

OVERHAUL MANUAL

TO: ALL HOLDERS OF AUXILIARY POWER UNIT ACCESS DOOR ASSEMBLY OVERHAUL MANUAL,
 52-46-10

REVISION NO. 7, DATED SEP 5/91

HIGHLIGHTS

DESCRIPTION OF CHANGE	TOPICS AFFECTED												
	D & O	D / A S S Y	C l e a n i n g	I n s p / C h k	R e p a i r	A S S Y	F / C	T e s t	T / S h o o t i n g	S / T o o l s	S t o r a g e	I P L	L / O v e r h a u l
Added 65-50904-625 top assembly per latest engineering release												X	

AUXILIARY POWER UNIT ACCESS DOOR ASSEMBLY

52-46-10

BOEING P/N 65-50904-2, -271, -625

AIRLINE P/N

THE FOLLOWING DIRECTIVES APPLY TO THIS SUBJECT:

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVES	DATE DIRECTIVE INCORPORATED INTO TEXT
		PRR 30680-43	Nov 15/68

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LIST OF EFFECTIVE PAGES

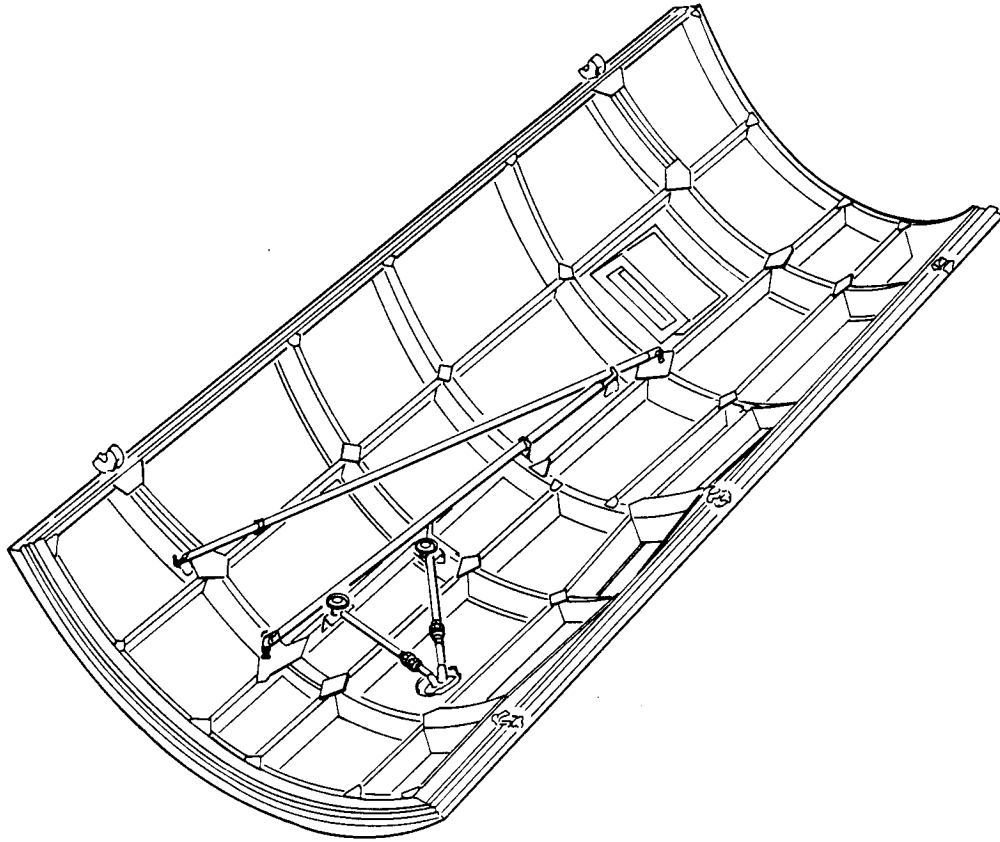
* Indicates pages revised, added or deleted in latest revision
F Indicates foldout pages - print one side only

PAGE	DATE	PAGE	DATE	PAGE	DATE
52-46-10					
* T-1	Sep 5/91				
T-2	BLANK				
* LEP-1	Sep 5/91				
LEP-2	BLANK				
T/C-1	Nov 15/68				
T/C-2	BLANK				
1	Mar 5/86				
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502	Nov 15/68				
901	Nov 15/68				
902	BLANK				
* 1101	Sep 5/91				
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* 1112	Sep 5/91				
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OVERHAUL MANUALAUXILIARY POWER UNIT ACCESS DOOR

Auxiliary Power Unit Access Door Assembly
Figure 1

DESCRIPTION AND OPERATION**1. Description**

- A. The door is a frame structure, fabricated of aluminum alloy. The structure is covered with a two piece bonded aluminum alloy skin. Latches and hinges are made of corrosion resistant steel. A drain mast is installed in the left forward side of the door to drain off fluids from the auxiliary power unit. This mast is made of epoxy-reinforced fiberglass.

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A cooling duct is installed in the aft end of the door. This duct is made of corrosion resistant steel. The door is formed to fair in with the outer mold line of the fuselage when it is closed and latched.

2. Operation

- A. The door is opened and closed with three tension latches. These latch assemblies are of a fail-safe design. They lock the door with an overcenter toggle.

