

TO: ALL HOLDERS OF CARRY-ALL COMPARTMENT TWO-LINK ARTICULATED HINGE ASSEMBLY  
OVERHAUL MANUAL, 25-21-05

REVISION NO. 3, DATED MAR 1/00

HIGHLIGHTS

DESCRIPTION OF CHANGE	TOPICS AFFECTED												
	D & O	D / A s s y	C l e a n i n g	I n s p / C h k	R e p a i r	A s s y	F / C	T e s t	T / S h o o t i n g	S / T o o l s	S t o r a g e	I P L	L / O v e r h a u l
Changed the pin (135) part number from MS20392-2C17 to MS20392-3C17 to agree with engineering drawings							X					X	
Changed the ID specified for the bushing (165) from 0.3146-0.3156 to 0.2510-0.2520							X						
Changed the washer (130) part number from AN960PD10L to AN960PD416L												X	
Added gas springs that are optional to the 65C16855-3 gas spring assy (item 65)												X	

Mar 1/00

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HIGHLIGHTS  
Page 1 of 1



OVERHAUL MANUAL

# CARRY-ALL COMPARTMENT TWO-LINK ARTICULATED HINGE ASSEMBLY

## 25-21-05

BOEING P/N 65C20204-21 thru -28

AIRLINE P/N

THE FOLLOWING DIRECTIVES APPLY TO THIS SUBJECT:

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVES	DATE DIRECTIVE INCORPORATED INTO TEXT
		PRR 24374 PRR 24577 PRR 32880 PRR 33036	Jun 5/85 Jun 5/85 Jun 5/85 Jun 5/85

LIST OF EFFECTIVE PAGES					
* Indicates pages revised, added or deleted in latest revision					
F Indicates foldout pages - print one side only					
PAGE	DATE	PAGE	DATE	PAGE	DATE
25-21-05					
T-1	Jun 5/85				
T-2	BLANK				
* LEP-1	Mar 1/00				
LEP-2	BLANK				
T/C-1	Jun 5/85				
T/C-2	BLANK				
1	Jun 1/97				
2	BLANK				
101	Jun 5/85				
102	BLANK				
301	Jun 5/85				
302	BLANK				
401	Jun 5/85				
402	Jun 5/85				
403	Jun 5/85				
404	Jun 5/85				
405	Jun 5/85				
406	Jun 1/97				
407	Jun 1/97				
408	Jun 1/97				
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410	BLANK				
501	Jun 5/85				
502	Jun 1/97				
503	Jun 5/85				
504	BLANK				
601	Jun 1/97				
602	Jun 5/85				
* 603	Mar 1/00				
604	BLANK				
1101	Jun 1/97				
1102	Jun 1/97				
* 1103	Mar 1/00				
* 1104	Mar 1/00				
* 1105	Mar 1/00				
1106	BLANK				



OVERHAUL MANUAL

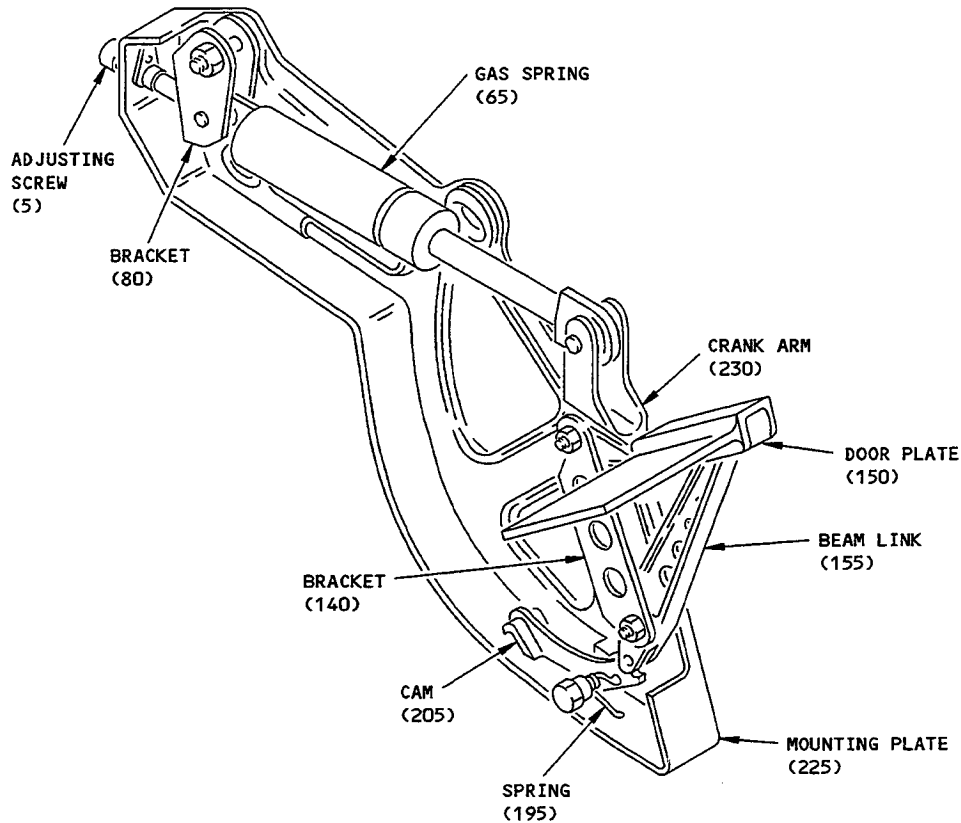
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\*[1] Special instructions not required. Use standard industry practices and the information contained in 20-30-01 and 20-30-03.

\*[2] Special instructions not required. Use standard industry practices and the information contained in 20-44-02.

## OVERHAUL MANUAL

CARRY-ALL COMPARTMENT TWO-LINK ARTICULATED HINGE ASSEMBLY

Carry-All Compartment Two-Link Articulated Hinge Assembly  
 Figure 1

DESCRIPTION AND OPERATION

1. The carry-all compartment two-link articulated hinge assembly includes brackets, gas spring, crank arm, beam link, cam, door plate, and mounting plate, with fasteners, bushings, bearings, and springs. The hinge is a pivot for the carry-all compartment door, and the gas spring prevents sudden door motion.

2. Leading Particulars (approximate)

Length — 11 inches  
 Width — 1.5 inches  
 Height — 6 inches  
 Weight — 1 pound

**OVERHAUL MANUAL****DISASSEMBLY**

1. Remove cotter pin (85), pin (100), washers (90, 92) and bushing (95). Separate gas spring (65) from crank arm (230).
2. Remove nut (25), bolt (40), washer (30) and bushing (35). Lift out gas spring (65) with attached bracket (80).
3. Remove cotter pin (60), pin (45), washer (55), bushing (50). Remove bracket (80) from gas spring (65). Do not remove bushings (70) from gas spring unless necessary for repair or replacement. Refer to appropriate vendor's instructions for overhaul of gas spring (75).
4. Remove spring pins (145) and door plate (150).
5. Remove nut (105), bolt (120), washer (110), bushing (115) and crank arm (230). Do not remove bushing (235) or bearing (240) from crank arm unless necessary for repair or replacement.
6. Remove cotter pin (125), pin (135), washer (130), bracket (140) and beam link (155). Do not remove bearing (160), bushings (165, 175) or pin (170) unless necessary for repair or replacement.
7. Remove nut (185), lock (220) (if applicable), bolt (190), washers (210, 215), spring (195), cam (205), and spacer (200).
8. Remove screw (5) and spacer (10) from mounting plate (225). Do not remove nut plate (15) unless necessary for repair or replacement.

