

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-400-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE BODY FLOOR BEAMS BELOW LAVS/GALLEYS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 100		ACCESS PANELS 1001 1003				

MECH	INSP	MPD ITEM NUMBER 53-400-00	
<p>GENERAL VISUAL - INTERNAL: BODY FLOOR BEAMS BELOW AND ADJACENT TO GALLEYS AND LAVATORIES (AREA DEFINED BY EACH OPERATOR'S INTERIOR CONFIGURATION).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF THE INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 			

EFFECTIVITY	GEN VISUAL	BODY FLOOR BEAMS BELOW LAVS/GALLEYS
		53-400-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

BODY FLOOR BEAMS BELOW LAVS/GALLEYS

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-404-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA LWR FUSLAGE	RELATED TASK	INTERVAL 00015 MOS	NOTE	PHASE 110XX	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE CARGO DOOR CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE	
ZONES 100		ACCESS PANELS 122AR 154AR					

MECH	INSP	MPD ITEM NUMBER 53-404-00
<p>DETAILED - INTERNAL: CARGO DOOR BEARSTRAPS INSIDE THE CARGO HANDLING/LOADER CONTROL PANEL CUTOUTS.</p> <p>INTERVAL NOTE: OR 3,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: APPLICABLE TO AIRPLANES PRIOR TO LINE NUMBER 496 WHICH HAVE THE CARGO HANDLING/LOADER CONTROL PANEL INSTALLED. REFER TO SERVICE LETTER 767-SL-53-20 FOR ADDITIONAL INFORMATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 		

5 6 9 0	EFFECTIVITY	DETAILED IN	CARGO DOOR CUTOUTS
			53-404-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	DETAILED IN	CARGO DOOR CUTOUTS
		53-404-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-406-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1A	PHASE 10101	MPD REV 012	TASK CARD REVISION DEC 22/07
TASK GEN VISUAL	TITLE BODY EXTERNAL SKIN			STRUCTURAL ILLUSTRATION REFERENCE 1-1 1-2 1-3 2-4	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 100 200		ACCESS PANELS 811 821 822 831 833 835 836 837 838 841 843 845 846 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-406-00				
		<p>GENERAL VISUAL - EXTERNAL: DOOR STOPS, SILLS, AND FRAMES.</p> <p>ACCESS NOTE: ACCORDING TO EACH OPERATOR'S DOOR CONFIGURATION. NOT APPLICABLE TO DEACTIVATED DOORS ON SF AIRPLANES.</p>				

5 6 9 2	EFFECTIVITY	GEN VISUAL	BODY EXTERNAL SKIN
			53-406-00 PAGE 1 OF 1 DEC 22/07

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-408-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED EX		TITLE LARGE FWD CARGO DOOR HINGES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 100		ACCESS PANELS 821				

MECH	INSP	MPD ITEM NUMBER 53-408-00				
		<p>DETAILED - EXTERNAL: LARGE FORWARD CARGO DOOR HINGES.</p> <p>AIRPLANE NOTE: AIRPLANES WITH LARGE FORWARD CARGO DOOR.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>				

EFFECTIVITY	DETAILED EX	LARGE FWD CARGO DOOR HINGES	53-408-00	PAGE 1 OF 2	APR 22/08
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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED EX

LARGE FWD CARGO DOOR HINGES

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-410-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA LWR FUSLAGE	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 011	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE LWR BODY SKIN AT ANTENNA LOCATIONS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 100			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-410-00
<p>GENERAL VISUAL - EXTERNAL: EXTERNAL BODY SKIN IN AREAS OF ANTENNA LOCATIONS BETWEEN STA 92.5 AND STA 1582.</p> <p>INTERVAL NOTE: S 1C IF SERVICE LETTER 767-SL-51-23 IS ACCOMPLISHED OR 15 MONTHS IF SERVICE LETTER 767-SL-51-23 IS NOT ACCOMPLISHED.</p> <p>AIRPLANE NOTE: SERVICE LETTER 767-SL-51-23. TASK APPLICABLE TO AIRPLANE LINE NUMBERS PRIOR TO 448 UNTIL THE RECOMMENDATIONS OF THE SERVICE LETTER ARE INCORPORATED AT ALL APPLICABLE ANTENNA LOCATIONS OR UNTIL ALL APPLICABLE ANTENNAS HAVE BEEN REINSTALLED USING BMS 3-27 AS SPECIFIED IN THE MAINTENANCE MANUAL.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 		

5 6 9 5	EFFECTIVITY	GEN VISUAL	LWR BODY SKIN AT ANTENNA LOCATIONS
			53-410-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
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WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

LWR BODY SKIN AT ANTENNA LOCATIONS

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-412-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED EX		TITLE DOOR STOPS AND CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE 1-1 1-3 1-50 8-1		APPLICABILITY AIRPLANE ENGINE ALL ALL
ZONES 100		ACCESS PANELS 811 821 822				

MECH	INSP	MPD ITEM NUMBER 53-412-00
<p>DETAILED - EXTERNAL: DOOR STOPS AND VISIBLE ADJACENT CUT-OUT STRUCTURE WITHIN AIRCRAFT CONTOUR AT ALL CARGO DOOR LOCATIONS INCLUDING DOOR SILLS AND FRAME CHORDS.</p> <p>INTERVAL NOTE: S 1C OR 3,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

5 6 9 7	EFFECTIVITY	DETAILED EX	DOOR STOPS AND CUTOUTS
			53-412-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED EX

DOOR STOPS AND CUTOUTS

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-414-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY STA. 132.5 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE 1-4	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 111			ACCESS PANELS 111AL			

MECH	INSP	MPD ITEM NUMBER 53-414-00
		<p>DETAILED - INTERNAL: BODY STA. 132.5 BULKHEAD.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): DISPLACES MOISTURE (E.G., BMS 3-23). OR</p>

5 6 9 9	EFFECTIVITY	DETAILED IN	BODY STA. 132.5 BULKHEAD
			53-414-00 PAGE 1 OF 2 APR 22/08

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TASK CARD

BOEING CARD NO. 53-414-00
AIRLINE CARD NO.

MECH	INSP

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

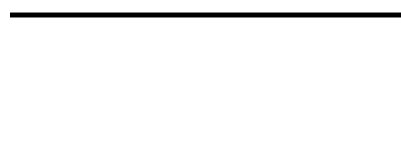
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

BODY STA. 132.5 BULKHEAD

53-414-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-416-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE FUSELAGE LOWER LOBE		STRUCTURAL ILLUSTRATION REFERENCE 1-5 1-6 1-8 1-9		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 113 114			ACCESS PANELS 1001 1003 113AL			

MECH	INSP	MPD ITEM NUMBER 53-416-00
		<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR FROM BS 132.5 TO BS 168.5/188.5 BELOW CREW CABIN FLOOR.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE
		53-416-00 PAGE 1 OF 2 APR 22/08

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6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

FUSELAGE LOWER LOBE
53-416-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-418-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD EQUIP BAY ACCESS DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE 1-8	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 113 114			ACCESS PANELS 1001 1003 113AL			

MECH	INSP	MPD ITEM NUMBER 53-418-00
<p>DETAILED - INTERNAL: BODY SKIN, DOUBLER, FRAMES, ADJACENT TO THE ACCESS DOOR CUTOUT INCLUDING FITTINGS AND BACK UP STRUCTURES, AND FORWARD EQUIPMENT BAY ACCESS DOOR INTERNAL STRUCTURE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

EFFECTIVITY	DETAILED IN	FWD EQUIP BAY ACCESS DOOR CUTOUT
		53-418-00 PAGE 1 OF 2 APR 22/08

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6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

FWD EQUIP BAY ACCESS DOOR CUTOUT

53-418-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-420-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 6C	PHASE S 2C	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE FUSELAGE EXTERIOR SURFACE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 111 113 114			ACCESS PANELS 111AL 113AL			

MECH	INSP	MPD ITEM NUMBER 53-420-00
<p>GENERAL VISUAL - EXTERNAL: FUSELAGE EXTERIOR SURFACES FROM BS 132.5 TO BS 168.5/188.5 BELOW CREW CABIN FLOOR, THE FORWARD PRESSURE BULKHEAD AT BS 132.5.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

5 7 0 5	EFFECTIVITY	GEN VISUAL	FUSELAGE EXTERIOR SURFACE
			53-420-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FUSELAGE EXTERIOR SURFACE
		53-420-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-428-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA NOSE W/W	RELATED TASK	INTERVAL S 1A	PHASE 10101	MPD REV 011	TASK CARD REVISION DEC 22/06
TASK GEN VISUAL	TITLE NOSE LANDING GEAR WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE 1-10		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 115 116		ACCESS PANELS 1004 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-428-00
		<p>GENERAL VISUAL - EXTERNAL: NOSE LANDING GEAR WHEEL WELL.</p> <p>ACCESS NOTE: AS VIEWED FROM THE GROUND.</p>

EFFECTIVITY	GEN VISUAL	NOSE LANDING GEAR WHEEL WELL
		53-428-00 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-430-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 007	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE AREA OUTBD & ABOVE NOSE WHEELWELL		STRUCTURAL ILLUSTRATION REFERENCE 1-7		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 117 118		ACCESS PANELS 1001 1003 113AL 119AL					

MECH	INSP	MPD ITEM NUMBER 53-430-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR BODY FLOOR BEAMS AND NOSE LANDING GEAR WHEEL WELL FLAT PRESSURE PANELS INCLUDING THE GEAR TRUNNION FITTING, THE AFT LOCK LINK FITTING, THE DRAG BRACE FITTING, AND THE ACTUATOR SUPPORT BEAM FITTING.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 		

EFFECTIVITY	GEN VISUAL	AREA OUTBD & ABOVE NOSE WHEELWELL
		53-430-00 PAGE 1 OF 2 APR 22/08

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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	AREA OUTBD & ABOVE NOSE WHEELWELL
		53-430-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-432-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 007	TASK CARD REVISION DEC 22/06
TASK DETAILED IN		TITLE BULKHEAD AND NOSE LANDING GEAR W/W		STRUCTURAL ILLUSTRATION REFERENCE 1-7 1-8		APPLICABILITY AIRPLANE ENGINE ALL ALL
ZONES 117 118			ACCESS PANELS 1001 1003 119AL			

MECH	INSP	MPD ITEM NUMBER 53-432-00
<p>DETAILED - INTERNAL: BODY SKIN, BULKHEAD AND NOSE LANDING GEAR WHEEL WELL SIDE PANEL AT INTERSECTION OF THE BODY STA 287 BULKHEAD AND THE AFT END OF THE NOSE LANDING GEAR WHEEL WELL SIDE PANEL.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>		