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DISASSEMBLY

1. Removal and Disassembly of Cowl Support Rods (See figure 1102.)
 - A. Remove nut (1), washer (2), and spring washer (3). Remove rod assembly (4).
 - B. Remove cotter pin (7), nut (8), washers (9), bolt (10), and spring washers (11) from rod assembly (4). Remove eyebolt (6).

NOTE: Do not disassemble rest of rod assembly (4) unless parts must be replaced.
 - C. Remove nut (1), washer (2), and spring washer (3). Remove rod assembly (21).
 - D. Remove cotter pin (24), nut (25), washers (26), bolt (27), and spring washer (28) from rod assembly (21). Remove eyebolt (23).

NOTE: Do not disassemble rest of rod assembly (21) unless parts must be replaced.
 - E. Remove nuts (38), washers (39), screws (40, 41 and 42). Remove plate assemblies (43).
 - F. Remove nuts (46), washers (47), and screws (48). Remove spring clips (49).
2. Removal of Drains (See figure 1103.)
 - A. Remove nuts (1), union (2), and sleeves (3).
 - B. Remove nuts (4), washers (5), bolts (6), and spacers (7). Remove cup assembly (8).
 - C. Remove nuts (9), union (10), and sleeves (11).
 - D. Remove bolts (12), washers (13), and spacers (14). Remove cup assembly (15).
 - E. Remove nuts (16), washers (17), and bolts (18). Remove doubler assembly (19) and drain mast assembly (20).
3. Removal of Latches and Seals (See figure 1104.)
 - A. Remove nuts (1), bolts (2), and washers (3). Remove latch assemblies (4).
 - B. Remove nut (7), bolt (8), washer (10), and spacer (9).

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C. Remove seals (21, 27, and 31).

NOTE: Tension bearings (5), bearing plates (11 and 12), stop fittings (15), retainers (19, 25, and 29), and fillers (20, 26, and 30) are illustrated for reference. These parts are permanently fastened, and should not be removed unless parts must be replaced.

4. Removal of Hinges and Lock Levers (See figure 1105.)

A. Remove nuts (1), bolts (2), washers (3, 4, and 5), and clamp-up bushings (6). Remove lock levers (11).

B. Remove nuts (7), bolts (8), washers (9), and spacers (10). Remove hinges (12).

C. Remove nuts (13), washers (14), and shear pins (15).

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CLEANING

1. Wash and rinse all metal parts in dry cleaning solvent, Specification P-D-680, or equivalent.
2. Remove stubborn accumulations of dirt with a stiff-bristle brush. Do not use a metallic brush.
3. Drain, and dry all parts thoroughly with clean, lint-free cloth, or dry compressed air.
4. Clean fiberglass mast assembly (20, figure 1103) with solvent, EMS 3-2C, Type 2, or equivalent.

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INSPECTION/CHECK

1. Visual

- A. Examine all metal parts for scratches, corrosion, dents, and other damage. Use a strong light and 10-power magnification.
- B. Examine fiberglass mast assembly (20, figure 1103) for delamination, defects in potting, and corrosion of tubes.
- C. Examine entire interior of structure for loose fasteners, damage, and general condition of paint and finish.

OVERHAUL MANUALREPAIR

1. Repair

- A. Use standard industry practices for repair of this component.
- B. For structural repairs, refer to the Boeing 737 Structural Repair Manual 52-00 and 52-40.
- C. For repairs to fiberglass components, refer to the following Boeing 737 Structural Repair Manuals:
 - (1) SRM 51-40-09 for 737-100, -200
 - (2) SRM 51-70-06 for 737-300, -400, -500

2. Refinish

NOTE: Refer to 20-30-02 for stripping of protective finishes and to 20-41-01 for explanation of F and SRF finish codes.

A. Fig. 1102

- (1) Tube (5, 22) -- Alodize per 20-30-03 and apply one coat primer BMS 10-11, Type 1 per 20-41-02 (SRF-2.901) on interior and exterior surfaces. Material: Aluminum alloy.
- (2) Rod end (12, 29) -- Cadmium plate with post-plate chromate treatment (F-1.20) all over. Material: 15-5PH CRES, 150-170 ksi or 17-4PH, 140-160 ksi.
- (3) Rod end (14, 31) -- Alodize or chromic acid anodize and apply one coat primer BMS 10-11, Type 1 (SRF-2.30) all over. Material: Aluminum alloy.

B. Fig. 1103

- (1) Cup assembly (8, 15) -- Passivate per 20-30-03 (F-8.07) all over. Material: 321 or 347 CRES.
- (2) Drain mast assembly (20) -- Apply Dexter 28-C-1 static conditioner plus Dexter 8-W-5 surfacer per 20-10-06 (SRF-14.672) followed by conductive coating BMS 10-21, Type 1 per 20-10-06 (SRF-14.68) all over. On all surfaces except surface of flange which fays with door, apply one coat primer BMS 10-11, Type 1 per 20-41-02 (SRF-12.205) plus gray gloss enamel BMS 10-60, BAC color 707 (SRF-14.9813).