**OVERHAUL MANUAL****INSPECTION/CHECK**

1. Check all parts for obvious defects in accordance with standard industry practices. Refer to Fits and Clearances for design dimensions and wear limits.
2. Spring (195) -- Deflect arm 45 degrees. Load shall be 0.50-0.70 pound-inches (torque or moment load).



## OVERHAUL MANUAL

REPAIR

## 1. Materials

- A. Primer -- BMS 10-11, Type 1 (Ref 20-60-02)
- B. Lacquer -- Flat White (F-14.92) (Ref 20-41-01)
- C. Retaining Compound -- Loctite 601 (Ref 20-60-04)

## 2. Repair minor defects in accordance with standard industry practices. Refer to Fits and Clearances for design dimensions and wear limits. Other repair consists of restoration of original finish. Refer to Refinish for details.

## 3. Refinish

NOTE: Refer to 20-30-02 for stripping of protective finishes and to 20-41-01 for explanation of F and SRF finish codes.

- A. Gas spring (65) -- Fig. 401.
- B. Bracket (80) -- Passivate (F-17.09) all over. Material: 17-7PH CRES, 150-170 ksi.
- C. Bracket (140) -- Fig. 402.
- D. Door plate (150), mounting plate (225) -- Apply Alodine 1000 (F-14.05) all over. Discoloration is allowed. Material: Al alloy.
- E. Beam link (180) -- Fig. 404.
- F. Spring (195) -- Cadmium plate (F-15.02). Material: Music wire, QQ-W-470.
- G. Cam (205) -- Fig. 405.
- H. Lock (220) -- No finish. Material: AISI 301 CRES, hardened.
- I. Arm (245) -- Fig. 407.

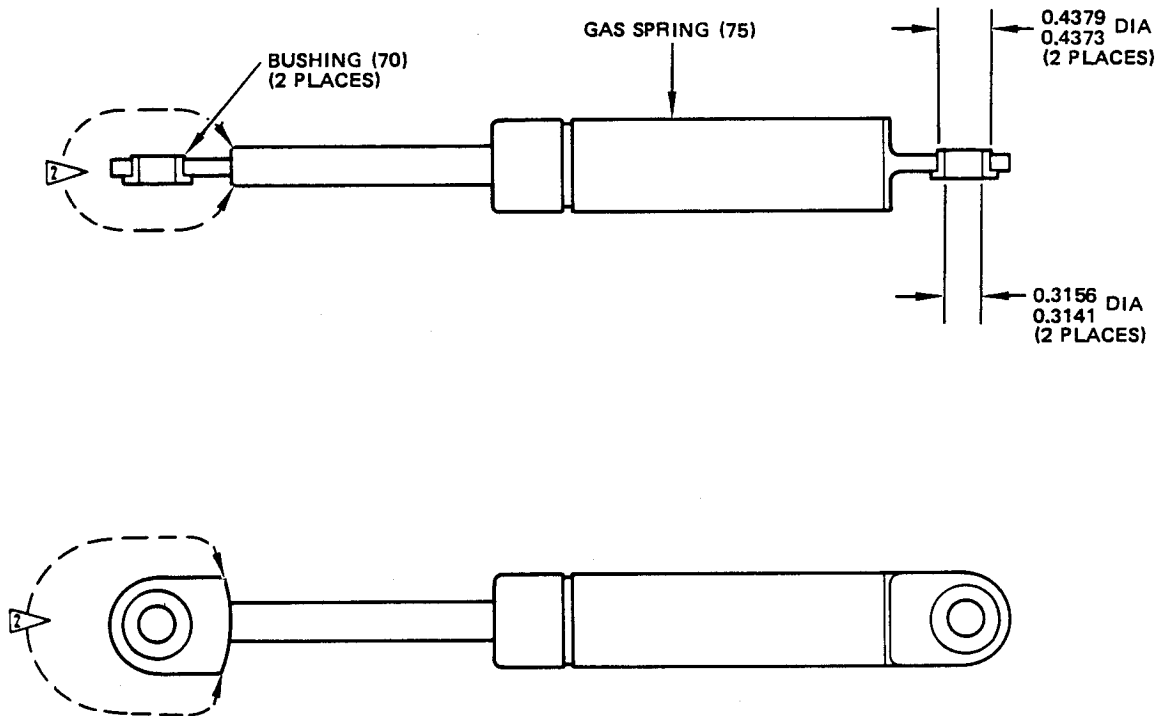


OVERHAUL MANUAL

4. Replacement

- A. Bushings (70, 165, 235) -- Remove defective bushing. Install new bushing by press fit method per 20-50-03. Check dimensions and machine as required (Fig. 401, 406).
- B. Bushing (175) -- Drive out pin (170) and remove defective bushing. Fit new bushing between link lugs and install pin (170). Check that bushing rotates freely after installation (Fig. 403).
- C. Bearings (160, 240) -- Remove defective bearing. Install new bearing with Loctite 601 compound (Fig. 403, 406).

## OVERHAUL MANUAL





NOTE: REFER TO APPROPRIATE VENDOR'S INSTRUCTIONS  
FOR OVERHAUL OF GAS SPRING (75)

**REFINISH**

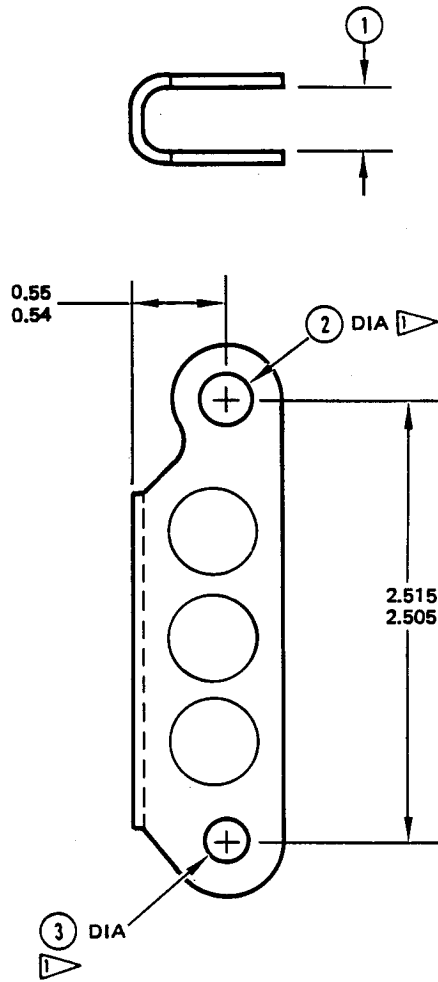
NO FINISH, EXCEPT  
APPLY PRIMER AND LACQUER  
AS NOTED 

ALL DIMENSIONS ARE IN INCHES

-  INSTALL PER 20-50-03
-  APPLY PRIMER, BMS 10-11, TYPE 1 (F-20.02)  
AND LACQUER, FLAT WHITE (F-14.92), EXCEPT  
ON BUSHING

GAS SPRING (65) 65C16855-3


OVERHAUL MANUAL



	(1)	(2)	(3)
DESIGN DIM	0.38 0.37	0.3156 0.3141	0.2505 0.2495
REPAIR LIMIT	—	—	—

REFINISH

CHROMIC ACID ANODIZE (F-17.04)

