

# COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST

# CFM56 ENGINE THROTTLE FUEL CONTROL BOX ASSEMBLY

## PART NUMBER 315A1040–10, –11, –3, –4, –5, –6, –7, –8, –9

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PUBLISHED BY BOEING COMMERCIAL AIRPLANES GROUP, SEATTLE, WASHINGTON, USA A DIVISION OF THE BOEING COMPANY PAGE DATE: Jul 01/2009



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## **COMPONENT MAINTENANCE MANUAL**

Revision No. 26 Jul 01/2009

To: All holders of CFM56 ENGINE THROTTLE FUEL CONTROL BOX ASSEMBLY 76-11-13.

Attached is the current revision to this COMPONENT MAINTENANCE MANUAL

The COMPONENT MAINTENANCE MANUAL is furnished either as a printed manual, on microfilm, or digital products, or any combination of the three. This revision replaces all previous microfilm cartridges or digital products. All microfilm and digital products are reissued with all obsolete data deleted and all updated pages added.

For printed manuals, changes are indicated on the List of Effective Pages (LEP). The pages which are revised will be identified on the LEP by an R (Revised), A (Added), O (Overflow, i.e. changes to the document structure and/or page layout), or D (Deleted). Each page in the LEP is identified by Chapter-Section-Subject number, page number and page date.

Pages replaced or made obsolete by this revision should be removed and destroyed.

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## COMPONENT MAINTENANCE MANUAL

Location of Change

Description of Change NO HIGHLIGHTS





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#### TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR 33577	JUN 5/84
		PRR 33695	DEC 5/84
		PRR 34487	DEC 5/88
	76-2		MAR 5/90
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76-1021			DEC 5/91
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315A1040-76-01			DEC 5/91
	76-4		NOV 1/99

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All revisions to this manual will be accompanied by transmittal sheet bearing the revision number. Enter the revision number in numerical order, together with the revision date, the date filed and the initials of the person filing.

Revision		Filed		Revision		Filed	
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Revision		Filed		Revision		Filed	
Number	Date	Date	Initials	Number	Date	Date	Initials

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When the temporary revision is incorporated or cancelled, and the pages are removed, enter the date the pages are removed and the initials of the person who removed the temporary revision.

Temporary	Revision	Ins	serted	Rei	moved	Tempora	ary Revision	Inserted		Removed	
Number	Date	Date	Initials	Date	Initials	Date	Initials	Number	Date	Date	Initials

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#### INTRODUCTION

#### 1. General

- A. The instructions in this manual supply the data necessary to do the maintenance functions together with the test, fault isolation, repair, and replacement of the defective parts.
- B. This manual is divided into different parts:
  - (1) Title Page
  - (2) Transmittal Letter
  - (3) Highlights
  - (4) List of Effective Pages
  - (5) Table of Contents
  - (6) Temporary Revision & Service Bulletin Record
  - (7) Record of Revisions
  - (8) Record of Temporary Revisions
  - (9) Introduction
  - (10) Procedures & IPL Sections
- C. Components that can be repaired have a different repair number for each specified repair. To find the repair number location of a component, look in the Repair-General procedure at the beginning of the REPAIR section. The Repair-General procedure also has an explanation of the True Position Dimension symbols used.
- D. All dimensions, measures, quantities and weights included are in English units. When metric equivalents are given they will be in the parentheses that follow the English units.
- E. The introduction to the Illustrated Parts List (IPL) shows how the IPL data is used.
- F. Design changes, optional parts, configuration differences and Service Bulletin modifications may cause different part numbers. These part numbers are identified in the IPL with an alphabetical letter which is added to the end of the basic item number. This new item number is referred to as an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless shown differently.
- G. The tool reference numbers found in the individual procedures and in the Special Tools, Fixtures, and Equipment section are used to identify if a tool is a standard tool (STD-XXXX), a commercial tool (COM-XXXX), or a Special Tool (SPL-XXXX). This reference number is also used to distinguish between tools with similar names in the same procedure. These reference numbers are for use in the documentation only. They are not to be used for ordering tools.







#### CFM - 56 ENGINE THROTTLE FUEL - CONTROL BOX ASSEMBLY - DESCRIPTION AND OPERATION

#### 1. Description

- A. The Engine Throttle Fuel Control Box Assembly uses a bearing-mounted gear to control the fuel flow to the CFM-56 engine.
- B. The gear is located inside the box and meshes with a gear rack and a "synchro" or transducer.

#### 2. Operation

- A. The throttle push-pull cable controls the linear movement of the rack inside the control box assembly.
- B. The rack then turns the control box gear and then this turns the transmitter assembly called the "power lever angle synchro."
- C. A fork extends out from the control box gear to position the power lever.
- D. When the power lever is in the correct position, then the transmitter assembly sends the angular position back to the auto-throttle servo computer.

