TO: ALL HOLDERS OF WING LEADING EDGE SLAT ASSEMBLIES NO. 1 AND 6 OVERHAUL MANUAL, 57-56-01

REVISION NO. 24, DATED JUL 1/09

HIGHLIGHTS

DESCRIPTION OF CHANGE	TOPICS AFFECTED												
	D&O	D/Assy	Cleaning	Insp/Chk	Repair	Assy	F/C	Test	T/Shooting	S/Tools	Storage	IPL	L/Overhaul
Added (optional) ring, item 212 in IPL												X	

WING LEADING EDGE SLAT ASSEMBLIES NO. 1 AND NO. 6

57-56-01

BOEING P/N 65-46421-107, -108, -155, -156, -165, -166, -167, -168, -195, -196, -197, -198, -199, -200, -201, -202, -203 thru -208, -213, -214, -217, -218, -221 thru -224, -227 thru -238

AIRLINE P/N

THE FOLLOWING DIRECTIVES APPLY TO THIS SUBJECT:

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVES	DATE DIRECTIVE INCORPORATED INTO TEXT
57-1013, Rev 1		PRR 31252	Sep 10/70
57-1023		PRR 31305	Sep 10/70
57-1028, Rev 2		PRR 31310	Sep 10/70
30-1011		PRR 31960-15	Jun 10/72
		PRR 32121-9	Jun 10/72
		PRR 32121-17	Jun 10/72
		PRR 32121-30	Dec 25/72
57-1080			Dec 25/73
		PRR 32597	Jul 5/76
57-1100			Jan 5/77
		PRR 32767-1	Jan 5/78
		PRR 32767-3	Jan 5/78
		PRR 32767-2	Jul 5/78
30-1015		PRR 32767-4	Jan 5/79
		PRR 32757-3	Jan 5/79
		PRR 32708	Jan 5/79
		PRR 32944	Jan 5/80
		PRR 33180-30	Jun 5/84
		PRR 33316	Mar 5/85
57-1080, Rev 1			Mar 5/90
57-1080, Rev 3	57-25		Dec 1/96
57-1080, Rev 2			Mar 1/05

LIST OF EFFECTIVE PAGES

* Indicates pages revised, added or deleted in latest revision
F Indicates foldout pages - print one side only

	PAGE	DATE		PAGE	DATE		PAGE	DATE
	57-56-01			1111	Mar 1/05			
	T-1	Mar 1/05		1112	Mar 1/05			
	T-2	BLANK		1113	Mar 1/05			
*	LEP-1	Jul 1/09		1114	Mar 1/05			
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	201	Sep 10/70		1118F	BLANK			
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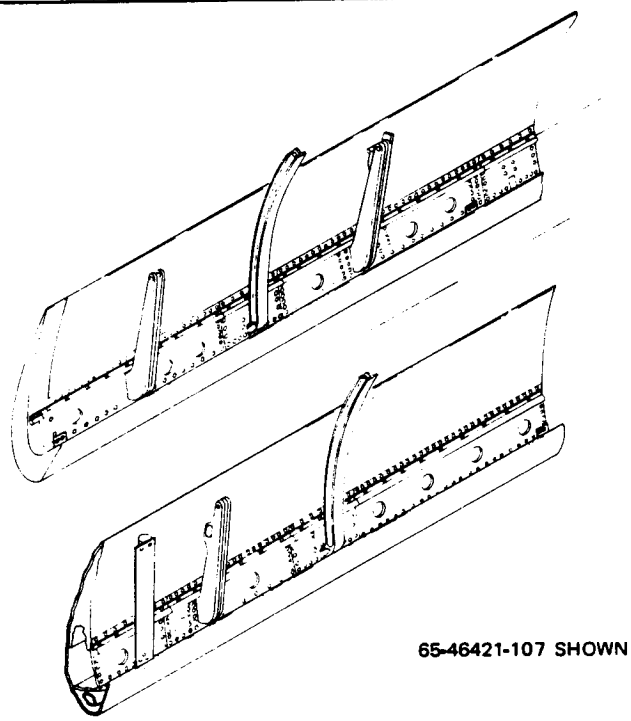


Wing Leading Edge Slat Assemblies No. 1 and No. 6
Figure 1

DESCRIPTION AND OPERATION

1. Description

   A. The lift capability of the 737 wing is supplemented by a set of two Krueger type flaps and three slats, which are installed on the leading edge of the wing. These lightweight leading edge devices work in coordination with the trailing edge flaps to improve operation of the airplane at low speeds, and enable takeoff in minimum distance.

B. Three slats are installed on the leading edge of each wing between the wing tip and the engine nacelle. The slats are numbered 1 through 3, from left to right, starting at the left wing tip; and 4 through 6, starting at the right engine nacelle. Each slat on the right wing is the opposite of its counterpart on the left wing.

C. Each slat is of conventional sheetmetal construction, with a trailing edge made of honeycomb sandwich structure. The basic structure is a leading edge beam, to which chordwise ribs are attached. These ribs are covered with inner and outer alclad skins. Anti-icing ducts are installed in the enclosed area just ahead of the leading edge beam, and are attached to the ends of the slat. These tubes are fed by a telescoping duct which attaches to the main thermal anti-icing system of the wing. The slat is supported by two tracks which move between roller bearings attached to support ribs in the wings. It is supported also by three auxiliary roller tracks installed between the main tracks.

2. Operation

A. Each leading edge slat is actuated by a two-position cylinder type hydraulic actuator. An integral part of each actuator is a locking mechanism which locks the slat in either the fully extended or the fully closed position. This precludes possibility of the slat being blown open or closed if hydraulic power fails.

DISASSEMBLY

1. Place wing leading edge slat assembly in a suitable holding fixture, and disassemble as follows: (See figure 1101.)

   A. Remove seals and retainers.

      (1) Remove bolts (1), retainer (2) and seal (3).

      (2) Remove bolts (4), retainer (5) and seal (6).

      (3) Remove seal (7) by scraping off with a sharp-edged wooden or plastic tool.

      (4) Remove bolts (8), retainer (9) and seal (10).

      (5) Remove bolts (11), retainer (12) and seal (13).

      (6) Remove bolts (14), retainer (15) and seal (16).

      (7) Remove bolts (17), retainer (18) and seal (19).

      (8) Remove bolts (20), retainer (21) and seal (22).

      (9) Remove bolts (23), retainer (24) and seal (25).

      (10) Remove bolts (26), retainer (27) and seal (28).

      (11) Remove bolts (29), retainer (30) and seal (31).

      (12) Remove seal (32) by scraping off with a sharp-edged wooden or plastic tool.

      (13) Remove bolts (33), retainer (34) and seal (35).

   B. Remove inner skins and tip fairing assembly.

      NOTE: Do not remove rivets (36, 39 and 42), bolts (37, 40 and 43), and upstops (38, 41 and 44) unless repair or replacement is required.

      (1) Remove bolts (45) and tip fairing assembly (46).

      (2) Remove bolts (47 and 48), washers (49), and inner skin (50).

(3) Remove bolts (51, 52, and 53), washers (54), and clip-on nuts (55). Remove inner skin (56) and clamps (58) if applicable.

(4) Remove grommet (57) from inner skin (56).

(5) Remove bolts (59 and 60), washers (61), and inner skin (62).

(6) Remove grommet (63) from inner skin (62).

(7) Remove bolts (64 and 65), washers (66), and inner skin (67).

C. Remove track assemblies, arm assemblies and associated parts.

NOTE: Serrated plates (93 and 94), aligning clips (95 and 96), bushings (99 and 100), and arm assemblies (101, 113, 113A, 127 and 127A) are shown for reference. These parts, and components of arm assemblies, are permanently installed and should not be removed unless parts must be replaced.

(1) Remove nuts (69), washers (70), and bolts (71). If nuts, P/N BACN10JD5 are used, remove cotter pins (68) before removing nuts.

(2) Remove nuts (73), washers (74), bolts (75), and serrated washers (76). If nuts, P/N BACN10JD6 are used, remove cotter pins (72) before removing nuts. Remove track assemblies (77) and track attach fitting assemblies (80 and 84).

(3) Separate track assemblies from track attach fitting assemblies by removing cotter pins (88), nuts (89), washers (90, 91A, 91B, 91C) if installed, bolts (91), and bushings (92).

D. Remove anti-icing duct installation, spray tube installation and associated parts from basic structure (211).

NOTE: Items (148 through 210) removed in this paragraph are a part of the anti-icing duct installation and spray tube installation and may, or may not, be installed on the assembly when it is received for overhaul.

(1) Remove screws (149) and washers (150).

(2) Remove spray tube assembly (151) from slat assembly by pulling tube out through opening in end-closure rib.

(3) Remove bolts (152).

(4) Remove spray tube assembly (153) from slat assembly by pulling tube out through opening in end-closure rib.

(5) Remove bolts (155), shim (156), and plate assembly (157), if applicable, loosening one end of door assembly (164).

NOTE: Do not remove rivets (158) and nutplates (159) from plate (160) unless repair or replacement is required.

(6) Remove screws (161), washers (162), shim (163), and door assembly (164).

(7) Remove bolts (165), plate assembly (166), and hinge assembly (170), or rivets (183) and hinge assembly (190), as applicable, from door assembly (164).

NOTE: Do not remove rivets (167) and nutplates (168) from plate (169) unless repair or replacement is required.

(8) Disassemble hinge assembly (170) if applicable.

(a) Remove hinge pin (176), separating hinge halves (177 and 178).

(b) Remove hinge pin (180), separating hinge halves (181 and 182).

NOTE: Do not remove nut (171), washers (172), screw (173) and rivets (174), separating hinge assemblies (175 and 179), unless repair or replacement is required.

Do not remove rivets (184) from hinge assemblies (185 and 190) unless repair or replacement is required.

(9) Remove pins (186 and 187), separating hinge halves (188 and 189).

(10) Remove pins (191 and 192), separating hinge halves (193 and 194).

(11) Remove pins (197 and 198), separating hinge halves (199 and 203).

NOTE: Do not remove rivets (195) from hinge assembly (196).

Do not remove rivets (201) and nutplates (202) from hinge half (203) unless repair or replacement is required.

Do not remove rivets (204) attaching channel (205) to panel (206) unless repair or replacement is required.

(12) Remove bolts (141, 142, 143) and doubler (144), or bolts (147D, 147E, 147F) and doublers (147A, 147B) and filler (147C) from beam on slat assembly.

(13) Remove bearings (207) from tee assembly (208).

(14) Remove tee assembly (208) from slat assembly through opening in slat beam.

(15) Remove O-ring packing (209) from retainer on tee assembly (210).

CLEANING

1. General

   A. Wash and rinse all metal parts except bearings in solvent, Specification P-D-680, or equivalent.

   B. Remove stubborn accumulations of dirt with a stiff-bristle brush. Do not use a metallic brush.

   C. On slat assemblies incorporating Boeing Service Bulletin Number 57-1028, Revision 2, if the slat is not to be installed on the airplane from which it was removed, remove the aerodynamic smoother and clean slat surfaces. Refer to Subject 20-60-01, Cleaners.

   D. Dry parts thoroughly with clean, lint-free cloth, or with dry, compressed air.

   E. For further information, refer to Subject 20-30-03, General Cleaning Procedures.

2. Bearings

   A. Clean all bearings per Subject 20-30-01, Cleaning and Relubricating Antifriction bearings.

      CAUTION: BEARING (79, FIGURE 1101) IS TEFLON LINED. CLEAN ONLY BY SPECIAL METHOD GIVEN IN REFERENCE SUBJECT.

INSPECTION/CHECK

1. Visual Check

A. Examine all metal parts for cracks, corrosion, scratches, dents, and damage. Use a strong light and a minimum of 10-power magnification.

B. Examine all painted or plated surfaces for defects.

C. Check bearings for corrosion, roughness, binding, and excessive radial and axial play.

D. Check all bushing and bolt holes for excessive and eccentric wear.

E. Examine entire basic structural assembly for corrosion, loose fasteners, damage, and general condition of paint and finish.

F. Examine honeycomb and bonded structure for evidence of delamination, internal moisture, scratches, and contour defects.

(1) Tap surface of honeycomb structure lightly with a coin or plastic rod. Go over entire surface. Normal structure will produce a solid, metallic sound; delaminated areas will produce a dull, hollow sound; and areas containing moisture will produce a dull, solid sound.

(2) Examine areas suspected of containing moisture radiographically to determine extent of damage.

(3) Determine contour defects by laying a straightedge across the surface of the panel. Raised areas indicate delamination. Warp of panels also can be determined with the straightedge.

(4) Examine edges of panel carefully for cuts and abrasions. Delamination starts very easily from damage to an edge member of a honeycomb panel.

G. On slat assemblies incorporating Boeing Service Bulletin Number 57-1028, Revision 2, examine surface of aerodynamic smoother and sealant for defects, cuts, cracks and abrasion as applicable.

2. Special Check (See Fig. 1101.)

A. If visual examination discloses evidence of defects in any of the parts listed, perform the following checks:

CAUTION: TO PREVENT DAMAGE TO ANTIFRICTION BEARINGS, IT IS NECESSARY TO PROTECT THEM FROM INSPECTION FLUID WHEN PERFORMING MAGNETIC PARTICLE OR DYE PENETRANT EXAMINATION OF COMPONENTS CONTAINING BEARINGS. AN ADEQUATE EXAMINATION CAN BE MADE BY CAREFULLY MASKING OFF THE BEARING AND APPLYING THE FLUID BY BRUSH INSTEAD OF BY DIPPING.

NOTE: It is not necessary to press out a bearing to check inside of component bore unless crack indications are detected during visual examination.

(1) Dye penetrant check -- track attach fittings (81 and 85), arms (102, 114 or 114A, and 128 or 128A), ribs-auxiliary track (102B, 102C), and spray tube assemblies (151 and 153).

(2) Magnetic particle check -- track (78) and serrated plates (93 and 94).

REPAIR

1. Repair (Fig. 1101)

   A. Remove minor defects with standard industry practices. Refer to Fits and Clearances for design dimensions and wear limits.

   B. For slat repair, refer to 57-50-3, Boeing 737 Structural Repair Manual.

   C. For repairs to honeycomb structure, refer to 51-40-6 Boeing 737 Structural Repair Manual.

   D. Refer to 36-10-03 for repair of spray tube assemblies (151, 153) and tee assembly (210).

   E. Track assembly (77), 65-49448-1, -7, -11:

      (1) Blend out worn areas up to acceptable upper or lower flange thickness of 0.09-0.11 inch for 65-49448-1, -7, and 0.10- 0.12 inch for 65-49448-11.

      (2) Shot peen the blended areas per SOPM 20-10-03, shot size 0.016-0.033, intensity 0.015A2.

2. Refinish (Fig. 1101)

   NOTE: Refer to 20-30-02 for stripping of protective finishes and to 20-41-01 for explanation of F and SRF finish codes.

   A. Track (78) -- Cadmium-titanium plate (F-1.308, which replaces F-1.181) all over. Then magnetic particle examine. Apply primer BMS 10-11, type 1 (SRF-12.205) and gray enamel BMS 10-11, type 2 (SRF-12.63) all over, but not in bearing hole, and no enamel on surfaces touched by slat rollers. Material: 4340M steel, 270-300 ksi.