 PERPENDICULAR TO SURFACES WITHIN 0.001 TIR

REPAIR

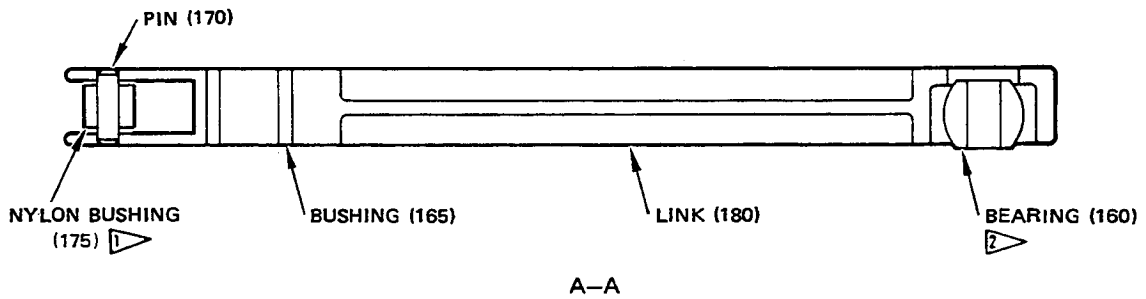
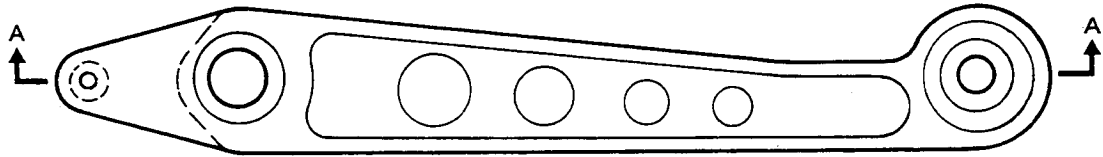
(SAME AS REFINISH)

$\sqrt{125}$  MACHINE FINISH

MATERIAL: AL ALLOY

ALL DIMENSIONS ARE IN INCHES

BRACKET (140) 65C16891-60

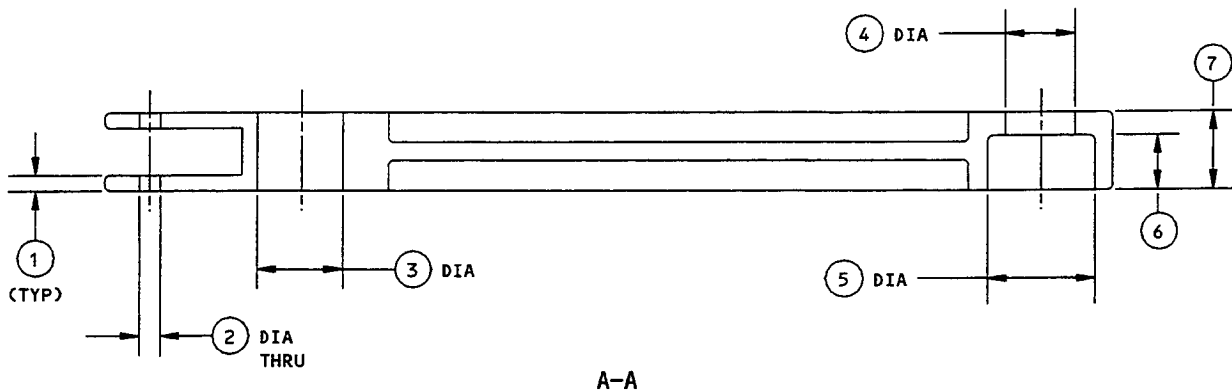
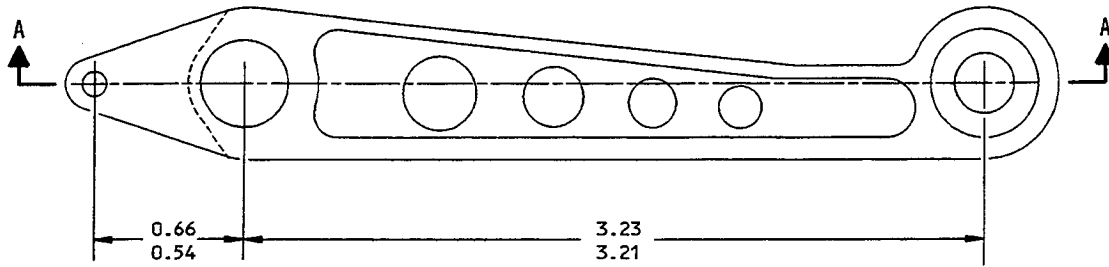


**1** BUSHING MUST ROTATE FREELY AFTER INSTALLATION

**2** INSTALL WITH LOCTITE 601.

BEAM LINK (155) 65C20204-13

OVERHAUL MANUAL



A-A

	①	②	③	④	⑤	⑥	⑦
DESIGN DIM	0.06 0.05	0.088 0.087	0.3754 0.3748	0.31 0.29	0.4702 0.4687	0.27 0.26	0.34 0.33
REPAIR LIMIT	-	-	-	-	-	-	-

**REFINISH**

CHEMICAL TREAT (F-14.05).  
 DISCOLORATION IS PERMITTED

**REPAIR**

(SAME AS REFINISH)