#### 3. Leading Particulars (approximate)

- A. Height 4 inches
- B. Width 11 inches
- C. Thickness 2 inches







315A1040-3 Thru -11 CFM-56 Engine Throttle Fuel Control Box Assembly Figure 1

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**TESTING AND FAULT ISOLATION** 

## (NOT APPLICABLE)





#### DISASSEMBLY

#### 1. General

- A. This procedure contains the data necessary to disassemble the engine throttle fuel control box assembly.
- B. Disassemble the control box assembly only when necessary to do the specified procedures that follow:
  - (1) To complete any type of testing and fault isolation.
  - (2) To find out if the parts are in a serviceable condition.
  - (3) To make any necessary repairs to the control box assembly.
  - (4) To put the control box assembly back into a serviceable condition.
- C. Refer to IPL Figure 1 for the item numbers.

#### 2. Disassembly

A. Consumable Materials

NOTE: Equivalent substitutes may be used.

Reference	Description	Specification
G01048	Lockwire - Corrosion Resistant Steel (0.032	In. Dia.) NASM20995~

B. Parts Replacement

**NOTE**: These parts that follow are recommended for replacement. Replacement of other parts can be by in-service experience.

- (1) Lockwire MS20995C32lockwire, G01048
- C. Procedure
  - (1) Remove nut (5) and washer (10).

**CAUTION:** CAREFULLY DISENGAGE GEAR TEETH TO PREVENT DAMAGE.

- (2) Separate the gear assy (15A) including the attached support fitting (43 or 45) from the housing (130).
  - (a) Do not disassemble the gear assy (15A) or the support fitting (43 or 45) and do not remove the dowel pins (47, 50) unless repair or replacement is specified.
  - (b) If necessary to disassemble the gear assembly (15A), remove the sealant around the nut(5) by touching the sealant with a soldering iron.
- (3) Remove lockwire from the nuts (60, 75, 85).
- (4) Do the steps that follow to disassemble the housing (130).
  - (a) Remove nuts (85) and washers (90) and pull eccentric pins (95) out of housing (130).

**NOTE**: The bearings (100) will drop out of the slot in the housing as the pins are removed.

- (b) Remove the rack (55) from the housing (130).
- (c) Remove the coupling nut (60).
- (d) Remove the sleeve (65) and the shield (70).
- (e) Loosen the nut (75) and remove the adapter (80).





(f) Remove the nuts (105A), the washers (110), and the bolts (115).

(5) Separate the clamps (120) and remove the synchro (125A) from the housing (130).





## CLEANING

#### 1. General

- A. This procedure has the data necessary to clean the engine throttle fuel control box assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) subjects identified in this procedure.
- C. Refer to IPL Figure 1 for item numbers.

#### 2. Cleaning

A. Consumable Materials

**NOTE**: Equivalent substitutes may be used.

	Reference	Description	Specification
	G50316	Cloth - Clean, Dry, Lint-free, White, Cotton	
В.	References		
	Reference	Title	
	SOPM 20-30-03	GENERAL CLEANING PROCEDURES	

- C. Procedure
  - (1) Clean all parts but bearings (30, 100, IPL Figure 1) by standard industry practices and information contained in SOPM 20-30-03.
  - (2) Clean carbon graphite bearings (100) with a clean cotton cloth, G50316 only. Do not use solvent or lubricant.
  - (3) Clean bearings (30) by the vendor's instructions.





#### **CHECK**

#### 1. General

- A. This procedure has the data necessary to find defects in the material of the specified parts.
- B. Refer to FITS AND CLEARANCES for the design dimension and wear limits.
- C. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in this procedure.
- D. Refer to IPL Figure 1 for item numbers.

#### 2. Check

A. References

Reference	Title
SOPM 20-20-01	MAGNETIC PARTICLE INSPECTION
SOPM 20-20-02	PENETRANT METHODS OF INSPECTION

- B. Procedure
  - (1) Use standard industry practices to examine all of the parts for defects or damage.
  - (2) Examine carbon graphite bearings (100, IPL Figure 1) for external damage, too much radial play, concentricity, and signs of cracks or breakdown of the carbon graphite.

**NOTE**: Replace these bearings if you find more than just small defects.

**CAUTION:** TO PREVENT DAMAGE FROM MAGNETIZATION, BE SURE TO REMOVE BEARING (30) BEFORE THE MAGNETIC PARTICLE INSPECTIONS.

- (3) Do a magnetic particle inspection (SOPM 20-20-01) Clamp (25A), gear (40A thru 40F), fitting (43 or 46), rack (55), clamp (120), housing (130).
- (4) Do a penetrant inspection (SOPM 20-20-02) Shield (70), pin (95).
- (5) Do an inspection of the teeth on the gear (40A thru 40F) and the rack (55) for too much wear.
  - **NOTE**: Replace parts if you find wear or galling that is more necessary for these parts to work correctly in service.





#### **REPAIR**

#### 1. Content

A. Repair, replacement, and refinish procedures are included in separate REPAIR sections as follows:

Table 601:					
P/N	NAME	REPAIR			
315A1045	GEAR	1-1, 1-2			
315A1019	FITTING	2-1			
315T1029	PIN	3-1			
	MISCELLANEOUS PARTS REFINISH	4-1			

#### 2. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in applicable repair procedures are shown in REPAIR-GENERAL, Figure 601.





