



COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

INBOARD TRAILING EDGE FORE FLAP ASSEMBLY

PART NUMBER

**65-46430-100SP, -111SP, -112SP, -121SP, -122SP,
-127SP, -128SP, -129SP, -130SP, -137SP, -138SP,
-141SP, -142SP, -161SP, -162SP, -163SP, -164SP,**

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PART NUMBER (Cont.)

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-189SP, -190SP, -43, -44, -75, -76, -79, -80, -83, -84, -87SP, -88SP, -97SP, -98SP, -99SP**

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COMPONENT MAINTENANCE MANUAL

Revision No. 29
Jul 01/2009

To: All holders of INBOARD TRAILING EDGE FORE FLAP ASSEMBLY 57-53-26.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

ATTENTION

IF YOU RECEIVE PRINTED REVISIONS, PLEASE VERIFY THAT YOU HAVE RECEIVED AND FILED THE PREVIOUS REVISION. BOEING MUST BE NOTIFIED WITHIN 30 DAYS IF YOU HAVE NOT RECEIVED THE PREVIOUS REVISION. REQUESTS FOR REVISIONS OTHER THAN THE PREVIOUS REVISION WILL REQUIRE A COMPLETE MANUAL REPRINT SUBJECT TO REPRINT CHARGES SHOWN IN THE DATA AND SERVICES CATALOG.

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TRANSMITTAL LETTER
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Location of Change

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REPAIR-GENERAL

Description of Change

Changed the material list per the latest source information.

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HIGHLIGHTS

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A = Added, R = Revised, D = Deleted, O = Overflow

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TEMPORARY REVISION AND SERVICE BULLETIN RECORD

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		PRR 31030	DEC 05/84
		PRR 32121-21	DEC 05/84
		PRR 33084	DEC 05/84
		PRR 33135-7	DEC 05/84
		PRR 33120-84	DEC 05/84
		PRR 33180-30	DEC 05/84
		PRR 34131	DEC 05/88
	57-21		SEP 05/91
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		PRR 35249	JUN 01/97

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TR AND SB RECORD

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All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed		Revision		Filed	
Number	Date	Date	Initials	Number	Date	Date	Initials



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Revision		Filed		Revision		Filed	
Number	Date	Date	Initials	Number	Date	Date	Initials

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REVISION RECORD

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All temporary revisions to this manual will be accompanied by a cover sheet bearing the temporary revision number. Enter the temporary revision number in numerical order, together with the temporary revision date, the date the temporary revision is inserted and the initials of the person filing. When the temporary revision is incorporated or cancelled, and the pages are removed, enter the date the pages are removed and the initials of the person who removed the temporary revision.

Temporary Revision		Inserted		Removed		Temporary Revision		Inserted		Removed	
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials



COMPONENT MAINTENANCE MANUAL

INTRODUCTION

1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
 - (1) Title Page
 - (2) Transmittal Letter
 - (3) Highlights
 - (4) List of Effective Pages
 - (5) Table of Contents
 - (6) Temporary Revision & Service Bulletin Record
 - (7) Record of Revisions
 - (8) Record of Temporary Revisions
 - (9) Introduction
 - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.

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INTRODUCTION

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COMPONENT MAINTENANCE MANUAL

INBOARD FLAP FOREFLAP ASSEMBLY - DESCRIPTION AND OPERATION

1. Description and Operation

- A. The foreflap consists of sheet metal covered nose ribs, a spar, and a curved, machine tapered honeycomb trailing edge. Both upper and lower surfaces of the flap are covered with clad aluminum skin panels. Three supports extend through the lower surface to connect the foreflap to tracks carried in the midflap. A toggle assembly is installed on each end of the foreflap. These toggles move along cams mounted on the midflap carriages.

2. Leading Particulars (approximate)

- A. Length – 132 inches
- B. Width – 16 inches
- C. Thickness – 4 inches
- D. Weight – 35 pounds

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DESCRIPTION AND OPERATION

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TESTING AND FAULT ISOLATION

(NOT APPLICABLE)

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TESTING AND FAULT ISOLATION

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DISASSEMBLY

1. General

- A. This procedure has the data necessary to disassemble the inboard flap foreflap assembly.
- B. Disassemble this component only as necessary to complete fault isolation, determine the serviceability of parts, perform required repairs, and restore the unit to serviceable condition.

2. Disassembly

- A. Use standard industry practices for disassembly of this component.

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DISASSEMBLY

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CLEANING

1. General

- A. This procedure has the data necessary the inboard flap foreflap assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for the item numbers.

2. Cleaning

A. References

Reference	Title
SOPM 20-30-01	CLEANING AND RELUBRICATING BEARINGS
SOPM 20-30-03	GENERAL CLEANING PROCEDURES

B. Procedure

- (1) Clean all parts, except bearings, using standard industry practices and information contained in SOPM 20-30-03.
- (2) Clean bearings (20, 65, 95, IPL Figure 1) in accordance with SOPM 20-30-01.

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CLEANING

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COMPONENT MAINTENANCE MANUAL

CHECK

1. General

- A. This procedure has the data necessary to find defects in the material of the specified parts.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Check

A. References

Reference	Title
SOPM 20-20-01	MAGNETIC PARTICLE INSPECTION
SOPM 20-20-02	PENETRANT METHODS OF INSPECTION
737 NDT Part 1, 51-05-01	Tap Test Inspection of Honeycomb Sandwich Structure
737 NDT Part 4, 51-00-02	Full Depth Honeycomb and Laminate Structure Inspection
737 NDT Part 9, 51-00-01	Non-Destructive Testing

B. Procedure

- (1) Check all parts for obvious defects using standard industry practices. If you find visible defects or suspect possible defects, do a magnetic particle or penetrant check as specified below.
 - (a) Magnetic particle examine the following parts per SOPM 20-20-01 if visual examination discloses defects.
 - 1) Supports (180, 185, 186, 187)
 - 2) Torque tubes (220, 240)
 - 3) Brackets (280, 281, 282, 335)
 - 4) Toggle (60)
 - (b) Penetrant check actuation bracket (525) per SOPM 20-20-02 if visual examination discloses defects.
- (2) Do a check of the honeycomb structure and bonded parts for evidence of delamination, internal water, scratches, and contour defects.
 - (a) If you see delamination or impact damage when you do a visual check, do an ultrasonic inspection or a tap test to find all damage. Refer to the 737 NDT Part 4, 51-00-02 or 737 NDT Part 1, 51-05-01.

NOTE: For the tap test, use a solid metal disk and tap the surface area lightly but firmly. The void areas will produce a dull sound as opposed to a sharp sound on a solid bonded area. Refer to the 737 NDT Part 1, 51-05-01.
 - (b) Do a check on areas you suspect of containing water with the radiographic or thermographic method. Refer to the 737 NDT Part 9, 51-00-01.
 - (c) Do a check on the edges of the panel carefully for cuts and abrasions. Delamination starts very easily from damage to an edgemember of honeycomb panel.
- (3) Refer to the applicable 737 Structural Repair Manual Chapter 57 for allowable damage data.

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CHECK

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REPAIR

1. Contents

A. Repair, refinish and replacement procedures are included in separate sections as follows:

Table 601:

PART NUMBER	NAME	REPAIR
69-35336	TOGGLE ASSY	1-1, 1-2
69-73105	ACTUATOR BRACKET ASSY	2-1
- - -	MISCELLANEOUS PARTS REFINISH	3-1
69-37815	TORQUE TUBE ASSY	4-1
69-38802	SUPPORT ASSY	5-1

2. Standard Practices

A. Refer to the following standard practices as applicable, for details of procedures in individual repairs.

- 20-00-00 Introduction
- SOPM 20-41-01 Decoding Table for Boeing Finish Codes
- SOPM 20-41-02 Application of Chemical and Solvent Resistant Finishes
- SOPM 20-42-02 Low Hydrogen Embrittlement Cadmium-Titanium Alloy Plating
- SOPM 20-42-05 Bright Cadmium Plating
- SOPM 20-50-02 Installation of Safetying Devices
- SOPM 20-50-03 Bearing Removal, Installation and Retention
- SOPM 20-50-07 Lubrication
- SOPM 20-50-19 General Sealing
- SOPM 20-60-02 Finishing Materials
- SOPM 20-60-03 Lubricants
- SOPM 20-60-04 Miscellaneous Materials

3. Materials

NOTE: Equivalent substitutes can be used.

A. Enamel

- (1) BMS 10-11, Type 2, Color BAC707 Gray coating, C50075
- (2) BMS 10-60, Type 1, Color BAC707 Gray coating, C00700

B. Grease

- (1) MIL-G-23827 grease, D00013
- (2) BMS 3-24 grease, D00015

C. Lockwire – MS20995NC32 lockwire, G50347

D. Primer – BMS 10-11, Type 1 primer, C00259

E. Sealant – BMS 5-95 sealant, A00247

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REPAIR - GENERAL

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F. Adhesive – BMS 5 -126 (BAC 5010, Type 38) adhesive, A01070

4. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in the applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.

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REPAIR - GENERAL

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—	STRAIGHTNESS	∅	DIAMETER
□	FLATNESS	S ∅	SPHERICAL DIAMETER
⊥	PERPENDICULARITY (OR SQUARENESS)	R	RADIUS
//	PARALLELISM	SR	SPHERICAL RADIUS
○	ROUNDNESS	()	REFERENCE
⊙	CYLINDRICITY	BASIC	A THEORETICALLY EXACT DIMENSION USED
⌒	PROFILE OF A LINE	(BSC)	TO DESCRIBE SIZE, SHAPE OR LOCATION OF
⌓	PROFILE OF A SURFACE	OR	A FEATURE. FROM THIS FEATURE PERMISSIBLE
◎	CONCENTRICITY	DIM	VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR
≡	SYMMETRY		NOTES.
∠	ANGULARITY	-A-	DATUM
↗	RUNOUT	(M)	MAXIMUM MATERIAL CONDITION (MMC)
↗↗	TOTAL RUNOUT	(L)	LEAST MATERIAL CONDITION (LMC)
□	COUNTERBORE OR SPOTFACE	(S)	REGARDLESS OF FEATURE SIZE (RFS)
∇	COUNTERSINK	(P)	PROJECTED TOLERANCE ZONE
⊕	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)	FIM	FULL INDICATOR MOVEMENT

EXAMPLES

— 0.002	STRAIGHT WITHIN 0.002	◎ ∅ 0.0005 C	CONCENTRIC TO DATUM C WITHIN 0.0005 DIAMETER
⊥ 0.002 B	PERPENDICULAR TO DATUM B WITHIN 0.002	≡ 0.010 A	SYMMETRICAL WITH DATUM A WITHIN 0.010
// 0.002 A	PARALLEL TO DATUM A WITHIN 0.002	∠ 0.005 A	ANGULAR TOLERANCE 0.005 WITH DATUM A
○ 0.002	ROUND WITHIN 0.002	⊕ ∅ 0.002 (S) B	LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
⊙ 0.010	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	⊥ ∅ 0.010 (M) A	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010 INCH DIAMETER, PERPENDICULAR TO DATUM A, AND EXTENDING 0.510 INCH ABOVE DATUM A, MAXIMUM MATERIAL CONDITION
⌒ 0.006 A	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM A	0.510 (P)	
⌓ 0.020 A	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.020 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE	2.000	THEORETICALLY EXACT DIMENSION IS 2.000
		OR	
		2.000	
		BSC	

True Position Dimensioning Symbols
Figure 601

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REPAIR - GENERAL

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TOGGLE ASSEMBLY - REPAIR 1-1

69-35336-1, -2, -3

1. General

- A. This repair gives the data to repair and refinish the inboard flap foreflap toggle assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- E. Refer to IPL Figure 1 for item numbers.

2. Bushing (90) Replacement

- A. Remove the old bushings.
- B. If you find defects on the toggle surfaces, refer to REPAIR 1-2 for repair instructions.
- C. Install replacement bushings (90, 90A) by the shrink-fit method (SOPM 20-50-03) with sealant, A00247.
- D. Install replacement bushings (90B) by the shrink-fit or press-fit method (SOPM 20-50-03).

3. Bearing (95) Replacement

- A. Toggle assembly 69-35336-1, -2
 - (1) Remove lockwire, G50347 from spring pin (100).
 - (2) Remove spring pin (100), bolt assembly (105A) and the bearing (95).
 - (3) Install the bearing (95) and bolt assembly (105A).
 - (4) Install spring pin (100).
 - (5) Lockwire spring pin (100) per SOPM 20-50-02 with lockwire, G50347.
- B. Toggle assembly 69-35336-3
 - (1) Remove staked material from the end of bolt (105).
 - (2) Push the bolt out from the toggle. The bolt is a shrink-fit in the toggle.
 - (3) Remove the bearing. Discard the old bolt.
 - (4) Install a replacement bearing and a replacement bolt in the toggle. Use the shrink-fit method of SOPM 20-50-03 to install the bolt.
 - (5) Stake the hollow end of the pin over the toggle to a minimum diameter of 0.27 inch.
 - (6) Make sure the bearing turns freely by light finger pressure.

4. Bearing (65) Replacement

- A. Remove the old bearing, bushing (85 or 91), washers (70, 71), nut (as applicable), and cotter pin (80).

CAUTION: INCORRECT INSTALLATION OF THE COTTER PIN AND THE CASTELLATED NUT CAN RESULT IN EXCESSIVE WEAR TO THE PARTS.

- B. Apply a thin layer of grease, D00015 to the shank and the threads of bearing (65).

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REPAIR 1-1

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- C. Install a replacement bearing with bushing (85 or 91), washer (70), and nut (75). Tighten nut (75) to 50-100 pound-inches.
 - (1) Use the washer (71) as required to align the slot of the castellated nut (75) with the cotter pin hole while torqued to 50-100 pound-inches.
- D. Install cotter pin (80) per SOPM 20-50-02.

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REPAIR 1-1
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TOGGLE - REPAIR 1-2

65C33100-1, -3, -4, 69-35329-1

1. General

- A. This repair gives the data to repair and refinish the inboard flap foreflap toggle.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- E. Refer to IPL Figure 1 for item numbers.

2. Holes for Bolts (REPAIR 1-2, Figure 601)

- A. Machine as required, within repair limits, to remove defects.
- B. Refinish as indicated.
- C. If applicable, make repair bushings (REPAIR 1-2, Figure 602) to adjust for the material removed in REPAIR 1-2, Paragraph 2.A..
- D. Install the bushings by the shrink-fit method of SOPM 20-50-03.
- E. Machine the bores of the bushings to design dimensions and finish. Be sure to restore the countersink, if necessary.

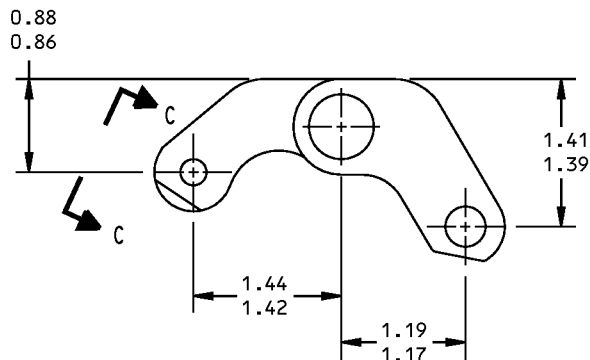
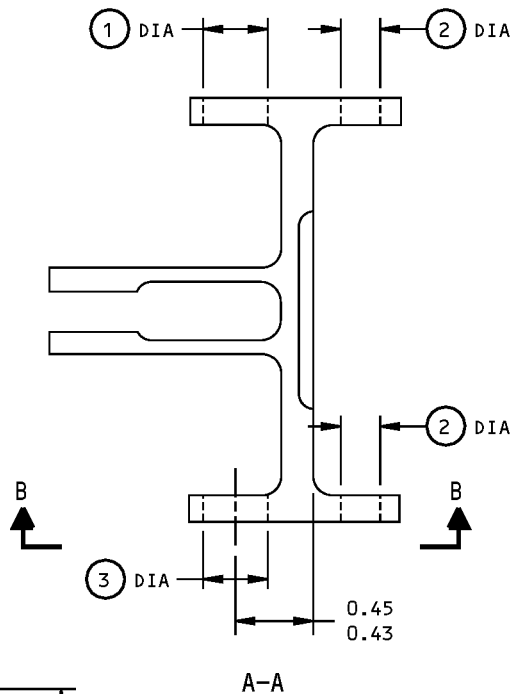
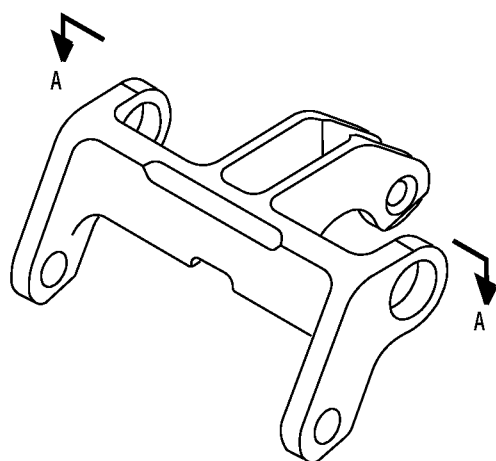
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REPAIR 1-2

