C26304-11		.	BULKHEAD	ASSY					C, G	1
-190D	65C30918-1			DELETED							
-190E	65C26304-15		.	BULKHEAD	ASSY					E	1
				(LIMITED	USAGE)						
-190F	65C26304-15		.	BULKHEAD	ASSY					G, I, K, M, O, Q, S	1
-194	65C26304-2		.	BULKHEAD	ASSY					B	1
				(OPT ITEM	195)						
				(LIMITED	USAGE)						
-195	65C26304-8		.	BULKHEAD	ASSY					B	1
				(OPT ITEM	195A)						
				(LIMITED	USAGE)						
-195A	65C26304-12		.	BULKHEAD	ASSY					B, F, J	1
				(LIMITED	USAGE)						
-195B	65C26304-12			DELETED							
-195C	65C26304-12		.	BULKHEAD	ASSY					D, H	1
-195D	65C30918-2			DELETED							
-195E	65C26304-16		.	BULKHEAD	ASSY					F	1
				(LIMITED	USAGE)						
-195F	65C26304-16		.	BULKHEAD	ASSY					H, J, L, N, P, R, T	1
200	69-73736-1		.	SUPPORT	ASSY-OUTBD	STRUT				A	1
-200A	65C30479-1		.	SUPPORT	ASSY-OUTBD	STRUT				C, E	1
				(OPT ITEM	200B)						
-200B	69-73736-1		.	SUPPORT	ASSY-OUTBD	STRUT				C, E	1
				(OPT ITEM	200A)						
-200C	65C30479-1		.	SUPPORT	ASSY-OUTBD	STRUT				G	1
				(OPT ITEM	200D)						
				(LIMITED	USAGE)						
-200D	69-73736-1		.	SUPPORT	ASSY-OUTBD	STRUT				G	1
				(OPT ITEM	200C)						
				(LIMITED	USAGE)						
-200E	69-30479-1		.	SUPPORT	ASSY					I, K, M, O, Q, S	1
				(OPT ITEM	200F)						
-200F	69-30479-7		.	SUPPORT	ASSY					I, K, M, O, Q, S	1
				(OPT ITEM	200E)						
-205	69-73736-2		.	SUPPORT	ASSY-OUTBD	STRUT				B	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-205A	65C30479-2		.							D, F	1
-205B	69-73736-2		.							D, F	1
-205C	65C30479-2		.							H	1
-205D	69-73736-2		.							H	1
-205E	65C30479-2		.							J, L, N, P, R, T	1
-205F	65C30479-8		.							J, L, N, P, R, T	1
210	BACB30MY6K5		.								4
215	BACC30M		.								4
			-----*								
220	69-73736-3		..								1
-220A	65C30479-3		..								1
-220B	65C30479-9		..								1
-225	69-73736-4		..								1
-225A	65C30479-4		..								1
-225B	65C30479-10		..								1
230	BACB28AA4B013		..								1
235	BACB28AA6B016		..								1
240	BACB28AA4B024		..								1
245	BACB28AA6B026		..								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
250	BACB28AA6B018		.	.							4
255	BACB28U8C037		.	.							2
260	65C26304-9		.	.							1
-260A	65C26304-5		.	.							1
-260B	65C26304-13		.	.							1
-260C	65C30918-3										
-260D	65C26304-17		.	.							1
-265	65C26304-10		.	.							1
-265A	65C26304-6		.	.							1
-265B	69C26304-14		.	.							1
-265C	65C30918-4										
-265D	65C26304-18		.	.							1
-270	65C26302-1										
270A	65C26302-7		.						A, E, G		1
-270B	65C26302-7		.						C		1
-270C	65C26302-7		.						I, K		1
-270D	65C30937-1		.						A, E, G		1
-270E	65C30937-1		.						I, K		1
-270F	65C30937-1										
-270G	65C26302-7		.						M, O, Q, S		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-275	65C26302-2										
275A	65C26302-8								. FITTING ASSY-INBD (OPT ITEM 275D)	B, F, H	1
-275B	65C26302-8								. FITTING ASSY-INBD	D	1
-275C	65C26302-8								. FITTING ASSY-INBD (OPT ITEM 275E) (LIMITED USAGE)	I, K	1
-275D	65C30937-2								. FITTING ASSY-INBD (OPT ITEM 275A)	B, F, H	1
-275E	65C30937-2								. FITTING ASSY-INBD (OPT ITEM 275C) (LIMITED USAGE)	J, L	1
-275F	65C30937-2								DELETED		
-275G	65C26302-8								. FITTING ASSY-INBD	N, P, R, T	1
280	BACB28AA6B022								. . BUSHING		2
285	BACB28U8C045								. . SLEEVE		1