— STRAIGHTNESS	Ø	DIAMETER				
	sØ	SPHERICAL DIAMETER				
$\perp$ perpendicularity (or squareness)	R	RADIUS				
// PARALLELISM	SR	SPHERICAL RADIUS				
O ROUNDNESS	()	REFERENCE				
<ul> <li>✓ CYLINDRICITY</li> <li>✓ PROFILE OF A LINE</li> <li>△ PROFILE OF A SURFACE</li> <li>Ø CONCENTRICITY</li> <li>≡ SYMMETRY</li> <li>∠ ANGULARITY</li> <li>✓ RUNOUT</li> <li>✓ TOTAL RUNOUT</li> <li>└ COUNTERBORE OR SPOTFACE</li> <li>✓ COUNTERSINK</li> <li>↔ THEORETICAL EXACT POSITION</li> </ul>	BASIC (BSC) OR DIM -A- (1) (3) (P)	A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE. FROM THIS FEATURE PERMIS- SIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES. DATUM MAXIMUM MATERIAL CONDITION (MMC) LEAST MATERIAL CONDITION (LMC) REGARDLESS OF FEATURE SIZE (RFS) PROJECTED TOLERANCE ZONE				
OF A FEATURE (TRUE POSITION)	FIM	FULL INDICATOR MOVEMENT				
	EXAMPLE	<u>s</u>				
- 0.002 STRAIGHT WITHIN 0.002	ØØ	0.0005 C CONCENTRIC TO DATUM C				
U 0.002 B PERPENDICULAR TO DATUM B		WITHIN 0.0005 DIAMETER				
	=	= 0.010 A SYMMETRICAL WITH DATUM A WITHIN 0.010				
WITHIN 0.002	Z	0.005 A ANGULAR TOLERANCE 0.005				
O.002 ROUND WITHIN 0.002		WITH DATUM A				
0.010 CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRI CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER	.c ⊕Ø0	D.002 ③ B LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE				
○ 0.006 A EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM A	⊥Ø0 0.510	D.010 (m) A AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010 INCH DIAMETER, PERPENDICULAR TO DATUM A, AND EXTENDING 0.510 INCH ABOVE DATUM A, MAXIMUM MATERIAL CONDITION				
○ 0.020 A SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.020 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFI	) LE	2.000 THEORETICALLY EXACT OR DIMENSION IS 2.000 2.000 BSC				
True Position Dimensioning Symbols						

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Figure 601



## **COMPONENT MAINTENANCE MANUAL**

#### **GEAR ASSEMBLY - REPAIR 1-1**

#### 315A1045-6, -8, -10, -15, -16, -17

#### 1. General

- A. This procedure has the data necessary to repair the gear assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in the repair.
- C. Refer to IPL Figure 1 for the item numbers.

#### 2. Support Fitting Replacement

A. References

Reference	Title
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT

- B. Procedure
  - **NOTE**: Support fitting (43, 46) is not part of the gear assy (15A), but the fitting is usually removed with the gear assy. It is included here for ease of repair.
  - (1) Do the steps that follow to replace the support fitting (43, 46).
    - (a) Press out the initial support fitting.
    - (b) Refer to REPAIR 2-1 for the repair procedure of the support fitting.
    - (c) Use the shrink-fit method (SOPM 20-50-03) to install the new support fitting into the bearing (30).
  - (2) Refer to REPAIR 1-2 if the replacement of the bearing (30) is necessary.





#### **GEAR ASSEMBLY - REPAIR 1-2**

#### 315A1045-6, -8, -10, -15, -16, -17

#### 1. General

- A. This procedure has the data necessary to repair and refinish the gear assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in the repair.
- C. Refer to IPL Figure 1 for the item numbers.

#### 2. Repair Procedures

A. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-50-03	BEARING AND BUSHING REPLACEMENT

- B. Bearing Replacement
  - (1) Drill out the rivets (20A) and remove the clamp ring (25A).
  - (2) Remove the bearing (30) (SOPM 20-50-03).
  - (3) Press the new bearing into the gear (40A thru 40F) (SOPM 20-50-03).
  - (4) Align the clamp ring (25A) with rivet holes on the gear (40A thru 40F) and make sure that the large cutouts on the gear are not blocked. Attach the rivets (20A).
  - (5) Refer to REPAIR 1-1 to install the support fitting (43, 46).
- C. Refinish
  - **NOTE**: For stripping of protective finishes, refer to SOPM 20-30-02. For decoding table for Boeing finish codes, refer to SOPM 20-41-01.
  - (1) Gears (40A thru 40F) Passivate (F-17.25) all over. Material: 17-4PH CRES
  - (2) Clamp Ring (25A) Passivate (F-17.25) all over. Material: 15-5PH CRES, 150-170 ksi







#### **SUPPORT FITTING ASSEMBLY - REPAIR 2-1**

#### 315A1019-3, -9

#### 1. General

- A. This procedure has the data necessary to repair the support fitting assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in the repair.
- C. Refer to IPL Figure 1 for item numbers.