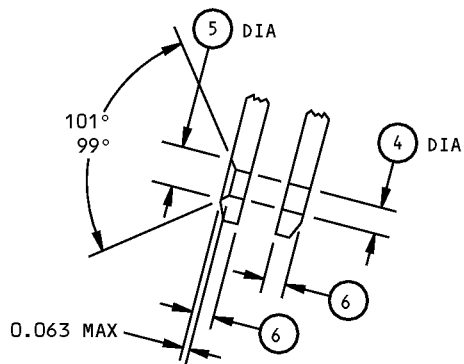
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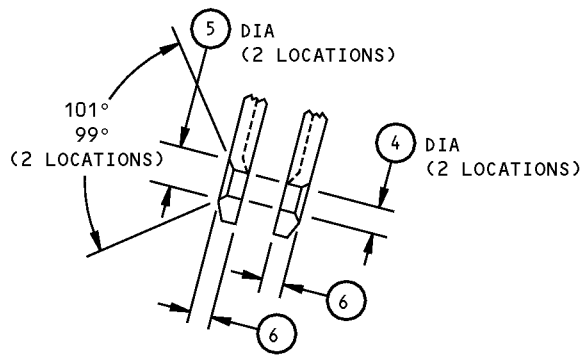
COMPONENT MAINTENANCE MANUAL



B-B



65C33100-1
69-35329-1
C-C



65C33100-3,-4
C-C

65C33100-1,-3,-4 69-35329-1 Toggle Repair and Refinish
Figure 601 (Sheet 1 of 2)

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REPAIR 1-2
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		①	②	③	④	⑤	⑥
69-35329-1	DESIGN DIM	0.6255 0.6248	0.378 0.375	0.6255 0.6248	0.2553 0.2513	0.420 0.390	0.21 0.19
	REPAIR LIMIT	---	---	---	0.2600 0.3160	 	---
65C33100-1	DESIGN DIM	0.6255 0.6248	0.379 0.375	0.6255 0.6248	0.2553 0.2513	0.408 0.398	0.22 0.20
	REPAIR LIMIT	---	---	---	0.2600 0.3160	 	---
65C33100-3,-4	DESIGN DIM	0.6255 0.6248	0.379 0.375	0.6255 0.6248	0.2500 0.2495	0.370 0.360	0.22 0.20
	REPAIR LIMIT	---	---	---	---	---	---

REFINISH

69-35329-1 - CADMIUM PLATE AND APPLY PRIMER
BMS 10-11, TYPE 1 (F-16.01) BUT
NOT IN HOLES. APPLY ENAMEL
BMS 10-11, TYPE 2 (SRF-12.63)
BUT NOT IN HOLES.

65C33100-1 - CADMIUM PLATE AND APPLY PRIMER
BMS 10-11, TYPE 1 (F-16.01) BUT
NOT IN HOLES. APPLY ENAMEL
BMS 10-11, TYPE 2 (F-21.02) BUT
NOT IN HOLES.

65C33100-3,-4 - PASSIVATE (F-17.09)

RESTORATION TO DESIGN DIMENSION NOT
REQUIRED

LIMIT FOR INSTALLATION OF REPAIR BUSHINGS
(FIG. 602)

REPAIR

REF

125/ ALL MACHINED SURFACES UNLESS SHOWN
DIFFERENTLY

MATERIAL: 69-35329-1, 65C33100-1:
4340 STEEL, 160-180 KSI

65C33100-3,-4:
15-5PH CRES, 180-200 KSI

ALL DIMENSIONS ARE IN INCHES

65C33100-1,-3,-4 69-35329-1 Toggle Repair and Refinish
Figure 601 (Sheet 2 of 2)

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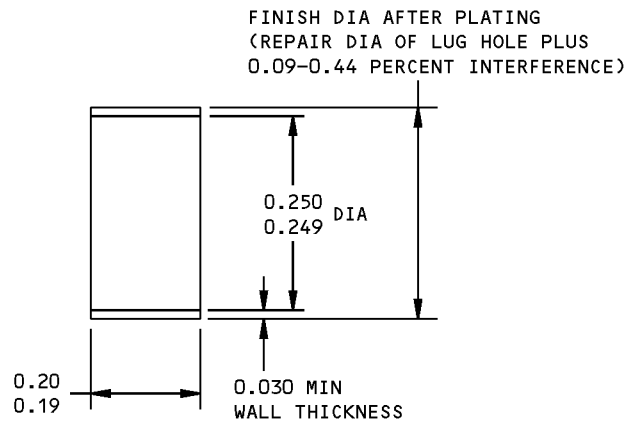
REPAIR 1-2

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125/ ALL MACHINED SURFACES

BREAK SHARP EDGES 0.01-0.02R

FINISH: CADMIUM PLATE (F-15.06)

MATERIAL: 15-5PH OR 17-4PH CRES
(180-200 KSI)

ALL DIMENSIONS ARE IN INCHES

HOLE LOCATION (4) FIG. 601

Repair Bushing Details
Figure 602

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REPAIR 1-2

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COMPONENT MAINTENANCE MANUAL

ACTUATION BRACKET ASSEMBLY - REPAIR 2-1

69-73105-1, -2, -7, -8

1. General

- A. This repair gives the data to repair and refinish the inboard flap foreflap actuation bracket assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- E. Refer to IPL Figure 1 for item numbers.

2. Bushing (550, 555) Replacement

- A. Remove the old bushings.
- B. Install replacement bushings by the shrink-fit method of SOPM 20-50-03.
- C. Fillet seal the bushing flanges with sealant, A00247.
- D. Apply a top coat of primer, C00259 (F-20.02) to fillet seal.

3. Refinish

- A. Bracket (535, 540) Chromic acid anodize and apply primer, C00259 (F-18.13). Material: Al alloy.

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REPAIR 2-1

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MISCELLANEOUS PARTS REFINISH - REPAIR 3-1

1. General

- A. This procedure has the data necessary to refinish the parts, which are not given in the specific repairs.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to IPL Figure 1 for the item numbers.

2. Refinish

- A. Repair of these parts is only replacement of the original finish. Refer to REPAIR 3-1, Table 601 for refinish details.

Table 601: Refinish Details

IPL FIG. & ITEM	MATERIAL	FINISH
Fig. 1		
Bracket (280, 281, 282, 335)	4330M Steel, 220-240 ksi	Cadmium-titanium plate (F-1.308, which replaces F-1.181). Apply primer, C00259 (F-12.205). Apply enamel coating, C50075 (SRF-12.63).
Retainer (520)	Al alloy	Chemical treat or chromic acid anodize, and apply primer, C00259 (SRF-2.30). Apply enamel coating, C50075 (SRF-12.63).
Foreflap Assy (1, 5)	- - -	After assembly, apply mask to toggle assy (55), torque tube assemblies (215, 235), nameplate (130), rubstrip (500) and seals (445, 450, 505, 510). Apply enamel coating, C50075 (SRF-14.9813).
Rib (608, 609)	Al alloy	Chromic acid anodize and apply primer, C00259 (F-18.13).
Rib (610)	Al alloy	On interior and exterior surfaces, apply primer, C00259 (F-18.06).

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REPAIR 3-1

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COMPONENT MAINTENANCE MANUAL

TORQUE TUBE ASSEMBLY - REPAIR 4-1

69-37815-1, -7, -8

1. General

- A. This repair gives the data to repair and refinish the inboard flap foreflap torque tube assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- E. Refer to IPL Figure 1 for item numbers.

2. Nutplate Replacement (REPAIR 4-1, Figure 601)

- A. Remove the rivets and the old nutplate and the filler.
- B. Install a replacement nutplate and filler with faying surface sealant and new rivets per SOPM 20-50-19.

3. Repair of the 1.970-1.972 Torque Tube Diameter (REPAIR 4-1, Figure 601)

- A. Machine as necessary within repair limits to remove defects.
- B. Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
- C. Shot peen as specified in REPAIR 4-1, Figure 601.
- D. Build up the machined surfaces with nickel plate. Machine the nickel plate to design dimensions and finish as specified in REPAIR 4-1, Figure 601.

4. Repair of Attach Holes

- A. Method 1 – Nickel plate buildup
 - (1) Machine as necessary, as shown in REPAIR 4-1, Figure 601, within repair limits to remove defects.
 - (2) Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
 - (3) Shot peen as specified in REPAIR 4-1, Figure 601.
 - (4) Build up the surfaces with nickel plate. Machine the nickel plate to design dimensions and finish as specified in REPAIR 4-1, Figure 601.
- B. Method 2 – Installation of repair bushing
 - (1) Machine as necessary, as shown in REPAIR 4-1, Figure 601, within repair limits to remove defects.
 - (2) Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
 - (3) Refinish as specified in REPAIR 4-1, Figure 601.
 - (4) Make a repair bushing (REPAIR 4-1, Figure 602) as required to adjust for the material removed.
 - (5) Install the repair bushing by the shrink-fit method (SOPM 20-50-03) with sealant, A00247 on mating surfaces.

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REPAIR 4-1

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- (6) Machine the bushing bore to design dimensions and finish as specified in REPAIR 4-1, Figure 601.
- C. Method 3 – Installation of oversize bolts
- (1) Machine the holes to 0.3125-0.3165 diameter, as shown in REPAIR 4-1, Figure 601.
 - (2) Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
 - (3) Refinish as specified in REPAIR 4-1, Figure 601.
 - (4) Machine the mating holes in the support assembly oversize (REPAIR 5-1).
 - (5) Make a note to be sure to use the oversize bolt and related washer and nut in the oversized holes.

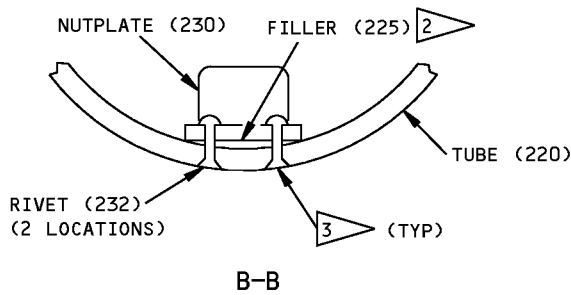
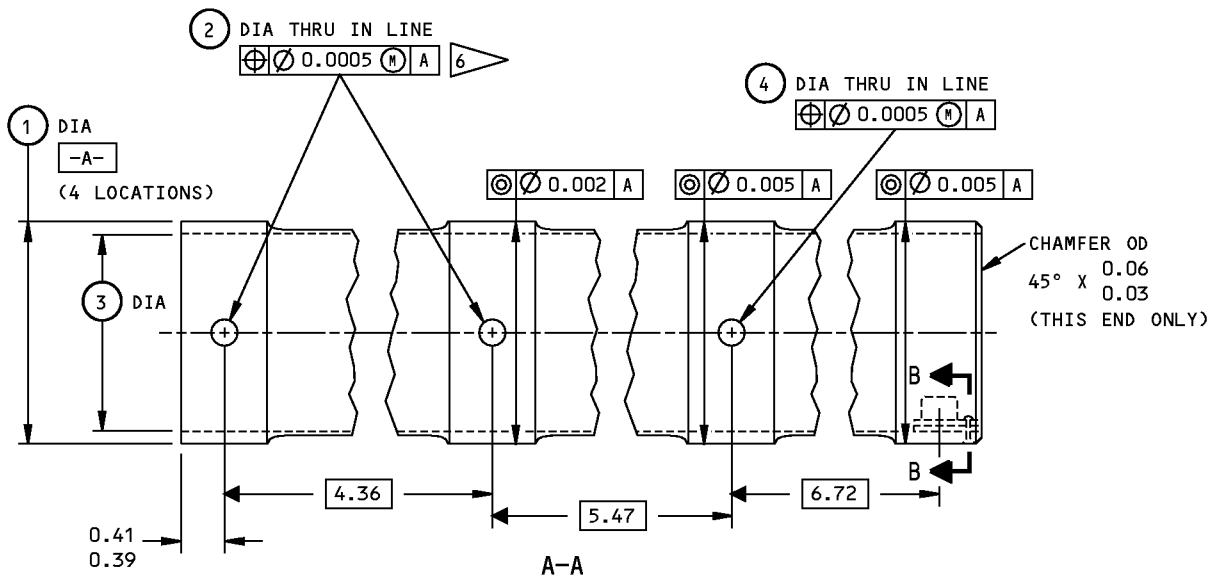
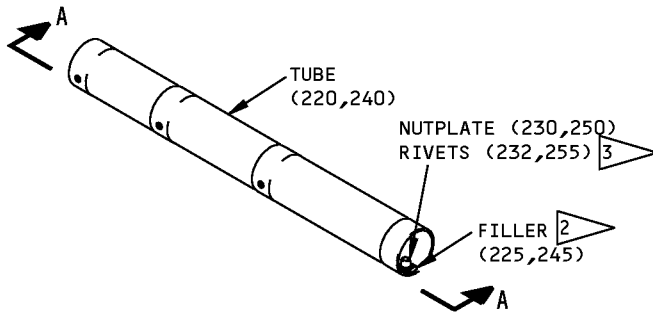
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REPAIR 4-1

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69-37815-1, -7, -8 Torque Tube Repair and Refinish
Figure 601 (Sheet 1 of 2)

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REPAIR 4-1
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	①	②	③	④
DESIGN DIM	1.972 1.970	0.254 0.250	1.820 1.800	0.254 0.250
REPAIR LIMIT	1.950 ①	0.313 ① 0.3165 ④ 0.3125 ④ 0.3500 ⑤	--	--

REFINISH

- FILLER (225,245) - CHROMIC ACID ANODIZE AND APPLY BMS 10-11 TYP 1 PRIMER (F-18.13)
- TUBE (220) - CADMIUM PLATE AND APPLY PRIMER (F-16.01) ON OD. APPLY BMS 10-11, TYPE 1 PRIMER (F-20.03) ON ID.
- TUBE (240) - CADMIUM PLATE (SRF-1.611) AND APPLY PRIMER (SRF-12.205) ON OD. APPLY BMS 10-11, TYPE 1 PRIMER (F-20.02) ON ID

- ① FOR 69-37815-1:
LIMIT FOR SULFAMATE NICKEL PLATE BUILDUP (SION 20-42-09) AND MACHINE TO DESIGN DIMENSIONS AND FINISH
- ② INSTALL WITH BMS 5-95 SEALANT
- ③ HEAD OF RIVET MUST NOT BE ABOVE TUBE OD SURFACE
- ④ FOR 69-37815-1:
RANGE FOR INSTALLATION OF 5/16-DIAMETER BOLT (150)
- ⑤ FOR 69-37815-1:
LIMIT FOR INSTALLATION OF REPAIR BUSHINGS (FIGURE 602) AND STANDARD BOLT (150)
- ⑥ IF REPAIR BUSHINGS AS SHOWN IN FIGURE 602 ARE TO BE INSTALLED, THEN THIS CONTROL ALSO APPLIES TO THE BUSHING INNER DIAMETERS AFTER BUSHINGS ARE INSTALLED

REPAIR

- REF ① THRU ⑥
125 MACHINE FINISH
SHOT PEEN (SOPM 20-10-03): 0.016-0.033 SHOT SIZE
0.008 A2 INTENSITY
- MATERIAL: FILLER (225,245) - AL ALLOY
TUBE (220,240) - 4330M STEEL,
180-200 KSI
- ITEM NUMBERS REFER TO IPL FIG. 1
- ALL DIMENSIONS ARE IN INCHES

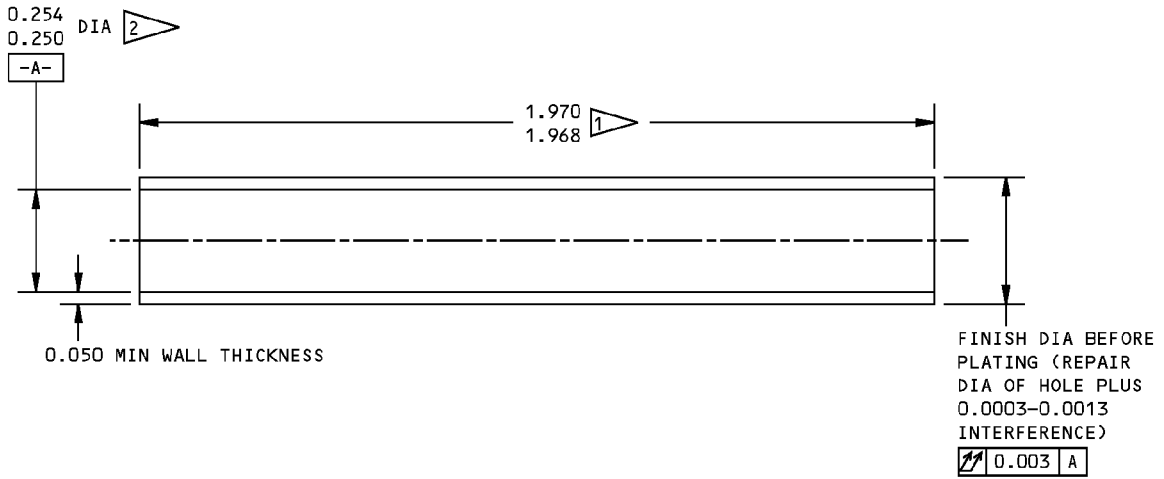
69-37815-1, -7, -8 Torque Tube Repair and Refinish
Figure 601 (Sheet 2 of 2)

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- 1 ADJUST THE LENGTH TO FIT FLUSH WITH THE SURFACE OR 0.010 MAXIMUM BELOW THE SURFACE
- 2 IF NECESSARY, YOU CAN LEAVE THIS DIAMETER UNDERSIZE IN ORDER TO MEET THE POSITIONAL TOLERANCE AFTER INSTALLATION, AS SHOWN IN FIG. 601

125/ MACHINE FINISH
 BREAK SHARP EDGES 0.01-0.02R
 PENETRANT CHECK (SOPM 20-20-02)
 CADMIUM PLATE (0.0003-0.0005 THICK, F-15.06)
 BUT NOT ON INSIDE DIAMETER
 MATERIAL: 15-5PH OR 17-4PH CRES, 180-200 KSI
 DIMENSIONS APPLY BEFORE PLATING
 ALL DIMENSIONS ARE IN INCHES

HOLE LOCATION (2) FIG. 601 - REPAIR BUSHING

Repair Bushing Details
 Figure 602

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REPAIR 4-1
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SUPPORT ASSEMBLY - REPAIR 5-1

69-38802-3, -4, -9, -10

1. General

- A. This repair gives the data to repair and refinish the inboard flap foreflap support assembly.
- B. Refer to REPAIR-GENERAL, Paragraph 2. for the Standard Overhaul Practices Manual (SOPM) subjects identified in this procedure.
- C. Refer to REPAIR-GENERAL, Paragraph 3. for the description of the consumable codes identified in this procedure.
- D. Refer to REPAIR-GENERAL, Figure 601 for the Standard True Position Dimensioning Symbols shown in the repair.
- E. Refer to IPL Figure 1 for item numbers.