125/ MACHINE FINISH

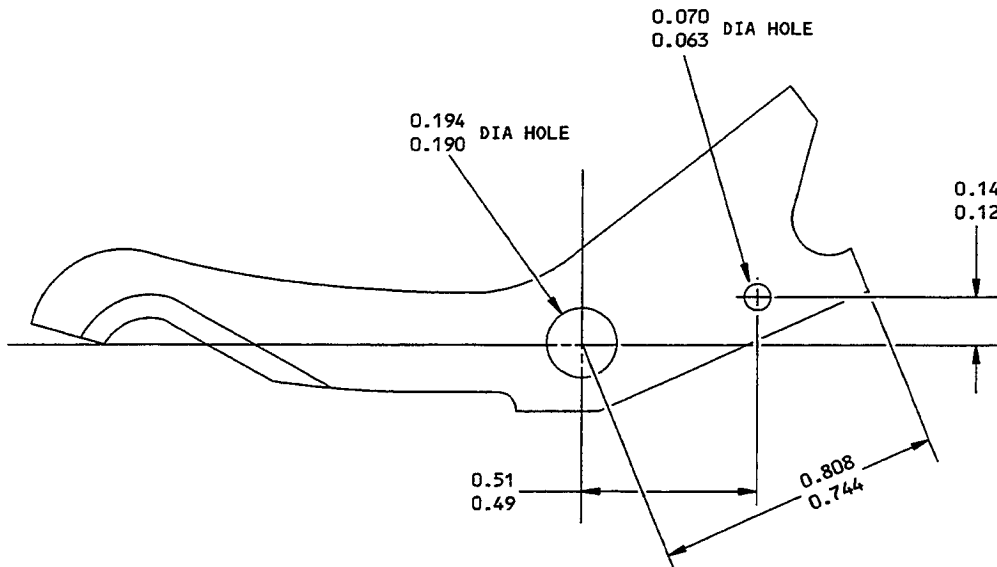
MATERIAL: AL ALLOY

ALL DIMENSIONS ARE IN INCHES

LINK (180) 65C19933-1

Link Details  
 Figure 404

**BOEING**  
**COMMERCIAL JET**  
 OVERHAUL MANUAL



REFINISH

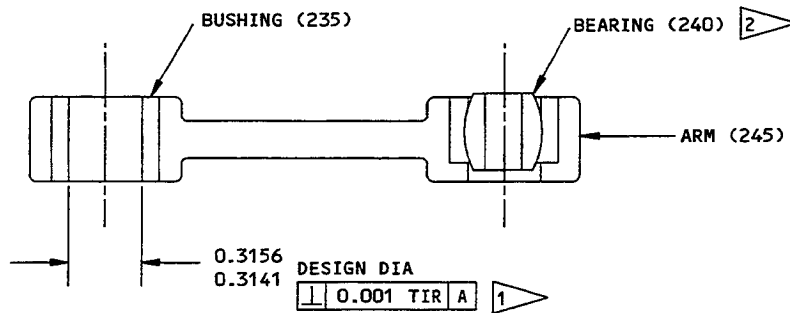
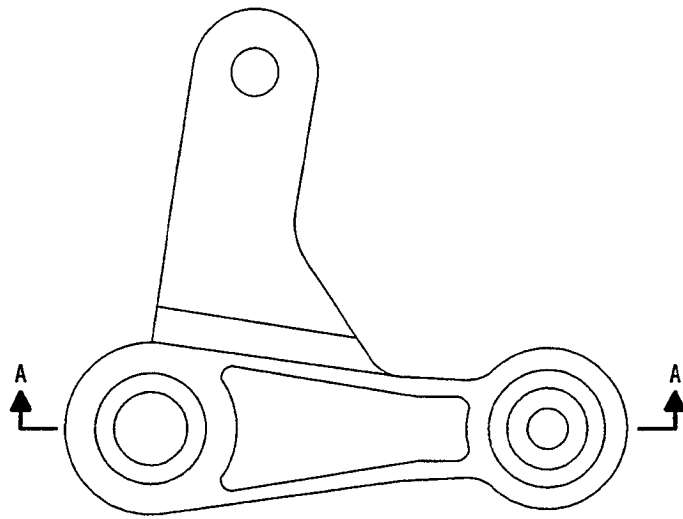
CHEMICAL TREAT (F-2.21 OR F-14.05).  
 DISCOLORATION IS PERMITTED

REPAIR

(SAME AS REFINISH)  
 MATERIAL: AL ALLOY  
 ALL DIMENSIONS ARE IN INCHES

CAM (205) 65C19935-3

Cam Details  
 Figure 405



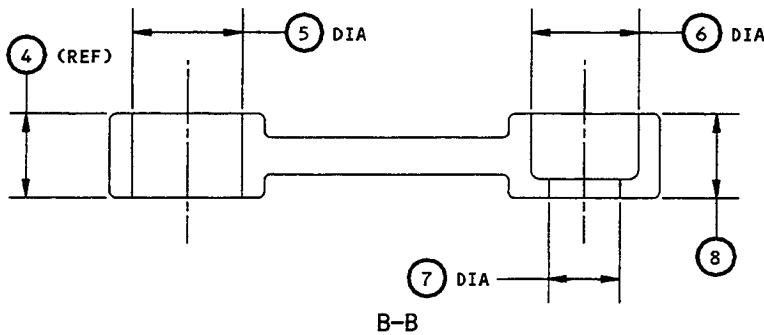
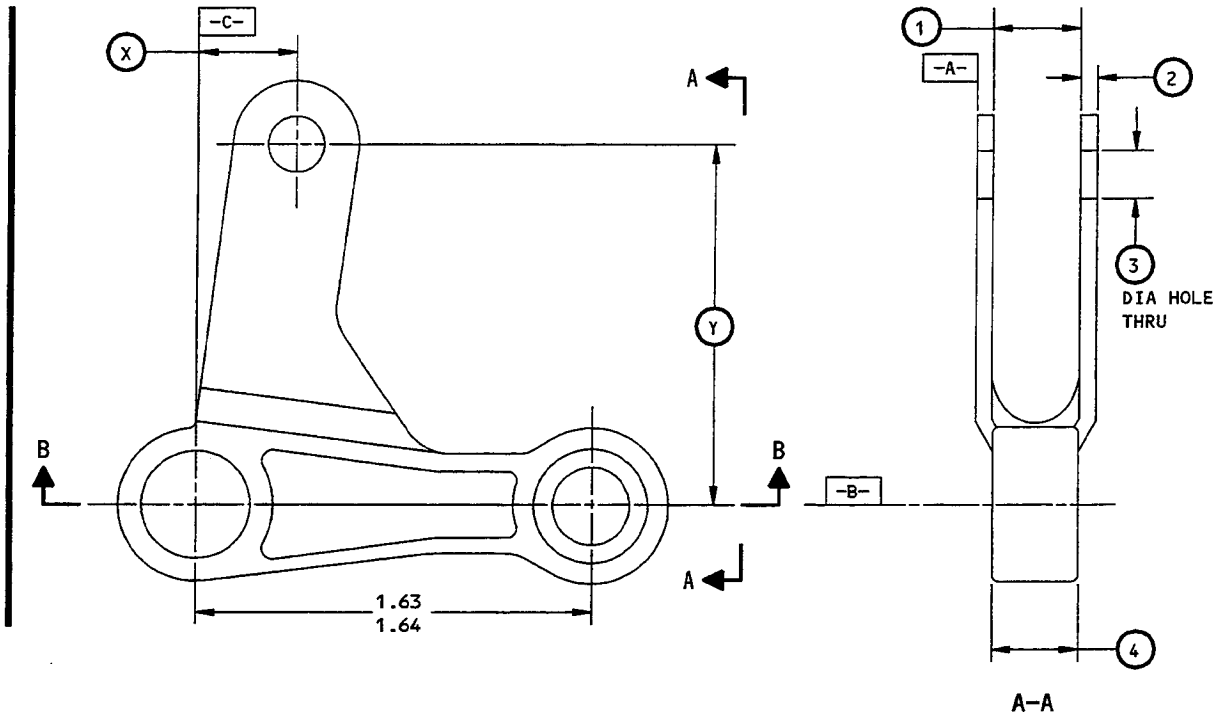
A-A

CRANK ARM (230) 65C20204-9 THRU -12

- 1 FOR DATUM -A-, SEE FIG. 407
- 2 INSTALL WITH LOCTITE 601

Crank Arm Parts Replacement  
Figure 406

OVERHAUL MANUAL



	(X)	(Y)
65C19937-1	0.40 0.38	1.46 1.44
65C19937-2	0.34 0.32	1.23 1.21
65C19937-3	0.27 0.25	0.99 0.97
65C19937-4	0.24 0.22	0.88 0.86

	(1)	(2)	(3)	(4)	(5)	(6)	(7)	(8)
DESIGN DIM	0.38 0.37	0.08 0.06	0.194 0.190	0.34 0.33	0.4379 0.4373	0.4702 0.4687	0.316 0.313	0.27 0.26
REPAIR LIMIT	-	-	-	-	-	-	-	-

**REFINISH**

CHEMICAL TREAT (F-14.05).  
DISCOLORATION IS PERMITTED

**REPAIR**

(SAME AS REFINISH)