#### 2. Pin Replacement (REPAIR 2-1, Figure 601)

A. Consumable Materials

**NOTE**: Equivalent substitutes may be used.

Reference	Description	Specification
A00081	Adhesive - Silicone Rubber - RTV 106	BAC5010, Type 74
A00160	Sealant - Firewall - Hydraulic Fluid Resistant	BMS5-63
A00900	Sealant - Silicone, RTV - Dow Corning 93-006-1	

B. References

Reference	Title
SOPM 20-50-04	INSTALLATION OF PERMANENT PINS AND PLUGS IN DRILL PASSAGES
SOPM 20-50-19	GENERAL SEALING
SOPM 20-60-04	MISCELLANEOUS MATERIALS

- C. Procedure
  - **NOTE**: For miscellaneous materials, refer to SOPM 20-60-04. For sealants use BMS 5-63 sealant, A00160, Dow Corning 93-006-1 sealant, A00900 or RTV 106 adhesive, A00081
  - (1) Remove the old pins (47, 50) (SOPM 20-50-04).
  - (2) Use the press-fit or shrink-fit method (SOPM 20-50-04) to install the new pins (47, 50) up to the specified depth.
  - (3) Seal (SOPM 20-50-19) the pin holes with sealant.







ITEM NUMBERS REFER TO IPL FIG. 1 ALL DIMENSIONS ARE IN INCHES

315A1019-3,-9 Support Fitting Parts Replacement Figure 601





#### PIN - REPAIR 3-1

#### 315T1029-1

#### 1. General

- A. This procedure has the data necessary to repair and refinish the pin.
- B. Refer to REPAIR-GENERAL, Figure 601 for the True Position Dimensioning Symbols used in the repair.

#### 2. Plating Repair

A. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES

- B. Procedure
  - **NOTE**: For stripping of protective finishes, refer to SOPM 20-30-02. For decoding table for Boeing finish codes, refer to SOPM 20-41-01.
  - (1) Repair is only replacement of the original finish. Refer to REPAIR 3-1, Figure 601 for the refinish instructions.







REFINISH CADMIUM PLATE (F-15.02) ALL OVER UNLESS SHOWN BY

1>NO PLATING ON THESE SURFACES

REPAIR (SAME AS REFINISH) 63 ALL MACHINED SURFACES MATERIAL: 304 CRES ALL DIMENSIONS ARE IN INCHES

315A1029-1 Pin Repair and Refinish Figure 601





## **COMPONENT MAINTENANCE MANUAL**

#### **MISCELLANEOUS PARTS REFINISH - REPAIR 4-1**

#### 1. General

- A. This procedure has the data necessary to refinish the parts which are not given in the specified repairs.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in the repair.
- C. Refer to IPL Figure 1 for the item numbers.

#### 2. Refinish of Other Parts

A. References

Reference	Title
SOPM 20-30-02	STRIPPING OF PROTECTIVE FINISHES
SOPM 20-41-01	DECODING TABLE FOR BOEING FINISH CODES
SOPM 20-50-08	APPLICATION OF BONDED SOLID FILM LUBRICANTS
SOPM 20-60-03	LUBRICANTS

#### B. Procedure

**NOTE**: For stripping of protective finishes, refer to SOPM 20-30-02. For decoding table for Boeing finish codes, refer to SOPM 20-41-01. For lubricants, refer to SOPM 20-60-03.

- (1) Repair of the parts listed in this repair is only replacement of the original finish.
- (2) Refer to REPAIR 4-1, Table 601 for the refinish details.

#### Table 601: Refinish Details

IPL FIG. & ITEM	MATERIAL	FINISH
Fig. 1		
Fitting (43,46)		Passivate (F-17.25, which replaces F-17.09).
Rack (55)	15-5PH CRES 180-200 ksi	Passivate (F-17.25, which replaces F-17.09). Apply ARMOLOY chrome plate or thin dense chrome plate (F- 15.43). Then apply dry film lubricant, Sermalube, Type 20 or Teceram 520 SOPM 20-50-08.
Shield (70)	304 CRES	Passivate (F-17.25, which replaces F-17.09).
Adapter (80)	15-5PH CRES 180-200 ksi	Passivate (F-17.25, which replaces F-17.09).
Clamp (120)	15-5PH CRES 150-170 ksi	Passivate (F-17.25, which replaces F-17.09).
Housing (130)	17-4PH CRES 130 ksi min	Passivate (F-17.25, which replaces F-17.09).



#### ASSEMBLY

#### 1. General

- A. This procedure has the data necessary to assemble the CFM-56 engine throttle fuel control box assembly.
- B. Refer to the Standard Overhaul Practice Manual (SOPM) for the SOPM subjects identified in the assembly.
- C. Refer to IPL Figure 1 for item numbers.