2. Parts Replacement

- A. 69-38802-3, -4
 - (1) Remove the old rub strip.
 - (2) With the old rubstrip or the support as a guide, drill a hole through the replacement rubstrip for the rivet. Be sure to countersink the hole at the top surface of the rubstrip.
 - (3) Bond a replacement rub strip in position with adhesive, A01070 per SOPM 20-50-12, special method II. Install a new rivet (175).
- B. 69-38802-9, -10
 - (1) Remove the old bushing and the old rubstrip.
 - (2) With the old rub strip or the support as a guide, drill a hole through the replacement rub strip for the rivet. Be sure to countersink the hole at the top surface of the rub strip.
 - (3) Bond a replacement rub strip in position with adhesive, A01076 per SOPM 20-50-12. Install a new rivet (176) with sealant, A00247 (F-19.48).
 - (4) Install a replacement bushing (168) with sealant, A00247 by the shrink-fit method of SOPM 20-50-03. Make sure that the flange is on the same side as the rubstrip.
 - (5) Seal the flange of the bushing with sealant, A00247.

3. Repair of Attach Holes

- A. Method 1 – Installation of repair bushing
 - (1) Machine as necessary within repair limits to remove defects.
 - (2) Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
 - (3) Refinish as specified in REPAIR 5-1, Figure 601.
 - (4) Make a repair bushing (REPAIR 5-1, Figure 602) as required to adjust for the material removed.
 - (5) Install the repair bushing by the shrink-fit method (SOPM 20-50-03) with sealant, A00247 on mating surfaces.
 - (6) Machine the bushing bore to design dimensions and finish as specified in REPAIR 5-1, Figure 601.
- B. Method 2 – Installation of oversize bolts

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REPAIR 5-1

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- (1) Machine as necessary within repair limits to remove defects.
- (2) Do a class A magnetic particle check of machined areas as specified in the SOPM 20-20-01.
- (3) Refinish as specified in REPAIR 5-1, Figure 601.
- (4) Machine the mating holes in the torque tube oversize (REPAIR 4-1).
- (5) Make a note to be sure to use the oversize bolt and related washer and nut in the oversized holes.

4. Refinish

- A. Supports (180, 185, 186, 187) - Cadmium-titanium plate (F-15.01) and apply primer, C00259 (F-20.02), but no primer, C00259 in the holes. Material: 4330M steel, 220-240 ksi.

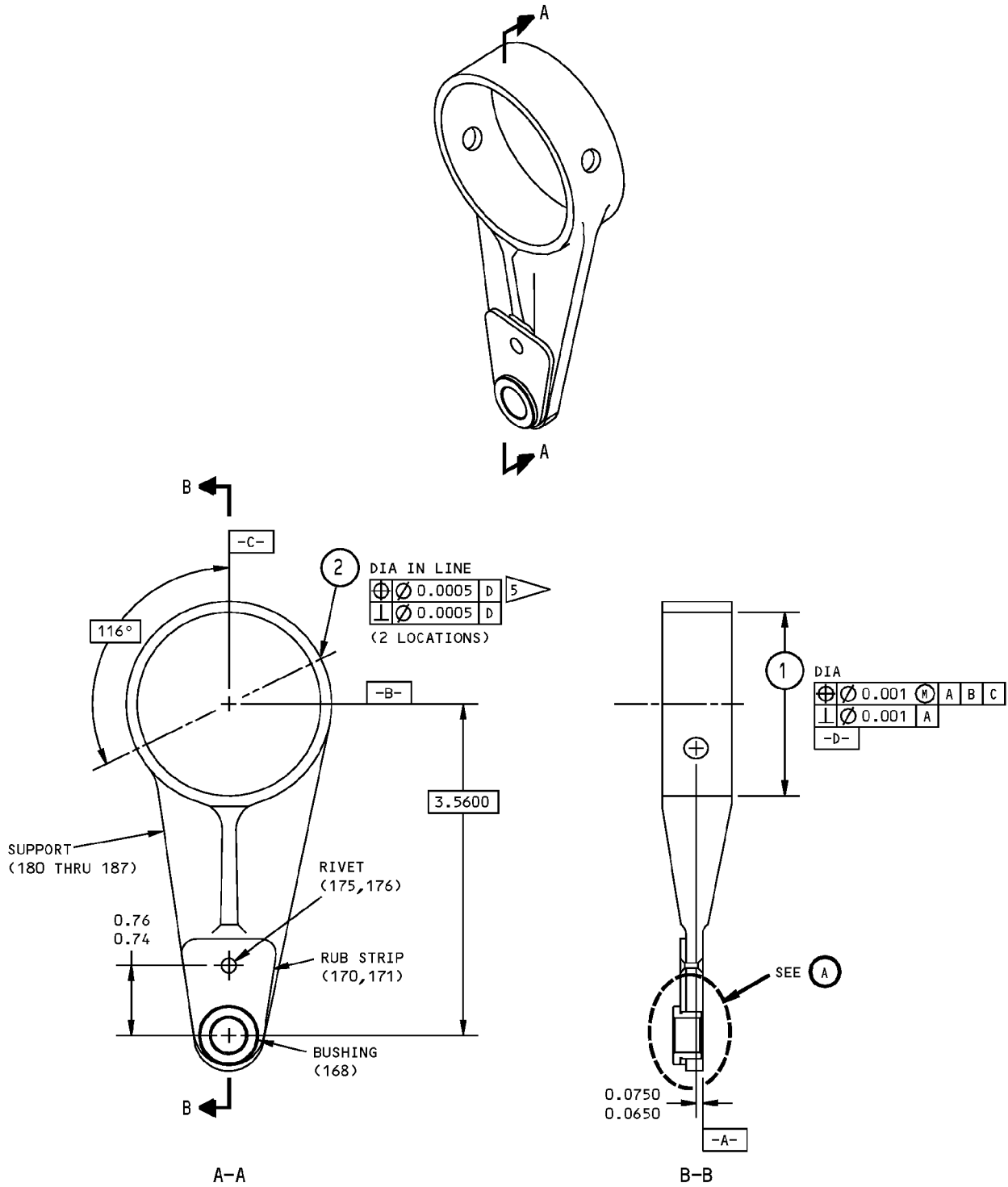
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69-38802-3,-4,-9,-10 Support Repair and Refinish
Figure 601 (Sheet 1 of 2)

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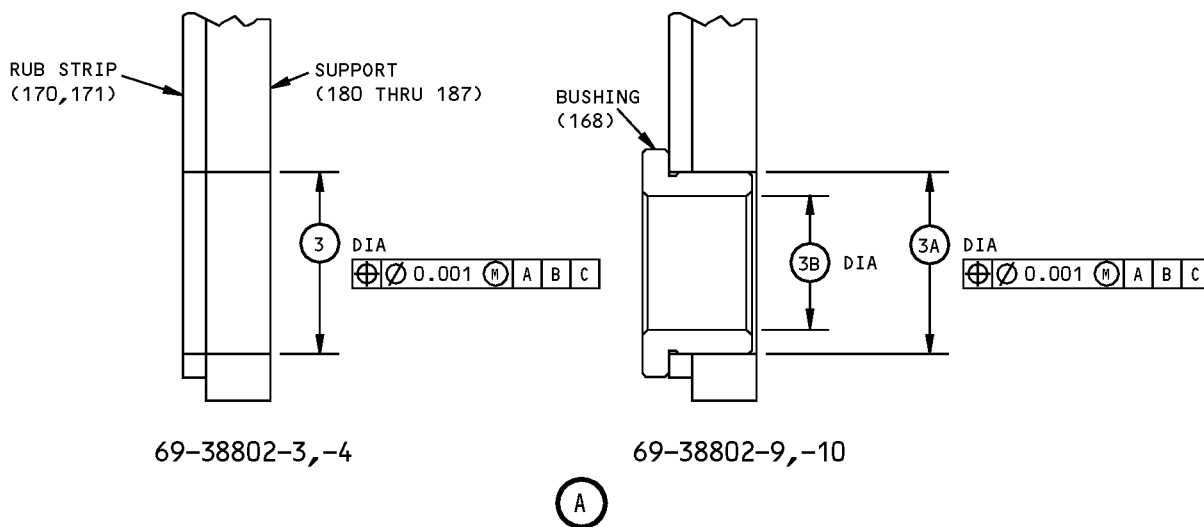
REPAIR 5-1

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	①	②	③ ①	③A ②	③B ②
DESIGN DIM	1.978 1.976	0.254 0.250	0.379 0.375	0.5003 0.4993	0.379 0.375
REPAIR LIMIT	---	0.3165 ③ 0.3125 ③ 0.3500 ④	---	---	---

REFINISH

CADMIUM-TITANIUM PLATE (F-15.01) (0.0003-0.0005 THICK ON DIAMETER ①). APPLY BMS 10-11, TYPE 1 PRIMER (F-20.02). BUT NO PRIMER IN HOLES

- ① FOR 69-38802-3,-4
- ② FOR 69-38802-9,-10
- ③ LIMIT FOR INSTALLATION OF 5/16-DIAMETER BOLT (150,190)
- ④ LIMIT FOR INSTALLATION OF REPAIR BUSHING AND STANDARD BOLT (150,190)
- ⑤ IF REPAIR BUSHINGS AS SHOWN IN FIG. 602 ARE TO BE INSTALLED, THEN THIS CONTROL ALSO APPLIES TO THE BUSHING INNER DIAMETERS AFTER BUSHINGS ARE INSTALLED

REPAIR

REF ③ ④

125/ MACHINE FINISH

SHOT PEEN (SOPM 20-10-03) BUT NOT IN HOLES:
RC 55-65 SHOT HEAT TREAT
0.017-0.046 SHOT SIZE
0.008-0.013 A2 INTENSITY

MATERIAL: 4330M STEEL, 220-240 KSI

DIMENSIONS APPLY AFTER PLATING

ALL DIMENSIONS ARE IN INCHES

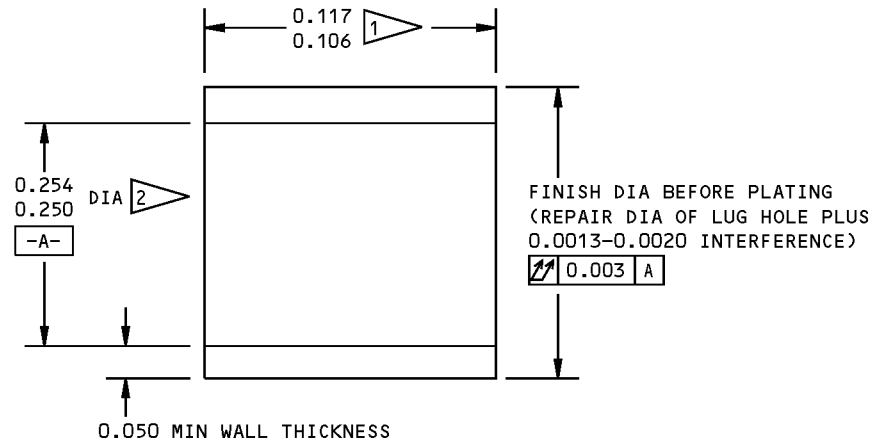
69-38802-3,-4,-9,-10 Support Repair and Refinish
Figure 601 (Sheet 2 of 2)

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- 1 ADJUST THE LENGTH TO FIT FLUSH WITH THE SURFACE OR 0.010 MAXIMUM BELOW THE SURFACE
- 2 IF NECESSARY, YOU CAN LEAVE THIS DIAMETER UNDERSIZE IN ORDER TO MEET THE POSITIONAL TOLERANCE AFTER INSTALLATION, AS SHOWN IN FIG. 601

REPAIR

125 ✓ ALL MACHINED SURFACES UNLESS SHOWN DIFFERENTLY

BREAK SHARP EDGES 0.01-0.02 R

MAGNETIC PARTICLE CHECK (SOPM 20-20-01)

CADMIUM PLATE (0.0003-0.0005 THICK, F-15.06)(OPTIONAL ON INTERNAL SURFACES)

MATERIAL: 15-5PH OR 17-4PH CRES, 180-200 KSI

DIMENSIONS APPLY BEFORE PLATING

ALL DIMENSIONS ARE IN INCHES

Repair Bushing Details
Figure 602

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ASSEMBLY

1. General

- A. This procedure has the necessary data to assemble the inboard flap foreflap assembly.
- B. Refer to the Standard Overhaul Practices Manual (SOPM) for the SOPM subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

2. Assembly

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
C00259	Primer - Chemical And Solvent Resistant Finish, Epoxy Resin	BMS10-11, Type I
D00013	Grease - Aircraft And Instrument Grease	MIL-PRF-23827 (NATO G-354) (Supersedes MIL-G-23827)
G02166	Lockwire - Nickel-Copper Alloy (Monel)(0.020 inch Diameter)	NASM20995N~ C20
G02436	Lockwire - Monel (0.040 In. Dia.)	NASM20995N~ C40

B. References

Reference	Title
SOPM 20-50-01	BOLT AND NUT INSTALLATION
SOPM 20-50-02	INSTALLATION OF SAFETYING DEVICES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT
SOPM 20-50-07	LUBRICATION
SOPM 20-60-02	FINISHING MATERIALS
SOPM 20-60-03	LUBRICANTS
SOPM 20-60-04	MISCELLANEOUS MATERIALS

C. Procedure

NOTE: For bolt and nut installation, refer to SOPM 20-50-01. For finishing materials, refer to SOPM 20-60-02. For lubricants, refer to SOPM 20-60-03. For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Use standard industry practices plus the following steps.
- (2) Install all lockbolts with wet primer, C00259.
- (3) Install all bearings (20, 20A, 20B, 65, 65A) per SOPM 20-50-03. Coat all surfaces with a light coat of grease, D00013 according to SOPM 20-50-07 prior to installation.

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ASSEMBLY

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- (4) Lightly coat all surfaces of bolts (150, 190, 210, 285, 310, 365) with grease, D00013 according to SOPM 20-50-07 prior to installation.
- (5) Tighten bolts (210, 285, 340, 560A, 580A) to 65-90 pound-inches.
- (6) Tighten bolt (310) to 130-180 pound-inches.
- (7) Tighten bolt (365) to 220-360 pound-inches.
- (8) Secure bolts (560A, 580A) with lockwire, G02166 per SOPM 20-50-02 after installation.
- (9) Secure bolts (285, 310, 340, 365) with lockwire, G02436 per SOPM 20-50-02 after installation.
- (10) Delaminate shims (290, 315, 345, 370), as required, to provide trailing edge adjustments on installation. Only one shim allowed per location with maximum permissible gap, after delamination, of 0.002 inches. Install shims (290, 315, 345, 370) with wet or dry primer, C00259.
- (11) Delaminate shim (555), as required, to provide trailing edge adjustments on installation; a maximum of two shims per location are allowed. Maximum permissible gap, after delamination, is 0.002 inch. Install shim (555) with wet or dry primer, C00259.
- (12) Tighten nuts (10, 25, 75) to 50-100 pound-inches.
- (13) Install bolt (150, 190) and nut (160, 200) finger-tight only.

