

TO: ALL HOLDERS OF ENGINE CONTROLS COMPONENTS ASSEMBLY OVERHAUL MANUAL,
 76-14-11

REVISION NO. 11, DATED NOV 1/02

HIGHLIGHTS

DESCRIPTION OF CHANGE	TOPICS AFFECTED												
	D & O	D / Assy	Cleaning	Inspect / Check	Repair	Assy	F / C	Test	T / Shooting	S / Tools	Storage	IPL	L / Overhaul
Revised the finishes for 65-45136-() drum assemblies					X								
Edited without technical change					X								
Added 65-45136-17, -18, -19 drum assemblies with changed cams per PRR 34640, and 65-45136-20 drum assembly without a cam												X	
Updated the vendors list												X	

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ENGINE CONTROL COMPONENTS

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BOEING P/N 65-23761-4, -5, -15 thru -20
 65-45136-1, -4, -6, -7 thru -15, -17 thru -20
 65-46506-1 thru -10, -21, -22, -29, -30, -35 thru -48
 65C16481-301 thru -328, -330 thru -357
 65-35354-3 thru -6

AIRLINE P/N

THE FOLLOWING DIRECTIVES APPLY TO THIS SUBJECT:

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVES	DATE DIRECTIVE INCORPORATED INTO TEXT
76-1002		PRR 30795 PRR 31270 PRR 31030-14 PRR 31113 PRR 31231	May 15/69 May 15/69 Aug 15/69 Aug 15/69 Aug 15/69
71-1016 71-1016, Rev 1		MC 3510-4K MC 3510-4K PRR 31883 PRR 31926	Jun 10/70 Jun 10/70 Jun 10/70 Jun 10/70
22-1011 22-1012 28-1020			Sep 10/71 Sep 10/71 Sep 25/73 Dec 25/74 Jun 25/75 Dec 25/75
76-1006		PRR 32056 PRR 32388 PRR 32885-10 PRR 32540 PRR 32747 PRR 32755 PRR 32755-R PRR 32811	Jul 5/78 Jul 5/78 Jul 5/78 Jul 5/78 Jul 5/78 Jul 5/78 Jul 5/78
28-1035 76-1007 76-1006 R2		PRR 33076 PRR 32902-R PRR 34640	Jul 5/78 Jul /582 Dec 5/83 Nov 1/02

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LIST OF EFFECTIVE PAGES

* Indicates pages revised, added or deleted in latest revision
 F Indicates foldout pages - print one side only

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76-14-11		1120	Jun 5/88		
* T-1	Nov 1/02	1121	BLANK		
T-2	BLANK	1122	Jun 5/88		
* LEP-1	Nov 1/02	* 1123	Nov 1/02		
LEP-2	BLANK	* 1124	Nov 1/02		
T/C-1	Jun 5/88				
T/C-2	BLANK				
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2	BLANK				
101	Jun 5/88				
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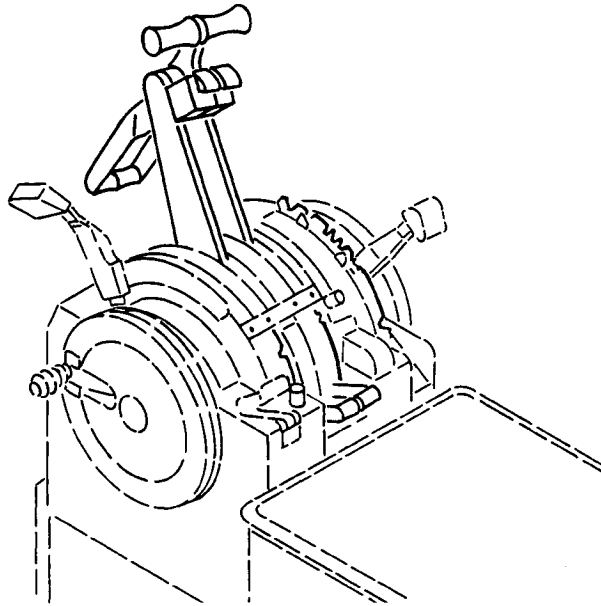
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65-23761 69-35354
65-45136 65C16481
65-46506

BOEING 
COMMERCIAL JET
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ENGINE CONTROL COMPONENTS



Engine Control Components
Figure 1

DESCRIPTION AND OPERATION

1. The engine control components consist of thrust handle assemblies, reverse-thrust-detent roller assemblies, start system lever assemblies, and engine start drum assemblies. Linkages between components are made in the control stand upper mechanism assembly (Ref 27-16-11).
2. The thrust handle assembly includes a reverse thrust lever with: an interlock which permits movement of a reverse thrust lever only when the thrust lever is in forward idle position, a stow detent/stop function. The reverse-thrust-detent roller assembly engages the thrust drums, allowing a feel of the position of thrust or reverse thrust levers. Each start lever connects to a start drum through linkage, and lever action is transmitted through the linkage to rotate the drum.
3. Leading Particulars (Approximate)

A.	Component	<u>Length</u>	<u>Width</u>	<u>Height</u>
(1)	Thrust handle assy	8 in	4 in	
(2)	Reverse-thrust-detent roller assy	2 in	2 in	3 in
(3)	Start system lever assy	5 in	5 in	
(4)	Engine start drum assy	3 in	2 in	3 in

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65-45136 69-35354
65C16481



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DISASSEMBLY

NOTE: Identify assemblies and parts, including washers, shims and spacers, to facilitate reassembly.

1. Disassemble thrust handle assembly (1) (IPL Fig. 1101):

NOTE: Refer to use codes in parts list to determine applicability of items to configuration being disassembled.

- A. Use standard industry practices to disassemble thrust handle assembly.

CAUTION: USE CARE TO AVOID DAMAGE TO WIRE INSULATION.

NOTE: Do not remove or disassemble the following parts unless necessary for repair or replacement:

Drum (5) from drum brake (59). Inserts (9, 112) from drum (11, 111). Bearing (12, 12A) from drum (10, 5). Rivet (18) and link (19) from lever (22). Rivets (31) or housing (33) from handle or lever (58). Rivet (46), washer (47), pawl (48) or pin (43A), rocker arm (43B) from lever (58). Wire bundle (49), terminal (50), rivets (51), wire guide (52), nut (53), screw (54), clamp (55), conduit (56), rivet (57), lever (58), drum (59) or bearing (60).

2. Disassemble reverse-thrust-detent roller assembly (1) (IPL Fig. 1102):

- A. Remove spring (5), nut (10), washer (20), eyebolt (25), and spacer (30) or nut (15).
- B. Remove nut (35), washer (40), bolt (45), spacer (50), and cam arm assemblies (55) from bracket assembly (95).

NOTE: Do not remove rivets (60), cam (65), bearing (85), and cam arm (90) from cam arm assemblies (55); or rivet (70), bearing (75), and cam arm (80) from cam assembly (65) unless necessary for repair.

Do not remove rivets (100), bracket (105), and support (110) from bracket assembly (95) unless necessary for repair.

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3. Disassemble start system lever assembly (1) (IPL Fig. 1103):
 - A. Remove screw (5), knob (10), rivet (15), washers (20), spacer (25), and rod assembly or link assembly (30).

NOTE: Do not separate tube (60) from fork (65) and rod end assembly (70) or remove bearing (80) from rod end (75) unless necessary for repair.

NOTE: Do not separate links (40, 45) or remove bearing (50) and sleeve (52) unless necessary for repair.
 - B. Align levers (90, 105) as necessary; use punch to remove pin (85). Remove engine start lever (90), spring (95), and stop (100) from lever assembly (105).

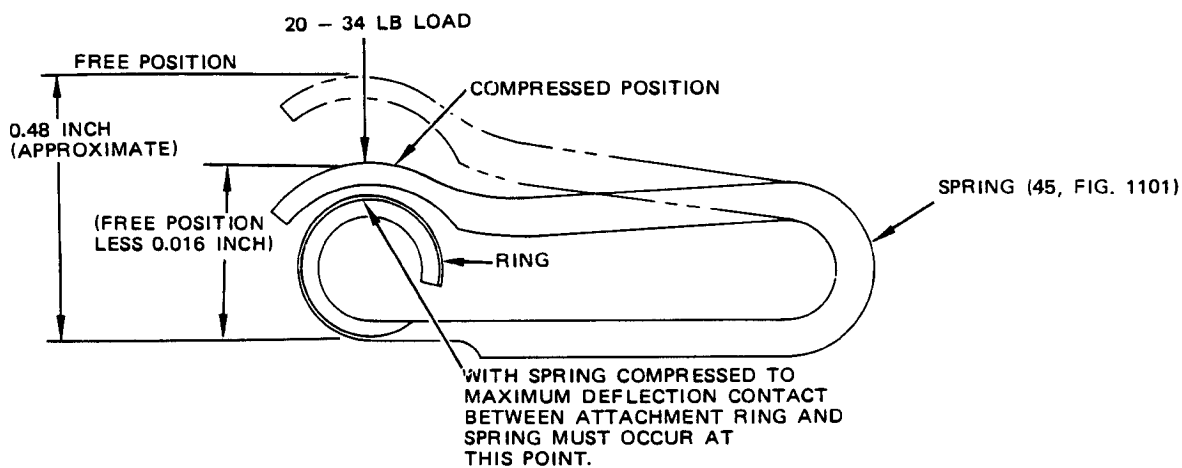
NOTE: Do not remove bearing (115), rivet (120), or handle stop (125) from lever (110) unless necessary for repair.
4. Disassemble drum assembly (1) (IPL Fig. 1104):
 - A. Remove bolts (5, 65), washers (10, 70), nuts (60), cams (15, 20, 25) and plate (30).

NOTE: Do not remove bearings (35, 40) and insert (55) from drum (50) unless necessary for repair.

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INSPECTION/CHECK

1. Check all parts for obvious defects in accordance with standard industry practices.
2. Penetrant examine per 20-20-02 -- (Fig. 1101) Drum (11), reverse thrust lever (22), rocker arm (43B), lever (58), thrust brake drum (59); (Fig. 1103) Engine-start lever (90), start-lever assy (110); (Fig. 1104) Drum (45).
3. Magnetic particle examine per 20-20-01 -- (Fig. 1101) Pawl (48).
4. Check springs (45, Fig. 1101) (5, Fig. 1102) (95, Fig. 1103) per Fig. 301.