#### 2. Assembly (IPL Figure 1, ASSEMBLY, Figure 701)

A. Consumable Materials

**NOTE:** Equivalent substitutes may be used.

Reference	Description	Specification
A00253	Adhesive - Epoxy, 2 Part, RT Cure	BMS5-109, Type II, Class 2
G01048	Lockwire - Corrosion Resistant Steel (0.032 In. Dia	a.) NASM20995 <sup>~</sup> C32

B. References

Reference	Title
SOPM 20-50-01	BOLT AND NUT INSTALLATION
SOPM 20-50-02	INSTALLATION OF SAFETYING DEVICES
SOPM 20-50-12	APPLICATION OF ADHESIVES
SOPM 20-60-04	MISCELLANEOUS MATERIALS

C. Procedure

**NOTE:** For bolt and nut installation, refer to SOPM 20-50-01. For miscellaneous materials, refer to SOPM 20-60-04.

- (1) Attach rack shield (70) to housing (130) with sleeve fitting (65) and coupling nut (60).
- (2) Install the rack (55) into the housing (130) as shown in ASSEMBLY, Figure 701.
- (3) Put eccentric pin (95) through housing (130) to attach bearing (100) in position.
- (4) Attach eccentric pin (95) with washer (90) and nut (85), but do not fully tighten the nut.
- (5) Turn eccentric pins (95) until rack (55) is at its lowest position and the distance between the rack and the center of the housing (130) is at a maximum.
- (6) Turn the gear assy (15A) with attached parts until rig marks are positioned as shown in ASSEMBLY, Figure 701.
- (7) Carefully press gear assy (15A) into housing (130). Make sure that the gear teeth engage teeth on rack (55) and that dowel pins (47,50) on support fitting (45) engage holes in housing.
- (8) Install washer (10) and nut (5) on gear assy (15A).
  - (a) Tighten nut (5) to 370-690 pound-inches.
  - (b) Solvent clean area (do not abrade), mask, and inject adhesive, A00253 around the edges of nut (5) (SOPM 20-50-12).

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- (c) Fill cavity of fuel control box assemblies 315A1040-6 thru -11 only (FITS AND CLEARANCES, Figure 801).
- (d) Cure adhesive, A00253 at a maximum of 190°F for 15 minutes.
- (9) Adjust eccentric pins (95) to set the minimum backlash, not to more than 0.006 inch, between rack (55) and gear assy (15A).

**NOTE**: Backlash must be at a minimum over full stroke of rack with no evidence of binding. Measure the backlash at the location shown in ASSEMBLY, Figure 701.

- (10) Put the control box assy in a vertical position.
- (11) Pull rack (55) out until it is at the approximate maximum travel position.
- (12) To make sure the rack travels down to its lower position, apply a load not larger than 0.4 pounds.
- (13) Align rig mark on flange of synchro (125A) with rig mark on control box housing (130) and align dots on synchro gear with rig mark on gear assy (15A).
- (14) Put clamps (120) over synchro (125A) and secure with bolts (115), washers (110), and nuts (105A). Make sure that the backlash between synchro (125A) and gear assy (15A) is not more than 0.006 inch.
- (15) Install the adapter (80) with nut (75) in the control box housing (130). Do not fully tighten nut (75) at this time.
- (16) Tighten nuts (85) and use MS20995C32 lockwire, G01048 to connect them together by the doubletwist procedure (SOPM 20-50-02).
- (17) Secure coupling nut (60) to housing (ASSEMBLY, Figure 701) with MS20995C32 lockwire, G01048 using the double-twist procedure (SOPM 20-50-02).
- (18) Adjust the rack (55) and the gear assembly (15A) positions as follows:
  - (a) Turn the rack (55) and the gear assembly (15A) until the rig marks are engaged.
  - (b) Put a 0.25-inch diameter rig pin into the housing (130) at the hole marked -R-. Be sure the gear rig marks are aligned at both locations.
  - (c) Measure dimension "B" from the surface of the cable connector and scribe a line on the housing at that point.
  - (d) Remove the rig pin from the housing.
  - (e) Extend the rack an additional 2.00 inches (same dimension as dimension "D") from scribe mark.
  - (f) Lightly push the rack to the opposite end, but stop when you feel resistance. Make sure that the stroke is within the dimension "C" limits shown.
- (19) Adjust the adapter (80) to dimension "A", and then tighten the nut (75) to the housing (130).
- (20) Install a MS20995C32 lockwire, G01048 on the nut (75), the housing (130) and the adapter (80) by the double-twist procedure (SOPM 20-50-02).
- (21) Before you move fuel control box, temporarily tie back the electrical connector for the synchro.







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	DIMENSIONS				
PART NUMBER	$(\mathbf{k})$	В	C		
315A1040-3 THRU -7	0.360 0.340	4.449 4.429	5.125	6.449 6.429	
315A1040-8,-9	0.160 0.140	4.250 4.230	5.125	6.250 6.230	
315A1040-10,-11	0.385 0.365	4.474 4.454	5.125	6.474 6.454	

THE MINIMUM LIMIT OF TRAVEL

ITEM NUMBERS REFER TO IPL FIG. 1 All dimensions are in inches

Assembly Details Figure 701 (Sheet 2 of 2)

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#### FITS AND CLEARANCES



Fits and Clearances Figure 801 (Sheet 1 of 2)

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			Design Dimension				Service Wear Limit		Limit
Ref Letter	Ma† Iter	ting n No.	Dimer	nsion	Assembly Clearance		Dimension		Maximum
Fig.801	IPL	Fig.1	Min	Max	Min	Max	Min	Max	Clearance
	ID	40A	1.4998	1.5002	0,0008	0.0007		1.5033	0.0070
A	OD	30	1.4994	1.5001	0.0008		1.4970		0.0032
D	ID	30	1.0623	1.0627	0 0002	0 0000		1.0651	0 0070
В	OD	43,45	1.0618	1.0621	0.0002	0.0009	1.0597		0.0030
<u>,</u>	ID	30	1.0623	1.0627	0.0007	0.0077		1.0678	0.0050
L	OD	130A	1.0590	1.0620	0.0003	0.0037	1.0569		0.0058
D	55		0.1700	0.1715	0.00	4 3	0.1480	0.016	0 3
E	40A	$\bigvee$	2.2227	2.2235	0.00	4 3	2.2178	0.016	0 3

