



OVERHAUL MANUAL

TO: ALL HOLDERS OF RATIO CHANGER ASSEMBLY OVERHAUL MANUAL, 27-67-11

REVISION NO. 25, DATED NOV 1/08HIGHLIGHTS

DESCRIPTION OF CHANGE	TOPICS AFFECTED												
	D & O	D / A s s y	C l e a n i n g	I n s p / C h k	R e p a i r	A s s y	F / C	T e s t	T / S h o o t i n g	S / T o o l s	S t o r a g e	I P L	L / O v e r h a u l
Add replacement instructions for fillers (21, 26)					X								

Nov 1/08

27-67-11
HIGHLIGHTS
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RATIO CHANGER ASSEMBLY

27-67-11

| BOEING P/N 65-46370-3 thru -21, -23, -24, -26, -27, -28

AIRLINE P/N

THE FOLLOWING DIRECTIVES APPLY TO THIS SUBJECT:

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVES	DATE DIRECTIVE INCORPORATED INTO TEXT
27-1017		PRR 30294 PRR 30386 PRR 30533 PRR 30533-2 PRR 30618 PRR 30042-7 PRR 31045 PRR 31171 PRR 31254 PRR 31706 PRR 32039 PRR 32121-10 PRR 32983 PRR 35005-72 PRR 38156 PRR 38274 PRR 38398	Nov 15/67 Nov 15/67 Nov 15/67 Nov 15/67 Nov 15/67 Mar 10/70 Mar 10/70 Mar 10/70 Mar 10/70 Mar 10/70 Dec 25/72 Dec 25/72 Jul 5/80 Jun 1/95 Jul 1/00 Jul 1/03 Jul 1/03

LIST OF EFFECTIVE PAGES

* Indicates pages revised, added or deleted in latest revision
 F Indicates foldout pages - print one side only

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2	Mar 5/84	1109	Nov 1/07		
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102	Nov 1/07	1111	Jul 1/08		
103	Jul 1/08	1112	Nov 1/07		
104	BLANK	1112A	Nov 1/07		
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202	BLANK	1112C	Nov 1/07		
301	Jul 1/03	1112D	Nov 1/07		
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402	Mar 5/87	1113	Nov 1/07		
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1102	Nov 1/07				



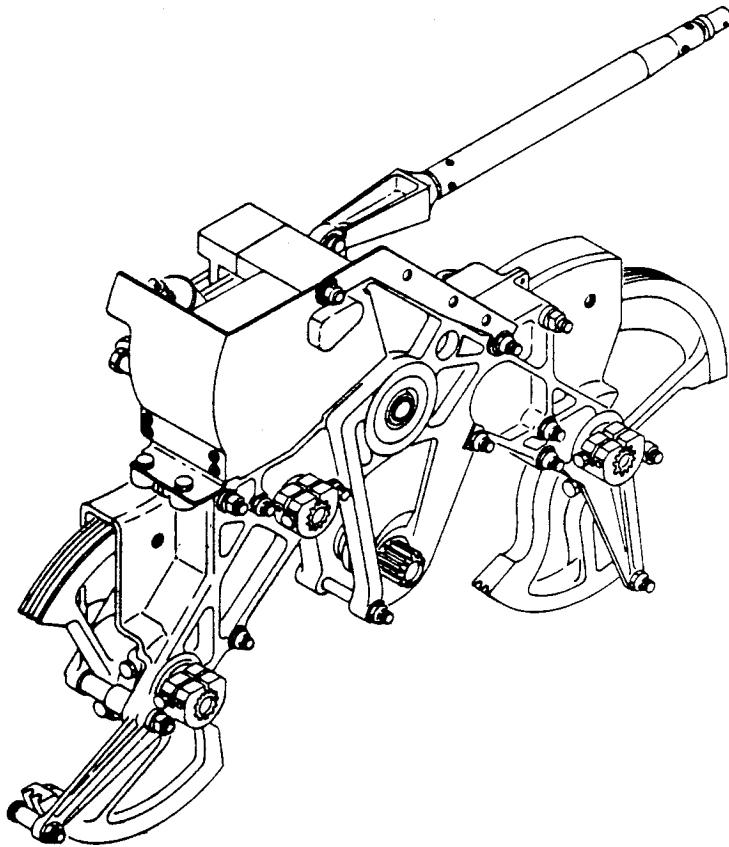
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| * [1] Use applicable procedures per 20-44-02 of the Standard Overhaul Practices Manual, D6-51702.

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RATIO CHANGER ASSEMBLY

Ratio Changer Assembly
Figure 1

DESCRIPTION AND OPERATION

1. Description

- A. The ratio changer assembly is a mechanical device which receives aileron and speed brake control inputs and transfers them to the spoiler mixer. For a given magnitude of input from the aileron system, the ratio changer varies the output to the mixer depending on speed brake lever setting. The mixer combines aileron and speed brake inputs to initiate spoiler action.

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- B. The assembly consists of forward and aft frames which house the aileron system input rod, speed brake input quadrants, a mechanical linkage system, connecting shafts to the spoiler mixer, and spoiler output quadrants. On some assemblies the speed brake input quadrants are integrated with a no-back clutch assembly to prevent movement of the speed brake system without inputs from the flight deck.

2. Operation

- A. The aileron system input rod is connected to a bellcrank which is slotted to receive two rollers. The bellcrank is connected by means of these rollers to a lever and a link. The lever is connected to a crank on the output shaft to the mixer. The link is attached to a fork lever which is connected to the speed brake quadrant by a two-part link.
- B. Rotation of the speed brake quadrant to raise the spoilers causes the rollers to move toward the bellcrank pivot. This reduces the magnitude of the lateral control input to the spoiler control system when speed brakes are used. The amount of reduction depends on the speed brake setting.

3. Leading Particulars (Approximate)

Length -- 26 inches
Height -- 16 inches
Width -- 6 inches
Weight -- 20 pounds

DISASSEMBLY

1. Procedure (Fig. 1101)

- A. Remove bolts (1, 5), washers (2, 3, 6), nut (4) and cover assembly (7).

NOTE: Do not disassemble cover assembly (7). Repair as a unit.

- B. Remove bolts (11), washers (12), nuts (13), and spacers (14).

- C. Remove bolts (16), washers (17), nuts (18), bracket assemblies (19 and 24), and strap assembly (97).

NOTE: Do not remove bearings (22) and sleeve (23) unless repair or replacement is necessary. Do not remove bonded fillers (21, 26) from brackets (20, 25).

- D. Remove bolts (27), washers (28), nuts (29), bracket (30), and on assemblies 65-46370-5 and -6, remove shield assembly (189).

- E. Remove collars (31), bushings (32), and lockbolts (33).

- F. Remove bolt (34), spacer (35), washer (36), and nut (37).

- G. Remove bolts (38, 38A, 42), spacers (15, 39, 43), washers (40, 44), nuts (41, 45), and on assemblies 65-46370-5 and -6, remove shield assembly (190).

- H. On assemblies 65-46370-3 thru -12, remove bolts (88), washers (89), nuts (90), speed brake quadrants (87), and spacer (91).

- I. Remove bolts (51, 55), washers (52, 56), nuts (53, 57), and output couplings (50, 54).

- J. Remove bolts (46, 46A, 46B), washers (47, 47A, 48), bushing (47B), and nuts (49).

- K. Remove aft frame assembly (172). On assemblies 65-46370-13 and -14 only, remove parts (127 thru 129) attaching input lever assembly (130) to output shaft of no-back assembly (191), then remove no-back assembly. Carefully lift all mechanism assemblies from forward frame assembly (179).

NOTE: Do not remove bolt (187), inserts (174, 181, 182) or bearings (175 through 178, and 183 through 186) from frames (173, 180) unless repair or replacement is necessary. Refer to manufacturer's instructions for servicing of no-back assembly.

L. Disassemble bolts (61, 66, 73, 78), washers (62, 67, 74, 79), nuts (63, 68, 75, 80), quadrant shafts (58, 70), washers (69, 81), spacers (59, 64, 71, 76), lower quadrants (60, 72), and upper quadrants (65, 77).

M. Remove cotter pin (85), bolt (82), washer (83), nut (84), and lever (86).

N. Remove lever shaft bolt assembly (93), bushing (96), washer (99), bearings (98), nut (100), cotter pin (101), output lever assembly (102) and input lever assembly (105).

NOTE: Do not remove bearings (104, 107) from levers (103, 106) or fitting (95) from bolt (94) unless repair or replacement is necessary.

O. Remove lockbolt (108) and collar (109) to remove input rod assembly (110) from input lever assembly (105).

NOTE: Do not remove rivets (114, 114A), clevis (113), or plug (112) from tube (111) unless repair or replacement is necessary.

P. Remove bolt (115), washer (116), nut (117), cotter pin (118), and link assembly (119).

NOTE: Do not remove bearings (121) from link (120) unless repair or replacement is necessary.

Q. Remove bolt (122), washer (123), bearing (124), nut (125), cotter pin (126), and speed brake input lever assembly (130).

NOTE: Do not remove bearings (132) from lever (131) unless repair or replacement is necessary.

R. On assys 65-46370-3 thru -12, remove bolt (127), washer (128), nut (129), and spacers (133). Pull shaft (92) out of lever assembly (130).

S. Remove collars (134), lever shaft bushings (135), lockbolts (136), and fork lever (143). Remove bearing (137) from link (142). On assemblies 65-46370-24, -26, -27 and -28 remove solid link (138E).

T. Remove bolts (152, 156), washers (153, 157), nuts (154, 158), cotter pins (155, 159), and lower scissor link (160).

- U. Remove bolts (167, 168), washer (169), spacer (170), collars (171), link assembly (139), link (142), and spacer (188). Remove bushings (163) from trunnion (161).

NOTE: Link assembly (139) and link (142) are a matched set and should be kept together after disassembly. Do not remove bearing (141) from link (140) unless repair or replacement is necessary.

- V. Remove pins (166), washers (164), and nuts (165) from link assembly (139) and link (142).

- W. On assemblies 65-46370-3 thru -21.

- (1) Remove nut (144) then tighten nut (145) until spring load is removed from cover (148). Cut lockwire and remove cover (148).
- (2) Slowly loosen nut (145) until all spring load is released, then remove nut (145), washers (146), retainers (149), optional laminated shim (149A), and springs (150, 151) from rod (147). 65-46370-21 may have optional configuration using five washers (146) and shim (146A), see IPL Fig. 1101, Detail F.

NOTE: Record thickness of laminated shim (149A) for use in assembly.

- X. On assembly 65-46370-23, remove nuts (144, 145), washers (146), spacers (150A, 150B) and optional laminated shim (149A) from rod (147).

NOTE: Record thickness of laminated shim (149A) for use in assembly.

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CLEANING

1. General

CAUTION: VAPOR DEGREASING OF TITANIUM PARTS IS RESTRICTED. REFER TO 20-30-03 FOR CLEANING OF TITANIUM AND TITANIUM ALLOY PARTS.

- A. Clean all parts except bearings using standard industry practices per 20-30-03.
- B. Clean and relubricate bearings using standard industry practices per 20-30-01.

INSPECTION/CHECK

1. Visual Check (See Fig. 1101.)

- A. Examine all metal parts for cracks, burrs, and corrosion, using strong light and a minimum 10-power magnification.
- B. Examine internal and external threads for cross-threading or stripping.
- C. Check all plated and painted surfaces for blistering or flaking.
- D. Check bearings for roughness, binding, and excessive radial or axial play.
- E. Examine bearing (22) and bushing (162) for cuts, nicks, grooves, and dents.

2. Special Check (See Fig. 1101.)

- A. If visual examination discloses evidence of defects in any of listed parts, perform the following checks:
 - (1) Magnetic particle check -- brackets (20, 25, 30), couplings (50, P/N 65-51681-1; 54, P/N 65-51681-2, -4), spacers (59, 71, P/N 69-40325-1), levers (86, P/N 65-52299-2, -4; 106), shafts (58, P/N 69-38907-1, -2; 70, P/N 69-38907-2, -4; 92, P/N 65-53599-1, -4), bolt (94), rod (147), retainer (149), springs (150 and 151), and pin (166).
 - (2) Fluorescent dye penetrant check -- couplings (50, P/N 65-51681-3; 54, P/N 65-51681-5), shafts (58, P/N 69-38907-5; 70, P/N 69-38907-6; 92, P/N 65-53599-7), spacers (59, 71, P/N 69-40325-2; 133, P/N 69-40324-4; 188), quadrants (60, 65, 72, 77, 87), levers (86, P/N 65-52299-3; 103, 131, 143), clevis (113), links (120, 140, 142, 160), solid link (138E), cover (148), spacer (150A), trunnion (161), and frames (173, 180).

B. Check springs (150 and 151) in accordance with figure 301.

NOTE: Springs shall not wobble when rolled on a flat surface. No permanent set shall result from test load.

Index No. Fig. 1101	Free Length (inches)	Test Length (inches)	Check Load (pounds)
150	1.72	1.60 1.37	65 to 85 208 to 228
151	2.03	1.60 1.37	65 to 85 105 to 125

Spring Check Data
Figure 301

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REPAIR
1. Repair (Fig. 1101)

- A. Repair minor defects using standard industry practices.
- B. If following parts are worn beyond limits shown in Fig. 601 rework as follows:
 - (1) Frames (173 and 180)
 - (a) Machine hole as required, within maximum dimension shown in Fig. 401.
 - (b) Install applicable aluminum alloy bushing (0.040-inch minimum wall thickness), (7075-T6, 7075-T73, 2024-T3 or 2024-T4) per 20-50-03, with bushing interference shown in Fig. 401.

