

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-001-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE LOWER HALF OF FUSELAGE - EXTERIOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 100			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-100-01Z
		<p>*** LOWER HALF OF FUSELAGE - EXTERIOR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ALL FUSELAGE INSTALLATIONS IN MAJOR ZONE 100.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

0 0 6 0	EFFECTIVITY	GEN VISUAL	LOWER HALF OF FUSELAGE - EXTERIOR
			06-001-01 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	LOWER HALF OF FUSELAGE - EXTERIOR
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-002-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE RADOME		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 111			ACCESS PANELS 111AL			

MECH	INSP	MPD ITEM NUMBER 0600-111-01Z
		<p>*** RADOME PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

0 0 6 2	EFFECTIVITY	GEN VISUAL	RADOME
			06-002-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	RADOME
		06-002-01 PAGE 2 OF 2 APR 22/08

STATION
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DATE



BOEING CARD NO. 06-003-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE AREAS FORWARD OF NLG WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 113 114			ACCESS PANELS 113AL			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** AREAS FORWARD OF NLG WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>	<p>0600-113-01Z 0600-114-01Z</p>

0 0 6 4	EFFECTIVITY	GEN VISUAL	AREAS FORWARD OF NLG WHEEL WELL
			06-003-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	AREAS FORWARD OF NLG WHEEL WELL
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STATION
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BOEING CARD NO. 06-004-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA NOSE W/W	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE NOSE LANDING GEAR WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 115 116			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER
		<p>*** NOSE LANDING GEAR WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>0600-115-01Z 0600-116-01Z</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING OF THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS

0 0 6 6	EFFECTIVITY	GEN VISUAL	NOSE LANDING GEAR WHEEL WELL
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MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

NOSE LANDING GEAR WHEEL WELL

06-004-01

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STATION
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DATE



BOEING CARD NO. 06-005-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS OUTBD AND ABOVE NLG WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 117 118			ACCESS PANELS 119AL			

MECH	INSP	MPD ITEM NUMBER				
		*** AREAS OUTBD AND ABOVE NLG WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.				0600-117-01Z 0600-118-01Z

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EFFECTIVITY	GEN VISUAL	AREAS OUTBD AND ABOVE NLG WHEEL WELL	06-005-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-006-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE MAIN EQUIPMENT CENTER	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 119		ACCESS PANELS 119AL NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-119-01Z
<p>*** MAIN EQUIPMENT CENTER PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND EQUIPMENT NOT REQUIRED.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 0 6 9	EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER
			06-006-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

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 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

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 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

MAIN EQUIPMENT CENTER

06-006-01

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-007-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE MAIN EQUIPMENT CENTER	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 119		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER 0600-119-02Z			
		<p>*** MAIN EQUIPMENT CENTER PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF MAIN EQUIPMENT CENTER DOOR.</p>			

EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER	06-007-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-008-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 013	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE FORWARD CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 121 122			ACCESS PANELS 821 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** FORWARD CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION, SIDEWALL PANELS, CEILING PANELS NOT REQUIRED.</p>				<p>0600-121-01Z 0600-122-01Z</p>

EFFECTIVITY	GEN VISUAL	FORWARD CARGO COMPARTMENT	06-008-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-009-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE FORWARD CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 121 122		ACCESS PANELS 1211 1212 1221 821 NOTE				

MECH	INSP	MPD ITEM NUMBER					
		<p>*** FORWARD CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1212 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1221 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1211 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT LEFT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p>				0600-121-02Z	0600-122-02Z

EFFECTIVITY	GEN VISUAL	FORWARD CARGO COMPARTMENT	06-009-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
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BOEING CARD NO. 06-010-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS BELOW FWD CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 123 124			ACCESS PANELS 1231 821			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS BELOW FWD CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1231 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL 25-52-03.</p>				<p>0600-123-01Z 0600-124-01Z</p>

EFFECTIVITY	GEN VISUAL	AREAS BELOW FWD CARGO COMPARTMENT	06-010-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
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TASK CARD

BOEING CARD NO. 06-011-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AREAS AFT OF FWD CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 125 126		ACCESS PANELS 1251 821				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS AFT OF FWD CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p>				<p>0600-125-01Z 0600-126-01Z</p>

EFFECTIVITY	GEN VISUAL	AREAS AFT OF FWD CARGO COMPARTMENT
		06-011-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-012-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 013	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS ABOVE WING CENTER SECTION		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 131 132			ACCESS PANELS 1311 1312 2411 NOTE			

MECH	INSP	MPD ITEM NUMBER
		<p>*** AREAS ABOVE WING CENTER SECTION PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: CABIN EQUIPMENT BETWEEN STA 785 AND 955.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1312 REQUIRES REMOVAL OF THE CABIN FLOOR PANELS, STA 785 TO 955, PER MAINTENANCE MANUAL 53-01-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES REMOVAL OF THE MID CABIN EQUIPMENT, STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1311 REQUIRES REMOVAL OF THE CABIN FLOOR COVERING, STA 785 TO 955, PER MAINTENANCE MANUAL 25-27-01.</p>
		0600-131-01Z 0600-132-01Z

EFFECTIVITY	GEN VISUAL	AREAS ABOVE WING CENTER SECTION
		06-012-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-013-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA WG CTR SECT	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING CENTER SECTION		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 133 134		ACCESS PANELS 1331 134AZ 134BZ 134CZ 134DZ 134EZ 134FZ 134GZ 136KZ 194HR				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** WING CENTER SECTION PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>AIRPLANE NOTE: AIRPLANES WITH 2-BAY OR 4-BAY CENTER SECTION FUEL TANK.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1331 REQUIRES DEFUEL/PURGE OF THE AUXILIARY FUEL TANKS (CENTER WING, 2 BAY OR 4 BAY) PER MAINTENANCE MANUAL 28-11-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 				0600-133-01Z 0600-134-01Z

EFFECTIVITY	GEN VISUAL	WING CENTER SECTION
		06-013-02 PAGE 1 OF 2 APR 22/08

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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

WING CENTER SECTION

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-014-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ECS BAY	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENVIRONMENTAL CONTROL SYSTEM BAYS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 135 136		ACCESS PANELS 193LL 193ML 193NL 194GR 194LR 194NR				

MECH	INSP	MPD ITEM NUMBER	
		<p>*** ENVIRONMENTAL CONTROL SYSTEM BAYS PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>	<p>0600-135-01Z 0600-136-01Z</p>

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EFFECTIVITY	GEN VISUAL	ENVIRONMENTAL CONTROL SYSTEM BAYS
		06-014-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-015-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 013	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS ABOVE MLG WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 141 142			ACCESS PANELS 1411 1412 2411 NOTE			

MECH	INSP	MPD ITEM NUMBER
		<p>*** AREAS ABOVE MLG WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: CABIN EQUIPMENT BETWEEN STA 955 AND 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1412 REQUIRES REMOVAL OF THE CABIN FLOOR PANELS, STA 955 TO 1065, PER MAINTENANCE MANUAL 53-01-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES REMOVAL OF THE MID CABIN EQUIPMENT, STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1411 REQUIRES REMOVAL OF THE CABIN FLOOR PANEL/COVERING, STA 955 TO 1065, PER MAINTENANCE MANUAL 53-01-01.</p>
		0600-141-01Z 0600-142-01Z

EFFECTIVITY	GEN VISUAL	AREAS ABOVE MLG WHEEL WELL
		06-015-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-016-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L MAIN W/W	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE LEFT MAIN WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 143			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 0600-143-01Z
<p>*** LEFT MAIN WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING OF THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

EFFECTIVITY	GEN VISUAL	LEFT MAIN WHEEL WELL
		06-016-01 PAGE 1 OF 2 APR 22/08

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AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

LEFT MAIN WHEEL WELL

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-017-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R MAIN W/W	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE RIGHT MAIN WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 144			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 0600-144-01Z
		<p>*** RIGHT MAIN WHEEL WELL PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING OF THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS

EFFECTIVITY	GEN VISUAL	RIGHT MAIN WHEEL WELL
		06-017-01 PAGE 1 OF 2 APR 22/08

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AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

RIGHT MAIN WHEEL WELL

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-018-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AREAS FWD OF AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151 152		ACCESS PANELS 1511 822 NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS FWD OF AFT CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES IN CONJUNCTION WITH ZONAL TASKS 0600-153-01Z, 0600-154-01Z.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FWD WALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p>				<p>0600-151-01Z 0600-152-01Z</p>

EFFECTIVITY	GEN VISUAL	AREAS FWD OF AFT CARGO COMPARTMENT
		06-018-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-019-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AREAS FWD OF AFT CARGO COMPARTMENT			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151 152		ACCESS PANELS 1511 1531 1532 1541 822 NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS FWD OF AFT CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES IN CONJUNCTION WITH ZONAL TASKS 0600-153-02Z, 0600-154-02Z.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FWD WALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1531 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT LEFT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p>				0600-151-02Z 0600-152-02Z

0 0 8 6	EFFECTIVITY	GEN VISUAL	AREAS FWD OF AFT CARGO COMPARTMENT
			06-019-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-020-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AREAS FWD OF AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151 152		ACCESS PANELS 1511 1551 822				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS FWD OF AFT CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES IN CONJUNCTION WITH ZONAL TASKS 0600-155-01Z, 0600-156-01Z.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FWD WALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1551 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL 25-52-03.</p>				0600-151-03Z 0600-152-03Z

EFFECTIVITY	GEN VISUAL	AREAS FWD OF AFT CARGO COMPARTMENT	06-020-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-021-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AFT CARGO COMPARTMENT - LEFT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 153			ACCESS PANELS 822 NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-153-01Z
		<p>*** AFT CARGO COMPARTMENT - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION, SIDEWALL PANELS AND CEILING PANELS NOT REQUIRED.</p>

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EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT - LEFT
		06-021-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-022-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AFT CARGO COMPARTMENT - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 153		ACCESS PANELS 1531 1532 822 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-153-02Z
		<p>*** AFT CARGO COMPARTMENT - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1531 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p>

EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT - LEFT
		06-022-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-023-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AFT CARGO COMPARTMENT - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 154		ACCESS PANELS 822 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-154-01Z
		<p>*** AFT CARGO COMPARTMENT - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION, SIDEWALL PANELS AND CEILING PANELS NOT REQUIRED.</p>

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EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT - RIGHT
		06-023-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-024-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE AFT CARGO COMPARTMENT - RIGHT			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 154		ACCESS PANELS 1532 1541 822 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-154-02Z
		<p>*** AFT CARGO COMPARTMENT - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL 25-52-01.</p>

EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT - RIGHT
		06-024-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-025-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS BELOW AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 155 156			ACCESS PANELS 1551 822			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** AREAS BELOW AFT CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1551 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL 25-52-03.</p>	<p>0600-155-01Z 0600-156-01Z</p>

EFFECTIVITY	GEN VISUAL	AREAS BELOW AFT CARGO COMPARTMENT	06-025-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-026-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE BULK CARGO COMPARTMENT - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 161		ACCESS PANELS 811 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-161-01Z				
		<p>*** BULK CARGO COMPARTMENT - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION, SIDEWALL PANELS AND CEILING PANELS NOT REQUIRED.</p>				

EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT - LEFT	06-026-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-027-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE BULK CARGO COMPARTMENT - LEFT			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 161		ACCESS PANELS 1611 1612 811 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-161-02Z
		<p>*** BULK CARGO COMPARTMENT - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1611 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT LEFT SIDE PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1612 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL

0 0 9 4	EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT - LEFT
			06-027-01 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

BULK CARGO COMPARTMENT - LEFT

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-028-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE BULK CARGO COMPARTMENT - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 162		ACCESS PANELS 811 NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-162-01Z
		<p>*** BULK CARGO COMPARTMENT - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION, SIDEWALL PANELS AND CEILING PANELS NOT REQUIRED.</p>

0 0 9 6	EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT - RIGHT
			06-028-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-029-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 015	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE BULK CARGO COMPARTMENT - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 162			ACCESS PANELS 1612 1621 811 NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-162-02Z
		<p>*** BULK CARGO COMPARTMENT - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION AND FLOORING NOT REQUIRED.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1612 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1621 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT RIGHT SIDE PANELS PER MAINTENANCE MANUAL 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL

EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT - RIGHT
		06-029-01 PAGE 1 OF 2 APR 22/08

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MECH	INSP

DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):

- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.

WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

BULK CARGO COMPARTMENT - RIGHT
06-029-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-030-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS BELOW BULK CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 163 164			ACCESS PANELS 1631 811			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS BELOW BULK CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1631 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT FLOOR PANELS.</p>				<p>0600-163-01Z 0600-164-01Z</p>

0 0 9 9	EFFECTIVITY	GEN VISUAL	AREAS BELOW BULK CARGO COMPARTMENT
			06-030-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-031-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREAS AFT OF BULK CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 165 166			ACCESS PANELS 1651 811			

MECH	INSP		MPD ITEM NUMBER
		<p>*** AREAS AFT OF BULK CARGO COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1651 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT AFT WALL PANELS.</p>	<p>0600-165-01Z 0600-166-01Z</p>

0 1 0 0	EFFECTIVITY		GEN VISUAL	AREAS AFT OF BULK CARGO COMPARTMENT
				06-031-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-032-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE WING TO BODY FAIRINGS - FWD LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 191 193		ACCESS PANELS 191AL 193GL 193NL				

MECH	INSP	MPD ITEM NUMBER			
		*** WING TO BODY FAIRINGS - FWD LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.			0600-191-01Z 0600-193-01Z

0 1 0 1	EFFECTIVITY	GEN VISUAL	WING TO BODY FAIRINGS - FWD LEFT
			06-032-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-033-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE WING TO BODY FAIRINGS - FWD RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 192 194		ACCESS PANELS 192AR 194HR 194LR				

MECH	INSP	MPD ITEM NUMBER			
		*** WING TO BODY FAIRINGS - FWD RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.			0600-192-01Z 0600-194-01Z

0 1 0 2	EFFECTIVITY	GEN VISUAL	WING TO BODY FAIRINGS - FWD RIGHT
			06-033-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-034-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING TO BODY FAIRINGS - AFT LEFT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 195 197			ACCESS PANELS 195JL 195RL 195SL 5002			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** WING TO BODY FAIRINGS - AFT LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED (0600-195-01Z ONLY).</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENSION OF THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 	<p>0600-195-01Z 0600-197-01Z</p>

0 1 0 3	EFFECTIVITY	GEN VISUAL	WING TO BODY FAIRINGS - AFT LEFT
			06-034-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

WING TO BODY FAIRINGS - AFT LEFT

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-035-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING TO BODY FAIRINGS - AFT RIGHT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 196 198			ACCESS PANELS 196JR 1981 5002			

MECH	INSP	MPD ITEM NUMBER
		<p>*** WING TO BODY FAIRINGS - AFT RIGHT 0600-196-01Z PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES. 0600-198-01Z</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED (0600-196-01Z ONLY).</p> <p>ACCESS NOTE: SPECIAL ACCESS 1981 REQUIRES EXTENSION/ RETRACTION OF THE RAM AIR TURBINE PER MAINTENANCE MANUAL 29-21-00.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENSION OF THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE

0 1 0 5	EFFECTIVITY	GEN VISUAL	WING TO BODY FAIRINGS - AFT RIGHT
			06-035-01 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

WING TO BODY FAIRINGS - AFT RIGHT

06-035-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-036-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE UPPER HALF OF FUSELAGE-EXTERIOR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 200			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-200-01Z
<p>*** UPPER HALF OF FUSELAGE-EXTERIOR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ALL FUSELAGE INSTALLATIONS IN MAJOR ZONE 200.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

EFFECTIVITY	GEN VISUAL	UPPER HALF OF FUSELAGE-EXTERIOR
		06-036-01 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	UPPER HALF OF FUSELAGE-EXTERIOR
		06-036-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-037-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CREW CABIN	RELATED TASK CONTROL CABIN	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE CONTROL CABIN		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 211 212			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER	
		*** CONTROL CABIN PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.	0600-211-01Z 0600-212-01Z

EFFECTIVITY	GEN VISUAL	CONTROL CABIN	06-037-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-038-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE PASSENGER CABIN-SECTION 41		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 221 222			ACCESS PANELS 831 841 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** PASSENGER CABIN-SECTION 41</p> <p>PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF SIDEWALL PANELS NOT REQUIRED BUT ENTRY/SERVICE DOORS SHOULD BE OPEN.</p>				<p>0600-221-01Z</p> <p>0600-222-01Z</p>

0 1 1 0	EFFECTIVITY	GEN VISUAL	PASSENGER CABIN-SECTION 41
			06-038-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-039-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREA ABOVE CEILING PASS CABIN-SEC 41		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 223 224			ACCESS PANELS 2231 2232 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREA ABOVE CEILING PASS CABIN-SEC 41 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION NOT REQUIRED BUT ENTRY/SERVICE DOORS SHOULD BE CLOSED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2231 REQUIRES REMOVAL OF THE SCULPTURED CEILING PANEL - STA 367 TO 455.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2232 REQUIRES REMOVAL OF THE FORWARD LOWERED CEILING PANEL NO. 1 - FORWARD OF STA 284.</p>				<p>0600-223-01Z 0600-224-01Z</p>

0 1 1 1	EFFECTIVITY	GEN VISUAL	AREA ABOVE CEILING PASS CABIN-SEC 41
			06-039-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-040-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 010	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE PASSENGER CABIN - SECTION 43	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE PASS ALL		
ZONES 231 232		ACCESS PANELS NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** PASSENGER CABIN - SECTION 43 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF SIDEWALL PANELS NOT REQUIRED.</p>			<p>0600-231-01Z 0600-232-01Z</p>	

0 1 1 2	EFFECTIVITY	GEN VISUAL	PASSENGER CABIN - SECTION 43
			06-040-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-041-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREA ABOVE PASS CABIN CEILING-SEC 43		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 233 234			ACCESS PANELS 2331 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREA ABOVE PASS CABIN CEILING-SEC 43 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2331 REQUIRES REMOVAL OF THE SCULPTURED CEILING PANEL - STA 434 TO 785.</p>				<p>0600-233-01Z 0600-234-01Z</p>

0 1 1 3	EFFECTIVITY	GEN VISUAL	AREA ABOVE PASS CABIN CEILING-SEC 43
			06-041-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-042-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 010	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE PASSENGER CABIN-SECTION 45		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 241 242		ACCESS PANELS NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** PASSENGER CABIN-SECTION 45 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF SIDEWALL PANELS OR EMERGENCY ESCAPE HATCHES NOT REQUIRED.</p>			<p>0600-241-01Z 0600-242-01Z</p>	

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EFFECTIVITY	GEN VISUAL	PASSENGER CABIN-SECTION 45	06-042-01	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-043-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 010	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREA ABOVE PASS CABIN CEILING-SEC 45		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 243 244			ACCESS PANELS 2431 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREA ABOVE PASS CABIN CEILING-SEC 45 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF INSULATION NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES REMOVAL OF THE SCULPTURED CEILING PANEL - STA 785 TO 1065 IS REQUIRED.</p>				<p>0600-243-01Z 0600-244-01Z</p>

0 1 1 5	EFFECTIVITY	GEN VISUAL	AREA ABOVE PASS CABIN CEILING-SEC 45
			06-043-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-044-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 010	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE PASSENGER CABIN - SECTION 46		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 251 252		ACCESS PANELS 833 843 NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** PASSENGER CABIN - SECTION 46 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: REMOVAL OF SIDEWALL PANELS NOT REQUIRED BUT ENTRY/SERVICE DOORS SHOULD BE OPEN.</p>				<p>0600-251-01Z 0600-252-01Z</p>

EFFECTIVITY	GEN VISUAL	PASSENGER CABIN - SECTION 46
		06-044-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-045-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 010	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE AREA ABOVE PASS CABIN CEILING-SEC 46		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 253 254			ACCESS PANELS 2531 NOTE			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREA ABOVE PASS CABIN CEILING-SEC 46 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SCULPTURED PANELS: STA 1065 TO 1439, AFT LOWERED CEILING PANELS NO. 1 AND 2, STA 1515 TO 1569. REMOVAL OF INSULATION NOT REQUIRED, BUT ENTRY/SERVICE DOORS SHOULD BE CLOSED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2531 REQUIRES REMOVAL OF THE SCULPTURED CEILING PANEL - STA 1065 TO 1569.</p>				<p>0600-253-01Z 0600-254-01Z</p>

0 1 1 7	EFFECTIVITY	GEN VISUAL	AREA ABOVE PASS CABIN CEILING-SEC 46
			06-045-01 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-046-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMPENNAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE AREAS AFT OF PRESS B/H TO BS 1725.5		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 311 312			ACCESS PANELS 312AR			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS AFT OF PRESS B/H TO BS 1725.5 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>				0600-311-01Z 0600-312-01Z