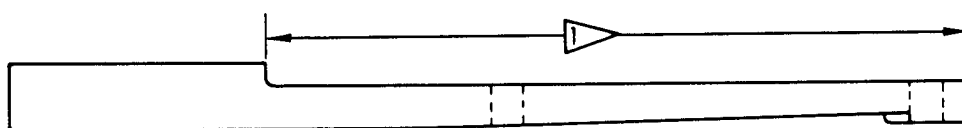
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C. Fig. 1104

- (1) Bearing (5) -- Cadmium plate per 20-42-05 (F-1.191) all over.
Material: 17-4PH CRES, 180 ksi minimum.
- (2) Plate (11, 12) -- Cadmium plate with post-plate chromate treatment per 20-42-05 (F-1.1923) all over. Material: 4130 steel, 160-180 ksi.
- (3) Fitting (15), retainer (19, 25, 29), filler (20, 26, 30) -- Alodize or chromic acid anodize and apply one coat primer BMS 10-11, type 1 (SRF-2.30) all over. Material: Aluminum alloy.

D. Fig. 1105

- (1) Lever (11) -- Passivate per 20-30-03 (F-8.07) all over.
Material: 301 CRES.
 - (2) Hinge (12) -- Passivate per 20-30-03 (F-8.70) all over. Apply two coats primer BMS 10-11, type 1 per 20-41-02 (F-20.03) per Fig. 401. Material: 17-4PH CRES, 180-200 ksi.
 - (3) Pin (15) -- Apply hard chrome plate per 20-42-03 (F-1.90) to head. Cadmium plate per 20-42-05 type 2 (F-1.202) shank, slot, threads and under surface of head. Material: 4330M steel, 180-200 ksi.
- E. Touch up entire interior of assembly as necessary with primer, BMS 10-11, type 1.



 APPLY TWO COATS PRIMER BMS 10-11
TYPE 1 PER 20-41-02 (F-20.03)

HINGE (12, FIG. 1105)

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3. Replacement

- A. Replace all parts worn or damaged beyond simple repair.
- B. Replace all cotter pins at each overhaul.
- C. Replace seals (21, 27, 31, Fig. 1104) at each overhaul.
 - (1) Remove old seal by scraping off with a sharp-edged wooden or plastic tool. Remove rivet (23) also to remove seal (31).
 - (2) Bond new seal in place as directed per 20-50-12, Type 60. Install rivet (23) through seal (31).
- D. Replace worn bushing (45, Fig. 1102) as follows:
 - (1) Remove bushing.
 - (2) Coat O.D. of bushing and plate (44, Fig. 1102) bore with primer BMS 10-11, type 1.
 - (3) Install bushing wet.
- E. Replace spring washers (3, 11, 28, Fig. 1102) at each overhaul.
- F. Replace safety pins (15, 32, Fig. 1102) at each overhaul.

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ASSEMBLY

1. Installation of Hinges and Lock Levers (Fig. 1105)
 - A. Install forward shear pin (15) with washers (14) and nut (13), or quick release pin (16) and chain (17).
 - B. Locate forward hinge (12). Install spacer (10), bolt (8), washer (9), and nut (7).
 - C. Locate forward cowl panel lock lever (11). Install clamp-up bushing (6), washers (3, 4, and 5), bolt (2), and nut (1).
 - D. Install aft shear pin (15) with washers (14) and nut (13), or quick release pin (16) and chain (17).
 - E. Locate aft hinge (12). Install spacer (10), bolt (8), washer (9), and nut (7).
 - F. Locate aft cowl panel lock lever (11). Install clamp-up bushing (6), washers (3, 4, and 5), bolt (2), and nut (1).
2. Installation of Latches (Fig. 1104)
 - A. Install spacers (9), bolts (8), washers (10), and nuts (7) in three places.
 - B. Locate forward latch assembly (4). Install bolt (2), washer (3), and nut (1).
 - C. Locate center latch assembly (4). Install bolt (2), washer (3), and nut (1).
 - D. Locate aft latch assembly (4). Install bolt (2), washer (3), and nut (1).
3. Installation of Drains (Fig 1103)
 - A. Locate drain mast assembly (20) and doubler assembly (19). Install bolts (18), washers (17), and nuts (16).
 - B. Locate cup assembly (15). Attach with spacers (14), washers (13), and bolts (12).
 - C. Attach drain tube of cup assembly to small drain tube in drain mast with sleeves (11), union (10), and nuts (9).
 - D. Locate cup assembly (8). Attach with spacers (7), bolts (6), washers (5), and nuts (4).
 - E. Attach drain tube of cup assembly (8) to large drain tube in drain mast with sleeves (3), union (2), and nuts (1).

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4. Installation of Cowl Support Rods (See figure 1102.)
 - A. Install eyebolt (23) on cowl support rod assembly (21) with spring washer (28), washers (26), bolt (27), nut (25), and cotter pin (24).
 - B. Install eyebolt (6) on rod assembly (4) with spring washer (11), washers (9), bolt (10), nut (8), and cotter pin (7).
 - C. Install forward plate assembly (43) with four screws (42), eight washers (39), and four nuts (38).
 - D. Install aft plate assembly (43) with two screws (40), two screws (41), eight washers (39), and four nuts (38).
 - E. Install rod assembly (21) with spring washer (3), washer (2), and nut (1).
 - F. Install rod assembly (4) with spring washer (3), washer (2), and nut (1).