EFFECTIVITY	DETAILED IN	BULKHEAD AND NOSE LANDING GEAR W/W
		53-432-00 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-438-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD COMPT	RELATED TASK	INTERVAL	PHASE 99XXX	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY FLOOR BEAM - STA 287		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 118		ACCESS PANELS 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-438-00
<p>DETAILED - INTERNAL: FLOOR BEAMS AT BODY STA 287.</p> <p>INTERVAL NOTE: S 4C OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: AIRPLANES LINE NUMBER 1 TO 422.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

EFFECTIVITY	DETAILED IN	BODY FLOOR BEAM - STA 287
		53-438-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

BODY FLOOR BEAM - STA 287

53-438-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



SAS **BOEING**

 767

TASK CARD

BOEING CARD NO.
53-440-00
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL			S 4C	14848	012	APR 22/08
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
GEN VISUAL		FUSELAGE LOWER LOBE INTERIOR				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
119		1001 1003 119AL				

MECH	INSP	MPD ITEM NUMBER	
		53-440-00	
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR FLOOR BEAMS, SEAT TRACKS, BODY STA 287 BULKHEAD, BODY SKIN, STRINGERS, AND FRAMES FROM THE LOWER SURFACE OF THE CABIN FLOOR TO BL 0.0.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 			

EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE INTERIOR
		53-440-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE INTERIOR
		53-440-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-442-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL	PHASE 14040	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD ENTRY/SERVICE DOOR CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 119		ACCESS PANELS 1001 119AL NOTE				

MECH	INSP	MPD ITEM NUMBER 53-442-00
<p>DETAILED - INTERNAL: BODY SKIN, DOUBLER, STRINGERS, FRAMES AND LOWER SILL WITHIN 22 INCHES FORWARD AND AFT OF THE FORWARD ENTRY, CREW ENTRY AND SERVICE DOOR CUTOUTS BETWEEN STRINGER S-20 TO STRINGER S-23.</p> <p>INTERVAL NOTE: ACCOMPLISH INSPECTION ONLY WHEN AN OPPORTUNITY (REPAIR OR PERMANENT REMOVAL OF AIRPLANE FROM SERVICE) ARISES WHICH REQUIRES OR PERMITS REMOVAL OF EQUIPMENT/ ELECTRICAL PANELS NECESSARY TO GAIN ACCESS.</p> <p>ACCESS NOTE: ACCESS TO THIS AREA IS RESTRICTED DUE TO INSTALLATIONS AND INSPECTION SHOULD BE ACCOMPLISHED AS ACCESS PERMITS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED 		

EFFECTIVITY	DETAILED IN	FWD ENTRY/SERVICE DOOR CUTOUTS
		53-442-00 PAGE 1 OF 3 APR 22/08

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INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD

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EFFECTIVITY	DETAILED IN	FWD ENTRY/SERVICE DOOR CUTOUTS
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TASK CARD

BOEING CARD NO. 53-442-00
AIRLINE CARD NO.

MECH	INSP
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(INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

FWD ENTRY/SERVICE DOOR CUTOUTS

53-442-00

PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-444-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE MAIN EQUIPMENT CENTER ACCESS DOOR	STRUCTURAL ILLUSTRATION REFERENCE 1-12		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 119		ACCESS PANELS 119AL				

MECH	INSP	MPD ITEM NUMBER 53-444-00
<p>GENERAL VISUAL - INTERNAL: MAIN EQUIPMENT CENTER ACCESS DOOR OUTER FRAME, STOPS, AND LATCH.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 7 1 8	EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER ACCESS DOOR
			53-444-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER ACCESS DOOR
		53-444-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-446-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN	TITLE MAIN EQUIPMENT CENTER	STRUCTURAL ILLUSTRATION REFERENCE 1-11 1-13		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 119		ACCESS PANELS 1001 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-446-00
<p>DETAILED - INTERNAL: BODY SKIN, STRINGERS, AND FRAMES, FROM STRINGER S-36L TO S-36R AND FROM 10 INCHES FORWARD AND 10 INCHES AFT OF THE MAIN EQUIPMENT CENTER ACCESS DOOR AND LOWER PORTION OF BODY STA. 287 BULKHEAD BETWEEN LBL 40 AND RBL 40.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

5 7 2 0	EFFECTIVITY	DETAILED IN	MAIN EQUIPMENT CENTER
			53-446-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

MAIN EQUIPMENT CENTER

53-446-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-448-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE MAIN EQUIPMENT CENTER ACCESS DOOR	STRUCTURAL ILLUSTRATION REFERENCE 1-12		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 119		ACCESS PANELS 1191 119AL				

MECH	INSP	MPD ITEM NUMBER 53-448-00
<p>GENERAL VISUAL - INTERNAL: MAIN EQUIPMENT CENTER ACCESS DOOR INTERNAL STRUCTURE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1191 REQUIRES THE REMOVAL OF THE ELECTRICAL/ELECTRONICS BAY DOOR LINING/INSULATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER ACCESS DOOR
		53-448-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER ACCESS DOOR
		53-448-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-450-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL	PHASE 99XXX	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY FLOOR BEAMS - STA 287 TO 347		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 119		ACCESS PANELS 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-450-00
<p>DETAILED - INTERNAL: BODY FLOOR BEAMS BETWEEN BODY STA 287 AND STA 347.</p> <p>INTERVAL NOTE: S 4C OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: AIRPLANES LINE NUMBER 1 TO 422.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

EFFECTIVITY	DETAILED IN	BODY FLOOR BEAMS - STA 287 TO 347
		53-450-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

BODY FLOOR BEAMS - STA 287 TO 347

53-450-00

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-452-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	MPD REV 24848	TASK CARD REVISION 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE LOWER LOBE		STRUCTURAL ILLUSTRATION REFERENCE 1-42		APPLICABILITY AIRPLANE ALL		ENGINE ALL
ZONES 121		ACCESS PANELS 1001 1003 1211 1212 821					

MECH	INSP	MPD ITEM NUMBER 53-452-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 355 TO BS 738.5, BETWEEN S-20L AND S-32L.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE
		53-452-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE
		53-452-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-454-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	PHASE 24848	MPD REV 012	TASK CARD REVISION APR 22/08	
TASK GEN VISUAL		TITLE LOWER LOBE INTERIOR - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE 1-14 1-15 1-16 1-42			APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 122			ACCESS PANELS 1001 1002 1003 1005 1212 1221 1222 821					

MECH	INSP		MPD ITEM NUMBER 53-454-00
		<p>GENERAL VISUAL - INTERNAL: LOWER LOBE INTERIOR BS 355 TO BS 738.5, BETWEEN S-20R AND S-32R, INCLUDING FLOOR BEAMS, BODY SKIN, STRINGERS, SEAT TRACKS, AND FRAMES FROM LOWER SURFACE OF CABIN FLOOR TO UPPER SURFACE OF CARGO FLOOR.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 	

5 7 2 8	EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - RIGHT
		53-454-00	PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - RIGHT
		53-454-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-456-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION DEC 22/08
TASK DETAILED IN		TITLE FORWARD CARGO DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 122		ACCESS PANELS 1223 1231 821				

MECH	INSP	MPD ITEM NUMBER 53-456-00
<p>DETAILED - INTERNAL: FORWARD CARGO DOOR CUTOUT LOWER MAIN SILL UPPER AND LOWER SURFACE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	DETAILED IN	FORWARD CARGO DOOR LOWER SILL
		53-456-00 PAGE 1 OF 2 DEC 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	FORWARD CARGO DOOR LOWER SILL
			53-456-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-458-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL 00060 MOS	NOTE	PHASE 140XX	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD CARGO COMPARTMENT DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE	
ZONES 122		ACCESS PANELS 1003 1005 1221 821					

MECH	INSP	MPD ITEM NUMBER 53-458-00
<p>DETAILED - INTERNAL: EDGE FRAMES OF THE FORWARD CARGO COMPARTMENT DOOR CUTOUT.</p> <p>INTERVAL NOTE: OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>NOTE: HISTORICALLY, BOEING HAS NOTED THIS AREA TO BE SUSCEPTIBLE TO CORROSION. SHOULD CORROSION BE DETECTED DURING THE PERFORMANCE OF THIS TASK/INSPECTION, CONSULT THE APPROPRIATE CORRECTIVE ACTION IN THE CORROSION PREVENTION PROCEDURES IN THE MODEL APPROPRIATE AIRCRAFT MAINTENANCE MANUAL (AMM). CORROSION INHIBITING COMPOUNDS IN THIS AREA SHOULD ALSO BE MAINTAINED IN ACCORDANCE WITH THE PROCEDURES IN THE AMM.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1221 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS 		

EFFECTIVITY	DETAILED IN	FWD CARGO COMPARTMENT DOOR CUTOUT
		53-458-00 PAGE 1 OF 3 APR 22/08

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FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)

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EFFECTIVITY	DETAILED IN	FWD CARGO COMPARTMENT DOOR CUTOUT
		53-458-00 PAGE 2 OF 3 APR 22/08

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-458-00
AIRLINE CARD NO.

MECH	INSP

- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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			53-458-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-460-00
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION	
AIRPL	FUSELAGE		S 4C	14848	014	APR 22/08	
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY	
DETAILED IN		BODY BELOW FWD CARGO COMPART FLOOR		1-17 1-18		AIRPLANE ENGINE	
						ALL ALL	
ZONES			ACCESS PANELS				
123 124			1005 1211 1221 1223 1231 1232 821 NOTE				

MECH	INSP	MPD ITEM NUMBER
		53-460-00
<p>DETAILED - INTERNAL: LOWER LOBE INTERIOR BODY SKIN, STRINGERS, KEEL BEAM EXTENSION, ROLLER TRAYS, AND FRAMES FROM UPPER SURFACE OF CARGO FLOOR TO THE LOWEST BODY SKIN INCLUDING AREA WITHIN 22 INCHES OF THE FORWARD CARGO COMPARTMENT DOOR CUTOUT LOWER MAIN SILL.</p> <p>ACCESS NOTE: NO PERMANENTLY ATTACHED STRUCTURE IS TO BE REMOVED TO ACCOMPLISH THIS INSPECTION. SEVERAL CONFIGURATIONS OF ELECTRONICS BAY COOLING SKIN HEAT EXCHANGERS HAVE BEEN INSTALLED IN THIS AREA. THESE HEAT EXCHANGERS COVER VARIOUS STRUCTURES IN THIS AREA. AS AN ALTERNATIVE TO REMOVAL OF THE HEAT EXCHANGER, ACCESS FOR THIS INSPECTION MAY BE GAINED BY REMOVING THE LOWER PLENUM AND BY LOOSENING AND LIFTING THE LOWER EDGE OF THE HEAT EXCHANGER. VIEWING AIDS WILL BE REQUIRED IF USING THIS TECHNIQUE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1231 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1232 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT BALL TRANSFER PANEL PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL</p>		

EFFECTIVITY		DETAILED IN	BODY BELOW FWD CARGO COMPART FLOOR
			53-460-00 PAGE 1 OF 3 APR 22/08

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REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION

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EFFECTIVITY	_____	DETAILED IN	BODY BELOW FWD CARGO COMPART FLOOR
			53-460-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-460-00
AIRLINE CARD NO.