Ref Letter Fig. 601	Maximum Hole Diameter (inches)	Bushing Interference (inches)	
		Minimum	Maximum
A, J	0.8693	0.0005	0.0010
B, K	1.4943	0.0010	0.0030
C, L	1.6210	0.0010	0.0030
D, M	1.4335	0.0010	0.0030

Rework of Frames
Figure 401

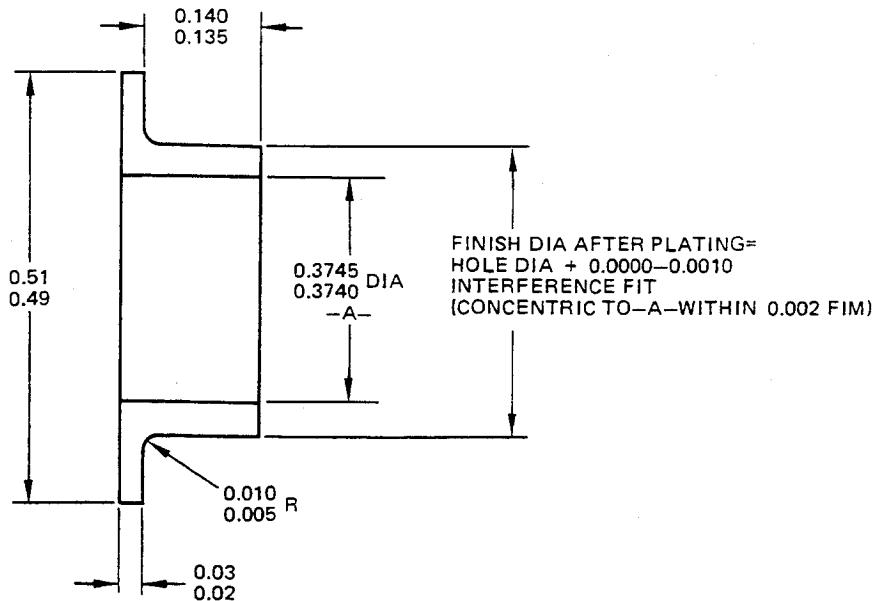
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(2) Trunnion (161)

- (a) Machine holes through trunnion as required, within 0.3800 inch maximum diameter.
- (b) Make two oversize bushings per Fig. 401A to replace bushings (163).
- (c) Install replacement bushings per 20-50-03.

(3) Lower scissor link (160).

- (a) Machine ID to 0.3300 inch maximum.
- (b) Install applicable steel bushing with 0.0010-0.0030 inch bushing interference.
- (c) Ream bushing to 0.250-0.251 -inch diameter.



CADMUM PLATE (F-4.201) EXCEPT ON ID
MATERIAL: AL-BRONZE OR AL-NI-BRONZE

BREAK SHARP EDGES
ALL DIMENSIONS ARE IN INCHES

Oversize Bushing Details
Figure 401A

2. Refinish (Fig. 1101)

NOTE: Refer to SOPM 20-30-02 for stripping of protective finishes and to SOPM 20-41-01 for decoding of F and SRF finish symbols and their BAC equivalents.

A. If painted or plated surfaces are worn or chipped, refinish listed items as indicated in following steps:

- (1) Cover assembly (7, P/N 65-51683-1) -- Manually apply colored film (F-2.940) to metal parts only, then apply one coat primer, BMS 10-11, Type 1 (SRF-12.205) all over.
- (2) Cover assembly (7, P/N 69-59043-1, -4, -7) -- Alodize or chromic acid anodize and apply BMS 10-11, Type 1 primer (SRF-2.30) all over, plus BMS 10-60, white enamel (SRF-14.9812) on 69-59043-4, -7.
- (3) Bolt (11), nut (13), bushings (32, 96, and 135), guard (38A), spacers (59, 69-40325-1; 71, 69-40325-1; 133, 69-40324-2), and washers (69 and 81) -- Cadmium plate (F-1.1926) all over.
- (4) Spacers (14 and 15) -- Chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.19) all over.
- (5) Bracket assemblies (19, 69-38901-1, -4) and (24, 69-40318-1) -- Apply one coat primer, BMS 10-11, Type 1 (F-20.02) all over except on bearings.
- (6) Bracket assemblies (24, 69-40318-4) -- Apply no finish (F-25.01).
- (7) Bracket (20, 69-38901-2) -- Cadmium-titanium plate and apply post-plate treatment (F-15.01) all over per SOPM 20-42-02. Material is 4340 bar, heat-treated to 180-200 ksi.
- (8) Bracket (25, 69-40318-2) -- Cadmium-titanium alloy plate (F-1.308) all over per SOPM 20-42-02. Material is 4340 bar, heat-treated to 180-200 ksi.
- (9) Bracket (25, 69-10318-5) -- Cadmium-titanium plate and apply post-plate treatment (F-15.01) and apply one coat primer, BMS 10-11, Type 1 (SRF-20.02) all over except in holes, per SOPM 20-42-02. Material is 4340 bar, heat-treated to 180-200 ksi.
- (10) Lever (86, 65-52299-2 and -4), and Bolt (94) -- Cadmium-titanium alloy plate (F-1.181) all over per SOPM 20-42-02. Material is 4340 bar, heat-treated to 180-200 ksi.
- (11) Filler (21) -- Chromic acid anodize (F-17.04) all over.
- (12) Filler (26, 69-40318-3) -- Chromic acid anodize (F-2.20) all over.
- (13) Filler (26, 69-40318-6) -- Chromic acid anodize or Boric acid - sulphuric acid anodize (F-17.31) all over, then apply one coat of BMS 10-11, Type 1 primer (F-20.02) all over.

- (14) Bracket (30) -- Cadmium-titanium alloy plate (F-1.181) all over and apply teflon-filled coating (SRF-14.9624) per Fig. 402. Material is 4340 bar, heat-treated to 180-200 ksi.
- (15) Couplings (50, P/N 65-51681-1, 54, P/N 65-51681-2, -4) -- Cadmium-titanium alloy plate (F-1.181) all over and apply one coat primer, BMS 10-11, Type 1 (SRF-12.205) except on splines.
- (16) Shafts (58, P/N 69-38907-1, -3; 70, P/N 69-38907-2, -4) -- Cadmium plate (F-1.1926) per Fig. 402. Material is CRES 17-4PH per AMS 5643, heat-treated to 180-200 ksi.
- (17) Spacers (59, P/N 69-40325-2; 64; 71, P/N 69-40325-2; 76; 133, P/N 69-40324-4) -- Chromic acid anodize (F-2.20) all over.
- (18) Quadrants (60, 65, 72, 77, 87) -- Chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.19) plus BMS 10-60 white enamel (SRF-14.9812) all over except no primer or paint on inside diameter of shaft holes.
- (19) Shafts (92, P/N 65-53599-1, -4), rod (147), retainer (149), and pins (166) -- Cadmium plate (F-1.1926) all over. Material is 4340 bar, heat-treated to 180-200 ksi.
- (20) Strap (97A) -- Alodize or chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.30) and BMS 10-60 white enamel (SRF-14.9812) all over.
- (21) Levers (103, 131, 143, P/N 65-75324-1, -3), clevis (113), and links (120, 140, 142, 160) -- Alodize or chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.30) all over except no primer in bearing holes.
- (22) Lever (106) -- Apply one coat primer, BMS 10-11, Type 1 (SRF-12.205) per Fig. 402. Material is 17-4PH per AMS 5643, heat-treated to 180-200 ksi.
- (23) Tube (111), lever (143, P/N 65-51679-1, 2, -3), and cover (148) -- Chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.19) all over except no primer in bearing holes.
- (24) Plug (112) -- Chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.19) all over except no primer on threads and on entire length of 0.625-inch diameter surface.
- (25) Springs (150, 151) -- Cadmium plate and apply two coats primer, color Y (SRF-1.92) all over.
- (26) Spacer (150A) -- Boric acid-sulfuric acid anodize or chromic acid anodize (F-17.31). Plus, apply BMS 10-11, Type 1 primer (F-20.03) except do not apply primer in the thru hole in center of spacer. Optional to the above -- chemical treat and apply BMS 10-11, Type 1 primer (F-18.06), except do not apply primer in the thru hole in the center of spacer.
- (27) Trunnion (161) -- Alodize or chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.30) all over per Fig. 402.

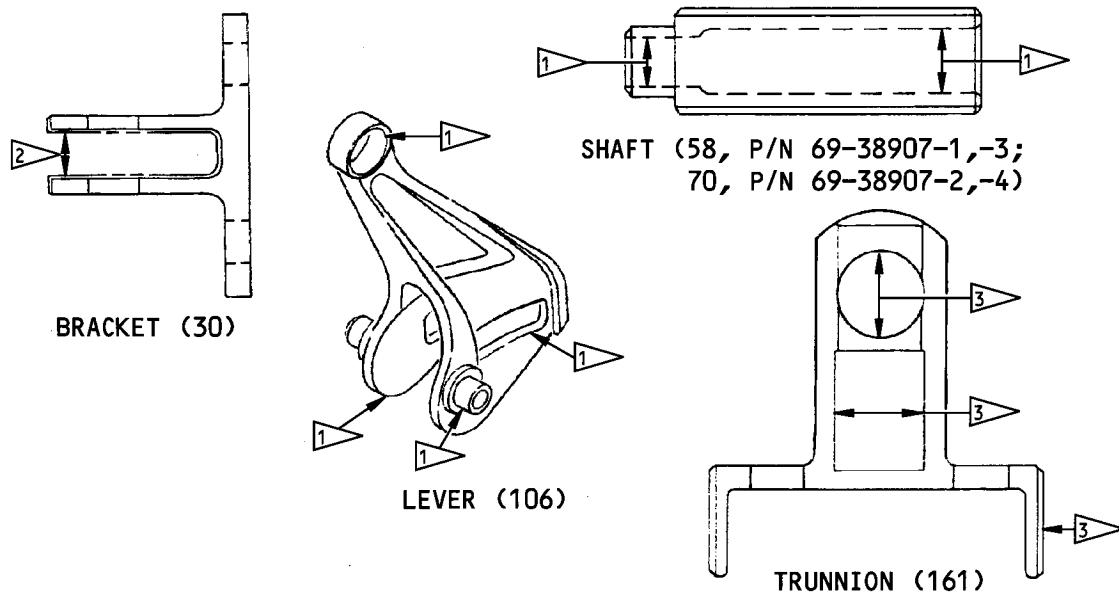
- (28) Bushing (163) -- Cadmium plate (F-4.201) except on inside diameter.
- (29) Frames (173 and 180) -- Sulfuric acid anodize (F-2.201) plus apply one coat primer, BMS 10-11, Type 1 (SRF-12.205) and BMS 10-60 white enamel (SRF-14.9812) all over except no paint or primer in bearing holes.
- (30) Spacer (188) -- Alodize or chromic acid anodize and apply one coat primer, BMS 10-11, Type 1 (SRF-2.30) all over except no primer in 0.165 and 0.3105-inch diameter holes.
- (31) Solid link (138E) -- Boric acid-sulfuric acid anodize or chromic acid anodize (F-17.31). Apply two coats of primer, BMS 10-11, Type 1 (F-20.03) all over except no primer in bearing holes.

3. Replacement

- A. Replace all parts damaged beyond simple repair.
- B. Replace all lockwire and cotter pins at each overhaul.
- C. If bearings require replacement, press out bearing and install new bearing per SOPM 20-50-03. Install bearings (22, 104, 107, 121, 132, 141, 175, 176, 177, 178, 183, 184, 185, and 186) with wet BMS 10-11, Type 1 primer.
- D. If bushing (163) is worn beyond limits stated in Fig. 601 replace with applicable oversize bushing.
- E. If fillers (21, 26) are separated from brackets (20, 25), install filler (21, 26) into bracket (20, 25) per SOPM 20-50-12, Type 38.

4. Materials

- A. Primer -- BMS 10-11, Type 1
- B. Corrosion Preventive Compound -- MIL-C-11796, class 3



- [1] NO FINISH ON THESE SURFACES
- [2] APPLY SRF-14.9624 ON THESE SURFACES
- [3] NO PRIMER ON THESE SURFACES

Refinish Diagram
Figure 402

ASSEMBLY

1. General

- A. Insert bearings (124, 137, 142A) with corrosion preventive compound, MIL-C-11796 applied to outer race contact surfaces. Install pins (166) with corrosion preventive compound applied to all contact surfaces.
- B. Apply grease, MIL-G-23827, to all mating splines prior to assembly. Apply grease to all contact surfaces of parts specified in assembly instructions below.
- C. Any combination of light and regular washers may be used to align slots in castellated nuts with holes in bolts, to permit installation of cotter pins.
- | D. On assemblies 65-46370-24, -26, -27, -28, install bearing (138F) with sealant BMS 5-95 and roller swage housing over bearing per SOPM 20-50-03.