0 1 1 8	EFFECTIVITY	GEN VISUAL	AREAS AFT OF PRESS B/H TO BS 1725.5
			06-046-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	AREAS AFT OF PRESS B/H TO BS 1725.5
		06-046-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-046-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMPENNAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 009	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE AREAS AFT OF PRESS B/H TO BS 1725.5		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 311 312			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER				
		<p>*** AREAS AFT OF PRESS B/H TO BS 1725.5 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE WITH TAIL SKID EXTENDED.</p> <p>AIRPLANE NOTE: APPLICABLE TO 767-300 AND 767-400ER</p>				<p>0600-311-02Z 0600-312-02Z</p>

0 1 2 0	EFFECTIVITY	GEN VISUAL	AREAS AFT OF PRESS B/H TO BS 1725.5
			06-046-02 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-047-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMPENNAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 004	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE STABILIZER TORSION BOX COMPT-LEFT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 313			ACCESS PANELS 312AR 313AL			

MECH	INSP	MPD ITEM NUMBER 0600-313-01Z
<p>*** STABILIZER TORSION BOX COMPT-LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

0 1 2 1	EFFECTIVITY	GEN VISUAL	STABILIZER TORSION BOX COMPT-LEFT
			06-047-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):**
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.**
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.**

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EFFECTIVITY	GEN VISUAL	STABILIZER TORSION BOX COMPT-LEFT
		06-047-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-048-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMPENNAGE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 004	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE STABILIZER TORSION BOX COMPT-RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 314			ACCESS PANELS 312AR 313AL			

MECH	INSP	MPD ITEM NUMBER 0600-314-01Z
<p>*** STABILIZER TORSION BOX COMPT-RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

0 1 2 3	EFFECTIVITY	GEN VISUAL	STABILIZER TORSION BOX COMPT-RIGHT
			06-048-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	STABILIZER TORSION BOX COMPT-RIGHT
		06-048-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-049-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA APU COMPT	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE APU COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 315 316			ACCESS PANELS 315AL 316AR			

MECH	INSP		MPD ITEM NUMBER
		<p>*** APU COMPARTMENT</p> <p>PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES AS VISIBLE WITH APU INSTALLED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>	<p>0600-315-01Z</p> <p>0600-316-01Z</p>

EFFECTIVITY		GEN VISUAL	APU COMPARTMENT
			06-049-01 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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6

EFFECTIVITY

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GEN VISUAL

APU COMPARTMENT

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-050-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA APU COMPT	RELATED TASK	INTERVAL APU CNG	PHASE 99XXX	MPD REV 018	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE APU COMPARTMENT	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 315 316		ACCESS PANELS 313AL 315AL 315BL 316AR				

MECH	INSP					MPD ITEM NUMBER
		<p>*** APU COMPARTMENT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF AREA EXPOSED BY APU REMOVAL.</p>				<p>0600-315-02Z 0600-316-02Z</p>

EFFECTIVITY	GEN VISUAL	APU COMPARTMENT	06-050-01	PAGE 1 OF 1	APR 22/06
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-051-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABLIZER	RELATED TASK	INTERVAL S 8C	PHASE 24896	MPD REV 003	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE VERT STAB - REMOVABLE LEADING EDGE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 321		ACCESS PANELS 321C 321D 321E 321K				

MECH	INSP	MPD ITEM NUMBER 0600-321-01Z			
		<p>*** VERT STAB - REMOVABLE LEADING EDGE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

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EFFECTIVITY	GEN VISUAL	VERT STAB - REMOVABLE LEADING EDGE
		06-051-01 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-052-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABLIZER	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE VERT STAB-AUX SPAR TO FRONT SPAR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 322		ACCESS PANELS 322AL 322AR 323AL 323AR				

MECH	INSP	MPD ITEM NUMBER 0600-322-01Z	
		<p>*** VERT STAB-AUX SPAR TO FRONT SPAR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	VERT STAB-AUX SPAR TO FRONT SPAR
		06-052-01 PAGE 1 OF 1 APR 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-053-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABLIZER	RELATED TASK	INTERVAL S 8C	PHASE 24896	MPD REV 003	TASK CARD REVISION APR 22/06
TASK GEN VISUAL		TITLE VERT STAB - AUX SPAR TO FRONT SPAR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 322			ACCESS PANELS 321BL 321BR 321C 321D 321E 321FZ 321GZ 321HZ 321K 321LZ 321MZ 321NZ 321PZ 321QZ 321RZ 321SZ 322AL 322AR 322CL			

MECH	INSP	MPD ITEM NUMBER 0600-322-02Z
		<p>*** VERT STAB - AUX SPAR TO FRONT SPAR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>

0 1 3 0	EFFECTIVITY	GEN VISUAL	VERT STAB - AUX SPAR TO FRONT SPAR
			06-053-01 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-054-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABILIZER	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE VERT STAB - FRONT SPAR TO REAR SPAR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 323			ACCESS PANELS 311AZ 312AR 323AL 323AR			

MECH	INSP	MPD ITEM NUMBER 0600-323-01Z
<p>*** VERT STAB - FRONT SPAR TO REAR SPAR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE, FIN-BODY DECK TO RIB 10.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

0 1 3 1	EFFECTIVITY	GEN VISUAL	VERT STAB - FRONT SPAR TO REAR SPAR
			06-054-01 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	VERT STAB - FRONT SPAR TO REAR SPAR
		06-054-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-055-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABLIZER	RELATED TASK	INTERVAL S 8C	PHASE 24896	MPD REV 003	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE VERT STAB - FRONT SPAR TO REAR SPAR			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 323		ACCESS PANELS 324PL 324QZ 324RZ 324SL 324TZ 324UZ 324WL 324XZ 324YZ 324ZZ				

MECH	INSP	MPD ITEM NUMBER 0600-323-02Z				
		<p>*** VERT STAB - FRONT SPAR TO REAR SPAR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE ABOVE RIB 10.</p>				

0 1 3 3	EFFECTIVITY	GEN VISUAL	VERT STAB - FRONT SPAR TO REAR SPAR
			06-055-01 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-056-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABILIZER	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE VERT STAB-REAR SPAR TO TRAILING EDGE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 324			ACCESS PANELS 324BL 324EL 324GL 324JL 324LL 324PL 324SL 324WL			

MECH	INSP	MPD ITEM NUMBER 0600-324-01Z
		<p>*** VERT STAB-REAR SPAR TO TRAILING EDGE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

0 1 3 4	EFFECTIVITY	GEN VISUAL	VERT STAB-REAR SPAR TO TRAILING EDGE
			06-056-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):**
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.**
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.**

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EFFECTIVITY

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GEN VISUAL

VERT STAB-REAR SPAR TO TRAILING EDGE
06-056-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-057-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABILIZER	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK EX SURVEILL		TITLE VERT STAB-REAR SPAR TO TRAILING EDGE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 324			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-324-02Z
		<p>*** VERT STAB-REAR SPAR TO TRAILING EDGE PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

0	EFFECTIVITY	EX SURVEILL	VERT STAB-REAR SPAR TO TRAILING EDGE
1			
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6			06-057-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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EX SURVEILL

VERT STAB-REAR SPAR TO TRAILING EDGE

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-058-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABILIZER	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE RUDDER		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 325			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-325-01Z
		<p>*** RUDDER PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

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EFFECTIVITY	GEN VISUAL	RUDDER	06-058-01	PAGE 1 OF 2	APR 22/08
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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	RUDDER
		06-058-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-059-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA V STABILIZER	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE VERT STABILIZER-TIP		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 326			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-326-01Z
<p>*** VERT STABILIZER-TIP PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

0 1 4 0	EFFECTIVITY	GEN VISUAL	VERT STABILIZER-TIP
			06-059-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):**
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.**
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.**

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EFFECTIVITY

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GEN VISUAL

VERT STABILIZER-TIP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-060-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABILIZER	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE HORIZONTAL STAB CENTER SECTION		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 331 341			ACCESS PANELS 312AR 341AZ			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** HORIZONTAL STAB CENTER SECTION PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED (0600-341-01Z ONLY).</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 	<p>0600-331-01Z 0600-341-01Z</p>

0 1 4 2	EFFECTIVITY	GEN VISUAL	HORIZONTAL STAB CENTER SECTION
			06-060-01 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY _____ GEN VISUAL HORIZONTAL STAB CENTER SECTION

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-061-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABILIZER	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE HORIZONTAL STAB CENTER SECTION		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 331 341			ACCESS PANELS 312AR 313AL			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** HORIZONTAL STAB CENTER SECTION PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>	<p>0600-331-02Z 0600-341-02Z</p>

0 1 4 4	EFFECTIVITY	GEN VISUAL	HORIZONTAL STAB CENTER SECTION
			06-061-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

HORIZONTAL STAB CENTER SECTION

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-062-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABLIZER	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 011	TASK CARD REVISION AUG 22/06
TASK GEN VISUAL		TITLE HORZ STAB-FRONT TO REAR SPAR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 334 344			ACCESS PANELS 335AAZ 335EB 335GB 335HB 345AAZ 345EB 345GB 345HB			

MECH	INSP	MPD ITEM NUMBER			
		<p>*** HORZ STAB-FRONT TO REAR SPAR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF AREA FORWARD OF REAR SPAR, AND BETWEEN RIBS 8 AND 11 (OTHER AREAS REQUIRE NO ADDITIONAL ZONAL GENERAL VISUAL INSPECTION TASK).</p>			
		0600-334-01Z 0600-344-01Z			

0 1 4 6	EFFECTIVITY	GEN VISUAL	HORZ STAB-FRONT TO REAR SPAR
			06-062-01 PAGE 1 OF 1 AUG 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-063-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABLIZER	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 013	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE HORZ STAB-REAR SPAR TO STAB TE			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 335 345		ACCESS PANELS 335CB 335DB 335EB 335GB 335HB 345CB 345DB 345EB 345GB 345HB				

MECH	INSP					MPD ITEM NUMBER
		<p>*** HORZ STAB-REAR SPAR TO STAB TE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF AREAS BETWEEN RIB 3 (STABILIZER STATION 107.9/122.6) AND RIB 11 (STABILIZER STATION 311.6).</p>				0600-335-01Z 0600-345-01Z

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EFFECTIVITY	GEN VISUAL	HORZ STAB-REAR SPAR TO STAB TE
		06-063-01 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-064-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABLIZER	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION DEC 22/06
TASK GEN VISUAL	TITLE HORZ STAB-REAR SPAR TO STAB TE			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 335 345		ACCESS PANELS 335BB 335CB 335DB 335JB 335KB 335LB 335MB 345BB 345CB 345DB 345JB 345KB 345LB 345MB				

MECH	INSP					MPD ITEM NUMBER
		<p>*** HORZ STAB-REAR SPAR TO STAB TE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF AREAS OUTBOARD OF RIB 11.</p>				<p>0600-335-02Z 0600-345-02Z</p>