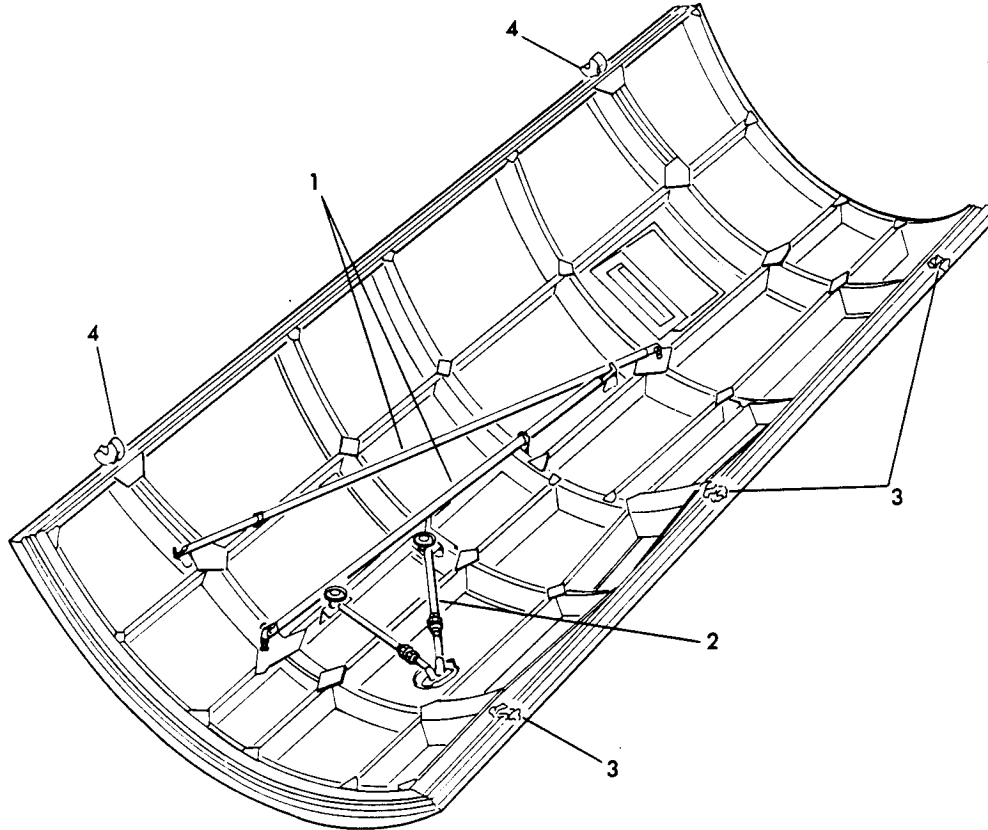
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STORAGE INSTRUCTIONS

1. Secure all loose rods to prevent movement and damage to assembly.
2. Wrap assembly in nonabsorbent material, and store in a cool, dry area, preferably humidity controlled.
3. For further instructions, refer to Subject 20-44-02, "Temporary Protective Coatings." Refer also to Subject 20-70-01, "Protection, Storage, and Handling of Airplane Components."

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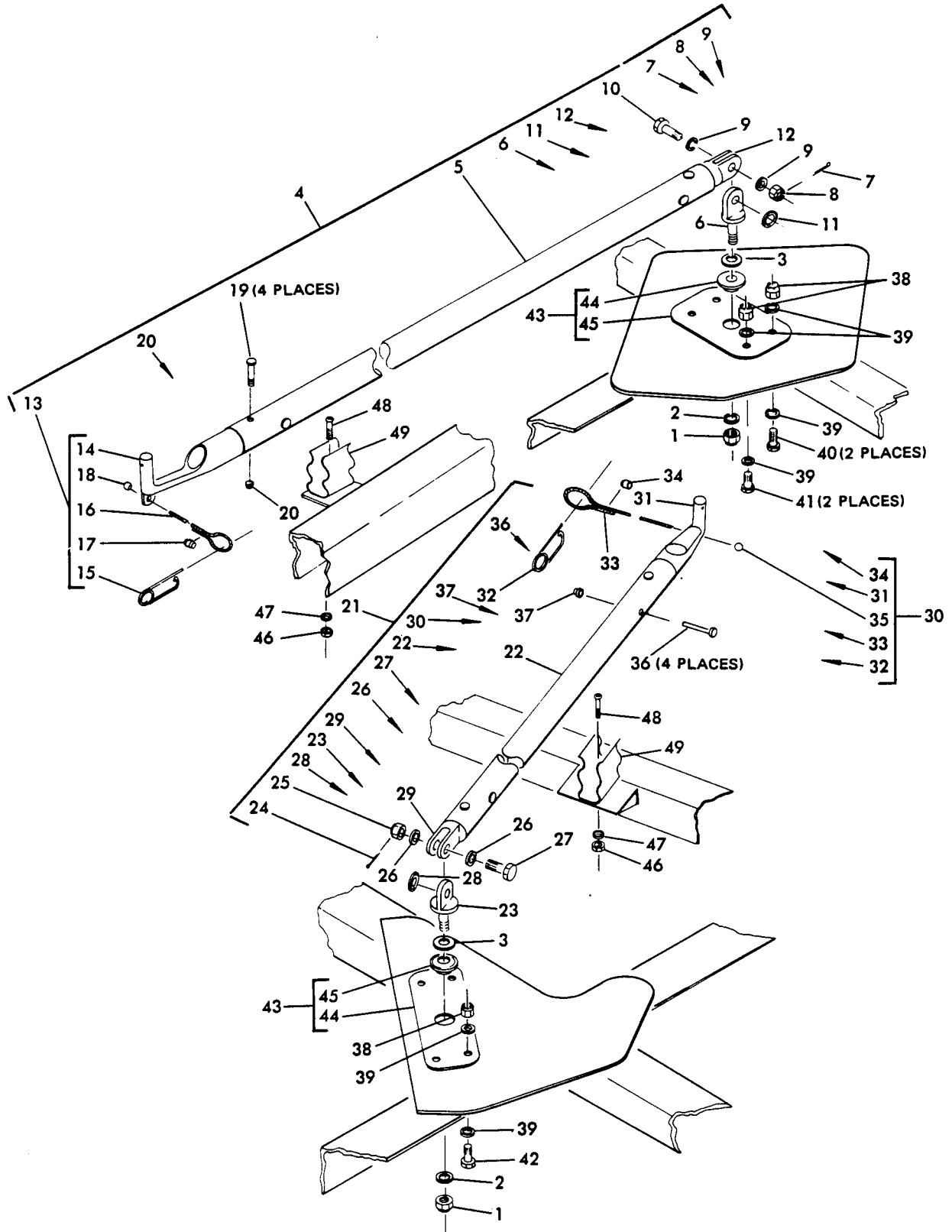
ILLUSTRATED PARTS LIST