125 / MACHINE FINISH

MATERIAL: AL ALLOY

ALL DIMENSIONS ARE IN INCHES

ARM (245) 65C19937-1 THRU 4

Arm Details  
Figure 407

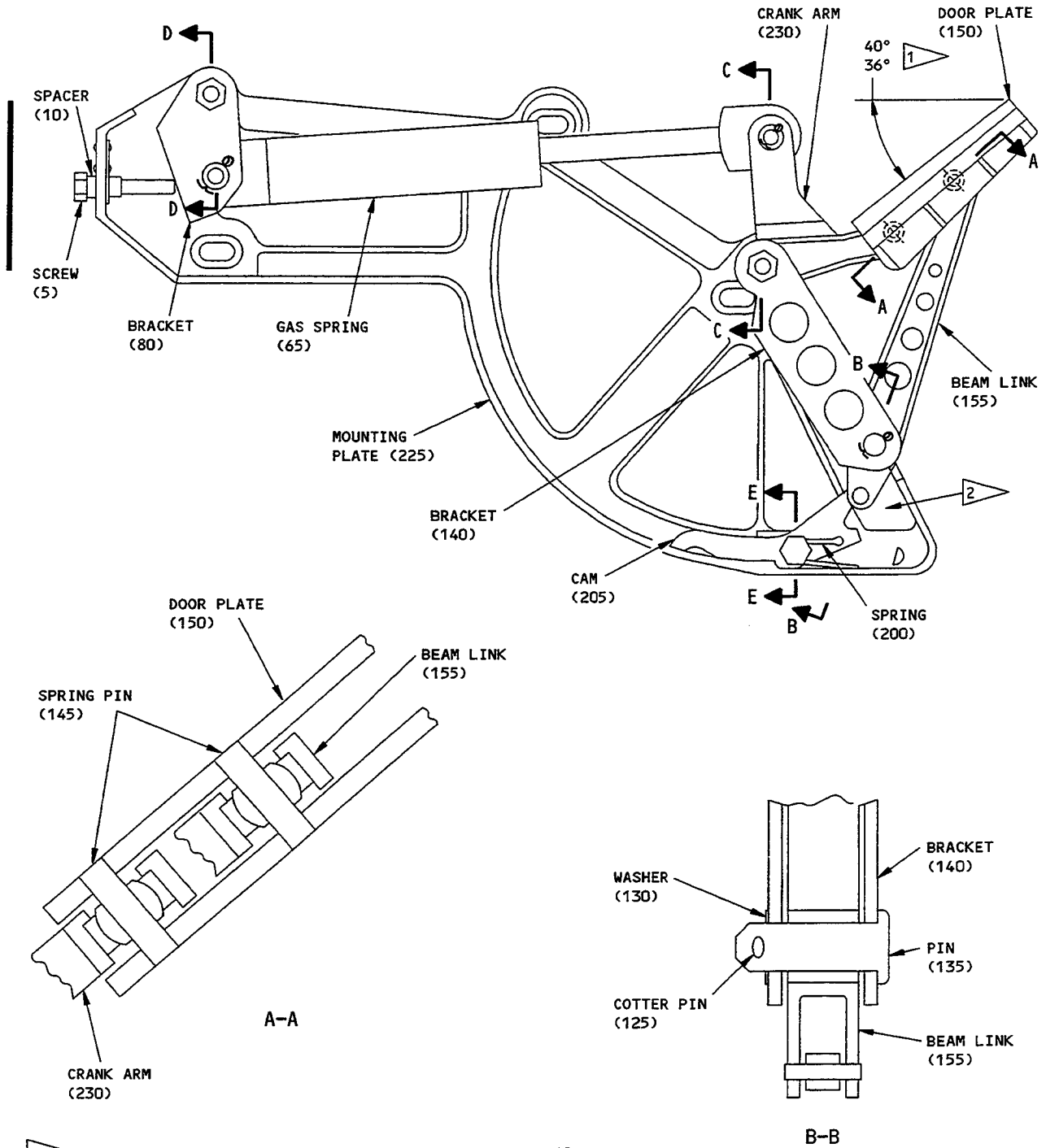


ASSEMBLY

NOTE: Refer to Fig. 501 for details of parts placement.

1. Install cam (205) and spring (195) in mounting plate (225) with spacer (200), washers (215, 210), bolt (190), lock (220) (if applicable), nut (185). Hook arm of spring into hole provided in cam as parts are installed.
2. Install crank arm (230) and bracket (140) in mounting plate (225) with bushing (115), bolt (120), washer (110) and nut (105).
3. Install lower ends of bracket (140) and beam link (155) in mounting plate (225) with pin (135), washer (130) and cotter pin (125).
4. Swing beam link upwards and connect with crank arm (230) using door plate (150) and spring pins (145).
5. Install bracket (80) on body end of gas spring (65) with bushing (50), pin (45), washer (55) and cotter pin (60). Then install upper end of bracket (80) in mounting plate (225) with bushing (35), bolt (40), washer (30) and nut (25).
6. Install blade end of gas spring (65) on crank arm (230) with bushing (95), washers (92, 90) as applicable, pin (100) and cotter pin (85).
7. Slip spacer (10) onto shank of screw (5), then thread screw (5) into nutplate (15) in end of mounting plate. Adjust screw (5) to preset door plate to angle shown.
8. Release cam (205) and cycle linkage by rotating clockwise, then return to original position. Motion shall be free of binding or interference.

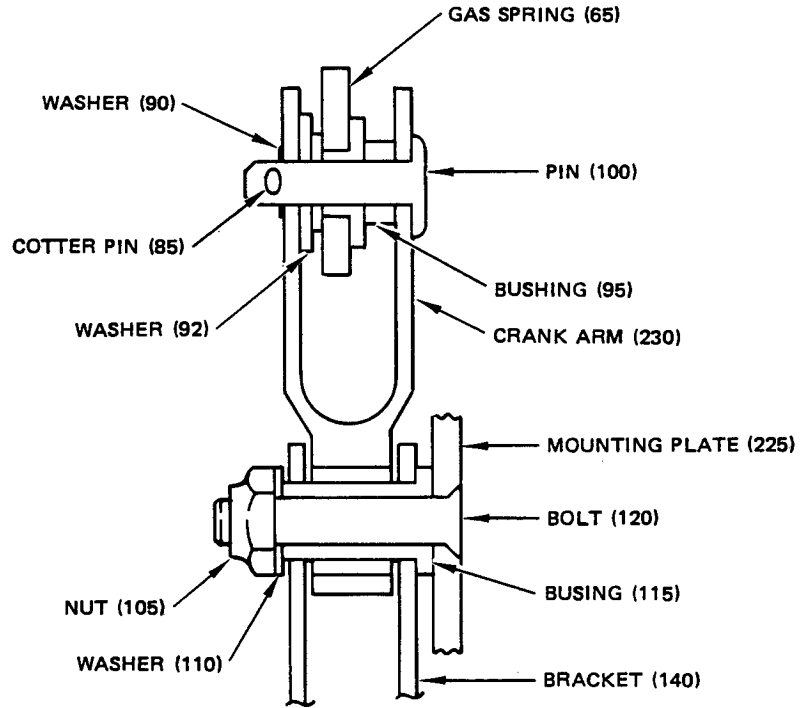
OVERHAUL MANUAL



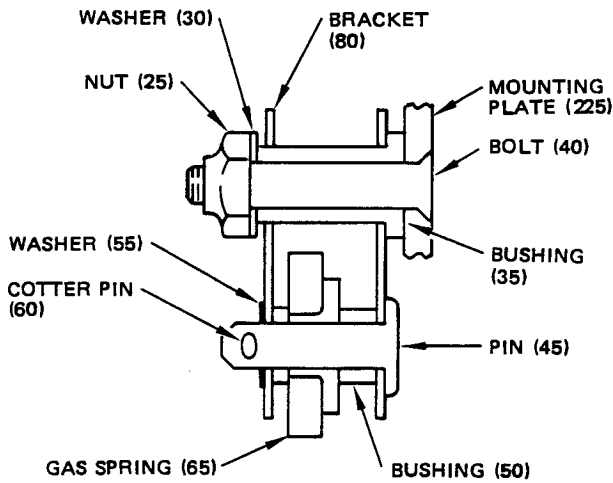
- 1 PRESET HINGE TO THIS POSITION BY ADJUSTING SCREW (5)
- 2 RELEASE CAM (205) AND CYCLE LINKAGE BY ROTATING CLOCKWISE THEN RETURNING TO POSITION SHOWN. MOTION SHALL BE FREE OF INTERFERENCE.