   B. Fittings (81, 85) -- Chemical treat or chromic acid anodize and apply primer BMS 10-11, type 1 (SRF-2.30) all over, but no primer in bearing hole. Material: Al alloy.

   C. Serrated plates (93, 94) -- Cadmium plate (F-1.1926) all over and apply primer BMS 10-11, type 1 (SRF-12.205) all over, but no primer on serrations. Material: 4340 steel, 180-200 ksi.

   D. Arms (102, 114, 128) -- Chemical treat or chromic acid anodize and apply primer BMS 10-11, type 1 (SRF-2.30) all over, but no primer in bushing holes. Material: Al alloy.

E. Arms (114A, 65-55526-7, -8) -- Chemical treat or chromic acid anodize and apply one coat primer BMS 10-11, type 1 (SRF-2.30) and gray gloss enamel BMS 10-60, color BAC707 (SR-14.9813) all over, except no primer or paint in holes. Material: Al alloy.

F. Arms (128A, 65-55525-9, -10) -- Chemical treat or chromic acid anodize and apply one coat primer BMS 10-11, type 1 (SRF-2.30) and gray gloss enamel BMS 10-60, color BAC707 (SRF-14.9813) all over, except no primer or paint in holes. Material: Al alloy.

G. Door (164, 69-38641-29) -- Clear chemical treat (SRF-14.011) on all surfaces, followed by one coat primer BMS 10-11, type 1 and gloss enamel BMS 10-60, type 2 (F-14.9863-7067) on all internal surfaces and edges. Material: Al alloy.

H. Skin (50; 56, 65-46421-173 and -174; 62, 65-46421-169 and -170; and 67) -- Chemical treat the interior and exterior surfaces, and apply one coat of primer BMS 10-11, Type 1 (F-18.06) and apply finishes as shown in Fig. 401. Material: Al alloy

I. Retainer (2, 5, 12, 15, 18, 21, 27, 30, 34) -- Chromic acid anodize (F-2.26).

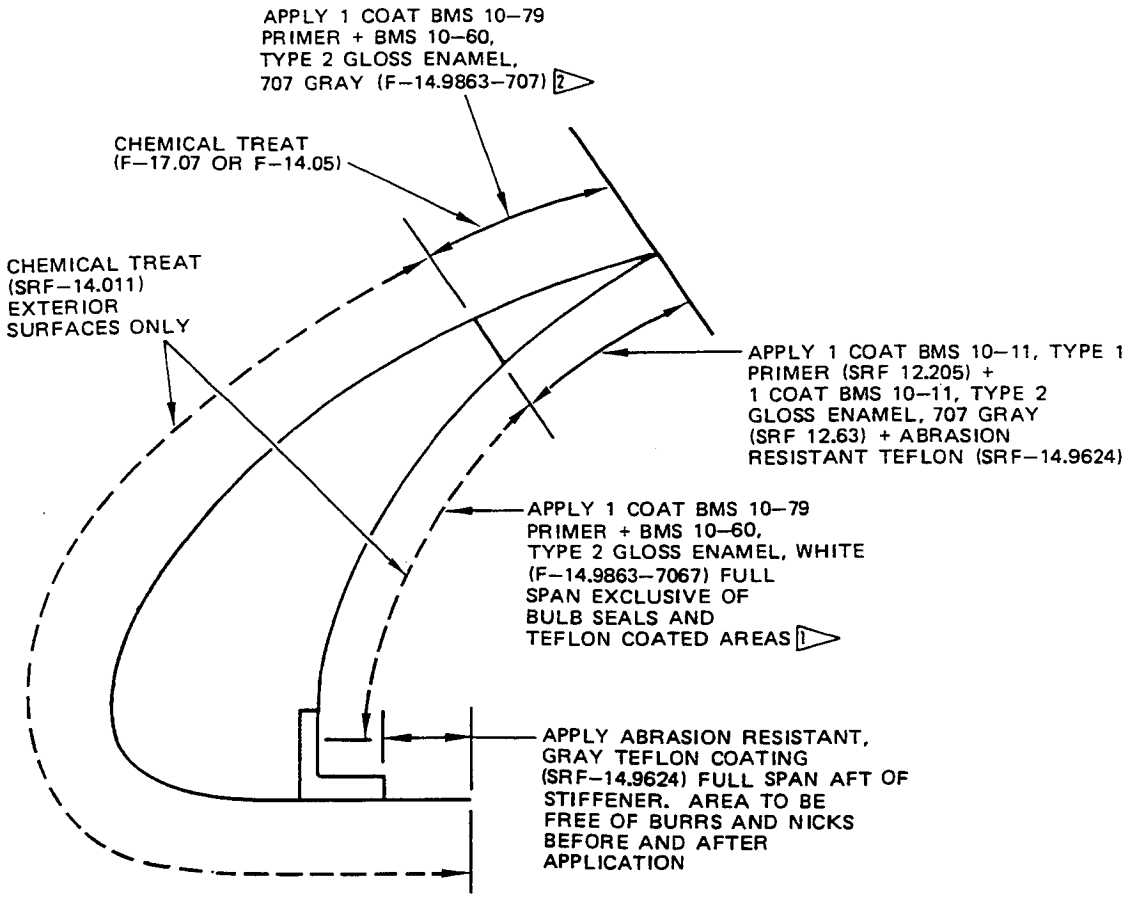
J. Deleted

K. Doublers (147A, 147B), filler (147C) -- Chromic acid anodize and apply one coat primer BMS 10-11, Type 1 (F-18.05) all over. Material: Al alloy.

L. Touch up all interior surfaces of basic assembly with primer, BMS 10-11, Type 1.

M. On slat assemblies incorporating Boeing Service Bulletin Number 57-1028, Revision 2, restore aerodynamic smoother and sealant as applicable if slat is to be installed on airplane from which it was removed. For information on aerodynamic smoothing sealant, refer to 20-50-11, Application of Aerodynamic Smoothing Sealant.

N. Bearing (215) -- Apply dry film lubricant BMS 3-8, class A, per 20-50-08, method 3, to ID only. Material: Al-bronze.



1 ASSEMBLIES 65-45421-199, -200, -205 THRU -208, -213, -214, -217, -218, -221, -222. PREFERRED OPTIONAL FINISH ON EARLIER ASSEMBLIES.

2 ASSEMBLIES 65-46421-217, -218, -221, -222. PREFERRED OPTIONAL FINISH ON EARLIER ASSEMBLIES.

SLAT ASSEMBLY SECTIONAL VIEW

Refinish Details
Figure 401

3. Replacement (Fig. 1101)

A. Replace all parts worn beyond simple repair.

B. Replace all cotter pins at each overhaul.

C. Replace O-ring packing (209) and all rubber or fabric seals at each overhaul.

D. Replace doubler (144) if installed, and attaching hardwares with doublers (147A, 147B) and filler (147C) and attaching hardware. Bond filler (147C) on doubler (147B) per 20-50-12, type 60. If nutplates (147I) installed on doubler (147B) require replacement, carefully drill out rivets (147J). Install serviceable items (147I) in doubler (147B) with new rivets (147J).

E. Replace seals (7 and 32) by bonding in place as directed in 20-50-12, type 60.

F. Replace grommets (57 and 63) at each overhaul.

(1) Trim grommet (57, BACG20ZA790) to fit cutout in inner skin (56) and bond in place as directed in 20-50-12, type 38. Install grommet (57, NAS1368N3B) by pressing into cutout. Do not bond.

(2) Trim grommet (63, BACG20ZA400 or BACG20ZA1275) to fit cutout in inner skin (62), and bond in place as directed in 20-50-12, type 38.

G. Replace defective bearings (79) as follows:

(1) Cut old bearing out of housing. Coat faying surfaces of new bearing and housing with grease, Specification MIL-G-23827, and install new bearing in housing. Roller swage bearing in place as directed in 20-50-03, Bearing Installation and Retention.

H. Replace all worn or defective bushings as follows:

(1) Press old bushing out of housing with a mandrel.

(2) Coat faying surfaces of new bushing and housing with primer, Specification BMS 10-11, type 1, and press bushing into housing.

I. Replace worn or damaged installation items.

(1) If nutplates (159 and 168) installed on plates (160 and 169) require replacement, carefully drill out rivets (158 and 167), as applicable, to separate the parts. Reassemble serviceable items (159 and 168) on plates (160 and 169) with new rivets (158 and 167).

NOTE: If hinge half (178 or 181) requires replacement, replace both hinge halves. Do not install rivets (174) on replacement hinge halves.

(2) Replace worn or damaged hinge half (200). However, if nutplates (202) installed on hinge half (203) require replacement, carefully drill out rivets (201) to separate the parts. Reassemble serviceable items (202 and 203) with new rivets (201).

(3) If channel (205) or panel (206) requires replacement, carefully drill out rivets (183 and 195), if applicable, and rivets (204) to separate parts. Reassemble serviceable items (205 and 206) with new rivets (204). Reinstall hinge halves (177 or 194, and 199), if applicable, with new rivets (183 and 195).

ASSEMBLY

1. Place basic structure (211) in a suitable holding fixture, and assemble parts as follows (Fig. 1101):

   A. Install anti-icing duct installation, spray tube installation and associated parts.

      NOTE: Items (148 thru 219) installed in this paragraph are a part of the anti-icing duct installation and spray tube installation; therefore, installation of these items may or may not be required during overhaul.

      (1) Install O-ring packing (209) in retainer on tee assembly (208).

      (2) Install tee assembly (208) in slat assembly through opening in slat beam.

      (3) Install bearings (207 or 215) on tee assembly (208).

      (4) Install doublers (147A, 147B) and filler (147C) on slat assembly with bolts (147D, 147E, 147F). Bond filler (147C) on doubler (147B) per 20-50-12, type 60. Coat bolts (147D, 147E, 147F) with grease, MIL-G-23827, before installing.

      (5) Assemble hinge halves (200 and 199) by installing pins (198 and 197).

      (6) If hinge assembly (190) is to be installed, assemble hinge halves (194 and 193) by installing pins (192 and 191).

      (7) If hinge assembly (185) is to be installed, assemble hinge halves (189 and 188) by installing pins (187 and 186).

         NOTE: Rivets (184) are installed after adjusting slat position on the wing.

      (8) If hinge assembly (170) is to be installed, assemble parts as follows:

         (a) Assemble hinge halves (182 and 181), on hinge assembly (179), by installing hinge pin (180).

         (b) Assemble hinge halves (178 and 177), on hinge assembly (175), by installing hinge pin (176).

         (c) Install but do not tighten screw (173), washers (172) and nut (171) on hinge assembly (170).

            NOTE: Rivets (174) are installed after adjusting slat position on the wing.

(9) Install door assembly (164) on tee assembly (208) with shim (163), washers (162), and bolts (161).

(10) Attach door assembly (164) to slat structure with shim (156), plate assembly (157), if applicable, and bolts (155).

(11) Prepare faying surfaces of slat structure and end flange on spray tube assembly (153) for electrical bond as described in 20-11-03.

(12) Install spray tube assembly (153) in slat assembly by sliding tube through opening in end-closure rib. Seat end of tube in bearing (207 or 215). Secure spray tube assembly (153) to slat end-closure rib with bolts (152).

(13) Check resistance across bond with an ohmmeter. Maximum allowable resistance is 0.010 ohm. Remove tube and repeat steps (11) and (12), if necessary , to obtain acceptable bond.

(14) Prepare faying surfaces of slat structure and end flange on spray tube assembly (151) for electrical bond as described in 20-11-03.

(15) Install spray tube assembly (151) in slat assembly by sliding tube through opening in end-closure rib. Seat end of tube in bearing (207 or 215). Secure spray tube assembly (151) to slat end-closure rib with washers (150) and screws (149).

(16) Check resistance across bond with an ohmmeter. Maximum allowable resistance across bond is 0.010 ohm. Remove tube and repeat steps (14) and (15), if necessary, to obtain acceptable bond.

(17) Assemble items (211 thru 218) on tee duct assembly (219).

(18) Tighten end cap (211), 250-370 lb-in. and lockwire, using double twist method per 20-50-02.

(19) Coat inside diameter of duct assembly (219) with grease MIL-G-23827.

B. Install track assemblies, arm assemblies and associated parts.

(1) Apply a layer of grease, MIL-G-23827, to bolts (91). Attach track assemblies (77) to track attach fitting assemblies (80, 84) with bushings (92), bolts (91), washers (91A, 90), nuts (89), and cotter pins (88). Install washers (91B, 91C), as necessary, to get the gap shown in Fig. 501.

(2) Locate track attach fitting assembly (84) on slat. Coat bolts (71 and 75) with grease, MIL-G-23827. Install bolts with serrated washers (76), washers (70 and 74), and nuts (69 and 73). If castellated nuts are used, loosely install cotter pins (68 and 72).

NOTE: Leave nuts and cotter pins loose. Position of track attach fitting must be adjusted when slat is installed on wing.

(3) Locate track attach fitting assembly (80) on slat. Coat bolts (71 and 75) with grease, MIL-G-23827. Install bolts with serrated washers (76), washers (70 and 74), and nuts (69 and 73). If castellated nuts are used, loosely install cotter pins (68 and 72).

NOTE: Leave nuts and cotter pins loose. Position of track attach fitting must be adjusted when slat is installed on wing.

C. Install inner skins and tip fairing assembly.

NOTE: Do not tighten bolts. Skin panels must be removed to adjust position of attach fittings when slat is installed on wing. Coat all bolts with grease, MIL-G-23827, before installation.

(1) Install inner skin (67) with washers (66) and bolts (64 and 65).

(2) Install inner skin (62) with washers (61) and bolts (59 and 60).

(3) Install inner skin (56) and clamps (58), if applicable, with bolts (51, 52 and 53), washers (54), and clip-on nuts (55), if applicable.

(4) Install inner skin (50) with washers (49) and bolts (47 and 48).

(5) Install tip fairing assembly (46) with bolts (45).

D. Install seals and retainers.

NOTE: All seals are fabricated with excess material on each end. Trim each seal on installation to match adjacent seals. Allowable gap between ends of seals is 0.00 to 0.06 inch, except as noted.

Coat all bolts attaching seals and retainers with grease, MIL-G-23827, before installation.

(1) Install seal (35) and retainer (34) with bolts (33).

(2) Install seal (31) and retainer (30) with bolts (29).

(3) Install seal (28) and retainer (27) with bolts (26).

(4) Install seal (25) and retainer (34) with bolts (33).

(5) Install seal (22) and retainer (34) with bolts (33).

(6) Install seal (19) and retainer (34) with bolts (33).

(7) Install seal (16) and retainer (34) with bolts (33).

(8) Install seal (13) and retainer (34) with bolts (33).

(9) Install seal (10) and retainer (34) with bolts (33).

(10) Install seal (6) and retainer (34) with bolts (33).

(11) Install seal (3) and retainer (34) with bolts (33).