65-74599-1, 69-35369-1

Spring Check Data
Figure 301 (Sheet 1)

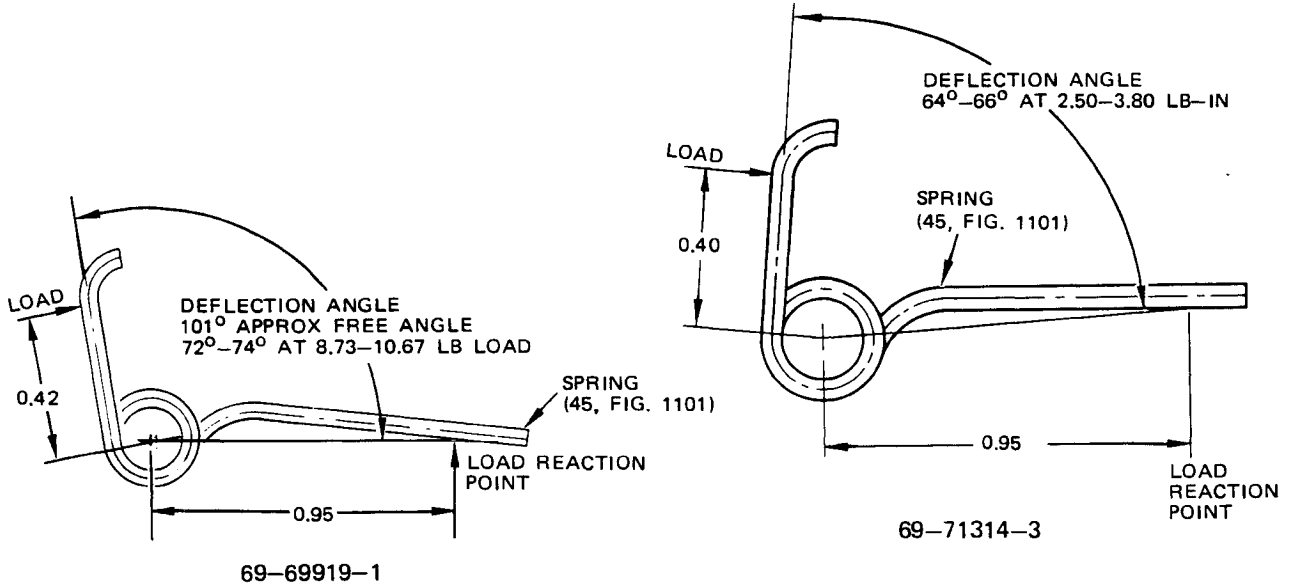


Fig. and Index No.	Approximate Free Length (Inches)	Test Length (Inches)	Allowable Load Limits (Pounds)
1102-5	1.80	2.46	21.6 - 26.4
		2.67	27.0 - 33.0
1102-5	1.80	2.45	11.5 - 12.5
		2.65	14.5 - 15.5
1103-95	1.29	0.94	0.81 - 1.81
		0.56	2.24 - 3.24

Spring Check Data
 Figure 301 (Sheet 2)

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REPAIR

1. Materials

- A. Solvent, Methyl Ethyl Ketone -- TT-M-261 (Ref 20-60-01)
- B. Bonding Adhesive -- BMS 5-31, Type 51 (Ref 20-50-12)
- C. Electrical Insulating Sealant Compound -- BMS 5-37, Class B (Ref 20-60-04)
- D. Lacquer -- Flat black color 37038, TT-L-20 (Ref 20-60-02)
- E. Primer -- BMS 10-11, Type 1 (Ref 20-60-02)
- F. Solvent -- P-D-680 (Ref 20-60-01)
- G. Dry Film Lube -- BMS 3-8, Class A (Ref 20-50-08)

2. Repair minor defects in accordance with standard industry practices.

3. Refinish

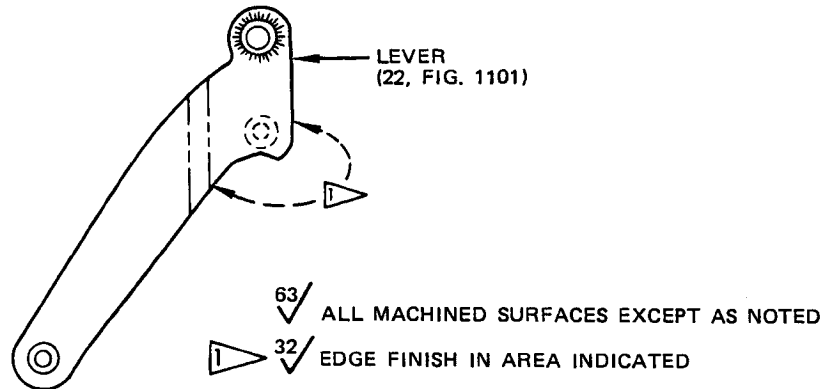
NOTE: Refer to 20-30-02 for stripping of protective finishes, and to 20-41-01 for explanation of F and SRF finish codes.

- A. Reverse thrust special bolt (3, Fig. 1101) -- Cadmium plate (F-1.1913) all over. Hold diameter of shank to 0.3108- to 0.3117 inch after plating. Material: 8630 steel (160-180 ksi).
- B. Bearing retainer (7, Fig. 1101), retainer (38, Fig. 1101), drum brake (59, Fig. 1101) (65-1634-1, -2, -5, -6), bracket (105, Fig. 1102), support (110, Fig. 1102), tube (60, Fig. 1103), fork (65, Fig. 1103) -- Alodize and apply one coat primer BMS 10-11, Type 1 (SRF-2.30) all over except no primer in bearing bore or grooves, or on faying surfaces. Material: Al alloy.
- C. Drum (11, Fig. 1101) -- Alodize and apply one coat primer BMS 10-11, Type 1 (SRF-2.19) all over except faying surfaces of bearing bore and cam detent seats. Material: Al alloy.
- D. Special Bolt (15, Fig. 1101) -- Cadmium plate (F-1.1913) all over. Hold diameter of shank to 0.1885- to 0.1894 inch after plating. Apply dry lubricant (F-19.10) to shank 0.0002- to 0.0005 inch thick, except in no case shall there be an interference between special bolt (15) and bushing (16). Material: 8630 steel (160-180 ksi).
- E. Reverse thrust link (19, Fig. 1101), reverse thrust lever (22, Fig. 1101), housing (33, Fig. 1101), cover (35, Fig. 1101), stop bolt (43, Fig. 1101), lever (58, Fig. 1101), engine-start lever (110, Fig. 1103). Material: CRES (301, 302, 347, 17-4, 17-7) (150-200 ksi) and 4340 steel (125-145 ksi).

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- (1) Apply dull-matte finish, dry abrasive blast, all over per 20-30-03. Use 180 or finer garnet, aluminum oxide or silicon carbide grit for dry blasting.
- (2) Apply additional finish to the following:
 - (a) Reverse thrust link (19, Fig. 1101) -- Apply dry lubricant (F-19.10) to all surfaces included in "S-curve" portion which have rounded edge (throughout approximate 4-inch total length).
 - (b) Lever (58, Fig. 1101) -- Nickel plate (F-1.803) all over. Optional finish chrome plate (F-14.111).
 - (c) Engine start lever (90, Fig. 1103) -- Apply dry lubricant (F-19.10) to all sliding surfaces, approximately 2.5 inches.
 - (d) Reverse thrust lever (22, Fig. 1101) -- Refinish per Fig. 401.

REFINISH
CHROME PLATE (F-14.111)
ALL OVER



Lever Refinish
Figure 401

- F. Control lever knob (21, Fig. 1101), knob assembly (24, Fig. 1101), paddle (41, Fig. 1101), knob (10, Fig. 1103) -- Apply lacquer per TT-L-20, flat black color 37038, Federal Standard 595, to fill letters. Material: Plastic.
- G. Retainer (28, Fig. 1101) -- Alodize and apply one coat primer BMS 10-11, Type 1 (F-2.31) plus lacquer (F-14.903-705) all over except omit primer on threads. Material: Al alloy.

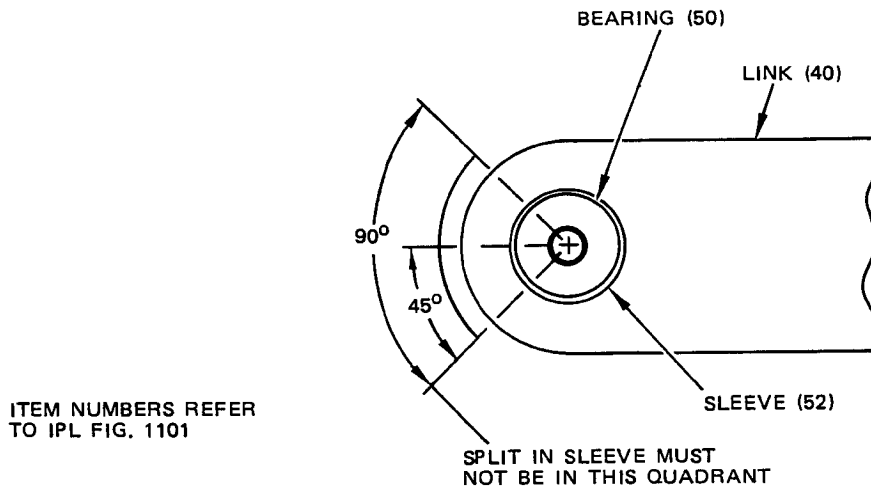
- H. Spring (45, Fig. 1101, 65-74599-1) -- Chrome plate (F-15.04), 0.003- to 0.004-inch thick, a length of 0.65- to 0.75 inch on spring-end which will contact reverse thrust lever (22). Do not plate flexing end which is supported by pin (44). Maintain 32-microinch finish on plated part; grinding permitted. Passivate (F-8.07) all over except on plated area. Material: 17-7PH CRES (TH1050 per BAC5619).
- I. Spring (45, Fig. 1101, 69-35369-1) -- Chrome plate (F-15.02) external surface of spring tongue. Maintain 32-microinch finish on plating, grinding permitted. Apply dry film lube (F-19.10) over plated area and edges, overspray permitted. Plate to within approximately 0.32 inch of 180-degree bend (flexing end). Material: CRES AISI, Type 302, spring temper per QQ-V1-423, comp 302, cond B.
- J. Spring (45, Fig. 1101, 69-69919-1, 69-71314-3) -- Apply two coats primer, BMS 10-11, Type 1 (F-20.03) all over. Material: 9254 chrome-silicon wire per QQ-W-412 comp 2, Type 1.
- K. Rocker arm (43, Fig. 1101) -- Cadmium plate (F-15.06) all over, except no plating in hole. Apply solid film lubricant (F-19.10). Material: Manganese bronze per AMS 4862, opt Al-Ni-Br per AMS 4640.
- L. Spacer (44A, Fig. 1101) -- Cadmium plate (F-15.06) all over, except no plating in hole. Material: Al-Ni-Br per AMS 4640.
- M. Pawl (48, Fig. 1101) -- Cadmium plate (F-1.20) or zinc plate (F-1.205) all over. Material: 17-4PH CRES (160-220 ksi).
- N. Spring (5, Fig. 1102) -- Cadmium plate and apply two coats primer (SRF-1.92) all over, except omit primer from 69-34830. Bake 3 hours at 350-400°F after plating. Material: Music wire.
- O. Eyebolt (25, Fig. 1102, 66-1551) -- Cadmium plate (F-1.20) or zinc plate (F-1.205) all over. Material: 8630 or 4130 steel.
- P. Cam arm (80, Fig. 1102) -- Cadmium plate and apply one coat primer BMS 10-11, Type 1 (SRF-1.285) all over. Material: 8630 steel.
- Q. Cam arm (90, Fig. 1102) -- Anodize and apply one coat primer BMS 10-11, Type 1 (SRF-2.19) all over except in bearing bore. Material: Al alloy.
- R. Rod end (75, Fig. 1103) -- Cadmium plate and apply one coat primer BMS 10-11, Type 1 (F-1.30) all over except no primer in bearing bore. Material: 8630 or 4130 steel.
- S. Spring (95, Fig. 1103), stop (100, Fig. 1103) -- Cadmium plate (F-1.20) all over. Bake 3 hours at 350-400°F. Material: Music wire.
- T. Lever (110, Fig. 1103) -- Alodize (SRF-2.30) all over except in bearing bore, plus dry lubricant (F-19.10) in slotted area which mates and slides on engine-start lever (90, Fig. 1103). Material: Al alloy.