2. Assemble Ratio Changer (Fig. 1101)

- A. Attach pins (166) to link assembly (139) and link (142), using washers (165) and nuts (164).
- B. Place bushings (163) in trunnion (161). Insert trunnion into spacer (188) and attach link assembly (139) and link (142) with bolts (167, 168), washer (169), spacer (170), and collars (171).
- C. Insert teflon bushing (162) and bungee rod (147) into trunnion (161). Insert bearing (142A) into link (142). Connect lower scissor link (160) to upper scissor assembly (138) and bungee rod, with bolts (152, 156), washers (153, 157), and nuts (154, 158). Tighten nuts to 12-15 lb-in. and install cotter pins (155, 159).
- D. Assemble bungee assembly for assemblies 65-46370-3 thru -21 (Fig. 501)
 - (1) Place retainers (149), springs (150, 151), washers (146), and nut (145) on bungee rod (147). On assembly 65-46370-21 use five washers (146) and shim (146A) when using optional cover (148, P/N 69-49784-2). Tighten nut (145) until distance "A" between upper surface of upper retainer and lower face of trunnion (161) is approximately 1.70 in. Check spring preload and adjust nut (145) as necessary to obtain spring preload of 260-285 lb for 65-46370-3 thru -20 or 360-385 lb for 65-46370-21.
 - (2) Screw cover (148) to finger-tight contact with upper retainer (149). Loosen nut (145) and rotate cover minimum amount to align a slot with lockwire hole in trunnion (161). Check that spring load is still 260-285 lb. Install lockwire and seal. Check that seal clears all components through total travel.
 - (3) Tighten nut (145) finger-tight against upper retainer (149) and install nut (144). Hold nut (145) while tightening nut (144) to 14-20 lb-in.

- (4) Check that there is no free play between cover (148) and nut (145), or equivalently, that there is no relative movement between upper scissors link assembly (138) and lower scissors link (160). Adjust nuts (144, 145), as necessary, per step (3) above to eliminate movement.

E. Assemble bungee assembly 65-46370-23 (Fig. 1101).

- (1) Place spacers (150A, 150B), washers (146), nuts (144, 145) on bungee rods (147).
- (2) Screw cover (148) to finger-tight contact with spacer (150B). Rotate cover minimum amount to align a slot with lockwire hole in trunnion (161). Install lockwire and seal. Check that seal clears all components through total travel.
- (3) Tighten nut (145) finger-tight against spacer (150A) and install nut (144). Hold nut (145) while tightening nut (144) to 14-20 lb-in.
- (4) Check that there is no free play between cover (148) and nut (145), or equivalently, that there is no relative movement between upper scissors link assembly (138) and lower scissors link (160). Adjust nuts (144, 145), as necessary per step (3) above to eliminate movement.

F. On assemblies 65-46370-3 thru -21, -23, insert bearing (137) into upper scissor link assembly (138). On assemblies 65-46370-24, -26, -27, -28, insert bearing (138F) into solid link assembly (138). Install bushings (135) in fork lever (143) with grease, then attach fork lever to link assembly with lockbolts (136) and collars (134).

G. Insert bearing (124) in lever (130). Apply grease to wear surfaces of lever. Attach lever to lower scissor link (160) on assemblies 65-46370-3 thru -21, -23 or solid link (138E) on assemblies 65-46370-24, -26, -27, -28, with bolt (122), washers (123), and nut (125). Tighten nut to 12-15 lb-in. and install cotter pin (126).

H. Connect link assembly (119) to fork lever (143) with bolt (115), washer (116), and nut (117). Tighten nut to 12-15 lb-in. and install cotter pin (118).

I. Assemble rod assembly (110) by installing clevis (113) onto tube (111) with wet BMS 10-11 (F-20.06).

J. Apply grease to wear surfaces of lever (105) and attach rod assembly (110) to lever with lockbolt (108) and collar (109).

K. Apply grease to bushing (96) and bolt (93) or bolt assembly (93). Insert bushing into output lever (102). Attach output lever and lever (105) to link (119) with bolt or bolt assembly (93), bearings (98), washer (99), and nut (100). Tighten nut to 12-15 lb-in. and install cotter pin (101). Bend ends of cotter pin back around nut and push into slots in nut. Fill lubrication fitting in bolt assembly (93) with grease and remove excess.

L. Apply grease to wear surfaces of lever (86). Connect lever to lever (102) with bolt (82), washer (83), and nut (84). Tighten nut to 50-70 lb-in. and install cotter pin (85).

M. Install bolt (187), if replaced, into frame (180) and insert (181).

- N. On assemblies 65-46370-3 thru -12, -15, install linkage as follows:
 - (1) Apply grease to shaft (92) and insert shaft into input lever assembly (130). Align center groove in shaft with bolt (127), washer (128), and nut (129). Tighten nut to 50-70 lb-in.
 - (2) Apply grease to spacers (91, 133) and quadrants (87). Position spacer (91) and quadrants on shaft (92). Install bolts (88), washers (89), and nuts (90). Tighten nuts to 50-70 lb-in.
 - (3) Insert one spacer (133) into bearing (183) in forward frame assembly (179). Install assembled parts into frame by inserting shaft (92) into spacer (133), lug on lever (105) into bearing (185), and lug on lever (86) into bearing (184).

- O. On assemblies 65-46370-13 and -14, -16 thru -19, -20, -21, -23, -24, -26, -27, -28, install linkage as follows:
 - (1) Apply grease MIL-PRF-23827 to spacers (133) and splined shaft of no-back assembly (191). Insert one spacer into bearing (183) in forward frame assembly (179), then insert no-back assembly shaft from opposite side of frame.
 - (2) Install assembled linkages into frame by positioning input lever assembly (130) on no-back shaft and inserting lug on lever (105) into bearing (185) and lug on lever (86) into bearing (184).
 - (3) Install bolt (127), washer (128), and nut (129) to secure input lever to no-back shaft. Tighten nut to 50-70 lb-in.

- P. Apply grease to contact surfaces of quadrants (60, 65, 72, 77), shafts (58, 70), spacers (64, 76). Position quadrants (65, 77) on shafts (58, 70). Install spacers (64, 76) and position quadrants (60, 72) on quadrant shafts. Install bolts (61, 66, 73, 78), washers (62, 67, 74, 79), and nuts (63, 68, 75, 80). Tighten nuts to 50-70 lb-in.

- Q. Install washers (69, 81) on quadrant shafts (58, 70) and insert shafts into bearings (185) in forward frame (179).

- R. Place spacers (14, 15) in position. Apply grease to spacers (59, 71) and install on quadrant shafts. Install remaining spacer (133) on shaft (92) or no-back assembly (191). Install aft frame (172).

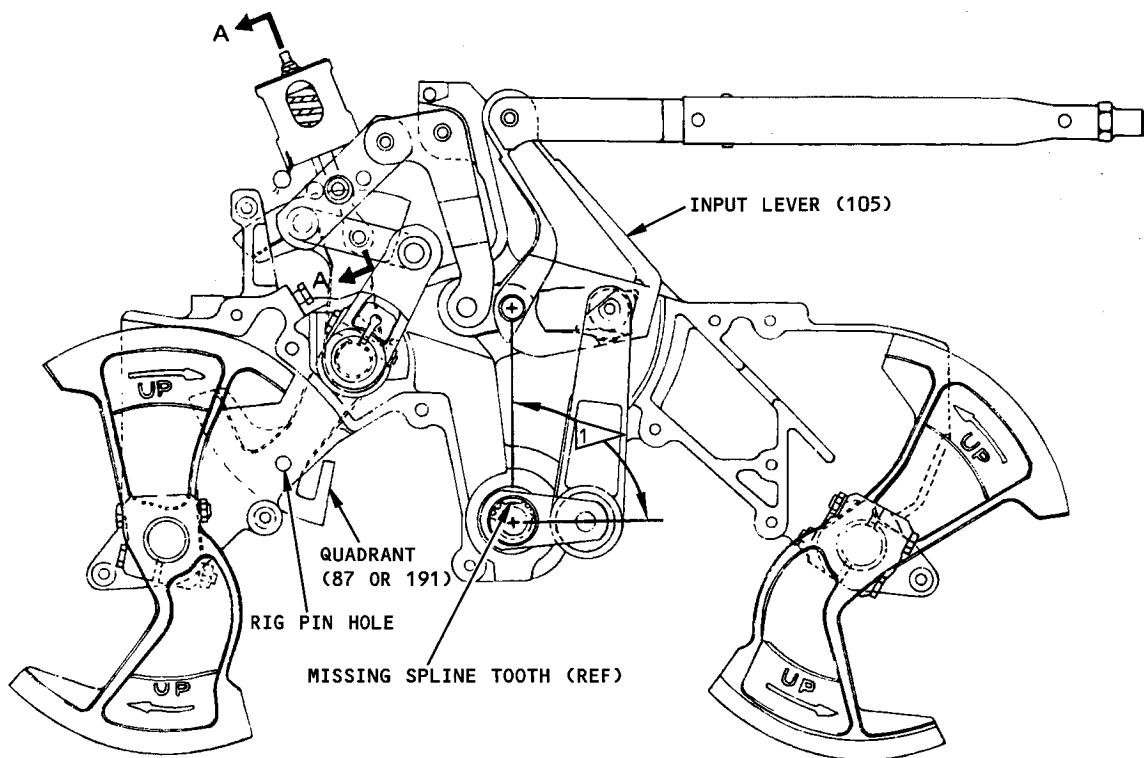
- S. Install spacer (43) on bolt (42) and insert bolt through forward frame (179), spacer (15), aft frame (172), and shield (190). Install washer (44), and nut (45) and tighten nut to 50-70 lb-in.

- T. Install bolts (46, 46A, 46B), washers (47, 47A, 48), bushing (47B), and nuts (49). Install bolts (38, 38A), spacers (39), washers (40), and nuts (41). Install bolt (34), spacer (35), washer (36), and nut (37). Tighten nuts (37, 41, 49) to 50-70 lb-in.

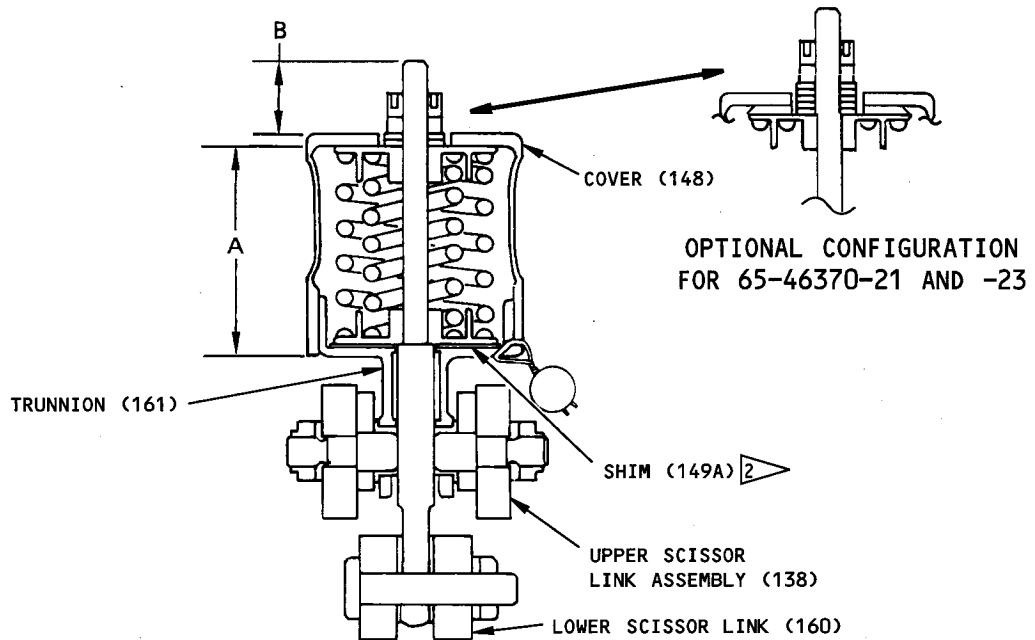
- U. Attach bracket (30) to forward frame (179) and shield (189) to aft frame (172) with bolts (27), washers (28), and nuts (29). Attach bracket (19) to forward frame (179) and shield (190) to aft frame (172) with bolt (16), washer (17), and nut (18).