-290	65C26302-5								DELETED		
290A	65C26302-9								. . FITTING (USED ON ITEMS 270A, 270B, 270C)		1
-290B	65C30937-3								. . FITTING (USED ON ITEMS 270D, 270E)		1
-295	65C26302-6								DELETED		
295A	65C26302-10								. . FITTING (USED ON ITEMS 275A, 275B, 275C)		1
-295B	65C30937-4								. . FITTING (USED ON ITEMS 275D, 275E)		1
-320	65C26303-1								DELETED		
320A	65C26303-7								. FITTING ASSY-OUTBD (OPT ITEM 320D)	A, C, E, G	1
-320B	65C26303-7								. FITTING ASSY-INBD (OPT ITEM 320H) (LIMITED USAGE)	I	1
-320C	65C26303-7								. FITTING ASSY-INBD (OPT ITEM 320I)	K, M, O, Q, S	1
-320D	65C30938-1								. FITTING ASSY-INBD (OPT ITEM 320A, 320G)	A, C, E, G	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-320E	65C30938-1									I	1
-320F	65C30938-1										
-320G	65C30938-9									A, C, E, G	1
-320H	65C30938-9									I	1
-320I	65C30938-9									K, M, O	1
-325	65C26303-2									, Q, S	
325A	65C26303-8									B, D, F, H	1
-325B	65C26303-8									J	1
-325C	65C26303-8									L, N, P, R, T	1
-325D	65C30938-2									B, D, F, H	1
-325E	65C30938-2									J	1
-325F	65C30938-2										
-325G	65C30938-10									B, D, F, H	1
-325H	65C30938-10									J	1
-325I	65C30938-10									L, N, P, R, T	1
330	BACB28AAB012										2
335	BACB28U8C027										1
-340	65C26303-5										
340A	65C26303-9										1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-340B	65C30938-3		. .	FITTING							1
				(USED ON ITEMS 320D, 320E)							
-340C	65C30938-7		. .	FITTING							1
				(USED ON ITEMS 320G THRU 320I)							
-345	65C26303-6			DELETED							
345A	65C26303-10		. .	FITTING							1
				(USED ON ITEMS 325A, 325B, 325C)							
-345B	65C30938-4		. .	FITTING							1
				(USED ON ITEMS 325D, 325E)							
-345C	65C30938-8		. .	FITTING							1
				(USED ON ITEMS 325G THRU 325I)							
365	BACB30LR3-6		. .	BOLT							56
370	BACB30LR3-6		. .	BOLT							4
375	65C26301-43		. .	PANEL-FWD ACCESS					A, C, E, G		1
-380	65C26301-44		. .	PANEL-FWD ACCESS					B, D, F, H		1
-381	65C26301-77		. .	PANEL-FWD ACCESS					I, K, M, O, Q, S		1
-382	65C26301-78		. .	PANEL-FWD ACCESS					J, L, N, P, R, T		1
385	65C26301-45		. .	PANEL-AFT ACCESS					A, C, E, G		1
-390	65C26301-46		. .	PANEL-AFT ACCESS					B, D, F, H		1
-391	65C26301-75		. .	PANEL-AFT ACCESS					I, K, M		1
-392	65C26301-76		. .	PANEL-AFT ACCESS					J, L, N		1
-395	65C26300-20			DELETED							
395A	65C26300-42		. .	SHIM					A, B		1
				(OPT ITEM 395B)							
				(LIMITED USAGE)							
-395B	BACS40R08F175F		. .	SHIM					A, B		1
				(OPT ITEM 395A)							
				(LIMITED USAGE)							
-395C	BACS40R08F175F		. .	SHIM					C-N		1
-395D	BACS40R08F175F		. .	SHIM					O-T		1
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
400	65C26300-21		.	SHIM						A, B	AR
				(OPT ITEM 400F, 400L)							
-400F	BACS40R08F88F		.	SHIM						A, B	2
				(OPT ITEM 400, 400L)							
-400G	BACS40R08F88F		.	SHIM						C-N	2
-400H	BACS40R08F88F		.	SHIM						O-T	2
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
-400L	BACS40R08F155F		.	SHIM						A, B	2
				(OPT ITEM 400, 400F)							
-400M	BACS40R08F155F		.	SHIM						C-N	2
-400N	BACS40R08F155F		.	SHIM						O-T	2
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
405	65C26300-22		.	SHIM						A, B	AR