EFFECTIVITY	GEN VISUAL	HORZ STAB-REAR SPAR TO STAB TE
		06-064-01 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-065-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABILIZER	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE HORZ STAB-REAR SPAR TO STAB TE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 335 345			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** HORZ STAB-REAR SPAR TO STAB TE PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>	<p>0600-335-03Z 0600-345-03Z</p>

0 1 4 9	EFFECTIVITY	GEN VISUAL	HORZ STAB-REAR SPAR TO STAB TE
			06-065-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

HORZ STAB-REAR SPAR TO STAB TE

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-067-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA H STABILIZER	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE HORIZONTAL STABILIZER-TIP		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 338 348			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** HORIZONTAL STABILIZER-TIP PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>	<p>0600-338-01Z 0600-348-01Z</p>

0 1 5 1	EFFECTIVITY	GEN VISUAL	HORIZONTAL STABILIZER-TIP
			06-067-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

HORIZONTAL STABILIZER-TIP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-098-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 2C	11224	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE L WING LE TO FRONT SPAR - INBOARD			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL
ZONES 511			ACCESS PANELS 5004 511CB 511DB 511EB 511GB 511HB 511MB NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-511-01Z
		<p>*** L WING LE TO FRONT SPAR - INBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE. ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: 511GB IS FOR V0001-V0136 ONLY.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

0 1 5 3	EFFECTIVITY	GEN VISUAL	L WING LE TO FRONT SPAR - INBOARD
			06-098-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

L WING LE TO FRONT SPAR - INBOARD

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-098-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE R WING LE TO FRONT SPAR - INBOARD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 611			ACCESS PANELS 5004 611CB 611DB 611EB 611GB 611HB 611MB NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-611-01Z
		<p>*** R WING LE TO FRONT SPAR - INBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: 611GB IS FOR V0001-V0136 ONLY.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

0 1 5 5	EFFECTIVITY	GEN VISUAL	R WING LE TO FRONT SPAR - INBOARD
			06-098-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP
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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

R WING LE TO FRONT SPAR - INBOARD

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-099-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION AUG 22/08
TASK GEN VISUAL		TITLE L WING LE TO FRONT SPAR - INBOARD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 511			ACCESS PANELS 5004 511AB 511BB 511FB 511GB 511JB 511KB 511LB 511NB 511QB 511RB NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-511-02Z
		<p>*** L WING LE TO FRONT SPAR - INBOARD</p> <p>PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (PERFORMING THIS TASK PRECLUDES NEED FOR ZONAL TASK 0600-511-01Z)</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: 511GB IS FOR V0001-V0136 ONLY.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

0 1 5 7	EFFECTIVITY	GEN VISUAL	L WING LE TO FRONT SPAR - INBOARD
			06-099-01-1 PAGE 1 OF 2 AUG 22/08

MECH	INSP

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

L WING LE TO FRONT SPAR - INBOARD

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-099-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 2C	MPD REV 018	TASK CARD REVISION AUG 22/08
TASK GEN VISUAL		TITLE R WING LE TO FRONT SPAR - INBOARD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 611		ACCESS PANELS 5004 611AB 611BB 611FB 611GB 611JB 611KB 611LB 611NB 611QB 611RB NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-611-02Z
		<p>*** R WING LE TO FRONT SPAR - INBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (PERFORMING THIS TASK PRECLUDES THE NEED FOR ZONAL TASK 0600-611-01Z). CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: 611GB IS FOR V0001-V0136 ONLY.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE

0 1 5 9	EFFECTIVITY	GEN VISUAL	R WING LE TO FRONT SPAR - INBOARD
			06-099-01-2 PAGE 1 OF 2 AUG 22/08

MECH	INSP
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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

R WING LE TO FRONT SPAR - INBOARD

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-100-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE L WING LE TO FRONT SPAR - INBOARD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 511			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-511-03Z	
		<p>*** L WING LE TO FRONT SPAR - INBOARD PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

0 1 6 1	EFFECTIVITY	GEN VISUAL	L WING LE TO FRONT SPAR - INBOARD
			06-100-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-100-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE R WING LE TO FRONT SPAR - INBOARD	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 611		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER 0600-611-03Z				
		<p>*** R WING LE TO FRONT SPAR - INBOARD PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>				

0 1 6 2	EFFECTIVITY	GEN VISUAL	R WING LE TO FRONT SPAR - INBOARD
			06-100-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-101-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE INBOARD SLAT (NO. 6) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 512			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-512-01Z
<p>*** INBOARD SLAT (NO. 6) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 6 3	EFFECTIVITY	GEN VISUAL	INBOARD SLAT (NO. 6) - LEFT WING
			06-101-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
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 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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6
4

EFFECTIVITY

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GEN VISUAL

INBOARD SLAT (NO. 6) - LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-101-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE INBOARD SLAT (NO. 7) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 612			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-612-01Z
<p>*** INBOARD SLAT (NO. 7) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 6 5	EFFECTIVITY	GEN VISUAL	INBOARD SLAT (NO. 7) - RIGHT WING
			06-101-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP
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GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

INBOARD SLAT (NO. 7) - RIGHT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-102-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 2C	11224	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE L WING LE TO FRONT SPAR - OUTBOARD			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 521			ACCESS PANELS 5004 521AAB 521ABB 521ACB 521ADB 521AGB 521AHB 521AJB 521AKB 521CB 521DB 521FB 521GB 521KB 521LB 521MB 521NB 521QB 521UB 521VB 521WB 521XB				

MECH	INSP	MPD ITEM NUMBER 0600-521-01Z					
		<p>*** L WING LE TO FRONT SPAR - OUTBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p>					

EFFECTIVITY	GEN VISUAL	L WING LE TO FRONT SPAR - OUTBOARD
		06-102-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-102-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE R WING LE TO FRONT SPAR - OUTBOARD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 621		ACCESS PANELS 5004 621AAB 621ABB 621ACB 621ADB 621AGB 621AHB 621AJB 621AKB 621CB 621DB 621FB 621GB 621KB 621LB 621MB 621NB 621UB 621VB 621WB 621XB				

MECH	INSP	MPD ITEM NUMBER
		0600-621-01Z
<p>*** R WING LE TO FRONT SPAR - OUTBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p>		

EFFECTIVITY	GEN VISUAL	R WING LE TO FRONT SPAR - OUTBOARD
		06-102-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-103-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 2C	PHASE 12424	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE L WING LE TO FRONT SPAR - OUTBOARD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 521			ACCESS PANELS 5004 521AB 521AEB 521AFB 521ALB 521AMB 521ANB 521BB 521EB 521HB 521JB 521PB 521QB 521RB 521TB 521YB 521ZB			

MECH	INSP	MPD ITEM NUMBER 0600-521-02Z
<p>*** L WING LE TO FRONT SPAR - OUTBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (PERFORMING THIS TASK PRECLUDES THE NEED FOR ZONAL TASK 0600-521-01Z). CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

0 1 6 9	EFFECTIVITY	GEN VISUAL	L WING LE TO FRONT SPAR - OUTBOARD
			06-103-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

L WING LE TO FRONT SPAR - OUTBOARD

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-103-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 2C	11224	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE R WING LE TO FRONT SPAR - OUTBOARD			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 621			ACCESS PANELS 5004 621AB 621AEB 621AFB 621ALB 621AMB 621ANB 621BB 621EB 621HB 621JB 621PB 621QB 621RB 621SB 621TB 621YB 621ZB NOTE				

MECH	INSP	MPD ITEM NUMBER 0600-621-02Z
<p>*** R WING LE TO FRONT SPAR - OUTBOARD PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE. (PERFORMING THIS TASK PRECLUDES NEED FOR ZONAL TASK 0600-621-01Z) ACCESS NOTE: AIRPLANE WILL HAVE EITHER ACCESS PANEL 621SB OR ACCESS PANELS 621PB, 621QB, AND 621RB. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00. THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

0 1 7 1	EFFECTIVITY	GEN VISUAL	R WING LE TO FRONT SPAR - OUTBOARD
			06-103-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):

- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.

WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	R WING LE TO FRONT SPAR - OUTBOARD
			06-103-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-104-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 5) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 522			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-522-01Z
<p>*** OUTBOARD SLAT (NO. 5) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 7 3	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 5) - LEFT WING
			06-104-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	OUTBOARD SLAT (NO. 5) - LEFT WING
			06-104-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-104-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 8) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 622		ACCESS PANELS 5004				

MECH	INSP	MPD ITEM NUMBER 0600-622-01Z
		<p>*** OUTBOARD SLAT (NO. 8) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 1 7 5	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 8) - RIGHT WING
			06-104-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD SLAT (NO. 8) - RIGHT WING

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-105-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 5) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 522			ACCESS PANELS 522DT			

MECH	INSP	MPD ITEM NUMBER 0600-522-02Z
		<p>*** OUTBOARD SLAT (NO. 5) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>

0 1 7 7	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 5) - LEFT WING
			06-105-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-105-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE OUTBOARD SLAT (NO. 8) - RIGHT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 622		ACCESS PANELS 622DT				

MECH	INSP	MPD ITEM NUMBER 0600-622-02Z				
		<p>*** OUTBOARD SLAT (NO. 8) - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>				

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EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 8) - RIGHT WING
		06-105-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-106-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 4) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 523		ACCESS PANELS 5004				

MECH	INSP	MPD ITEM NUMBER 0600-523-01Z
		<p>*** OUTBOARD SLAT (NO. 4) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 1 7 9	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 4) - LEFT WING
			06-106-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
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 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 4) - LEFT WING
		06-106-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-106-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 9) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 623			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-623-01Z
<p>*** OUTBOARD SLAT (NO. 9) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 8 1	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 9) - RIGHT WING
			06-106-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD SLAT (NO. 9) - RIGHT WING

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-107-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 4) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 523			ACCESS PANELS 523AT 523FT			

MECH	INSP	MPD ITEM NUMBER 0600-523-02Z	
		<p>*** OUTBOARD SLAT (NO. 4) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

0 1 8 3	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 4) - LEFT WING
			06-107-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-107-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 9) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 623			ACCESS PANELS 623AT 623FT			

MECH	INSP	MPD ITEM NUMBER 0600-623-02Z	
		<p>*** OUTBOARD SLAT (NO. 9) - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE</p>	

0 1 8 4	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 9) - RIGHT WING
			06-107-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-108-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 3) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 524			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-524-01Z
<p>*** OUTBOARD SLAT (NO. 3) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 8 5	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 3) - LEFT WING
			06-108-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	OUTBOARD SLAT (NO. 3) - LEFT WING
			06-108-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-108-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 10) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 624			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-624-01Z
		<p>*** OUTBOARD SLAT (NO. 10) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 1 8 7	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 10) - RIGHT WING
			06-108-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD SLAT (NO. 10) - RIGHT WING