- V. Attach bracket (24) to forward frame (179) and strap (97) to aft frame (172) with bolt (16), washer (17), and nut (18). Install bolt (11) through remaining holes in brackets (19, 24) and install washers (12), as required, and nuts (13). Install bolts (11) through forward frame (179), spacers (14), and aft frame (172) and install washers (12) and nuts (13). Tighten nuts (18, 29) to 50-70 lb-in. Tighten nuts (13) to 40-50 lb-in.
- W. Apply grease to bushings (32). Insert bushings into fork lever (143). Connect fork lever to frames (172, 179) with lockbolts (33), and collars (31).
- X. Check assembly rig position (Fig. 501).
 - (1) Adjust bolt (187) to obtain 37 degrees 45 minutes to 38 degrees rotation of lever (130).
 - (2) Insert 5/16-in. dia rig pin through speedbrake quadrant (87), or no-back (191) quadrant and frame assemblies (172, 179).
 - (3) Rotate input lever (105) to align centerline of missing spline tooth on lever (86) within one degree of line connecting centers of rotation of the two levers (105, 86). Hold levers in position.
 - (4) Remove rig pin and rotate speedbrake or no-back quadrant clockwise (as viewed from aft frame) until rollers (98) are at or very near the ends of slots (away from pivot) in input lever (105).
 - (5) With rollers held in position, check that rig pin can be inserted through quadrant and frames and removed freely, without binding.
 - (6) If rig pin cannot be inserted and removed freely, for assemblies 65-46370-3 thru -21, adjust scissor spring assembly as follows:
 - (a) Measure dimension "B" from top of bungee rod (147) to upper surface of cover (148).
 - (b) Remove cover (148) and measure distance "A" (Ref step D.(1)).
 - (c) Remove nuts (144, 145), washers (146), retainers (149), and springs (150, 151). Install laminated shim (149A) in trunnion (161) then install lower retainer (149).
 - (d) With shim tightly sandwiched between lower retainer and trunnion, and with retainer firmly seated on shoulder of bungee rod (147), adjust shim thickness until rig pin holes are aligned.
 - (e) Reassemble bungee assembly parts (144 through 151) per par. D except increase dimension "A" by shim thickness. Check that dimension "B" is unchanged and that spring preload is still 260-285 lb.
 - (7) If rig pin cannot be inserted and removed freely, for assembly 65-46370-23, adjust as follows:
 - (a) Remove cover (148), nuts (144, 145), washers (146) and spacers (150A, 150B). Install laminated shim (149A) in trunnion (161). Adjust shim thickness until rig pin holes align.

- (b) Reassemble bungee assembly parts (144 thru 150B).
- Y. Fasten cover (7) and upper end of strap (97) to frames (172, 179) with bolt (1), washers (2, 3), and nut (4). Install bolts (5) and washers (6). Tighten nut (4) to 50-70 lb-in.
- Z. Attach coupling (50) to shaft (92) or no-back assembly (191) with bolt (51), washer (52), and nut (53). Attach couplings (54) to quadrant shafts (58, 70) with bolts (55), washers (56), and nuts (57). Tighten nuts (53, 57) to 50-70 lb-in.
3. Materials
- A. Corrosion Preventive Compound -- MIL-C-11796, class 3 (SOPM 20-60-03)
 - B. Grease -- BMS 3-33 or MIL-G-23827 (SOPM 20-60-03)
 - C. Primer -- BMS 10-11, Type 1, Grade A (SOPM 20-41-02)



1 △ 93° 38' WHEN MISSING TOOTH IS ALIGNED (REF)



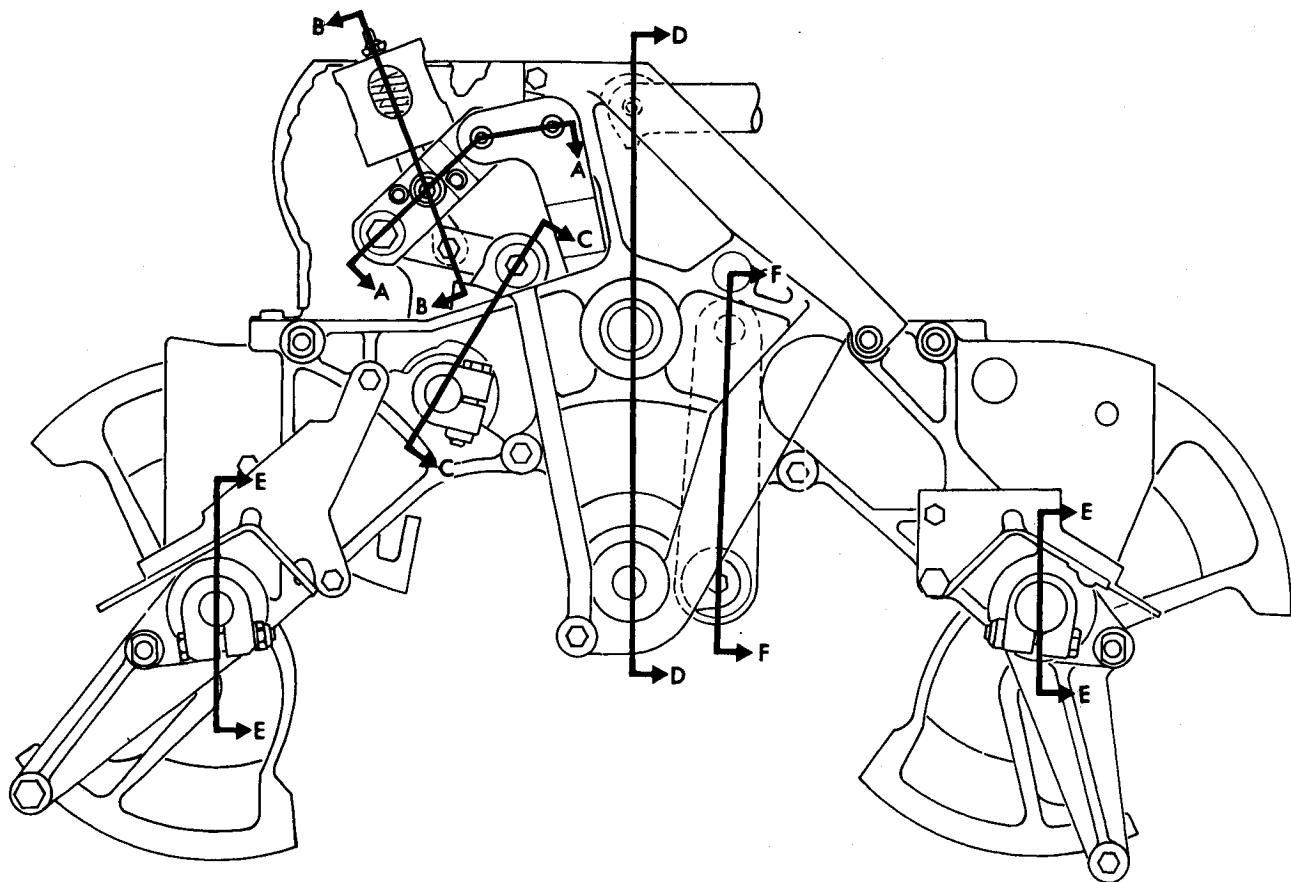
2 △ OPTIONAL

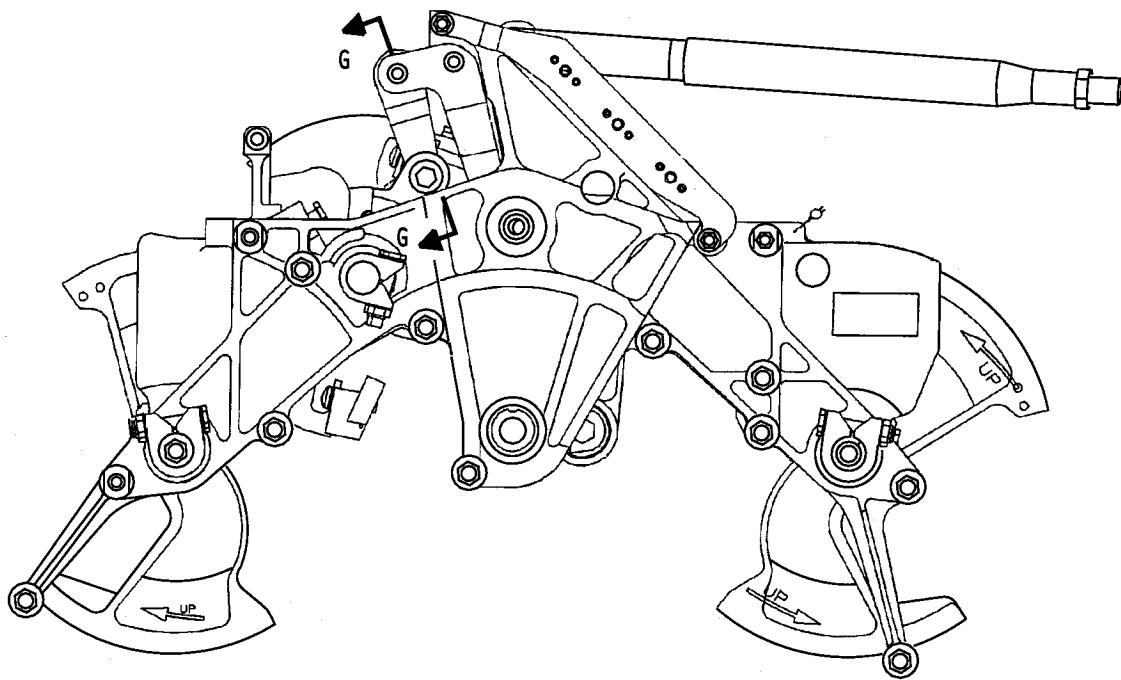
FOR 65-46370-3 THRU -21 AND -23

Assembly Rigging Details
Figure 501

FITS AND CLEARANCES

1. The fits and clearances table lists design dimensions and service wear limits for close tolerance parts of the assembly that are subject to wear or corrosion. Unless otherwise specified, parts should be returned to the design dimensions whenever rework is accomplished.
2. Clearances are given to aid assembly of the components. The values given in the Maximum Allowable Clearance column are the maximum permitted to ensure proper functioning of the unit. If assembled parts fail to meet this requirement, one or more of the parts must be rejected. Parts that are rejected should be reworked if within the rework limits given in the Repair procedure; if not within rework limits, the parts should be scrapped. It is recommended that the design clearances be used as the guiding assembly criteria when newly reworked parts are assembled.





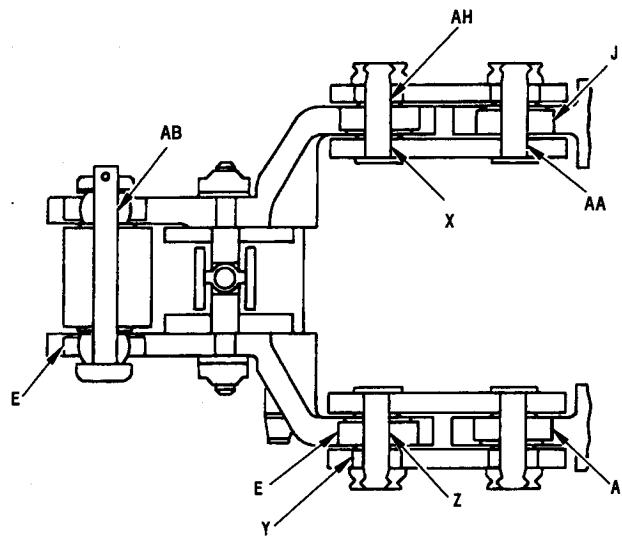
ASSEMBLIES 65-46370-24,-26,-27,-28

Fits and Clearances
Figure 601 (Sheet 1A)

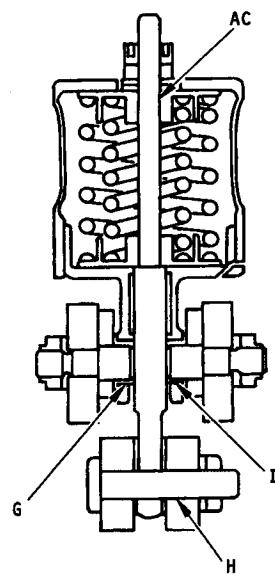
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SECTION A-A

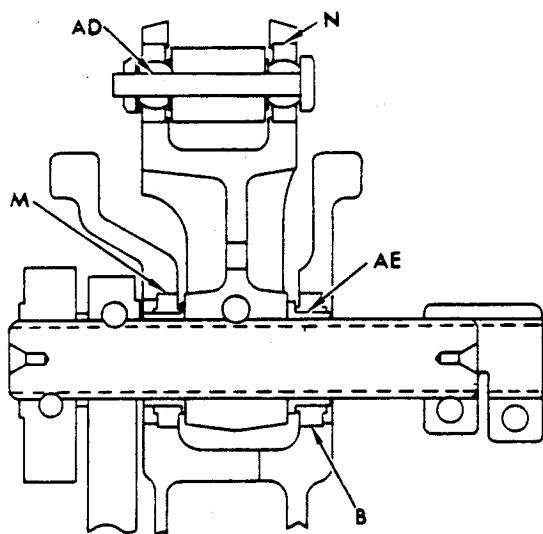


SECTION B-B

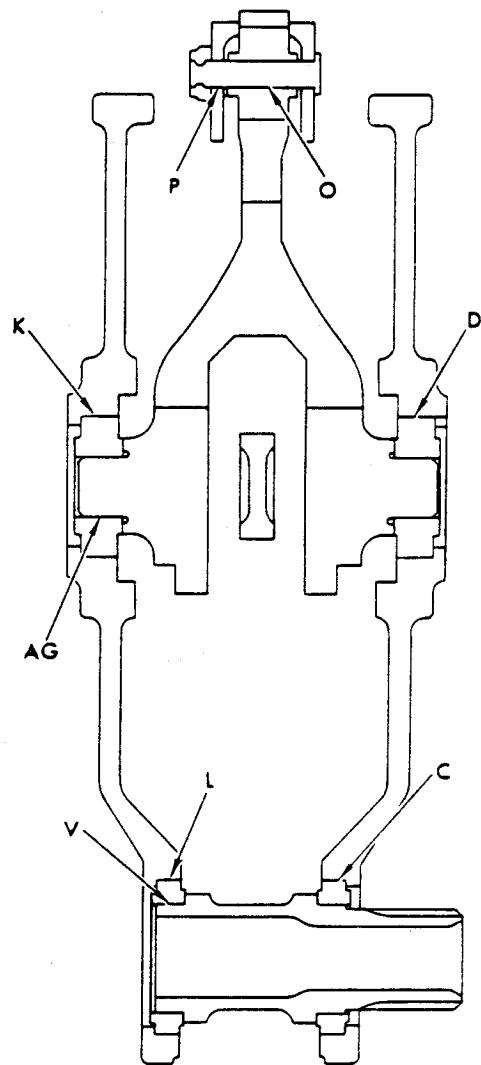
Fits and Clearances
 Figure 601 (Sheet 2)