Auxiliary Power Unit Access Door Assembly
Figure 1101

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-	65-50904-2 65-50904-271 65-50904-625										
1			AUXILIARY POWER UNIT ACCESS DOOR ASSY								
2			AUXILIARY POWER UNIT ACCESS DOOR ASSY								
3			AUXILIARY POWER UNIT ACCESS DOOR ASSY								
4			. COWL SUPPORT RODS AND ATTACHMENTS (FIG. 1102)								
			. DRAINS AND ATTACHMENTS (FIG. 1103)								
			. LATCHES AND SEALS (FIG. 1104)								
			. HINGES AND LOCKING LEVERS (FIG. 1105)								

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Cowl Support Rods and Attachments
Figure 1102

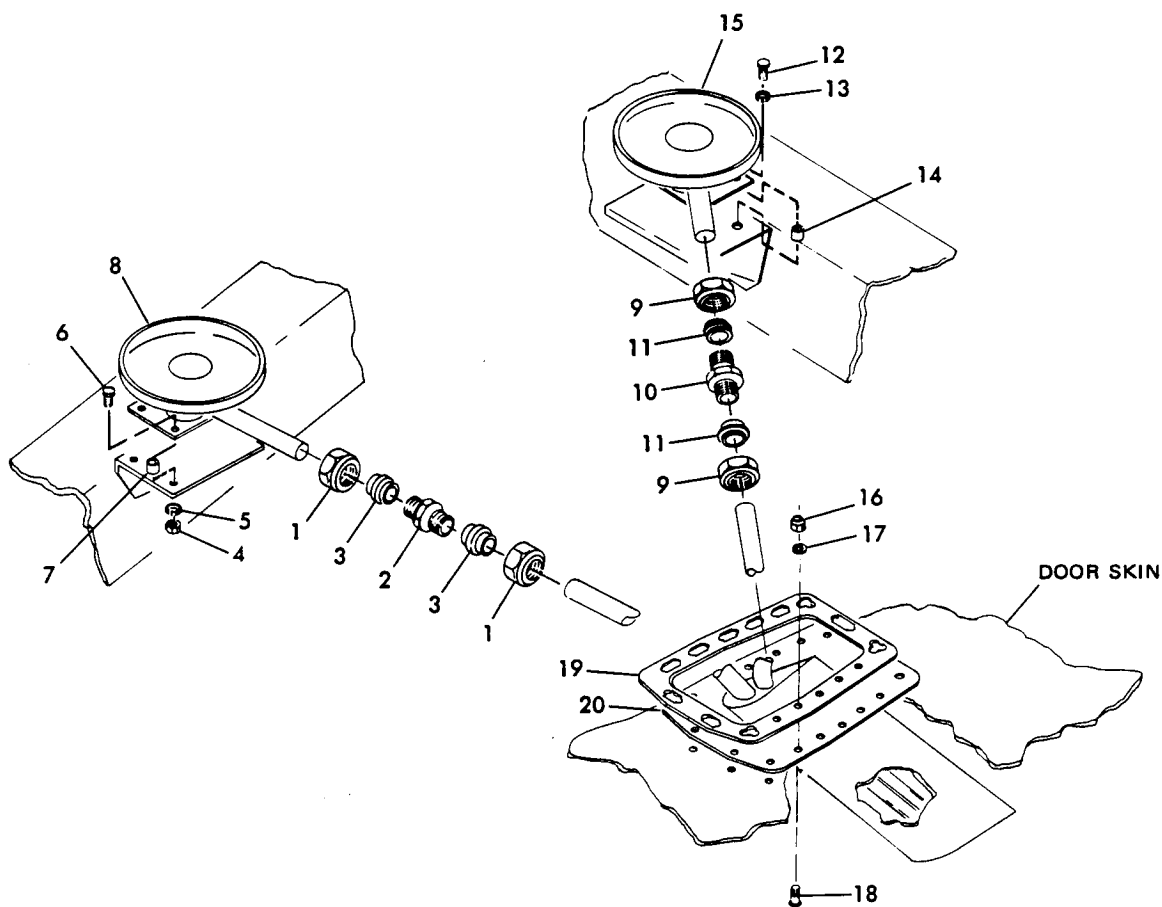
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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1102-			COWL SUPPORT RODS AND ATTACHMENTS								
1	BACN10JC4C										2
2	AN960C416L										2
3	3502-18-04- 4102										2
4	65-50999-501										1
5	65-50999-503										1
6	66-23351-1										1
7	MS24665-151										1
8	BACN10JD103										1
9	AN960PD10L										2
10	BACB30NE3D7										1
11	3539-14-02- 4102										1
12	63-2892										1
13	65-26230-6										1
14	69-20293-1										1
15	21471										1
16	BACC13G208- 065CA										1
17	28-1C										1
17	BACT14A4										1
18	BACT14B2										1
19	BACB30GZ5-12										4
20	BACC30P5										4
21	65-50999-505										1
21	65-50999-507										1
22	65-50999-506										1
22	65-50999-508										1
23	66-23351-1										1
24	MS24665-151										1
25	BACN10JD103										1
26	AN960PD10L										2
27	BACB30NE3D7										1
28	3539-14-02- 4102										1
29	63-2892										1
30	65-26230-6										1
31	69-20293-1										1
32	21471										1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1102-33	BACC13G208-065CA		.	.	.	CABLE ASSEMBLY					1
34	28-1C		.	.	.	SLEEVE, CABLE, V76691 (REPLS BACT14A4)					1
34	BACT14A4		.	.	.	TERMINAL (OPT TO 28-1C)					1
35	BACT14B2		.	.	.	TERMINAL					1
36	BACB30GZ5-12		.	.	LOCKBOLT						4
37	BACC30P5		.	.	COLLAR						4
38	BACN10JC3		.	NUT							8
39	AN960PD10L		.	WASHER							16
40	NAS623-3-5		.	SCREW							2
41	NAS623-3-4		.	SCREW							2
42	NAS623-3-3		.	SCREW							4
43	66-19379-2		.	PLATE ASSEMBLY							2
44	66-19379-3		.	.	PLATE						1
45	BACB28B5-260		.	.	BUSHING						1
46	BACN10JC3		.	NUT							2
47	AN960PD10L		.	WASHER							2
48	NAS623-3-1		.	SCREW							2
49	66-8350		.	CLIP, SPRING							2