				(OPT ITEM 405F, 405L)							
-405F	BACS40R08F88F		.	SHIM						A, B	2
				(OPT ITEM 405, 405L)							
-405G	BACS40R08F88F		.	SHIM						C-N	2
-405H	BACS40R08F88F		.	SHIM						O-T	2
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
-405L	BACS40R08F155F		.	SHIM						A, B	2
				(OPT ITEM 405, 405F)							
-405M	BACS40R08F155F		.	SHIM						C-N	2
-405N	BACS40R08F155F		.	SHIM						O-T	2
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
-410	65C26300-23			DELETED							
410A	65C26300-42		.	SHIM						A, B	1
				(OPT ITEM 410B) (LIMITED USAGE)							
-410B	BACS40R08F175F		.	SHIM						A, B	1
				(OPT ITEM 410A) (LIMITED USAGE)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-410C	BACS40R08F175F		.	SHIM						C-N	1
-410D	BACS40R08F175F		.	SHIM						O-T	1
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
-415	BACS40R022C125F			DELETED							
415A	65C26300-43		.	SHIM						A, B	1
				(OPT ITEM 415B) (LIMITED USAGE)							
-415B	BACS40R08F120F		.	SHIM						A, B	1
				(OPT ITEM 415A) (LIMITED USAGE)							
-415C	BACS40R08F120F		.	SHIM						C-N	1
-415D	BACS40R08F120F		.	SHIM						O-T	1
				(INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5)							
420	BACS40R025E130F		.	SHIM							AR
425	BACS40R020C128F		.	SHIM						A-D	AR
-425A	BACS40R020C128F		.	SHIM						E, F	AR
				(LIMITED USAGE)							
430	BACR15BA3AD		.	RIVET							120
435	BACN10JS3A4CD		.	NUTPLATE							36
440	BACN10JS3A3CD		.	NUTPLATE							12
445	BACN10KE3DCD		.	NUTPLATE							4
450	BACN10YF33CD		.	NUTPLATE							5
455	BACN10JS3A2CD		.	NUTPLATE							2
460	BACN10YF34CD		.	NUTPLATE							1
465	NAS500FC10		.	SPACER-NUTPLATE							4
475	NAS6703-30		.	BOLT						A-J	1
476	NAS6704-46		.	BOLT						K-T	1
480	BACW10P183AL		.	WASHER						A-J	2
480A	BACW10P33AL		.	WASHER						K-T	2
481	65C33043-1		.	HINGE SUPT FTG ASSY						K, M, O, Q, S	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-482	65C33043-2		.							L, N, P, R, T	1
482C	KJB270306B		.	.							2
-485	BACB28X6D055										
485A	69-37234-28		.							A-N	2
-485B	69-37234-30		.							A-N	2
490	NAS74A3E113P		.							A-J	1
491	BACB28AK04-285		.							K-T	1
495	BACN10JC3CD		.							A-J	1
496	BACN10YR4CD		.							K-T	2
500	69-57706-2		.							A-J	3
501	69-77100-1		.							A-J	2
-501A	69-77100-1		.							K-T	2
505	NAS6703-14		.							A-L	2
-505A	NAS6704-13		.							A-L	2
-505B	NAS6704-14		.							K, L	2
-505C	NAS6704-13		.							K, L	2
-505D	NAS6704-13		.							M, N	2
510	BACW10P183AL		.								4
-510A	BACW10BP4DP		.								4
515	NAS74A3E011P		.							A-L	2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-515A	BACB28AK04-070		.	BUSHING						A-L	2
				(POST SB 737-57-1167R3)							
-515B	NAS74A3E011P		.	BUSHING						K, L	2
				(PRE SB 737-57-1167R1)							
				(LIMITED USAGE)							
-515C	BACB28AK04-070		.	BUSHING						K, L	2
				(POST SB 737-57-1167R1)							
				(LIMITED USAGE)							
-515D	BACB28AK04-070		.	BUSHING						M, N	2
520	BACN10JC3CD		.	NUT							2
				(PRE SB 737-57-1167R3)							
-520A	BACN10YR4CD		.	NUT							2
				(POST SB 737-57-1167R3)							
-525	65C26300-39		.	DOOR ASSY						A	1
				(LIMITED USAGE)							
-525A	65C26300-47		.	DOOR ASSY						A, C	1
				(LIMITED USAGE)							
-525B	65C26300-47			DELETED							