65-46370

BOEING
COMMERCIAL JET
OVERHAUL MANUAL



SECTION C-C



SECTION D-D

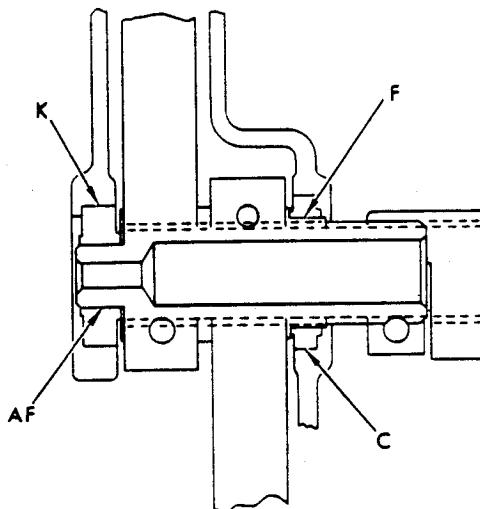
Jun 10/71

Fits and Clearances
Figure 601 (Sheet 3)

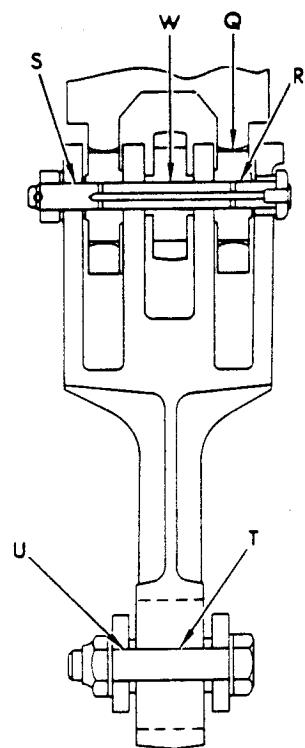
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BOEING 
COMMERCIAL JET
OVERHAUL MANUAL

65-46370

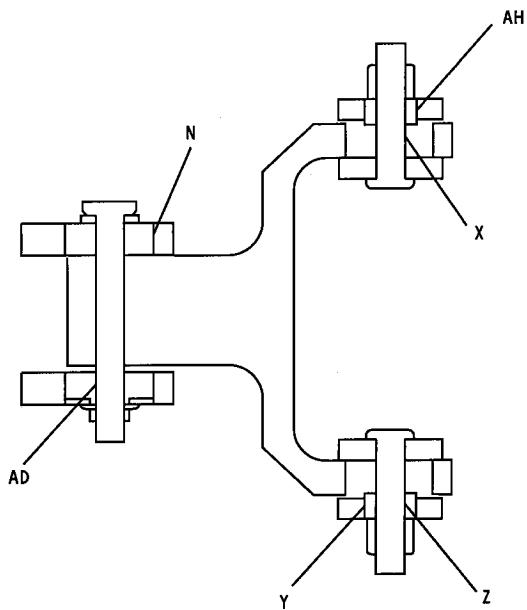


SECTION E-E



SECTION F-F

Fits and Clearances
Figure 601 (Sheet 4)



SECTION G-G
ASSEMBLIES 65-46370-24,-26,-27,-28

Fits and Clearances
Figure 601 (Sheet 4A)

Ref Letter Fig. 601	Mating Item No. Fig. 1101	Design Dimensions				Service Wear Limits		
		Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inch)		Maximum Allowable Clearance (inch)
		Min	Max	Min	Max	Min	Max	
A	ID 173	0.7488	0.7493	0.0012	0.0002	0.7495	0.7525	0.0030
	OD 178	0.7495	0.7500					
B	ID 173	1.3736	1.3743	*[1] 0.0014	*[1] 0.0002	1.3745	1.3785	0.0040
	OD 177	1.3745	1.3750					
C	ID 173	1.500	1.5010	0.0000	0.0020	1.4990	1.5030	0.0040
	OD 176	1.4990	1.5000					
D	ID 173	1.3125	1.3135	0.0000	0.0020	1.3115	1.3155	0.0040
	OD 175	1.3115	1.3125					
E	ID 140 (65-46368-5)	0.7500	0.7508	0.0000	0.0013	0.7495	0.7525	0.0030
	OD 141	0.7495	0.7500					
F	ID 176	1.0618	1.0632	*[1] 0.0013	*[1] 0.0004	1.0622	1.0632	0.0010
	OD 59,71	1.0628	1.0631					
G	ID 161	0.3735	0.3740	*[1] 0.0010	0.0000	0.3735	0.3750	0.0015
	OD 163	0.3740	0.3745					
H	ID 160	0.2500	0.2510	0.0005	0.0025	0.2481	0.2540	0.0059
	OD 152	0.2485	0.2495					
I	ID 163	0.3115	0.3125	0.0005	0.0025	0.3090	0.3140	0.0050
	OD 166	0.3100	0.3110					
J	ID 180	0.7488	0.7493	*[1] 0.0012	*[1] 0.0002	0.7495	0.7495	0.0000
	OD 186	0.7495	0.7500					

 Fits and Clearances
 Figure 601 (Sheet 5)

Ref Letter Fig. 601	Mating Item No. Fig. 1101	Design Dimensions				Service Wear Limits			
		Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inch)		Maximum Allowable Clearance (inch)	
		Min	Max	Min	Max	Min	Max		
K	ID 180	1.3736	1.3743	*[1] 0.0014	*[1] 0.0002		1.3745	0.0000	
	OD 185	1.3745	1.3750			1.3745			
L	ID 180	1.5000	1.5010		0.0000	0.0020		1.5030	0.0040
	OD 184	1.4999	1.5000				1.4990		
M	ID 180	1.3125	1.3135		0.0000	0.0020		1.3155	0.0040
	OD 183	1.3115	1.3125				1.3115		
N	ID 131	0.7500	0.7508		0.0000	0.0013		0.7525	0.0030
	OD 132	0.7495	0.7500				0.7495		
O	ID 107	0.2495	0.2500		0.0000	0.0015		0.2500	0.0019
	OD 108	0.2485	0.2495				0.2481		
P	ID 113	0.2495	0.2505		0.0000	0.0025		0.2515	0.0035
	OD 108	0.2480	0.2495				0.2480		
Q	ID 106	0.8770	0.8790		0.0020	0.0050		0.8840	0.010
	OD 98	0.8740	0.8750				0.8740		
R	ID 98	0.2493	0.2500	*[1] 0.0002		0.0015		0.2500	0.0015
	OD 94	0.2485	0.2495				0.2485		
S	ID 103	0.2495	0.2505		0.0000	0.0020		0.2540	0.0059
	OD 94	0.2485	0.2495				0.2481		
T	ID 104	0.3120	0.3125		0.0000	0.0015		0.3125	0.0019
	OD 82	0.3110	0.3120				0.3106		

Fits and Clearances
Figure 601 (Sheet 6)

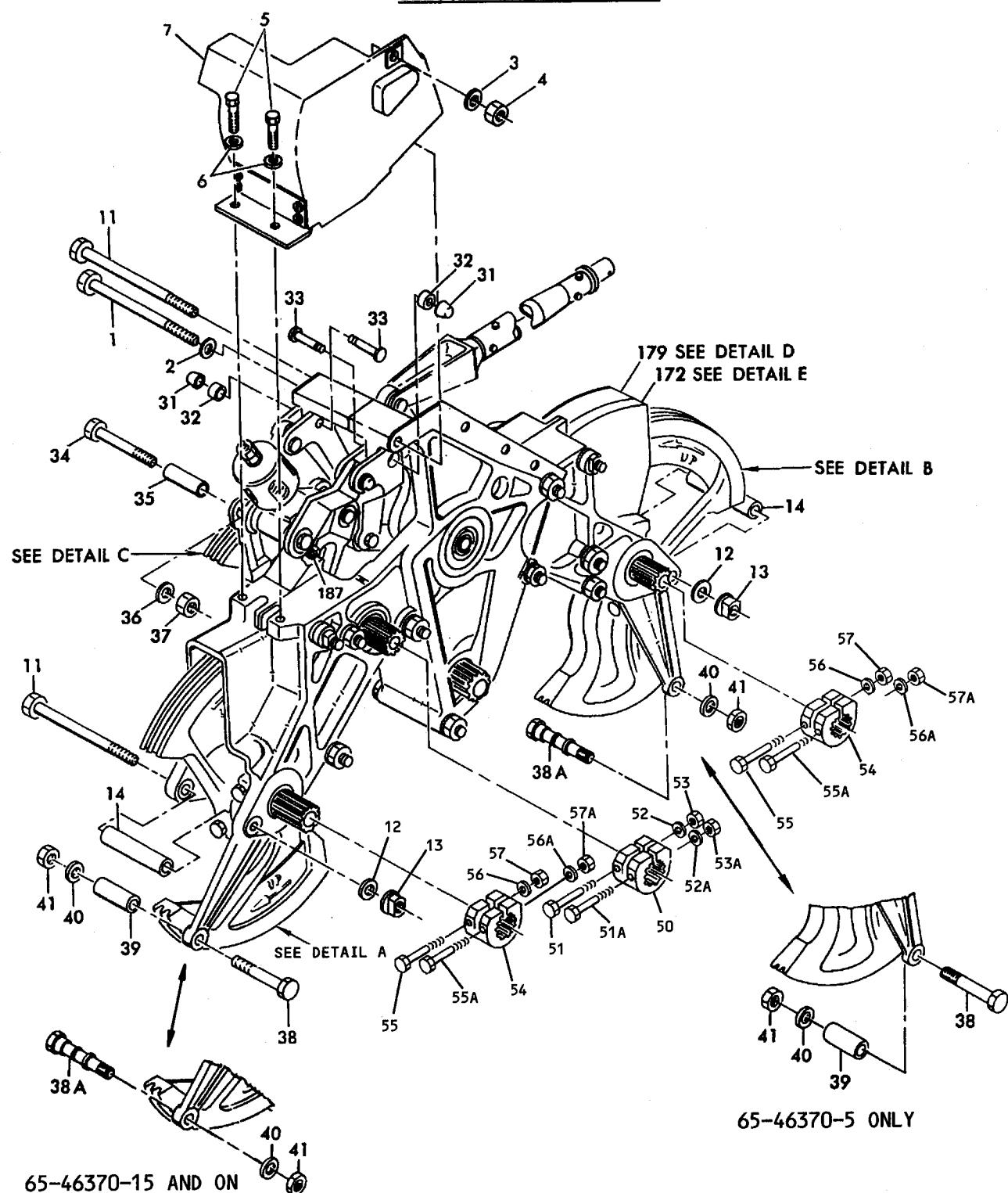
Ref Letter Fig. 601	Mating Item No. Fig. 1101	Design Dimensions				Service Wear Limits		
		Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inch)		Maximum Allowable Clearance (inch)
		Min	Max	Min	Max	Min	Max	
U	ID 86	0.3120	0.3130	0.0000	0.0020	0.3106	0.3165	0.0059
	OD 82	0.3110	0.3120					
V	ID 184	1.0625	1.0632	0.0003	0.0014	1.0614	1.0632	0.0018
	OD 86	1.0618	1.0622					
W	ID 121	0.2495	0.2500	0.0000	0.0015	0.2481	0.2500	0.0019
	OD 94	0.2485	0.2495					
X	ID 143	0.2495	0.2505	0.0000	0.0020	0.2481	0.2540	0.0059
	OD 136	0.2485	0.2495					
Y	ID 143	0.4370	0.4380	0.0000	0.0020	0.435	0.439	0.0040
	OD 135	0.4360	0.4370					
Z	ID 141, 138F	0.2495	0.2500	0.0000	0.0020	0.2480	0.2500	0.0020
	OD 136	0.2480	0.2495					
AA	ID 143	0.2495	0.2505	0.0000	0.0025	0.2480	0.2540	0.0060
	OD 33	0.2480	0.2495					
AB	ID 141	0.2495	0.2500	0.0000	0.0015	0.2481	0.2500	0.0019
	OD 156	0.2485	0.2495					
AC	ID 149	0.1900	0.1910	0.0010	0.0025	0.1880	0.1920	0.0040
	OD 147	0.1885	0.1890					
AD	ID 132	0.2495	0.2500	0.0000	0.0015	0.2481	0.2500	0.0019
	OD 122	0.2485	0.2495					