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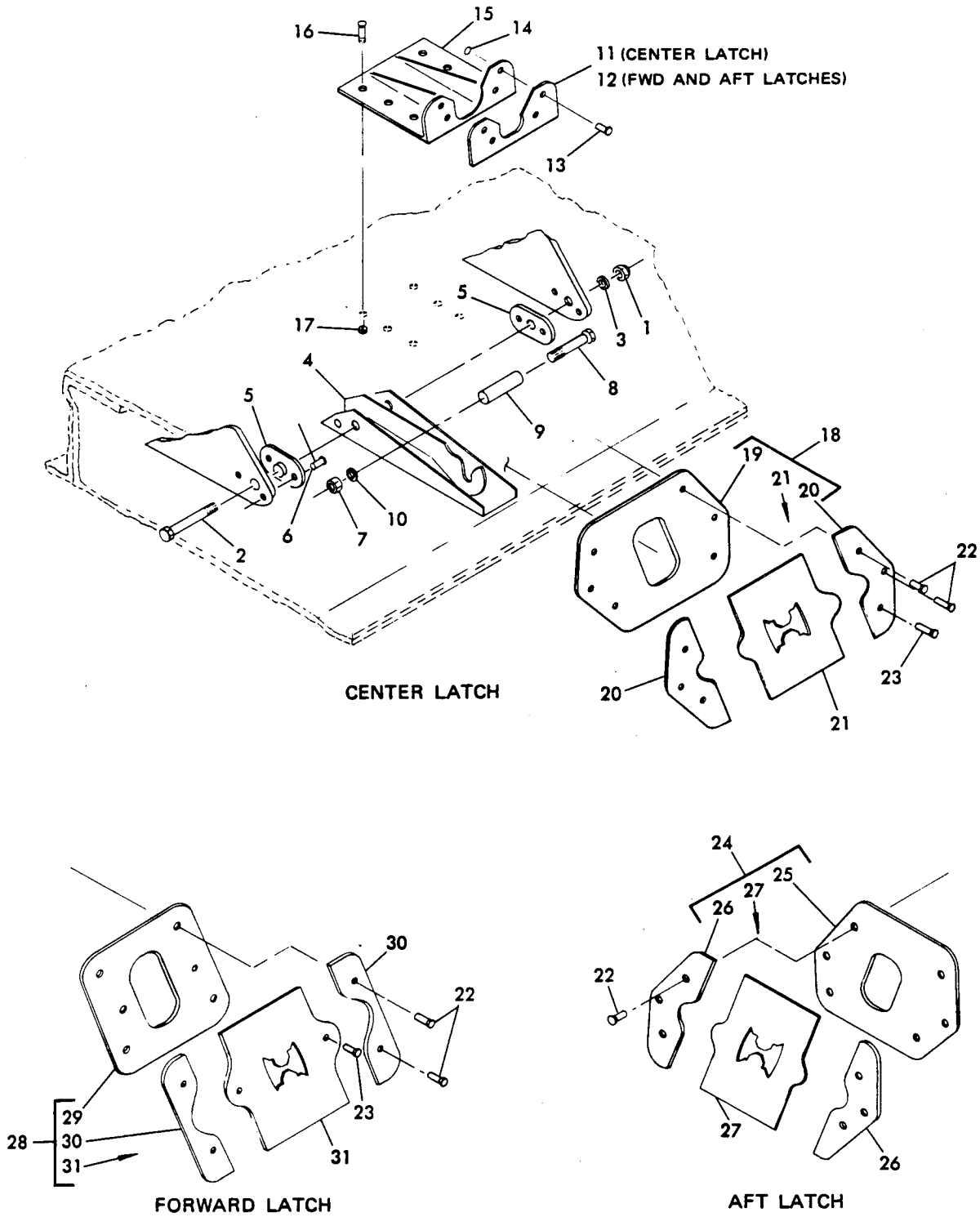