-525C	65C26300-79		.	DOOR ASSY						C	1
				(LIMITED USAGE)							
-525D	65C26300-85		.	DOOR ASSY						K	1
				(LIMITED USAGE)							
-525E	65C26300-79		.	DOOR ASSY						E, G, I	1
-525F	65C26300-89		.	DOOR ASSY						K	1
				(LIMITED USAGE)							
-525G	65C26300-89		.	DOOR ASSY						M	1
-525H	65C26300-101		.	DOOR ASSY						O, Q, S	1
-530	65C26300-40		.	DOOR ASSY						B	1
				(LIMITED USAGE)							
-530A	65C26300-48		.	DOOR ASSY						B, D	1
				(LIMITED USAGE)							
-530B	65C26300-48			DELETED							
-530C	65C26300-80		.	DOOR ASSY						D	1
				(LIMITED USAGE)							
-530D	65C26300-86		.	DOOR ASSY						L	1
				(LIMITED USAGE)							
-530E	65C26300-80		.	DOOR ASSY						F, H, J	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-530F	65C26300-90		.							L	1
-530G	65C26300-90		.							N	1
-530H	65C26300-102		.							P, R, T	1
535	NAS6703-65		. .								1
536	NAS6704-66		. .								1
540	BACW10P44AL		. .								2
545	BACW10P35AL		. .								4
546	BACW10P186C		. .								1
-550	BACB28X6D055										
550A	69-37234-28		. .								2
-550B	69-37234-30		. .								2
-555	BACB28X6D060										
555A	69-37234-29		. .								2
-555B	69-37234-31		. .								2
560	NAS74A3E315P		. .								1
561	BACB28AK04-420		. .								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
565	69-57704-3		. .								3
566	69-77184-1		. .								2
-566A	69-77184-1		. .								3
570	BACN10JC3CD		. .								1
571	NAS1805-4		. .								1
575	65-70930-27		. .								1
580	65-70930-28		. .								1
585	NAS6703-35		. .								1
586	NAS6704-35		. .								1
590	BACW10P183AL		. .								2
591	BACW10P33AL		. .								2
595	BACN10JC3CD		. .								1
-600	BACB28X6D055										DELETED

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
600A	69-37234-28								. . BUSHING (USED ON ITEMS 525, 525A, 525C, 525E, 530, 530A, 530C, 530E)		4
-600B	69-37234-30								. . BUSHING (OPT ITEM 600A) (USED ON ITEMS 525, 525A, 525C, 525E, 530, 530A, 530C, 530E)		4
605	NAS74A3E113P								. . BUSHING (USED ON ITEMS 525, 525A, 525C, 525E, 530, 530A, 530C, 530E)		1
606	BACB28AK04-200								. . BUSHING (USED ON ITEMS 525D, 525F, 525G, 525H, 530D, 530F, 530G, 530H)		1
610	BACN10JC3CD								. . NUT (USED ON ITEMS 525, 525A, 530, 530A)		1
611	BACN10YR4CD								. . NUT (USED ON ITEMS 525D, 525F, 525G, 525H, 530D, 530F, 530G, 530H)		1
615	65-70930-22								. . CHANNEL-SPRING SPT (USED ITEMS 525, 525A, 525C, 525E, 530, 530A, 530C, 530E) (PRE SB 737-57-1167R3)		1
-616	65C33034-1								. . LINK ASSY (USED ON ITEMS 525D, 530D)		1
-616A	65C33034-1								. . LINK ASSY (USED ON ITEMS 525F, 525G, 530F, 530G) (OPT ITEM 616B) (LIMITED USAGE)		1
-616B	65C33034-5								. . LINK ASSY (USED ON ITEMS 525F, 525G, 525H, 530F, 530G, 530H) (POST SB 737-57-1167R3)		1
616L	65C33034-3								. . . LINK (USED ON ITEMS 616, 616A)		1
-616N	65C33034-7								. . . LINK (USED ON ITEM 616B)		1
617	KJB270306B								. . . BUSHING		4
-618	65C33034-2								. . LINK ASSY (USED ON ITEMS 525D, 530D)		2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-618A	65C33034-2		. .								2
-618B	65C33034-6		. .						R3)		2
618L	65C33034-3		. . .								1
-618N	65C33034-7		. . .								1
619	KJB270306B		. . .								4
620	BACB30NN3K6		. .								4
-620A	BACB30LH3-6		. .								4