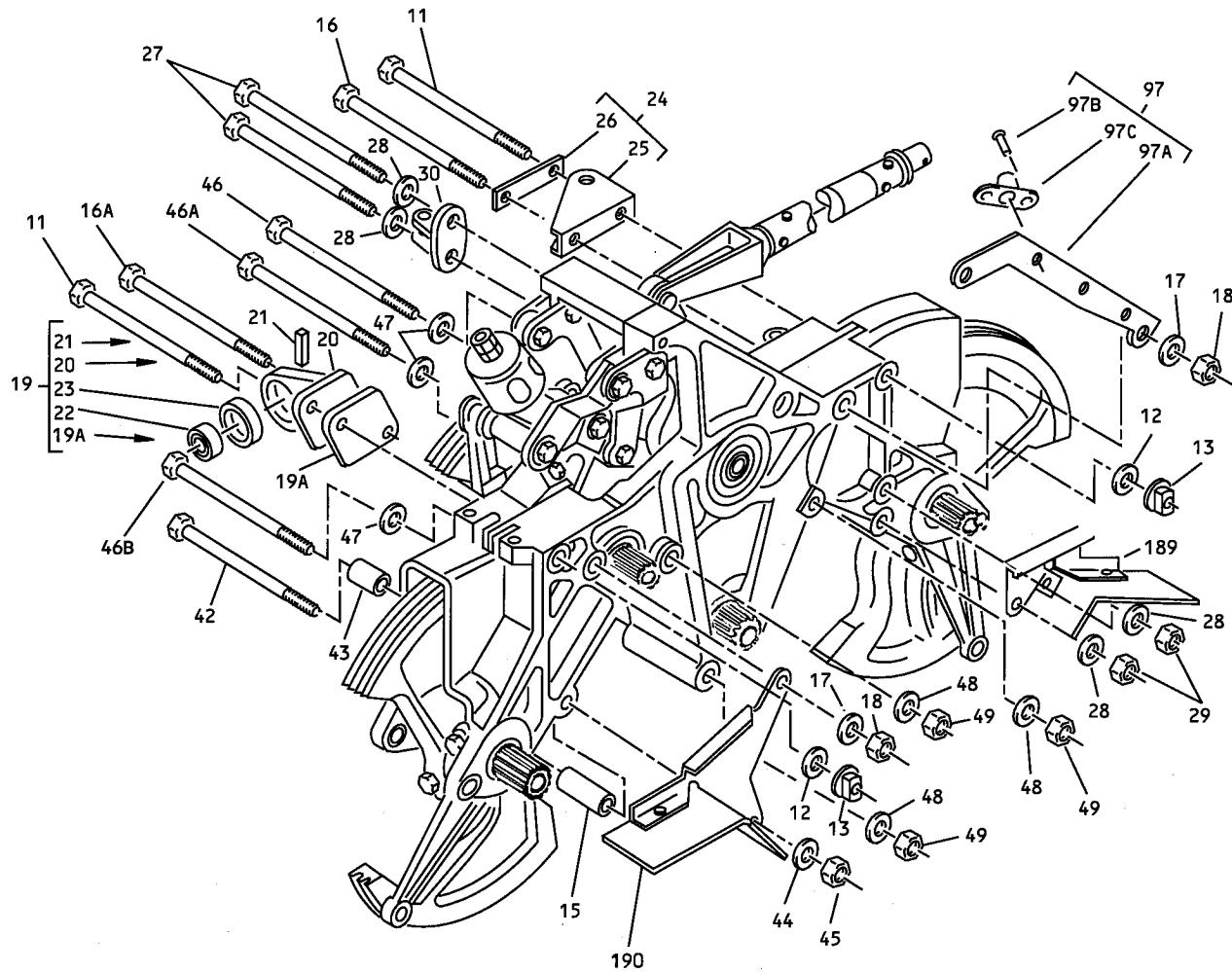
 Fits and Clearances
 Figure 601 (Sheet 7)

Ref Letter Fig. 601	Mating Item No. Fig. 1101	Design Dimensions				Service Wear Limits		
		Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inch)		Maximum Allowable Clearance (inch)
		Min	Max	Min	Max	Min	Max	
AE	ID 176	0.8743	0.8757	*[1] 0.0013	0.0004	0.8747	0.8757	0.0010
	OD 133	0.8753	0.8756					
AF	ID 177,185	0.6245	0.6250	*[1] 0.0008	0.0000	0.6240	0.6250	0.0010
	OD 58,70	0.6250	0.6253					
AG	ID 185	0.6245	0.6250	0.0000	0.0010	0.6200	0.6250	0.0050
	OD 106	0.6240	0.6245					
AH	ID 135	0.2500	0.2505	0.0005	0.0025	0.2480	0.2540	0.0060
	OD 136	0.2480	0.2495					

*[1] Interference fit

Fits and Clearances
Figure 601 (Sheet 8)

ILLUSTRATED PARTS LIST

 Ratio Changer Assembly
 Figure 1101 (Sheet 1)

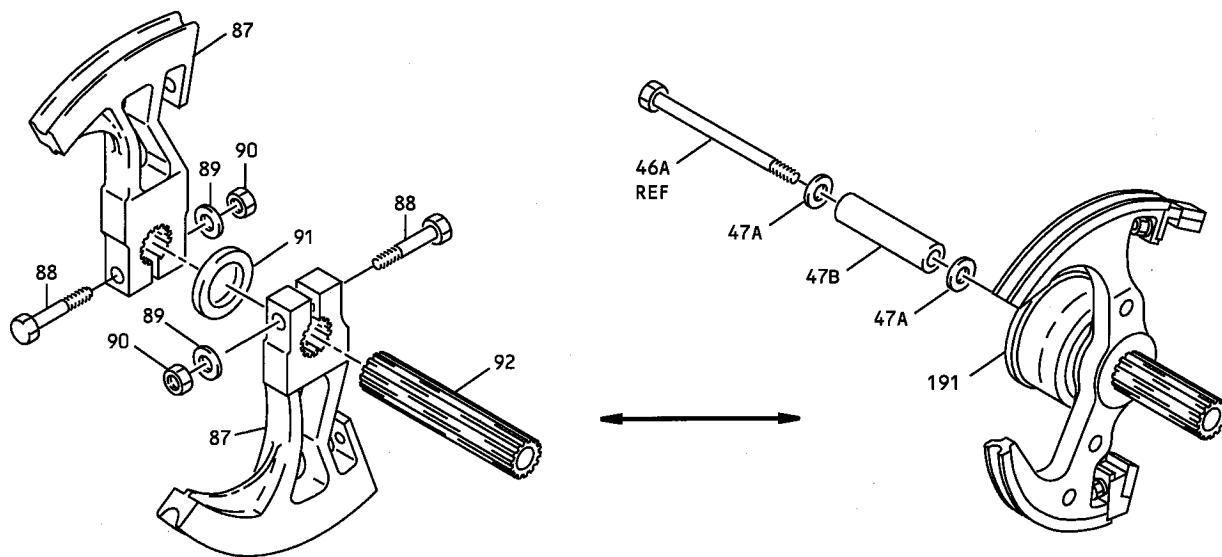
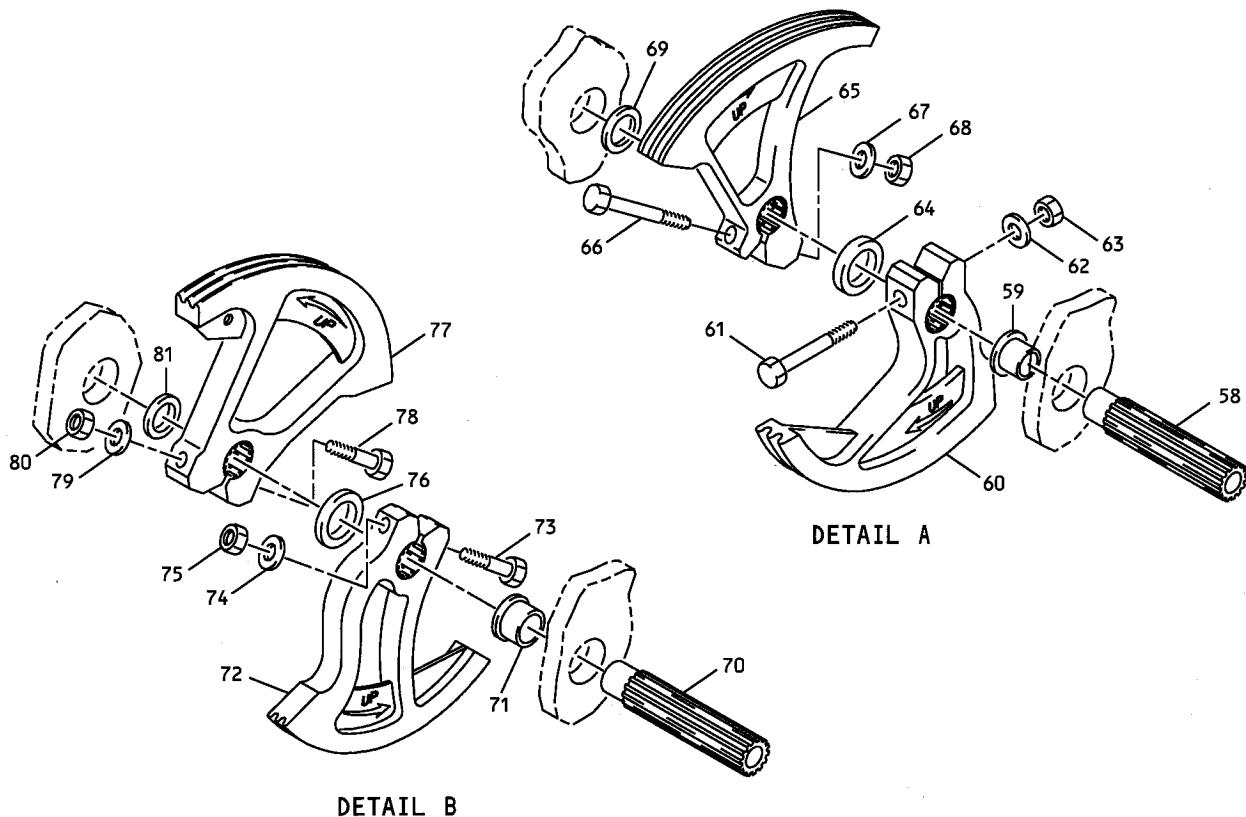


Ratio Changer Assembly
 Figure 1101 (Sheet 2)

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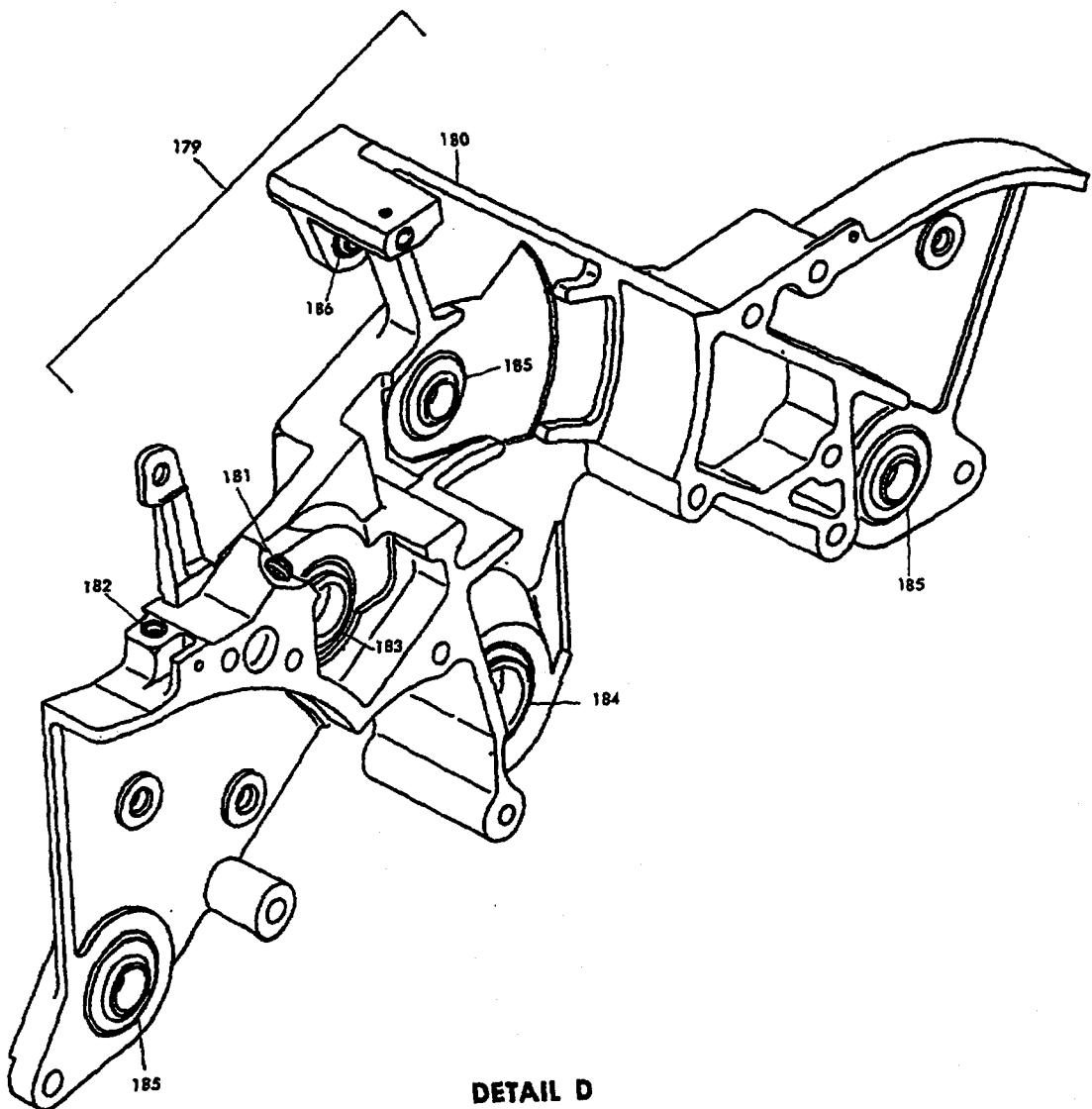
65-46370-3 THRU -12,-15