Drains and Attachments
Figure 1103

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1103-			DRAINS AND ATTACHMENTS								
1	MS21921-10S										2
1	MS21921-10K										2
1	MS21921-10J										2
2	MS21902-10C										1
3	MS21922-10C										2
4	BACN10JC3										4
5	AN960C10										4
6	BACB3ONE3-3										4
7	NAS42HT6-8										4
8	69-50896-501										1
8	69-50896-1										1
8	69-50896-8										1
9	MS21921-8S										2
9	MS21921-8K										2
9	MS21921-8J										2
10	MS21902-8C										1
11	MS21922-8C										2
12	BACB3ONE3-3										4
13	AN960C10										4
14	NAS42HT6-8										4
15	69-50896-502										1
15	69-50896-5										1
15	69-50896-9										1
16	BACN10JC3C										5
17	AN960PD10										5
18	BACB30LU3-3										16
19	65-66224-9										1
20	65-66224-11										1
20	65-66224-12										1
20	65-66224-502										1

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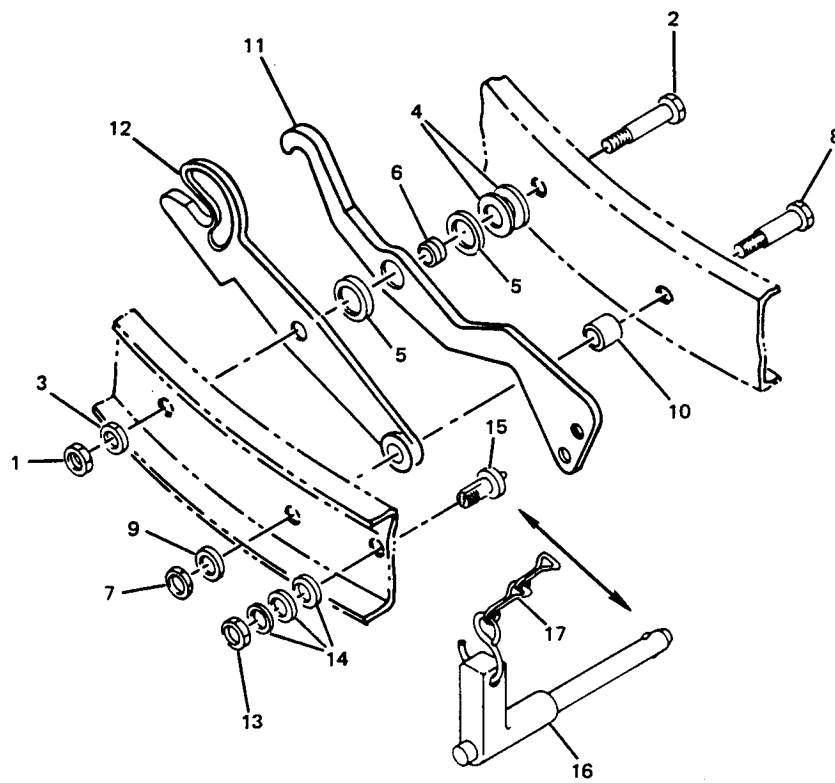


Latches and Seals
 Figure 1104

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1104-			LATCHES AND SEALS								
1	BACN10JC3										3
2	BACB3ONE3-29										3
3	AN960C10L										12
4	H888-1										3
5	69-36658-2										6
6	BACR15CE5D										12
7	BACN10JC3										3
8	BACB3ONE3-25										3
9	NAS43HT3-90										3
10	AN960C10L										3
11	69-37946-1										1
12	69-37946-2										2
13	BACB30FM5-5										12
14	BACC30M5										12
15	69-37944-1										3
16	BACB30GW6-5										18
17	BACC30K6										18
18	69-56008-3										1
19	69-56008-6										1
20	69-56008-7										1
21	69-56008-8										1
22	MS20470D6										14
23	BACR15CE6D										4
24	69-56008-4										1
25	69-56008-9										1
26	69-56008-10										1
27	69-56008-11										1
28	69-56008-5										1
29	69-56008-12										1
30	69-56008-13										1
31	69-56008-14										1