MECH	INSP

COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED
 IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	BODY BELOW FWD CARGO COMPART FLOOR
			53-460-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-464-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 014	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY BELOW FWD CARGO FLOOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 123 124			ACCESS PANELS 1005 1231 1232 821 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-464-00
<p>DETAILED - INTERNAL: FRAME UPPER CHORDS AT NUT PLATE LOCATIONS BELOW THE CARGO FLOOR.</p> <p>INTERVAL NOTE: S 4C OR 12,000 CYCLES,WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: AIRPLANES LINE NUMBER 1 TO 383.</p> <p>ACCESS NOTE: NO PERMANENTLY ATTACHED STRUCTURE IS TO BE REMOVED TO ACCOMPLISH THIS INSPECTION.</p> <p>OF THIS TASK/INSPECTION, CONSULT THE APPROPRIATE CORRECTIVE ACTION IN THE CORROSION PREVENTION PROCEDURES IN THE MODEL APPROPRIATE AIRCRAFT MAINTENANCE MANUAL (AMM). CORROSION INHIBITING COMPOUNDS IN THIS AREA SHOULD ALSO BE MAINTAINED IN ACCORDANCE WITH THE PROCEDURES IN THE AMM.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1231 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1232 REQUIRES REMOVAL OF THE FORWARD CARGO COMPARTMENT BALL TRANSFER PANEL PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <p>1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE</p>		

EFFECTIVITY	DETAILED IN	BODY BELOW FWD CARGO FLOOR
		53-464-00 PAGE 1 OF 3 APR 22/08

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MECH	INSP

- BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)

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EFFECTIVITY	_____	DETAILED IN	BODY BELOW FWD CARGO FLOOR
			53-464-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-464-00
AIRLINE CARD NO.

MECH	INSP

- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

BODY BELOW FWD CARGO FLOOR

53-464-00

PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-472-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FWD CARGO	RELATED TASK	INTERVAL 00060 MOS	NOTE	PHASE 140XX	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD LARGE CARGO DOOR CUTOUT			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 124			ACCESS PANELS 1231 821 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-472-00
<p>DETAILED - INTERNAL: FAIL-SAFE STRAP FOR LOWER MAIN SILL INNER CHORD OF THE LARGE FORWARD CARGO DOOR CUTOUT.</p> <p>INTERVAL NOTE: OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: AIRPLANES WITHIN LINE NUMBER 1 TO 377 WITH LARGE FORWARD CARGO DOOR.</p> <p>ACCESS NOTE: NO PERMANENTLY ATTACHED STRUCTURE IS TO BE REMOVED TO ACCOMPLISH THIS INSPECTION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1231 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, 		

5 7 4 1	EFFECTIVITY	DETAILED IN	FWD LARGE CARGO DOOR CUTOUT
			53-472-00 PAGE 1 OF 3 APR 22/08

MECH	INSP
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- CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

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EFFECTIVITY

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DETAILED IN

FWD LARGE CARGO DOOR CUTOUT

SAS



767

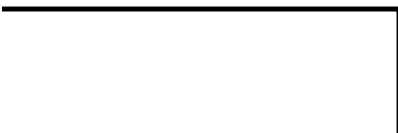
TASK CARD

BOEING CARD NO. 53-472-00
AIRLINE CARD NO.

MECH	INSP	
		7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

FWD LARGE CARGO DOOR CUTOUT

53-472-00

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-474-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE AREA AFT OF FWD CARGO COMPARTMENT	STRUCTURAL ILLUSTRATION REFERENCE 1-19			APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 125 126		ACCESS PANELS 1001 1003 1251 821					

MECH	INSP	MPD ITEM NUMBER 53-474-00
		<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR (BS 738.5 TO BS 785.9, S-20 TO BL-0.0), BODY FLOOR BEAMS, SEAT TRACKS, BODY SKIN, BULKHEADS, STRINGERS, FRAMES, AND OVERWING AND UNDERWING LONGERON EXTENSIONS FROM LOWER SURFACE OF CABIN FLOOR TO STRINGER S-35.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL

EFFECTIVITY	GEN VISUAL	AREA AFT OF FWD CARGO COMPARTMENT
		53-474-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY GEN VISUAL AREA AFT OF FWD CARGO COMPARTMENT

STATION
TAIL NO.
DATE

SAS **BOEING**
767
TASK CARD

BOEING CARD NO. 53-476-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE AREA AFT OF FWD CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE 1-17 1-19		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 125 126			ACCESS PANELS 1001 1003 1005 1251 821			

MECH	INSP		MPD ITEM NUMBER 53-476-00
		<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR, BS 738.5 TO BS 785.9, BETWEEN S-32 AND BL 0.0, INCLUDING SKINS, STRINGERS, FRAMES, BULKHEADS AND LONGERONS, LONGITUDINAL LAP SPLICES, CIRCUMFERENTIAL JOINTS, AND FRONT SPAR BULKHEAD LOWER WING CHORD.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL 	

EFFECTIVITY		GEN VISUAL	AREA AFT OF FWD CARGO COMPARTMENT
			53-476-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

AREA AFT OF FWD CARGO COMPARTMENT

53-476-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-478-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE KEEL BEAM EXTENSION 738.5 - 785.9		STRUCTURAL ILLUSTRATION REFERENCE 1-17	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 125 126			ACCESS PANELS 1001 1003 1251 821			

MECH	INSP	MPD ITEM NUMBER 53-478-00
<p>DETAILED - INTERNAL: KEEL BEAM EXTENSION CHORDS AND BODY SKIN, STRINGERS, AND FRAMES BETWEEN STRINGER S-35L AND S-35R.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

5 7 4 8	EFFECTIVITY	DETAILED IN	KEEL BEAM EXTENSION 738.5 - 785.9
			53-478-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

KEEL BEAM EXTENSION 738.5 - 785.9

53-478-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-480-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY STA 785.9 BULKHEAD INNER CHORD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 125 126			ACCESS PANELS 1001 1003 1251 821			

MECH	INSP	MPD ITEM NUMBER 53-480-00
<p>DETAILED - INTERNAL: BODY STATION 785.9 BULKHEAD INNER CHORD BETWEEN STRINGERS S-20 AND S-21.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE 		

5 7 5 0	EFFECTIVITY	DETAILED IN	BODY STA 785.9 BULKHEAD INNER CHORD
			53-480-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):

- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	BODY STA 785.9 BULKHEAD INNER CHORD
			53-480-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-482-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA A/C MIX BAY	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FRONT SPAR BULKHEAD BELOW FLOOR		STRUCTURAL ILLUSTRATION REFERENCE 1-51		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 125 126		ACCESS PANELS 1001 1003 1251 821				

MECH	INSP	MPD ITEM NUMBER 53-482-00
<p>GENERAL VISUAL - INTERNAL: WING FRONT SPAR BULKHEAD AND SIDE OF BODY SPLICE, INCLUDING AREA OF ATTACHMENT OF BODY SKIN TO FRONT SPAR AND VERTICAL STIFFENERS IN AREA OF LOWER FRONT SPAR CHORD.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1251 REQUIRES THE REMOVAL OF THE FORWARD CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

5 7 5 2	EFFECTIVITY	GEN VISUAL	FRONT SPAR BULKHEAD BELOW FLOOR
			53-482-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):

- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.

WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FRONT SPAR BULKHEAD BELOW FLOOR
		53-482-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-494-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE INTERIOR FUSELAGE INSPECTION			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 131 132			ACCESS PANELS 1001 1311 1312 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-494-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 785.9 TO BS 955.1, BETWEEN CABIN FLOOR AND THE WING CENTER SECTION UPPER SURFACE.</p> <p>ACCESS NOTE: REMOVE CABIN EQUIPMENT FROM BODY STA 785.9 TO 955.1.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

EFFECTIVITY	GEN VISUAL	INTERIOR FUSELAGE INSPECTION
		53-494-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

INTERIOR FUSELAGE INSPECTION

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-496-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE WING CENTER SECTION UPPER SURFACE		STRUCTURAL ILLUSTRATION REFERENCE 1-21 1-22		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 131 132			ACCESS PANELS 1001 1311 1312 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-496-00
<p>DETAILED - INTERNAL: STRUCTURE BELOW FLOOR BETWEEN BODY STA. 786 AND STA. 955.1 AND WING CENTER SECTION UPPER SURFACE INCLUDING: BODY SKIN. BODY FRAMES AND BODY STA 786 AND 955.1 BULKHEADS, FORWARD AND AFT SIDES. SIDE OF BODY SPLICE CHORD. BODY STA 786 SKIN/STRINGER CIRCUMFERENTIAL SPLICE.</p> <p>ACCESS NOTE: CABIN EQUIPMENT FROM BODY STA 785 TO STA 955 (PASS AIRPLANES ONLY).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1311 REQUIRES REMOVAL OF THE CABIN FLOOR COVERING STA 785 TO 955 PER MAINTENANCE MANUAL REFERENCE 25-27-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1312 REQUIRES REMOVAL OF THE CABIN FLOOR PANELS STA 785 TO 955 PER MAINTENANCE MANUAL REFERENCE 53-01-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR 		

EFFECTIVITY	DETAILED IN	WING CENTER SECTION UPPER SURFACE
		53-496-00 PAGE 1 OF 3 APR 22/08

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INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF

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EFFECTIVITY	_____	DETAILED IN	WING CENTER SECTION UPPER SURFACE
			53-496-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-496-00
AIRLINE CARD NO.