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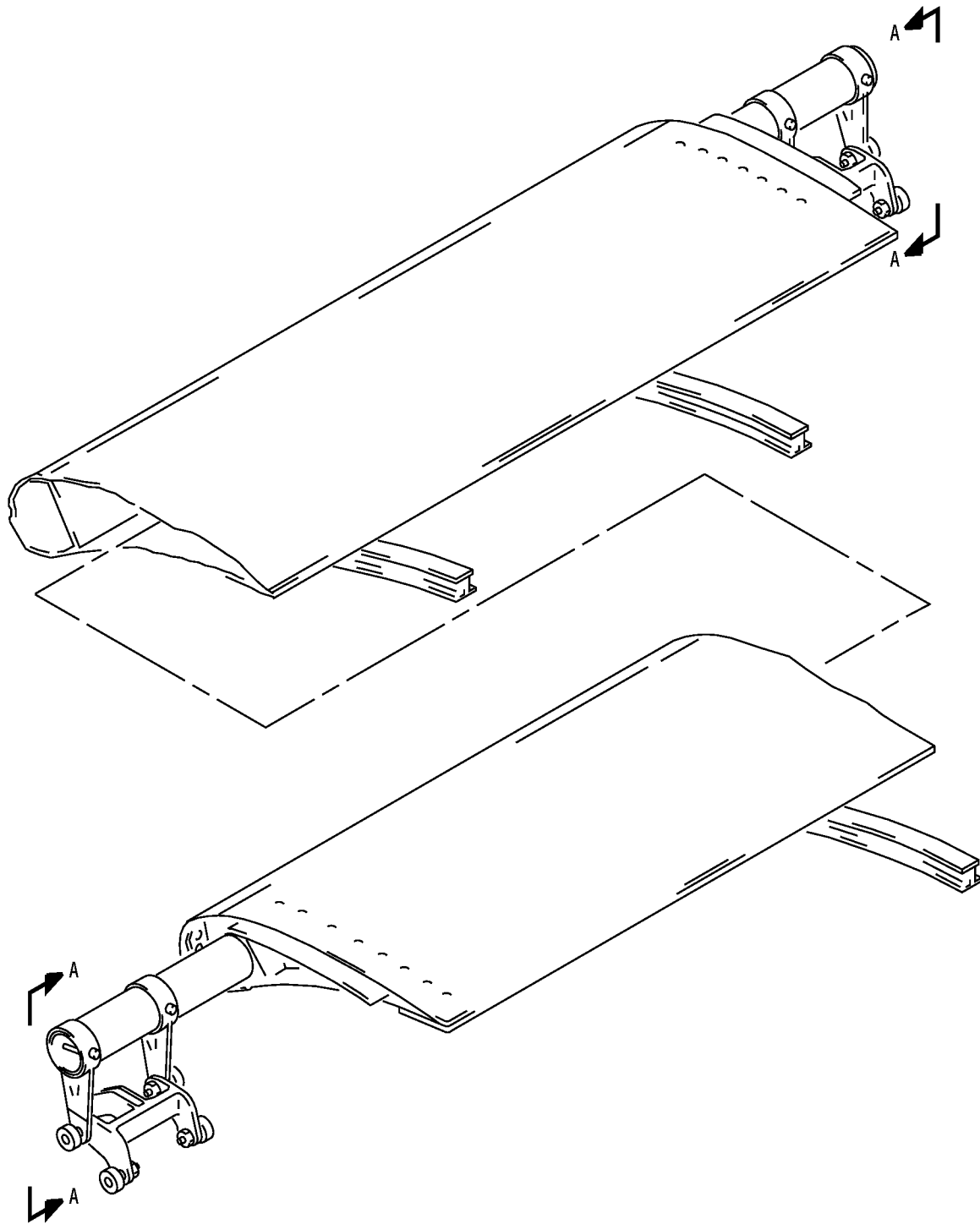
ASSEMBLY

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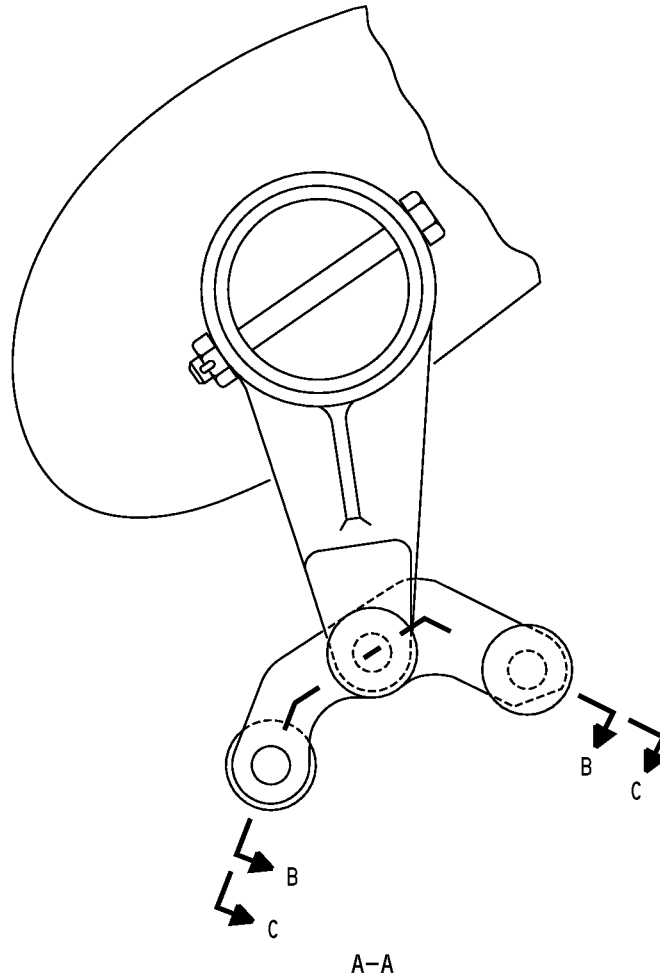
FITS AND CLEARANCES



Fits and Clearances
Figure 801 (Sheet 1 of 4)

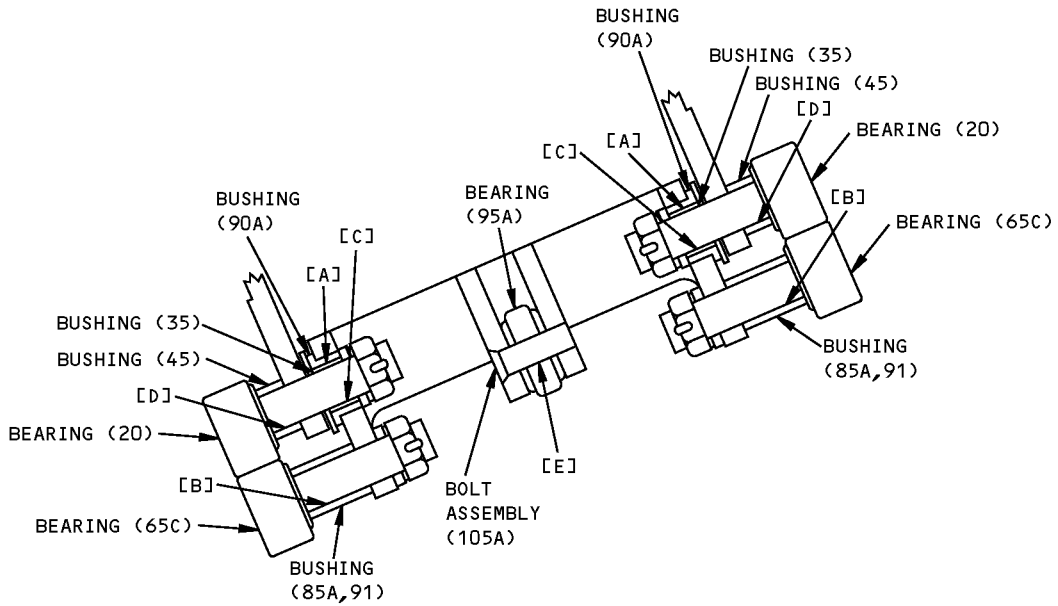
57-53-26
FITS AND CLEARANCES
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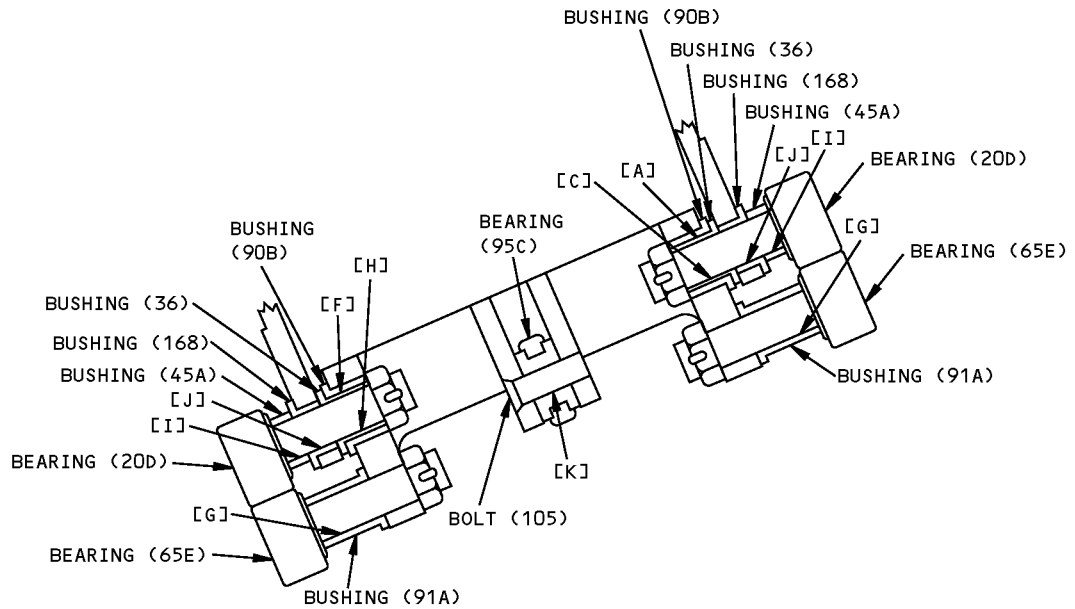


Fits and Clearances
Figure 801 (Sheet 2 of 4)

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B-B



C-C

Fits and Clearances
Figure 801 (Sheet 3 of 4)

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
FITS AND CLEARANCES

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REF LETTER	REF IPL	DESIGN DIMENSION*				SERVICE WEAR LIMIT*		
	FIG. 1, MATING ITEM NO.	DIMENSION		ASSEMBLY CLEARANCE 		DIMENSION		MAXIMUM CLEARANCE
		MIN	MAX	MIN	MAX	MIN	MAX	
[A]	ID 90A	0.5030	0.5040	0.0017	0.0034		0.5081	0.0068
	OD 35	0.5006	0.5013			0.4962		
[B]	ID 85A	0.3750	0.3755	0.0000	0.0020		0.3800	0.0050
	OD 65C	0.3735	0.3750			0.3700		
[B]	ID 91	0.3750	0.3765	0.0000	0.0030		0.3800	0.0050
	OD 65C	0.3735	0.3750			0.3700		
[C]	ID 35	0.3750	0.3765	0.0000	0.0025		0.3800	0.0050
	OD 20	0.3740	0.3750			0.3700		
[D]	ID 45	0.3750	0.3765	0.0000	0.0025		0.3800	0.0050
	OD 20	0.3740	0.3750			0.3700		
[E]	ID 95A	0.2493	0.2500	-0.0020	-0.0002			
	OD 105A	0.2502	0.2513					
[F]	ID 90B	0.5014	0.5024	0.0001	0.0018			
	OD 36	0.5006	0.5013					
[G]	ID 91A	0.3750	0.3765	0.0000	0.0030			
	OD 65E	0.3735	0.3750					
[H]	ID 36	0.3750	0.3765	0.0000	0.0030			
	OD 20D	0.3735	0.3750					
[I]	ID 45A	0.3750	0.3765	0.0000	0.0030			
	OD 20D	0.3735	0.3750					
[J]	ID 168	0.3750	0.3765	0.0000	0.0030			
	OD 20D	0.3735	0.3750					
[K]	ID 95C	0.2495	0.2500	-0.0015	-0.0005			
	OD 105	0.2505	0.2510					

* ALL DIMENSIONS ARE IN INCHES





 NEGATIVE VALUE DENOTES INTERFERENCE FIT

Fits and Clearances
Figure 801 (Sheet 4 of 4)


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REF IPL		NAME	TORQUE*	
FIG. NO.	ITEM NO.		POUND-INCHES	POUND-FEET
1	10	NUT	50-100	
1	25	NUT	50-100	
1	75	NUT	50-100	
1	150	BOLT		
1	160	NUT		
1	190	BOLT		
1	200	NUT		
1	210	SCREW	65-90	
1	285	BOLT	65-90	
1	310	BOLT	130-180	
1	340	BOLT	65-90	
1	365	BOLT	220-360	
1	430	NUT	50-100	
1	560	BOLT	65-90	
1	580	BOLT	65-90	

* REFER TO SOPM 20-50-01 FOR TORQUE VALUES OF STANDARD FASTENERS.

 FINGER TIGHT

Torque Table
Figure 802



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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

(NOT APPLICABLE)

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SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

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ILLUSTRATED PARTS LIST

1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
.	Assembly					
.	Attaching parts for assembly					
.	.	Detail parts for assembly				
.	.	Subassembly				
.	.	Attaching parts for subassembly				
.	.	.	Detail parts for subassembly			
.	.	.	Sub-subassembly			
.	.	.	Attaching parts for subassembly			
.	.	.	.	Details parts for sub-subassembly		
						Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
- (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
- (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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Optional (OPT)	The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)	The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)	The part replaces and is interchangeable with, or is an alternative to, the initial part.

VENDOR CODES

Code	Name
21335	TIMKEN US CORPORATION DIV FAFNIR 336 MECHANIC STREET LEBANON, NH 03766-0267 FORMERLY FAFNIR BRG AND TEXTRON INC FAFNIR DIV IN NEW BRITAIN, CONNECTICUT ; FORMERLY TORRINGTON CO THE SPECIAL PRODUCTS DIV SUB OF THE INGERSOLL-RAND CO V8D210 FORMERLY TORRINGTON CO FAFNIR BEARING DIV IN TORRINGTON, CT
50632	KAMATICS CORP SUB OF KAMAN CORP 1335 BLUE HILLS ROAD BLOOMFIELD, CONNECTICUT 06002-1304