625	BACW10P183AL		. .								4
630	BACN10JC3CD		. .								4
635	69-57704-1		. .								1
-635A	69-57704-5		. .								1
-640	69-57704-2		. .								1
-640A	69-57704-6		. .								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
644	BACR15CE6AD8								. . RIVET (POST SB 737-57-1242)		1
645	BACB30NN3K8								. . BOLT (PRE SB 737-57-1242)		7
-645A	BACB30NN3K6								. . BOLT (POST SB 737-57-1242)		6
650	BACW10P183AL								. . WASHER (PRE SB 737-57-1242)		7
-650A	BACW10P183AL								. . WASHER (POST SB 737-57-1242)		6
655	BACN10JC3CD								. . NUT (USED ON ITEMS 525, 525A, 530, 530A) (PRE SB 737-57-1242)		7
-655A	BACN10JC3CD								. . NUT (POST SB 737-57-1242)		6
660	69-57705-3								. . STRIP-RUB (USED ON ITEM 525)		1
660A	69-57705-5								. . STRIP-RUB (USED ON ITEMS 525A, 525C THRU 525H) (PRE SB 737-57-1242)		1
-660B	69-57705-7								. . STRIP-RUB (USED ON ITEMS 525A, 525C THRU 525H) (POST SB 737-57-1242)		1
-665	69-57705-4								. . STRIP-RUB (USED ON ITEM 530)		1
-665A	69-57705-6								. . STRIP-RUB (USED ON ITEMS 530A, 530C THRU 530H) (PRE SB 737-57-1242)		1
-665B	69-57705-8								. . STRIP-RUB (USED ON ITEMS 530A, 530C THRU 530H) (POST SB 737-57-1242)		1
670	69-67237-1								. . SHIM-SEAL PLATE		AR
680	65-70947-27								. . DOOR ASSY (USED ON ITEMS 525, 525A) (PRE SB 737-57-1167R3)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-681	65-70947-287		. .								1
682	65-70947-293		. .								1
-683	65-70947-295		. .								1
-683A	65-70947-295										
-685	65-70947-28		. .								1
-686	65-70947-288		. .								1
-687	65-70947-294		. .								1
-688	65-70947-296		. .								1
-688A	65-70947-296										
690	BACB28W6D018		. . .								2
695	BACB28W6D040		. . .								1
-695A	KJB291906B		. . .								2
700	65-70947-149		. . .								1
-701	65-70947-289		. . .								1
702	65-70947-289		. . .								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-705	65-70947-150		. . .	DOOR							1
				(USED ON ITEM 685)							
-706	65-70947-290		. . .	DOOR							1
				(USED ON ITEM 686)							
-707	65-70947-290		. . .	DOOR							1
				(USED ON ITEMS 687, 688)							
-708	65C33609-1		. .	FITTING-HINGE							1
				(USED ON ITEM 525H)							
-708A	65C33609-2		. .	FITTING-HINGE							1
				(USED ON ITEM 530H)							
-708B	65C33609-5		. .	FITTING-HINGE							1
				(USED ON ITEM 525H)							
-708C	65C33609-6		. .	FITTING-HINGE							1
				(USED ON ITEM 530H)							
-709	BACR15BA6AD7		. RIVET								1
				(POST SB 737-57-1167R3)							
710	BACB30NN3K7		. BOLT						A, B		3
				(LIMITED USAGE)							
715	AN960-10L		. WASHER								3
720	NAS1804-3		. NUT								3
725	69-57703-1		. FITTING-HINGE						A-F		1
				(PRE SB 737-57-1167R3)							
-725A	69-57703-1		. FITTING-HINGE						G, H		1
				(PRE SB 737-57-1167R3)							
-725B	65C30443-1		. FITTING-HINGE						C-F		1
				(OPT ITEM 725)							
				(PRE SB 737-57-1167R3)							
-725C	69-57703-1		DELETED								
-725D	65C30443-1		. FITTING-HINGE						G, H		1
				(OPT ITEM 725A)							
				(PRE SB 737-57-1167R1)							
-725E	65C30443-1		. FITTING-HINGE						K, L		1
				(LIMITED USAGE)							
				(PRE SB 737-57-1167R3)							
-725F	65C33609-1		. FITTING-HINGE						K		1
				(LIMITED USAGE)							
-725G	65C33609-2		. FITTING-HINGE						L		1