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-109-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 3) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 524			ACCESS PANELS 524AT 524FT			

MECH	INSP	MPD ITEM NUMBER 0600-524-02Z	
		<p>*** OUTBOARD SLAT (NO. 3) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

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EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 3) - LEFT WING
		06-109-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-109-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE OUTBOARD SLAT (NO. 10) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 624		ACCESS PANELS 624AT 624FT				

MECH	INSP	MPD ITEM NUMBER 0600-624-02Z	
		<p>*** OUTBOARD SLAT (NO. 10) - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

0 1 9 0	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 10) - RIGHT WING
			06-109-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-110-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 2) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 525			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-525-01Z
<p>*** OUTBOARD SLAT (NO. 2) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 9 1	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 2) - LEFT WING
			06-110-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD SLAT (NO. 2) - LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-110-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 11) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 625		ACCESS PANELS 5004				

MECH	INSP	MPD ITEM NUMBER 0600-625-01Z
<p>*** OUTBOARD SLAT (NO. 11) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 9 3	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 11) - RIGHT WING
			06-110-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD SLAT (NO. 11) - RIGHT WING

06-110-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-111-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 2) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 525			ACCESS PANELS 525AT 525FT			

MECH	INSP	MPD ITEM NUMBER	
		*** OUTBOARD SLAT (NO.2) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.	0600-525-02Z

0 1 9 5	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 2) - LEFT WING
			06-111-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-111-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE OUTBOARD SLAT (NO. 11) - RIGHT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 625		ACCESS PANELS 625AT 625FT				

MECH	INSP	MPD ITEM NUMBER 0600-625-02Z			
		<p>*** OUTBOARD SLAT (NO. 11) - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

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EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 11) - RIGHT WING 06-111-01-2 PAGE 1 OF 1 AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-112-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 1) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 526			ACCESS PANELS 5004			

MECH	INSP	MPD ITEM NUMBER 0600-526-01Z
<p>*** OUTBOARD SLAT (NO. 1) - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 1 9 7	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 1) - LEFT WING
			06-112-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	OUTBOARD SLAT (NO. 1) - LEFT WING
			06-112-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-112-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 12) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 626		ACCESS PANELS 5004				

MECH	INSP	MPD ITEM NUMBER 0600-626-01Z
		<p>*** OUTBOARD SLAT (NO. 12) - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5004 REQUIRES EXTENSION OF THE LEADING EDGE SLATS PER MAINTENANCE MANUAL 27-81-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 1 9 9	EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 12) - RIGHT WING
			06-112-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

OUTBOARD SLAT (NO. 12) - RIGHT WING

06-112-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-113-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 1) - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 526		ACCESS PANELS 526AT 526FT				

MECH	INSP	MPD ITEM NUMBER 0600-526-02Z			
		<p>*** OUTBOARD SLAT (NO. 1) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 1) - LEFT WING
		06-113-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-113-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING LE	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD SLAT (NO. 12) - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 626			ACCESS PANELS 626AT 626FT			

MECH	INSP	MPD ITEM NUMBER 0600-626-02Z	
		<p>*** OUTBOARD SLAT (NO. 12) - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	OUTBOARD SLAT (NO. 12) - RIGHT WING
		06-113-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-114-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE CENTER AUXILIARY TANK - LEFT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE	
ZONES 531		ACCESS PANELS 1331 531AB 531BB				

MECH	INSP	MPD ITEM NUMBER 0600-531-01Z
<p>*** CENTER AUXILIARY TANK - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1331 REQUIRES DEFUELING AND PURGING THE AUXILIARY FUEL TANKS (CENTER WING, 2 BAY OR 4 BAY) PER MAINTENANCE MANUAL 28-11-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

0 2 0 3	EFFECTIVITY	GEN VISUAL	CENTER AUXILIARY TANK - LEFT WING
			06-114-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP
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AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

CENTER AUXILIARY TANK - LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-114-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE CENTER AUXILIARY TANK - RIGHT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 631		ACCESS PANELS 1331 631AB 631BB				

MECH	INSP	MPD ITEM NUMBER 0600-631-01Z
<p>*** CENTER AUXILIARY TANK - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 1331 REQUIRES DEFUELING AND PURGING THE AUXILIARY FUEL TANKS (CENTER WING, 2 BAY OR 4 BAY) PER MAINTENANCE MANUAL 28-11-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

0 2 0 5	EFFECTIVITY	GEN VISUAL	CENTER AUXILIARY TANK - RIGHT WING
			06-114-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

CENTER AUXILIARY TANK - RIGHT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-115-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/06
TASK GEN VISUAL		TITLE MAIN TANK (INBD OF RIB 10)-LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 532			ACCESS PANELS 5001 532AB 532BB 532DB 532EB 532FB			

MECH	INSP	MPD ITEM NUMBER 0600-532-01Z
<p>*** MAIN TANK (INBD OF RIB 10)-LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5001 REQUIRES DRAINING AND PURGING THE LEFT MAIN AND CENTER AUXILIARY FUEL TANKS PER MAINTENANCE MANUAL 28-11-00.</p>		

EFFECTIVITY	GEN VISUAL	MAIN TANK (INBD OF RIB 10)-LEFT WING
		06-115-01-1 PAGE 1 OF 1 AUG 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-115-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE MAIN TANK(INBD OF RIB 10)-RIGHT WING			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 632		ACCESS PANELS 6001 632AB 632BB 632DB 632EB 632FB				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** MAIN TANK (INBD OF RIB 10)-RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 6001 REQUIRES DRAINING AND PURGING THE RIGHT MAIN AND CENTER AUXILIARY FUEL TANKS PER MAINTENANCE MANUAL 28-11-00.</p>				0600-632-01Z

EFFECTIVITY	GEN VISUAL	MAIN TANK(INBD OF RIB 10)-RIGHT WING
		06-115-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-116-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TANK	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE DRY BAY - INBOARD LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 533			ACCESS PANELS 533AB			

MECH	INSP	MPD ITEM NUMBER 0600-533-01Z
<p>*** DRY BAY - INBOARD LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	GEN VISUAL	DRY BAY - INBOARD LEFT WING
		06-116-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):**
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.**
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.**

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EFFECTIVITY

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GEN VISUAL

DRY BAY - INBOARD LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-116-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TANK	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE DRY BAY - INBOARD RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 633			ACCESS PANELS 633AB			

MECH	INSP	MPD ITEM NUMBER 0600-633-01Z
<p>*** DRY BAY - INBOARD RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	GEN VISUAL	DRY BAY - INBOARD RIGHT WING
		06-116-01-2 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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2

EFFECTIVITY

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GEN VISUAL

DRY BAY - INBOARD RIGHT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-117-01-1
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	L WING TANK		4C	14848	018	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
GEN VISUAL		MAIN TANK (OB OF RIB 10) - LEFT WING				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
541		5001 541AB 541BB 541CB 541DB 541EB 541FB 541GB 541HB 541JB 541KB 541LB 541MB 541NB 541PB 541QB 541SB 541TB 541UB 541VB 541WB 541XB				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** MAIN TANK (OB OF RIB 10) - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5001 REQUIRES DRAINING AND PURGING THE LEFT MAIN AND CENTER AUXILIARY FUEL TANKS PER MAINTENANCE MANUAL 28-11-00.</p>				
		0600-541-01Z				

0 2 1 3	EFFECTIVITY	GEN VISUAL	MAIN TANK (OB OF RIB 10) - LEFT WING
			06-117-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-117-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 009	TASK CARD REVISION AUG 22/05					
TASK GEN VISUAL	TITLE MAIN TANK (OB OF RIB 10)- RIGHT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE						
ZONES 641			ACCESS PANELS								
			6001	641AB	641BB	641CB	641DB	641EB	641FB	641GB	641HB
			641JB	641KB	641LB	641MB	641NB	641PB	641QB	641SB	641TB
			641UB	641VB	641WB	641XB					

MECH	INSP	MPD ITEM NUMBER									
		<p>*** MAIN TANK (OB OF RIB 10)- RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 6001 REQUIRES DRAINING AND PURGING THE RIGHT MAIN AND CENTER AUXILIARY FUEL TANKS PER MAINTENANCE MANUAL 28-11-00.</p>									
		0600-641-01Z									

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EFFECTIVITY	GEN VISUAL	MAIN TANK (OB OF RIB 10)- RIGHT WING
		06-117-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-118-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE SURGE TANK - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 542			ACCESS PANELS 5401 542AB 542BB 542CB			

MECH	INSP	MPD ITEM NUMBER 0600-542-01Z
		<p>*** SURGE TANK - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5401 REQUIRES OPENING THE LEFT SURGE TANK PER MAINTENANCE MANUAL 28-11-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 2 1 5	EFFECTIVITY	GEN VISUAL	SURGE TANK - LEFT WING
			06-118-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

SURGE TANK - LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-118-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TANK	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE SURGE TANK - RIGHT WING	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 642		ACCESS PANELS 6401 642AB 642BB 642CB				

MECH	INSP	MPD ITEM NUMBER 0600-642-01Z
<p>*** SURGE TANK - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 6401 REQUIRES OPENING THE RIGHT SURGE TANK PER MAINTENANCE MANUAL 28-11-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

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EFFECTIVITY	GEN VISUAL	SURGE TANK - RIGHT WING
		06-118-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

SURGE TANK - RIGHT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-119-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA LEFT WING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING TIP - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 544			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-544-01Z
		<p>*** WING TIP - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

EFFECTIVITY	GEN VISUAL	WING TIP - LEFT WING
		06-119-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	WING TIP - LEFT WING
			06-119-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-119-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA RIGHT WING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING TIP - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 644			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-644-01Z	
		<p>*** WING TIP - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>	

EFFECTIVITY	GEN VISUAL	WING TIP - RIGHT WING
		06-119-01-2 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	WING TIP - RIGHT WING
			06-119-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-120-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE REAR SPAR TO MLG SUPPORT BEAM-L WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 551			ACCESS PANELS 551TB 551UB			

MECH	INSP	MPD ITEM NUMBER 0600-551-01Z	
		<p>*** REAR SPAR TO MLG SUPPORT BEAM-L WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	REAR SPAR TO MLG SUPPORT BEAM-L WING
		06-120-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-120-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE REAR SPAR TO MLG SUPPORT BEAM-R WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 651		ACCESS PANELS 651TB 651UB				

MECH	INSP	MPD ITEM NUMBER 0600-651-01Z			
		<p>*** REAR SPAR TO MLG SUPPORT BEAM-R WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

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EFFECTIVITY	GEN VISUAL	REAR SPAR TO MLG SUPPORT BEAM-R WING
		06-120-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-121-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK W-06-123-01-1	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE L MLG SUPT BEAM & REAR SPAR TO TE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 552			ACCESS PANELS 552AB 552CB 552HB 561AB 561BB 561CB			