- U. Handle stop (125, Fig. 1103) -- Apply primer BMS 10-11, Type 1 (SRF-12.206) all over. Material: 17-4 PH CRES (180-200 ksi).
- V. Links (40, 45, Fig. 1103) -- Anodize (F-2.20) all over. Material: Al alloy.
- W. Drum (50, Fig. 1104) -- Anodize (F-17.05). Apply BMS 10-11, Type 1 primer (F-20.02) all over but not in bearing bore and groove. Material: Al alloy.
- X. Link assembly (30, Fig. 1103) -- Apply BMS 10-11, Type 1 primer (SRF-12.205) but not in holes or on bearing (50, Fig. 1103). Material: Al alloy.
- Y. Plate (30, Fig. 1104) -- Chromic acid anodize (F-17.04). Material: Al alloy.
- Z. Drum brake (59, Fig. 1101) (65-1634-11, -12) -- Chemical treat surface or chromic acid anodize (F-18.05). Apply BMS 10-11, Type 1 primer. Material: Al alloy.

4. Replacement

- A. Bearings (12, 12A, Fig. 1101) (85, Fig. 1102) (80, Fig. 1103) -- Install replacement bearings (SOPM 20-50-03). Roller swage bearings (12, 12A, 85), and press fit bearing (8).
- B. Bearings (60, Fig. 1101) (115, Fig. 1103) (35, 40, Fig. 1104) -- Install replacement bearings and roller swage them per SOPM 20-50-03.
- C. Cam follower bearing (75, Fig. 1102) and retaining rivet (70, Fig. 1102) -- Install in cam arm (80, Fig. 1102). If the lube fitting in the cam follower bearing is out above the bearing, remove the fitting and plug the hole with CRES plug 66-13328-3.
- D. Bearing (50, Fig. 1103) -- Install with sleeve (52) in link (40) and roller swage per SOPM 20-50-03. Be sure to turn the split in sleeve (52) as shown in Fig. 402. Sleeve must be flush with bearing 0.005 inch after swaging.

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Bearing Replacement
Figure 402

- E. Helical coil insert (9, Fig. 1101) (55, Fig. 1104) -- Install with wet primer, BMS 10-11, Type 1, 1/4 to 1/2 turn below surface of part. Remove tang.
- F. Thrust knob insulating disk (25, Fig. 1101).
- (1) Remove all remnants of defective part from knob assembly (24, Fig. 1101). Clean replacement contact areas with methyl ethyl ketone. Dry the surfaces.
 - (2) Prepare BMS 5-31, Type 51 adhesive. Thoroughly blend 30 parts by weight of accelerator to 100 parts by weight of base resin.
NOTE: Pot life is 1-1/2 hours at 60°F or 20 minutes at 77°F.
 - (3) Apply adhesive to contact surface of thrust knob insulating disk (25, Fig. 1101) and knob assembly (24, Fig. 1101). Assemble and apply sufficient pressure to ensure contact throughout contact area. There should be evidence of extruding excess adhesive. Wipe out excess adhesive, but do not smear adjacent surfaces. Maintain only sufficient pressure to hold parts in place until adhesive sets.
NOTE: Methyl ethyl ketone may be used in minimum amount to clean out excess adhesive. Avoid contamination of bonding adhesive.
 - (4) Cure 12 hours at 70°F, or 2 hours at 150 to 160°F or 4 hours at 120 to 130°F.

- G. Switch (29, 39, Fig. 1101) -- Remove solder-terminal lugs and replace with crimp lugs (50, Fig. 1101). Salvage terminal screws and lockwashers to connect wiring. Salvage mounting nut and lockwasher to use for installation with retainer (28, Fig. 1101).
- H. Wire bundle assembly (49, Fig. 1101).
- (1) Remove defective wire bundle assembly (49) and remnants of potting material to permit installation of replacement wire bundle assembly. Clean potting areas with solvent, Specification P-D-680 or equivalent. Dry with clean cloth or dry air.
 - (2) Remove wire bundle outer teflon covering from that portion of wiring which will be in switch assembly (32) or knob assembly (24). Remove sufficient covering to provide slack for installation and removal of switch (29) in knob assembly (24).
 - (3) Pull wire bundle assembly (49) through wire guide (52), conduit (56), lever (58), and into switch housing (33). Stripped wire must be pulled entirely through lever (58) and entrance hole in switch housing (33) so that teflon bundle covering will protect wiring from insulating compound.
 - (4) Form a suitable dam in lever (58) where wire bundle (49) passes through lever (58) to restrict entrance of insulating compound. Provide enough area to bond cable to lever, but limit total bulk of compound to prevent interference with function of lever and components.
 - (5) Measure and prepare sealant BMS 5-37, Class B, electrical insulating sealant compound in accordance with manufacturer's instructions.
 - (6) Inject insulating compound into prepared areas in lever (58) to fill cavity; build up a fillet around cable in switch housing (33). Use care to prevent excess compound entering lever areas and switch housing.
 - (7) Inject sufficient compound into top end of conduit (56) to seal it; mold it flush with top.
 - (8) Cure according to manufacturer's instructions.
 - (9) When cure is complete, remove dam. Remove excess compound.
- I. Rocker arm (43B, Fig. 1101) -- Install in lever (58, Fig. 1101) with pin (43A, Fig. 1101).

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ASSEMBLY

1. Assemble Engine-Start Drum Components (Fig. 1104)
 - A. Install cams (15) with bolts (5) and washers (10).
 - B. Install cams (20, 25) and plate (30) with bolts (65), washers (70) and nuts (60).
2. Assemble Start System Lever Assembly (Fig. 1103)
 - A. Position spring (95) on stop (100) and place in start lever assembly (105), aligning hole in stop (100) with that in lever assembly (105). Slide engine-start lever (90) into place in start lever assembly (105), aligning holes in each with that in stop (100). Insert pin (85) through hole in start lever assembly (105) and into position free of lever assembly, but pinning engine-start lever (90) to stop (100).
 - B. Install knob (10) with screw (5).
 - C. Install rod or link assembly (30), spacer (25), washers (20), and rivet (15).
3. Assemble Control Stand Reverse-Thrust-Detent Roller Assembly (Fig. 1102)
 - A. Position cam arm assemblies (55) and spacer (50) in bracket assembly (95); install bolt (45), washer (40) and nut (35).
 - B. Install spacers (30) or nut (15), if applicable, eyebolts (25), washers (20), and nuts (10) as necessary.
 - C. Attach springs (5) to eyebolts (25) and cam arm assemblies (55).
4. Assemble Thrust Handle Assembly (Fig. 1101)
 - A. Use standard industry practices for assembly of this component, and the following additional procedures.
 - B. Position switch (39) so that paddle (41) contacts stop on housing and switch actuates when paddle is pressed. Tighten screw until switch cannot be rotated. Paddle must rotate freely without binding.
 - C. Install bearing (8) per 20-50-03.
 - D. Install link (19) and attach lever (22) on drum (59) with bolt (3), washer(s) (4) and nut (3). Use washers as required for clearance. Use MS20995NC40 lockwire to secure nut (3), install lockwire per 20-50-02, single method.

CAUTION: LOCKWIRE SHALL NOT EXTEND ABOVE SURFACE OF DRUM.

E. For all thrust handle assemblies:

- (1) Reverse thrust lever stow detent/stop components.
 - (a) Position spring (45, 65-74599-1 or 69-35369-1) in lever (58) and install pin (44), stop bolt (43), and nut (42). Tighten nut to 4-6 lb-in. Align pin (44) so that chamfered ends protrude equally from each side. Grind or file chamfered ends until they are flush with lever sides. Touch up finish on lever (58) as necessary.
 - (b) Position spring (45, 69-69919-1, 69-71314-3) in lever (58), with arched end in rocker arm (43B) slot, and install pin (44). Pull other end of spring forward sufficiently to clear 0.094- to 0.097-inch hole and install pin (44). Install pins (44) as in (a) above.

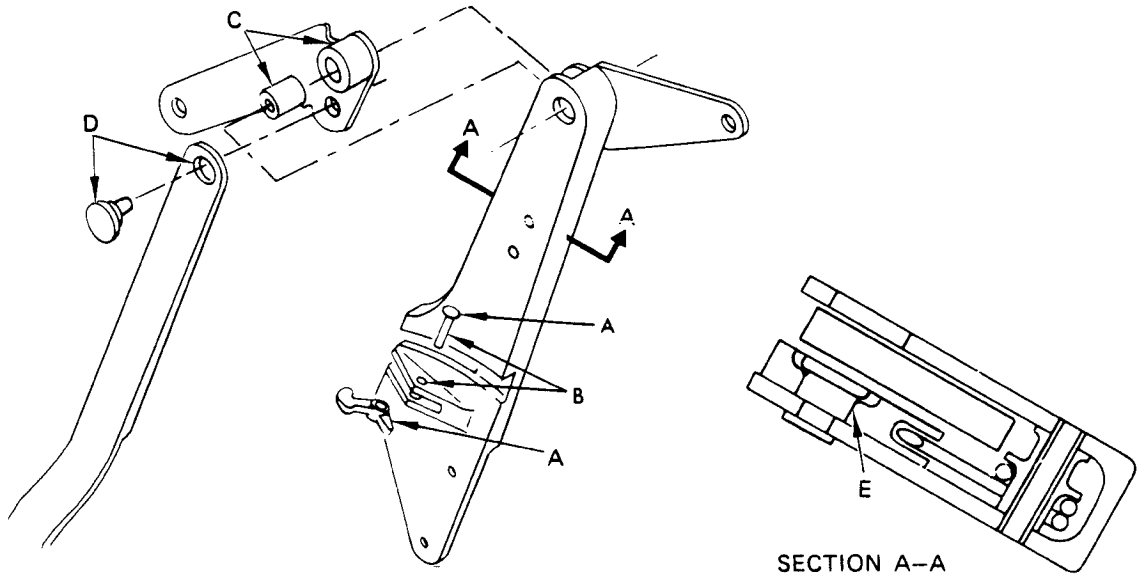
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69-35354