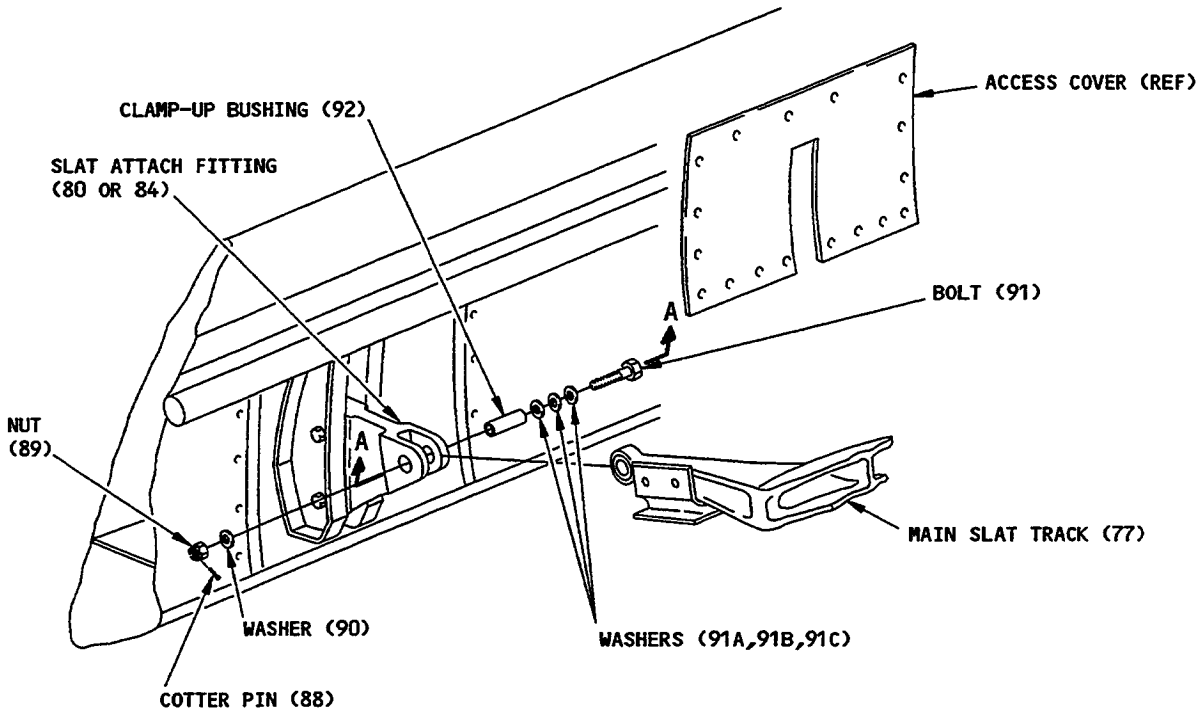
2. Materials

NOTE: Equivalent subsstitutes can be used.

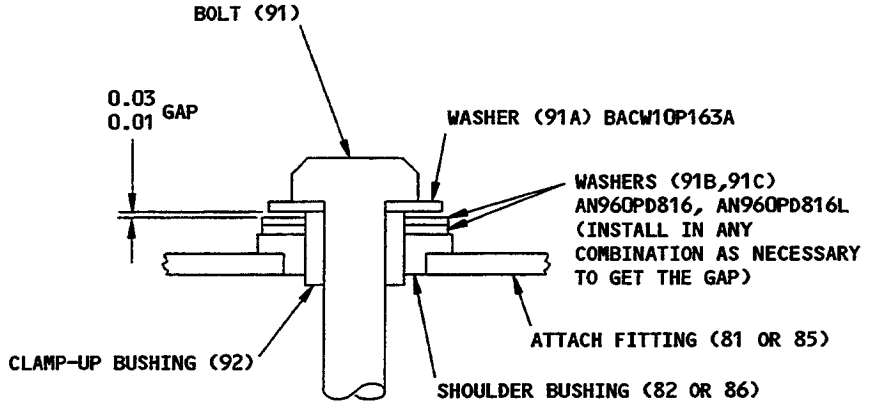
A. Adhesive -- Type 60 (Ref. 20-50-12)

B. Grease -- MIL-G-23827 (Ref 20-60-03)

C. Primer -- BMS 10-11, Type 1 (Ref 20-60-02)



TYPICAL SLAT TRACK INSTALLATION



A-A

ITEM NUMBERS REFER TO FIG. 1101
ALL DIMENSIONS ARE IN INCHES

Post SB 57-1080 Slat Track to Fitting Installation Details
Figure 501

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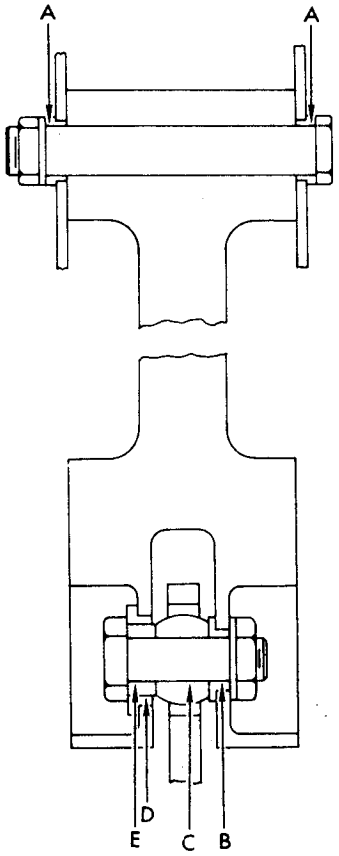
FITS AND CLEARANCES

1. The fits and clearances table lists design dimensions and service wear limits for close tolerance parts of the assembly that are subject to wear or corrosion. Unless otherwise specified, parts should be returned to the design dimensions whenever rework is accomplished.

2. Clearances are given to aid assembly of the components. The values given in the Maximum Allowable Clearance column are the maximum permitted to ensure proper functioning of the unit. If assembled parts fail to meet this requirement, one or more of the parts must be rejected. Parts that are rejected should be reworked if within the rework limits given in the Repair procedure; if not within rework limits, the parts should be scrapped. It is recommended that the design clearances be used as the guiding assembly criteria when newly reworked parts are assembled.



Fits and Clearances
Figure 601 (Sheet 1)

Ref Letter Fig.601	Mating Item No. Fig.1101		Design Dimensions				Service Wear Limits		
			Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inches)		Maximum Allowable Clearance (inch)
			Min	Max	Min	Max	Min	Max	
A	ID	99	0.3125	0.3140	0.0005	0.0030	0.3065	0.3180	0.0060
	OD	71	0.3110	0.3120					
B	ID	83,87	0.3125	0.3140	0.0005	0.0030	0.0365	0.3180	0.0060
	OD	91	0.3110	0.3120					
C	ID	79	0.3120	0.3125	0.0000	0.0015	0.3090	0.3150	0.0030
	OD	91	0.3110	0.3120					
D	ID	82,86	0.5000	0.5015	0.0020	0.0050	0.4910	0.5070	0.0090
	OD	92	0.4960	0.4980					
E	ID	92	0.3120	0.3135	0.0000	0.0025	0.3070	0.3170	0.0050
	OD	91	0.3110	0.3120					

Fits and Clearances
Figure 601 (Sheet 2)

STORAGE INSTRUCTIONS

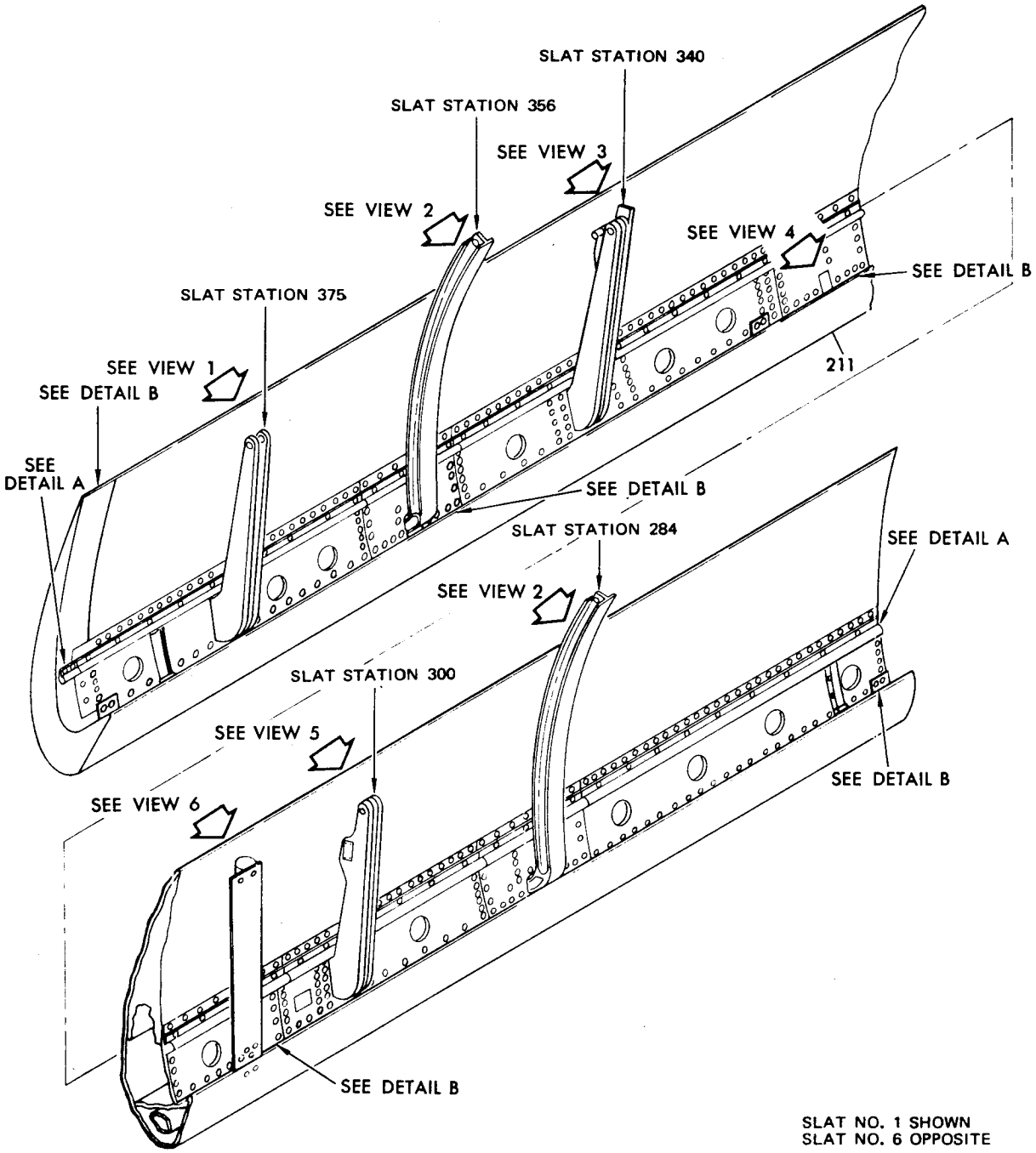
1. Tie or tape track assemblies (77, figure 1101) and tee assembly (208) in place to prevent movement and damage to adjacent parts in slat.

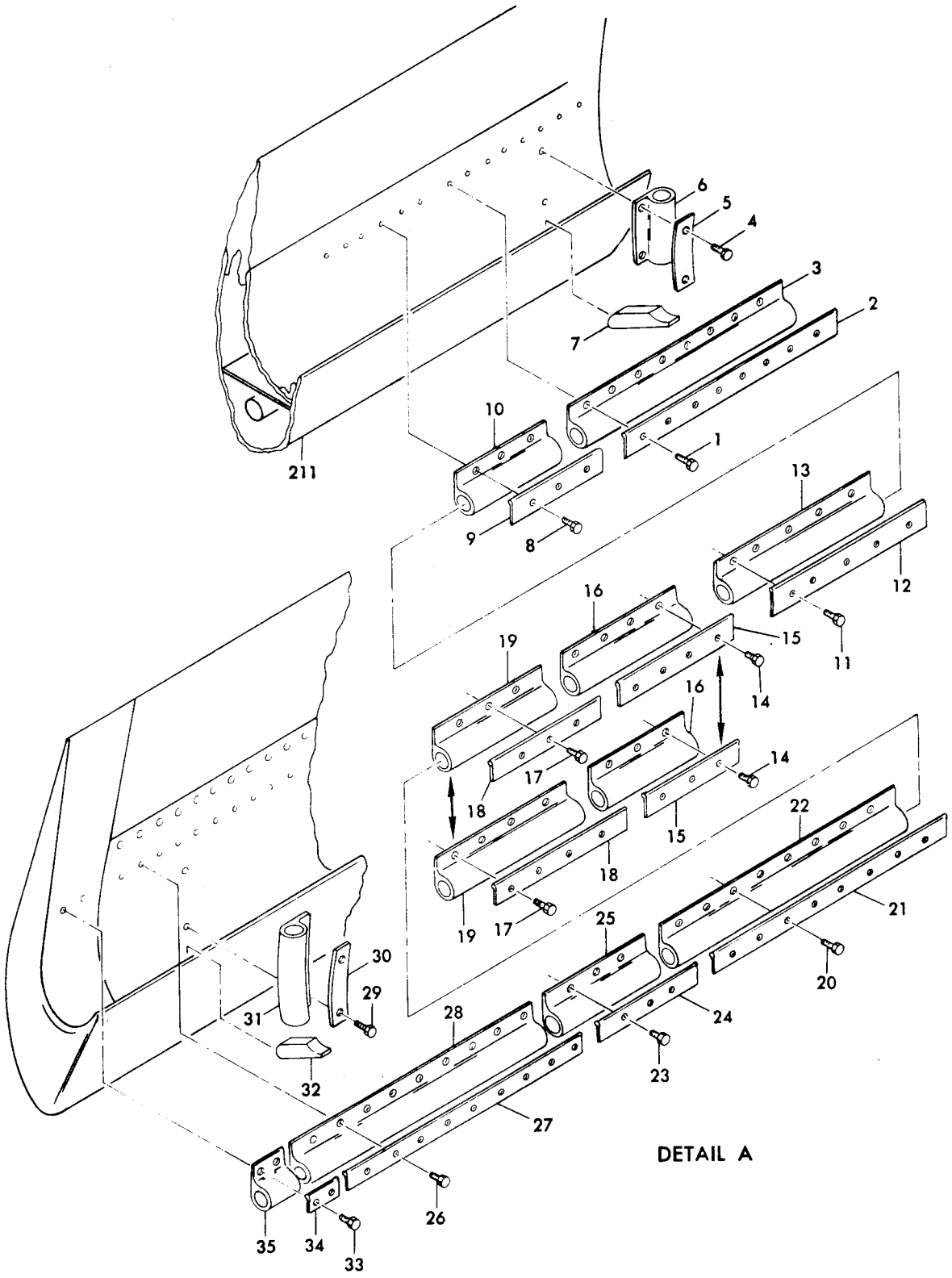
2. Wrap entire assembly in nonabsorbent material, and store in a cool, dry area, preferably humidity controlled.

3. For further information, refer to Subjects 20-44-02, Temporary Protective Coatings, and 20-70-01, Protection, Storage, and Handling of Airplane Components.