MECH	INSP	MPD ITEM NUMBER 0600-552-01Z	
		<p>*** L MLG SUPT BEAM & REAR SPAR TO TE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	L MLG SUPT BEAM & REAR SPAR TO TE
		06-121-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-121-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK W-06-123-01-2	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE R MLG SUPT BEAM & REAR SPAR TO TE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 652			ACCESS PANELS 652AB 652CB 652HB 661AB 661BB 661CB			

MECH	INSP	MPD ITEM NUMBER 0600-652-01Z
		<p>*** R MLG SUPT BEAM & REAR SPAR TO TE PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>

EFFECTIVITY	GEN VISUAL	R MLG SUPT BEAM & REAR SPAR TO TE
		06-121-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-122-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE INBOARD TRAILING EDGE FLAP - L WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 555			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-555-01Z
		<p>*** INBOARD TRAILING EDGE FLAP - L WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 2 2 7	EFFECTIVITY	GEN VISUAL	INBOARD TRAILING EDGE FLAP - L WING
			06-122-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

INBOARD TRAILING EDGE FLAP - L WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-122-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE INBOARD TRAILING EDGE FLAP - R WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 655			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-655-01Z
		<p>*** INBOARD TRAILING EDGE FLAP - R WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 2 2 9	EFFECTIVITY	GEN VISUAL	INBOARD TRAILING EDGE FLAP - R WING
			06-122-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	INBOARD TRAILING EDGE FLAP - R WING
		06-122-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-123-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK W-06-121-01-1	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE INBOARD AILERON - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 556			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-556-01Z
		<p>*** INBOARD AILERON - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (AILERON LEADING EDGE AREA ONLY) IN CONJUNCTION WITH ZONAL TASK 0600-552-01Z.</p>

EFFECTIVITY	GEN VISUAL	INBOARD AILERON - LEFT WING
		06-123-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-123-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK W-06-121-01-2	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE INBOARD AILERON - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 656			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-656-01Z	
		<p>*** INBOARD AILERON - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (AILERON LEADING EDGE AREA ONLY) IN CONJUNCTION WITH ZONAL TASK 0600-652-01Z.</p>	

EFFECTIVITY	GEN VISUAL	INBOARD AILERON - RIGHT WING
		06-123-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-124-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK W-06-125-01-1	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE REAR SPAR TO WING TE - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 561			ACCESS PANELS 5002 5003 561LB 561MB 561NB 561RBX			

MECH	INSP	MPD ITEM NUMBER 0600-561-01Z
<p>*** REAR SPAR TO WING TE - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5003 REQUIRES RAISING THE SPOILERS PER MAINTENANCE MANUAL 27-61-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 		

EFFECTIVITY	GEN VISUAL	REAR SPAR TO WING TE - LEFT WING
		06-124-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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4

EFFECTIVITY

GEN VISUAL

REAR SPAR TO WING TE - LEFT WING

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-124-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK W-06-125-01-2	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE REAR SPAR TO WING TE - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 661			ACCESS PANELS 5002 5003 661LB 661MB 661NB 661RBX			

MECH	INSP		MPD ITEM NUMBER
		<p>*** REAR SPAR TO WING TE - RIGHT WING</p> <p>PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>ACCESS NOTE: SPECIAL ACCESS 5003 REQUIRES RAISING THE SPOILERS PER MAINTENANCE MANUAL 27-61-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 	0600-661-01Z

EFFECTIVITY		GEN VISUAL	REAR SPAR TO WING TE - RIGHT WING
			06-124-01-2 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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6

EFFECTIVITY

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GEN VISUAL

REAR SPAR TO WING TE - RIGHT WING

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-125-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK W-06-124-01-1	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE OUTBOARD AILERON - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 567			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-567-01Z
		<p>*** OUTBOARD AILERON - LEFT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (AILERON LEADING EDGE AREA ONLY) IN CONJUNCTION WITH ZONAL TASK 0600-561-01Z.</p>

EFFECTIVITY	GEN VISUAL	OUTBOARD AILERON - LEFT WING
		06-125-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-125-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK W-06-124-01-2	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/06
TASK GEN VISUAL		TITLE OUTBOARD AILERON - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 667			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-667-01Z
		<p>*** OUTBOARD AILERON - RIGHT WING PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE (AILERON LEADING EDGE AREA ONLY) IN CONJUNCTION WITH ZONAL TASK 0600-661-01Z.</p>

EFFECTIVITY	GEN VISUAL	OUTBOARD AILERON - RIGHT WING
		06-125-01-2 PAGE 1 OF 1 APR 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-126-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD AILERON - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 567			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-567-02Z
<p>*** OUTBOARD AILERON - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

0 2 3 9	EFFECTIVITY	GEN VISUAL	OUTBOARD AILERON - LEFT WING
			06-126-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD AILERON - LEFT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-126-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OUTBOARD AILERON - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 667			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-667-02Z
<p>*** OUTBOARD AILERON - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

0 2 4 1	EFFECTIVITY	GEN VISUAL	OUTBOARD AILERON - RIGHT WING
			06-126-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
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- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OUTBOARD AILERON - RIGHT WING

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-127-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OB FLAP SUPPORT FAIRING - L IB FLAP		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 571			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-571-01Z
<p>*** OB FLAP SUPPORT FAIRING - L IB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

EFFECTIVITY	GEN VISUAL	OB FLAP SUPPORT FAIRING - L IB FLAP
		06-127-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
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 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	OB FLAP SUPPORT FAIRING - L IB FLAP
			06-127-01-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-127-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OB FLAP SUPPORT FAIRING - R IB FLAP		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 671			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-671-01Z
		<p>*** OB FLAP SUPPORT FAIRING - R IB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

EFFECTIVITY	GEN VISUAL	OB FLAP SUPPORT FAIRING - R IB FLAP
		06-127-01-2 PAGE 1 OF 2 APR 22/08

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MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OB FLAP SUPPORT FAIRING - R IB FLAP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-128-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE IB FLAP SUPPORT FAIRING - L OB FLAP		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 572			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-572-01Z
		<p>*** IB FLAP SUPPORT FAIRING - L OB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

0 2 4 7	EFFECTIVITY	GEN VISUAL	IB FLAP SUPPORT FAIRING - L OB FLAP
			06-128-01-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
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 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

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 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

IB FLAP SUPPORT FAIRING - L OB FLAP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-128-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE IB FLAP SUPPORT FAIRING - R OB FLAP		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 672			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-672-01Z
<p>*** IB FLAP SUPPORT FAIRING - R OB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

0 2 4 9	EFFECTIVITY	GEN VISUAL	IB FLAP SUPPORT FAIRING - R OB FLAP
			06-128-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

IB FLAP SUPPORT FAIRING - R OB FLAP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-129-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OB FLAP SUPPORT FAIRING - L OB FLAP		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 573			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-573-01Z
		<p>*** OB FLAP SUPPORT FAIRING - L OB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE

EFFECTIVITY	GEN VISUAL	OB FLAP SUPPORT FAIRING - L OB FLAP
		06-129-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OB FLAP SUPPORT FAIRING - L OB FLAP

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-129-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING TE	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OB FLAP SUPPORT FAIRING - R OB FLAP		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 673			ACCESS PANELS 5002			

MECH	INSP	MPD ITEM NUMBER 0600-673-01Z
<p>*** OB FLAP SUPPORT FAIRING - R OB FLAP PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 5002 REQUIRES EXTENDING THE TRAILING EDGE FLAPS PER MAINTENANCE MANUAL 27-51-00.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

EFFECTIVITY	GEN VISUAL	OB FLAP SUPPORT FAIRING - R OB FLAP
		06-129-01-2 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

OB FLAP SUPPORT FAIRING - R OB FLAP

06-129-01-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-130-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA NOSE GEAR	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE NOSE LANDING GEAR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 711			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-711-01Z	
		<p>*** NOSE LANDING GEAR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	NOSE LANDING GEAR
		06-130-01 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-131-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L MAIN GEAR	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE LEFT MAIN LANDING GEAR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 731			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-731-01Z	
		<p>*** LEFT MAIN LANDING GEAR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

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EFFECTIVITY	GEN VISUAL	LEFT MAIN LANDING GEAR
		06-131-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-131-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R MAIN GEAR	RELATED TASK 1C	INTERVAL	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE RIGHT MAIN LANDING GEAR	STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL			
ZONES 741	ACCESS PANELS					

MECH	INSP	MPD ITEM NUMBER 0600-741-01Z
		<p>*** RIGHT MAIN LANDING GEAR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>

EFFECTIVITY	GEN VISUAL	RIGHT MAIN LANDING GEAR
		06-131-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-132-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE BULK CARGO COMPARTMENT DOOR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 811		ACCESS PANELS 8111				

MECH	INSP	MPD ITEM NUMBER 0600-811-01Z
		<p>*** BULK CARGO COMPARTMENT DOOR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE. CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: SPECIAL ACCESS 8111 REQUIRES REMOVAL OF THE BULK CARGO DOOR LINING/INSULATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

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EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT DOOR
		06-132-01 PAGE 1 OF 2 APR 22/08

MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

BULK CARGO COMPARTMENT DOOR

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-133-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 009	TASK CARD REVISION APR 22/06
TASK GEN VISUAL		TITLE BULK CARGO COMPARTMENT DOOR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 811			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-811-02Z				
		<p>*** BULK CARGO COMPARTMENT DOOR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p>				

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EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT DOOR	06-133-01	PAGE 1 OF 1	APR 22/06
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-134-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 009	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE FORWARD CARGO COMPARTMENT DOOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 821			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-821-01Z
		<p>*** FORWARD CARGO COMPARTMENT DOOR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p>

EFFECTIVITY	GEN VISUAL	FORWARD CARGO COMPARTMENT DOOR
		06-134-01 PAGE 1 OF 1 APR 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-135-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FORWARD CARGO COMPARTMENT DOOR	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 821		ACCESS PANELS 8211 821A 821B 821C 821D 821E 821F 821G 821H 821J 821K				

MECH	INSP	MPD ITEM NUMBER 0600-821-02Z
<p>*** FORWARD CARGO COMPARTMENT DOOR PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>AIRPLANE NOTE: AIRPLANES WITH LARGE FORWARD CARGO DOOR.</p> <p>CPC NOTE: CORROSION PREVENTION AND CONTROL BASIC TASK REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 8211 REQUIRES REMOVAL OF THE FORWARD CARGO DOOR LINING/INSULATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

EFFECTIVITY	GEN VISUAL	FORWARD CARGO COMPARTMENT DOOR
		06-135-02 PAGE 1 OF 2 APR 22/08

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MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

FORWARD CARGO COMPARTMENT DOOR

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-136-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 010	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE AFT CARGO COMPARTMENT DOOR	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 822		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER 0600-822-01Z
		<p>*** AFT CARGO COMPARTMENT DOOR PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p>

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EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT DOOR
		06-136-01 PAGE 1 OF 1 APR 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-137-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE AFT CARGO COMPARTMENT DOOR	STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY AIRPLANE ENGINE	
ZONES 822		ACCESS PANELS 822 8221 822A 822B 822C 822D 822E 822F				
MECH	INSP	MPD ITEM NUMBER				

*** AFT CARGO COMPARTMENT DOOR
PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.

AIRPLANE NOTE: APPLICABLE TO PASS AND SF AIRPLANES.

