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#### GPA Group plc

PAGE DATE	CODE	PAGE	DATE	CODE	PAGE	DATE	CODE
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STARTING							
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R = REVISED, A = ADDED OR D = DELETED 632 JAN 20/09 D633N632

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#### CHAPTER 80 - STARTING

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Low Max Motoring (Fig. 104) No N3 Rotation (Fig. 103) Start Switch Failed to Hold in GND (Fig. 106)		104 103 106	
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Start Switch Returned to AUTO

Starter VALVE Light On (Fig.

(Fig. 107)

105)

## 80-CONTENTS

RGUI

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FAULT CODE	LOG BOOK REPORT	FAULT ISOLATION REFERENCE
80 11 XA	(01=L,02=R) Starting problem was encountered by the flight crew which is not covered by the fault code diagrams.	
80 11 XB	(01=L,02=R) Starting problem was encountered by the flight crew which is not covered by the fault code diagrams.	SSM 80-00-00, SSM 80-11-01, SSM 80-11-02
80 11 XC	(01=L,02=R) Starting problem was encountered by the flight crew which is not covered by the fault code diagrams.	SSM 80-00-00, SSM 80-11-01, SSM 80-11-02
80 11 01	(01=L,02=R) Engine N3 failed to rotate with start selector in GND, start VALVE light was off and duct pressure was normal.	FIM 80-11-00/101, Fig. 103, Block 1
80 11 02	(01=L,02=R) Engine max N3 motoring speed low during start, duct pressure was normal with start valve open.	FIM 80-11-00/101, Fig. 104, Block 1
80 11 03	(01=L,02=R) Engine start VALVE light (failed to, remained) on and EICAS msg: (L,R) ENG STARTER (did not, did) displ with start selector in GND, eng failed to rotate.	FIM 80-11-00/101, Fig. 105, Block 1
80 11 04	(01=L,02=R) Engine start selector sw failed to hold in GND during start, eng rotated normal when held in GND.	FIM 80-11-00/101, Fig. 106, Block 1
80 11 05	(01=L,02=R) Engine start selector sw failed to hold in GND during start, eng failed to rotate when held in GND.	FIM 80-11-00/101, Fig. 106, Block 1
80 11 08	(01=L,02=R) Engine start VALVE light remained on and EICAS msg (L/R) STARTER CUTOUT displayed during start when start selector returned to AUTO.	FIM 80-11-00/101, Fig. 107, Block 1

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FAULT CODE	LOG BOOK REPORT	FAULT ISOLATION REFERENCE
80 11 11	(01=L,02=R) Engine start selector returned to AUTO before 47% N3,% during start.	FIM 80-11-00/101, Fig. 106, Block 1
80 11 12	(01=L,02=R) Engine start selector failed to return to AUTO at 47% N3 during start. Manually selected to AUTO.	FIM 80-11-00/101, Fig. 108, Block 1
80 11 13	(01=L,02=R) Engine had impending hot start. Starter cutout early at% N3.	FIM 80-11-00/101, Fig. 106, Block 1
80 11 15	(01=L,02=R) Engine start VALVE light (failed to, remained) on and EICAS msg (L,R) ENG STARTER (did not, did) display with start selector in GND. Engine rotation was normal.	FIM 80-11-00/101, Fig. 105, Block 22