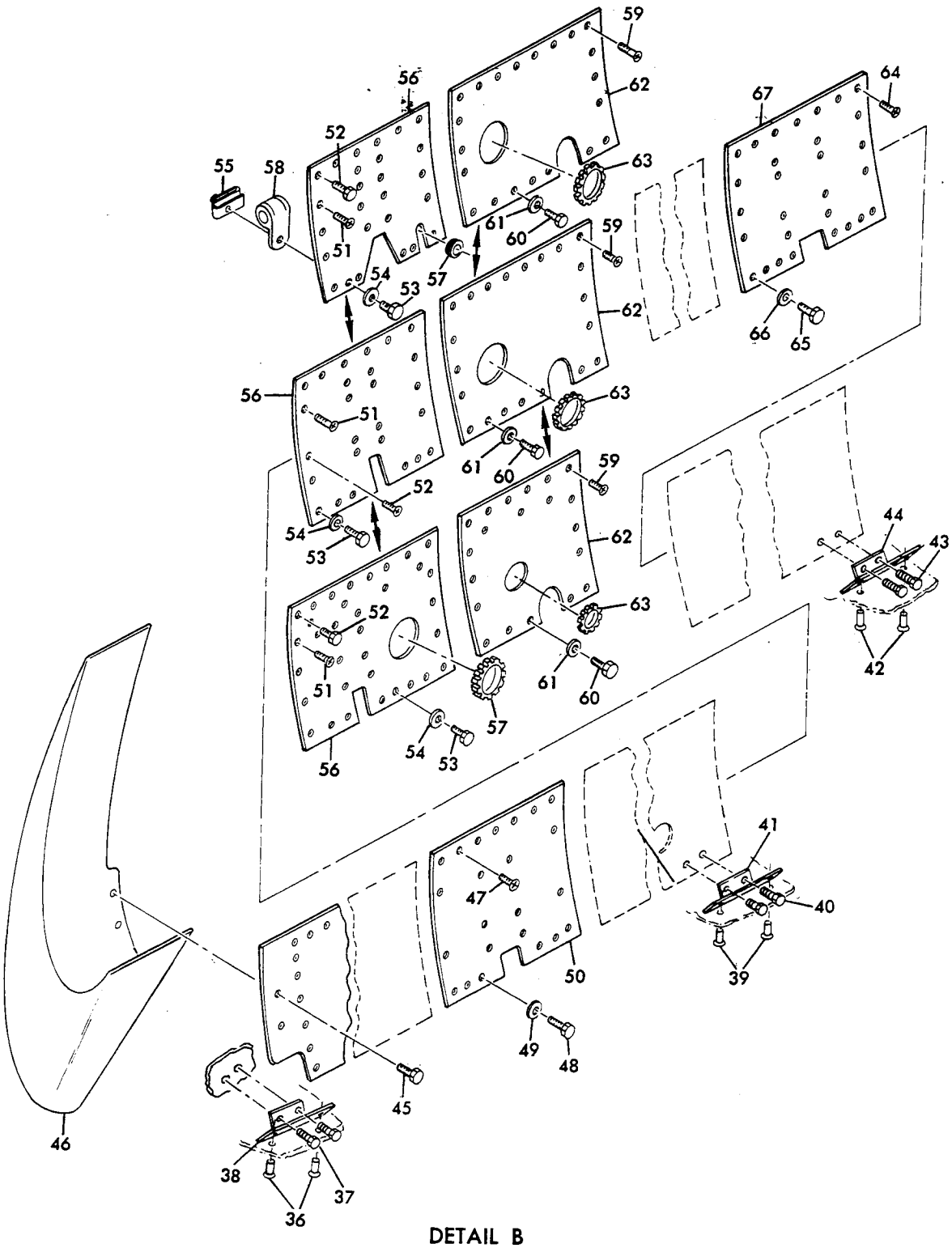
ILLUSTRATED PARTS LIST

1. Exploded View





Wing Leading Edge Slat Assemblies No. 1 and 6
Figure 1101 (Sheet 2)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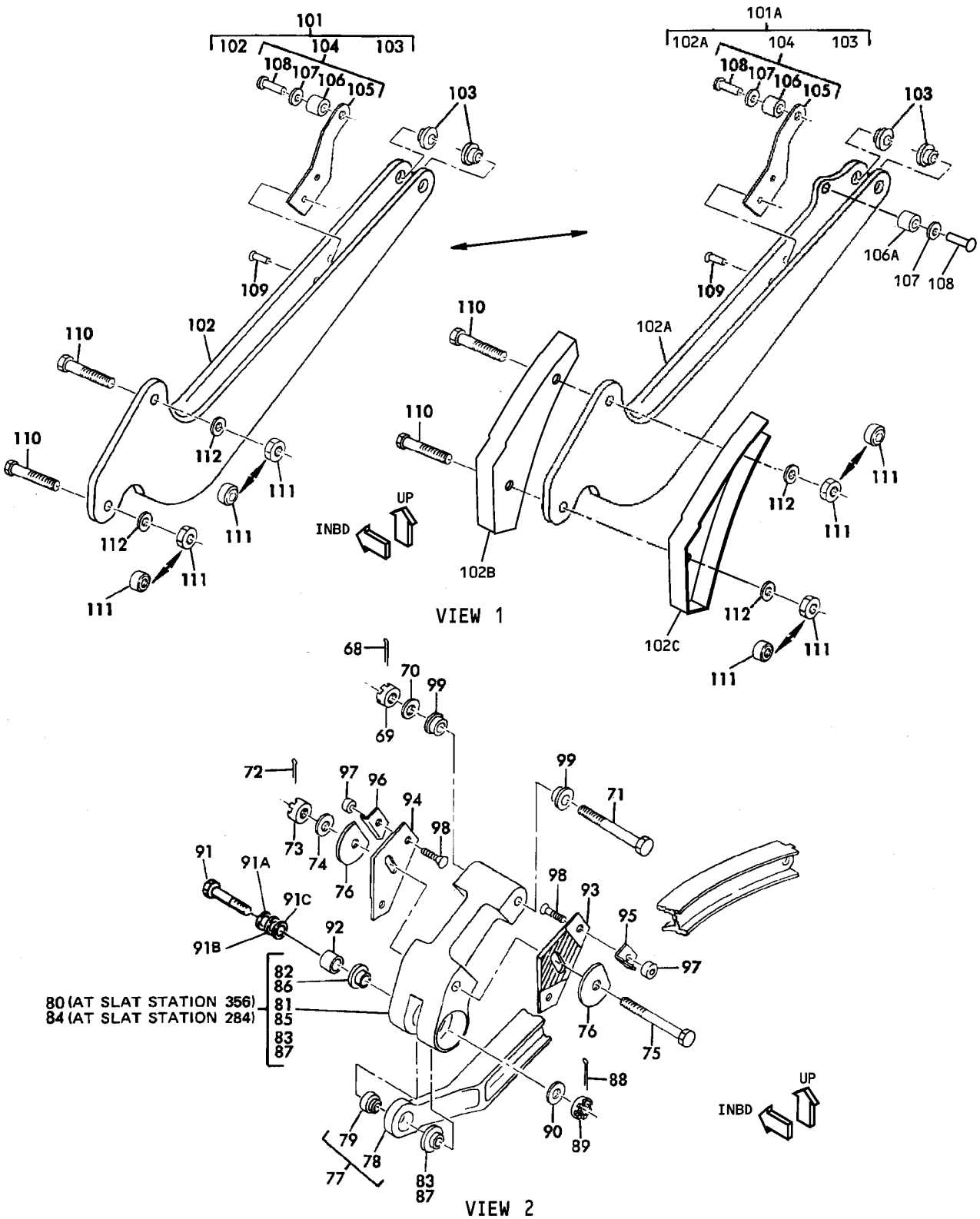
DETAIL B



Wing Leading Edge Slat Assemblies No. 1 and 6
Figure 1101 (Sheet 4)

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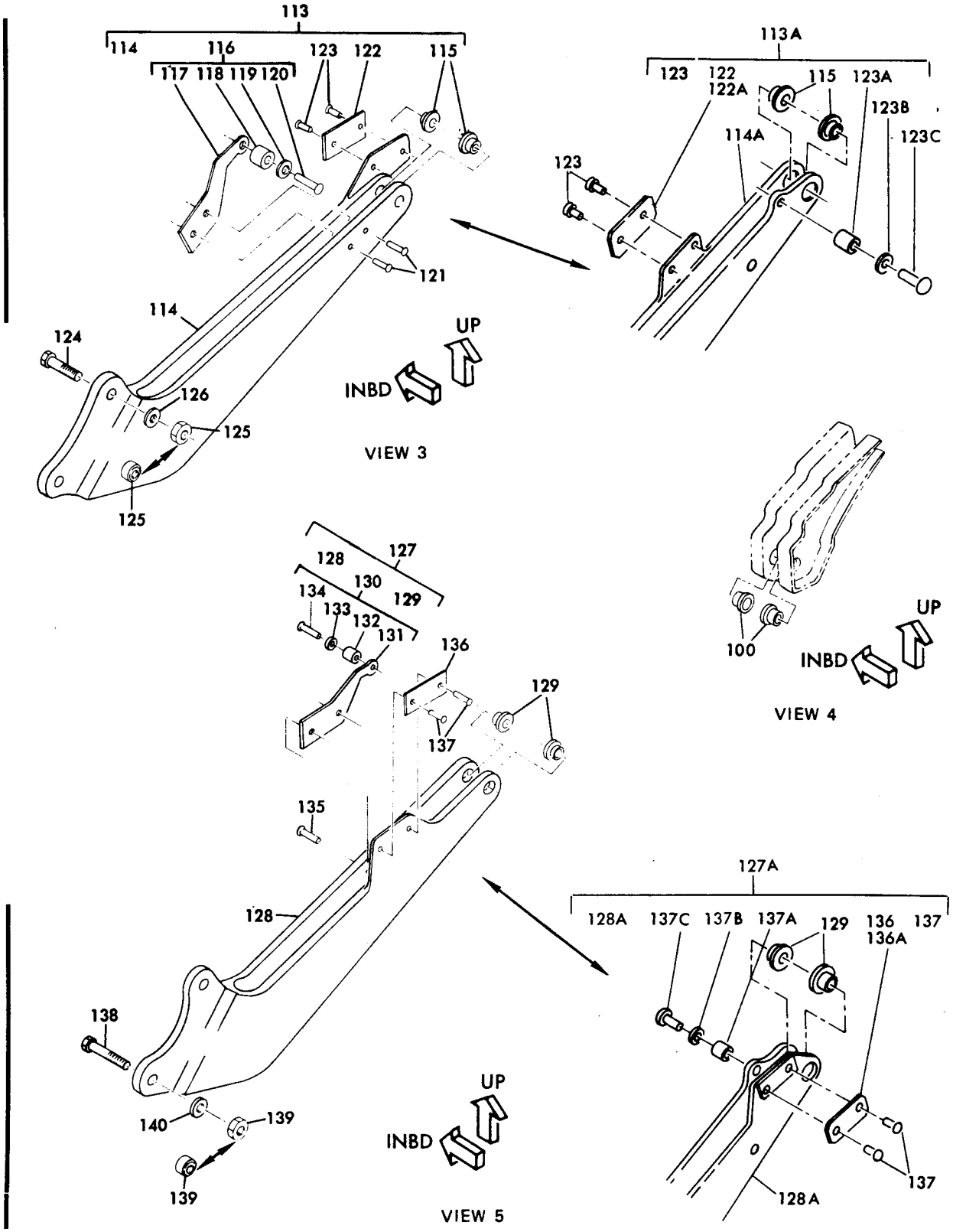
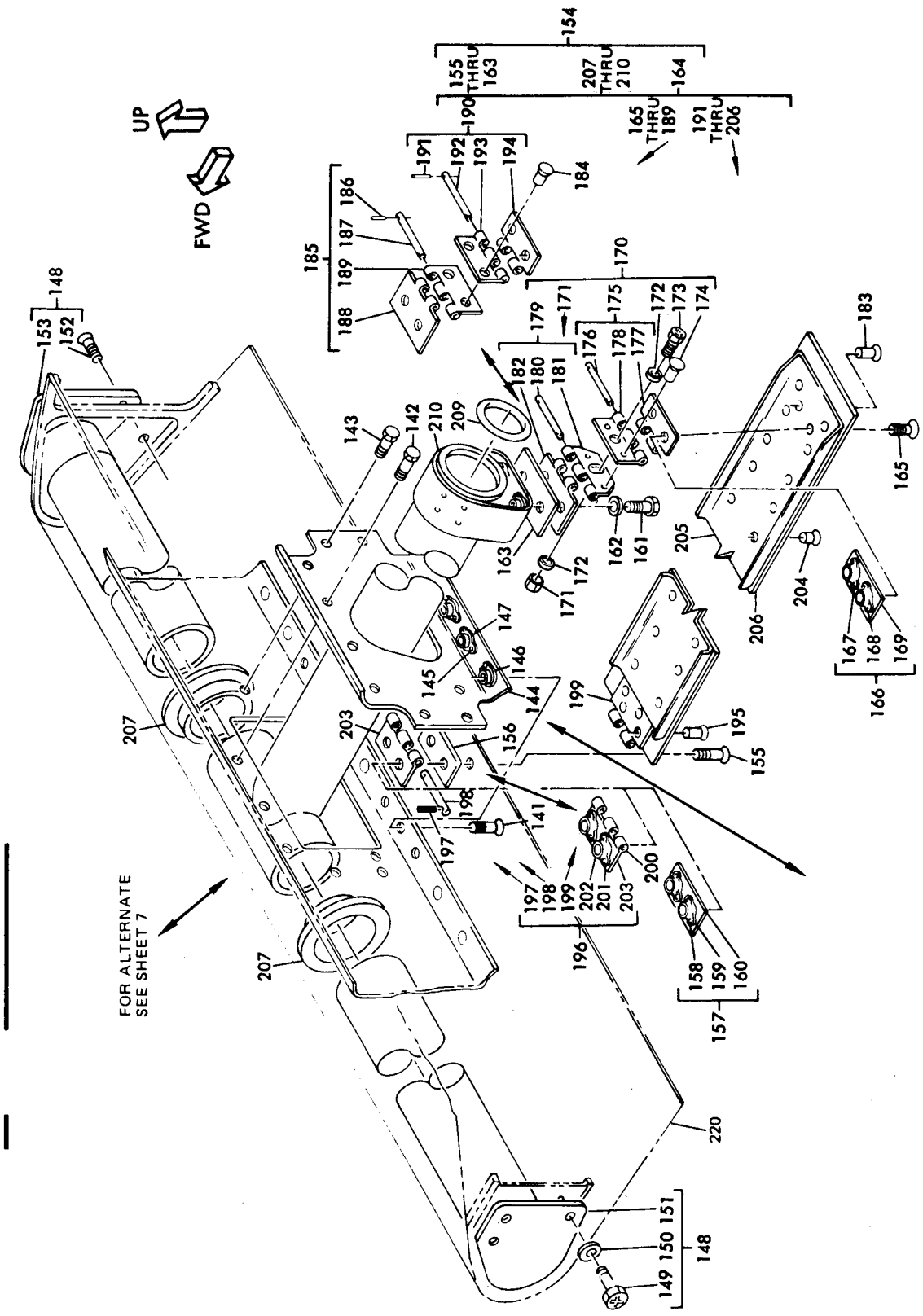
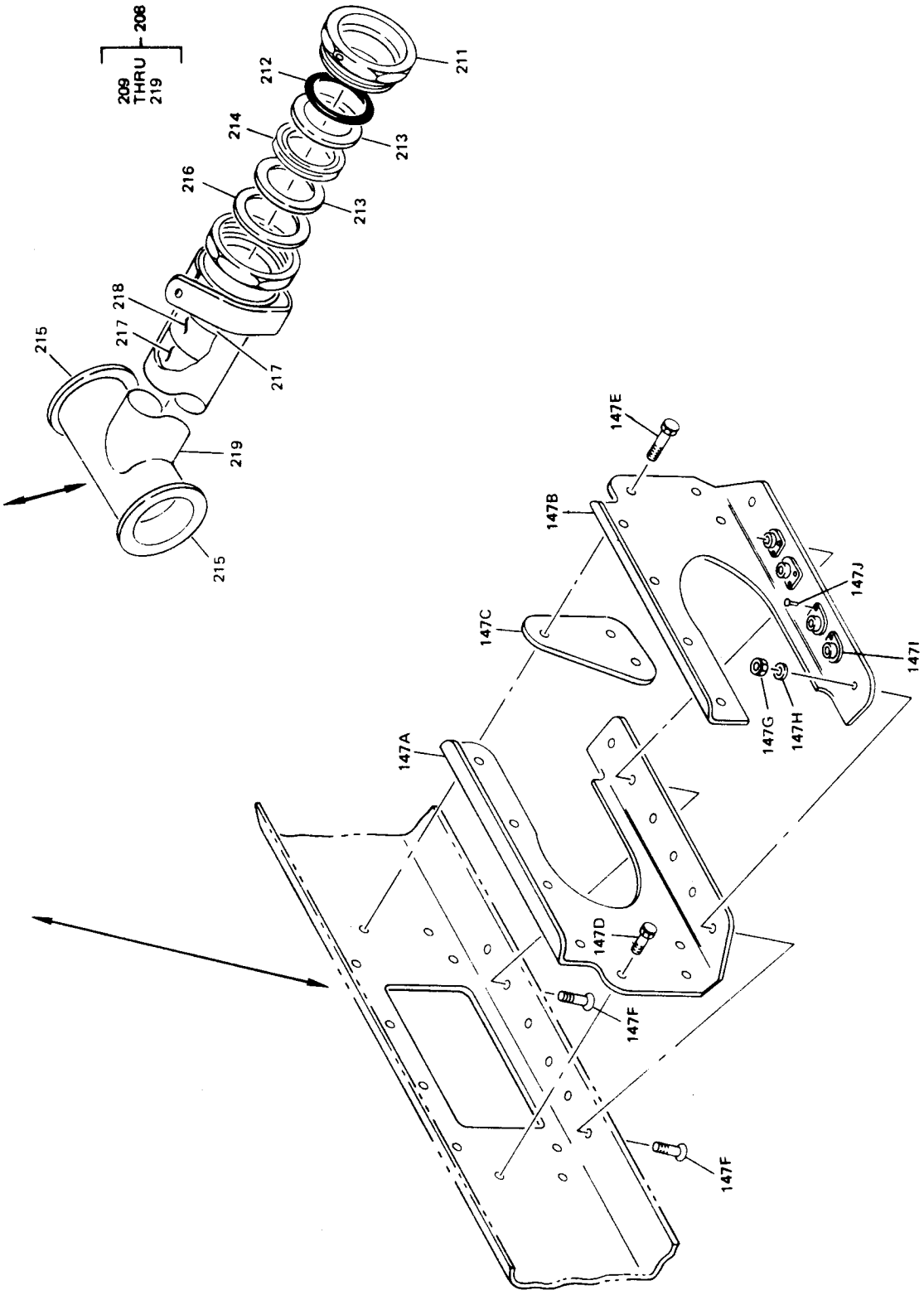