MECH	INSP

- FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY		DETAILED IN	WING CENTER SECTION UPPER SURFACE
			53-496-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-502-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 013	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE KEEL BEAM & WING CENTER SECTION		STRUCTURAL ILLUSTRATION REFERENCE 1-28 1-30 1-31		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 139		ACCESS PANELS 1391 139AL 139BL 139CL 139DL 139EZ 139FZ				

MECH	INSP	MPD ITEM NUMBER 53-502-00
<p>GENERAL VISUAL - INTERNAL: KEEL BEAM WEBS INCLUDING SHEAR WEB AT BODY STA. 863, STIFFENERS, AND CHORDS; AND WING CENTER SECTION LOWER SURFACE AND BODY BULKHEADS AT BODY STA. 785.9 AND STA. 955.1.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

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			53-502-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY _____ GEN VISUAL KEEL BEAM & WING CENTER SECTION

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-504-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE WING CENTER SECTION TENSION FITTING		STRUCTURAL ILLUSTRATION REFERENCE 1-29	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 139		ACCESS PANELS 139CL 139EZ 139FZ				

MECH	INSP	MPD ITEM NUMBER 53-504-00
<p>DETAILED - INTERNAL: TENSION FITTING INSTALLATION ON THE WING CENTER SECTION LOWER SURFACE AT BL 0 FROM THE REAR SPAR TO WING STRINGER S-1L.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

5 7 6 1	EFFECTIVITY	DETAILED IN	WING CENTER SECTION TENSION FITTING
			53-504-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	WING CENTER SECTION TENSION FITTING
			53-504-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-506-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE LOWER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 141 142		ACCESS PANELS 1001 1411 1412 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-506-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR, BS 955.1 TO BS 1065, BELOW CABIN FLOOR, INCLUDING THE CANTED PRESSURE DECK AND AFT TWO FEET OF WING CENTER SECTION UPPER SKIN.</p> <p>ACCESS NOTE: REMOVE CABIN EQUIPMENT FROM BODY STA 955.1 TO 1065.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

5 7 6 3	EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR
			53-506-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

LOWER LOBE INTERIOR
53-506-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-508-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BELOW FLOOR STRUCT - STA 955 TO 1065		STRUCTURAL ILLUSTRATION REFERENCE 1-52	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 141 142			ACCESS PANELS 1001 1411 1412 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-508-00
<p>DETAILED - INTERNAL: STRUCTURE BELOW FLOOR BETWEEN BODY STA. 955 AND STA. 1065 INCLUDING: FLOOR BEAMS; BODY FRAMES AND BODY STA. 1065 BULKHEAD, FWD AND AFT SIDES; PRESSURE DECK; BODY STA. 1065 SKIN/STRINGER CIRCUMFERENTIAL SPLICE; MAIN LANDING GEAR SUPPORT FITTING ATTACHMENT AT BODY STA. 1043.</p> <p>ACCESS NOTE: CABIN EQUIPMENT FROM BODY STA 955 TO STA 1065 (PASS AIRPLANES ONLY).</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1411 REQUIRES REMOVAL OF FLOOR PANELS/COVERINGS STA 955 TO 1065 PER MAINTENANCE MANUAL REFERENCE 25-27-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1412 REQUIRES REMOVAL OF FLOOR PANELS STA 955 TO 1065 PER MAINTENANCE MANUAL REFERENCE 53-01-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH 		

EFFECTIVITY	DETAILED IN	BELOW FLOOR STRUCT - STA 955 TO 1065
		53-508-00 PAGE 1 OF 3 APR 22/08

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- STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER

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EFFECTIVITY	_____	DETAILED IN	BELOW FLOOR STRUCT - STA 955 TO 1065
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- REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY		DETAILED IN	BELOW FLOOR STRUCT - STA 955 TO 1065
			53-508-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-510-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA LWR FUSLAGE	RELATED TASK	INTERVAL 00060 MOS	NOTE	PHASE 140XX	MPD REV 015	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BELOW FLOOR STRUCT - STA 955 TO 1065		STRUCTURAL ILLUSTRATION REFERENCE 1-52		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 141 142			ACCESS PANELS 1001 1411 1412 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-510-00
<p>DETAILED - INTERNAL: SLOPING PRESSURE DECK WEB IN AREAS OF DRAIN SCREENS BETWEEN BODY STA 955 AND STA 1065.</p> <p>INTERVAL NOTE: OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>ACCESS NOTE: CABIN EQUIPMENT FROM BODY STA 955 TO STA 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1411 REQUIRES REMOVAL OF FLOOR PANELS/COVERINGS STA 955 TO 1065 PER MAINTENANCE MANUAL REFERENCE 25-27-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1412 REQUIRES REMOVAL OF FLOOR PANELS STA 955 TO 1065 PER MAINTENANCE MANUAL REFERENCE 53-01-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE 		

5 7 6 8	EFFECTIVITY	DETAILED IN	BELOW FLOOR STRUCT - STA 955 TO 1065
			53-510-00 PAGE 1 OF 3 APR 22/08

MECH	INSP
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ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F

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EFFECTIVITY	_____	DETAILED IN	BELOW FLOOR STRUCT - STA 955 TO 1065
			53-510-00 PAGE 2 OF 3 APR 22/08

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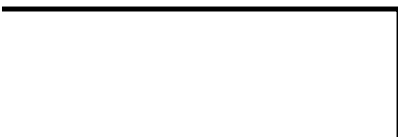
TASK CARD

BOEING CARD NO. 53-510-00
AIRLINE CARD NO.

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		<p>- SELECTED AREAS NOTED.</p> <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY



DETAILED IN

BELOW FLOOR STRUCT - STA 955 TO 1065

53-510-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-518-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1A	PHASE 10101	MPD REV 018	TASK CARD REVISION DEC 22/06
TASK GEN VISUAL	TITLE MAIN LANDING GEAR WHEEL WELL STRUCT		STRUCTURAL ILLUSTRATION REFERENCE 1-32 1-33 1-36		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 143		ACCESS PANELS 1004				

MECH	INSP	MPD ITEM NUMBER 53-518-00
		<p>GENERAL VISUAL - INTERNAL: MAIN LANDING GEAR WHEEL WELL STRUCTURE.</p>

EFFECTIVITY	GEN VISUAL	MAIN LANDING GEAR WHEEL WELL STRUCT
		53-518-00 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-520-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE TRANSVERSE BEAMS		STRUCTURAL ILLUSTRATION REFERENCE 1-36	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 143			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-520-00
<p>DETAILED - INTERNAL: TRANSVERSE BEAM LOWER CHORDS AT BODY STA 999 BETWEEN BL 22 AND BL 33, BODY STA 1021 BETWEEN BL 0 AND BL 69, AND BODY STA 1043 BETWEEN BL 0 AND BL 42.</p> <p>AIRPLANE NOTE: APPLICABLE TO 767-300 AND -400ER.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

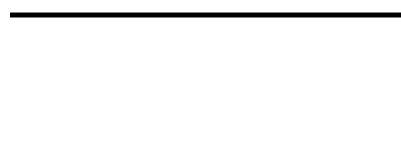
5 7 7 2	EFFECTIVITY	DETAILED IN	TRANSVERSE BEAMS
			53-520-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

TRANSVERSE BEAMS

53-520-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-522-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE MAIN LANDING GEAR DOOR UPLOCK		STRUCTURAL ILLUSTRATION REFERENCE 1-36 1-37		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 143			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-522-00
<p>DETAILED - INTERNAL: MAIN LANDING GEAR DOOR UPLOCK.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

5 7 7 4	EFFECTIVITY	DETAILED IN	MAIN LANDING GEAR DOOR UPLOCK
			53-522-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

MAIN LANDING GEAR DOOR UPLOCK

53-522-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-524-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE REAR SPAR BULKHEAD AT WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE 1-32 1-35 1-38 1-39		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 143			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-524-00
<p>DETAILED - INTERNAL: BODY STA. 955.1 REAR SPAR BULKHEAD AND WING REAR SPAR TERMINAL FITTING BETWEEN BL 62 AND WBL 97.9 INCLUDING SPAR WEB ADJACENT TO FITTING, REAR SPAR LOWER CHORD SPLICE AND OVERWING LONGERON.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

5 7 7 6	EFFECTIVITY	DETAILED IN	REAR SPAR BULKHEAD AT WHEEL WELL
			53-524-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

REAR SPAR BULKHEAD AT WHEEL WELL

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-526-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE TRANSVERSE BEAMS AT MAIN WHEEL WELLS		STRUCTURAL ILLUSTRATION REFERENCE 1-36	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 143			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-526-00
<p>DETAILED - INTERNAL: TRANSVERSE BEAM LOWER CHORDS AT BODY STA 1021 BETWEEN BL 20 AND BL 69 AND BODY STA 1043 BETWEEN BL 0 AND BL 14.</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND SF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

EFFECTIVITY	DETAILED IN	TRANSVERSE BEAMS AT MAIN WHEEL WELLS
		53-526-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

TRANSVERSE BEAMS AT MAIN WHEEL WELLS

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-528-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE REAR SPAR AT KEEL BEAM		STRUCTURAL ILLUSTRATION REFERENCE 1-53	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 143 144			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-528-00
<p>DETAILED - INTERNAL: WING REAR SPAR AT BL 0 IN AREA OF ATTACHMENT TO THE KEEL BEAM.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

EFFECTIVITY	DETAILED IN	REAR SPAR AT KEEL BEAM
		53-528-00 PAGE 1 OF 2 APR 22/08

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- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

REAR SPAR AT KEEL BEAM

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-530-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1A	PHASE 10101	MPD REV 018	TASK CARD REVISION DEC 22/06
TASK GEN VISUAL	TITLE MAIN LANDING GEAR WHEEL WELL STRUCT		STRUCTURAL ILLUSTRATION REFERENCE 1-32 1-33 1-36		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 144		ACCESS PANELS 1004				

MECH	INSP	MPD ITEM NUMBER 53-530-00
		GENERAL VISUAL - INTERNAL: MAIN LANDING GEAR WHEEL WELL STRUCTURE.