DETAIL C

 65-46370-13,-14,-16,-17,-18,
 -19,-20,-21,-23,-24,
 -26,-27,-28

 Ratio Changer Assembly
 Figure 1101 (Sheet 3)

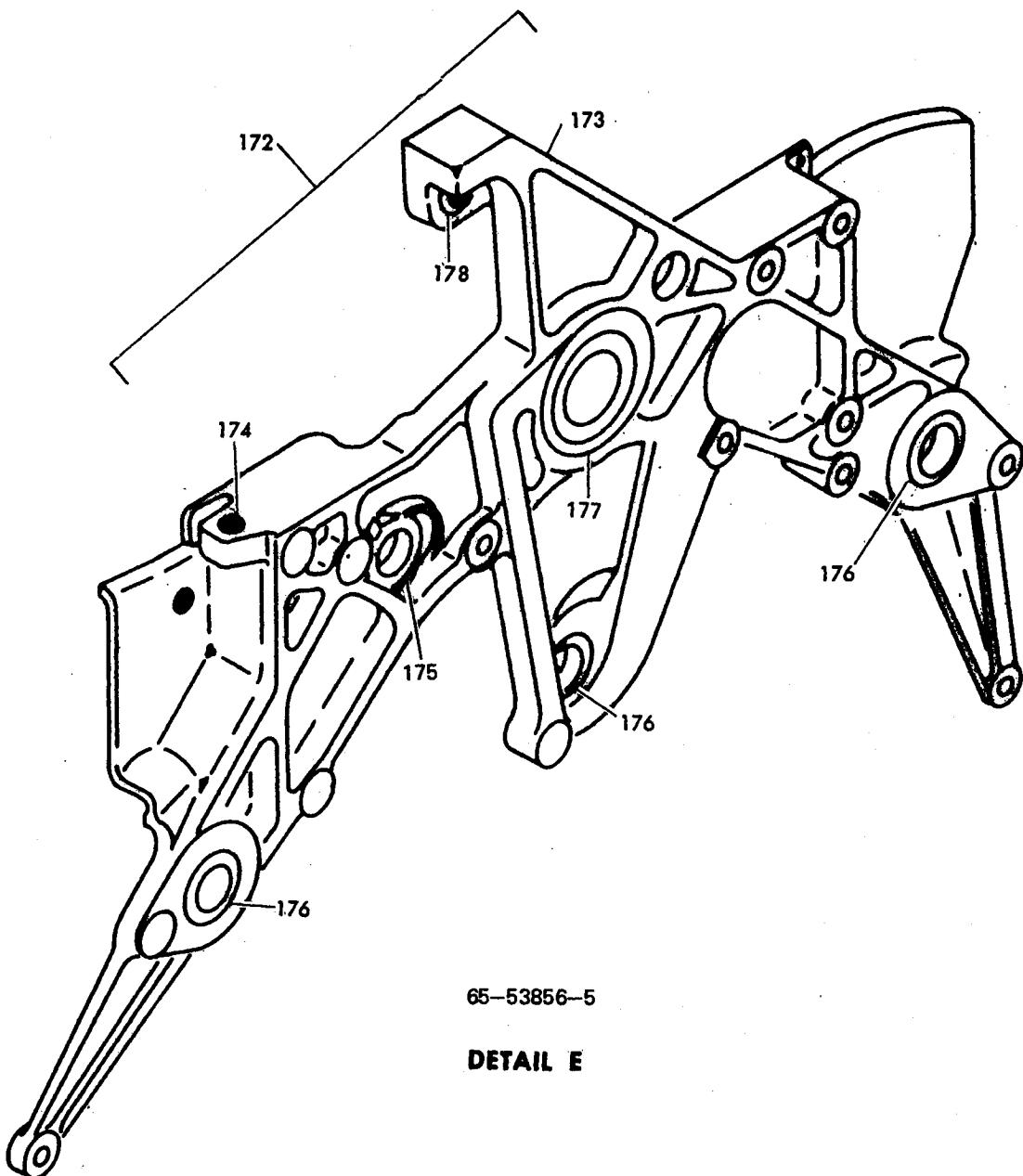
BOEING
OVERHAUL MANUAL



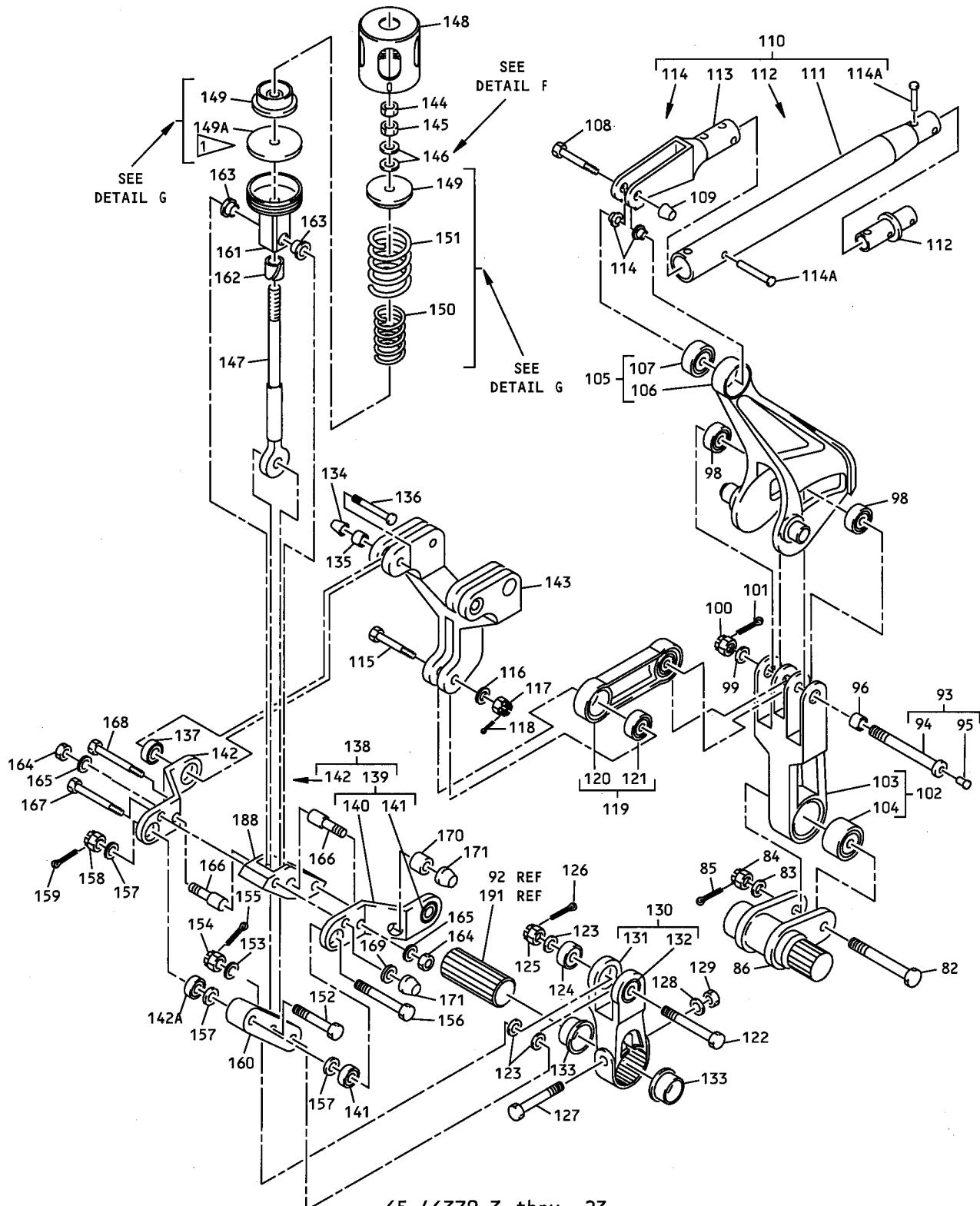
DETAIL D

65-46370

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OVERHAUL MANUAL

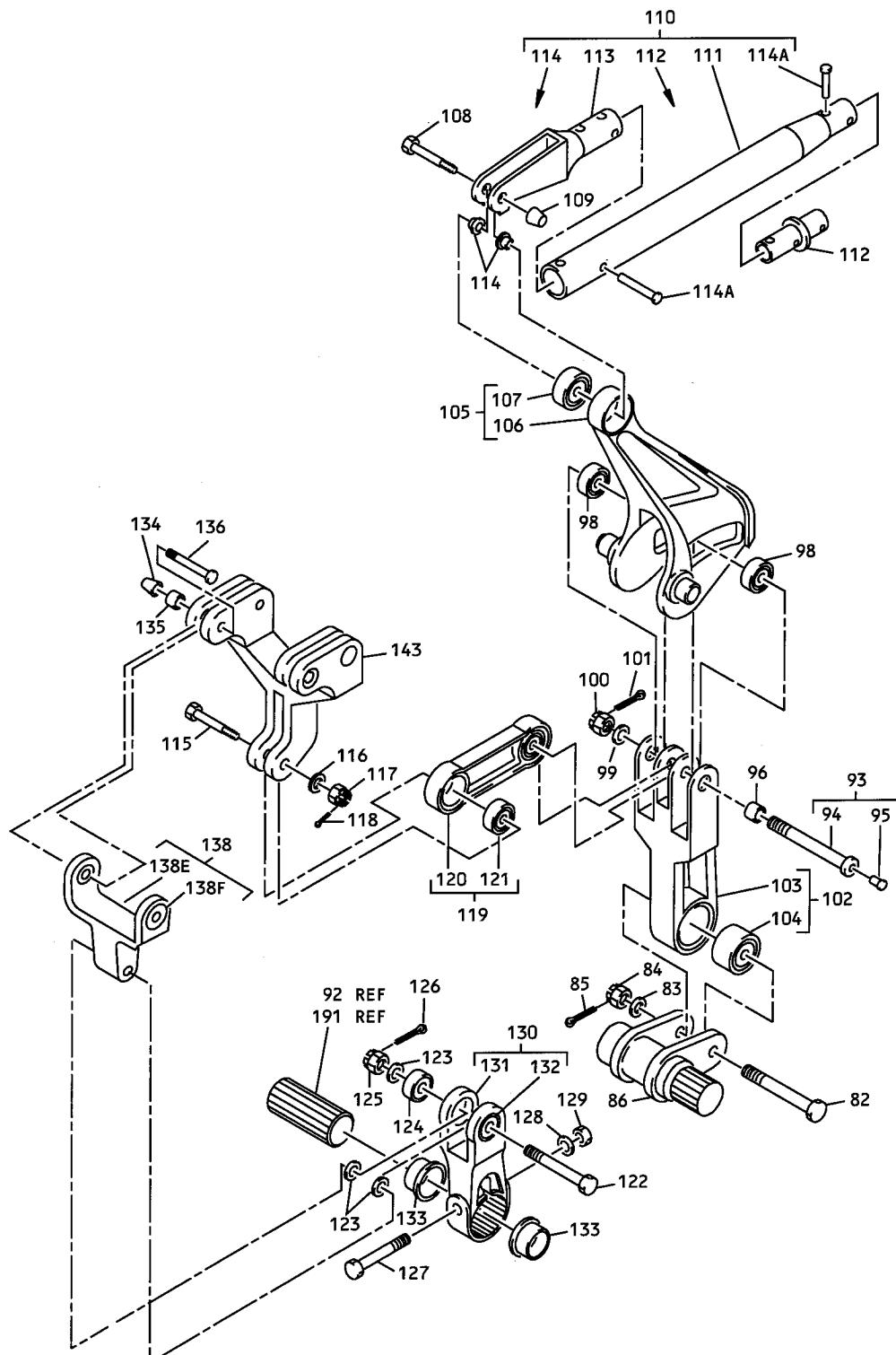


OVERHAUL MANUAL



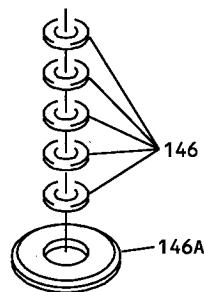
65-46370-3 thru -23

Ratio Changer Assembly
Figure 1101 (Sheet 6)

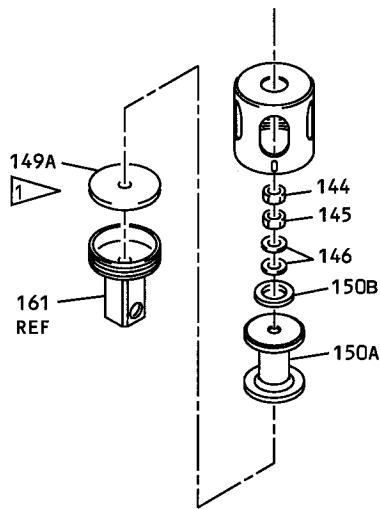


65-46370-24,-26,-27,-28

Ratio Changer Assembly
Figure 1101 (Sheet 6A)