Wing Leading Edge Slat Assemblies No. 1 and 6
Figure 1101 (Sheet 5)



Wing Leading Edge Slat Assemblies No. 1 and 6
Figure 1101 (Sheet 6)



Wing Leading Edge Slat Assemblies No. 1 and 6
Figure 1101 (Sheet 7)

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
	65-46421-107		WLE SLAT ASSY NO. 1	A	RF
	65-46421-108		WLE SLAT ASSY NO. 6	B	RF
	65-46421-155		WLE SLAT ASSY NO. 1 (PRE SB 737-57-1080R2) *[27]*[28]	JA	RF
	65-46421-156		WLE SLAT ASSY NO. 6 (PRE SB 737-57-1080R2) *[27]*[29]	KA	RF
	65-46421-165		WLE SLAT ASSY NO. 1 (SB 57-1023)	C	RF
	65-46421-166		WLE SLAT ASSY NO. 6 (SB 57-1023)	D	RF
	65-46421-167		WLE SLAT ASSY NO. 1 (PRE SB 737-57-1080R2) *[27]*[28]	LA	RF
	65-46421-168		WLE SLAT ASSY NO. 6 (PRE SB 737-57-1080R2) *[27]*[29]	MA	RF
	65-46421-195		WLE SLAT ASSY NO. 1	E	RF
	65-46421-196		WLE SLAT ASSY NO. 6	F	RF
	65-46421-197		WLE SLAT ASSY NO. 1 (PRE SB 737-57-1080R2) *[27]*[28]	NA	RF
	65-46421-198		WLE SLAT ASSY NO. 6 (PRE SB 737-57-1080R2) *[27]*[29]	OA	RF
	65-46421-199		WLE SLAT ASSY NO. 1	G	RF
	65-46421-200		WLE SLAT ASSY NO. 6	H	RF
	65-46421-201		WLE SLAT ASSY NO. 1 (PRE SB 737-57-1080R2) *[28]	PA	RF
	65-46421-202		WLE SLAT ASSY NO. 6 (PRE SB 737-57-1080R2) *[27]*[28]	QA	RF
	65-46421-203		WLE SLAT ASSY NO. 1	S	RF
	65-46421-204		WLE SLAT ASSY NO. 6	T	RF
	65-46421-205		WLE SLAT ASSY NO. 1	I	RF
	65-46421-206		WLE SLAT ASSY NO. 6	J	RF
	65-46421-207		WLE SLAT ASSY NO. 1	K	RF
	65-46421-208		WLE SLAT ASSY NO. 6	L	RF
	65-46421-213		WLE SLAT ASSY NO. 1	M	RF
	65-46421-217		WLE SLAT ASSY NO. 1 *[21]	U	RF
	65-46421-221		WLE SLAT ASSY NO. 1 *[22]	W	RF
	65-46421-214		WLE SLAT ASSY NO. 6	N	RF
	65-46421-218		WLE SLAT ASSY NO. 6 *[21]	V	RF
	65-46421-222		WLE SLAT ASSY NO. 6 *[22]	X	RF
	65-46421-223		WLE SLAT ASSY NO. 1 *[25]	Y	RF
	65-46421-224		WLE SLAT ASSY NO. 6 *[26]	Z	RF
	65-46421-227		WLE SLAT ASSY NO. 1	O	RF
	65-46421-228		WLE SLAT ASSY NO. 6	P	RF
	65-46421-229		WLE SLAT ASSY NO. 1 *[25]	BA	RF
	65-46421-230		WLE SLAT ASSY NO. 6 *[26]	CA	RF
	65-46421-231		WLE SLAT ASSY NO. 1	Q	RF
	65-46421-232		WLE SLAT ASSY NO. 6	R	RF

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
	65-46421-233		WLE SLAT ASSY NO. 1 *[25]	DA	RF
	65-46421-234		WLE SLAT ASSY NO. 6 *[26]	EA	RF
	65-46421-235		WLE SLAT ASSY NO. 1 *[25]	FA	RF
	65-46421-236		WLE SLAT ASSY NO. 6 *[26]	GA	RF
	65-46421-237		WLE SLAT ASSY NO. 1	HA	RF
	65-46421-238		WLE SLAT ASSY NO. 6	IA	RF
1	BACB30LU3-2		. BOLT	A-LST	8
1	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	8
2	65-46421-69		. RETAINER	ACEGIK MOQSU W BA HA	1
2	65-46421-70		. RETAINER (OPP)	BDFHJL NPRTV X CA IA	1
3	65-46421-131		. SEAL	ACEGIK MOQSU W BA HA	1
3	65-46421-132		. SEAL (OPP)	BDFHJL NPRTVX CA IA	1
4	NAS1103-2		. BOLT	A-L	2
4	BACB30NM3K2		. BOLT (REPLS NAS1103-2)	M-X BA CA HA IA	2
5	65-46421-133		. RETAINER	ACEGIK MOQSU W BA HA	1
5	65-46421-148		. RETAINER (OPP)	BDFHJN LPRTVX CA IA	1
6	65-46421-137		. SEAL	ACEGIK MOQSU W BA HA	1
6	65-46421-138		. SEAL (OPP)	BDFHJL NPRTV X CA IA	1
7	65-46421-140		. SEAL		1
8	BACB30LU3-2		. BOLT	A-LST	3
8	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
9	69-44971-1		. RETAINER		1
10	65-46421-125		. SEAL		1
11	BACB30LU3-2		. BOLT	A-LST	5