Assembly Details  
Figure 501 (Sheet 1)

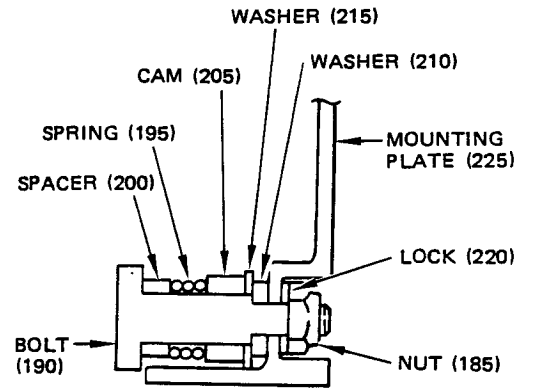
**OVERHAUL MANUAL**



C-C

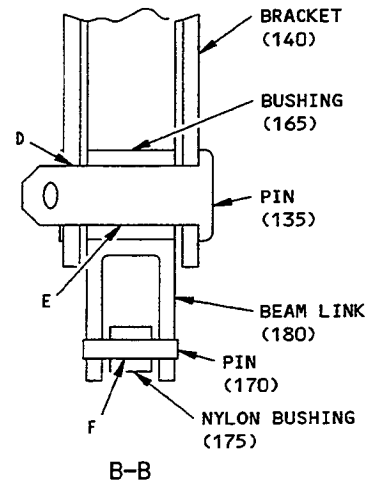
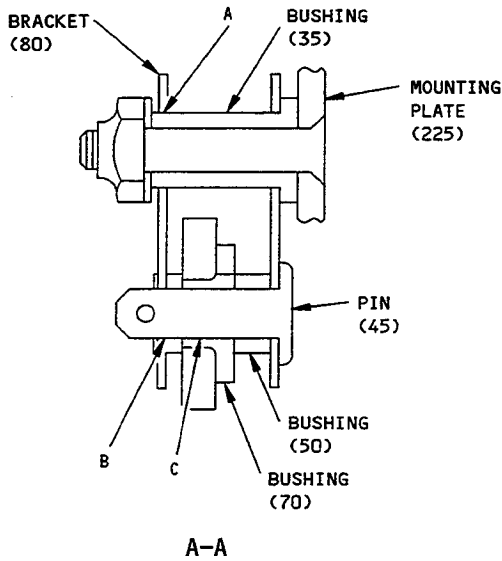
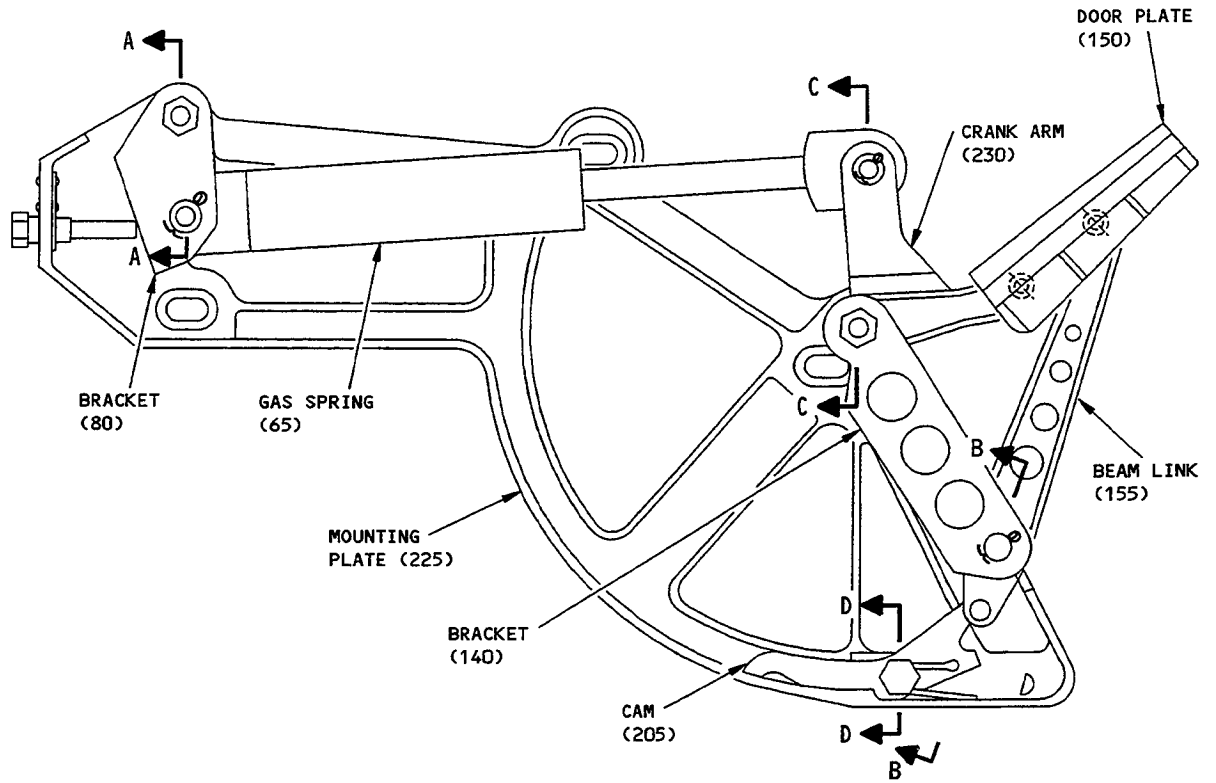


D-D

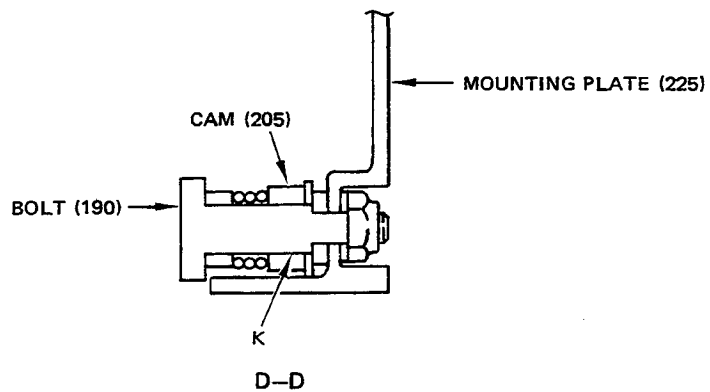
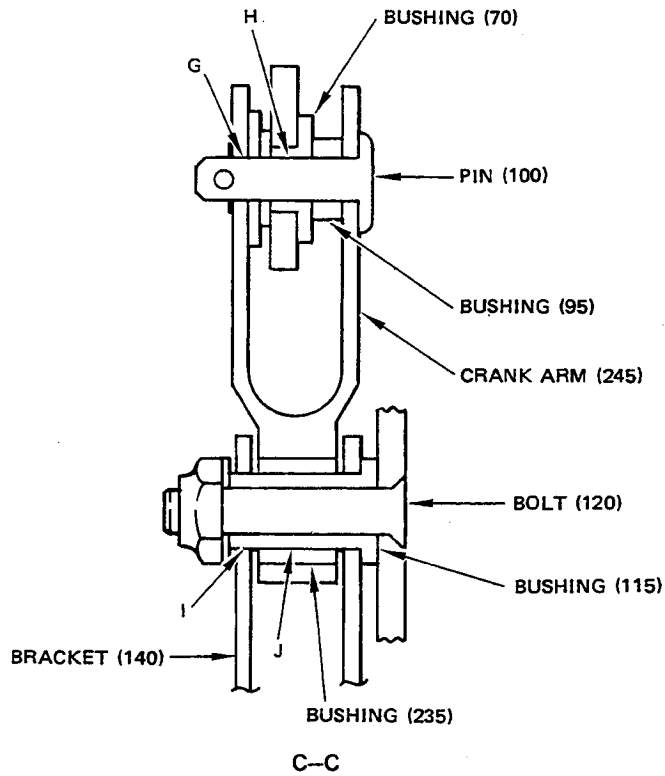