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RB.211 ENGINES

#### **ENGINE STARTING SYSTEM**

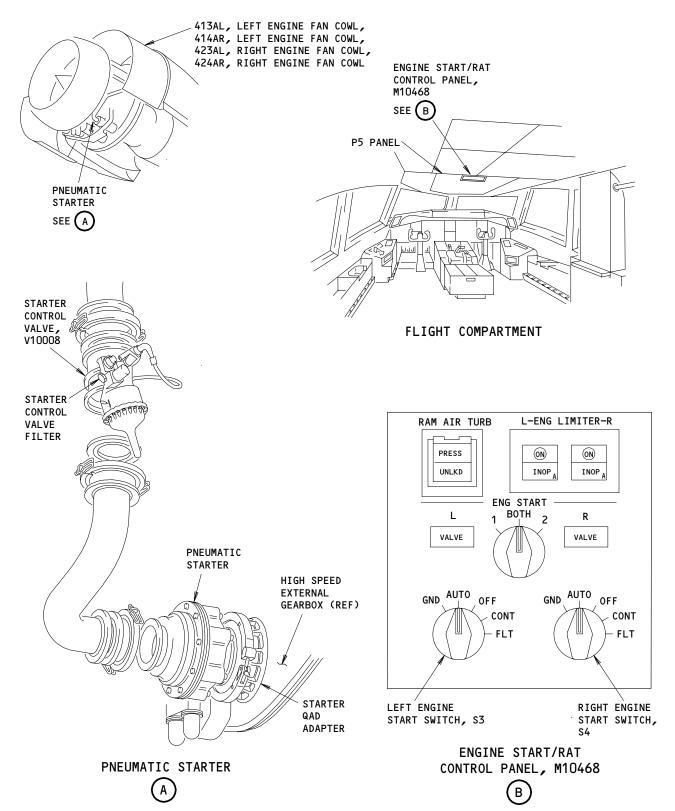
	_			
COMPONENT	FIG. 102 SHT	QTY	ACCESS/AREA	REFERENCE
ADAPTER - STARTER QAD		2	413AL,414AR, LEFT ENGINE FAN COWL, 423AL,424AR, RIGHT ENGINE FAN COWL	80-11-07
FILTER - STARTER CONTROL VALVE		2	413AL,414AR, LEFT ENGINE FAN COWL, 423AL,424AR, RIGHT ENGINE FAN COWL	80-11-08
PANEL - ENG START/RAT CONT, M10468 RELAYS - (31-01-36/101) LEFT ENGINE CONTROLS POWER, K10208 LEFT ENGINE START DISAGREE, K10209 LEFT ENGINE START 2, K10247 LEFT ENGINE START 3, K10248 LEFT ENGINE START TO STOP, K10212 RELAYS - (31-01-37/101) RIGHT ENGINE CONTROLS POWER, K10220 RIGHT ENGINE START DISAGREE, K10221 RIGHT ENGINE START 2, K10250 RIGHT ENGINE START 3, K10251 RIGHT ENGINE START TO STOP, K10224		1	FLIGHT COMPARTMENT, P5	*
STARTER - PNEUMATIC		2	413AL,414AR, LEFT ENGINE FAN COWL, 423AL,424AR, RIGHT ENGINE FAN COWL	80-11-01
SWITCH - LEFT ENGINE START, S3		1	FLIGHT COMPARTMENT, P5, ENGINE START/RAT CONTROL PANEL, M10468	*
SWITCH - RIGHT ENGINE START, S4		1	FLIGHT COMPARTMENT, P5, ENGINE START/RAT CONTROL PANEL, M10468	*
VALVE - STARTER CONTROL, V10008		2	413AL,414AR, LEFT ENGINE FAN COWL, 423AL,424AR, RIGHT ENGINE FAN COWL	80-11-08

<sup>\*</sup> SEE THE WDM EQUIPMENT LIST

Engine Starting System - Component Index Figure 101

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Engine Starting System - Component Location Figure 102