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OVERHAUL MANUAL

FITS AND CLEARANCES



SECTION A-A

THRUST HANDLE ASSEMBLY
ENGINE NO. 1 SHOWN

		Design Dimensions				Service Wear Limits		
Ref Letter Fig. 601	Mating Item No. Fig. 1101	Dimensions (inches)		Assembly Clearance (inch)		Dimension Limits (inches)		Maximum Allowable Clearance (inch)
		Min	Max	Min	Max	Min	Max	
A	ID 48	0.125	0.130	*[1] 0.004	0.009		0.143	0.014
	OD 46	0.121	0.129			0.117		
B	ID 58	0.1285	0.1335	*[1] 0.0005	0.0125		0.1490	0.020
	OD 46	0.1210	0.1290			0.117		
C	ID 22	0.3150	0.3160	0.0014	0.0029		0.3215	0.0079
	OD 16	0.3131	0.3136			0.3101		
D	ID 19	0.375	0.379	0.002	0.009		0.389	0.016
	OD 18	0.370	0.373			0.366		
E	ID 43B	0.190	0.191	0.003	0.005			
	OD 43A	0.186	0.187					

*[1] Interference fit

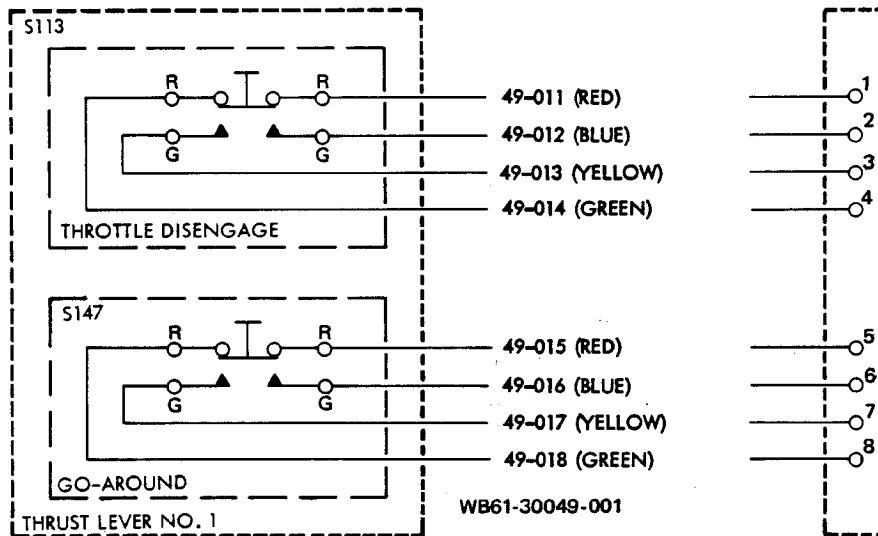
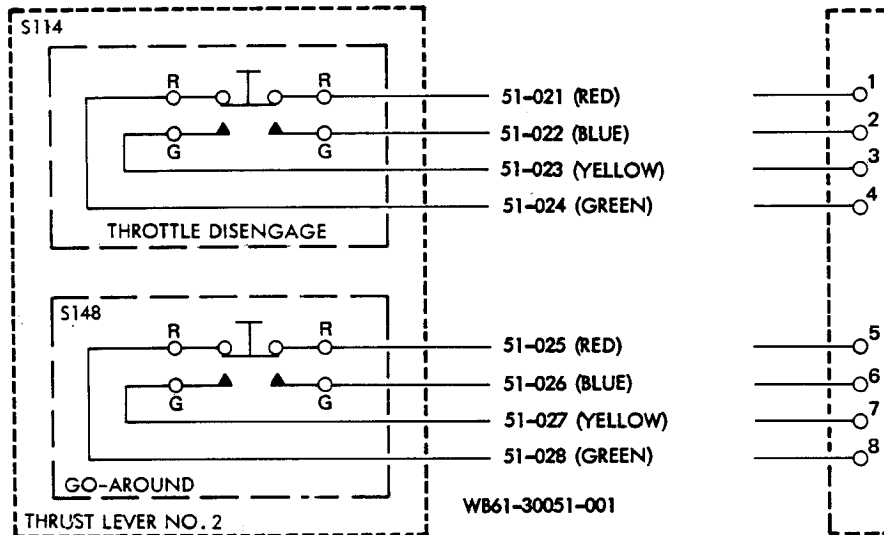
Fits and Clearances
 Figure 601 (Sheet 2)

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TESTING

1. Test electrical continuity through switches (29, Fig. 1101), (if applicable, 39, Fig. 1101) and wire bundle assemblies (49, Fig. 1101) (Fig. 701).
2. Verify continuity (less than 1 ohm) or no continuity (infinite resistance) between pins (Fig. 702).



Wire Bundle Schematic
 Figure 701

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Switch Position	From Pin	To Pin	Reading
Released	1	4	Continuity
	2	3	No Continuity
	5	8	Continuity
	6	7	No Continuity
Depressed	1	4	No Continuity
	2	3	Continuity
	5	8	No Continuity
	6	7	Continuity

Continuity Check
Figure 702

3. Check mechanical operation (Fig. 1101)

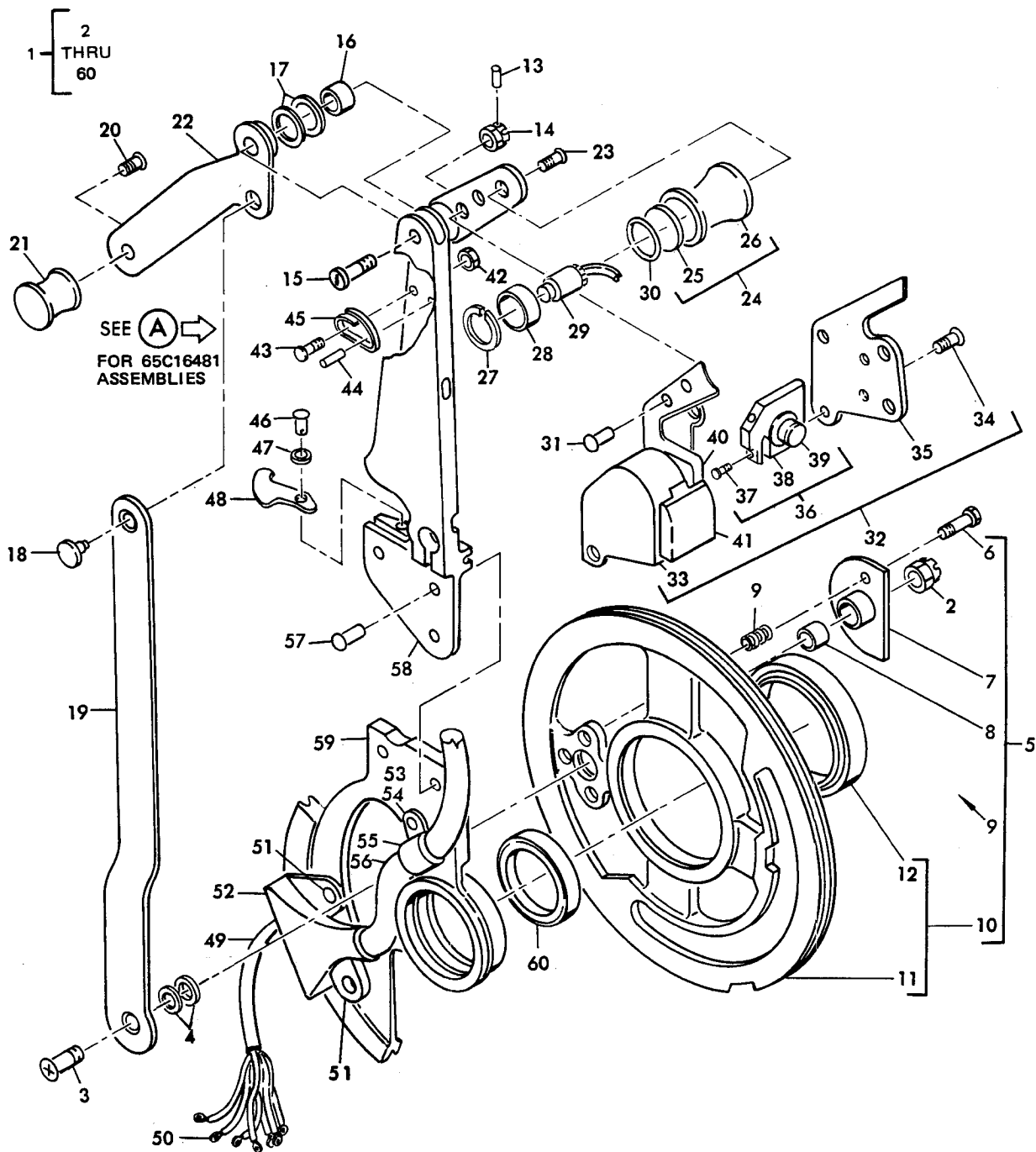
- A. Operate thrust lever and reverse thrust lever through full travel and verify mechanism, including pawl (48) and rocker arm (43B), operates freely without binding.
- B. Slowly move lever (22) from stowed position to clear rocker arm (43B). Force required to move the lever shall be 1.5-4.5 lbs measured at knob (21), 90 degrees to a line through the knob centerline and lever pivot point (bolt (15)).
- C. Slowly move lever (22) to stowed position. Force required to rotate rocker arm shall not exceed 4.5 lbs, measured as in step (2).

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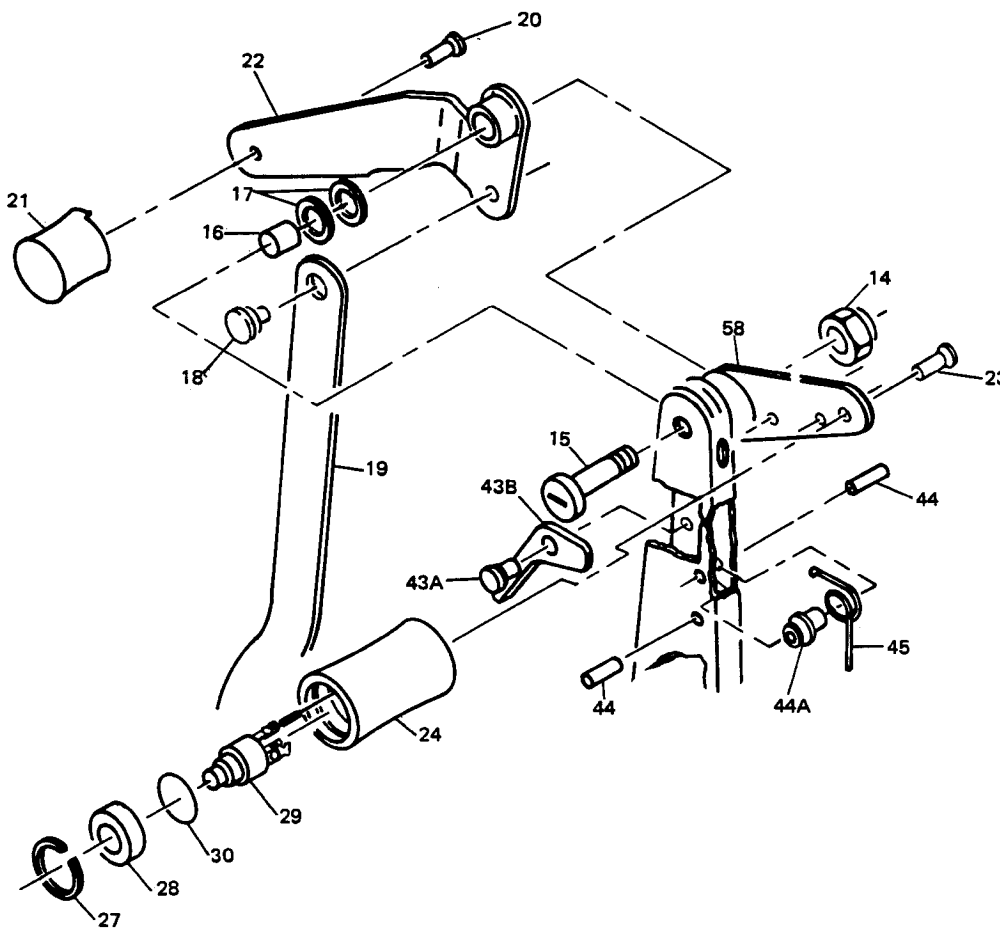
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ILLUSTRATED PARTS LIST