E-E

**OVERHAUL MANUAL**  
**FITS AND CLEARANCES**



Fits and Clearances  
Figure 601 (Sheet 1)



Ref Letter Fig. 601	Mating Item No. Fig. 1101	Design Dimensions				Service Wear Limits		
		Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inch)		Maximum Allowable Clearance (inch)
		Min	Max	Min	Max	Min	Max	
A	ID 80	0.3270	0.3310	0.0134	0.0179			
	OD 35	0.3131	0.3136					
B	ID 80	0.190	0.194	0.004	0.010			
	OD 45	0.184	0.186					
C	ID 70	0.3141	0.3156	0.0005	0.0025			
	OD 45	0.3131	0.3136					
D	ID 140	0.2495	0.2505	0.0015	0.0045			
	OD 135	0.2460	0.2480					
E	ID 165	0.2510	0.2520	0.0030	0.0060			
	OD 135	0.2460	0.2480					
F	ID 175	N.A.	N.A.					
	OD 170	0.089	0.090					
G	ID 245	0.190	0.194	0.004	0.010			
	OD 100	0.184	0.186					
H	ID 70	0.3141	0.3156	0.1281	0.1316			
	OD 100	0.1840	0.1860					
I	ID 140	0.3141	0.3156	0.0005	0.0025			
	OD 115	0.3131	0.3136					
J	ID 235	0.3141	0.3156	0.0005	0.0025			
	OD 115	0.3131	0.3136					
K	ID 205	0.1900	0.1940	0.0015	0.0061			
	OD 190*[1]	0.1879	0.1885					
K	ID 205	0.1900	0.1940	0.0005	0.0050			
	OD 190*[2]	0.1890	0.1895					

\*[1] BACB30PF3-7 or BACB30PW3-7 bolts (190)

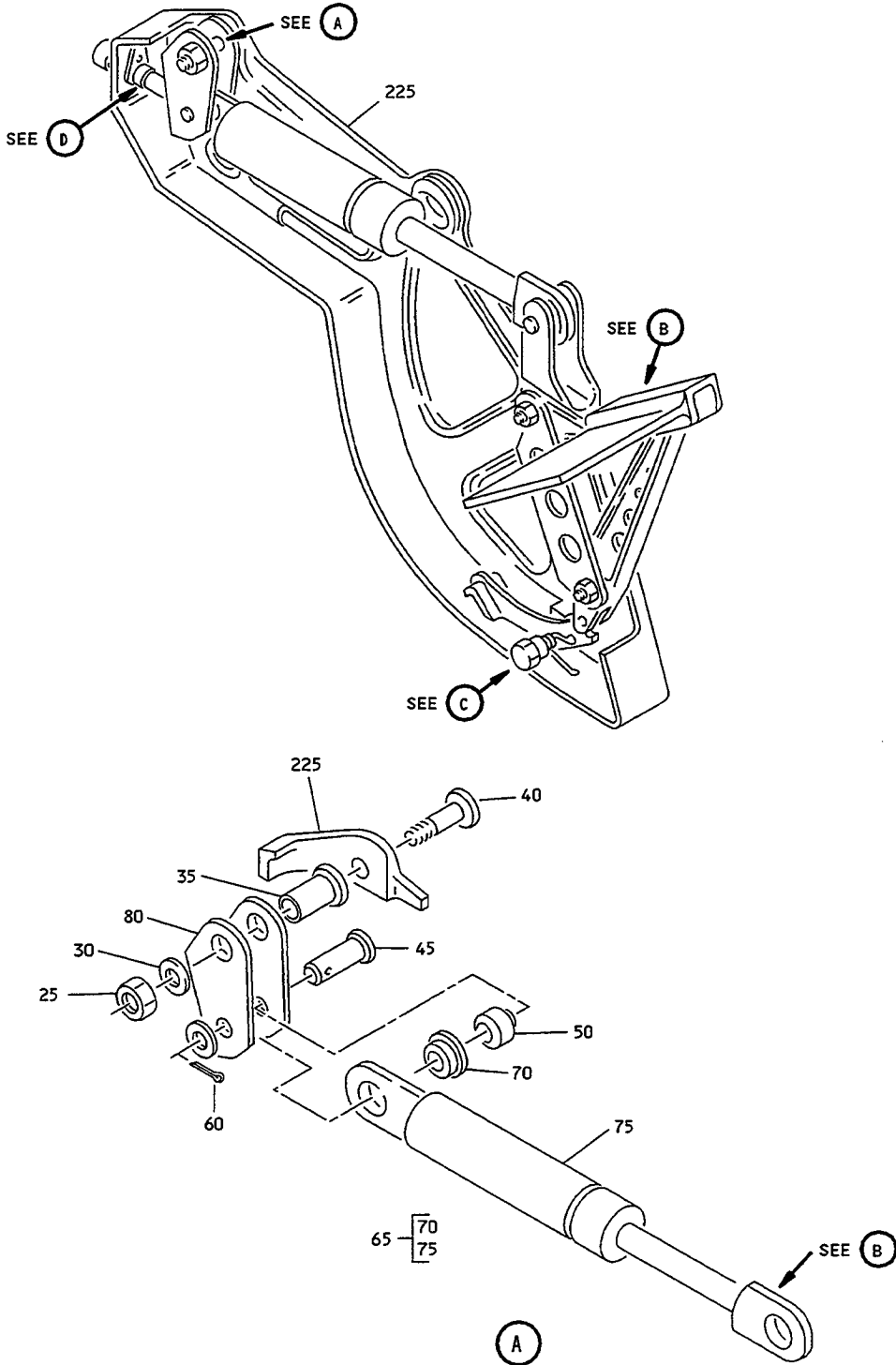
\*[2] BACB30PU3-7 bolts (190)

Fits and Clearances  
 Figure 601 (Sheet 3)