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-11	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
12	65-46421-55		. RETAINER	ACEGIK MOQSU W BA HA	1
12	65-46421-56		. RETAINER (OPP)	BDFHJL NPRTVX CA IA	1
13	65-46421-119		. SEAL	ACEGIK MOQSU W BA HA	1
13	65-46421-120		. SEAL (OPP)	BDFHJL NPRTVX CA IA	1
14	BACB30LU3-2		. BOLT	ABST	4
14	BACB30LU3-2		. BOLT (SB 57-1023)	C-L	3
14	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
15	65-46421-50		. RETAINER *[20]	AB	1
15	65-46421-181		. RETAINER (SB 57-1023)	CEGIKM OQSUW BA HA	1
15	65-46421-192		. RETAINER (OPP-181)	DFHJLN PRTVX CA IA	1
16	65-46421-121		. SEAL	AB	1
16	65-46421-179		. SEAL (SB 57-1023)	CEGIKM OQSUW BA HA	1
16	65-46421-188		. SEAL (OPP-179)	DFHJLN PRTVX CA IA	1
17	BACB30LU3-2		. BOLT (SB 57-1023)	C-LST	4
17	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
17	BACB30LU3-2		. BOLT	AB	3
18	69-44971-2		. RETAINER	AB	1
18	65-46421-182		. RETAINER (SB 57-1023)	CEGIKM OQSUW BA HA	1
18	65-46421-194		. RETAINER (OPP-182)	DFHJLN PRTVX CA IA	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-19	65-46421-180		. SEAL (SB 57-1023)	CEGIKM OQSUW BA HA	1
19	65-46421-190		. SEAL (OPP)	DFHJLN PRTVX CA IA	1
19	65-46421-127		. SEAL *[20]	AB	1
20	BACB30LU3-2		. BOLT	A-LST	8
20	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
21	65-46421-51		. RETAINER	ACEGIK MOQSU W BA HA	1
21	65-46421-52		. RETAINER (OPP)	BDFHJLN PRTVX CA IA	1
22	65-46421-117		. SEAL	ACEGIK MOQSU W BA HA	1
22	65-46421-118		. SEAL (OPP)	BDFHJLN PRTVX CA IA	1
23	BACB30LU3-2		. BOLT	A-LST	3
23	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
24	69-44971-1		. RETAINER		1
25	65-46421-125		. SEAL		1
26	BACB30LU3-2		. BOLT	A-LST	8
26	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	3
27	65-46421-67		. RETAINER	ACEGIK MOQSU W BA HA	1
27	65-46421-68		. RETAINER (OPP)	BDFHJLN PRTVX CA IA	1
28	65-46421-129		. SEAL	ACEGIK MOQSU W BA HA	1
28	65-46421-130		. SEAL (OPP)	BDFHJLN PRTVX CA IA	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
29	NAS1103-2		. BOLT	A-L	2
29	BACB30NM3K2		. BOLT (REPLS NAS1103-2)	M-X BA CA HA IA	2
30	65-46421-134		. RETAINER	ACEGIK MOQSU W BA HA	1
30	65-46421-147		. RETAINER (OPP)	BDFHJL NPRTVX CA IA	1
31	65-46421-135		. SEAL	ACEGIK MOQSU W BA HA	1
31	65-46421-136		. SEAL (OPP)	BDFHJLN PRTVX CA IA	1
32	65-46421-139		. SEAL		1
33	BACB30LU3-2		. BOLT	A-LST	2
33	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	2
34	65-46421-63		. RETAINER		1
35	65-46421-123		. SEAL		1
36	BACR15BA6D		. RIVET (REPLS MS20426D6)		2
37	BACB30LU3-6		. BOLT	A-LST	2
37	BACB30NN3K6		. BOLT (REPLS BACB30LU3-6)	M-X BA CA HA IA	2
38	69-44924-3		. UPSTOP	A-PS-X BA CA HA IA	1
38	69-73512-1		. UPSTOP (REPLS 69-44924-3)	QR	1
39	BACR15BA6D		. RIVET (REPLS MS20426D6)		2
40	BACB30LU3-6		. BOLT	A-LST	2
40	BACB30NN3K6		. BOLT (REPLS BACB30LU3-6)	M-X BA CA HA IA	2
41	69-44923-2		. UPSTOP	A-PS-X BA CA HA IA	1
41	69-73511-1		. UPSTOP (REPLS 69-44923-2)	QR	1
42	BACR15BA6D		. RIVET (REPLS MS20426D6)		2
43	BACB30LU3-6		. BOLT	A-LST	2
43	BACB30NN3K6		. BOLT (REPLS BACB30LU3-6)	M-X BA CA HA IA	2
44	69-44922-2		. UPSTOP	A-PST	1
44	69-73510-1		. UPSTOP (REPLS 69-44922-2)	QR	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
45	BACB30LU3-2		. BOLT	A-LST	11
45	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	11
46	65-67110-1		. FAIRING ASSY, TIP	ACEGIK MOQSU W BA HA	1
46	65-67110-2		. FAIRING INSTL, TIP	BDFHJL NPRTVX CA IA	1
47	BACB30LU3-4		. BOLT	A-LST	17
47	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	17
48	NAS1103-4		. BOLT		6
49	AN960PD10L		. WASHER		AR
50	65-46421-11		. SKIN, INNER	ACEGIK MOQUW BA HA	1
50	65-46421-12		. SKIN, INNER	BDFHJL NPRVXC A IA	1
51	BACB30LU3-4		. BOLT	A-LST	6
51	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	6
52	NAS1103-4		. BOLT (USED WITH 65-46421-149,-150, -173, OR -174)		*[1]
52	BACB30LU3-4		. BOLT (USED WITH 65-46421-15 OR -16)	A-LST	14
52	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	14
53	NAS1103-4		. BOLT		*[2]
54	AN960PD10L		. WASHER		AR
55	BACN10FX1		. NUT, CLIP-ON (USED WITH BACC10DK2) *[20]	AB	4
56	65-46421-173		. SKIN, INNER (SB 57-1023)	CEGIKM OQUW BA HA	1
56	65-46421-174		. SKIN, INNER (SB 57-1023)	DFHJLN PRVX CA IA	1
56	65-46421-149		. SKIN, INNER *[20]	A	1
56	65-46421-150		. SKIN, INNER *[20]	B	1
56	65-46421-15		. SKIN, INNER *[20]	A	1
56	65-46421-16		. SKIN, INNER *[20]	B	1
57	BACG20ZA790		. GROMMET		1
57	NAS1368N3B		. GROMMET	AB	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
58	BACC10DK2		. CLAMP *[20]	AB	4
59	BACB30LU3-4		. BOLT	A-LST	*[3]
59	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	*[3]
60	NAS1103-4		. BOLT		*[4]
61	AN960PD10L		. WASHER		*[4]
62	65-46421-169		. SKIN, INNER (SB 57-1023)	CEGIKM S	1
62	65-46421-170		. SKIN, INNER (SB 57-1023)	DFHJLN T	1
62	65-46421-17		. SKIN, INNER *[20]	A	1
62	65-46421-18		. SKIN, INNER *[20]	B	1
62	65-46421-111		. SKIN, INNER	OQUWY BA HA	1
62	65-46421-112		. SKIN, INNER	PRVXZ CA IA	1
63	BACG20ZA400		. GROMMET (SB 57-1023)	C-M	1
63	BACG20ZA1275		. GROMMET *[20]		1
64	BACB30LU3-4		. BOLT	A-LST	20
64	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	20
65	NAS1103-4		. BOLT		8
66	AN960PD10L		. WASHER		8
67	65-46421-21		. SKIN, INNER	ACEGIK MOQUW BA HA	1
67	65-46421-22		. SKIN, INNER	BDFHJL NPRVX CA IA	1
68	MS24665-134		. PIN, COTTER *[5] (SB 57-1013)		2
69	BACN10JD5		. NUT *[5] (SB 57-1013)		2
69	BACN10JC5		. NUT (REPLS NAS679A5) *[6]*[20]		2
70	MS20002-5		. WASHER		2
71	BACB30NF5D30		. BOLT *[5] (SB 57-1013)		2
71	BACB30NR5DK28		. BOLT (REPLS BACB30NF5D28)	M-X BA CA HA IA	2
71	BACB30NF5D28		. BOLT *[5]*[20]	A-LOP ST	2
71	NAS1105-30		. BOLT *[6]*[20]	A-L	2
71	BACB30NR5K30		. BOLT (REPLS NAS1105-30)	M-X BA CA HA IA	2
72	MS24665-285		. PIN, COTTER *[7] (SB 57-1013)		2
73	BACN10JD6		. NUT *[7] (SB 57-1013)		2
73	BACN10JC6		. NUT (REPLS NAS679A6) *[8]*[20]	A-L	2
74	MS20002-6		. WASHER		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
75	BACB30NF6D34		. BOLT *[7] (SB 57-1013)		2
75	BACB30NF6D33		. BOLT (REPLS NAS1106-34) *[7]*[20]	A-LST	2
75	BACB30NR6DK33		. BOLT (REPLS BACB30NF6D33)	M-X BA CA HA IA	2
75	NAS1106-34		. BOLT *[8]*[20]		2
76	69-40666-3		. WASHER, SERRATED		4
77	65-49448-1		. TRACK ASSY	ABCDIJ HA IA	2
77	65-49448-7		. TRACK ASSY	EFGH KL	2
77	65-49448-11		. TRACK ASSY	M-X BA CA	2
78	65-49448-2		. . TRACK (USED ON 65-49448-1)		1
78	65-49448-8		. . TRACK (USED ON 65-49448-7)		1
78	65-49448-12		. . TRACK (USED ON 65-49448-11)		1
79	03-728-0312		. . BEARING, V09455 (BOEING 10-60545-112S)(PREF)		1
79	SBS10ATC24		. . BEARING, V21335 (BOEING 10-60545-112S)(PREF)		1
79	YTA119		. . BEARING, V77896 (BOEING 10-60545-112S)(PREF)		1
79	BLFN5-003		. . BEARING, V81376 (BOEING 10-60545-112S)(PREF)		1
79	KSBG5N5		. . BEARING, V97613 (BOEING 10-60545-112S)(PREF)		1
79	MS21232-5		. . BEARING (OPT)		1
80	69-40694-1		. FITTING ASSY, TRACK ATTACH		1
81	69-40694-2		. . FITTING		1
82	BACB28X8B10		. . BUSHING		1
83	BACB28X5B10		. . BUSHING		1
84	69-40611-1		. FITTING ASSY, TRACK ATTACH		1
85	69-40611-2		. . FITTING		1
86	BACB28X8B10		. . BUSHING		1
87	BACB28X5B10		. . BUSHING		1
88	MS24665-134		. PIN, COTTER		2
89	BACN10JD5		. NUT (REPLS AN320-5)		2
90	MS20002-5		. WASHER		2
91	BACB30GE5D13		. BOLT		2
91A	AN960C816L		DELETED		
91A	BACW10P163A		. WASHER (POST SB 57-1080) (REPLS AN960C816L)		2
91B	AN960PD816		. WASHER (POST SB 57-1080)		AR
91C	AN960PD816L		. WASHER (POST SB 57-1080)		AR
92	NAS74A5-004P		. BUSHING		2
93	69-40628-1		. PLATE, SERRATED		2
94	69-40628-2		. PLATE, SERRATED		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
95	69-43504-3		. CLIP, ALIGNING		2
96	69-43504-4		. CLIP, ALIGNING		2
97	BACC30K6		. COLLAR		8
98	BACB30GY6-3		. LOCKBOLT		8
99	BACB28X5B5		. BUSHING		4
100	69-37867-43		. BUSHING		2
100	69-37867-26		. BUSHING (OPT TO 69-37867-43)		2
101	69-54932-1		. ARM ASSY (LIMITED)	A	1
101	69-54932-2		. ARM ASSY (LIMITED)	B	1
101A	69-54931-1		. RIB-ARM ASSY (LIMITED)	A	1
101A	69-54931-1		. RIB-ARM ASSY	C	1
101A	69-54931-2		. RIB-ARM ASSY (LIMITED)	B	1
101A	69-54931-2		. RIB-ARM ASSY	D	1
101A	69-54931-3		. RIB-ARM ASSY	EGI	1
101A	69-54931-4		. RIB-ARM ASSY	FHJ	1
101A	69-54931-5		. RIB-ARM ASSY	KMUW OQ HA	1
101A	69-54931-6		. RIB-ARM ASSY	LNVXP RIA	1
102	69-54932-3		. . ARM (USED ON 69-54932-1)		1
102	69-54932-4		. . ARM (USED ON 69-54932-2)		1
102A	69-54932-1		. . ARM ASSY (USED ON 69-54931-1)		1
102A	69-54932-2		. . ARM ASSY (USED ON 69-54931-2)		1
102A	69-54932-5		. . ARM ASSY (USED ON 69-54931-3) (OPT TO 65-79963-3)		1
102A	65-79963-3		. . ARM ASSY-REWORK (USED ON 69-54931-3) (OPT TO 69-54932-5)		1
102A	69-54932-6		. . ARM ASSY (USED ON 69-54931-4) (OPT TO 65-79963-4)		1
102A	65-79963-4		. . ARM ASSY-REWORK (USED ON 69-54931-4) (OPT TO 69-54932-6)		1
102A	69-54932-5		. . ARM (USED ON 69-54931-5)		1
102A	69-54932-6		. . ARM (USED ON 69-54931-6)		1
102B	69-54929-1		. . RIB AUXILIARY TRACK (USED ON 69-54931-1,-3)		1
102B	69-54929-2		. . RIB AUXILIARY TRACK (USED ON 69-54931-2,-4)		1
102B	69-54929-3		. . RIB AUXILIARY TRACK (LIMITED) (USED ON 69-54931-5)		1
102B	69-54929-4		. . RIB AUXILIARY TRACK (LIMITED) (USED ON 69-54931-6)		1
102C	69-54930-1		. . RIB AUXILIARY TRACK (USED ON 69-54931-1,-3)		1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-102C	69-54930-2		. . RIB AUXILIARY TRACK (USED ON 69-54931-2,-4)		1
102C	69-54930-3		. . RIB AUXILIARY TRACK (LIMITED) (USED ON 69-54931-5)		1
102C	69-54930-4		. . RIB AUXILIARY TRACK (LIMITED) (USED ON 69-54931-6)		1
103	BACB28X4B18		. . BUSHING		2
104	69-50764-2		. . ARM ASSY (USED ON 69-54932-1)		1
104	69-50764-1		. . ARM ASSY (USED ON 69-54932-2)		1
105	69-50764-3		. . . ARM		1
106	69-50764-5		. . . SPACER (REPLS 69-50764-4) (USED ON 69-50764-1,-2)		1
106	69-50764-4		DELETED		
106A	NAS42DD5-20		. . . SPACER (USED ON 69-54932-5,-6)		1
107	AN960PD8		. . . WASHER		1
108	BACR15BB5D		. . . RIVET (REPLS MS20470D5)		1
109	BACR15BA5D		. . RIVET (REPLS MS20426D5)		2
110	NAS1104-9		. BOLT *[20]		3
110	NAS1104-8		. BOLT *[20]		2
110	BACB30GW8-8		. LOCKBOLT *[20]		2
111	BACN10JC4		. NUT (REPLS NAS679A4W) (USED WITH NAS1104-9)		3
111	BACN10JC4		. NUT (REPLS NAS679A4W) (USED WITH NAS1104-8)		2
111	BACC30K8		. COLLAR (USED WITH BACB30GW8-8)		2
112	AN960PD416		. WASHER (USED WITH NAS1104-9)		3
112	AN960PD416		. WASHER (USED WITH NAS1104-8)		2
113	65-55526-1		. ARM ASSY	ACI	1
113	65-55526-2		. ARM ASSY	BDJ	1
113A	65-55526-5		. ARM ASSY (PREF)	EGKMO QUW BA HA	1
113A	65-79963-5		. ARM ASSY (OPT)	EG	1
113A	65-55526-6		. ARM ASSY (PREF)	FHLNP RVX CA IA	1
113A	65-79963-6		. ARM ASSY (OPT)	FH	1
114	65-55526-3		. . ARM (USED ON 65-55526-1)		1
114	65-55526-4		. . ARM (USED ON 65-55526-2)		1
114A	65-55526-7		. . ARM (USED ON 65-55526-5)		1
114A	*[11]		. . ARM ASSY (USED ON 65-79963-5)		1
114A	65-55526-8		. . ARM (USED ON 65-55526-6)		1
114A	*[11]		. . ARM ASSY (USED ON 65-79963-6)		1
115	BACB28X4B18		. . BUSHING		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
116	69-50764-1		. . ARM ASSY (USED ON 65-55526-1)		1
116	69-50764-2		. . ARM ASSY (USED ON 65-55526-2)		1
117	69-50764-3		. . . ARM		1
118	69-50764-5		. . . SPACER (REPLS 69-50764-4)		1
118	69-50764-4		. . . SPACER		1
119	AN960PD8		. . . WASHER		1
120	BACR15BB5D		. . . RIVET (REPLS MS20470D5)		1
121	BACR15BA5D		. . RIVET (REPLS MS20426D5) (USED ON 65-55526-1, -2)		2
122	66-24198-7		. . ACTUATOR, SENSOR (USED ON 65-55526-1,-2,-5,-6)		1
122A	69-62797-5		. . SENSOR (USED ON 65-79963-5)		1
122A	69-62797-6		. . SENSOR (USED ON 65-79963-6)		1
123	BACR15BA3A		. . RIVET (REPLS MS20426A3)		2
123A	NAS42DD5-20		. . SPACER (USED ON 65-55526-5, -6)		1
123B	AN960PD8		. . WASHER (USED ON 65-55526-5, -6)		1
123C	BACR15BB5D		. . RIVET (USED ON 65-55526-5, -6)		1
124	NAS1104-7		. BOLT *[20]		2
124	BACB30GW8-6		. LOCKBOLT *[20]		2
125	BACN10JC4		. NUT (REPLS NAS679A4W) (USED WITH NAS1104-7)		2
125	BACC30K8		. COLLAR (USED WITH BACB30GW8-6)		2
126	AN960PD416		. WASHER (USED WITH NAS1104-7)		2
127	65-55525-1		. ARM ASSY	ACI	1
127	65-55525-2		. ARM ASSY	BDJ	1
127A	65-55525-7		. ARM ASSY (PREF)	EG	1
127A	65-79962-1		. ARM ASSY (OPT)	EG	1
127A	65-55525-8		. ARM ASSY (PREF)	FH	1
127A	65-79962-2		. ARM ASSY (OPT)	FH	1
127A	65-55525-11		. ARM ASSY	GKMOQ UW BA HA	1
127A	65-55525-12		. ARM ASSY	HLNPR VX CA IA	1
128	65-55525-5		. . ARM (USED ON 65-55525-1)		1
128	65-55525-6		. . ARM (USED ON 65-55525-2)		1
128A	65-55525-9		. . ARM (USED ON 65-55525-7)		1
128A	*[12]		. . ARM ASSY (USED ON 65-79962-1)		1
128A	65-55525-10		. . ARM (USED ON 65-55525-8)		1
128A	*[12]		. . ARM ASSY (USED ON 65-79962-2)		1
128A	65-55525-13		. . ARM (USED ON 65-55525-11)		1
128A	65-55525-14		. . ARM (USED ON 65-55525-12)		1
129	BACB28X4B18		. . BUSHING		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
130	69-50764-2		. . ARM ASSY (USED ON 65-55525-1)		1
130	69-50764-1		. . ARM ASSY (USED ON 65-55525-2)		1
131	69-50764-3		. . . ARM		1
131	69-50764-5		. . . SPACER (REPLS 69-50764-4)		1
132	69-50764-4		. . . SPACER		1
133	AN960PD8		. . . WASHER		1
134	BACR15BB5D		. . . RIVET (REPLS MS20470D5)		1
135	BACR15BA5D		. . RIVET (REPLS MS20426D5) (USED ON 65-55525-1,-2)		1
136	66-24198-7		. . ACTUATOR, SENSOR (USED ON 65-55525-1,-2,-7,-8)		1
136A	69-62797-1		. . SENSOR (USED ON 65-79962-1)		1
136A	69-62797-2		. . SENSOR (USED ON 65-79962-2)		1
137	BACR15BA3A		. . RIVET (REPLS MS20426A3)		2
137A	NAS42DD5-20		. . SPACER (USED ON 65-55525-7,-8)		1
137B	AN960PD8		. . WASHER (USED ON 65-55525-7,-8)		1
137C	BACR15BB5D		. . RIVET (USED ON 65-55525-7,-8)		1
138	NAS1105-7		. BOLT *[20]		2
138	BACB30GW10-6		. LOCKBOLT *[20]		2
139	BACN10JC5		. NUT (REPLS NAS679A5) (USED WITH NAS1105-7)		2
139	BACC30K10		. COLLAR (USED WITH BACB30GW10-6)		2
140	AN960PD516		. WASHER (USED WITH NAS1105-7)		2
141	BACB30LU3-3		. BOLT		6
142	NAS1103-2		. BOLT	A-L	4
142	BACB30NM3K2		. BOLT (REPLS NAS1103-2)	M-X BA CA HA IA	4
143	NAS1103-3		. BOLT		6
144	69-43539-1		. DOUBLER (REPLD BY 147A,147B,147C)		1
145	BACR15BA3D		. RIVET (REPLS MS20426D3)		12
146	BACN10PC3		. NUTPLATE		4
147	BACN10JN3		. NUTPLATE (REPLS NAS1068A3)		2
147A	69-68803-1		. DOUBLER (ADDED BY SB 57-1100)		1
147B	69-68803-2		. DOUBLER (ADDED BY SB 57-1100)		1
147C	69-68803-3		. FILLER (ADDED BY SB 57-1100)		1
147D	BACB30NF3-2		. BOLT		3
147E	BACB30NF3-3		. BOLT		7
147F	BACB30LU3-4		. BOLT	A-LST	6
147F	BACB30NN3K4		. BOLT (REPLS BACB30LU3-4)	M-X BA CA HA IA	6
147G	BACN10JC3		. NUT *[14]		2
147H	AN960PD10		. WASHER *[14]		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
147I	BACN10JP3D		. NUTPLATE *[15]		3
147J	MS20426D3		. RIVET		AR
			INSTALLATION ITEMS		
148	65-54753-3		SPRAY TUBE INSTL	ACEGIK MOQSU WY BA DA FA HA JA LA NA PA	1
148	65-54753-4		SPRAY TUBE INSTL	BDFHJL NPRTVX Z CA EA GA IA KA MA OA QA	1
149	BACS12CB3-7		. SCREW		3
150	AN960D10L		. WASHER		3
151	65-54754-12		. TUBE ASSY, SPRAY (USED ON 65-54753-3)		1
151	65-54754-13		. TUBE ASSY, SPRAY (USED ON 65-54753-4)		1
152	BACB30FL3-2		. BOLT		3
153	65-54755-12		. TUBE ASSY, SPRAY (USED ON 65-54753-3)		1
154	65-51582-15		DUCT INSTL, ANTI-ICING	KMOQU W BA HA	1
154	65-51582-1		DUCT INSTL, ANTI-ICING	ACEGI	1
154	65-51582-16		DUCT INSTL, ANTI-ICING	LNPRVX CA IA	1
154	65-51582-2		DUCT INSTL, ANTI-ICING	BDFHJ	1
154	65-51582-9		DUCT INSTL, TAI	S JA LA NA PA	1
154	65-51582-10		DUCT INSTL, TAI	T KA MA OA QA	1
155	BACB30LU3-2		. BOLT (REPLS BACB30FL3-1) (SB 30-1011)	A-LST	2
155	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	2
156	65-51582-7		. SHIM *[9]		1
157	69-38655-19		. PLATE ASSY *[10] *[16] (SB 30-1011)		1
158	BACR15BA3D		. . RIVET (REPLS MS20426D3)		4
159	BACN10JN3		. . NUTPLATE (REPLS NAS1068A3)		2
160	69-38655-18		. . PLATE		1
161	BACS12CB3-8		. SCREW *[9]		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
161	BACB30NE3-2		. BOLT (REPLS BACB30NF3-2) *[10] (SB 30-1011)		2
161	BACS12CB3-5		. SCREW *[17]		2
162	AN960-10L		. WASHER *[9]		2
162	AN960PD10		. WASHER *[10] (SB 30-1011)		2
162	AN960-10L		. WASHER *[17]		2
163	65-51582-8		. SHIM *[9]		1
164	69-38641-29		. DOOR ASSY (PREF)		1
164	69-38641-23		. DOOR ASSY (OPT TO 65-51582-29) *[20] (SB 30-1011)		1
164	69-38641-17		. DOOR ASSY (USED ON 65-51582-1) *[20] (OPT TO 69-38641-23 AND -29)		1
164	69-38641-1		. DOOR ASSY (USED ON 65-51582-1) *[20]		1
164	69-38641-18		. DOOR ASSY (USED ON 65-51582-2) *[20] (OPT TO 69-38641-23 AND -29)		1
164	69-38641-2		. DOOR ASSY (USED ON 65-51582-2) *[20]		1
165	BACB30LU3-2		. . BOLT (USED ON 69-38641-17,-18,-23)	A-LST	2
165	BACB30NN3K2		. BOLT (REPLS BACB30LU3-2)	M-X BA CA HA IA	2
166	69-38655-19		. . PLATE ASSY (USED ON 69-38641-17, -18,-23,-29)		1
167	BACR15BA3D		. . . RIVET (REPLS MS20426D3)		4
168	BACN10JN3		. . . NUTPLATE (REPLS NAS1068A3)		2
169	69-38655-18		. . . PLATE		1
170	69-38655-21		. . HINGE ASSY (USED ON 69-38641-17, -18,-23,-29)		1
171	BACN10JC3		. . . NUT		1
172	AN960PD10		. . . WASHER		2
173	BACS12CB3-8		. . . SCREW		1
174	BACR15BA6D		. . . RIVET (REPLS MS20426D6) (SB 30-1011)		2
175	69-3865-20		. . . HINGE ASSY		1
176	69-38655-27		. . . . PIN, HINGE (SB 30-1011)		1
177	69-38655-16		. . . . HINGE HALF		1
178	69-38655-23		. . . . HINGE HALF		1
179	69-38655-24		. . . HINGE ASSY		1
180	69-38655-26		. . . . PIN, HINGE		1
181	69-38655-25		. . . . HINGE HALF		1
182	69-38655-29		. . . . HINGE HALF		1
183	BACR15BA6D		. . RIVET (REPLS MS20426D6) (USED ON 69-38641-1, -2)		4
184	MS20615-5M5		. . RIVET (USED ON 69-38641-1,-2)		3