Handle Assemblies
Figure 1101 (Sheet 1)

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65C16481 ASSEMBLIES



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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-			<u>ENGINE CONTROL COMPONENTS</u>								
1	65-46506-1		HANDLE ASSY, LH (PRE SB 76-1006)							A	RF
1	65-46506-3		HANDLE ASSY, LH (PRE SB 76-1006)							B	RF
1	65-46506-5		HANDLE ASSY, LH (PRE SB 76-1006)							C	RF
1	65-46506-7		HANDLE ASSY, LH (PRE SB 76-1006)							D	RF
1	65-46506-9		HANDLE ASSY, LH (PRE SB 76-1006)							E	RF
1	65-46506-21		HANDLE ASSY, LH (PRE SB 76-1006)							F	RF
1	65-46506-29		HANDLE ASSY, LH (SB 22-1012, 76-1002) (PRE SB 76-1006)							G	RF
1	65-46506-35		HANDLE ASSY, LH (SB 76-1002) (PRE SB 76-1006)							H	RF
1	65-46506-37		HANDLE ASSY, LH (SB 76-1002) (PRE SB 76-1006)							I	RF
1	65-46506-39		HANDLE ASSY, LH (SB 76-1002) (PRE SB 76-1006)							J	RF
1	65-46506-41		HANDLE ASSY, LH (SB 22-1011, 22-1012, 76-1002) (PRE SB 76-1006)							K	RF
1	65-46506-43		HANDLE ASSY, LH (SB 22-1012, 76-1002, 76-1002) (PRE SB 76-1006)							L	RF
1	65-46506-45		HANDLE ASSY, LH (PRE SB 76-1006)							M	RF
1	65-46506-47		HANDLE ASSY, LH (PRE SB 76-1006)							N	RF
1	65C18257-7		DELETED								
1	65C18257-11		DELETED								
1	65C18257-13		DELETED								
1	65C18257-15		DELETED								
1	65C18257-17		DELETED								
1	65C16481-301		HANDLE ASSY, LH (PRE SB 76-1006)							T	RF
1	65C16481-303		HANDLE ASSY, LH (PRE SB 76-1006)							U	RF
1	65C16481-305		HANDLE ASSY, LH (PRE SB 76-1006)							V	RF
1	65C16481-307		HANDLE ASSY, LH (PRE SB 76-1006)							W	RF
1	65C16481-309		HANDLE ASSY, LH (PRE SB 76-1006)							X	RF
1	65C16481-311		HANDLE ASSY, LH (PRE SB 76-1006)							Y	RF
1	65C16481-313		HANDLE ASSY, LH (PRE SB 76-1006)							Z	RF
1	65C16481-315		HANDLE ASSY, LH (PRE SB 76-1006)							BA	RF
1	65C16481-317		HANDLE ASSY, LH (PRE SB 76-1006)							CA	RF
1	65C16481-319		HANDLE ASSY, LH (PRE SB 76-1006)							DA	RF
1	65C16481-321		HANDLE ASSY, LH (PRE SB 76-1006)							EA	RF
1	65C16481-323		HANDLE ASSY, LH (PRE SB 76-1006)							FA	RF
1	65C16481-325		HANDLE ASSY, LH (PRE SB 76-1006)							GA	RF
1	65C16481-327		HANDLE ASSY, LH (PRE SB 76-1006)							HA	RF
1	65C16481-331		HANDLE ASSY, LH (POST SB 76-1006)							IA	RF
1	65C16481-333		HANDLE ASSY, LH (POST SB 76-1006)							JA	RF
1	65C16481-335		HANDLE ASSY, LH (POST SB 76-1006)							KA	RF

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-											
1	65C16481-337		HANDLE ASSY, LH (POST SB 76-1006)							LA	RF
1	65C16481-339		HANDLE ASSY, LH (POST SB 76-1006)							MA	RF
1	65C16481-341		HANDLE ASSY, LH (POST SB 76-1006)							NA	RF
1	65C16481-343		HANDLE ASSY, LH (POST SB 76-1006)							OA	RF
1	65C16481-345		HANDLE ASSY, LH (POST SB 76-1006)							PA	RF
1	65C16481-347		HANDLE ASSY, LH (POST SB 76-1006)							QA	RF
1	65C16481-349		HANDLE ASSY, LH (POST SB 76-1006)							RA	RF
1	65C16481-351		HANDLE ASSY, LH (POST SB 76-1006)							SA	RF
1	65C16481-353		HANDLE ASSY, LH (POST SB 76-1006)							TA	RF
1	65C16481-355		HANDLE ASSY, LH (POST SB 76-1006)							UA	RF
1	65C16481-357		HANDLE ASSY, LH (POST SB 76-1006)							VA	RF
1	65-46506-2		HANDLE ASSY, RH (PRE SB 76-1006)							WA	RF
1	65-46506-4		HANDLE ASSY, RH (PRE SB 76-1006)							XA	RF
1	65-46506-6		HANDLE ASSY, RH (PRE SB 76-1006)							YA	RF
1	65-46506-8		HANDLE ASSY, RH (PRE SB 76-1006)							ZA	RF
1	65-46506-10		HANDLE ASSY, RH (PRE SB 76-1006)							CB	RF
1	65-46506-22		HANDLE ASSY, RH (PRE SB 76-1006)							DB	RF
1	65-46506-30		HANDLE ASSY, RH (SB 22-1012, 76-1002) (PRE SB 76-1006)							EB	RF
1	65-46506-36		HANDLE ASSY, RH (SB 76-1002) (PRE SB 76-1006)							FB	RF
1	65-46506-38		HANDLE ASSY, RH (SB 76-1002) (PRE SB 76-1006)							GB	RF
1	65-46506-40		HANDLE ASSY, RH (SB 76-1002) (PRE SB 76-1006)							HB	RF
1	65-46506-42		HANDLE ASSY, RH (SB 22-1011, 22-1012, 76-1002)(PRE SB 76-1006)							IB	RF
1	65-46506-44		HANDLE ASSY, RH (SB 22-1012, 76-1002) (PRE SB 76-1006)							JB	RF
1	65-46506-46		HANDLE ASSY, RH (PRE SB 76-1006)							KB	RF
1	65-46506-48		HANDLE ASSY, RH (PRE SB 76-1006)							LB	RF
1	65C18257-8		DELETED								
1	65C18257-12		DELETED								
1	65C18257-14		DELETED								
1	65C18257-16		DELETED								
1	65C18257-18		DELETED								
1	65C16481-302		HANDLE ASSY, RH (PRE SB 76-1006)							RB	RF
1	65C16481-304		HANDLE ASSY, RH (PRE SB 76-1006)							SB	RF
1	65C16481-306		HANDLE ASSY, RH (PRE SB 76-1006)							TB	RF
1	65C16481-308		HANDLE ASSY, RH (PRE SB 76-1006)							UB	RF
1	65C16481-310		HANDLE ASSY, RH (PRE SB 76-1006)							VB	RF
1	65C16481-312		HANDLE ASSY, RH (PRE SB 76-1006)							WB	RF
1	65C16481-314		HANDLE ASSY, RH (PRE SB 76-1006)							XB	RF
1	65C16481-316		HANDLE ASSY, RH (PRE SB 76-1006)							YB	RF

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-											
1	65C16481-318									ZB	RF
1	65C16481-320									DC	RF
1	65C16481-322									EC	RF
1	65C16481-324									FC	RF
1	65C16481-326									GC	RF
1	65C16481-328									HC	RF
1	65C16481-330									IC	RF
1	65C16481-332									JC	RF
1	65C16481-334									KC	RF
1	65C16481-336									LC	RF
1	65C16481-338									MC	RF
1	65C16481-340									NC	RF
1	65C16481-342									OC	RF
1	65C16481-344									PC	RF
1	65C16481-346									QC	RF
1	65C16481-348									RC	RF
1	65C16481-350									SC	RF
1	65C16481-352									TC	RF
1	65C16481-354									UC	RF
1	65C16481-356									VC	RF
1	65C18257-67										
1	65C18257-68										
1	65C18257-71										
1	65C18257-72										
1	65c18257-73										
1	65C18257-74										
1	65C18257-75										
1	65C18257-76										
1	65C18257-77										
1	65B18257-78										
2	BACN10JD105										
3	63-1669										
3	66-26135-1									NT-VA RB-VC	1
4	AN960PD516										AR
4	AN960PD516L										AR
5	65-73668-1										1
5	65-73668-2										1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY		
			1	2	3	4	5	6	7				
1101-5	65-73668-7		.	D	R	U	M	A	S	S		1	
												1	
5	65-73668-8		.	D	R	U	M	A	S	S		1	
												1	
5	65-73668-13		.	D	R	U	M	A	S	S		1	
												1	
5	65-73668-14		.	D	R	U	M	A	S	S		1	
												1	
5	65-33892-1		.	D	R	U	M	A	S	S		1	
												1	
5	65-33892-2		.	D	R	U	M	A	S	S		1	
												1	
5	65-88646-4		.	D	R	U	M	A	S	S	N HA	1	
											VC		
5	65-88646-5		.	D	R	U	M	A	S	S	VA LB	1	
											HC		
5	65-88646-4												
5	65-88646-5												
5	65-88646-9		.	D	R	U	M	A	S	S	ED GD		
5	65-88646-10		.	D	R	U	M	A	S	S	FD HD	1	
5	65-88646-14											1	
5	65-88646-17		.	D	R	U	M	A	S	S	ID	1	
5	65-88646-18		.	D	R	U	M	A	S	S	JD	1	
5	65-88646-22												
5	65-73668-13		.	D	R	U	M	A	S	S	ED GD	1	
5	65-73668-14		.	D	R	U	M	A	S	S	FD HD	1	
6	BACB30LU3-1		.	.	B	O	L	T			*[1]	3	
6	BACB30LU3-2		.	.	B	O	L	T			*[2]	3	
6	NAS514P1032-6		.	.	S	C	R	E	W		*[14]	3	
7	66-1523-1		.	.	R	E	T	A	I	N	E	R	1
8	BACB10A683		.	.	B	E	A	R	I	N	G		1
												1	
9	MS21209F1-15		.	.	I	N	S	E	R	T			3
												3	
10	65-33892-3		.	.	D	R	U	M	A	S	ABE	1	
10	65-33892-4		.	.	D	R	U	M	A	S		1	
												1	