Mar 1/00

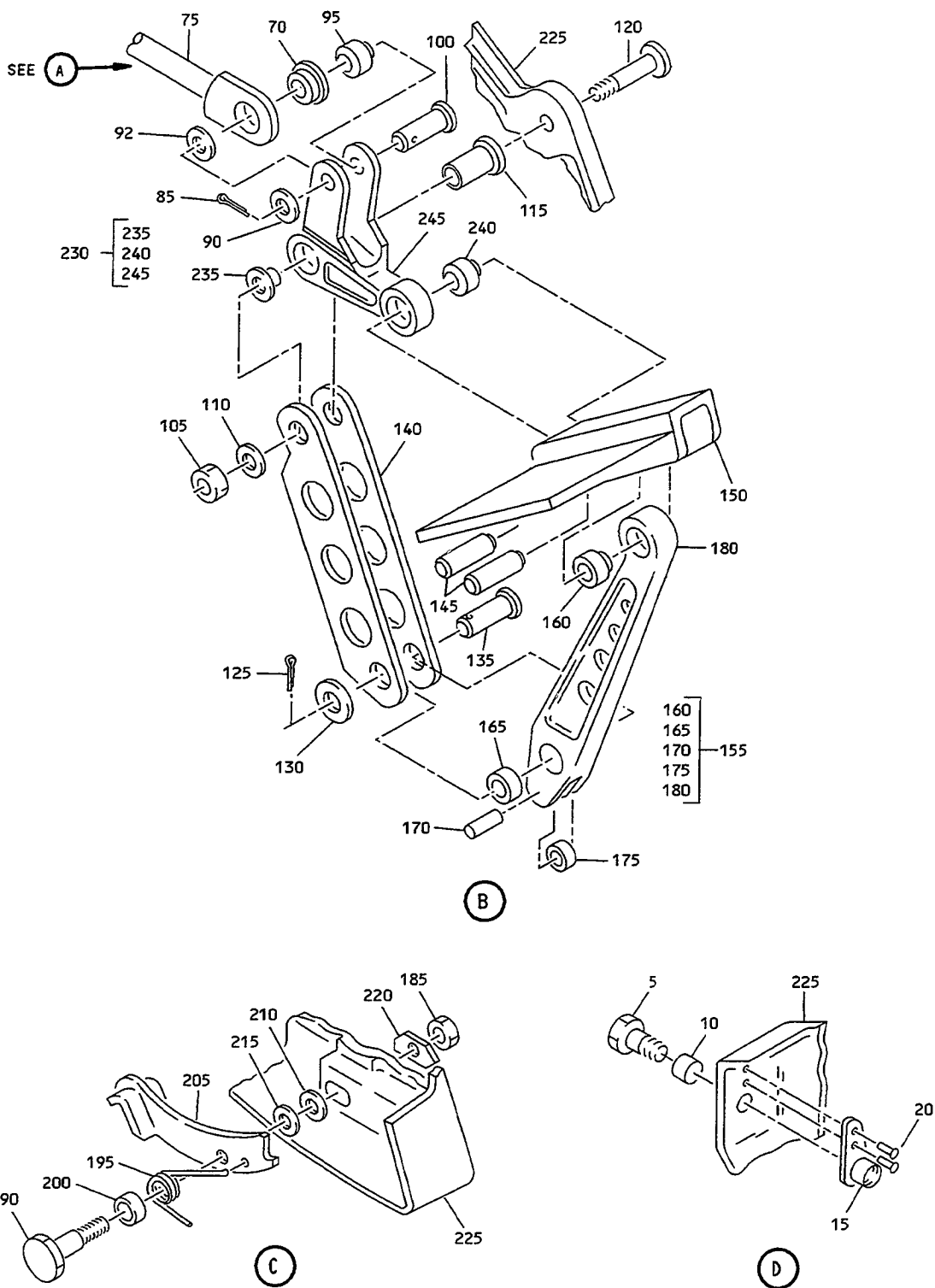
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ILLUSTRATED PARTS LIST



Carry-All Compartment Two-Link Articulated Hinge Assembly  
 Figure 1101 (Sheet 1)

**OVERHAUL MANUAL**



**Carry-All Compartment Two-Link Articulated Hinge Assembly  
 Figure 1101 (Sheet 2)**



FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101											
1	65C20204-21									A	RF
1	65C20204-22									B	RF
1	65C20204-23									C	RF
1	65C20204-24									D	RF
1	65C20204-25									E	RF
1	65C20204-26									F	RF
1	65C20204-27									G	RF
1	65C20204-28									H	RF
5	BACS12CB3-16										1
10	NAS43DD3-6										1
15	BACN10JP3B										1
15	BACN10JP3A										1
20	BACR15BB3D										2
25	BACN10JC3										1
30	BACW10P353S										1
35	BACB28X3C052										1
40	BACB30FN6-10										1
45	MS20392-2C17										1
50	BACB28Y3C037										1
55	AN960PD10L										1
60	MS24665-132										1
65	65C16855-3										1
70	BACB28W5B016										2
75	F0058										1
80	65C16891-59										1
85	MS24665-132										1
90	AN960PD10L										1
92	AN960PD516									E-H	1
95	BACB28Y3C037										1
100	MS20392-2C17										1
105	BACN10JC3										1
110	BACW10P353S										1
115	BACB28X3C052										1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101											
120	BACB30FN6-10										1
125	MS24665-132										1
130	AN960PD416L										1
130	NAS1149D0416J										1
135	MS20392-3C17										1
140	65C16891-60										1
145	MS16562-41										2
150	65C19936-1								ACEG		1
150	65C19936-2								BDFH		1
155	65C20204-13										1
160	ABG2										1
160	ABG2CR										1
160	ABG2SS										1
160	LHA2										1
160	LHSS2										1
160	LSS2										1
165	AA307-1										1
170	MS20253P2-035										1
175	N5042										1
175	13P003										1
180	65C19933-1										1
185	BACN10JC06										1
190	BACB30PF3-7										1
190	BACB30PU3-7										1
190	BACB30PW3-7										1
195	65C16897-5								ACEG		1
195	65C16897-6								BDFH		1
200	NAS43DD3-8										1
205	65C19935-3										1
210	BACW10P118S										1
215	NAS1515M3L										1
220	65C16851-24										1
225	65C19932-1								ACEG		1
225	65C19932-2								BDFH		1
225	65C19932-5								ACEG		1
225	65C19932-6								BDFH		1
230	65C20204-9								AB		1
230	65C20204-10								CD		1
230	65C20204-11								EF		1
230	65C20204-12								GH		1
235	AA401-25										1
240	ABG2										1
240	ABG2CR										1
240	ABG2SS										1
240	LHA2										1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101											
240	LHSS2		.	.	BEARING, SPHERICAL, V50294 (OPT)						1
240	LSS2		.	.	BEARING, SPHERICAL, V50294 (OPT)						1
245	65C19937-1		.	.	ARM (USED ON 65C20204-9)						1
245	65C19937-2		.	.	ARM (USED ON 65C20204-10)						1
245	65C19937-3		.	.	ARM (USED ON 65C20204-11)						1
245	65C19937-4		.	.	ARM (USED ON 65C20204-12)						1

\*[1] 65C19932-1 (OR -2 OPPOSITE) PLUS 65C16851-24 OPTIONAL TO 65C19932-5 (OR -6 OPPOSITE).

\*[2] SP-8252 (V24403), 646865 (V55787), C16-07672 (V5V313) OR P472 (V52861) ARE OPTIONAL TO 65C16855-3.

#### VENDORS

V08863	NYLOMATIC, 10 HEADLEY PLACE, FALLSINGTON, PENNSYLVANIA 19054-1401
V13764	MICRO PLASTICS, INC., HIGHWAY 178 N., FLIPPIN, ARKANSAS 72634
V24403	ENIDINE, INC., 7 CENTRE DR., ORCHARD PARK, NEW YORK 14127-2281
V50294	NEW HAMPSHIRE BALL BEARINGS, INC., 9730 INDEPENDENCE AVE., P.O. BOX 2515, CHATSWORTH, CALIFORNIA 91311-4323
V52861	AVM, HIGHWAY 76 E., P.O. BOX 729, MARION, SOUTH CAROLINA 29571
V55787	GAS SPRING CORP., 92 COUNTY LINE RD., COLMAR, PENNSYLVANIA 18915-9606
V5V313	SUSPA, INC., 3970 ROGER B. CHAFFEE BLVD., GRAND RAPIDS, MICHIGAN 49508-3440
V70417	CHRYSLER CORP., AMPLEX DIV., 6565 E. EIGHT-MILE RD., WARREN, MICHIGAN 48091-2949
V73134	HEIM, INC., INCOM INTERNATIONAL, INC., 60 ROUND RD., P.O. BOX 430, FAIRFIELD, CONNECTICUT 06430-5114