CPC NOTE: CORROSION PREVENTION AND CONTROL BASIC TASK
REQUIRED.

ACCESS NOTE: SPECIAL ACCESS 8221 REQUIRES REMOVAL OF
THE AFT CARGO DOOR LINING/INSULATION.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH
DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL
REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC
(E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO
ACCOMPLISH STEP 3).
IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE
TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE
BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH
STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE
CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR
INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G.,
GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED
INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION.
ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS
FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE
ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS,
CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL
DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE
FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL
FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE
PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

0600-822-02Z

EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT DOOR
		06-137-01 PAGE 1 OF 2 APR 22/08

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MECH	INSP	
		<p>6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):</p> <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23). <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35). <p>B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):</p> <ul style="list-style-type: none"> - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II) <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35). <p>C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:</p> <ul style="list-style-type: none"> - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS) - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1) - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED) - AREAS WITH ELECTRICAL ARC POTENTIAL - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES) - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

0 2 6 6	EFFECTIVITY	GEN VISUAL	AFT CARGO COMPARTMENT DOOR
			06-137-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-138-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ENTRY DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 011	TASK CARD REVISION APR 22/06
TASK GEN VISUAL	TITLE PASS CABIN DOOR - FORWARD ENTRY		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 831		ACCESS PANELS 831				

MECH	INSP	MPD ITEM NUMBER 0600-831-01Z
		<p>*** PASS CABIN DOOR - FORWARD ENTRY PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p>

EFFECTIVITY	GEN VISUAL	PASS CABIN DOOR - FORWARD ENTRY
		06-138-01 PAGE 1 OF 1 APR 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-139-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXIT - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 832			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-832-01Z
		<p>*** OVERWING EMERGENCY EXIT - LEFT PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>AIRPLANE NOTE: 767-200 AIRPLANES WITH SINGLE OVERWING EMERGENCY EXIT CONFIGURATION AND 200SF AIRPLANES WITH DEACTIVATED SINGLE OVERWING EMERGENCY EXIT CONFIGURATION.</p>

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EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXIT - LEFT
		06-139-01-1 PAGE 1 OF 1 AUG 22/07

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-139-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXIT - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 842			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 0600-842-01Z
		<p>*** OVERWING EMERGENCY EXIT - RIGHT PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>AIRPLANE NOTE: 767-200 AIRPLANES WITH SINGLE OVERWING EMERGENCY EXIT CONFIGURATION, AND 767 SF AIRPLANES WITH DEACTIVATED SINGLE OVERWING EMERGENCY EXIT CONFIGURATION.</p>

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EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXIT - RIGHT
		06-139-01-2 PAGE 1 OF 1 AUG 22/07

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-139-02-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXITS - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 832 834		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** OVERWING EMERGENCY EXITS - LEFT PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF THE ZONES.</p> <p>AIRPLANE NOTE: DUAL OVERWING EMERGENCY EXIT CONFIGURATION.</p>			<p>0600-832-01Z 0600-834-01Z</p>	

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EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - LEFT
		06-139-02-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-139-02-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION AUG 22/06
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXITS - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE	
ZONES 842 844			ACCESS PANELS			
NOTE	ALL					

MECH	INSP	MPD ITEM NUMBER				
		<p>*** OVERWING EMERGENCY EXITS - RIGHT. PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF THE ZONES.</p> <p>AIRPLANE NOTE: DUAL OVERWING EMERGENCY EXIT CONFIGURATION.</p>				<p>0600-842-01Z 0600-844-01Z</p>

EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - RIGHT
		06-139-02-2 PAGE 1 OF 1 AUG 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-140-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXIT - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 832		ACCESS PANELS 832 8321				

MECH	INSP	MPD ITEM NUMBER 0600-832-02Z
		<p>*** OVERWING EMERGENCY EXIT - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>AIRPLANE NOTE: 767-200 AIRPLANES WITH SINGLE OVERWING EMERGENCY EXIT CONFIGURATION AND 200SF AIRPLANES WITH DEACTIVATED SINGLE OVERWING EMERGENCY EXIT CONFIGURATION.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXIT - LEFT
		06-140-01-1 PAGE 1 OF 2 APR 22/08

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MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OVERWING EMERGENCY EXIT - LEFT

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-140-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE OVERWING EMERGENCY EXIT - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 842		ACCESS PANELS 8321 842				

MECH	INSP	MPD ITEM NUMBER 0600-842-02Z
		<p>*** OVERWING EMERGENCY EXIT - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>AIRPLANE NOTE: 767-200 AIRPLANES WITH SINGLE OVERWING EMERGENCY EXIT CONFIGURATION, AND 767 SF AIRPLANES WITH DEACTIVATED SINGLE OVERWING EMERGENCY EXIT CONFIGURATION.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

0 2 7 4	EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXIT - RIGHT
			06-140-01-2 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

OVERWING EMERGENCY EXIT - RIGHT

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-140-02-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OVERWING EMERGENCY EXITS - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 832 834			ACCESS PANELS 832 8321 834			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** OVERWING EMERGENCY EXITS - LEFT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF THE ZONES.</p> <p>AIRPLANE NOTE: DUAL OVERWING EMERGENCY EXIT CONFIGURATION.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 	<p>0600-832-02Z 0600-834-02Z</p>

0 2 7 6	EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - LEFT
			06-140-02-1 PAGE 1 OF 2 APR 22/08

MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - LEFT
		06-140-02-1 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-140-02-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA EMERG DOORS	RELATED TASK	INTERVAL 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE OVERWING EMERGENCY EXITS - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 842 844			ACCESS PANELS 8321 842 844			

MECH	INSP	MPD ITEM NUMBER
		<p>*** OVERWING EMERGENCY EXITS - RIGHT PERFORM INTERNAL GENERAL VISUAL INSPECTION OF THE ZONES.</p> <p>0600-842-02Z 0600-844-02Z</p> <p>AIRPLANE NOTE: DUAL OVERWING EMERGENCY EXIT CONFIGURATION.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

0 2 7 8	EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - RIGHT
			06-140-02-2 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	OVERWING EMERGENCY EXITS - RIGHT
		06-140-02-2 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-141-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ENTRY DOORS	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 011	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL	TITLE PASSENGER CABIN DOOR - AFT ENTRY	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL		
ZONES 833		ACCESS PANELS 833				

MECH	INSP	MPD ITEM NUMBER 0600-833-01Z				
		<p>*** PASSENGER CABIN DOOR - AFT ENTRY PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p> <p>INTERVAL NOTE: 1C (PASS), 4C (SF).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS AND SF AIRPLANES.</p> <p>NOTE: FOR SF AIRPLANES, INSPECTION IS DONE WITH DEACTIVATED DOORS PARTIALLY OPEN.</p>				

0 2 8 0	EFFECTIVITY	GEN VISUAL	PASSENGER CABIN DOOR - AFT ENTRY
			06-141-01 PAGE 1 OF 1 AUG 22/07

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-142-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ENTRY DOORS	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 011	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL	TITLE PASSENGER CABIN DOOR-FORWARD SERVICE	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL		
ZONES 841		ACCESS PANELS 841				

MECH	INSP	MPD ITEM NUMBER 0600-841-01Z
		<p>*** PASSENGER CABIN DOOR-FORWARD SERVICE PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p> <p>INTERVAL NOTE: 1C (PASS), 4C (SF).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS AND SF AIRPLANES.</p> <p>NOTE: FOR SF AIRPLANES, INSPECTION IS DONE WITH DEACTIVATED DOORS PARTIALLY OPEN.</p>

0 2 8 1	EFFECTIVITY	GEN VISUAL	PASSENGER CABIN DOOR-FORWARD SERVICE
			06-142-01 PAGE 1 OF 1 AUG 22/07

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-143-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ENTRY DOORS	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 011	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL	TITLE PASSENGER CABIN DOOR - AFT SERVICE	STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL		
ZONES 843		ACCESS PANELS 843				

MECH	INSP	MPD ITEM NUMBER 0600-843-01Z				
		<p>*** PASSENGER CABIN DOOR - AFT SERVICE PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE (WITH DOOR PARTIALLY OPEN).</p> <p>INTERVAL NOTE: 1C (PASS), 4C (SF).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS AND SF AIRPLANES.</p> <p>NOTE: FOR SF AIRPLANES, INSPECTION IS DONE WITH DEACTIVATED DOORS PARTIALLY OPEN.</p>				

0 2 8 2	EFFECTIVITY	GEN VISUAL	PASSENGER CABIN DOOR - AFT SERVICE
			06-143-01 PAGE 1 OF 1 AUG 22/07

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-144-01
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA L WING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE KRUEGER FLAP - LEFT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 513			ACCESS PANELS NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-513-01Z
<p>*** KRUEGER FLAP - LEFT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: LE FLAP EXTENDED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

EFFECTIVITY	GEN VISUAL	KRUEGER FLAP - LEFT WING
		06-144-01 PAGE 1 OF 2 APR 22/08

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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	KRUEGER FLAP - LEFT WING
		06-144-01 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-144-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 1 THRUST REVERSERS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 415 416		ACCESS PANELS 415AL 416AR NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** THRUST REVERSERS - ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: PORTIONS OF THIS ZONAL TASK REQUIRE THE FAN DUCT COWL AND THRUST REVERSER HALVES TO BE CLOSED.</p>				<p>0600-415-01Z 0600-416-01Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 1 THRUST REVERSERS
		06-144-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

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TASK CARD

BOEING CARD NO. 06-144-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 2 THRUST REVERSERS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 425 426		ACCESS PANELS 425AL 426AR NOTE				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** THRUST REVERSERS - ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: PORTIONS OF THIS ZONAL TASK REQUIRE THE FAN DUCT COWL AND THRUST REVERSER HALVES TO BE CLOSED.</p>				<p>0600-425-01Z 0600-426-01Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 2 THRUST REVERSERS
		06-144-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-144-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA R WING	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE KRUEGER FLAP - RIGHT WING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 613			ACCESS PANELS NOTE			