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY	
			1	2	3	4	5	6	7			
1101-10	65-73668-3		.	.	DRUM	ASSY	(USED	ON	65-46506-5,-7,-21;	65C16481-305,-307,-311,-334,-336,-340)		1
10	65-73668-4		.	.	DRUM	ASSY	(USED	ON	65-46506-2,-4,-10;	65C16481-302,-304,-310,-331,-333,-339)		1
10	65-73668-9		.	.	DRUM	ASSY	(USED	ON	65-46506-35,-37,-39;	65C16481-315,-317,-319,-344,-346,-348)		1
10	65-73668-10		.	.	DRUM	ASSY	(USED	ON	65-46506-36,-38,-40;	65C16481-316,-318,-320,-345,-347,-349)		1
10	65-73668-15		.	.	DRUM	ASSY	(USED	ON	65-46506-29,-41,-43,-45;	65C16481-313,-321,-323,-325,-342,-350,-352,-354)		1
10	65-73668-16		.	.	DRUM	ASSY	(USED	ON	65-46506-30,-42,-44,-46;	65C16481-314,-322,-324,-326,-343,-351,-353,-355)		1
10	65-88646-6		.	.	DRUM						NS VC	1
10	65-88646-7		.	.	DRUM						VA LB HC	1
10	65-88646-11		.	.	DRUM	ASSY					ED GD	1
10	65-88646-12		.	.	DRUM	ASSY					FD HD	1
10	65-88646-16				DELETED							
10	65-88646-19				DELETED							
10	65-88646-20				DELETED							
10	65-88646-24				DELETED							
11	65-33892-5		.	.	.	DRUM	(USED	ON	65-46506-1,-3,-9;	65C16481-301,-303,-309,-330,-332,-338)		
11	65-33892-6		.	.	.	DRUM	(USED	ON	65-46506-2,-4,-10;	65C16481-302,-304,-310,-331,-333,-339)		
11	65-73668-5		.	.	.	DRUM	(USED	ON	65-46506-5,-7,-21;	65C16481-305,-307,-311,-334,-336,-340)		1
11	65-73668-6		.	.	.	DRUM	(USED	ON	65-46506-6,-8,-22;	65C16481-306,-308,-312,-335,-337,-341)		1
11	65-73668-11		.	.	.	DRUM	(USED	ON	65-46506-35,-37,-39;	65C16481-315,-317,-319,-344,-346,-348)		1
11	65-73668-12		.	.	.	DRUM	(USED	ON	65-46506-36,-38,-40;	65C16481-316,-318,-320,-345,-347,-349)		1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY	
			1	2	3	4	5	6	7			
1101-11	65-73668-17		.	.	.	DRUM (USED ON 65-46506-29,-41,-43,-45; 65C16481-313,-321,-323,-325,-342,-350,-352,-354)						1
11	65-73668-18		.	.	.	DRUM (USED ON 65-46506-30,-42,-44,-46; 65C16481-314,-322,-324,-326,-343,-351,-353,-355)						1
12	BACB10A825		.	.	.	BEARING (REPLS BACB10BW37, MS20202KP37B, AN202KP37B)					*[13]	1
12A	BACB10BW37		.	.	BEARING						N HA VA LB HC	1
12A	BACB10AU37		.	.	BEARING (OPT)						N HA VA LB HC	1
13	NAS561C2-5		.	PIN, SPRING (PRE SB 76-1006)							*[11]	1
14	BACN10JD103AU		.	NUT (REPLS AN320C3 (PRE SB 76-1006)							*[11]	1
14	BACN10JC3		.	NUT (POST SB 76-1006)							*[15]	1
15	63-1409-1		.	BOLT, SPECIAL (PRE SB 76-1006)							*[11]	1
15	BACB30FM6A7U		.	BOLT (POST SB 76-1006)							*[15]	1
16	NAS75-3-012		.	BUSHING (PRE SB 76-1006)							*[11]	1
16	BACB28Y3M040		.	BUSHING (OPT)							*[11]	1
16	BACB28Y3M040		.	BUSHING (POST SB 76-1006)							*[15]	1
17	BACW10P24C		.	WASHER (PRE SB 76-1006)								AR
17	AN960C516L		.	WASHER (PRE SB 76-1006)								AR
17	69-69916-1		.	WASHER (4 MAX) (POST SB 76-1006)							*[15]	AR
17	69-69916-2		.	WASHER (POST SB 76-1006)							*[15]	1
18	66-11520		.	RIVET								1
19	69-1572-1		.	LINK, REVERSE THRUST							A-N T-HA IC-VC	1
19	69-1572-2		.	LINK, REVERSE THRUST							IA-LB RB-VC	1
20	BAS12BP3-8		.	SCREW								1
21	65C14183-15		.	KNOB, CONTROL LEVER							A-N T-HA IC-VC	1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY														
			1	2	3	4	5	6	7																
1101-21	65-2343-15		.	K	N	O	B	,	C	O	N	T	R	O	L	E	R	(O	P	T)	A-N	1	
21	65C14183-16		.	K	N	O	B	,	C	O	N	T	R	O	L	E	R						IA-LB	1	
21	65-2343-16		.	K	N	O	B	,	C	O	N	T	R	O	L	E	R	(O	P	T)	IA-LB	1	
22	69-1819-17		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	A-F	1
22	69-1819-18		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	WA-DB	1
22	69-1819-21		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	G-N	1
22	69-1819-22		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	EB-LB	1
22	69-1819-27		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	T-HA	1
22	69-1819-28		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	IC-VC	1
22	69-1819-37		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	DB-LB	1
22	69-1819-38		.	L	E	V	E	R	,	R	E	V	E	R	S	E	T	R	H	R	O	S	T	IA-VA	1
22A	BACS12BP04A4																								
22A	BACS12BP04C																								
22B	65C18253-1																								
22B	65C18253-2																								
22C	65C18257-19																								
22C	65C18257-22																								
22C	65C18257-20																								
22C	65C18257-46																								
22C	65C18257-47																								
22D	66-25944-2																								
22D	69-70862-1																								
22E	65C18255-5																								
22E	65C18255-6																								
22E	65C19962-1																								
22E	65C19962-2																								
22F	66-25946-1																								
22F	69-70866-1																								
22G	66-25939-1																								
22H	69-69983-1																								
22H	69-69983-2																								
22I	65C-18254-5																								
22I	65C-18254-6																								
22I	69-70867-1																								
22I	69-70867-1																								
22I	69-70867-2																								

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-											
22J	W0731-009		DELETED								
22K	69-69983-1		DELETED								
22K	69-69983-2		DELETED								
22L	BACS12BP04A4		DELETED								
22L	BACS12BP04C4		DELETED								
22M	69-69984-8		DELETED								
22M	69-69984-9		DELETED								
22M	69-69984-11		DELETED								
22M	69-69984-12		DELETED								
22N	69-69984-3		DELETED								
22N	69-69984-4		DELETED								
22N	69-69984-15		DELETED								
22N	69-69984-16		DELETED								
22O	69-69984-7		DELETED								
22O	69-69984-17		DELETED								
22P	MS21209C0410		DELETED								
22Q	66-25945-1		DELETED								
22R	65C18257-5		DELETED								
22R	65C18257-6		DELETED								
22R	65C18257-48		DELETED								
22R	65C18257-49		DELETED								
22S	69-69981-1		DELETED								
22S	69-69981-2		DELETED								
22S	69-69981-5		DELETED								
22S	69-69981-6		DELETED								
22T	66-25941-1		DELETED								
22U	69-69985-1		DELETED								
22U	69-70861-1		DELETED								
22V	66-25940-1		DELETED								
22W	61-32157-2		DELETED								
22W	61-32157-3		DELETED								
22X	66-25942-1		DELETED								
23	BACS12BP3-5		DELETED								
24	65-45117-1		. KNOB ASSY (USED ON 65-46506-1,-5, -35,-41; 65C18257-11,-71; 65C16481-301,-305,-315,-321,-330, -334,-344,-350)								1
24	65-45117-2		. KNOB ASSY (USED ON 65-46506-2,-6, -36,-42; 65C18257-12,-72; 65C16481-302,-306,-316,-322,-331, -335,-345,-351)								1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-24	65-45117-8		.							*[3]	1
24	65-45117-9		.							*[4]	1
25	69-40892-1		.	.							1
26	65-45117-3		.	.							1
27	MS16625-4086		.								1
28	69-35353-1		.							*[5]	1
28	69-35353-2		.							*[6]	1
29	G2006		.							*[7]	1
30	AN6227-12		.								1
31	BACR15BA4D		.								3
32	65-45115-1		.							A-D HIK LT-W BA	1
32	65-45115-2		.							*[8]	1
32	65-45115-6		.								1
32	65-45115-7		.								1
32	65-45115-12		.							M GA	1
32	65-45115-13		.							UA KB GC	1
32	65-45115-8		.								1
32	65-45115-9		.								1
32	65-45115-10		.							GJZ	1
32	65-45115-11		.							OA EB HB XB	1
32	65-45115-14										
32	65-45115-15										
32	65-45115-16										
32	65-45115-17										
33	65-51320-5		.	.							
33	65-51320-6		.	.							1
34	BAC12BP04C4		.	.							6
35	69-38721-1		.	.							1
35	69-38721-2		.	.							1