ALL

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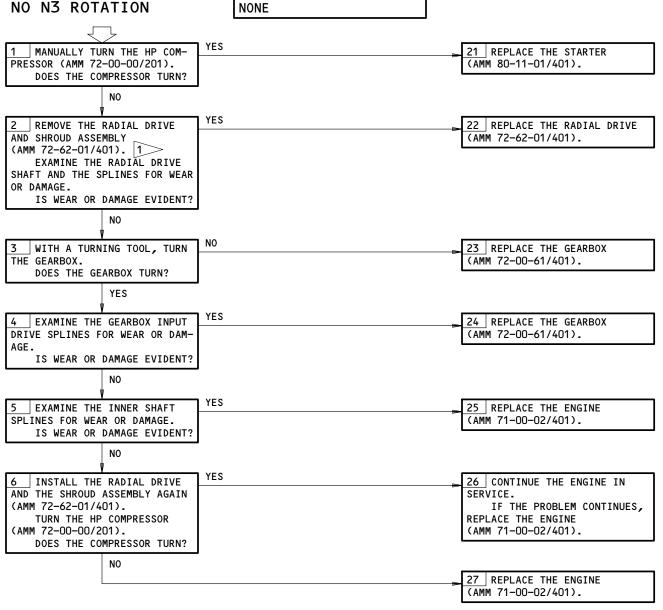
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R02

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# PREREQUISITES NONE



IF A STARTER ASSISTED RELIGHT OR A GROUND START OF THE ENGINE RESULTS IN A FAILED RADIAL DRIVE, THERE CAN BE OIL LEAKS INTO THE HP DRUM. REPLACE THE ENGINE.

70740

No N3 Rotation Figure 103

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**PREREQUISITES** LOW MAX MOTORING NONE NO DRY MOTOR THE ENGINE 21 REPLACE THE STARTER CON-TROL VALVE (MM 80-11-02/401). (MM 71-00-00/201) AND MAKE SURE THE STARTER CONTROL VALVE POSITION INDICATOR SHOWS OPEN. IS THE VALVE FULLY OPEN? YES EXAMINE THE STARTER AIR 22 REPLACE AND/OR REPAIR THE DUCTS FOR LEAKS. DUCTS IF IT IS NECESSARY. ARE LEAKS PRESENT? 23 REPLACE THE STARTER (MM 80-11-01/401).

Low Max Motoring Figure 104

80-11-00

#### **PREREQUISITES**

MAKE SURE THESE CIRCUIT BREAKERS ARE CLOSED: 11D19 (11D20)

MAKE SURE THE AIRPLANE IS IN THIS CONFIGURATION: ELECTRICAL POWER IS ON (AMM 24-22-00/201)

STARTER "VALVE" LIGHT ON

YES 1 PUT THE ENGINE START 21 REPLACE THE STARTER SWITCH TO GND POSITION. CONTROL VALVE MEASURE FOR 28V DC AT THE (AMM 80-11-02/401). ENGINE STARTER VALVE, CONNECTOR D1306, PIN 1 (PIN 2, GND) (WDM 71-51-22). IS THERE 28V DC? NO 22 EXAMINE THE CIRCUIT FROM THE ENGINE STARTER VALVE, CONNECTOR D1306, PIN 1, TO THE L (R) ENGINE START SWITCH, S3 (S4), CONNECTOR D3572 (D3574), PIN 15 (12) (WDM 71-51-11; 80-11-11, -21). REPAIR THE PROBLEMS THAT

Starter VALVE Light On Figure 105

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R03

YOU FIND.

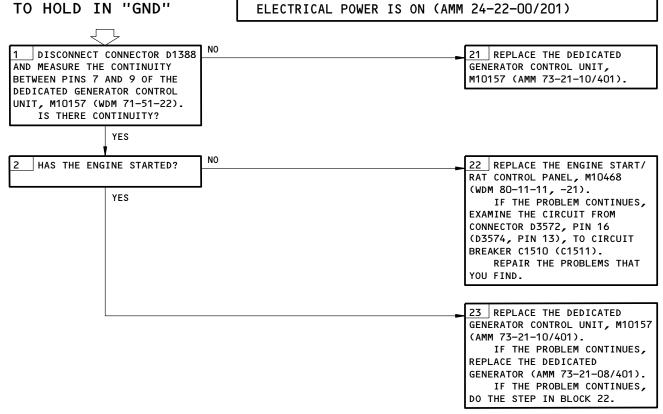
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RB 211 ENGINES