*[1] USED ON 65-50904-271



Hinge and Lock Levers
Figure 1105

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1105-			HINGES AND LOCKING LEVERS								
1	BACN10JC4										2
2	BACB3ONE4-13										2
3	AN960PD416										2
4	AN960C716L										4
5	BACW10P186C										2
6	NAS73-4-003										2
7	BACN10JC4										2
8	BACB3ONE4-13										2
9	AN960PD416										2
10	NAS43HT4-21										2
11	66-24175-2										2
11	66-24175-2										2
11	66-24175-501										2
11	66-24175-5										2
11	66-24175-5										2
12	65-47970-11										2
13	BACN10JC4										2
14	AN960PD416										6
15	69-36659-1										2
16	BACP18AMAL07HC										2
17	BACC19B8-060										2

*[1] LIMITED

*[2] USED ON 65-50904-2

*[3] USED ON 65-50904-271

*[4] PREFERRED SPARE TO 66-24175-2

*[5] USED ON 66-24175-5 ONLY

*[6] USED ON 65-50904-625

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Part No.	Fig. and Index No.	Qty per Assy	Part No.	Fig. and Index No.	Qty per Assy
AN960C10		AR	BACT14A4	1102-34	1
AN960C10L		AR	BACT14B2	1102-18	1
AN960C416L		AR	BACT14B2	1102-35	1
AN960C716L		AR	BACW10P186C	1105-5	2
AN960PD10		AR			
AN960PD10L		AR	H888-1	1104-4	3
AN960PD416		AR			
			MS20470D6		AR
BACB28B5-260	1102-45	1	MS21902-10C		AR
BACB30FM5-5	1104-13	12	MS21902-8C		AR
BACB30GW6-5	1104-16	18	MS21921-10J		AR
BACB30GZ5-12	1102-19	4	MS21921-10K		AR
BACB30GZ5-12	1102-36	4	MS21921-10S		AR
BACB30LU3-3	1103-18	16	MS21921-8J		AR
BACB30NE3-25	1104-8	3	MS21921-8K		AR
BACB30NE3-29	1104-2	3	MS21921-8S		AR
BACB30NE3-3	1103-6	4	MS21922-10C		AR
BACB30NE3-3	1103-12	4	MS21922-8C		AR
BACB30NE3D7	1102-10	1	MS24665-151		AR
BACB30NE3D7	1102-27	1			
BACB30NE4-13	1105-2	2	NAS42HT6-8		AR
BACB30NE4-13	1105-8	2	NAS43HT3-90		AR
BACCC13G208-065CA	1102-16	1	NAS43HT4-21		AR
BACCC13G208-065CA	1102-33	1	NAS623-3-1		AR
BACCC19B8-060	1105-17	2	NAS623-3-3		AR
BACC30K6	1104-17	18	NAS623-3-4		AR
BACC30M5	1104-14	12	NAS623-3-5		AR
BACC30P5	1102-20	4	NAS73-4-003		AR
BACC30P5	1102-37	4			
BACN10JC3	1102-38	8	21471	1102-15	1
BACN10JC3	1102-46	2	21471	1102-32	1
BACN10JC3	1103-4	4	28-1C	1102-17	1
BACN10JC3	1104-1	3	28-1C	1102-34	1
BACN10JC3	1104-7	3			
BACN10JC3C	1103-16	5	3502-18-04-4102	1102-3	2
BACN10JC4	1105-1	2	3539-14-02-4102	1102-11	1
BACN10JC4	1105-7	2	3539-14-02-4102	1102-28	1
BACN10JC4	1105-13	2			
BACN10JC4C	1102-1	2	63-2892	1102-12	1
BACN10JD103	1102-8	1	63-2892	1102-29	1
BACN10JD103	1102-25	1	65-26230-6	1102-13	1
BACP18AMAL07HC	1105-16	2	65-26230-6	1102-30	1
BACR15CE5D	1104-6	12	65-47970-11	1105-12	2
BACR15CE6D	1104-23	4	65-50904-2	1101	
BACT14A4	1102-17	1	65-50904-271	1101	
			65-50904-625	1101	

BOEING 
COMMERCIAL JET
OVERHAUL MANUAL

Part No.	Fig. and Index No.	Qty per. Assy	Part No.	Fig. and Index No.	Qty per. Assy
65-50999-501	1102-4	1			
65-50999-503	1102-5	1			
65-50999-505	1102-21	1			
65-50999-506	1102-22	1			
65-66224-11	1103-20	1			
65-66224-9	1103-19	1			
66-19379-2	1102-43	2			
66-19379-3	1102-44	1			
66-23351-1	1102-6	1			
66-23351-1	1102-23	1			
66-24175-2	1105-11	2			
66-24175-5	1105-11	2			
66-24175-501	1105-11	2			
66-8350	1102-49	2			
69-20293-1	1102-14	1			
69-20293-1	1102-31	1			
69-36658-2	1104-5	6			
69-36659-1	1105-15	2			
69-37944-1	1104-15	3			
69-37946-1	1104-11	1			
69-37946-2	1104-12	2			
69-50896-1	1103-8	1			
69-50896-5	1103-15	1			
69-50896-501	1103-8	1			
69-50896-502	1103-15	1			
69-56008-10	1104-26	1			
69-56008-11	1104-27	1			
69-56008-12	1104-29	1			
69-56008-13	1104-30	1			
69-56008-14	1104-31	1			
69-56008-3	1104-18	1			
69-56008-4	1104-24	1			
69-56008-5	1104-28	1			
69-56008-6	1104-19	1			
69-56008-7	1104-20	1			
69-56008-8	1104-21	1			
69-56008-9	1104-25	1			