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY													
			1	2	3	4	5	6	7															
1101-35	69-38721-3		.	.	C	O	V	E	R	(USED ON 65-45115-8,-10,-12)		1												
35	69-38721-4		.	.	C	O	V	E	R	(USED ON 65-45115-9,-11,-13)		1												
35	69-38721-5		DELETED																					
35	69-38721-6		DELETED																					
36	65-45115-3		.	.	M	O	U	N	T	ASSY		1												
37	NAS514P440-5P		.	.	.	S	C	R	E	W		1												
38	69-38734-1		.	.	.	R	E	T	A	I	N	E	1											
39	C2006		.	.	.	S	W	I	T	C	H	, V81640	1											
40	MS20392-1C19		.	.	P	I	N					1												
41	69-38720-1		.	.	P	A	D	D	L	E	(USED ON 65-45115-1,-2,-8,-9)	1												
41	65-38720-2		.	.	P	A	D	D	L	E	(USED ON 65-45115-12,-13)	1												
41	65-38720-3		.	.	P	A	D	D	L	E	(USED ON 65-45115-6,-7,-10,-11)	1												
42	22LH1660-40		.	.	N	U	T	,	V	72962 (PRE SB 76-1006)	*[9]	1												
42	92-1660-40		.	.	N	U	T	,	V	72962 (OPT) (PRE SB 76-1006)	*[9]	1												
42	BAC10DN40		.	.	N	U	T	(OPT)	(USED ON 65-46506-1 THRU -10,-21,-22) (PRE SB 76-1006)			1												
43	66-1624-1		.	.	S	T	O	P	B	O	L	T, REVERSE THRUST (PRE SB 76-1006)	A-F WA-DB	1										
43	66-1624-2		.	.	S	T	O	P	B	O	L	T, REVERSE THRUST (PRE SB 76-1006)	G-M EB-LB	1										
43A	69-69917-1		.	.	P	I	N				*[15]	1												
43B	65C16476-1		.	.	A	R	M	,	R	O	C	K	*[15]	1										
44	MS39086-130		.	.	P	I	N				*[11]	1												
44	MS39086-130		.	.	P	I	N				*[15]	2												
44A	69-69918-1		.	.	S	P	A	C	E	R		*[15]	1											
45	69-69919-1		.	.	S	P	R	I	N	G (PRE SB 76-1006)	*[12]	1												
45	65-74599-1		.	.	S	P	R	I	N	G (SB 76-1002) (PRE SB 76-1006)	*[9]	1												
45	69-35369-1		.	.	S	P	R	I	N	G (OPT) (PRE SB 76-1006)	A-D WA-ZA	1												
45	69-35369-1		.	.	S	P	R	I	N	G (USED ON 65-46506-1 THRU -10,-21,-22) (PRE SB 76-1006)		1												
45	69-71314-3		.	.	S	P	R	I	N	G (POST SB 76-1006)	IA-VA IC-VC	1												
46	63-1440		.	.	R	I	V	E	T			1												
47	5804-8-2		.	.	W	A	S	H	E	R, V86928		1												
48	69-1066-2		.	.	P	A	W	L	,	T	H	R	U	S	T	L	E	V	E	R	L	O	C	1
49	61-30049-001		.	.	B	U	N	D	L	E	A	S	S	, W	I	R	E		A-N T-HA	1				
49	61-30051-001		.	.	B	U	N	D	L	E	A	S	S	, W	I	R	E		IA-LB RB-HC	1				

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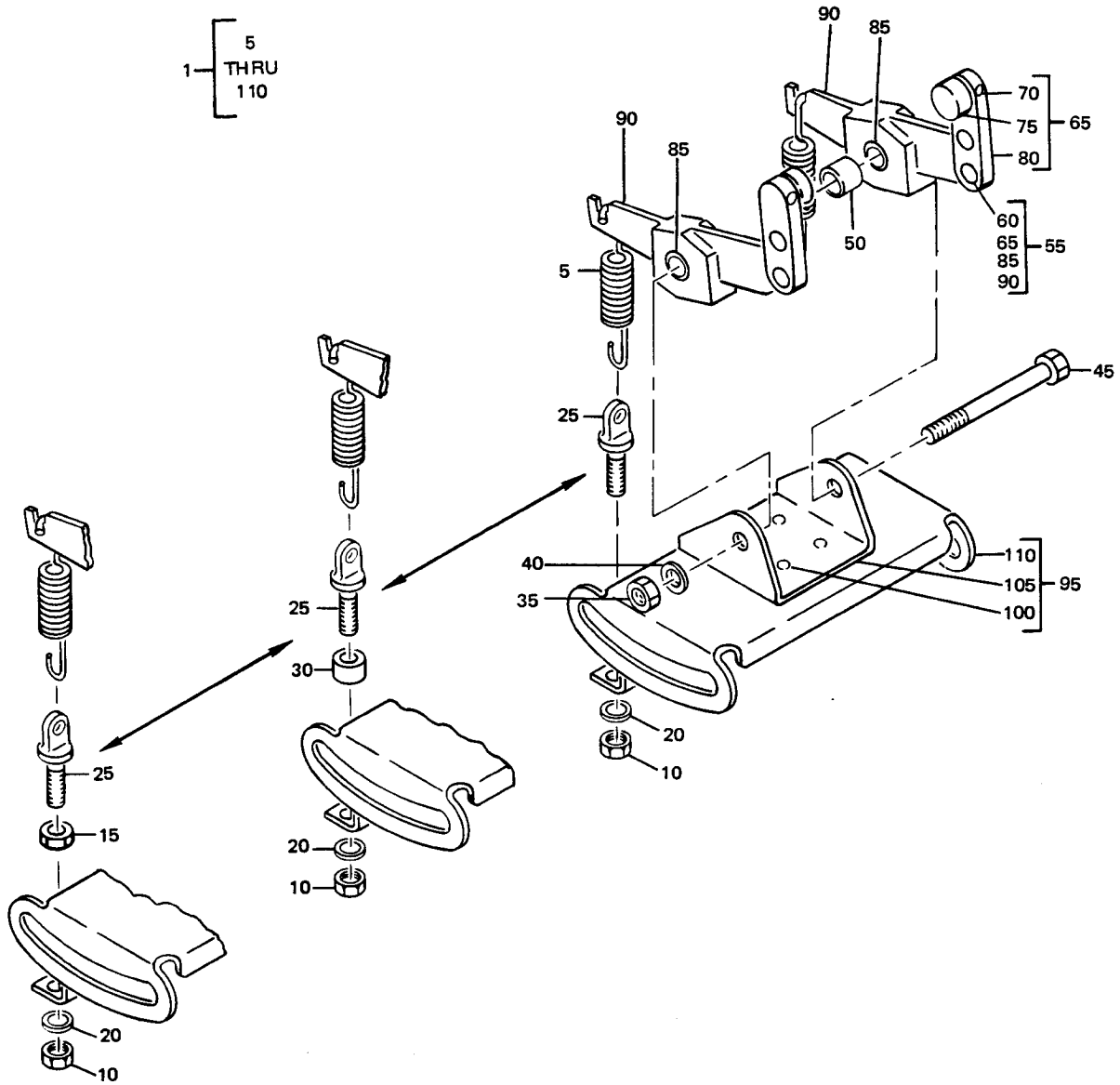
OVERHAUL MANUAL

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-50	BACT12AC43		.	.							AR
51	BACR15BA5B		.								2
52	65-58259-1		.							A-M	1
52	65-58259-2		.							T-GA	1
52	69-67277-1		.							IA-UA	1
52	69-68123-1		.							WA-KB	1
53	BACN10JC3		.							RB-GC	1
54	BACB30LU3-1		.							N HA	1
55	BACC10DK6		.							VA	1
55	AN742D6		.							LB HC	1
55	AN742D6		DELETED								1
56	69-45733-1		.								1
57	BACR15BA6D		.								3
58	69-40474-5		.							A-N	1
58	69-40474-6		.							T-HA	1
58	69-40474-9		.							IA-VA	1
58	69-40474-10		.							WA-LB	1
58	65C18256-4		DELETED							RB-HC	1
58	65C18256-5		DELETED								
58	65C18256-24		DELETED								
58	65C18256-25		DELETED								
58A	66-25938-1		DELETED								
58A	66-25938-2		DELETED								
58B	65C18252-1		DELETED								
58B	65C18252-2		DELETED								
58B	65C18252-7		DELETED								
58B	65C18252-8		DELETED								
58C	65C18252-3		DELETED								
58C	65C18252-4		DELETED								
58C	65C18252-9		DELETED								
58C	65C18252-10		DELETED								
58D	MS21209C0410		DELETED								
58D	MS21209C0410		DELETED								

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1101-											
58E	MS21209C0415										
58F	MS240273M										
58G	MS206153M										
58H	65C1825-13										
58H	65C18256-13										
58H	65C18256-9										
58H	65C18256-14										
58H	65C18256-10										
58I	66-25943-2										
59	65-1634-5								A-NT-HA	1	
59	65-1634-1								A-M T-GA	1	
59	65-1634-6								IA-LB RB-HC	1	
59	65-1634-2								IA-UA WA-KB RB-HC		
59	65-1634-11										
59	65-1634-12										
60	BACB10CF25PP										
60	BACB10AS25										
60	BACB10A300DD								A-N T-HA	1	
60	BACB10BW37										
60	BACB10AU37										

- *[1] ABEN JA MA XA CB IC JC MC VC
- *[2] CDF-L KA LA NA-UA YA ZA DB-KB KC LC NC-TC
- *[3] BD-GIJL-NUW-Z CA DA FA GA HA
- *[4] JA LA-OA QA RA TA UA VA XA ZA CB DB EB GB HB JB KB LB SB UB-XB ZB DC FC GC HC
- *[5] ACHKPQTV BA-GA IA-UA WA-KB NB RB-GC YC ZC
- *[6] BD-GIJL-NUW-Z CA DA FA GA HA JA-OA QA RA TA UA VA XA-EB GB HB JB-LB SB UB-XB ZB DC FC GC HC
- *[7] ACHKTV BA EA IA KA PA SA RB TB YB EC
- *[8] IA-LA PA QA SA TA WA-ZA FB GB IB JB RB-UB YB
- *[9] A-D G-N WA-ZA EB-LB
- *[10] A-N T-LB RB-HC
- *[11] A-N WA-KB
- *[12] T-HA RB-HC
- *[13] A-M T-Z BA-GA IA-UA WA-LB RB-GC
- *[14] VA-CB RB-ZB DC LB HC
- *[15] T-HA RB-HC IA-VA IC-VC



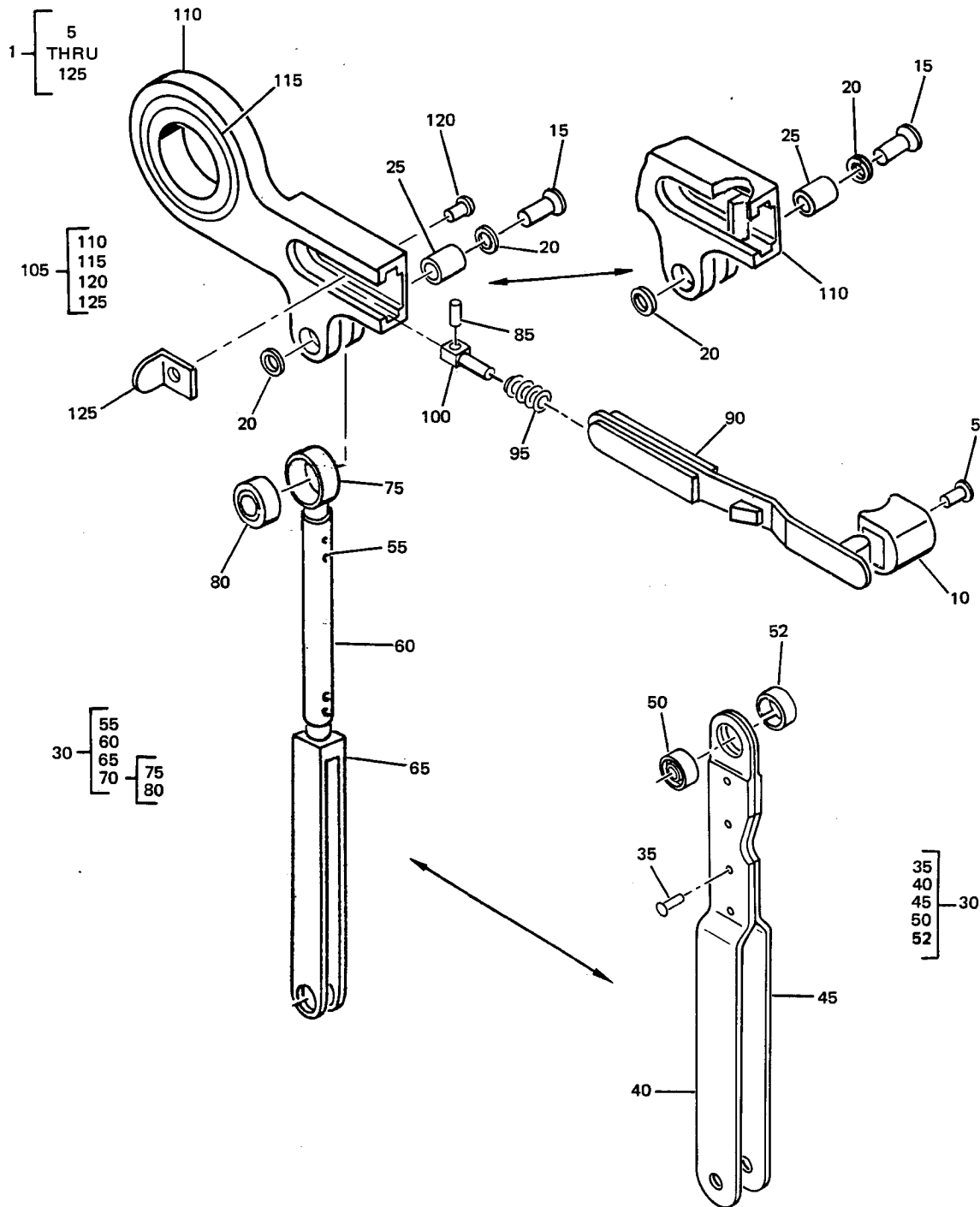
Control Stand Roller Assembly
Figure 1102