				(LIMITED USAGE)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-725H	65C33123-11		.	FITTING-HINGE						A-L	1
				(LH)							
				(POST SB 737-57-1167R3)							
-725I	65C33123-10		.	FITTING-HINGE						A-L	1
				(RH)							
				(POST SB 737-57-1167R3)							
-725J	65C33609-1		.	FITTING-HINGE						M	1
-725K	65C33609-2		.	FITTING-HINGE						N	1
730	BACB30NN3K7		.	BOLT							2
				(PRE SB 737-57-1167R3)							
-730A	BACB30NN3K7		.	BOLT							3
				(POST SB 737-57-1167R3)							
735	BACB30NN3K9		.	BOLT							1
740	AN960-10L		.	WASHER							3
				(PRE SB 737-57-1167R3)							
-740A	AN960-10L		.	WASHER							4
				(POST SB 737-57-1167R3)							
745	NAS1804-3		.	NUT							3
				(PRE SB 737-57-1167R3)							
-745A	NAS1804-3		.	NUT							4
				(POST SB 737-57-1167R3)							
750	69-73712-1		.	FITTING-HINGE						A, C, E	1
				(PRE SB 737-57-1167R3)							
-750A	69-73712-1			DELETED							
-750B	65C30448-1			DELETED							
-750C	69-73712-1		.	FITTING-HINGE						G	1
				(LIMITED USAGE)							
				(PRE SB 737-57-1167R3)							
-750D	65C30448-1			DELETED							
-750E	65C30448-5		.	FITTING-HINGE						G	1
				(LIMITED USAGE)							
				(OPT ITEM 750C)							
				(PRE SB 737-57-1167R3)							
-750F	65C30448-5		.	FITTING-HINGE						I	1
				(PRE SB 737-57-1167R3)							
-750G	65C30448-5		.	FITTING-HINGE						K	1
				(LIMITED USAGE)							
				(PRE SB 737-57-1167R3)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-750H	65C33609-5		.	FITTING-HINGE						K	1
				(LIMITED USAGE)							
-750I	65C33609-6		.	FITTING-HINGE						L	1
				(LIMITED USAGE)							
-750J	65C33123-9		.	FITTING-HINGE						A, C, E, G, I, K, L	1
				(POST SB 737-57-1167R3)							
-750K	65C33609-5		.	FITTING-HINGE						M	1
-750L	65C33609-6		.	FITTING-HINGE						N	1
-755	69-73712-2		.	FITTING-HINGE						B, D, F	1
-755A	69-73712-2			DELETED							
-755B	65C30448-2			DELETED							
-755C	69-73712-2		.	FITTING-HINGE						H	1
				(LIMITED USAGE)							
-755D	65C30448-2			DELETED							
-755E	65C30448-6		.	FITTING-HINGE						H	1
				(LIMITED USAGE)							
				(OPT ITEM 755C)							
				(PRE SB 737-57-1167R3)							
-755F	65C30448-6		.	FITTING-HINGE						J	1
				(PRE SB 737-57-1167R3)							
-755G	65C33123-12		.	FITTING-HINGE						H, J, L	1
				(POST SB 737-57-1167R3)							
-755H	65C30448-6		.	FITTING-HINGE						L	1
				(LIMITED USAGE)							
				(PRE SB 737-57-1167R1)							
760	BACB30NW6K6		.	HI-LOK							2
765	BACB30NW6K5		.	HI-LOK							14
770	BACB30NW6K4		.	HI-LOK							1
775	BACB30NW5K7		.	HI-LOK							4
-780	BACR15GA5			DELETED							
780A	BACB30NW5K		.	HI-LOK							4
-785	BACR15CE6M			DELETED							
785A	BACB30NW5K		.	HI-LOK							2
790	BACF3T03M6G06		.	FILLER							4
795	BACC30M		.	COLLAR							17

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
800	BACC30AB5C		.								4
805	69-74570-1		.							A, C, E	1
-805A	69-74570-15		.							A, C, E	1
-810	69-74570-2		.							B, D, F	1
-810A	69-74570-16		.							B, D, F	1
820	BACR15GA5-4		.								3
-820A	BACR15GA6-5		.								3
825	BACB30NW5K7		.								1
-825A	BACB30NW6K7		.								1
830	BACC30AB5C		.								1
-830A	BACC30AB6C		.								1
835	69-74570-3		.							A, C, E, G, I	1
-835A	69-74570-17		.							A, C, E, G, I, K	1
-835B	69-74570-3		.							K	1