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
185	69-38655-2		. . HINGE ASSY (USED ON 69-38641-1,-2)		1
186	MS51923-130		. . . PIN (REPLS MS51923-112)		2
187	MS20253P2-200		. . . PIN (REPLS MS20253P2-205)		1
188	69-38655-7		. HINGE HALF		1
189	69-38655-11		. HINGE HALF		1
190	69-38655-3		. HINGE ASSY (USED ON 69-38641-1,-2)		1
191	MS51923-130		. PIN (REPLS MS51923-112)		2
192	MS20253P2-175		. PIN (REPLS MS20253P2-180)		1
193	69-38655-9		. . . HINGE HALF		1
194	69-38655-8		. . . HINGE HALF		1
195	BACR15BA6D		. . RIVET (REPLS MS20426D6)		4
196	69-38655-4		. . HINGE ASSY (USED ON 69- 38641-1,-2)		1
196	69-38655-14		. . HINGE ASSY (USED ON 69-38641-17, -18,-23,-29)		1
197	MS51923-130		. . . PIN (REPLS MS51923-112) (USED ON 69-38641-1,-2)		2
198	MS20253P2-175		. . . PIN (REPLS MS20253P2-180) (USED ON 69-38641-1,-2)		1
198	69-38655-27		. . . PIN (USED ON 69-38641-17,-18, -23,-29)		1
199	69-38655-8		. . . HINGE HALF		1
200	69-38655-17		. HINGE HALF (USED ON 69-38641-17, -18,-23,-29)		1
201	BACR15BA3D		. . . RIVET (REPLS MS20426D3) (USED ON 69-38641-1, -2)		4
202	BACN10JN3		. . . NUTPLATE (REPLS NAS1068A3) (USED ON 69-38641-1, -2)		2
203	69-38655-10		. . . HINGE HALF (USED ON 69-38641-1, -2)		1
204	BACR15BA6D		. . RIVET (REPLS MS20426D6)		20
205	69-38644-1		. . CHANNEL (USED ON 69-38641-1,-2)		1
205	69-38644-4		. . CHANNEL (USED ON 69-38641-1,-18)		1
205	69-38644-7		. . CHANNEL (USED ON 69-38641-23,-29)		1
206	69-38641-7		. . PANEL (USED ON 69-38641-1,-17)		1
206	69-38641-8		. . PANEL (USED ON 69-38641-2,-18)		1
206	69-38641-26		. . PANEL (USED ON 69-38641-23,-29)		1
207	65-51583-8		. BEARING (USED ON 65-51582-1,-15)		2
207	65-51583-6		. BEARING (OPT TO 65-51583-8)		2
207	65-51583-2		. BEARING (OPT TO 65-51583-8)		2
207	65-51583-7		. BEARING (USED ON 65-51582-1,-16)		2
207	65-51583-5		. BEARING (OPT TO 65-51583-7)		2
207	65-51583-1		. BEARING (OPT TO 65-51583-7)		2
207	66-13255-1		. BEARING (USED ON 65-51582-1,-2) *[19]		2

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1 2 3 4 5 6 7	USE CODE	QTY PER ASSY
1101-					
208	65C17142-3		. TEE ASSY (REPLS 65-51570-1,-7 PER SB 30-1015)	ACEGI	1
208	65C17142-3		. TEE ASSY (SUPSDS 65C17142-1 ON 65-51582-15)	KMOQU WY BA DA FA	1
208	65C17142-1		. TEE ASSY (SUPSD BY 65C17142-3)	KMUW	1
208	65C17142-4		. TEE ASSY (REPLS 65-51570-2,-8 PER SB 30-1015)	BDFHJ	1
208	65C17142-4		. TEE ASSY (SUPSDS 65C17142-2 ON 65-51582-16)	LNPRVX Z CA EA GA	
208	65C17142-2		. TEE ASSY (SUPSD BY 65C17142-4)	LNVX	1
208	65-51570-1		. TEE ASSY (USED ON 65-51582-1)		1
208	65-51570-2		. TEE ASSY (USED ON 65-51582-2)		1
208	65-51570-7		. TEE ASSY (USED ON 65-51582-1) (PRE SB 30-1015)		1
208	65-51570-8		. TEE ASSY (USED ON 65-51582-2) (PRE SB 30-1015)		1
-208A	65C17106-1		. TEE ASSY, UPPER (REPLS 65-54756-1 PER SB 30-1015)		1
-208A	65-54756-1		. TEE ASSY, UPPER (PRE SB 30-1015)		1
209	BACP11L219		. . PACKING, O-RING		1
209	BACP11L218		DELETED		
210	*[13]		. . TEE ASSY		1
211	69-69944-2		. . CAP, END *[17]		1
211	69-69944-1		. . CAP, END *[17] (OPT)		1
212	M83248-1-218		. . O-RING *[17]		1
212	AS568-218-7115-70		. . RING *[17]		1
213	BACR12BP218		. . RING, BACK-UP *[17]		2
214	A2735		. . SEAL *[17] *[18]		1
215	66-13255-2		. . BEARING *[17]		2
216	69-69946-1		. . RETAINER *[17]		1
217	PGL-5276		. . BEARING *[17] V73680		2
218	69-69951-1		. . SPACER *[17]		1
219	65C17109-5		. . DUCT ASSY, TEE (USED ON 65C17142-3)		1
219	65C17109-1		. . DUCT ASSY, TEE (USED ON 65C17142-1)		1
219	65C17109-6		. . DUCT ASSY, TEE (USED ON 65C17142-4)		1
219	65C17109-1		. . DUCT ASSY, TEE (USED ON 65C17142-2)		1
220	*[13]		. BASIC STRUCTURE		1
-221	MS20995NC51		. . LOCK WIRE *[17]		AR

*[1] USE 14 WITH 65-46421-149 OR -150
USE 16 WITH 65-46421-173 OR -174
*[2] USE 6 WITH 65-46421-149 OR -150
USE 8 WITH 65-46421-173 OR -174
*[3] USE 14 WITH 65-46421-149 OR -150
USE 12 WITH 65-46421-173 OR -174
*[4] USE 7 WITH 65-46421-149 OR -150
USE 5 WITH 65-46421-173 OR -174
*[5] USE BACN10JD5 AND MS24665-134 WITH BACB30NR5D28 AND BACB30NR5D30
*[6] USE BACN10JC5 WITH NAS1105-30
*[7] USE BACN10JD6 AND MS24665-285 WITH BACB30NF6D33 AND BACB30NF6D34
*[8] USE BACN10JC6 WITH NAS1106-34
*[9] USED WITH 69-38641-1 AND -2
*[10] USED WITH 69-38641-17, -18 AND -23
*[11] NO BOEING PART NUMBER ASSIGNED; REWORKED FROM 65-55526-1 OR -2
*[12] NO BOEING PART NUMBER ASSIGNED; REWORKED FROM 65-55525-1 OR -2
*[13] NO BOEING PART NUMBER ASSIGNED
*[14] REPLACED BY NUTPLATE ON SOME ASSEMBLIES
*[15] REPLACES NUT AND WASHER (2 PLACES) ON SOME ASSEMBLIES
*[16] USED WITH 69-38641-29
*[17] USED WITH 65-51582-15 AND -16
*[18] PARKER HANNIFIN CORP., PARKER PACKING DIVISION, SALT LAKE CITY, UTAH
*[19] NOT USED WITH 65C17142-( ) TEE ASSYS
*[20] LIMITED USE
*[21] 65-46421-217, -221 IDENTICAL TO -213 EXCEPT FOR THE FOLLOWING DIFFERENCES: EXTERNAL FINISH; INNER SKINS 65-46421-169 USED ON - 213 AND 65-46421-225 USED ON -217, -221; TRAILING EDGE ASSEMBLIES 65-55536-35 USED ON -213, -217 AND 65-55536-41 USED ON -221.
*[22] 65-46421-218, -222 IDENTICAL TO -214 EXCEPT FOR THE FOLLOWING DIFFERENCES: EXTERNAL FINISH; INNER SKINS 65-46421-170 USED ON -214 AND 65-46421-226 USED ON -218, -222; TRAILING EDGE ASSEMBLIES 65-55536-36 USED ON -214, -218 AND 65-55536-42 USED ON -222.
*[23] DELETED
*[24] DELETED
*[25] SLAT ASSEMBLIES 65-46421-223, -229, -233 CONSIST OF THE SLAT ASSEMBLIES NO. 1 65-46421-221, -227, -231, -237, RESPECTIVELY; AND THE TAI DUCT INSTALLATION 65-54753-3, 65C17142-3.
*[26] SLAT ASSEMBLIES 65-46421-224, -230, -234 CONSIST OF THE SLAT ASSEMBLIES NO. 6 65-46421-222, -228, -232, -238, RESPECTIVELY; AND THE TAI DUCT INSTALLATION 65-54753-4, 65C17142-4.
*[27] NO EQUIVALENT BOEING PART NUMBER EXISTS FOR THOSE PARTS REWORKED BY SERVICE BULLETIN 737-57-1080R2.
*[28] SLAT ASSEMBLY PART NUMBERS 65-46421-155, -167, -197, -201 CONSIST OF SLAT ASSEMBLY PART NUMBERS 65-46421-107, -165, -195, -199 RESPECTIVELY, AND TAI DUCT/TUBE INSTALLATION PART NUMBERS 65-51582-9 AND 65-54753-3.
*[29] SLAT ASSEMBLY PART NUMBERS 65-46421-156, -168, -198, -202 CONSIST OF SLAT ASSEMBLY PART NUMBERS 65-46421-108, -166, -196, -200 RESPECTIVELY, AND TAI DUCT/TUBE INSTALLATION PART NUMBERS 65-51582-10 AND 65-54753-4.