> DIMENSION OVER 0.0576 INCH DIAMETER PIN

> dimension over 0.0576 inch diameter pin to center of bore

> MAXIMUM PERMITTED BACKLASH BETWEEN GEAR AND RACK > NEGATIVE VALUES ARE AN INTERFERENCE FIT

ALL DIMENSIONS ARE IN INCHES

Fits and Clearances Figure 801 (Sheet 2 of 2)

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REF	IPL	NAME	TORQUE*		
FIG. NO.	ITEM NO.	NAME	POUND-INCHES	POUND-FEET	
1	5	Nut	370–690	31-58	

\* REFER TO SOPM 20-50-01 FOR TORQUE VALUES OF STANDARD FASTENERS.

Torque Table Figure 802





SPECIAL TOOLS, FIXTURES, AND EQUIPMENT

## (NOT APPLICABLE)

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#### **ILLUSTRATED PARTS LIST**

#### 1. Introduction

- A. The Illustrated Parts List (IPL) contains an illustration and a list of component parts you can repair or replace. The Illustrated Parts Catalog (IPC) shows how to use the Boeing part number system.
- B. This shows how parts are related: The relation of each item to its next higher assembly (NHA) is shown in the NOMENCLATURE column. Use the indenture system that follows:

1	2	3	4	5	6	7
•	-	•	•	•	•	

- . Assembly
- . Attaching parts for assembly
- . Detail parts for assembly
- . . Subassembly
- . . Attaching parts for subassembly
- . . . . Detail parts for subassembly
- . . . Sub-subassembly
- . . . Attaching parts for subassembly
  - . . . . Details parts for sub-subassembly

Detail Installation Parts (Included only if installation parts may be sent to the shop as part of assembly)

- C. Each top assembly is given one use code letter (A, B, C, etc.) in the USAGE CODE column. All subsequent component parts in the list can have one or more of the use code letters to show effectivity to top assemblies. A component part without a use code applies to all top assemblies.
- D. An alphabetical letter is added after the item number for optional parts, parts changed by a Service Bulletin, configuration differences (except left-handed and right-handed parts), last engineering releases, and parts added between item numbers in a sequence. The alphabetical letter will not be shown on the illustration for equivalent parts of the same part number.
- E. Color-coded parts are identified with a single digit alpha following the dash number or with "SP" suffix. If the "SP" suffix is used, it represents consolidation of all color codes applicable for a given usage which are not separately listed. Orders for color-coded parts should include the registry number of the airplane for which the parts are ordered.
- F. If a part number is 15 characters long but will not fit in the part number column, the part number will be displayed with a "~" at the end of the line and will be continued on the next line. The "~" denotes that the part number continues on the next line.
- G. Parts changed by a Service Bulletin are shown by PRE SB XXXX and POST SB XXXX added to the NOMENCLATURE column.
  - (1) When a new top assembly is added by a Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the top assembly level only. The configuration differences at the detail part level are shown by use code letters.
  - (2) When the top assembly part number is not changed by the Service Bulletin, PRE SB XXXX and POST SB XXXX will be added at the detail level.
- H. Interchangeable Parts

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Optional (OPT)		The part is optional to and interchangeable with other parts that have the same item number.
Replaces, Replaced by and not interchangeable with (REPLACES, REPLACED BY AND NOT INTCHG/W)		The part replaces and is not interchangeable with the initial part.
Replaces, Replaced by (REPLACES, REPLACED BY)		The part replaces and is interchangeable with, or is an alternative to, the initial part.
		VENDOR CODES
Code	Name	
02758	NETWORKS 9750 DE SC CHATSWOF FORMERLY	S ELECTRONIC CORP U S BEARING DIV DTO AVENUE RTH, CALIFORNIA 91311-4409 Y U S BEARING DIV NETWORKS ELEC CORP
06989	API MOTIO WESTTOWN WEST CHES	N N ROAD AT WESTCHESTER PARK PO BOX 547 STER, PENNSYLVANIA 19380
34336	L AND S M. 2019 SOUTH WICHITA, K	ACHINE COMPANY INCORPORATED HWEST BLVD PO BOX 12264 XANSAS 67277
50632	Kamatics 1335 Blue Bloomfiei	CORP SUB OF KAMAN CORP HILLS ROAD LD, CONNECTICUT 06002-1304
55231	TRIBON BE 6200 HILLC CLEVELANI FORMERLY	ARING COMPANY REST DR D, OHIO 44125 ′ PURE CARBON COMPANY V80894
78118	SPLIT BALL HIGHWAY 4 LEBANON,	- BEARING DIV OF MPB CORP 4 NEW HAMPSHIRE 03766-7301
97393	Shur-lok 2541 White Irvine, Cai Formerly Formerly	CORPORATION E ROAD PO BOX 19584 LIFORNIA 92623-9584 ′ SHUR LOK CORP VB0060 ′ IN SANTA ANA, CALIFORNIA 92714