5 7 8 2	EFFECTIVITY	GEN VISUAL	MAIN LANDING GEAR WHEEL WELL STRUCT
			53-530-00 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-532-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE TRANSVERSE BEAMS AT MAIN WHEEL WELLS		STRUCTURAL ILLUSTRATION REFERENCE 1-36	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 144		ACCESS PANELS 1004				

MECH	INSP	MPD ITEM NUMBER 53-532-00
<p>DETAILED - INTERNAL: TRANSVERSE BEAM LOWER CHORDS AT BODY STA 999 BETWEEN BL 22 AND BL 33, BODY STA 1021 BETWEEN BL 0 AND BL 69, AND BODY STA 1043 BETWEEN BL 0 AND BL 42.</p> <p>AIRPLANE NOTE: APPLICABLE TO 767-300 AND -400ER.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

5 7 8 3	EFFECTIVITY	DETAILED IN	TRANSVERSE BEAMS AT MAIN WHEEL WELLS
			53-532-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

TRANSVERSE BEAMS AT MAIN WHEEL WELLS

53-532-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-534-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE MAIN LANDING GEAR DOOR UPLOCK		STRUCTURAL ILLUSTRATION REFERENCE 1-36 1-37		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 144			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-534-00
<p>DETAILED - INTERNAL: MAIN LANDING GEAR DOOR UPLOCK.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

5 7 8 5	EFFECTIVITY	DETAILED IN	MAIN LANDING GEAR DOOR UPLOCK
			53-534-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

MAIN LANDING GEAR DOOR UPLOCK

53-534-00

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STATION
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DATE



BOEING CARD NO. 53-536-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE REAR SPAR BULKHEAD AT WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE 1-32 1-35 1-38 1-39		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 144			ACCESS PANELS 1004			

MECH	INSP	MPD ITEM NUMBER 53-536-00
<p>DETAILED - INTERNAL: BODY STA. 955.1 REAR SPAR BULKHEAD AND WING REAR SPAR TERMINAL FITTING BETWEEN BL 62 AND WBL 97.9 INCLUDING SPAR WEB ADJACENT TO FITTING, REAR SPAR LOWER CHORD SPLICE AND OVERWING LONGERON.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

5 7 8 7	EFFECTIVITY	DETAILED IN	REAR SPAR BULKHEAD AT WHEEL WELL
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AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

REAR SPAR BULKHEAD AT WHEEL WELL

53-536-00

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STATION
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DATE



BOEING CARD NO. 53-538-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 018	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE TRANSVERSE BEAMS AT MAIN WHEEL WELLS		STRUCTURAL ILLUSTRATION REFERENCE 1-36	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 144		ACCESS PANELS 1004				

MECH	INSP	MPD ITEM NUMBER 53-538-00
<p>DETAILED - INTERNAL: TRANSVERSE BEAM LOWER CHORDS AT BODY STA 1021 BETWEEN BL 20 AND BL 69 AND BODY STA 1043 BETWEEN BL 0 AND BL 14.</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND SF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1004 REQUIRES OPENING/ CLOSING THE LANDING GEAR WHEEL WELL DOORS PER MAINTENANCE MANUAL REFERENCE 32-00-15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

5 7 8 9	EFFECTIVITY	DETAILED IN	TRANSVERSE BEAMS AT MAIN WHEEL WELLS
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MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

TRANSVERSE BEAMS AT MAIN WHEEL WELLS

53-538-00

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STATION
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DATE



BOEING CARD NO. 53-540-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE KEEL BEAM - AFT SECTION		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 149			ACCESS PANELS 149AL 149BL 149CL			

MECH	INSP	MPD ITEM NUMBER 53-540-00
<p>GENERAL VISUAL - INTERNAL: KEEL BEAM CHORDS, WEBS, STIFFENERS, BS 955.1 TO BS 1072.75, AND BODY BULKHEAD AT STA. 1081.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	GEN VISUAL	KEEL BEAM - AFT SECTION
		53-540-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	KEEL BEAM - AFT SECTION
			53-540-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-542-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE BODY FWD OF AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE 1-20 1-40		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 151 152		ACCESS PANELS 1001 1003 1511 822					

MECH	INSP	MPD ITEM NUMBER 53-542-00
<p>GENERAL VISUAL - INTERNAL: FLOOR BEAMS, SEAT TRACKS; BODY SKIN, STRINGERS, FRAMES, SPLICES & BODY STA. 1065 BULKHEAD FROM LOWER SURFACE OF CABIN FLOOR TO STRINGER S-32.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FORWARD WALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE 		

5 7 9 3	EFFECTIVITY	GEN VISUAL	BODY FWD OF AFT CARGO COMPARTMENT
			53-542-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-544-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE S 2C	14824	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY FWD OF AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE 1-20 1-40		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151		ACCESS PANELS 1001 1003 1005 1511 822					

MECH	INSP	MPD ITEM NUMBER 53-544-00
<p>DETAILED - INTERNAL: BL 26 AND 47 INTERCOSTALS; KEEL BEAM EXTENSION CHORDS AND BODY SKIN, STRINGERS, FRAMES, SPLICES AND BODY STA. 1065 BULKHEAD OUTER CHORD BETWEEN STRINGER S-32L AND BL 0.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FORWARD WALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE 		

EFFECTIVITY	DETAILED IN	BODY FWD OF AFT CARGO COMPARTMENT
		53-544-00 PAGE 1 OF 2 APR 22/08

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- FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-548-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE S 2C	14824	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY FWD OF AFT CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE 1-20 1-40		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 152		ACCESS PANELS 1001 1003 1005 1511 822					

MECH	INSP	MPD ITEM NUMBER 53-548-00
<p>DETAILED - INTERNAL: BL 26 AND 47 INTERCOSTALS; KEEL BEAM EXTENSION CHORDS AND BODY SKIN, STRINGERS, FRAMES, SPLICES AND BODY STA. 1065 BULKHEAD OUTER CHORD BETWEEN STRINGER S-32R AND BL 0.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FORWARD WALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE 		

EFFECTIVITY	DETAILED IN	BODY FWD OF AFT CARGO COMPARTMENT
		53-548-00 PAGE 1 OF 2 APR 22/08

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- FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	DETAILED IN	BODY FWD OF AFT CARGO COMPARTMENT
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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-550-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE LOWER LOBE INTERIOR			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE	
ZONES 153 154		ACCESS PANELS 1001 1003 1005 1531 1532 1541 822					

MECH	INSP	MPD ITEM NUMBER 53-550-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 1088.7 TO BS 1404, BETWEEN S-20 AND S-32.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1531 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT LEFT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR 		

EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE INTERIOR
		53-550-00 PAGE 1 OF 3 APR 22/08

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INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF

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EFFECTIVITY

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GEN VISUAL

FUSELAGE LOWER LOBE INTERIOR

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TASK CARD

BOEING CARD NO. 53-550-00
AIRLINE CARD NO.

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		<p>FLAMMABILITY PROPERTIES)</p> <ul style="list-style-type: none"> - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY

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GEN VISUAL

FUSELAGE LOWER LOBE INTERIOR

53-550-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-552-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO COMP	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE CARGO COMPARTMENT		STRUCTURAL ILLUSTRATION REFERENCE 1-40		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 154		ACCESS PANELS 1001 1003 1532 1541 822 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-552-00
<p>GENERAL VISUAL - INTERNAL: FLOOR BEAMS, SEAT TRACKS, AND BODY SKIN, STRINGERS, SKIN SPLICES, STRINGER SPLICES AND FRAMES FROM LOWER SURFACE OF CABIN FLOOR TO UPPER SURFACE OF CARGO FLOOR.</p> <p>ACCESS NOTE: RIGHT OUTBOARD CEILING PANELS (FROM BL 0) ONLY.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. 		

EFFECTIVITY	GEN VISUAL	CARGO COMPARTMENT
		53-552-00 PAGE 1 OF 3 APR 22/08

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ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH

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EFFECTIVITY	GEN VISUAL	CARGO COMPARTMENT
		53-552-00 PAGE 2 OF 3 APR 22/08

SAS



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TASK CARD

BOEING CARD NO. 53-552-00
AIRLINE CARD NO.

MECH	INSP
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- TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

CARGO COMPARTMENT

53-552-00

PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-554-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION DEC 22/08
TASK DETAILED IN		TITLE AFT CARGO DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 154		ACCESS PANELS 1223 1551 822				

MECH	INSP	MPD ITEM NUMBER 53-554-00
		<p>DETAILED - INTERNAL: AFT CARGO DOOR CUTOUT LOWER MAIN SILL UPPER AND LOWER SURFACE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

5 8 0 5	EFFECTIVITY	DETAILED IN	AFT CARGO DOOR LOWER SILL
			53-554-00 PAGE 1 OF 2 DEC 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

AFT CARGO DOOR LOWER SILL

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-556-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA AFT CARGO	RELATED TASK	INTERVAL 00060 MOS	NOTE	PHASE 140XX	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT CARGO COMPARTMENT DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE	
ZONES 154		ACCESS PANELS 1005 1541 822					

MECH	INSP	MPD ITEM NUMBER 53-556-00
<p>DETAILED - INTERNAL: EDGE FRAMES OF THE AFT CARGO COMPARTMENT DOOR CUTOUT.</p> <p>INTERVAL NOTE: OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

5 8 0 7	EFFECTIVITY	DETAILED IN	AFT CARGO COMPARTMENT DOOR CUTOUT
			53-556-00 PAGE 1 OF 2 APR 22/08

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- FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

AFT CARGO COMPARTMENT DOOR CUTOUT

53-556-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-558-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 014	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY BELOW AFT CARGO COMPART FLOOR		STRUCTURAL ILLUSTRATION REFERENCE 1-18		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 155 156			ACCESS PANELS 1003 1005 1531 1532 1541 1551 1552 822 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-558-00
<p>DETAILED - INTERNAL: LOWER LOBE INTERIOR, BODY SKIN, STRINGERS, ROLLER TRAYS, SKIN SPLICES, STRINGER SPLICES AND FRAMES FROM UPPER SURFACE OF CARGO FLOOR TO BL O.O INCLUDING AREA WITHIN 22 INCHES OF THE AFT CARGO COMPARTMENT DOOR CUTOUT LOWER MAIN SILL.</p> <p>ACCESS NOTE: NO PERMANENTLY ATTACHED STRUCTURE IS TO BE REMOVED TO ACCOMPLISH THIS INSPECTION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1551 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1552 REQUIRES REMOVAL OF AFT CARGO COMPARTMENT BALL TRANSFER PANEL PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., 		

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			53-558-00 PAGE 1 OF 3 APR 22/08

MECH	INSP
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GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)

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TASK CARD

BOEING CARD NO. 53-558-00
AIRLINE CARD NO.