MECH	INSP	MPD ITEM NUMBER 0600-613-01Z
		<p>*** KRUEGER FLAP - RIGHT WING PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p> <p>CPC NOTE: CORROSION PREVENTION & CONTROL BASIC TASK REQUIRED</p> <p>ACCESS NOTE: LE FLAP EXTENDED</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

0 2 8 7	EFFECTIVITY	GEN VISUAL	KRUEGER FLAP - RIGHT WING
			06-144-02 PAGE 1 OF 2 APR 22/08

MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	KRUEGER FLAP - RIGHT WING
		06-144-02 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-145-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK W-79-401-01-1	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE ENGINE 1		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 411			ACCESS PANELS 413AL 414AR 415AL 416AR 417AL 418AR			

MECH	INSP	MPD ITEM NUMBER	
		*** ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF THE ZONE.	0600-411-02Z

EFFECTIVITY	GEN VISUAL	ENGINE 1	06-145-01-1	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-145-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK W-79-401-01-2	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/07
TASK GEN VISUAL		TITLE ENGINE 2		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 421			ACCESS PANELS 423AL 424AR 425AL 426AR 427AL 428AR			

MECH	INSP	MPD ITEM NUMBER	
		*** ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.	0600-421-02Z

EFFECTIVITY	GEN VISUAL	ENGINE 2	06-145-01-2	PAGE 1 OF 1	AUG 22/07
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-146-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE ENGINE 1 FAN COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 413 414			ACCESS PANELS 413AL 414AR			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** FAN COWL PANELS - ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>	<p>0600-413-01Z 0600-414-01Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 1 FAN COWL PANELS	06-146-01-1 PAGE 1 OF 1 AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-146-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 2 FAN COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 423 424		ACCESS PANELS 423AL 424AR				

MECH	INSP	MPD ITEM NUMBER			
		*** FAN COWL PANELS - ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.			0600-423-01Z 0600-424-01Z

EFFECTIVITY	GEN VISUAL	ENGINE 2 FAN COWL PANELS	06-146-01-2	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-147-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 1 CORE COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 417 418		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER			
		*** CORE COWL PANELS - ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.			0600-417-01Z 0600-418-01Z

EFFECTIVITY	GEN VISUAL	ENGINE 1 CORE COWL PANELS	06-147-01-1	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-147-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 2 CORE COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 427 428		ACCESS PANELS				

MECH	INSP	MPD ITEM NUMBER			
		*** CORE COWL PANELS - ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.			0600-427-01Z 0600-428-01Z

EFFECTIVITY	GEN VISUAL	ENGINE 2 CORE COWL PANELS	06-147-01-2	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-148-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 1 FWD NACELLE STRUT FAIRING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 431		ACCESS PANELS 431AT 431BT 431DT 431ET				

MECH	INSP	MPD ITEM NUMBER 0600-431-01Z			
		<p>*** FWD NACELLE STRUT FAIRING - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	STRUT 1 FWD NACELLE STRUT FAIRING
		06-148-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-148-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 2 FWD NACELLE STRUT FAIRING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 441		ACCESS PANELS 441AT 441BT 441DT 441ET				

MECH	INSP	MPD ITEM NUMBER 0600-441-01Z			
		<p>*** FWD NACELLE STRUT FAIRING - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

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EFFECTIVITY	GEN VISUAL	STRUT 2 FWD NACELLE STRUT FAIRING
		06-148-01-2 PAGE 1 OF 1 AUG 22/05

STATION
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BOEING CARD NO. 06-149-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 1 CORE COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 417 418		ACCESS PANELS 417AL 418AR				

MECH	INSP	MPD ITEM NUMBER	
		<p>*** CORE COWL PANELS - ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>	<p>0600-417-02Z 0600-418-02Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 1 CORE COWL PANELS	06-149-01-1 PAGE 1 OF 1 AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-149-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE ENGINE 2 CORE COWL PANELS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 427 428			ACCESS PANELS 427AL 428AR			

MECH	INSP	MPD ITEM NUMBER			
		<p>*** CORE COWL PANELS - ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>			<p>0600-427-02Z 0600-428-02Z</p>

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EFFECTIVITY	GEN VISUAL	ENGINE 2 CORE COWL PANELS
		06-149-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-150-01-1
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE STRUT 1 UNDERWING FAIRING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 433			ACCESS PANELS 432FL			

MECH	INSP	MPD ITEM NUMBER 0600-433-01Z			
		<p>*** UNDERWING FAIRING - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

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EFFECTIVITY	GEN VISUAL	STRUT 1 UNDERWING FAIRING
		06-150-01-1 PAGE 1 OF 1 AUG 22/05

STATION
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BOEING CARD NO. 06-150-01-2
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 2 UNDERWING FAIRING		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 443		ACCESS PANELS 442FR				

MECH	INSP	MPD ITEM NUMBER 0600-443-01Z			
		<p>*** UNDERWING FAIRING - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	STRUT 2 UNDERWING FAIRING
		06-150-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-151-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE STRUT 1 MID AND AFT TORQUE BOXES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 434 436			ACCESS PANELS 434AL 436AL 436BL			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** MID AND AFT TORQUE BOXES - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>	<p>0600-434-01Z 0600-436-01Z</p>

EFFECTIVITY	GEN VISUAL	STRUT 1 MID AND AFT TORQUE BOXES
		06-151-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-151-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE STRUT 2 MID AND AFT TORQUE BOXES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 444 446			ACCESS PANELS 444AR 446AR 446BR			

MECH	INSP	MPD ITEM NUMBER	
		<p>*** MID AND AFT TORQUE BOXES - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p>	<p>0600-444-01Z 0600-446-01Z</p>

EFFECTIVITY	GEN VISUAL	STRUT 2 MID AND AFT TORQUE BOXES
		06-151-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-152-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 1 NOSE COWL		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 412		ACCESS PANELS 413AL 414AR				

MECH	INSP	MPD ITEM NUMBER 0600-412-02Z			
		<p>*** NOSE COWL - ENGINE 1 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	ENGINE 1 NOSE COWL	06-152-01-1	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-152-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE ENGINE 2 NOSE COWL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 422			ACCESS PANELS 423AL 424AR			

MECH	INSP		MPD ITEM NUMBER 0600-422-02Z
		<p>*** NOSE COWL - ENGINE 2 PERFORM EXTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>	

EFFECTIVITY	GEN VISUAL	ENGINE 2 NOSE COWL	06-152-01-2 PAGE 1 OF 1 AUG 22/05
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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-153-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE ENGINE 1 THRUST REVERSERS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 415 416		ACCESS PANELS 4151				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** THRUST REVERSERS - ENGINE 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 4151 REQUIRES DEPLOYMENT AND STOWAGE OF THE LEFT ENGINE THRUST REVERSER PER MAINTENANCE MANUAL 78-31-00.</p>				<p>0600-415-02Z 0600-416-02Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 1 THRUST REVERSERS
		06-153-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-153-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA ENGINE 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE ENGINE 2 THRUST REVERSERS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 425 426		ACCESS PANELS 4251				

MECH	INSP	MPD ITEM NUMBER				
		<p>*** THRUST REVERSERS - ENGINE 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 4251 REQUIRES DEPLOYMENT AND STOWAGE OF THE RIGHT ENGINE THRUST REVERSER PER MAINTENANCE MANUAL 78-31-00.</p>				<p>0600-425-02Z 0600-426-02Z</p>

EFFECTIVITY	GEN VISUAL	ENGINE 2 THRUST REVERSERS
		06-153-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-154-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 1 AFT NACELLE STRUT FAIRINGS			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 437		ACCESS PANELS 437AL 437AR 437BL 437BR 437DB				

MECH	INSP	MPD ITEM NUMBER 0600-437-01Z			
		<p>*** AFT NACELLE STRUT FAIRINGS - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	STRUT 1 AFT NACELLE STRUT FAIRINGS
		06-154-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 06-154-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 2 AFT NACELLE STRUT FAIRINGS			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 447		ACCESS PANELS 447AL 447AR 447BL 447BR 447DB				

MECH	INSP	MPD ITEM NUMBER 0600-447-01Z			
		<p>*** AFT NACELLE STRUT FAIRINGS - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE.</p>			

EFFECTIVITY	GEN VISUAL	STRUT 2 AFT NACELLE STRUT FAIRINGS
		06-154-01-2 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 06-155-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE STRUT 1 CORE COWL SKIRT FAIRINGS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 435			ACCESS PANELS 436CL 436CR			

MECH	INSP	MPD ITEM NUMBER 0600-435-01Z
		<p>*** CORE COWL SKIRT FAIRINGS - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE IN CONJUNCTION WITH ZONAL TASKS 0600-417-02Z, 0600-418-02Z.</p>

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EFFECTIVITY	GEN VISUAL	STRUT 1 CORE COWL SKIRT FAIRINGS
		06-155-01-1 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
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SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-155-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 2 CORE COWL SKIRT FAIRINGS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 445		ACCESS PANELS 446CL 446CR				

MECH	INSP	MPD ITEM NUMBER 0600-445-01Z				
		<p>*** CORE COWL SKIRT FAIRINGS - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE IN CONJUNCTION WITH ZONAL TASKS 0600-427-02Z, 0600-428-02Z.</p>				

0 3 1 0	EFFECTIVITY	GEN VISUAL	STRUT 2 CORE COWL SKIRT FAIRINGS
			06-155-01-2 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 06-156-01-1
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 1	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE STRUT 1 FORWARD TORQUE BOX		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 432			ACCESS PANELS 432CL			

MECH	INSP	MPD ITEM NUMBER 0600-432-01Z				
		<p>*** FORWARD TORQUE BOX - STRUT 1 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE IN CONJUNCTION WITH ZONAL TASK 0600-431-01Z.</p>				

EFFECTIVITY	GEN VISUAL	STRUT 1 FORWARD TORQUE BOX
		06-156-01-1 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 06-156-01-2
AIRLINE CARD NO.

SKILL ENGIN	WORK AREA STRUT 2	RELATED TASK	INTERVAL 1C	PHASE 11212	MPD REV 005	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL	TITLE STRUT 2 FORWARD TORQUE BOX		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL 4000	
ZONES 442		ACCESS PANELS 442CL				

MECH	INSP	MPD ITEM NUMBER 0600-442-01Z				
		<p>*** FORWARD TORQUE BOX - STRUT 2 PERFORM INTERNAL GENERAL VISUAL INSPECTION OF ZONE IN CONJUNCTION WITH ZONAL TASK 0600-441-01Z.</p>				

EFFECTIVITY	GEN VISUAL	STRUT 2 FORWARD TORQUE BOX
		06-156-01-2 PAGE 1 OF 1 AUG 22/05

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