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NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-46430-1		1	120	1
65-46430-100SP		1	5F	RF
65-46430-101		1	120D	1
65-46430-102		1	125D	1
65-46430-111SP		1	1G	RF
65-46430-112SP		1	5G	RF
65-46430-113		1	480B	1
65-46430-114		1	485B	1
65-46430-115		1	490D	1
65-46430-116		1	495D	1
65-46430-117		1	120E	1
65-46430-118		1	125E	1
65-46430-119		1	120F	1
65-46430-120		1	125F	1
65-46430-121SP		1	1H	RF
65-46430-122SP		1	5H	RF
65-46430-123		1	490A	1
65-46430-124		1	495A	1
65-46430-125		1	120G	1
65-46430-126		1	125G	1
65-46430-127SP		1	1J	RF
65-46430-128SP		1	5J	RF
65-46430-129SP		1	1K	RF
65-46430-130SP		1	5K	RF
65-46430-131		1	120H	1
65-46430-132		1	125H	1
65-46430-133		1	490C	1
65-46430-134		1	495C	1
65-46430-135		1	120I	1
65-46430-136		1	125I	1
65-46430-137SP		1	1L	RF
65-46430-138SP		1	5L	RF
65-46430-139		1	120J	1
65-46430-140		1	125J	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-46430-141SP		1	1M	RF
65-46430-142SP		1	5M	RF
65-46430-149		1	120K	1
65-46430-15		1	500	1
65-46430-150		1	125K	1
65-46430-151		1	120L	1
65-46430-152		1	125L	1
65-46430-153		1	511	1
65-46430-157		1	490G	1
65-46430-158		1	495G	1
65-46430-159		1	120M	1
65-46430-160		1	125M	1
65-46430-161SP		1	1N	RF
65-46430-162SP		1	5N	RF
65-46430-163SP		1	1P	RF
65-46430-164SP		1	5P	RF
65-46430-165SP		1	1Q	RA
65-46430-166SP		1	5Q	RF
65-46430-167SP		1	1R	RF
65-46430-168SP		1	5R	RF
65-46430-169		1	120N	1
65-46430-170		1	125N	1
65-46430-171		1	120P	1
65-46430-172		1	125P	1
65-46430-173SP		1	1S	RF
65-46430-174SP		1	5S	RF
65-46430-175		1	120Q	1
65-46430-176		1	125Q	1
65-46430-177SP		1	1T	RF
65-46430-178SP		1	5T	RF
65-46430-179		1	120R	1
65-46430-180		1	125R	1
65-46430-181SP		1	1U	RF
65-46430-182SP		1	5U	RF
65-46430-183		1	120S	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-46430-184		1	125S	1
65-46430-185SP		1	1V	RF
65-46430-186SP		1	5V	RF
65-46430-187		1	120T	1
65-46430-188		1	125T	1
65-46430-189SP		1	1W	RF
65-46430-190SP		1	5W	RF
65-46430-191		1	120U	1
65-46430-192		1	125U	1
65-46430-2		1	125	1
65-46430-25		1	130	1
65-46430-26		1	510	1
65-46430-27		1	505	1
65-46430-38		1	520	2
65-46430-43		1	1	RF
65-46430-44		1	5	RF
65-46430-503		1	490	1
65-46430-504		1	495	1
65-46430-505		1	480	1
65-46430-506		1	485	1
65-46430-52		1	480C	1
65-46430-522		1	495B	1
65-46430-523		1	490B	1
65-46430-53		1	485C	1
65-46430-54		1	490E	1
65-46430-55		1	495E	1
65-46430-75		1	1A	RF
65-46430-76		1	5A	RF
65-46430-77		1	120A	1
65-46430-78		1	125A	1
65-46430-79		1	1B	RF
65-46430-80		1	5B	RF
65-46430-83		1	1C	RF
65-46430-84		1	5C	RF
65-46430-85		1	120B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
65-46430-86		1	125B	1
65-46430-87SP		1	1D	RF
65-46430-88SP		1	5D	RF
65-46430-89		1	120C	1
65-46430-90		1	125C	1
65-46430-91		1	480D	1
65-46430-92		1	485D	1
65-46430-93		1	490F	1
65-46430-94		1	495F	1
65-46430-97SP		1	1E	RF
65-46430-98SP		1	5E	RF
65-46430-99SP		1	1F	RF
65-68278-23		1	460	1
65-68278-24		1	455	1
65-68278-30		1	445	1
65-68278-31		1	450	1
65-68278-44		1	475	1
65-68278-57		1	465	1
65-68278-58		1	470	1
65-68278-59		1	445A	1
65-68278-60		1	450A	1
65-68278-65		1	465A	1
65-68278-66		1	470A	1
65-68278-67		1	460A	1
65-68278-68		1	455A	1
65C32620-1		1	480A	1
65C32620-11		1	480E	1
65C32620-12		1	485E	1
65C32620-2		1	485A	1
65C33100-1		1	60A	1
65C33100-3		1	60B	1
65C33100-4		1	60C	1
65C37022-1		1	171	1
66-23212-1		1	105A	1
66-23212-2		1	115	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
66-26119-1		1	285A	1
		1	285B	1
		1	285C	1
		1	285F	1
		1	340A	2
		1	340B	2
		1	340C	2
		1	340F	2
66-26119-2		1	310A	1
		1	310B	1
		1	310C	1
		1	310G	1
66-26119-3		1	365A	2
		1	365B	2
		1	365C	2
		1	365G	2
66-26119-4		1	310F	1
66-26119-5		1	365F	2
69-35329-1		1	60	1
69-35336-1		1	55	2
69-35336-2		1	55A	2
69-35336-3		1	55B	2
69-35349-1		1	335	2
69-35349-3		1	280	1
69-35349-5		1	281	2
69-35349-7		1	282	1
69-37815-1		1	215	1
		1	235	1
69-37815-10		1	240B	1
69-37815-3		1	220	1
		1	240	1
69-37815-5		1	225	1
		1	245	1
69-37815-7		1	235A	1
69-37815-8		1	235B	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-37815-9		1	240A	1
69-38802-10		1	147	4
69-38802-11		1	186	1
69-38802-12		1	187	1
69-38802-3		1	140	2
69-38802-4		1	145	2
69-38802-5		1	180	1
69-38802-6		1	185	1
69-38802-7		1	170	1
69-38802-9		1	146	4
69-73105-1		1	525	1
		1	525A	1
69-73105-2		1	530	1
		1	530A	1
69-73105-3		1	535	1
69-73105-4		1	540	1
69-73105-7		1	525B	1
		1	525C	1
69-73105-8		1	530B	1
		1	530C	1
69-76975-1		1	608	1
69-76975-2		1	608A	1
69-76975-3		1	608B	1
69-76975-4		1	608C	1
69-76976-1		1	609H	1
69-76976-10		1	609C	1
69-76976-11		1	609S	1
69-76976-12		1	609T	1
69-76976-2		1	609I	1
69-76976-3		1	609	1
69-76976-4		1	609A	1
69-76976-5		1	609Q	1
69-76976-6		1	609R	1
69-76976-7		1	609J	1
69-76976-8		1	609K	1

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
69-76976-9		1	609B	1
69-76977-1		1	610	5
69-76977-2		1	610A	5
69-76977-3		1	610B	5
69-76977-4		1	610C	5
69-77698-1		1	105	1
69-78026-1		1	611D	1
69-78026-2		1	611E	1
69-78026-3		1	611F	1
69-78026-4		1	611G	1
69-78112-1		1	611	1
69-78112-2		1	611A	1
69-78112-3		1	611B	1
69-78112-4		1	611C	1
AL63858		1	95D	2
AN320-4		1	160	4
		1	200	2
AN320-5		1	160A	AR
AN960-416		1	155	4
		1	195	2
AN960-516		1	155A	AR
AN960-616		1	30	4
		1	70	2
AN960-616L		1	15	AR
		1	71	AR
AN960C816L		1	40	4
AN960KD10L		1	435A	6
AN960KD416		1	195B	2
AN960PD10L		1	435	4
BACB10AF6F14H		1	65	2
BACB10AF6F16H		1	20	4
BACB10B131		1	95	1
BACB10B196-11		1	65B	2
BACB10BH60F11		1	20A	4
		1	65A	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACB10BH60F12		1	20B	4
BACB10ET04		1	95A	1
BACB10FK6F14HS		1	65C	2
BACB10FK6F16HS		1	20C	4
BACB28AK06-062		1	85A	2
BACB28AT08B025C		1	90A	2
BACB28W4C026		1	545	1
BACB28W6B026		1	550	1
BACB28X6C051		1	36	4
BACB28X6C055		1	85B	2
		1	91	2
		1	91A	2
BACB28X6C23		1	168	1
BACB28X8B25		1	90	2
BACB28Y6C018		1	45A	4
BACB28Y6C43		1	35	4
BACB30DM4-8H		1	285	1
		1	285D	1
		1	340	2
		1	340D	2
BACB30DM5-9H		1	310	1
		1	310D	1
BACB30DM6-10H		1	365	2
		1	365D	2
BACB30LE4H10		1	560A	1
BACB30LE4H13		1	580A	1
BACB30LE4HK10		1	285H	1
		1	612	1
BACB30LE4HK13		1	310J	1
		1	613	1
BACB30LU3-3		1	430	4
BACB30NN3K4		1	430A	6
		1	430B	3
		1	614	3
BACB30NN4K6		1	430C	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	615	2
BACB30NR4DK36		1	150B	4
		1	430E	4
		1	616	4
BACB30NR4DK38		1	190B	2
		1	617	2
BACB30NR5DK36		1	150C	AR
BACB30NRDK38		1	430F	2
BACB30UU4K8H		1	285E	1
		1	340E	2
BACB30UU5K11N		1	310E	1
BACB30UU6K11H		1	365E	2
BACB30VF4K6		1	430D	2
		1	618	2
BACN10CP4H		1	300	1
		1	360	2
BACN10CP5H		1	330	1
BACN10CP6H		1	385	2
BACN10HC4		1	300A	1
		1	360A	2
		1	385A	2
		1	570	2
		1	619	5
BACN10HC5		1	330A	1
		1	620	1
BACN10HC6		1	621	2
BACN10JD104CD		1	160B	4
		1	200B	2
BACN10JD105CD		1	160C	AR
BACN10JD106		1	10	4
		1	10B	4
		1	75A	2
BACN10JD106CD		1	10A	4
		1	75B	2
BACN10JD4CD		1	200D	2

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
		1	622	2
BACN10JD6		1	25	4
		1	75	2
BACN10JN3CM		1	623	12
BACN10JP4DCD		1	230A	1
		1	250	1
BACN10YR3CD		1	440A	6
		1	624	3
BACR10V4		1	295	1
		1	350	2
		1	625	3
BACR10V5		1	320	1
		1	575	2
		1	626	3
BACR10V6		1	375	2
		1	627	2
BACR15BA3D		1	232	2
		1	255	2
BACR15DR3PAC2		1	176	4
BACS12N10-9		1	515	12
		1	628	12
BACS40R002B014F		1	605	3
BACS40R007B014F		1	629	1
BACS40R008B014F		1	630	1
BACS40R008C14F		1	631	1
BACS40R009B024F		1	632	1
BACS40R009C012F		1	633	1
BACS40R009C013F		1	634	1
BACS40R010C021F		1	635	1
BACS40R010F010F		1	636	1
BACS40R014D014F		1	607	3
		1	637	1
BACS40R07B14F		1	290	AR
		1	345	AR
BACS40R100F100F		1	555	AR

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
BACS40R14D14F		1	315	1
		1	370	AR
BACW10BN4C		1	305	1
		1	355	2
BACW10BN5C		1	325	1
BACW10BN6C		1	380	2
BACW10BP41CD		1	565	2
		1	638	2
BACW10BP4CD		1	305A	1
		1	355A	2
		1	380A	2
		1	585	3
BACW10BP5CD		1	325A	1
		1	588	1
BACW10BP6CD		1	590	2
BACW10DS4		1	355B	2
BACW10DS4S		1	305B	1
		1	306	1
		1	356	2
		1	380B	2
		1	380C	2
		1	595	3
		1	639	3
		1	325B	1
BACW10DS5S		1	326	1
		1	598	1
		1	640	1
		1	600	2
BACW10DS6S		1	641	2
		1	65G	2
CL66109		1	20	4
ITEM 20 T/W ITEMS 10 AND 15 IS OPTIONAL TO ITEM 20A T/W ITEM 10 OR OPTIONAL TO ITEM 20B T/W ITEM 15 AND 25		1	20A	4
		1	20B	4

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
KJB331308V		1	90B	2
KRP141500VT6-16		1	65D	2
		1	65E	2
KRP141500VT6-17		1	20D	4
		1	65F	2
KRP502-4VT		1	95C	1
MS16562-10		1	100	1
MS20426B3		1	175	1
MS20470D3		1	135	4
MS20995NC20		1	642	1
MS24665-134		1	165	4
		1	205	2
MS24665-153		1	205A	2
		1	643	2
MS24665-283		1	50	4
		1	80	2
NAS1104-36D		1	150	4
NAS1104-38D		1	190	2
NAS1105-36D		1	150A	AR
NAS1149C0632R		1	71A	AR
		1	71B	AR
NAS1149C0663R		1	31	4
		1	70A	2
		1	70B	2
NAS516-1		1	110	1
NAS517-4-6		1	210	2
NAS6704H8		1	285G	1
		1	340G	2
		1	644	3
NAS6705H11		1	310H	1
		1	645	1
NAS6706H11		1	365H	2
		1	646	2
NAS679A3W		1	440	4
NAS75-6-008		1	45	4

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PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
NAS75-6-020		1	85	2

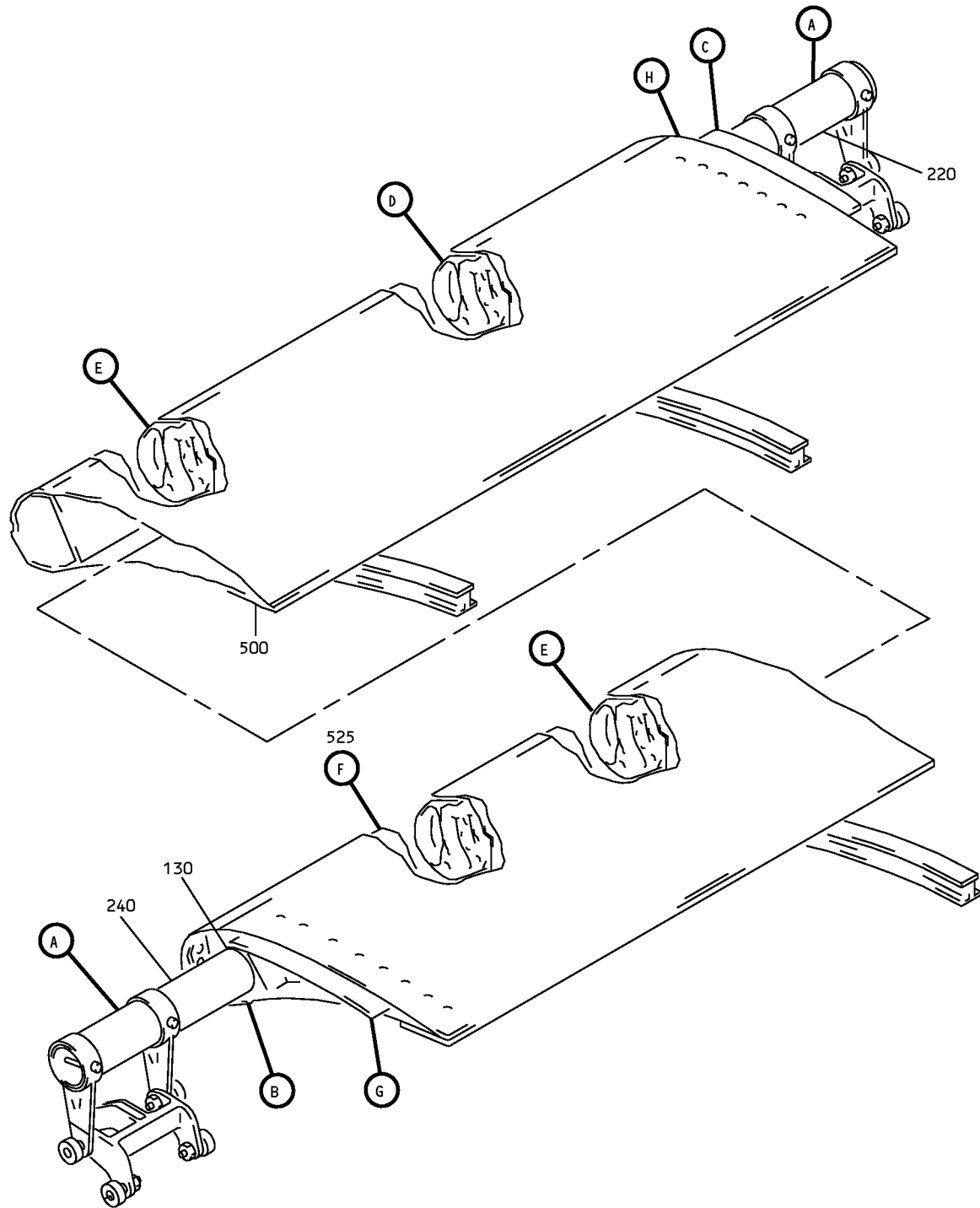
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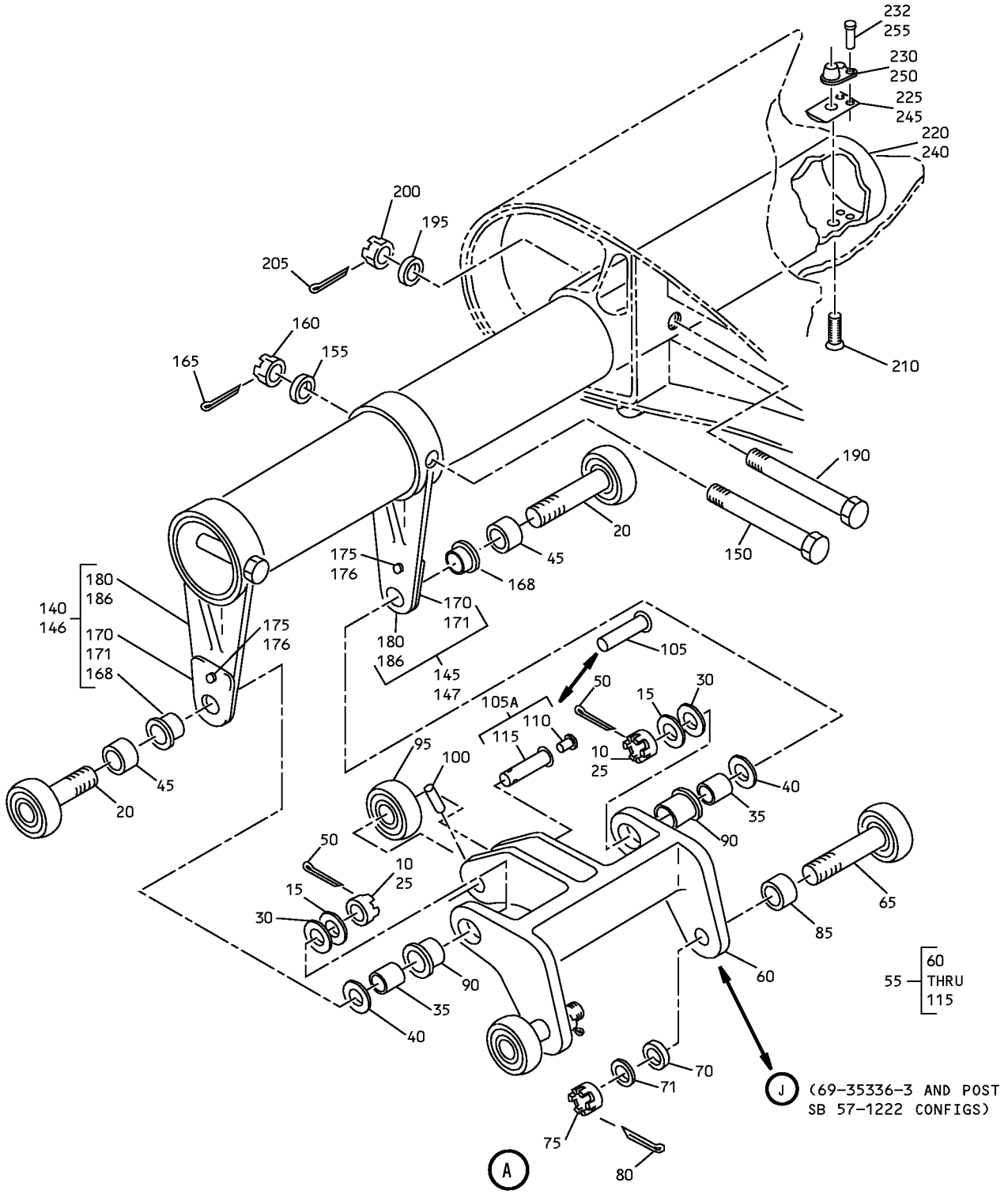
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Foreflap Assembly - Inboard Flap
 IPL Figure 1 (Sheet 1 of 8)