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		<ul style="list-style-type: none"> - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY



DETAILED IN

BODY BELOW AFT CARGO COMPART FLOOR

53-558-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-562-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CARGO FL	RELATED TASK	INTERVAL NOTE	PHASE 99XXX	MPD REV 014	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FRAME UPPER CHORDS AT NUT PLATE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 155 156			ACCESS PANELS 1005 1551 1552 822 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-562-00
<p>DETAILED - INTERNAL: FRAME UPPER CHORDS AT NUT PLATE LOCATIONS BELOW THE CARGO FLOOR.</p> <p>INTERVAL NOTE: S 4C OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>AIRPLANE NOTE: AIRPLANES LINE NUMBER 1 TO 383.</p> <p>ACCESS NOTE: NO PERMANENTLY ATTACHED STRUCTURE IS TO BE REMOVED TO ACCOMPLISH THIS INSPECTION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1551 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1552 REQUIRES THE REMOVAL OF THE AFT CARGO COMPARTMENT BALL MAT TRANSFER PANELS PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 		

EFFECTIVITY	DETAILED IN	FRAME UPPER CHORDS AT NUT PLATE
		53-562-00 PAGE 1 OF 3 APR 22/08

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		<p>3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.</p> <p>4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.</p> <p>5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.</p> <p>6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):</p> <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23). <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35). <p>B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):</p> <ul style="list-style-type: none"> - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II) <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35). <p>C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:</p> <ul style="list-style-type: none"> - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS) - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1) - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
		<p>EFFECTIVITY</p>

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	DETAILED IN	FRAME UPPER CHORDS AT NUT PLATE
		53-562-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-562-00
AIRLINE CARD NO.

MECH	INSP	
		<ul style="list-style-type: none"> - AREAS WITH ELECTRICAL ARC POTENTIAL - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES) - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY



DETAILED IN

FRAME UPPER CHORDS AT NUT PLATE

53-562-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-564-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 4C	PHASE S 2C	14824	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE EXTERIOR SURFACE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 155		ACCESS PANELS 155AL					

MECH	INSP	MPD ITEM NUMBER 53-564-00
<p>GENERAL VISUAL - EXTERNAL: FUSELAGE EXTERIOR SURFACE FROM BS 1088 TO BS 1404, BETWEEN S-32L AND BL 0.0.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 8 1 5	EFFECTIVITY	GEN VISUAL	FUSELAGE EXTERIOR SURFACE
			53-564-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FUSELAGE EXTERIOR SURFACE
		53-564-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-572-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE BULK CARGO COMPARTMENT	STRUCTURAL ILLUSTRATION REFERENCE 1-41 1-43		APPLICABILITY AIRPLANE ENGINE ALL ALL			
ZONES 161	ACCESS PANELS 1001 1003 1005 1611 1612 811 NOTE						

MECH	INSP	MPD ITEM NUMBER 53-572-00
<p>GENERAL VISUAL - INTERNAL: LOWER LOBE INTERIOR FLOOR BEAMS, SEAT TRACKS, BODY SKIN, STRINGERS, AND FRAMES FROM LOWER SURFACE OF CABIN FLOOR TO UPPER SURFACE OF CARGO FLOOR WITHIN 22 INCHES FORWARD AND AFT OF BULK CARGO COMPARTMENT DOOR CUTOUT.</p> <p>ACCESS NOTE: LEFT OUTBOARD CEILING PANELS ONLY.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1611 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT LEFT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1612 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED 		

EFFECTIVITY	GEN VISUAL	BULK CARGO COMPARTMENT
		53-572-00 PAGE 1 OF 3 APR 22/08

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MECH	INSP
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INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD

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EFFECTIVITY

GEN VISUAL

BULK CARGO COMPARTMENT

SAS



767

TASK CARD

BOEING CARD NO. 53-572-00
AIRLINE CARD NO.

MECH	INSP
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(INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

BULK CARGO COMPARTMENT

53-572-00

PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-574-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 010	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BULK CARGO DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 161		ACCESS PANELS 1223 811				

MECH	INSP	MPD ITEM NUMBER 53-574-00
<p>DETAILED - INTERNAL: BULK CARGO DOOR CUTOUT LOWER SILL UPPER SURFACE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1223 REQUIRES REMOVAL OF THE SCUFF PLATE AT THE LOWER SILL OF THE CARGO DOOR.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

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EFFECTIVITY	DETAILED IN	BULK CARGO DOOR LOWER SILL
		53-574-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	BULK CARGO DOOR LOWER SILL
			53-574-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-576-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL 00060 MOS	NOTE	PHASE 140XX	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT ENTRY/SERVICE DOOR CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE 2-12		APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 161 162			ACCESS PANELS 1003 1005 1611 1621 811				

MECH	INSP	MPD ITEM NUMBER 53-576-00
<p>DETAILED - INTERNAL: AFT ENTRY/SERVICE DOOR LOWER MAIN SILLS.</p> <p>INTERVAL NOTE: OR 12,000 CYCLES, WHICHEVER OCCURS FIRST.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1611 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT LEFT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1621 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT RIGHT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS 		

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EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR CUTOUTS
		53-576-00 PAGE 1 OF 3 APR 22/08

MECH	INSP
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FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)

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EFFECTIVITY

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DETAILED IN

AFT ENTRY/SERVICE DOOR CUTOUTS

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-576-00
AIRLINE CARD NO.

MECH	INSP	
		<ul style="list-style-type: none"> - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

5 8 2 4	EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR CUTOUTS
			53-576-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-578-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE LOWER LOBE INTERIOR - RIGHT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 162		ACCESS PANELS 1001 1612 1621 811					

MECH	INSP	MPD ITEM NUMBER 53-578-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 1404 TO BS 1540, BETWEEN S-20R TO S-32R.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 8 2 5	EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - RIGHT
			53-578-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

LOWER LOBE INTERIOR - RIGHT
53-578-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-582-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA BULK CARGO	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT ENTRY/SERVICE DOOR CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE 1-43		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 162			ACCESS PANELS 1003 1005 1611 1621 811				

MECH	INSP	MPD ITEM NUMBER 53-582-00
<p>DETAILED - INTERNAL: LOWER LOBE INTERIOR BODY SKIN, DOUBLER, STRINGERS, FRAMES, AND LOWER SILL WITHIN 22 INCHES FORWARD AND AFT OF THE AFT ENTRY AND SERVICE DOOR CUTOUTS BETWEEN STRINGER S-20 AND STRINGER S-23.</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1611 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT LEFT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1621 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT RIGHT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., 		

5 8 2 7	EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR CUTOUTS
			53-582-00 PAGE 1 OF 3 APR 22/08

MECH	INSP
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GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)

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EFFECTIVITY

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DETAILED IN

AFT ENTRY/SERVICE DOOR CUTOUTS

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-582-00
AIRLINE CARD NO.

MECH	INSP	
		<ul style="list-style-type: none"> - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

5 8 2 9	EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR CUTOUTS
			53-582-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-584-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BULK CARGO COMPARTMENT FLOOR & BELOW		STRUCTURAL ILLUSTRATION REFERENCE 1-41	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 163			ACCESS PANELS 1005 1631 811			

MECH	INSP	MPD ITEM NUMBER 53-584-00
<p>DETAILED - INTERNAL: LOWER LOBE INTERIOR BODY SKIN FROM BS 1404 TO BS 1540, BETWEEN S-32L AND BL 0.0, INCLUDING SKINS, STRINGERS, AND FRAMES FROM UPPER SURFACE OF CARGO FLOOR TO BL 0 WITHIN 22 INCHES OF THE BULK CARGO COMPARTMENT DOOR CUTOUT LOWER MAIN SILL.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1631 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT FLOOR PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

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EFFECTIVITY	DETAILED IN	BULK CARGO COMPARTMENT FLOOR & BELOW
		53-584-00 PAGE 1 OF 2 APR 22/08

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- FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

BULK CARGO COMPARTMENT FLOOR & BELOW

53-584-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-586-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE LOWER LOBE INTERIOR BODY SKIN		STRUCTURAL ILLUSTRATION REFERENCE 1-41	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 164			ACCESS PANELS 1005 1631 811			

MECH	INSP	MPD ITEM NUMBER 53-586-00
<p>DETAILED - INTERNAL: LOWER LOBE INTERIOR BODY SKIN FROM BS 1404 TO BS 1540, BETWEEN S-32R AND BL O.O, INCLUDING SKINS, STRINGERS, AND FRAMES FROM UPPER SURFACE OF CARGO FLOOR TO BL O.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1005 REQUIRES THE REMOVAL OF CARGO COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-52-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1631 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT FLOOR PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

5 8 3 2	EFFECTIVITY	DETAILED IN	LOWER LOBE INTERIOR BODY SKIN
			53-586-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	LOWER LOBE INTERIOR BODY SKIN
			53-586-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-588-00
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION	
AIRPL	FUSELAGE		S 8C	S 4C	24848	008 APR 22/08	
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY	
GEN VISUAL		FUSELAGE LOWER LOBE INTERIOR		1-43		AIRPLANE	ENGINE
						ALL	ALL
ZONES			ACCESS PANELS				
165 166			1001 1005 1611 1621 1651 811 NOTE				

MECH	INSP	MPD ITEM NUMBER
		53-588-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 1540 TO BS 1629, BETWEEN S-20 AND S-32. ACCESS NOTE: WASTE TANK REMOVAL IS RECOMMENDED IN ORDER TO FACILITATE INSPECTION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1651 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE 		

EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE INTERIOR
		53-588-00 PAGE 1 OF 2 APR 22/08

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- PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	FUSELAGE LOWER LOBE INTERIOR
		53-588-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-590-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE LOWER LOBE INTERIOR - LEFT			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 165 166			ACCESS PANELS 1001 1005 1651 811 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-590-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE LOWER LOBE INTERIOR STRUCTURE, BS 1540 TO BS 1629, BETWEEN S-32L AND S-32R, INCLUDING SKINS, STRINGERS, FRAMES, BULKHEADS, LONGERONS AND CARGO FLOOR STRUCTURE.</p> <p>ACCESS NOTE: WASTE AND POTABLE WATER TANK REMOVAL IF INSTALLED, IS RECOMMENDED IN ORDER TO FACILITATE INSPECTION</p> <p>INTERVAL NOTES: 1) IF LEVELING COMPOUND IS NOT PRESENT: THEN, IF A DUAL COAT OF CORROSION INHIBITOR PER PART 6B) OF THE BASIC TASK IS MAINTAINED A REPEAT OF S 4C IS ACCEPTABLE; IF DUAL COAT IS NOT USED IN THIS AREA THEN A REPEAT INTERVAL OF S 2C IS REQUIRED. 2) IF LEVELING COMPOUND IS PRESENT: THEN A REPEAT INTERVAL OF S 2C IS REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS: 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS,</p>		

EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - LEFT
		53-590-00 PAGE 1 OF 3 APR 22/08

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- CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

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EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - LEFT
		53-590-00 PAGE 2 OF 3 APR 22/08

MECH	INSP	
		7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	LOWER LOBE INTERIOR - LEFT
		53-590-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-592-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 013	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY BELOW FLOOR-REAR PRESSURE DOME		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 165 166			ACCESS PANELS 1001 1005 1651 811 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-592-00
		<p>DETAILED - INTERNAL: BODY SKIN, STRINGERS, FRAMES, BODY STA. 1582 SKIN/STRINGER SPLICE, AND LOWER 22 INCHES OF THE PRESSURE DOME FROM STRINGER S-28 TO BL 0.</p> <p>ACCESS NOTE: WASTE AND POTABLE WATER TANK REMOVAL, IF INSTALLED, IS RECOMMENDED IN ORDER TO FACILITATE INSPECTION.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1651 REQUIRES THE REMOVAL OF THE BULK CARGO COMPARTMENT AFT WALL PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

EFFECTIVITY	DETAILED IN	BODY BELOW FLOOR-REAR PRESSURE DOME
		53-592-00 PAGE 1 OF 2 APR 22/08

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4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

BODY BELOW FLOOR-REAR PRESSURE DOME

53-592-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-594-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 4C	PHASE S 2C	14824	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE EXTERIOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 165		ACCESS PANELS 163AL					

MECH	INSP	MPD ITEM NUMBER 53-594-00
<p>GENERAL VISUAL - EXTERNAL: FUSELAGE EXTERIOR SURFACE FROM BS 1540 TO BS 1582, BETWEEN S-20L AND BL 0.0.</p> <p>CPC NOTE: APPLY ITEM 6) OF THE BASIC TASK TO SKIN AT TOILET SERVICE PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

EFFECTIVITY	GEN VISUAL	FUSELAGE EXTERIOR
		53-594-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

FUSELAGE EXTERIOR

53-594-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-602-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE UPPER WING/SIDE OF BODY SPLICE		STRUCTURAL ILLUSTRATION REFERENCE 1-34		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 191		ACCESS PANELS 191AL 191BL 191CL 191EL 191FL 191GL				

MECH	INSP	MPD ITEM NUMBER 53-602-00
<p>GENERAL VISUAL - INTERNAL: WING UPPER SURFACE AND SIDE OF BODY SPLICE NORMALLY COVERED BY THE FAIRING BETWEEN BODY STA. 785.9 AND STA. 955.1.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 8 4 3	EFFECTIVITY	GEN VISUAL	UPPER WING/SIDE OF BODY SPLICE
			53-602-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	UPPER WING/SIDE OF BODY SPLICE
			53-602-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-606-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE OVERWING LONGERON		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 191		ACCESS PANELS 191FL 191GL				

MECH	INSP	MPD ITEM NUMBER 53-606-00
<p>DETAILED - INTERNAL: OVERWING LONGERON IN AREAS OF ATTACHMENTS AT BODY STA. 859.5, 883.5 AND 911.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	DETAILED IN	OVERWING LONGERON
		53-606-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

OVERWING LONGERON

53-606-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-608-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE UPPER WING/SIDE OF BODY SPLICE		STRUCTURAL ILLUSTRATION REFERENCE 1-34		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 192		ACCESS PANELS 192AR 192BR 192CR 192DR 192ER 192FR 192GR				

MECH	INSP	MPD ITEM NUMBER 53-608-00
<p>GENERAL VISUAL - INTERNAL: WING UPPER SURFACE AND SIDE OF BODY SPLICE NORMALLY COVERED BY THE FAIRING BETWEEN BODY STA. 785.9 AND STA. 955.1.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 8 4 7	EFFECTIVITY	GEN VISUAL	UPPER WING/SIDE OF BODY SPLICE
			53-608-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	UPPER WING/SIDE OF BODY SPLICE
		53-608-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-610-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE WING LWR SURFACE BL 62 -FAIRING EDGE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 192			ACCESS PANELS 192FR 192GR 194AR 194HR 194KR 194LR 194MR			

MECH	INSP	MPD ITEM NUMBER 53-610-00
<p>GENERAL VISUAL - INTERNAL: LOWER WING SURFACE BETWEEN BL 62 AND OUTBOARD EDGE OF WING/BODY FAIRING INCLUDING SIDE OF BODY SPLICE, WING FRONT SPAR SPLICE, BULKHEAD AT BODY STA. 785.9 AND BODY SKIN NORMALLY COVERED BY THE FAIRING FROM FORWARD EDGE OF FAIRING TO BODY STA. 955.1.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

5 8 4 9	EFFECTIVITY	GEN VISUAL	WING LWR SURFACE BL 62 -FAIRING EDGE
			53-610-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

GEN VISUAL

WING LWR SURFACE BL 62 -FAIRING EDGE

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-612-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE OVERWING LONGERON		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 192		ACCESS PANELS 192FR 192GR				

MECH	INSP	MPD ITEM NUMBER 53-612-00
<p>DETAILED - INTERNAL: OVERWING LONGERON IN AREAS OF ATTACHMENTS AT BODY STA. 859.5, 883.5 AND 911.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	DETAILED IN	OVERWING LONGERON
		53-612-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

OVERWING LONGERON

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-614-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION DEC 22/08
TASK GEN VISUAL		TITLE WING LWR SURF BL 62 - FAIRING EDGE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 193			ACCESS PANELS 193GL 193JL 193KL 193NL 193PL			

MECH	INSP	MPD ITEM NUMBER 53-614-00
<p>GENERAL VISUAL - INTERNAL:</p> <p>LOWER WING SURFACE BETWEEN BL 62 AND OUTBOARD EDGE OF WING/BODY FAIRING INCLUDING SIDE OF BODY SPLICE, WING FRONT SPAR SPLICE, BULKHEAD AT BODY STA. 785.9 AND BODY SKIN NORMALLY COVERED BY THE FAIRING FROM FORWARD EDGE OF FAIRING TO BODY STA. 955.1 AND THE PORTION OF THE OUTBOARD WING LOWER PANEL NORMALLY CONCEALED BY THE FAIRINGS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

5 8 5 3	EFFECTIVITY	GEN VISUAL	WING LWR SURF BL 62 - FAIRING EDGE
			53-614-00 PAGE 1 OF 2 DEC 22/08

MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

WING LWR SURF BL 62 - FAIRING EDGE

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-616-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE LWR SIDE OF BODY SPLICE PLATES		STRUCTURAL ILLUSTRATION REFERENCE 1-44 1-45		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 193			ACCESS PANELS 193PL			

MECH	INSP	MPD ITEM NUMBER 53-616-00
<p>DETAILED - INTERNAL: SIDE OF BODY SPLICE PLATES IN AREAS OF ATTACHMENT TO THE LOWER WING SURFACE BETWEEN THE REAR SPAR AND STRINGER S-10L.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	DETAILED IN	LWR SIDE OF BODY SPLICE PLATES
		53-616-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

LWR SIDE OF BODY SPLICE PLATES

53-616-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-618-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE LWR WING SURFACE AT REAR SPAR CHORD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 193		ACCESS PANELS 193PL				

MECH	INSP	MPD ITEM NUMBER 53-618-00
<p>DETAILED - INTERNAL: LOWER WING SURFACE AT THE REAR SPAR CHORD AND SPLICE STRINGERS S-6L AND S-10L FROM BL 62 TO OUTBOARD EDGE OF WING/BODY FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	DETAILED IN	LWR WING SURFACE AT REAR SPAR CHORD
		53-618-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	LWR WING SURFACE AT REAR SPAR CHORD
			53-618-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-620-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE WING LWR REAR SPAR SPLICE FITTING		STRUCTURAL ILLUSTRATION REFERENCE 1-35	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 193			ACCESS PANELS 193PL			

MECH	INSP	MPD ITEM NUMBER 53-620-00
<p>DETAILED - INTERNAL: WING LOWER SURFACE REAR SPAR SPLICE FITTING AT WBL 97.9.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).</p>		

EFFECTIVITY	DETAILED IN	WING LWR REAR SPAR SPLICE FITTING
		53-620-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
- WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	WING LWR REAR SPAR SPLICE FITTING
			53-620-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-622-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BL 70 UNDERWING LONGERON FITTING		STRUCTURAL ILLUSTRATION REFERENCE 1-46	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 193			ACCESS PANELS 193PL			

MECH	INSP	MPD ITEM NUMBER 53-622-00
<p>DETAILED - INTERNAL: BL 70 UNDERWING LONGERON FITTING INSTALLATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).</p>		

EFFECTIVITY	DETAILED IN	BL 70 UNDERWING LONGERON FITTING
		53-622-00 PAGE 1 OF 2 APR 22/08

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TASK CARD

BOEING CARD NO. 53-622-00
AIRLINE CARD NO.