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#### NUMERICAL INDEX

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
10602-00		1	100	2
315A1019-1		1	43	1
		1	46	1
315A1019-10		1	46B	1
315A1019-3		1	45	1
315A1019-7		1	46A	1
315A1019-9		1	45A	1
315A1026-1		1	25A	1
315A1036-3		1	55	1
315A1036-7		1	55A	1
315A1036-8		1	55B	1
315A1036-9		1	55C	1
315A1040-10		1	1H	RF
315A1040-11		1	1J	RF
315A1040-3		1	1A	RF
315A1040-4		1	1B	RF
315A1040-5		1	1C	RF
315A1040-6		1	1D	RF
315A1040-7		1	1E	RF
315A1040-8		1	1F	RF
315A1040-9		1	1G	RF
315A1043-1		1	130	1
315A1043-2		1	130A	1
315A1043-3		1	130B	1
315A1045-10		1	15C	1
315A1045-11		1	40C	1
315A1045-12		1	40D	1
315A1045-13		1	40E	1
315A1045-14		1	40F	1
315A1045-15		1	15D	1
315A1045-16		1	15E	1
315A1045-17		1	15F	1
315A1045-6		1	15A	1
315A1045-7		1	40A	1

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## COMPONENT MAINTENANCE MANUAL

PART NUMBER	AIRLINE PART NUMBER	FIGURE	ITEM	UNITS PER ASSEMBLY
315A1045-8		1	15B	1
315A1045-9		1	40B	1
315A1048-1		1	120	2
315A1088-1		1	70	1
315A1919-1		1	48	2
315T1005-1		1	75B	1
315T1029-1		1	95	2
315T4012-1		1	80	1
AN960C10L		1	90	2
AN960C416L		1	110	2
AN960C716		1	10	1
BACN10DP5J		1	60	1
BACN10GW7AS		1	5	1
BACN10JC4CM		1	105A	2
BCREFA1477		1	30	1
BCREFA1478		1	30A	1
BCREFA1479		1	30B	1
HS140-3		1	125A	1
HSP4TL104		1	100	2
KRP114804BT		1	100	2
LS5097		1	75	1
		1	75A	1
MS20427M3C6		1	20A	10
MS20819-5J		1	65	1
NAS509-3C		1	85A	2
NAS607-2-3P		1	47	2
		1	50	2
NAS6704-12		1	115	2
S17-24BIE3P515LY198		1	30	1
S17-24BIE5P515LY198		1	30A	1
S17-24BIE7P515LY198		1	30B	1
S315A110-3		1	125A	1
S315N166-1		1	100	2
SL2999-3C		1	85	2

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CFM-56 Engine Throttle Fuel Control Box Assembly IPL Figure 1 (Sheet 1 of 2)

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CFM-56 Engine Throttle Fuel Control Box Assembly IPL Figure 1 (Sheet 2 of 2)

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1–					
-1	315A1040-10		DELETED		
-1A	315A1040-3		BOX ASSY-ENG THROT FUEL CONT	А	RF
–1B	315A1040-4		BOX ASSY-ENG THROT FUEL CONT (PRE SB 737-76-1021) (PRE SB 737-76-1022)	В	RF
–1C	315A1040-5		BOX ASSY-ENG THROT FUEL CONT (POST SB 737-76-1021) (PRE SB 737-76-1022)	С	RF
–1D	315A1040-6		BOX ASSY-ENG THROT FUEL CONT (POST SB 737-76-1022)	D	RF
–1E	315A1040-7		BOX ASSY-ENG THROT FUEL CONT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	E	RF
–1F	315A1040-8		BOX ASSY-ENG THROT FUEL CONT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	F	RF
–1G	315A1040-9		BOX ASSY-ENG THROT FUEL CONT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	G	RF

-Item not Illustrated

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
–1H	315A1040-10		BOX ASSY-ENG THROT FUEL CONT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	Η	RF
-1J	315A1040-11		BOX ASSY-ENG THROT FUEL CONT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	J	RF
5	BACN10GW7AS		. NUT		1
10	AN960C716		. WASHER		1
15	315A1045-15		DELETED		
15A	315A1045-6		. GEAR ASSY (OPT ITEM 15B)	А, В	1
–15B	315A1045-8		. GEAR ASSY (OPT ITEM 15A)	А, В	1
-15C	315A1045-10		. GEAR ASSY-*[1] (POST SB 737-76-1021) (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	C, D, F-H	1

-Item not Illustrated

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FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE CODE	UNITS PER ASSY
1_					
-15D	315A1045-15		. GEAR ASSY-*[1] (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	C, D, F-H	1
–15E	315A1045-16		. GEAR ASSY (VARIABLE) (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	E	1
–15F	315A1045-17		. GEAR ASSY (VARIABLE) (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	J	1
20	MS20427M3C6		DELETED		
20A	MS20427M3C6		RIVET		10
25	315A1026-1		DELETED		
25A	315A1026-1		RING-BRG CLAMP		1
30	BCREFA1477		BEARING (V78118) (S17-24BIE3P515LY198) (OPT ITEM 30A, 30B)		1