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Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 2 of 8)

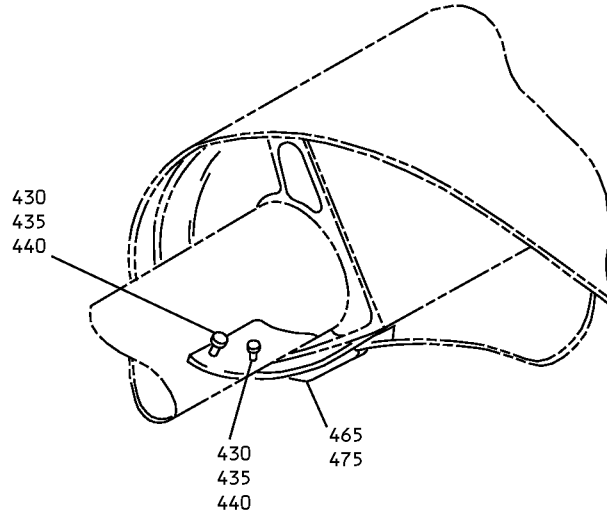
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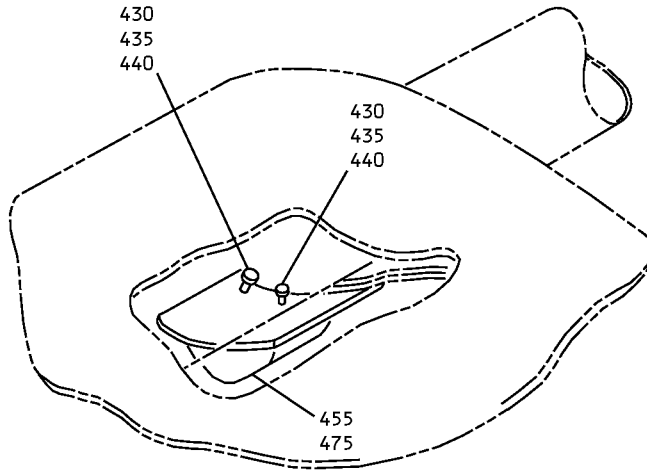
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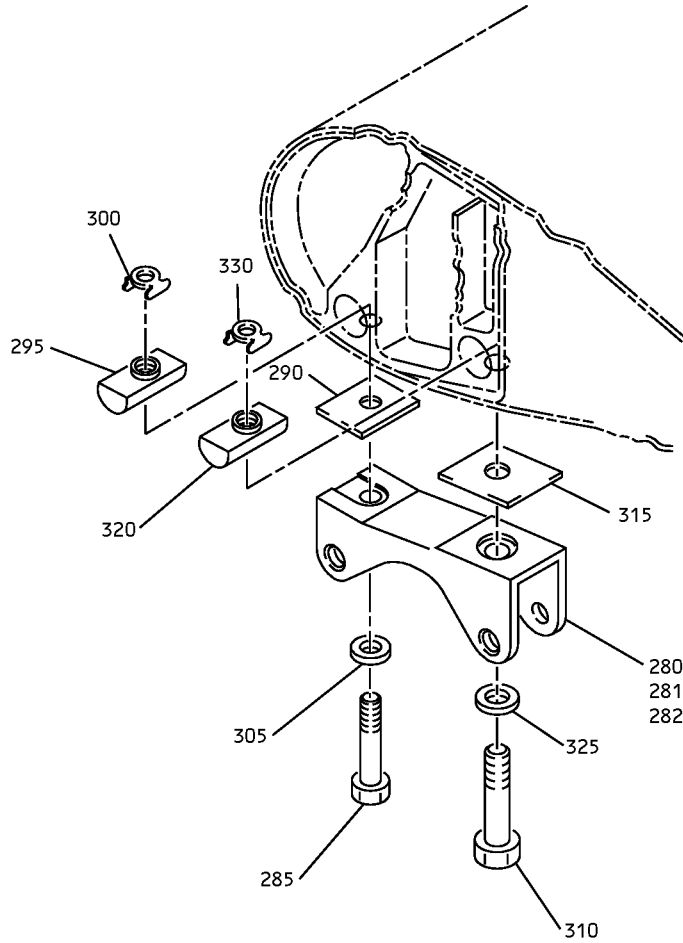
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C

Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 3 of 8)

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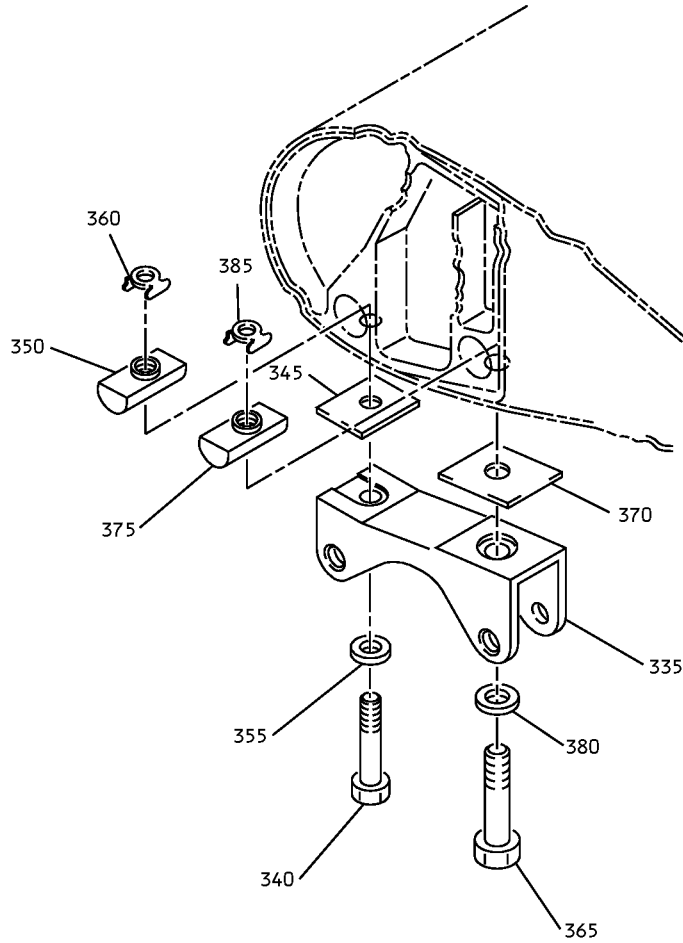


(WBL 84.96)



Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 4 of 8)

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(WBL 117.10, 159.50)

E

Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 5 of 8)

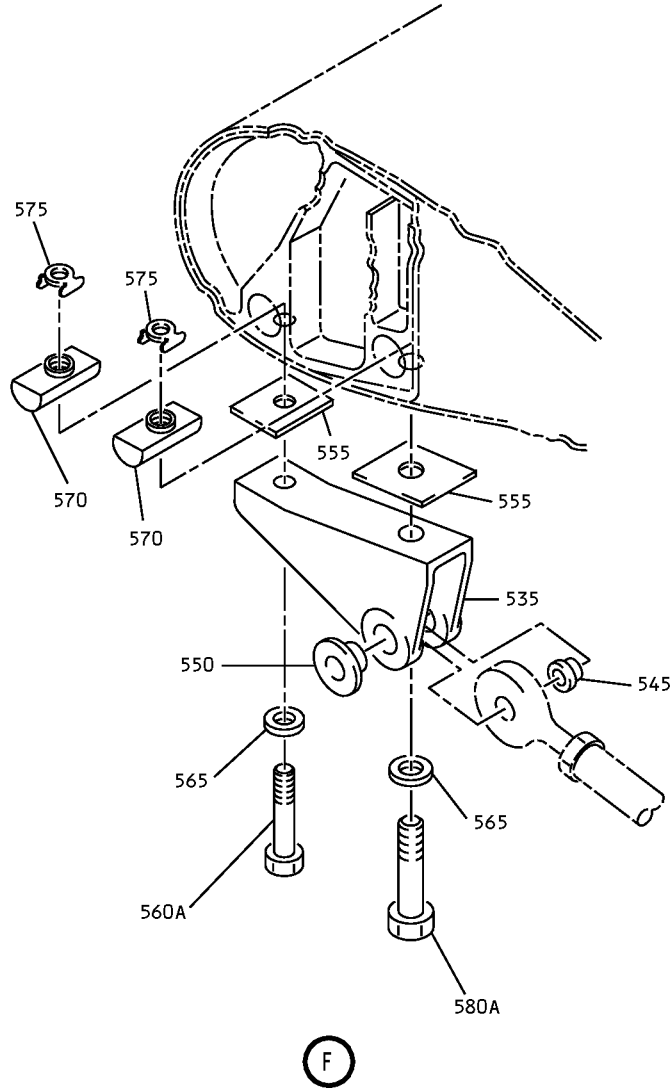
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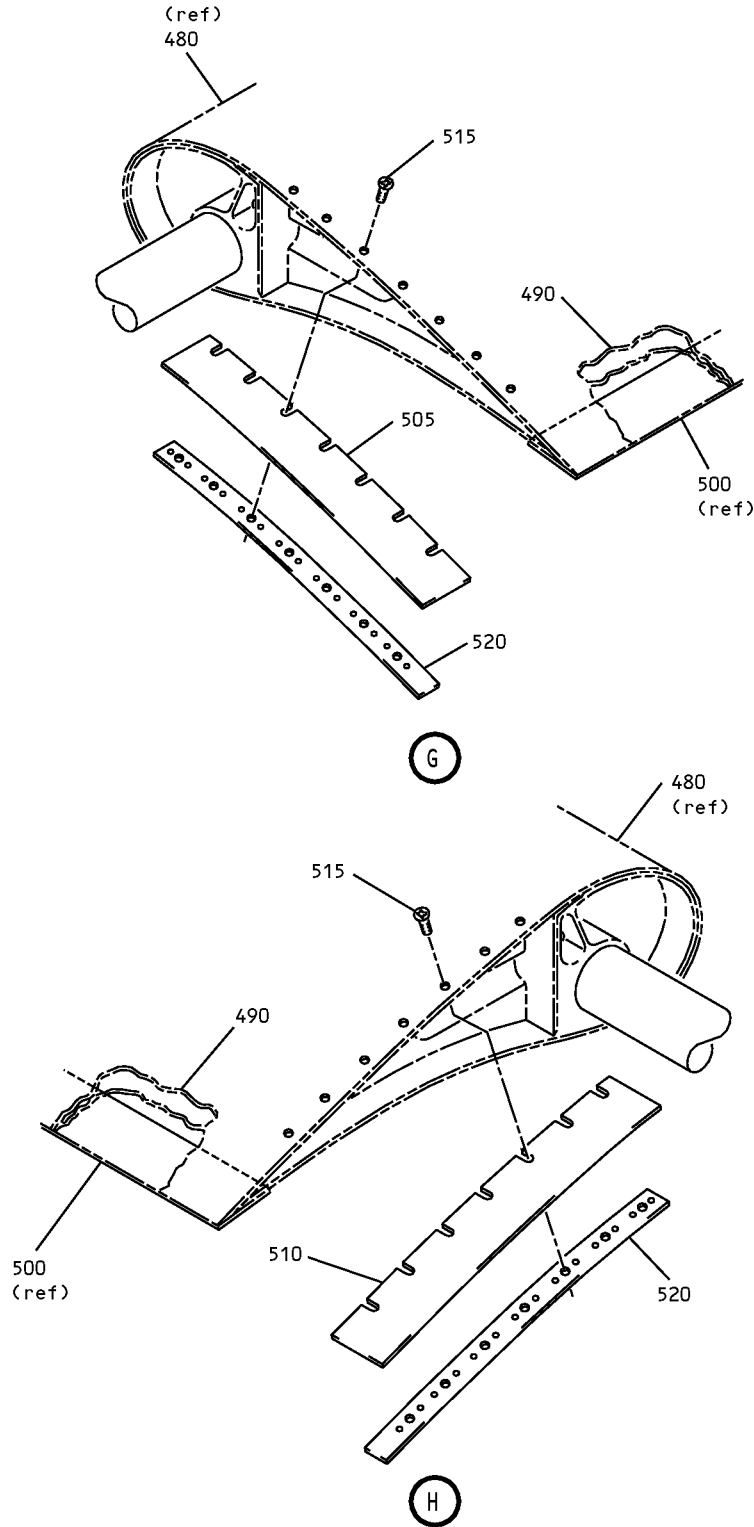
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Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 6 of 8)

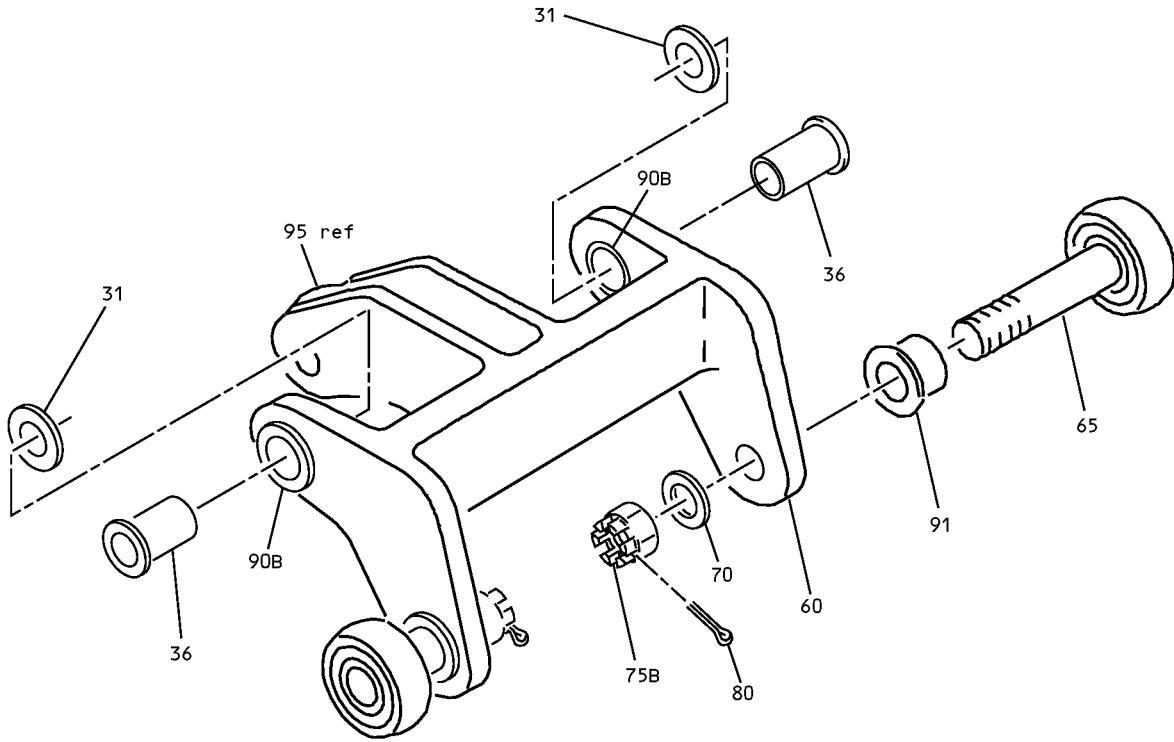
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Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 7 of 8)

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69-35336-3 AND CONFIGS POST SB 57-1222

J

Foreflap Assembly - Inboard Flap
IPL Figure 1 (Sheet 8 of 8)