-835C	69-74570-17		.							M, O, Q, S	1
-840	69-74570-4		.							B, D, F, H, J	1
-840A	69-74570-18		.							B, D, F, H, J, L	1
-840B	69-74570-4		.							L	1
-840C	69-74570-18		.							N, P, R, T	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
841	69-74570-5		.	STRIP-RUB, UPR						G	1
				(LIMITED USAGE)							
-842	69-74570-6		.	STRIP-RUB, UPR						H	1
				(LIMITED USAGE)							
843	69-74570-13		.	STRIP-RUB, UPR						G	1
				(LIMITED USAGE)							
-843A	69-74570-13		.	STRIP-RUB, UPR						I, K, M, O, Q, S	1
-844	69-74570-14		.	STRIP-RUB, UPR						H	1
				(LIMITED USAGE)							
-844A	69-74570-14		.	STRIP-RUB, UPR						J, L, N, P, R, T	1
-845	65C26306-13			DELETED							
-845A	65C26306-5			DELETED							
845B	65C26307-13		.	TAILCONE FITTING						A	1
				(LIMITED USAGE)							
-845C	65C26307-5		.	TAILCONE FITTING						A	1
				(OPT ITEM 845B,845D)							
				(LIMITED USAGE)							
-845D	65C26307-15		.	TAILCONE FITTING						A	1
				(LIMITED USAGE)							
-845E	65C26307-15		.	TAILCONE FITTING						C, E, G, I, K, M, O, Q	1
-845F	65C26307-17		.	TAILCONE FITTING						S	1
				(LIMITED USAGE)							
-845G	65C26307-19		.	TAILCONE FITTING						S	1
				(LIMITED USAGE)							
-850	65C26306-14			DELETED							
-850A	65C26306-6			DELETED							
-850B	65C26307-14		.	TAILCONE FITTING						B	1
				(LIMITED USAGE)							
				(OPT ITEM 850C)							
-850C	65C26307-6		.	TAILCONE FITTING						B	1
				(OPT ITEM 850B, 850D)							
				(LIMITED USAGE)							
-850D	65C26307-16		.	TAILCONE FITTING						B	1
				(LIMITED USAGE)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY						
			1	2	3	4	5	6	7								
1-																	
-850E	65C26307-16		.	T	A	I	L	C	O	N	E	F	I	T	T	D, F, H, J, L, N, P, R	1
-850F	65C26307-18		.	T	A	I	L	C	O	N	E	F	I	T	T	T	1
-850G	65C26307-20		.	T	A	I	L	C	O	N	E	F	I	T	T	T	1
855	BACR15BA3AD		.	R	I	V	E	T								82	
860	BACN10YF32CD		.	N	U	T	P	L	A	T	E					26	
865	BACN10KE3E2CD		.	N	U	T	P	L	A	T	E					2	
870	BACN10KE3DCD		.	N	U	T	P	L	A	T	E					2	
875	BACN10JS3A4CD		.	N	U	T	P	L	A	T	E					10	
880	BACN10KE3E3CD		.	N	U	T	P	L	A	T	E					1	
885	65C26300-30		.	S	P	A	C	E	R							4	
900	BACB30LR3-6		.	B	O	L	T									28	
-902	BACB30NN5K8		.	B	O	L	T									3	
-903	BACB30NN5K11		.	B	O	L	T									1	
905	BACB30LR3-6		.	B	O	L	T									4	
-907	BACB30NW6K5		.	H	I	-	L	O	K							6	
-910	BACB30LR3-6		.	B	O	L	T									6	
-910A	BACB30NN3K7		.	B	O	L	T									6	
-915	BACB30LR3-6		.	B	O	L	T									2	
-915A	BACB30NN3K5		.	B	O	L	T									2	
-920	BACB30LR3-6		.	B	O	L	T									1	
-920A	BACB30NN3K7		.	B	O	L	T									1	
925	NAS500FC10		.	S	P	A	C	E	R							1	
927	AN960-516L		.	W	A	S	H	E	R							4	
928	NAS1804-5		.	N	U	T										4	
929	BACC30M6		.	C	O	L	L	A	R							6	
930	BACB30VF3K3		.	B	O	L	T									2	

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY		
			1	2	3	4	5	6	7				
1-													
935	AN960XC9		.	W	A	S	H	E	R		2		
940	BACN10JC3CD		.	N	U	T					2		
945	69-57701-2		.	S	U	P	P	O	R	T	1		
950	BACB30NW6K2		.	H	I	L	O	K			1		
955	BACB30NW6K6		.	H	I	L	O	K			1		
960	BACB30NW6K3		.	H	I	L	O	K			1		
965	BACC30M6		.	C	O	L	L	A	R		3		
970	65-70938-3		.	F	A	I	R	I	N	G-WING PANEL, AFT, WING TO NACELLE STRUCT	1		
-975	65C26301-71		.	T	A	I	L	C	O	N	E-UPR (FOR DETAILS SEE FIG. 2)	O, P	1