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-Item not Illustrated



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1–					
–30A	BCREFA1478		BEARING (V78118) (S17-24BIE5P515LY198) (OPT ITEM 30, 30B)		1
–30B	BCREFA1479		BEARING (V78118) (S17-24BIE7P515LY198) (OPT ITEM 30, 30A)		1
30C	BCREFA1477		DELETED		
-30D	BCREFA1478		DELETED		
-30E	BCREFA1479		DELETED		
35	315A1045-12		DELETED		
–35A	315A1045-11		DELETED		
–35B	315A1045-7		DELETED		
–35C	315A1045-9		DELETED		
–35D	315A1045-13		DELETED		
–35E	315A1045-14		DELETED		
40	315A1019-3		DELETED		
40A	315A1045-7		GEAR (USED ON ITEM 15A)	А, В	1
-40B	315A1045-9		GEAR (USED ON ITEM 15B)	А, В	1
-40C	315A1045-11		GEAR (USED ON ITEM 15C)	C, D, F-H	1
-40D	315A1045-12		GEAR (USED ON ITEM 15D)	C, D, F-H	1
-40E	315A1045-13		GEAR	Е	1
-40F	315A1045-14		GEAR	J	1
-43	315A1019-1		. FITTING-SPRT	А	1

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-Item not Illustrated



FIG/ ITEM	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USAGE CODE	UNITS PER ASSY
1_					
-45	315A1019-3		. FITTING ASSY-SPRT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	B-D, F, H	1
-45A	315A1019-9		. FITTING ASSY-SPRT (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	E, G, J	1
46	315A1019-1		FITTING (OPT ITEM 46A)	B-D, F, H	1
46A	315A1019-7		FITTING (OPT ITEM 46)	B-D, F, H	1
–46B	315A1019-10		FITTING	E, G, J	1
47	NAS607-2-3P		PIN-DOWEL		2
48	315A1919-1		DOWEL		2
50	NAS607-2-3P		. PIN-DOWEL	А	2
-50A	315A1019-7		DELETED		
–50B	315A1019-10		DELETED		
55	315A1036-3		. RACK-GEAR (OPT ITEM 55A) (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	A-E	1

-Item not Illustrated

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FIG/	PART NUMBER	AIRLINE PART NUMBER	NOMENCLATURE	USAGE	UNITS PER ASSY
1_		HOMBEN		0002	
–55A	315A1036-7		. RACK-GEAR (OPT ITEM 55) (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	A-E	1
–55B	315A1036-8		. RACK-GEAR (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	F, G	1
–55C	315A1036-9		. RACK-GEAR (REWORKED BY COMPONENT SB 315A1040-76-01) (PERMITS FUEL CONTROL BOX ASSEMBLIES 315A1040-6 THRU -11 TO BE REWORKED INTO ANY OTHER 315A1040-6 THRU -11 C05FIGURATION BY CHANGING THE SUPPORT FITTING ASSEMBLY, RACK, AND GEAR ASSEMBLY)	H, J	1
60	BACN10DP5J		. NUT-COUPLING		1
65	MS20819-5J		. SLEEVE		1
65A	315A1036-7		DELETED		
–65B	315A1036-9		DELETED		
65C	315A1036-8		DELETED		
70	315A1088-1		. SHIELD		1
75	LS5097		. NUT (V34336)	А	1
–75A	LS5097		. NUT (V34336) (OPT ITEM 75B)	B-J	1

-Item not Illustrated

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FIG/		AIRLINE PART		USAGE	UNITS PER
	PART NUMBER	NUMBER	1234567	CODE	A351
1–	· · - <b>-</b> · · · ·				
–75B	315T1005-1		. NUT (OPT ITEM 75A)	B-J	1
80	315T4012-1		. ADAPTER		1
85	SL2999-3C		. NUT (V97393) (OPT ITEM 85A)		2
85A	NAS509-3C		. NUT (OPT ITEM 85)		2
90	AN960C10L		. WASHER		2
95	315T1029-1		. PIN-ECCENTRIC		2
–95A	NAS509-3C		DELETED		
100	10602-00		. BEARING-BACKUP (V55231) (SPEC S315N166-1) (OPT HSP4TL104 (V02758)) (OPT KRP114804BT (V50632))		2
105	315T1029-1		DELETED		
105A	BACN10JC4CM		. NUT		2
110	AN960C416L		. WASHER		2
–110A	VTB08600		DELETED		
–110B	10602-00		DELETED		
115	NAS6704-12		. BOLT		2
120	315A1048-1		. CLAMP-SYNCHRO		2
125	NAS6704-12		DELETED		
125A	HS140-3		. SYNCHRO-XMTR (V06989) (SPEC S315A110-3)		1
130	315A1043-1		. HOUSING	А	1
–130A	315A1043-2		. HOUSING (OPT ITEM 130B)	B-J	1
–130B	315A1043-3		. HOUSING (OPT ITEM 130A)	B-J	1
135	HS140-3		DELETED		
140	315A1043-3		DELETED		
-140A	315A1043-2		DELETED		

-Item not Illustrated

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