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FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1102-											
1	69-35354-3									A	RF
1	69-35354-4									B	RF
1	69-35354-5									C	RF
1	69-35354-6									D	RF
5	69-34830-1									A	2
5	69-55932-1									BCD	2
10	BACN10JC3										2
15	MS25082-3									D	2
20	AN960PD10L										AR
25	AN42B3A									AB	2
25	66-1551									AB	2
25	AN42B5A									C	2
25	66-1551-1									D	1
30	NAS43DD3-11									B	2
30	NAS43DD3-18									C	2
35	BACN10JC4										2
40	AN960PD10L										AR
45	NAS1104-30										1
50	NAS43DD4-32										1
55	69-17736-8										1
55	69-17736-9										1
60	MS20426D5										2
65	66-13328-1										1
70	NAS508M2-9										1
75	BACB10B209CF3										1
80	66-13328-2										1
85	BACB10A127										1
90	69-17736-7										1
95	69-35354-2										1
100	BACR15BA6D										AR
105	69-45173-1										1
110	69-455173-2										1



Start System Lever Assembly
Figure 1103

65-23761 69-35354
 65-45136 65C16481
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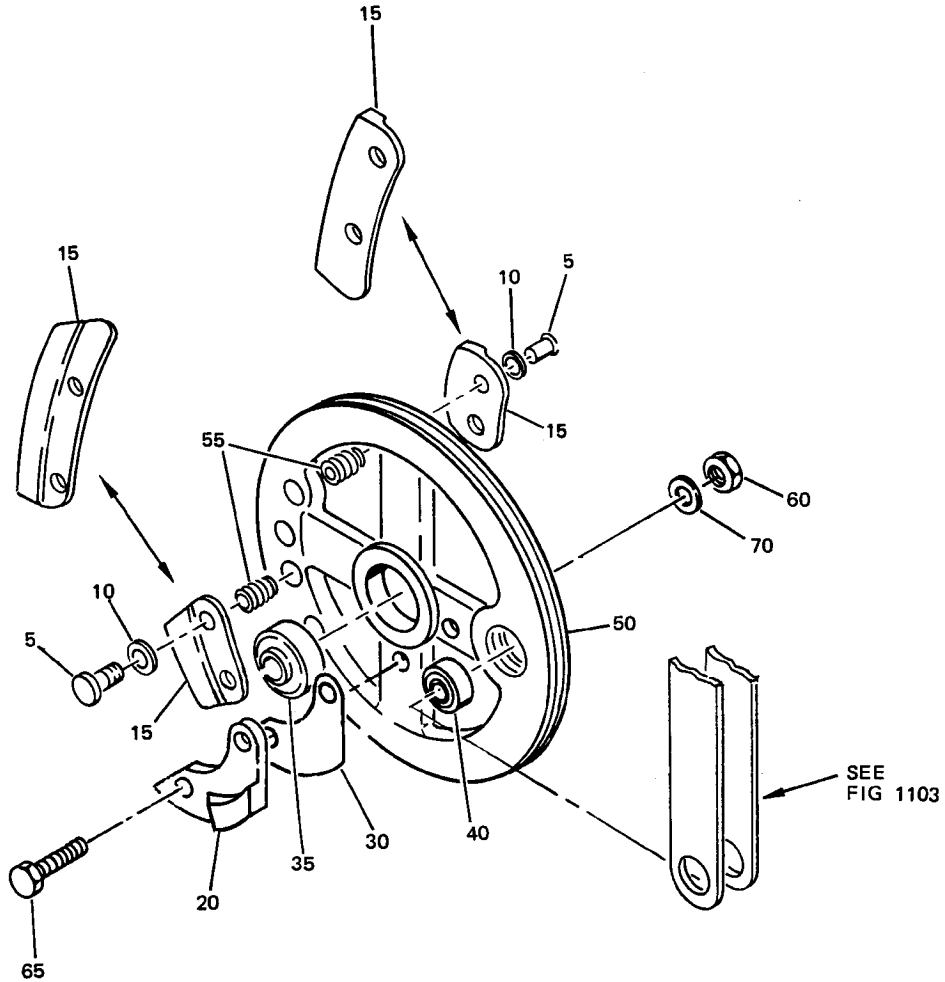


OVERHAUL MANUAL

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	N O M E N C L A T U R E							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1103-											
1	65-23761-4									A	RF
1	65-23761-5									B	RF
1	65-23761-15									C	RF
1	65-23761-16									D	RF
1	65-23761-17									E	RF
1	65-23761-18									F	RF
1	65-23761-19									G	RF
1	65-23761-20									H	RF
5	BACS12BP3-8										1
10	65C14183-30									ACEG	1
10	66-14506-4									ACEG	1
10	65C14183-31									BDFH	1
10	66-14506-5									BDFH	1
15	MS20615-3M										1
20	AN960-2L									AB	2
20	AN960-3L									C-H	2
25	NAS42HT3-30										1
30	69-1692									AB	1
30	69-66099-1									C-H	1
30	69-66099-4									C-H	1
35	MS20615-5									C-H	1
40	69-66099-2									C-H	1
45	69-66099-3									C-H	1
50	BACB10AC3L									C-H	1
52	66-26136-2									C-H	1
55	BACR15BB4D									AB	AR
60	69-1692-1									AB	1
65	66-1643									AB	1
70	66-1641									AB	1
75	66-1641-1										1
80	BACB10AC3L										1
85	NAS561C3-10										1
90	69-17300-1									BDF	1
90	69-17300-2									ACE	1
90	69-17300-5									DFH	1
90	69-17300-6									CEG	1
95	63-1408										1
100	63-1327-1										1
105	65-1659-8									AC	1
105	65C18080-1									C	1
105	65-1659-9									BD	1
105	65C18080-2									D	1
105	65C18080-7									EG	1

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THRU
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Engine Start Drum Assembly
Figure 1104

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1104-1	65-45136-1		DRUM ASSY, ENGINE START (PRE SB 28-1020)							A	RF
1	65-45136-4		DRUM ASSY, ENGINE START (PRE SB 28-1020)							B	RF
1	65-45136-7		DRUM ASSY, ENGINE START (PRE SB 28-1035)							C	RF
1	65-45136-8		DRUM ASSY, ENGINE START (PRE SB 28-1035)							D	RF
1	65-45136-9		DRUM ASSY, ENGINE START (PRE SB 28-1035)							E	RF
1	65-45136-10		DRUM ASSY, ENGINE START (PRE SB 28-1035)							F	RF
1	65-45136-11		DRUM ASSY, ENGINE START (PRE SB 28-1035)							G	RF
1	65-45136-12		DRUM ASSY, ENGINE START (PRE SB 28-1035)							H	RF
1	65-45136-13		DRUM ASSY, ENGINE START (PRE SB 28-1035)							I	RF
1	65-45136-14		DRUM ASSY, ENGINE START (PRE SB 28-1035)							J	RF
1	65-45136-15		DRUM ASSY, ENGINE START							K	RF
1	65-45136-17		DRUM ASSY, ENGINE START							L	RF
1	65-45136-18		DRUM ASSY, ENGINE START							M	RF
1	65-45136-19		DRUM ASSY, ENGINE START							N	RF
1	65-45136-20		DRUM ASSY, ENGINE START							O	RF
5	NAS623-3-3		. BOLT								4
10	AN960PD10L		. WASHER							A-L	4
10	NAS1149D0316J		. WASHER							MNO	4
15	66-23291-1		. CAM							ACDIJ	2
15	69-40908-1		. CAM							BE-H	2
15	69-40908-2		. CAM							KL	2
15	69-40908-3		. CAM							MN	2
20	69-63348-1		. CAM							CE	1
20	69-72045-1		. CAM							Gi	1
-25	69-63348-2		. CAM							DF	1
-25	69-72045-2		. CAM							HJ	1
30	69-72046-1		. PLATE							G-J	1
35	BACB10A544		. BEARING (REPLS MS20201KP8A)							A-L	1
35	BACB10BX8		. BEARING (REPLS BACB10A544, MS20201KP8A)							A-L	1
35	BACB10BX8		. BEARING							MNO	1

FIG. & ITEM NO.	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE							USE CODE	QTY PER ASSY
			1	2	3	4	5	6	7		
1104-40	BACB10A682		.							A-L	1
40	BACB10BX3		.							A-L	1
40	BACB10BX3		.							MNO	1
45	65-45136-2		.							ABK	1
45	65-45136-2		.							NO	1
45	65-45136-6		.							ABK	1
45	65-45136-6		.							C-J	1
45	65-45136-16		.							LM	1
50	65-45136-3		.	.							1
50	65-46136-5		.	.							1
55	MS21209F1-15		.	.							4
60	BACN10JC3		.							C-F	2
65	NAS623-3-6		.							C-F	2
65	NAS623-3-8		.							G-J	2
70	AN960PD10L		.							C-F	2

VENDORS

V72962 HARVARD INDUSTRIES, INC., 3 WERNER WAY, SUITE 210, LEBANON, NEW JERSEY 08833

V81640 EATON AEROSPACE LLC, 2250 WHITFIELD AVE., SARASOTA, FLORIDA 34243-3962

V86928 SEASTROM MANUFACTURING CO., INC., 456 SEASTROM ST., TWIN FALLS, IDAHO 83301-8526