OPTIONAL FOR 65-46370-21
DETAIL F



FOR 65-46370-23 ONLY
DETAIL G

► OPTIONAL

Ratio Changer Assembly
Figure 1101 (Sheet 7)

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Jul 1/08

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	USE CODE	QTY PER ASSY
			1 2 3 4 5 6 7		
1101-					
29	NAS679A4W		. NUT (REPLD BY BACN10JC4)		2
29	BACN10JC4		. NUT (REPLS NAS679A4W) (REPLD BY BACN10JC4CD)		2
29	BACN10JC4CD		. NUT (REPLS BACN10JC4)		2
30	69-42512-1		. BRACKET		1
31	BACC30K8		. COLLAR	A-U	2
31	BACC30BK8		. COLLAR	VWX	2
32	66-24917-2		. BUSHING, LEVER SHAFT		2
33	BACB30GW8-12		. LOCKBOLT	A-QU RST	2
33	BACB30FM8-12		DELETED		
33	BACB30VN8K12		. BOLT, PROTRUDING HD	VWX	2
34	NAS1104-19		. BOLT	CD	1
34	NAS1104-22		. BOLT (REPLS NAS1104-19)	ABE-JM	1
34	BACB30NF4-22		. BOLT	KLN-T	1
34	BACB30NF4-19		. BOLT	UVW	1
34	BACB30LJ4-19		. BOLT	X	1
35	NAS43DD4-75		. SPACER	A-Q	1
35	NAS43DD4-75FC		. SPACER	RST	1
35	NAS43DD4-70FC		. SPACER	UVWX	1
36	AN960-416L		. WASHER (REPLD BY NAS1149F0432P)		1
36	NAS1149F0432P		. WASHER (REPLS AN906-416L)		1
37	NAS679A4W		. NUT (REPLD BY BACN10JC4)		1
37	BACN10JC4		. NUT (REPLS NAS679A4W)(REPLD BY BACN10JC4CD)		1
37	BACN10JC4CD		. NUT (REPLS BACN10JC4)		1
38	NAS1104-18		. BOLT	ABD-J	1
38	NAS1104-18		. BOLT	C	2
38	BACB30NF4-19		. BOLT	KL	1
38A	69-57936-1		. GUARD (SB 27-1017)	ABD-L	1
38A	69-57936-1		. GUARD	M-X	2
39	NAS43DD4-60		. SPACER	C	2
39	NAS43DD4-60		. SPACER	ABD-L	1
39	NAS43DD4-50		DELETED		
40	AN960-416L		. WASHER		2
40	NAS1149F0432P		. WASHER (REPLS AN906-416L)		2
41	NAS679A4W		. NUT (REPLD BY BACN10JC4)		2
41	BACN10JC4		. NUT (REPLS NAS679A4W)(REPLD BY BACN10JC4CD)		2
41	BACN10JC4CD		. NUT (REPLS BACN10JC4)		2
42	NAS1104-48		. BOLT (REPLS NAS1104-47)	A-JM	1
42	BACB30NF4-47		. BOLT	KLN-T	1
42	BACB30NF4-49		. BOLT	UVW	1
42	BACB30LJ4-49		. BOLT	X	1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-97B	MS20426D3N		. . RIVET	A-U	6
97B	BACR15DR3AC2		. . RIVET	VWX	6
97C	BACN10JR3FR		. . NUTPLATE	A-U	3
97C	BACN10JR3CFD		. . NUTPLATE	VWX	3
98	BACB10BN04		. BEARING (REPLS BACB10B131)	ACDF	2
98	BACB10ET04		. BEARING (RPLS BACB10BN04)	ACDF	2
98	BACB10BN04P		. BEARING (REPLS BACB10B256R)	BEG-Q	2
98	BACB10ET04		. BEARING (RPLS BACB10BN04P)	BEG-Q	2
98	BACB10ET04		. BEARING (REPLS BACB10B25R)	RST	2
98	BACB10ET04J		. BEARING	U	2
98	BACB10ET04		. BEARING	VW	2
98	BACB10HH04		. BEARING	X	2
99	AN960-416L		. WASHER (REPLD BY NAS1149F0432P)		1
99	NAS1149F0432P		. WASHER (REPLS AN960-416L)	A-X	1
100	MS17826-4		. NUT	A-T	1
100	BACN11N4CD		. NUT	U	1
100	BACN11N104CD		. NUT	VWX	1
101	MS24665-134		. PIN, COTTER		1
101	BACP18BC02C0 6P		. PIN, COTTER (RPLS MS246650134)		1
102	65-46365-5		. LEVER ASSY, OUTPUT	ABE-X	1
102	65-46365-7		. LEVER ASSY, OUTPUT	CD	1
103	65-46365-6		. . LEVER (USED ON 65-46365-5)		1
103	65-46365-8		. . LEVER (USED ON 65-46365-7)		1
104	BACB10C161H		. . BEARING		1
105	65-46364-4		. LEVER ASSY	A	1
105	65-46364-6		. LEVER ASSY	CD	1
105	65-46364-8		. LEVER ASSY	BE-X	1
105	65-46364-4		. LEVER ASSY (OPT)	EF	1
106	65-46364-5		. . LEVER (USED ON 65-46364-4)		1
106	65-46364-7		. . LEVER (USED ON 65-46364-6)		1
106	65-46364-9		. . LEVER (USED ON 65-46364-8)		1
107	BACB10C160H		. . BEARING		1
108	BACB30GW8-15		. LOCKBOLT	A-QU RST	1
108	BACB30FM8-15		DELETED		
108	BACB30VN8K15		. LOCKBOLT	VWX	1
109	BACC30K8		. COLLAR		1
110	69-40326-3		. ROD ASSY, INPUT	ABE-P	1
110	69-40326-1		. ROD ASSY, INPUT	CD	1
110	69-40326-4		. ROD ASSY, INPUT	QRST-X	1
111	69-40326-2		. . TUBE		1
112	60-2271-2		. . PLUG		1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	USE CODE	QTY PER ASSY
			1 2 3 4 5 6 7		
1101-					
177	BACB10A684		. . BEARING (REPLD BY BACB10BX10)	A-U	1
177	BACB10BX10		. . BEARING (REPLS BACB10A684)	A-U	1
177	BACB10BX10		. . BEARING	VWX	1
178	KP4AE9894		. . BEARING, V21335(OPT TO BACB10A04A, BACB10A313)	A-U	1
178	BACB10AC04A		. . BEARING (PREF)	A-U	1
178	BACB10AC04A		. . BEARING	VWX	1
178	BACB10A313		. . BEARING (OPT TO BACB10AC04A, KPAE9894)	A-U	1
179	65-53854-3		. . FRAME ASSY, FORWARD	ABEF	1
179	65-53854-1		. . FRAME ASSY, FORWARD	CD	1
179	65-53854-5		. . FRAME ASSY, FORWARD	G-JM	1
179	65-53854-7		. . FRAME ASSY, FORWARD	KLN-U	1
179	65-53854-9		. . FRAME ASSY, FORWARD	VWX	1
180	65-53854-4		. . FRAME (USED ON 65-53854-3)		1
180	65-53854-2		. . FRAME (USED ON 65-53854-1)		1
180	65-53854-6		. . FRAME (USED ON 65-53854-5)		1
180	65-53854-8		. . FRAME (USED ON 65-53854-7)		1
180	65-53854-10		. . FRAME (USED ON 65-53854-9)		1
181	MS21209F4-15		. . INSERT	A-U	1
181	MS21209F4-15P		. . INSERT	VWX	1
182	MS21209F4-20		. . INSERT	A-U	1
182	MS21209F4-20P		. . INSERT	VWX	1
183	B540DDE9607		. . BEARING, V21335 (OPT TO BACB10AS14)		1
183	BACB10AS14		. . BEARING (PREF)	A-U	1
183	BACB10AS14		. . BEARING	VWX	1
184	B541DDE9607		. . BEARING, V21335 (OPT TO BACB10AS17)	A-U	1
184	BACB10AS17		. . BEARING (PREF)	A-U	1
184	BACB10AS17		. . BEARING	VWX	3
185	BACB10A684		. . BEARING (REPLD BY BACB10BX10)	A-U	3
185	BACB10BX10		. . BEARING (REPLS BACB10A684)	A-U	3
185	BACB10BX10		. . BEARING	VWX	1
186	KP4AE9894		. . BEARING, V21335 (OPT TO BACB10A313, BACB10AC04A)	A-U	1
186	BACB10AC04A		. . BEARING (PREF)	A-U	1
186	BACB10A313		. . BEARING (OPT TO BACB10AC04A, KP4AE9894)	A-U	1
186	BACB10AC04A		. . BEARING	VWX	1
187	BACB30KZ4-15		. . BOLT		1
187	NAS564-15		. . BOLT (OPT)	U	1
187	BACB30KZ4-15		. . BOLT (OPT)	U	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE	USE CODE	QTY PER ASSY
			1 2 3 4 5 6 7		
1101-					
188	65-75323-1		. SPACER	A-T	1
188	69-50383-1		. SPACER (OPT)	A-T	1
189	65-67908-4		. SHIELD ASSY	CD	1
190	65-67908-6		. SHIELD ASSY	CD	1
190	65-67908-13		DELETED		
191	10-62024-1		. NO-BACK ASSY (OPT TO 10-62024-4)	KN	1
191	10-62024-2		. NO-BACK ASSY (OPT TO 10-62024-4)	L	1
191	10-62024-3		. NO-BACK ASSY (OPT TO 10-62024-4)	OPQ	1
191	10-62024-4		. NO-BACK ASSY	RSTU	1
191	10-62024-4		. NO-BACK ASSY (OPT TO 10-62024-1, -2,-3)	KLNOPQ	1
191	A3091A000		. NO-BACK ASSY (PREF)	RSTU	1
191	A3091A000		. NO-BACK ASSY	VWX	1

VENDORS

V21335 TIMKEN US CORP., 336 MECHANIC ST., LEBANON, NEW HAMPSHIRE 03766-2614



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Part No.	Fig. and Index No.	Qty. per Assy.	Part No.	Fig. and Index No.	Qty. per Assy.
BACB30LJ4-19	1101-51	1	BACB30NF4-67	46A	1
BACB30LJ4-19	51A	1	BACB30NF4D14	115	1
BACB30LJ4-19	55	2	BACB30NF4D14	152	1
BACB30LJ4-19	55A	2	BACB30NF4D24	122	1
BACB30LJ4-26	61	1	BACB30NF4D24	156	1
BACB30LJ4-26	66	1	BACB30NF4D33	93	1
BACB30LJ4-26	73	1	BACB30NF4D34	93	1
BACB30LJ4-26	78	1	BACB30NF5D18	82	1
BACB30LJ4-44	16	1	BACB30VN8K12	136	1
BACB30LJ4-44	27	1	BACB30VN8K12	33	2
BACB30LJ4-44	46B	1	BACB30VN8K15	108	1
BACB30LJ4-46	16A	1	BACC30BK8	31	2
BACB30LJ4-49	17	1	BACC30K8	109	1
BACB30LJ4-56	46	1	BACC30K8	134	2
BACB30LJ4-58	1	1	BACC30K8	31	2
BACB30LJ4-67	46A	1	BACC30M5	171	2
BACB30LJ4D14	115	1	BACN10JC3	144	1
BACB30LJ4D24	122	1	BACN10JC3CD	144	
BACB30LJ4D33	93	1	BACN10JC4	129	1
BACB30LJ5D18	82	1	BACN10JC4	164	2
BACB30NF4-1	5	2	BACN10JC4	18	2
BACB30NF4-17	127	1	BACN10JC4	29	2
BACB30NF4-17	51	1	BACN10JC4	37	1
BACB30NF4-17	51A	1	BACN10JC4	4	1
BACB30NF4-18	55	2	BACN10JC4	41	2
BACB30NF4-19	38	1	BACN10JC4	45	
BACB30NF4-19	51	1	BACN10JC4	49	3
BACB30NF4-19	51A	1	BACN10JC4	53	1
BACB30NF4-19	55	2	BACN10JC4	57	2
BACB30NF4-19	55A	2	BACN10JC4	63	1
BACB30NF4-20	127	1	BACN10JC4	68	1
BACB30NF4-22	34	1	BACN10JC4	75	1
BACB30NF4-22	34	1	BACN10JC4	80	2
BACB30NF4-26	61	1	BACN10JC4	90	2
BACB30NF4-26	66	1	BACN10JC4CD	129	1
BACB30NF4-26	73	1	BACN10JC4CD	164	2
BACB30NF4-26	78	1	BACN10JC4CD	18	2
BACB30NF4-44	16	1	BACN10JC4CD	29	2
BACB30NF4-44	16A	1	BACN10JC4CD	37	1
BACB30NF4-44	27	2	BACN10JC4CD	4	1
BACB30NF4-44	46B	1	BACN10JC4CD	41	2
BACB30NF4-46	16A	1	BACN10JC4CD	45	1
BACB30NF4-47	42	1	BACN10JC4CD	49	3
BACB30NF4-49	42	1	BACN10JC4CD	53	1
BACB30NF4-56	46	1	BACN10JC4CD	53A	1
BACB30NF4-58	1	1	BACN10JC4CD	57	2
BACB30NF4-65	46A	1	BACN10JC4CD	57A	2