-975A	65C26301-71		.	T	A	I	L	C	O	N	E-UPR (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	Q, R	1
-975B	65C26301-91		.	T	A	I	L	C	O	N	E-UPR (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	Q, R	1
-975C	65C26301-91		.	T	A	I	L	C	O	N	E-UPR (FOR DETAILS SEE FIG. 2)	S, T	1
-980	65C26301-65		.	T	A	I	L	C	O	N	E-LWR (LH) (FOR DETAILS SEE FIG. 2)	O	1
-980A	65C26301-65		.	T	A	I	L	C	O	N	E-LWR (LH) (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	Q	1
-980B	65C26301-87		.	T	A	I	L	C	O	N	E-LWR (LH) (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	Q	1
-980C	65C26301-87		.	T	A	I	L	C	O	N	E-LWR (LH) (FOR DETAILS SEE FIG. 2)	S	1
-985	65C26301-66		.	T	A	I	L	C	O	N	E-LWR (RH) (FOR DETAILS SEE FIG. 2)	P	1

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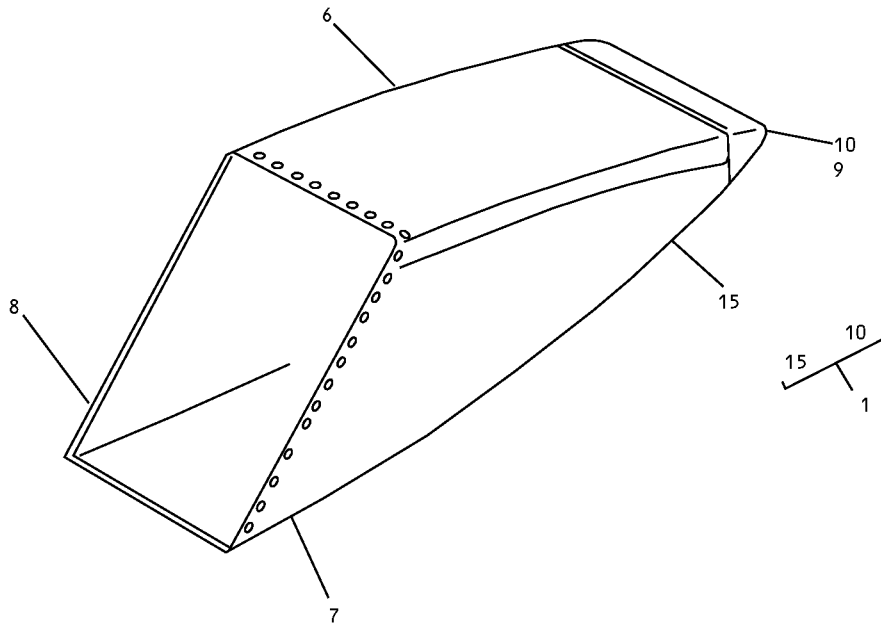
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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1- -985A	65C26301-66		. TAILCONE-LWR (RH) (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	R	1
-985B	65C26301-88		. TAILCONE-LWR (RH) (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	R	1
-985C	65C26301-88		. TAILCONE-LWR (RH) (FOR DETAILS SEE FIG. 2)	T	1
-990	65C31457-2		. CAP-END (FOR DETAILS SEE FIG. 2) (LIMITED USAGE)	O, P	1
-990A	65C31457-2		. CAP-END (FOR DETAILS SEE FIG. 2)	Q-T	1

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Inboard Flap Outboard Track Tailcone Assembly
IPL Figure 2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
2-												
1	65C26300-5											
-1A	65C26300-57											
-1B	65C26300-69											
-1C	65C26300-77											
-1D	65C26300-111											
-5	65C26300-6											
-5A	65C26300-58											
-5B	65C26300-70											
-5C	65-26300-78											
-5D	65-26300-112											
6	65C26301-71											
-6A	65C26301-91											

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2- 7	65C26301-65		TAILCONE-LWR (LH) (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5 FIG. 1)								RF
-7A	65C26301-87		TAILCONE-LWR (LH) (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5 FIG. 1)								RF
8	65C26301-66		TAILCONE-LWR (RH) (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5 FIG. 1)								RF
-8A	65C26301-88		TAILCONE-LWR (RH) (INSTALLATION PART USED WITH INBD FLAP OUTBOARD TRACK FAIRING ASSEMBLY ITEMS 1, 5 FIG. 1)								RF
9	65C31457-2		CAP-END (LIMITED USAGE)								1
10	69-73101-1		. CAP-END (OPT ITEM 10A) (PRE SB 737-57-1150)							A, B	1
-10A	69-73101-2		. CAP-END (OPT ITEM 10) (PRE SB 737-57-1150)							A, B	1
-10B	69-73101-2		. CAP-END (PRE SB 737-57-1150)							C, D	1
-10C	65C31457-1		. CAP-END (LIMITED USAGE)							E, F	1
-10D	65C31457-2		. CAP-END (LIMITED USAGE)							E, F	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
2- -10E	65C31457-2		.								G, H	1
10F	65C3518-1		.								G, H	1
10G	65C31457-1		.								A-D	1
10H	65C31457-2		.								A-D	1
-10I	65C35318-1		.								I, J	1
-10J	65C31457-2		.								I, J	1
-10K	65C31457-3		.								I, J	1
15	65C26301-51		.								A	1
-15A	65C26301-63		.								C, E	1
-15B	65C26301-69		.								G	1
-20	65C26301-52		.								B	1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
2- -20A	65C26301-64									D, F	1
-20B	65C26301-70									H	1

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