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1	65-46430-43									A	RF
-1A	65-46430-75									C	RF
-1B	65-46430-79									E	RF
-1C	65-46430-83									G	RF
-1D	65-46430-87SP									I	RF
-1E	65-46430-97SP									K	RF
-1F	65-46430-99SP									M	RF
-1G	65-46430-111SP									O	RF
-1H	65-46430-121SP									Q	RF
-1J	65-46430-127SP									S	RF
-1K	65-46430-129SP									U	RF
-1L	65-46430-137SP									W	RF
-1M	65-46430-141SP									Y	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-1N	65-46430-161SP									BA	RF
-1P	65-46430-163SP									DA	RF
1Q	65-46430-165SP									FA	RA
-1R	65-46430-167SP									HA	RF
-1S	65-46430-173SP									JA	RF
-1T	65-46430-177SP									LA	RF
-1U	65-46430-181SP									NA	RF
-1V	65-46430-185SP									PA	RF
-1W	65-46430-189SP									RA	RF
-5	65-46430-44									B	RF
-5A	65-46430-76									D	RF
-5B	65-46430-80									F	RF
-5C	65-46430-84									H	RF
-5D	65-46430-88SP									J	RF
-5E	65-46430-98SP									L	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-5F	65-46430-100SP									N	RF
-5G	65-46430-112SP									P	RF
-5H	65-46430-122SP									R	RF
-5J	65-46430-128SP									T	RF
-5K	65-46430-130SP									V	RF
-5L	65-46430-138SP									X	RF
-5M	65-46430-142SP									Z	RF
-5N	65-46430-162SP									CA	RF
-5P	65-46430-164SP									EA	RF
-5Q	65-46430-166SP									GA	RF
-5R	65-46430-168SP									IA	RF
-5S	65-46430-174SP									KA	RF
-5T	65-46430-178SP									MA	RF
-5U	65-46430-182SP									OA	RF
-5V	65-46430-186SP									QA	RF

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-5W	65-46430-190SP									SA	RF
10	BACN10JD106										4
-10A	BACN10JD106CD									FA-SA	4
-10B	BACN10JD106									S-X, BA- EA	4
15	AN960-616L										AR
20	BACB10AF6F16H									A-R, Y, Z	4
-20A	BACB10BH60F11									A-R, Y,Z	4
-20B	BACB10BH60F12									A-R, Y, Z	4
-20C	BACB10FK6F16HS									S-X, BA- EA	4
-20D	KRP141500VT6-17									FA-SA	4
25	BACN10JD6										4
30	AN960-616									A-GA	4
31	NAS1149C0663R									HA-SA	4
35	BACB28Y6C43									A-GA	4
36	BACB28X6C051									HA-SA	4
40	AN960C816L									A-GA	4
45	NAS75-6-008									A-GA	4
-45A	BACB28Y6C018									HA-SA	4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
50	MS24665-283										4
-55	69-35336-1									A-GA	2
-55A	69-35336-2									A-GA	2
-55B	69-35336-3									HA-SA	2
60	69-35329-1										1
-60A	65C33100-1										1
60B	65C33100-3										1
60C	65C33100-4										1
65	BACB10AF6F14H										2
-65A	BACB10BH60F11										2
-65B	BACB10B196-11										2
-65C	BACB10FK6F14HS										2
-65D	KRP141500VT6-16										2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1-												
-65E	KRP141500VT6-16		. .	FOLLOWER-CAM								2
				(V50632)								
				(USED ON ITEM 55B)								
				(ITEM 65F PLUS ONE ITEM 70B OR								
				ITEM 65G IS OPT ITEM 65E)								
-65F	KRP141500VT6-17		. .	FOLLOWER-CAM								2
				(V50632)								
				(USED ON ITEM 55B)								
				(ITEM 65F PLUS ONE ITEM 70B OR								
				ITEM 65G IS OPT ITEM 65E)								
-65G	CL66109		. .	FOLLOWER-CAM								2
				(V21335)								
				(USED ON ITEM 55B)								
				(ITEM 65F PLUS ONE ITEM 70B OR								
				ITEM 65G IS OPT ITEM 65E)								
70	AN960-616		. .	WASHER								2
				(USED ON ITEMS 55, 55A)								
				(PRE SB 737-57-1222)								
-70A	NAS1149C0663R		. .	WASHER								2
				(USED ON ITEMS 55, 55A)								
				(POST SB 737-57-1222)								
-70B	NAS1149C0663R		. .	WASHER								2
				(USED ON ITEM 55B)								
				(ITEM 65F PLUS ONE ITEM 70B OR								
				ITEM 65G IS OPT ITEM 65E)								
71	AN960-616L		. .	WASHER								AR
				(USED ON ITEM 55A)								
				(USE THIS WASHER AS								
				NECESSARY TO LET YOU INSTALL								
				ITEM 80, AND AT THE SAME TIME,								
				TIGHTEN ITEM 75 TO THE								
				SPECIFIED TORQUE)								
-71A	NAS1149C0632R		. .	WASHER								AR
				(USED ON ITEMS 55, 55A)								
-71B	NAS1149C0632R		. .	WASHER								AR
				(USED ON ITEM 55B)								
75	BACN10JD6		. .	NUT-CASTELLATED								2
				(REPLACED BY ITEM 75A)								
				(USED ON ITEMS 55, 55A)								
75A	BACN10JD106		. .	NUT-CASTELLATED								2
				(REPLACES ITEM 75)								
				(USED ON ITEMS 55, 55A)								

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
75B	BACN10JD106CD								. . NUT-CASTELLATED (USED ON ITEM 55B)		2
80	MS24665-283								. . PIN-COTTER		2
85	NAS75-6-020								. . BUSHING (REPLACED BY ITEM 85A) (USED ON ITEMS 55, 55A)		2
-85A	BACB28AK06-062								. . BUSHING (REPLACES ITEM 85) (USED ON ITEMS 55, 55A)		2
-85B	BACB28X6C055								. . BUSHING (USED ON ITEM 55B)		2
90	BACB28X8B25								. . BUSHING (REPLACED BY ITEM 90A) (USED ON ITEMS 55, 55A)		2
-90A	BACB28AT08B025C								. . BUSHING (REPLACES ITEM 90) (USED ON ITEMS 55, 55A)		2
90B	KJB331308V								. . BUSHING (USED ON ITEM 55B)		2
-91	BACB28X6C055								. . BUSHING (USED ON ITEMS 55, 55A) (POST SB 737-57-1222)		2
91A	BACB28X6C055								. . BUSHING (USED ON ITEM 55B)		2
92	BACB28Y6C018								DELETED		
95	BACB10B131								. . BEARING (REPLACED BY ITEM 95A) (USED ON ITEMS 55, 55A)		1
-95A	BACB10ET04								. . BEARING (REPLACES ITEM 95) (USED ON ITEMS 55, 55A)		1
-95C	KRP502-4VT								. . BEARING (V50632) (USED ON ITEM 55B)		1
-95D	AL63858								. . BEARING (V21335) (USED ON ITEM 55B) (OPT ITEM 95C)		2
100	MS16562-10								. . SPRING PIN (USED ON ITEM 105A)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
105	69-77698-1		.	.	BOLT (USED ON ITEM 55B)						1
105A	66-23212-1		.	.	BOLT ASSY (USED ON ITEMS 55, 55A)						1
110	NAS516-1		.	.	.FITTING-GREASE						1
115	66-23212-2		.	.	BOLT						1
-120	65-46430-1		.	.	FLAP ASSY (LH)				A		1
-120A	65-46430-77		.	.	FLAP ASSY (LH)				C, E		1
-120B	65-46430-85		.	.	FLAP ASSY (LH)				E, G		1
-120C	65-46430-89		.	.	FLAP ASSY (LH)				I, K		1
-120D	65-46430-101		.	.	FLAP ASSY (LH)				M, PA		1
-120E	65-46430-117		.	.	FLAP ASSY (LH)				O		1
-120F	65-46430-119		.	.	FLAP ASSY (LH)				Q		1
-120G	65-46430-125		.	.	FLAP ASSY (LH)				S		1
-120H	65-46430-131		.	.	FLAP ASSY (LH)				U		1
-120I	65-46430-135		.	.	FLAP ASSY (LH)				W		1
-120J	65-46430-139		.	.	FLAP ASSY (LH)				Y		1
-120K	65-46430-149		.	.	FLAP ASSY (LH)				U		1
-120L	65-46430-151		.	.	FLAP ASSY (LH)				DA		1
-120M	65-46430-159		.	.	FLAP ASSY (LH)				BA		1
-120N	65-46430-169		.	.	FLAP ASSY (LH)				HA		1
-120P	65-46430-171		.	.	FLAP ASSY (LH)				FA		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-120Q	65-46430-175		.	FLAP	ASSY					JA	1
				(LH)							
-120R	65-46430-179		.	FLAP	ASSY					LA	1
				(LH)							
-120S	65-46430-183		.	FLAP	ASSY					NA	1
				(LH)							
-120T	65-46430-187		.	FLAP	ASSY					RA	1
				(LH)							
-120U	65-46430-191		.	FLAP	ASSY					W	1
				(LH)							
-125	65-46430-2		.	FLAP	ASSY					B	1
				(RH)							
-125A	65-46430-78		.	FLAP	ASSY					D, F	1
				(RH)							
-125B	65-46430-86		.	FLAP	ASSY					F, H	1
				(RH)							
-125C	65-46430-90		.	FLAP	ASSY					J, L	1
				(RH)							
-125D	65-46430-102		.	FLAP	ASSY					N, QA	1
				(RH)							
-125E	65-46430-118		.	FLAP	ASSY					P	1
				(RH)							
-125F	65-46430-120		.	FLAP	ASSY					R	1
				(RH)							
-125G	65-46430-126		.	FLAP	ASSY					T	1
				(RH)							
-125H	65-46430-132		.	FLAP	ASSY					V	1
				(RH)							
-125I	65-46430-136		.	FLAP	ASSY					X	1
				(RH)							
-125J	65-46430-140		.	FLAP	ASSY					Z	1
				(RH)							
-125K	65-46430-150		.	FLAP	ASSY					V	1
				(RH)							
-125L	65-46430-152		.	FLAP	ASSY					EA	1
				(RH)							
-125M	65-46430-160		.	FLAP	ASSY					CA	1
				(RH)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-125N	65-46430-170		. FLAP ASSY (RH)	IA	1
-125P	65-46430-172		. FLAP ASSY (RH)	GA	1
-125Q	65-46430-176		. FLAP ASSY (RH)	KA	1
-125R	65-46430-180		. FLAP ASSY (RH)	MA	1
-125S	65-46430-184		. FLAP ASSY (RH)	OA	1
-125T	65-46430-188		. FLAP ASSY (RH)	SA	1
-125U	65-46430-192		. FLAP ASSY (RH)	X	1
130	65-46430-25		. . PLATE-IDENT ATTACHING PARTS		1
-135	MS20470D3		. . RIVET -----*		4
140	69-38802-3		. . SUPPORT ASSY (PRE SB 737-57-1222) (USED ON ITEMS 120-120L, 125-125L)		2
145	69-38802-4		. . SUPPORT ASSY (PRE SB 737-57-1222) (USED ON ITEMS 120-120L, 125-125L)		2
146	69-38802-9		. . SUPPORT ASSY (POST SB 737-57-1222) (USED ON ITEMS 120M-120U, 125M-125U)		4
147	69-38802-10		. . SUPPORT ASSY (POST SB 737-57-1222) (USED ON ITEMS 120M-120U, 125M-125U) ATTACHING PARTS		4
150	NAS1104-36D		. . BOLT (USED ON ITEMS 120-120E, 125-125E)		4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-150A	NAS1105-36D		. .								AR
-150B	BACB30NR4DK36		. .								4
-150C	BACB30NR5DK36		. .								AR
155	AN960-416		. .								4
-155A	AN960-516		. .								AR
160	AN320-4		. .								4
-160A	AN320-5		. .								AR
-160B	BACN10JD104CD		. .								4
-160C	BACN10JD105CD		. .								AR
165	MS24665-134		. .								4
168	BACB28X6C23		. . .								1
170	69-38802-7		. . .								1
171	65C37022-1		. . .								1
175	MS20426B3		. . .								1
176	BACR15DR3PAC2		. . .								4

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
180	69-38802-5		. . .	SUPPORT							1
				(USED ON ITEM 140)							
185	69-38802-6		. . .	SUPPORT							1
				(USED ON ITEM 145)							
186	69-38802-11		. . .	SUPPORT							1
				(USED ON ITEM 146)							
187	69-38802-12		. . .	SUPPORT							1
				(USED ON ITEM 147)							
190	NAS1104-38D		. .	BOLT							2
				(USED ON ITEMS 120-120E, 125-125E)							
-190A	NAS1105-38D		DELETED								
-190B	BACB30NR4DK38		. .	BOLT							2
				(USED ON ITEMS 120F-120U, 125F-125U)							
-190C	BACB30NR5DK38		DELETED								
195	AN960-416		. .	WASHER							2
				(USED ON ITEMS 120-120E, 120H-120U, 125-125E, 125H-125U)							
-195A	AN960-516		DELETED								
-195B	AN960KD416		. .	WASHER							2
				(USED ON ITEMS 120F, 120G, 120T, 125F, 125G, 125T)							
-195C	AN960KD516		DELETED								
200	AN320-4		. .	NUT							2
				(USED ON ITEMS 120-120E, 125-125E)							
-200A	AN320-5		DELETED								
-200B	BACN10JD104CD		. .	NUT							2
				(USED ON ITEMS 120F, 120G, 120T, 125F, 125G, 125T)							
-200C	BACN10JD105CD		DELETED								
-200D	BACN10JD4CD		. .	NUT							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-200E	BACN10JD5CD		DELETED								

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
205	MS24665-134										2
-205A	MS24665-153										2
-205B	MS24665-283										
210	NAS517-4-6										2
-215	69-37815-1										1
220	69-37815-3										1
225	69-37815-5										1
230	NAS1067A4										
230A	BACN10JP4DCD										1
232	BACR15BA3D										2
-235	69-37815-1										1
-235A	69-37815-7										1
-235B	69-37815-8										1
240	69-37815-3										1
-240A	69-37815-9										1
-240B	69-37815-10										1
245	69-37815-5										1
250	BACN10JP4DCD										1
255	BACR15BA3D										2
280	69-35349-3										1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-			ATTACHING PARTS								
281	69-35349-5		. . BRACKET (WBL 84.96 ONLY) (USED ON ITEMS 120G, 120H, 120J-120T, 125G, 125H, 125J-125T)								2
282	69-35349-7		. . BRACKET (WBL 84.96 ONLY) (USED ON ITEMS 120G, 120H, 120J-120T, 125G, 125H, 125J-125T)								1
285	BACB30DM4-8H		. . BOLT (USED ON ITEMS 120-120C, 125-125C)								1
-285A	66-26119-1		. . BOLT (OPT ITEM 285)							M-P	1
-285B	66-26119-1		. . BOLT							Q, R, Y, Z	1
-285C	66-26119-1		. . BOLT (OPT ITEM 285D) (USED ON ITEMS 120C, 120I, 125C, 125I)								1
-285D	BACB30DM4-8H		. . BOLT (OPT ITEM 285C) (USED ON ITEMS 120C, 125C)								1
-285E	BACB30JU4K8H		. . BOLT (LIMITED USAGE) (OPT ITEM 285F) (USED ON ITEMS 120G, 120H, 125G, 125H)								1
-285F	66-26119-1		. . BOLT (LIMITED USAGE) (OPT ITEM 285E) (USED ON ITEMS 120G, 120H, 125G, 125H)								1
-285G	NAS6704H8		. . BOLT (LIMITED USAGE) (USED ON ITEMS 120G, 120H, 125G, 125H)								1
-285H	BACB30LE4HK10		. . BOLT (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)								1
290	BACS40R07B14F		. . SHIM								AR