#### **PREREQUISITES**

MAKE SURE THESE CIRCUIT BREAKERS ARE CLOSED: 11D19 (11D20)

MAKE SURE THE AIRPLANE IS IN THIS CONFIGURATION: ELECTRICAL POWER IS ON (AMM 24-22-00/201)



Start Switch Failed to Hold in GND Figure 106

EFFECTIVITY-ALL

START SWITCH FAILED

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/ RB.211 ENGINES

STARTER "VALVE" LIGHT ON AFTER START SWITCH **RETURNED TO "AUTO"** 

**PREREQUISITES** NONE

EXAMINE THE STARTER CON-TROL VALVE POSITION INDICATOR. IS IT IN THE OPEN POSI-

NO

YES

21 REPLACE THE STARTER CON-TROL VALVE (AMM 80-11-02/401).

22 REPLACE THE LEFT (RIGHT) ENGINE START DISAGREE RELAY, K10209 (K10221)(WDM 80-11-11, -21).

IF THE PROBLEM CONTINUES, REPLACE THE LEFT (RIGHT) ENGINE START RELAY 3, K10248 (K10251).

IF THE PROBLEM CONTINUES, REMOVE THE EICAS COMPUTER (AMM 31-41-02/401).

EXAMINE THE CIRCUIT FROM EICAS COMPUTER TO THE LEFT (RIGHT) ENGINE START DISAGREE RELAY K10209 (K10221) AND TO THE LEFT (RIGHT) ENGINE START RELAY K10248 (K10251) (WDM 80-11-11,-21).

INSTALL THE EICAS COMPUTER.

Starter VALVE Light on After Start Switch Returned to AUTO Figure 107

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RB.211 ENGINES 

## **PREREQUISITES**

MAKE SURE THESE CIRCUIT BREAKERS ARE CLOSED: 11D19 (11D20)

MAKE SURE THE AIRPLANE IS IN THIS CONFIGURATION: ELECTRICAL POWER IS ON (AMM 24-22-00/201)

START SWITCH FAILED TO RETURN TO "AUTO" AT 47% N3 1>>

NO 1 REMOVE THE LEFT (RIGHT) 21 REPLACE THE ENGINE START/ ENGINE START RELAY 2, K10247 RAM AIR TURBINE CONTROL PANEL, (K10250), LEFT (RIGHT) ENGINE M10468 (WDM 80-11-11, START RELAY 3, K10248 WDM 80-11-21). (K10251) AND LEFT (RIGHT) ENGINE CONTROLS POWER RELAY, K10208 (K10220). DISCONNECT CONNECTOR D1306 AT THE ENGINE STARTER VALVE, v10008. PUT THE LEFT (RIGHT) ENGINE START SWITCH, S3 (S4), TO GND. DISCONNECT CONNECTOR D1388. MAKE SURE THE START SW MOVES TO AUTO. CONNECT CONNECTORS D1388 AND D1306. YFS **INSTALL RELAYS K10247** 22 REPLACE THE DEDICATED (K10250), K10248 (K10251), GENERATOR CONTROL UNIT, AND K10208 (K10220) M10157 (AMM 73-21-10/401). (WDM 80-11-11, WDM 80-11-21). IF THE PROBLEM CONTINUES, DID THE START SWITCH MOVE REPLACE THE DEDICATED TO AUTO? GENERATOR (AMM 73-21-08/401).

1 > IT IS ACCEPTABLE IF THE START SWITCH DOES NOT RETURN TO "AUTO" BY 47% N3, PROVIDED THAT THE START SWITCH DOES RETURN TO "AUTO" BY IDLE SPEED AND THE "EICAS" CAUTION MESSAGE "L (R) STARTER CUTOUT" DOES NOT DISPLAY.

> Start Switch Failed to Return to AUTO at 47% N3 Figure 108

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