- NOT ILLUSTRATED

VENDORS

V09455	DUKES, INC., 9060 WINNETKA AVE., NORTHRIDGE, CALIFORNIA 91324-3293
V21335	TIMKEN U.S. CORP., 336 MECHANIC ST., LEBANON, NEW HAMPSHIRE 03766-2614
V73680	GARLOCK, INC., DBA GARLOCK SEALING TECHNOLOGIES, 1666 DIVISION ST., PALMYRA, NEW YORK 14522-9343
V77896	REXNORD INDUSTRIES, INC., DIV. BEARING OPERATION, 2400 CURTISS ST., DOWNERS GROVE, ILLINOIS 60515-4037
V81376	RBC SOUTHWEST PRODUCTS, INC., 2240 BUENA VISTA, DUARTE, CALIFORNIA 91010-3318
V97613	DOVER DIVERSIFIED, INC., DBA KAHR BEARING, 5675 W. BURLINGAME RD., TUCSON, ARIZONA 85743-9453

Part No.	Fig. and Index No.	Qty. per Assy.
A2735	1101-214	1
AN960-10L	162	2
AN960-10L	162	2
AN960C816L	91A	
AN960D10L	150	3
AN960PD10	147H	2
AN960PD10	162	2
AN960PD10	172	2
AN960PD10L	49	AR
AN960PD10L	54	AR
AN960PD10L	61	*[4]
AN960PD10L	66	8
AN960PD416	112	3
AN960PD416	112	2
AN960PD416	126	2
AN960PD516	140	2
AN960PD8	107	1
AN960PD8	119	1
AN960PD8	123B	1
AN960PD8	133	1
AN960PD8	137B	1
AN960PD816	91B	AR
AN960PD8161	91C	AR
AS568-218-7115-70	212	1
BACB28X4B18	103	2
BACB28X4B18	115	2
BACB28X4B18	129	2
BACB28X5B10	83	1
BACB28X5B10	87	1
BACB28X5B5	99	4
BACB28X8B10	82	1
BACB28X8B10	86	1
BACB30FL3-2	152	3
BACB30GE5D13	91	2
BACB30GW10-6	138	2
BACB30GW8-6	124	2
BACB30GW8-8	110	2
BACB30GY6-3	98	8
BACB30LU3-2	1	8
BACB30LU3-2	8	3
BACB30LU3-2	11	5
BACB30LU3-2	14	4
BACB30LU3-2	14	3
BACB30LU3-2	17	4
BACB30LU3-2	17	3
BACB30LU3-2	20	8
BACB30LU3-2	23	3
BACB30LU3-2	26	8

Part No.	Fig. and Index No.	Qty. per Assy.
BACB30LU3-2	33	2
BACB30LU3-2	45	11
BACB30LU3-2	155	2
BACB30LU3-2	165	2
BACB30LU3-3	141	6
BACB30LU3-4	47	17
BACB30LU3-4	51	6
BACB30LU3-4	52	14
BACB30LU3-4	59	*[3]
BACB30LU3-4	64	20
BACB30LU3-4	147F	6
BACB30LU3-6	37	2
BACB30LU3-6	40	2
BACB30LU3-6	43	2
BACB30NE3-2	161	2
BACB30NF3-2	147D	3
BACB30NF3-3	147E	7
BACB30NF5D28	71	2
BACB30NF5D30	71	2
BACB30NF6D33	75	2
BACB30NF6D34	75	2
BACB30NM3K2	4	2
BACB30NM3K2	29	2
BACB30NM3K2	142	4
BACB30NN3K2	1	8
BACB30NN3K2	8	3
BACB30NN3K2	11	3
BACB30NN3K2	14	3
BACB30NN3K2	17	3
BACB30NN3K2	20	3
BACB30NN3K2	23	3
BACB30NN3K2	26	3
BACB30NN3K2	33	2
BACB30NN3K2	45	11
BACB30NN3K2	155	2
BACB30NN3K2	165	2
BACB30NN3K4	47	17
BACB30NN3K4	51	6
BACB30NN3K4	52	14
BACB30NN3K4	59	*[3]
BACB30NN3K4	64	20
BACB30NN3K4	147F	6
BACB30NN3K6	37	2
BACB30NN3K6	40	2
BACB30NN3K6	43	2
BACB30NR5DK28	71	2
BACB30NR5K30	71	2
BACB30NR6DK33	75	2

Part No.	Fig. and Index No.	Qty. per Assy.
BACC10DK2	1101-58	4
BACC30K10	139	2
BACC30K6	97	8
BACC30K8	111	2
BACC30K8	125	2
BACG20ZA1275	63	1
BACG20ZA400	63	1
BACG20ZA790	57	1
BACN10FX1	55	4
BACN10JC3	147G	2
BACN10JC3	171	1
BACN10JC4	111	3
BACN10JC4	111	2
BACN10JC4	125	2
BACN10JC5	69	2
BACN10JC5	139	2
BACN10JC6	73	2
BACN10JD5	69	2
BACN10JD5	89	2
BACN10JD6	73	2
BACN10JN3	147	2
BACN10JN3	159	2
BACN10JN3	168	2
BACN10JN3	202	2
BACN10JP3D	147I	4
BACN10PC3	146	4
BACP11L218	209	
BACP11L219	209	1
BACR12BP218	213	2
BACR15BA3A	123	2
BACR15BA3A	137	2
BACR15BA3D	145	12
BACR15BA3D	158	4
BACR15BA3D	167	4
BACR15BA3D	201	4
BACR15BA5D	109	2
BACR15BA5D	121	2
BACR15BA5D	135	1
BACR15BA6D	36	2
BACR15BA6D	39	2
BACR15BA6D	42	2
BACR15BA6D	174	2
BACR15BA6D	183	4
BACR15BA6D	195	4
BACR15BA6D	204	20
BACR15BB5D	108	1
BACR15BB5D	120	1

Part No.	Fig. and Index No.	Qty. per Assy.
BACR15BB5D	123C	1
BACR15BB5D	134	1
BACR15BB5D	137C	1
BACS12CB3-5	161	2
BACS12CB3-7	149	3
BACS12CB3-8	161	2
BACS12CB3-8	173	1
BACW10P163A	91A	2
BLFN5-003	79	1
KSBG5N5	79	1
M83248-1-218	212	1
MS20002-5	70	2
MS20002-5	90	2
MS20002-6	74	2
MS20253P2-175	192	1
MS20253P2-175	198	1
MS20253P2-200	187	1
MS20426D3	147J	AR
MS20615-5M5	184	3
MS20995NC51	221	AR
MS21232-5	79	1
MS24665-134	68	2
MS24665-134	88	2
MS24665-285	72	2
MS51923-130	186	2
MS51923-130	191	2
MS51923-130	197	2
NAS1103-2	4	2
NAS1103-2	29	2
NAS1103-2	142	4
NAS1103-3	143	6
NAS1103-4	48	6
NAS1104-7	124	2
NAS1104-8	110	2
NAS1104-9	110	3
NAS1105-7	138	2
NAS1106-34	75	2
NAS1368N3B	57	1
NAS42DD5-20	106A	1
NAS42DD5-20	123A	1
NAS42DD5-20	137A	1
NAS74A5-004P	92	2
NAS1103-4	52	*[1]
NAS1103-4	53	*[2]
NAS1103-4	60	*[4]
NAS1103-4	65	8
NAS1105-30	71	2

Part No.	Fig. and Index No.	Qty. per Assy.
PGL-5276	1101-217	2
SBS10ATC24	79	1
YTA119	79	1
03-728-0312	79	1
65-46421-107		RF
65-46421-108		RF
65-46421-11	50	1
65-46421-111	62	1
65-46421-112	62	1
65-46421-117	22	1
65-46421-118	22	1
65-46421-119	13	1
65-46421-12	50	1
65-46421-120	13	1
65-46421-121	16	1
65-46421-123	35	1
65-46421-125	10	1
65-46421-125	25	1
65-46421-127	19	1
65-46421-129	28	1
65-46421-130	28	1
65-46421-131	3	1
65-46421-132	3	1
65-46421-133	5	1
65-46421-134	30	1
65-46421-135	31	1
65-46421-136	31	1
65-46421-137	6	1
65-46421-138	6	1
65-46421-139	32	1
65-46421-140	7	
65-46421-147	30	1
65-46421-148	5	1
65-46421-149	56	1
65-46421-15	56	1
65-46421-150	56	1
65-46421-155		RF
65-46421-155		RF
65-46421-16	56	1
65-46421-165		RF
65-46421-166		RF
65-46421-167		RF
65-46421-168		RF
65-46421-169	62	1
65-46421-17	62	1
65-46421-170	62	1
65-46421-173	56	1

Part No.	Fig. and Index No.	Qty. per Assy.
65-46421-174	56	1
65-46421-179	16	1
65-46421-18	62	1
65-46421-180	19	1
65-46421-181	15	1
65-46421-182	18	1
65-46421-188	16	1
65-46421-190	19	1
65-46421-192	15	1
65-46421-194	18	1
65-46421-195		RF
65-46421-196		RF
65-46421-197		RF
65-46421-198		RF
65-46421-199		RF
65-46421-200		RF
65-46421-201		RF
65-46421-202		RF
65-46421-203		RF
65-46421-204		RF
65-46421-205		RF
65-46421-206		RF
65-46421-207		RF
65-46421-208		RF
65-46421-21	67	1
65-46421-213		RF
65-46421-214		RF
65-46421-217		RF
65-46421-218		RF
65-46421-22	67	1
65-46421-221		RF
65-46421-222		RF
65-46421-223		RF
65-46421-224		RF
65-46421-227		RF
65-46421-228		RF
65-46421-229		RF
65-46421-230		RF
65-46421-231		RF
65-46421-232		RF
65-46421-233		RF
65-46421-234		RF
65-46421-235		RF
65-46421-236		RF
65-46421-237		RF
65-46421-238		RF
65-46421-50	15	1

Part No.	Fig. and Index No.	Qty. per Assy.
65-46421-51	1101-21	1
65-46421-52	21	1
65-46421-55	12	1
65-46421-56	12	1
65-46421-63	34	1
65-46421-67	27	1
65-46421-68	27	1
65-46421-69	2	1
65-46421-70	2	1
65-49448-1	77	2
65-49448-11	77	2
65-49448-12	78	1
65-49448-2	78	1
65-49448-7	77	2
65-49448-8	78	1
65-51570-1	208	1
65-51570-2	208	1
65-51570-7	208	1
65-51570-8	208	1
65-51582-1	154	1
65-51582-10	154	1
65-51582-15	154	1
65-51582-16	154	1
65-51582-2	154	1
65-51582-7	156	1
65-51582-8	163	1
65-51582-9	154	1
65-51583-1	207	2
65-51583-2	207	2
65-51583-5	207	2
65-51583-6	207	2
65-51583-7	207	2
65-51583-8	207	2
65-54753-3	148	1
65-54753-4	148	1
65-54754-12	151	1
65-54754-13	151	1
65-54755-12	153	1
65-54756-1	-208A	1
65-55525-1	127	1
65-55525-10	128A	1
65-55525-11	127A	1
65-55525-12	127A	1
65-55525-13	128A	1
65-55525-14	128A	1
65-55525-2	127	1

Part No.	Fig. and Index No.	Qty. per Assy.
65-55525-5	128	1
65-55525-6	128	1
65-55525-7	127A	1
65-55525-8	127A	1
65-55525-9	128A	1
65-55526-1	113	1
65-55526-2	113	1
65-55526-3	114	1
65-55526-4	114	1
65-55526-5	113A	1
65-55526-6	113A	1
65-55526-7	114A	1
65-55526-8	114A	1
65-67110-1	46	1
65-67110-2	46	1
65-79962-1	127A	1
65-79962-2	127A	1
65-79963-3	102A	1
65-79963-4	102A	1
65-79963-5	113A	1
65-79963-6	113A	1
65C17109-1	219	1
65C17109-2	219	1
65C17109-5	219	1
65C17109-6	219	1
65C17106-1	208A	1
65C17142-1	208	1
65C17142-2	208	1
65C17142-3	208	1
65C17142-3	208	1
65C17142-4	208	1
65C17142-4	208	
66-13255-1	207	2
66-13255-2	215	2
66-24198-7	122	1
66-24198-7	136	1
69-37867-26	100	2
69-37867-43	100	2
69-38641-1	164	1
69-38641-17	164	1
69-38641-18	164	1
69-38641-2	164	1
69-38641-23	164	1
69-38641-26	206	1
69-38641-29	164	1
69-38641-7	206	1

Part No.	Fig. and Index No.	Qty. per Assy.
69-38641-8	1101-206	1
69-38644-1	205	1
69-38644-4	205	1
69-38644-7	205	1
69-38655-10	203	1
69-38655-11	189	1
69-38655-14	196	1
69-38655-16	177	1
69-38655-17	200	1
69-38655-18	160	1
69-38655-18	169	1
69-38655-19	157	1
69-38655-19	166	1
69-38655-2	185	1
69-38655-20	175	1
69-38655-21	170	1
69-38655-23	178	1
69-38655-24	179	1
69-38655-25	181	1
69-38655-26	180	1
69-38655-27	176	1
69-38655-27	198	1
69-38655-29	182	1
69-38655-3	190	1
69-38655-4	196	1
69-38655-7	188	1
69-38655-8	194	1
69-38655-8	199	1
69-38655-9	193	1
69-40611-1	84	1
69-40611-2	85	1
69-40628-1	93	2
69-40628-2	94	2
69-40666-3	76	4
69-40694-1	80	1
69-40694-2	81	1
69-43504-3	95	2
69-43504-4	96	2
69-43539-1	144	1
69-44922-2	44	1
69-44923-2	41	1
69-44924-3	38	1
69-44971-1	9	1
69-44971-1	24	1
69-44971-2	18	1

Part No.	Fig. and Index No.	Qty. per Assy.
69-50764-1	104	1
69-50764-1	116	1
69-50764-1	130	1
69-50764-2	104	1
69-50764-2	116	1
69-50764-2	130	1
69-50764-3	105	1
69-50764-3	117	1
69-50764-3	131	1
69-50764-4	106	1
69-50764-4	118	1
69-50764-4	132	1
69-50764-5	106	1
69-50764-5	118	1
69-50764-5	131	1
69-54929-1	102B	1
69-54929-2	102B	1
69-54929-3	102B	1
69-54929-4	102B	1
69-54930-1	102C	1
69-54930-2	102C	1
69-54930-3	102C	1
69-54930-4	102C	1
69-54931-1	101A	1
69-54931-2	101A	1
69-54931-3	101A	1
69-54931-4	101A	1
69-54931-5	101A	1
69-54931-6	101A	1
69-54932-1	101	1
69-54932-1	102A	1
69-54932-2	101	1
69-54932-2	102A	1
69-54932-3	102	1
69-54932-3	102A	1
69-54932-4	102	1
69-54932-4	102A	1
69-54932-5	102A	1
69-54932-6	102A	1
69-62797-1	136A	1
69-62797-2	136A	1
69-62797-5	122A	1
69-62797-6	122A	1
69-68803-1	147A	
69-68803-2	147B	

Part No.	Fig. and Index No.	Qty. per Assy.
69-68803-3	1101-147C	1
69-69944-1	211	1
69-69946-1	216	1
69-69951-1	218	1
69-73510-1	44	1
69-75511-1	41	1
69-73512-1	38	1
*[11]	114A	1
*[11]	114A	1
*[12]	128A	1
*[12]	128A	1
*[13]	210	1
*[13]	220	1

Part No.	Fig. and Index No.	Qty. per Assy.