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
295	BACR10V4								. . RETAINER		1
300	BACN10CP4H								. . NUT (USED ON ITEMS 120-120E, 125-125E)		1
-300A	BACN10HC4								. . NUT (USED ON ITEMS 120F-120U, 125F-125U)		1
305	BACW10BN4C								. . WASHER (USED ON ITEMS 120-120E, 125-125E)		1
-305A	BACW10BP4CD								. . WASHER (LIMITED USAGE) (USED ON ITEMS 120F, 120I, 120J, 125F, 125I, 125J)		1
-305B	BACW10DS4S								. . WASHER (LIMITED USAGE) (USED ON ITEMS 120F, 120I, 120J, 125F, 125I, 125J)		1
-306	BACW10DS4S								. . WASHER (USED ON ITEMS 120G, 120H, 125G, 125H)		1
310	BACB30DM5-9H								. . BOLT (OPT ITEM 310A) (USED ON ITEMS 120-120C, 125-125C)		1
-310A	66-26119-2								. . BOLT (OPT ITEM 310)	M-P	1
-310B	66-26119-2								. . BOLT	Q, R, Y, Z	1
-310C	66-26119-2								. . BOLT (USED ON ITEMS 120C, 120I, 125C, 125I)		1
-310D	BACB30DM5-9H								. . BOLT (OPT ITEM 310C) (USED ON ITEMS 120C, 125C)		1
-310E	BACB30UU5K11N								. . BOLT (LIMITED USAGE) (OPT ITEM 310F) (USED ON ITEMS 120G, 120H, 125G, 125H)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY	
			1	2	3	4	5	6	7			
1-												
-310F	66-26119-4											1
-310G	66-26119-2											1
-310H	NAS6705H11											1
-310J	BACB30LE4HK13											1
315	BACS40R14D14F											1
320	BACR10V5											1
325	BACW10BN5C											1
-325A	BACW10BP5CD											1
-325B	BACW10DS5S											1
-326	BACW10DS5S											1
330	BACN10CP5H											1
-330A	BACN10HC5											1
335	69-35349-1											2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-			ATTACHING PARTS								
340	BACB30DM4-8H		. . BOLT (OPT ITEM 340A) (USED ON ITEMS 120-120C, 125-125C)								2
-340A	66-26119-1		. . BOLT (OPT ITEM 340)							M-P	2
-340B	66-26119-1		. . BOLT							Q, R, Y, Z	2
-340C	66-26119-1		. . BOLT (OPT ITEM 340D) (USED ON ITEMS 120C, 120I, 125C, 125I)								2
-340D	BACB30DM4-8H		. . BOLT (OPT ITEM 340C) (USED ON ITEMS 120C, 125C)								2
-340E	BACB30UU4K8H		. . BOLT (LIMITED USAGE) (OPT ITEM 340F) (USED ON ITEMS 120G,120H,125G,125H)								2
-340F	66-26119-1		. . BOLT (LIMITED USAGE) (OPT ITEM 340E) (USED ON ITEMS 120G,120H,125G,125H)								2
-340G	NAS6704H8		. . BOLT (LIMITED USAGE) (USED ON ITEMS 120G, 120H, 125G, 125H)								2
345	BACS40R07B14F		. . SHIM								AR
350	BACR10V4		. . RETAINER								2
355	BACW10BN4C		. . WASHER (USED ON ITEMS 120-120E, 125-125E)								2
-355A	BACW10BP4CD		. . WASHER (LIMITED USAGE) (USED ON ITEMS 120F, 120I, 120J, 125F, 125I, 125J)								2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-355B	BACW10DS4		. .	WASHER							2
				(LIMITED USAGE)							
				(USED ON ITEMS 120F, 120I, 120J, 125F, 125I, 125J)							
-356	BACW10DS4S		. .	WASHER							2
				(USED ON ITEMS 120G, 120H, 125G, 125H)							
360	BACN10CP4H		. .	NUT							2
				(USED ON ITEMS 120-120E, 125-125E)							
-360A	BACN10HC4		. .	NUT							2
				(USED ON ITEMS 120F-120U, 125F-125U)							
365	BACB30DM6-10H		. .	BOLT							2
				(OPT ITEM 365A)							
				(USED ON ITEMS 120-120C, 125-125C)							
-365A	66-26119-3		. .	BOLT					M-P		2
				(OPT ITEM 365)							
-365B	66-26119-3		. .	BOLT					Q, R, Y, Z		2
-365C	66-26119-3		. .	BOLT							2
				(OPT ITEM 365D)							
				(USED ON ITEMS 120C, 120I, 125C, 125I)							
-365D	BACB30DM6-10H		. .	BOLT							2
				(OPT ITEM 365C)							
				(USED ON ITEMS 120C, 125C)							
-365E	BACB30UU6K11H		. .	BOLT							2
				(LIMITED USAGE)							
				(OPT ITEM 365F)							
				(USED ON ITEMS 120G, 120H, 125G, 125H)							
-365F	66-26119-5		. .	BOLT							2
				(LIMITED USAGE)							
				(OPT ITEM 365E)							
				(USED ON ITEMS 120G, 120H, 125G, 125H)							
-365G	66-26119-3		. .	BOLT							2
				(LIMITED USAGE)							
				(USED ON ITEMS 120G, 120H, 125G, 125H)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-365H	NAS6706H11										2
370	BACS40R14D14F										AR
375	BACR10V6										2
380	BACW10BN6C										2
-380A	BACW10BP4CD										2
-380B	BACW10DS4S										2
-380C	BACW10DS4S										2
385	BACN10CP6H										2
-385A	BACN10HC4										2
430	BACB30LU3-3										4
-430A	BACB30NN3K4										6
-430B	BACB30NN3K4										3
-430C	BACB30NN4K6										2

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-430D	BACB30VF4K6								. . BOLT (OPT ITEM 430C) (USED ON ITEMS 120F-120U, 125F-125U)		2
-430E	BACB30NR4DK36								. . BOLT (USED ON ITEMS 120F-120U, 125F-125U)		4
-430F	BACB30NRDK38								. . BOLT (USED ON ITEMS 120F-120U, 125F-125U)		2
435	AN960PD10L								. . WASHER (USED ON ITEMS 120-120E, 125-125E)		4
-435A	AN960KD10L								. . WASHER (USED ON ITEMS 120F-120U, 125F-125U)		6
440	NAS679A3W								. . NUT (USED ON ITEMS 120-120E, 125-125E)		4
-440A	BACN10YR3CD								. . NUT (USED ON ITEMS 120F-120U, 125F-125U)		6
-445	65-68278-30								. . SEAL INSTL-FOREFLAP TO LWR PANEL (USED ON ITEM 120)		1
-445A	65-68278-59								. . SEAL INSTL-FOREFLAP TO LWR PANEL (USED ON ITEMS 120A-120J)		1
-450	65-68278-31								. . SEAL INSTL-FOREFLAP TO LWR PANEL (USED ON ITEM 125)		1
-450A	65-68278-60								. . SEAL INSTL-FOREFLAP TO LWR PANEL (USED ON ITEMS 125A-125J)		1
455	65-68278-24								. . . BLOCK ASSY-END (USED ON ITEM 445, 445A) (OPT ITEM 445A)		1
-455A	65-68278-68								. . . BLOCK ASSY-END (USED ON ITEM 445, 445A) (OPT ITEM 445)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-460	65-68278-23		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 450, 450A)						
					(OPT ITEM 460A)						
-460A	65-68278-67		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 450, 450A)						
					(OPT ITEM 460)						
465	65-68278-57		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 445A)						
					(OPT ITEM 465A)						
-465A	65-68278-65		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 445A)						
					(OPT ITEM 465)						
-470	65-68278-58		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 450A)						
-470A	65-68278-66		. . .	BLOCK	ASSY-END						1
					(USED ON ITEM 450A)						
475	65-68278-44		STRIP-RUB							1
480	65-46430-505		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 120D)						
-480A	65C32620-1		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 120F-120J)						
-480B	65-46430-113		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 120E)						
-480C	65-46430-52		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEMS 120-120B)						
-480D	65-46430-91		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 120C)						
-480E	65C32620-11		. .	SKIN	ASSY						1
					(USED ON ITEMS 120G, 120H, 120K-120U)						
-485	65-46430-506		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 125D)						
-485A	65C32620-2		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEMS 125F-125J)						
-485B	65-46430-114		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEM 125E)						
-485C	65-46430-53		. .	SKIN	ASSY-LEADING EDGE						1
					(USED ON ITEMS 125-125B)						

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-485D	65-46430-92		.	.	SKIN ASSY-LEADING EDGE (USED ON ITEM 125C)						1
-485E	65C32620-12		.	.	SKIN ASSY-LEADING EDGE (USED ON ITEMS 125G, 125H, 125K-125U)						1
490	65-46430-503		.	.	HONEYCOMB ASSY (USED ON ITEM 120D)						1
-490A	65-46430-123		.	.	HONEYCOMB ASSY (LIMITED USAGE) (USED ON ITEMS 120F, 120J)						1
-490B	65-46430-523		.	.	HONEYCOMB ASSY (LIMITED USAGE) (USED ON ITEMS 120F, 120J)						1
-490C	65-46430-133		.	.	HONEYCOMB ASSY (USED ON ITEMS 120G-120I, 120K)						1
-490D	65-46430-115		.	.	HONEYCOMB ASSY (USED ON ITEM 120E)						1
-490E	65-46430-54		.	.	HONEYCOMB ASSY (USED ON ITEMS 120-120B)						1
-490F	65-46430-93		.	.	HONEYCOMB ASSY (USED ON ITEM 120C)						1
-490G	65-46430-157		.	.	HONEYCOMB ASSY (USED ON ITEMS 120L-120U)						1
-495	65-46430-504		.	.	HONEYCOMB ASSY (USED ON ITEM 125D)						1
-495A	65-46430-124		.	.	HONEYCOMB ASSY (LIMITED USAGE) (USED ON ITEMS 125F, 120J)						1
-495B	65-46430-522		.	.	HONEYCOMB ASSY (LIMITED USAGE) (USED ON ITEMS 125F, 120J)						1
-495C	65-46430-134		.	.	HONEYCOMB ASSY (USED ON ITEMS 125G-125I, 125K)						1
-495D	65-46430-116		.	.	HONEYCOMB ASSY (USED ON ITEM 125E)						1
-495E	65-46430-55		.	.	HONEYCOMB ASSY (USED ON ITEMS 125-125B)						1
-495F	65-46430-94		.	.	HONEYCOMB ASSY (USED ON ITEM 125C)						1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-495G	65-46430-158								. . HONEYCOMB ASSY (USED ON ITEMS 125L-125U)		1
500	65-46430-15								. . . STRIP-RUB		1
505	65-46430-27								. . SEAL (USED ON ITEMS 120-120P, 125-125P)		1
510	65-46430-26								. . SEAL (USED ON ITEMS 120-120J, 125-125J)		1
511	65-46430-153								. . SEAL (USED ON ITEMS 120I, 120K-120U, 125I, 125K-125U)		1
515	BACS12N10-9								. . SCREW		12
520	65-46430-38								. . RETAINER		2
-525	69-73105-1								. . BRACKET ASSY-ACTUATION (WBL 165.0) (USED ON ITEMS 120D-120F, 120J)		1
-525A	69-73105-1								. . BRACKET ASSY-ACTUATION (LIMITED USAGE) (WBL 165.0) (USED ON ITEMS 120G, 120H)		1
-525B	69-73105-7								. . BRACKET ASSY-ACTUATION (LIMITED USAGE) (WBL 165.0) (USED ON ITEMS 120G, 120H)		1
-525C	69-73105-7								. . BRACKET ASSY-ACTUATION (WBL 165.0) (USED ON ITEMS 120K, 120M-120T)		1
-530	69-73105-2								. . BRACKET ASSY-ACTUATION (WBL 165.0) (USED ON ITEMS 125D-125F, 125J)		1
-530A	69-73105-2								. . BRACKET ASSY-ACTUATION (LIMITED USAGE) (WBL 165.0) (USED ON ITEMS 125G, 125H)		1
-530B	69-73105-8								. . BRACKET ASSY-ACTUATION (LIMITED USAGE) (WBL 165.0) (USED ON ITEMS 125G, 125H)		1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-530C	69-73105-8		. .								1
535	69-73105-3		. . .								1
-540	69-73105-4		. . .								1
545	BACB28W4C026		. . .								1
550	BACB28W6B026		. . .								1
555	BACS40R100F100F		. .								AR
-560	BACB30LE4H12										
560A	BACB30LE4H10		. .								1
565	BACW10BP41CD		. .								2
570	BACN10HC4		. .								2
575	BACR10V5		. .								2
580	BACB30LE4H14										
580A	BACB30LE4H13		. .								1
-585	BACW10BP4CD		. .						Q, R, Y, Z		3
-588	BACW10BP5CD		. .						Q, R, Y, Z		1
-590	BACW10BP6CD		. .						Q, R, Y, Z		2
-595	BACW10DS4S		. .						Q-Z		3
-598	BACW10DS5S		. .						Q-Z		1
-600	BACW10DS6S		. .						Q-Z		2
-605	BACS40R002B014F		. .						Q, R, Y, Z		3
-607	BACS40R014D014F		. .						Q, R, Y, Z		3
-608	69-76975-1		. .								1

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-608A	69-76975-2		. .	RIB-WBL	117.10						1
											(USED ON ITEMS 125F-125K)
-608B	69-76975-3		. .	RIB-WBL	117.10						1
											(USED ON ITEMS 120L-120U)
-608C	69-76975-4		. .	RIB-WBL	117.10						1
											(USED ON ITEMS 125L-125U)
-609	69-76976-3		. .	RIB-WBL	84.96						1
											(USED ON ITEMS 120F-120K)
-609A	69-76976-4		. .	RIB-WBL	84.96						1
											(USED ON ITEMS 125F-125K)
-609B	69-76976-9		. .	RIB-WBL	84.96						1
											(USED ON ITEMS 120L-120U)
-609C	69-76976-10		. .	RIB-WBL	84.96						1
											(USED ON ITEMS 125L-125U)
-609H	69-76976-1		. .	RIB-WBL	159.50						1
											(USED ON ITEMS 120F-120K)
-609I	69-76976-2		. .	RIB-WBL	159.50						1
											(USED ON ITEMS 125F-125K)
-609J	69-76976-7		. .	RIB-WBL	159.50						1
											(USED ON ITEMS 120L-120U)
-609K	69-76976-8		. .	RIB-WBL	159.50						1
											(USED ON ITEMS 125L-125U)
-609Q	69-76976-5		. .	RIB-WBL	165.00						1
											(USED ON ITEMS 120F-120K)
-609R	69-76976-6		. .	RIB-WBL	165.00						1
											(USED ON ITEMS 125F-125K)
-609S	69-76976-11		. .	RIB-WBL	165.00						1
											(USED ON ITEMS 120L-120U)
-609T	69-76976-12		. .	RIB-WBL	165.00						1
											(USED ON ITEMS 125L-125U)
-610	69-76977-1		. .	RIB							5
											(ITEM 65F PLUS ONE ITEM 70B OR ITEM 65G IS OPT ITEM 65E) (USED ON ITEMS 120F-120K)
-610A	69-76977-2		. .	RIB							5
											(USED ON ITEMS 125F-125K)
-610B	69-76977-3		. .	RIB							5
											(USED ON ITEMS 120L, 120M, 120N, 120Q)

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			1	2	3	4	5	6	7		
1-											
-610C	69-76977-4		. .	RIB							5
				(USED ON ITEMS 125L, 125M, 125N, 125Q)							
-611	69-78112-1		. .	END BLOCK ASSY							1
				(USED ON ITEMS 125G, 125H, 125L-125Q, 125S)							
-611A	69-78112-2		. .	END BLOCK ASSY							1
				(USED ON ITEMS 120G, 120H, 120L-120Q, 120S)							
-611B	69-78112-3		. .	END BLOCK ASSY							1
				(USED ON ITEMS 120G, 120H, 120L-120Q, 120S)							
-611C	69-78112-4		. .	END BLOCK ASSY							1
				(USED ON ITEMS 125G, 125H, 125L-125Q, 125S)							
-611D	69-78026-1		. .	END BLOCK ASSY							1
				(USED ON ITEMS 125G, 125H, 125L, 125M, 125R, 125T, 125U)							
-611E	69-78026-2		. .	END BLOCK ASSY							1
				(USED ON ITEMS 120G, 120H, 120L, 120M, 120R, 120T, 120U)							
-611F	69-78026-3		. .	END BLOCK ASSY							1
				(USED ON ITEMS 120G, 120H, 120L, 120M, 120R, 120T, 120U)							
-611G	69-78026-4		. .	END BLOCK ASSY							1
				(USED ON ITEMS 125G, 125H, 125L, 125M, 125R, 125T, 125U)							
-612	BACB30LE4HK10		. .	BOLT							1
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-613	BACB30LE4HK13		. .	BOLT							1
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-614	BACB30NN3K4		. .	BOLT							3
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-615	BACB30NN4K6		. .	BOLT							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-616	BACB30NR4DK36		. .	BOLT							4
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-617	BACB30NR4DK38		. .	BOLT							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-618	BACB30VF4K6		. .	BOLT							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-619	BACN10HC4		. .	NUT-CYLINDRICAL							5
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-620	BACN10HC5		. .	NUT-CYLINDRICAL							1
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-621	BACN10HC6		. .	NUT-CYLINDRICAL							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-622	BACN10JD4CD		. .	NUT							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-623	BACN10JN3CM		. .	NUT PLATE							12
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-624	BACN10YR3CD		. .	NUT							3
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-625	BACR10V4		. .	RETAINER							3
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-626	BACR10V5		. .	RETAINER							3
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-627	BACR10V6		. .	RETAINER							2
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							
-628	BACS12N10-9		. .	SCREW							12
				(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)							

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE							USAGE CODE	UNITS PER ASSY
			1	2	3	4	5	6	7		
1-											
-629	BACS40R007B014F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-630	BACS40R008B014F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-631	BACS40R008C14F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-632	BACS40R009B024F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-633	BACS40R009C012F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-634	BACS40R009C013F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-635	BACS40R010C021F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-636	BACS40R010F010F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-637	BACS40R014D014F										
			. .	SHIM							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-638	BACW10BP41CD										
			. .	WASHER							2
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-639	BACW10DS4S										
			. .	WASHER							3
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-640	BACW10DS5S										
			. .	WASHER							1
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		
-641	BACW10DS6S										
			. .	WASHER							2
									(USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1-					
-642	MS20995NC20		. . LOCK WIRE (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		1
-643	MS24665-153		. . COTTER PIN (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		2
-644	NAS6704H8		. . BOLT (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		3
-645	NAS6705H11		. . BOLT (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		1
-646	NAS6706H11		. . BOLT (USED ON ITEMS 120G, 120H, 120K-120U, 125G, 125H, 125K-125U)		2
			-----*-----		

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