MECH	INSP

- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
- WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



DETAILED IN

BL 70 UNDERWING LONGERON FITTING

53-622-00

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STATION
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DATE



BOEING CARD NO. 53-624-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 013	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE WING LWR SURFACE BL 62 -FAIRING EDGE			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 194		ACCESS PANELS 194AR 194HR 194KR 194LR 194MR 194NR				

MECH	INSP	MPD ITEM NUMBER 53-624-00
<p>GENERAL VISUAL - INTERNAL: LOWER WING SURFACE BETWEEN BL 62 AND OUTBOARD EDGE OF WING/BODY FAIRING INCLUDING SIDE OF BODY SPLICE, WING FRONT SPAR SPLICE, BULKHEAD AT BODY STA. 785.9 AND BODY SKIN NORMALLY COVERED BY THE FAIRING FROM FORWARD EDGE OF FAIRING TO BODY STA. 955.1 AND THE PORTION OF THE OUTBOARD WING LOWER PANEL NORMALLY CONCEALED BY THE FAIRINGS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

EFFECTIVITY	GEN VISUAL	WING LWR SURFACE BL 62 -FAIRING EDGE
		53-624-00 PAGE 1 OF 2 APR 22/08

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GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

WING LWR SURFACE BL 62 -FAIRING EDGE

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STATION
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BOEING CARD NO. 53-626-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION AUG 22/09
TASK DETAILED IN		TITLE LWR SIDE OF BODY SPLICE PLATES		STRUCTURAL ILLUSTRATION REFERENCE 1-44 1-45		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 194			ACCESS PANELS 194MR			

MECH	INSP	MPD ITEM NUMBER 53-626-00
<p>DETAILED - INTERNAL: SIDE OF BODY SPLICE PLATES IN AREAS OF ATTACHMENT TO THE LOWER WING SURFACE BETWEEN THE REAR SPAR AND STRINGER S-10L.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

5 8 6 5	EFFECTIVITY	DETAILED IN	LWR SIDE OF BODY SPLICE PLATES
			53-626-00 PAGE 1 OF 2 AUG 22/09

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

LWR SIDE OF BODY SPLICE PLATES

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-628-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE LWR WING SURFACE AT REAR SPAR CHORD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 194			ACCESS PANELS 194MR			

MECH	INSP	MPD ITEM NUMBER 53-628-00
<p>DETAILED - INTERNAL: LOWER WING SURFACE AT THE REAR SPAR CHORD AND SPLICE STRINGERS S-6L AND S-10L FROM BL 62 TO OUTBOARD EDGE OF WING/BODY FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

5	EFFECTIVITY	DETAILED IN	LWR WING SURFACE AT REAR SPAR CHORD
8			53-628-00 PAGE 1 OF 2 APR 22/08
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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	LWR WING SURFACE AT REAR SPAR CHORD
			53-628-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-630-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE WING LWR REAR SPAR SPLICE FITTING		STRUCTURAL ILLUSTRATION REFERENCE 1-35	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 194			ACCESS PANELS 194MR			

MECH	INSP	MPD ITEM NUMBER 53-630-00
<p>DETAILED - INTERNAL: WING LOWER SURFACE REAR SPAR SPLICE FITTING AT WBL 97.9.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).</p>		

5 8 6 9	EFFECTIVITY	DETAILED IN	WING LWR REAR SPAR SPLICE FITTING
			53-630-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	WING LWR REAR SPAR SPLICE FITTING
			53-630-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-632-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BL 70 UNDERWING LONGERON FITTING		STRUCTURAL ILLUSTRATION REFERENCE 1-46	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 194			ACCESS PANELS 194MR			

MECH	INSP	MPD ITEM NUMBER 53-632-00
<p>DETAILED - INTERNAL: BL 70 UNDERWING LONGERON FITTING INSTALLATION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).</p>		

EFFECTIVITY	DETAILED IN	BL 70 UNDERWING LONGERON FITTING
		53-632-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
- WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	BL 70 UNDERWING LONGERON FITTING
			53-632-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-634-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 012	TASK CARD REVISION DEC 22/08
TASK GEN VISUAL	TITLE WING-TO-BODY FAIRING		STRUCTURAL ILLUSTRATION REFERENCE 1-49		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 195		ACCESS PANELS 195BL 195CL 195DL 195EL 195NL 195RL 195SL				

MECH	INSP	MPD ITEM NUMBER 53-634-00
		<p>GENERAL VISUAL - INTERNAL: BODY SKIN AND OUTBOARD WING UPPER PANEL NORMALLY COVERED BY THE WING/BODY FAIRING FROM BODY STA. 955.1 TO AFT EDGE OF FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

EFFECTIVITY	GEN VISUAL	WING-TO-BODY FAIRING
		53-634-00 PAGE 1 OF 2 DEC 22/08

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MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	WING-TO-BODY FAIRING
		53-634-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-636-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION DEC 22/08
TASK GEN VISUAL	TITLE SKIN UNDER WING/BODY AFT FAIRING		STRUCTURAL ILLUSTRATION REFERENCE 1-49		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 196		ACCESS PANELS 196BR 196CR 196DR 196ER 196KR 196NR				

MECH	INSP	MPD ITEM NUMBER 53-636-00
<p>GENERAL VISUAL - INTERNAL: BODY SKIN AND OUTBOARD WING UPPER PANEL NORMALLY COVERED BY THE WING/BODY FAIRING FROM BODY STA. 955.1 TO AFT EDGE OF FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

EFFECTIVITY	GEN VISUAL	SKIN UNDER WING/BODY AFT FAIRING
		53-636-00 PAGE 1 OF 2 DEC 22/08

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MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	SKIN UNDER WING/BODY AFT FAIRING
		53-636-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-638-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE SKIN UNDER WING/BODY AFT L FAIRING	STRUCTURAL ILLUSTRATION REFERENCE 1-49		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 197		ACCESS PANELS 197CL 197DL				

MECH	INSP	MPD ITEM NUMBER 53-638-00
<p>GENERAL VISUAL - INTERNAL: BODY SKIN AND OUTBOARD WING UPPER PANEL NORMALLY COVERED BY THE WING/BODY FAIRING FROM BODY STA. 955.1 TO AFT EDGE OF FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>		

5 8 7 7	EFFECTIVITY	GEN VISUAL	SKIN UNDER WING/BODY AFT L FAIRING
			53-638-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	GEN VISUAL	SKIN UNDER WING/BODY AFT L FAIRING
			53-638-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-640-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA W/B FAIRING	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE SKIN UNDER WING/BODY AFT R FAIRING		STRUCTURAL ILLUSTRATION REFERENCE 1-49	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 198			ACCESS PANELS 1981 198CR 198GR			

MECH	INSP	MPD ITEM NUMBER 53-640-00
		<p>GENERAL VISUAL - INTERNAL: BODY SKIN AND OUTBOARD WING UPPER PANEL NORMALLY COVERED BY THE WING/BODY FAIRING FROM BODY STA. 955.1 TO AFT EDGE OF FAIRING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p>

EFFECTIVITY	GEN VISUAL	SKIN UNDER WING/BODY AFT R FAIRING
		53-640-00 PAGE 1 OF 2 APR 22/08

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TASK CARD

BOEING CARD NO. 53-640-00
AIRLINE CARD NO.

MECH	INSP

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

SKIN UNDER WING/BODY AFT R FAIRING

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-642-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE STRUCTURE FWD LAVS, GALLEYS & DOORS		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 200		ACCESS PANELS 2221 2322 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-642-00
<p>GENERAL VISUAL - INTERNAL: FLOOR SUPPORT STRUCTURE UNDER FORWARD GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES FROM BS 276 TO BS 368 FOR PASSENGER AIRPLANES AND BS 244 TO BS 367 FOR FREIGHTER AIRPLANES.</p> <p>ACCESS NOTES: 1) REMOVE FORWARD GALLEYS AND LAVS (IF THE FLOOR PANELS UNDER THE GALLEYS ARE FACTORY SEALED AND THE FACTORY SEAL HAS NOT BEEN BROKEN THEN GALLEY REMOVAL IS NOT REQUIRED). 2) FLOOR PANELS THAT ARE FACTORY SEALED DO NOT HAVE TO BE REMOVED. 3) REMOVAL OF SEATS, CABIN EQUIPMENT, AND FLOOR PANELS TO GAIN ACCESS MAY BE REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL 		

EFFECTIVITY	GEN VISUAL	STRUCTURE FWD LAVS, GALLEYS & DOORS
		53-642-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):

- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):

- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.

WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:

- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	STRUCTURE FWD LAVS, GALLEYS & DOORS
		53-642-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-644-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 12C	PHASE S 4C	MPD REV 34848	TASK CARD REVISION 012 APR 22/08	
TASK GEN VISUAL		TITLE STRUCTURE CTR & AFT GAL, LAV & DOORS			STRUCTURAL ILLUSTRATION REFERENCE		
ZONES 200		ACCESS PANELS 1311 1312 1411 1412 NOTE					
MECH		INSP				MPD ITEM NUMBER 53-644-00	

GENERAL VISUAL - INTERNAL:
 FLOOR SUPPORT STRUCTURE UNDER CENTER AND AFT GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES.

ACCESS NOTES:

- 1) REMOVE CENTER AND AFT GALLEYS AND LAVS (IF THE FLOOR PANELS UNDER GALLEYS ARE FACTORY SEALED AND THE FACTORY SEAL HAS NOT BEEN BROKEN THEN THE GALLEY REMOVAL IS NOT REQUIRED).
- 2) FLOOR PANELS THAT ARE FACTORY SEALED DO NOT NEED TO BE REMOVED.
- 3) REMOVAL OF SEATS, CABIN EQUIPMENT, AND FLOOR PANELS TO GAIN ACCESS MAY BE REQUIRED.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
 IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE

EFFECTIVITY

GEN VISUAL

STRUCTURE CTR & AFT GAL, LAV & DOORS

53-644-00 PAGE 1 OF 2 APR 22/08

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- FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-646-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE SEAT TRACKS			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE PASS ALL
ZONES 200			ACCESS PANELS 2002 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-646-00					
		<p>GENERAL VISUAL - INTERNAL: SEAT TRACKS.</p> <p>ACCESS NOTE: REMOVAL OF SEATS, CABIN EQUIPMENT, AND FLOOR PANELS TO GAIN ACCESS MAY BE REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2002 REQUIRES REMOVAL OF THE SEAT TRACK COVERS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 					

EFFECTIVITY	GEN VISUAL	SEAT TRACKS	53-646-00	PAGE 1 OF 2	APR 22/08
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MECH	INSP

- GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

 - 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	SEAT TRACKS
		53-646-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-648-00
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION					
AIRPL	FUSELAGE		S 1A	10101	012	DEC 22/07					
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE			APPLICABILITY				
GEN VISUAL		BODY EXTERNAL SKIN		1-2	2-4	2-9	AIRPLANE	ENGINE			
				1-2	2-4	2-9	ALL	ALL			
ZONES			ACCESS PANELS								
100	200		831	833	835	836	837	838	841	843	845
			846	NOTE							

MECH	INSP	MPD ITEM NUMBER
		<p>53-648-00</p> <p>GENERAL VISUAL - EXTERNAL: BODY SKIN INCLUDING AREAS AROUND CUTOUTS.</p> <p>ACCESS NOTE: AS VIEWED FROM THE GROUND.</p>

EFFECTIVITY		GEN VISUAL	BODY EXTERNAL SKIN
			53-648-00 PAGE 1 OF 1 DEC 22/07

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-650-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 009	TASK CARD REVISION APR 22/08
TASK DETAILED EX		TITLE DOOR STOPS AND CUTOUTS		STRUCTURAL ILLUSTRATION REFERENCE 1-2 2-2 2-4 2-5 2-9		APPLICABILITY AIRPLANE ENGINE ALL ALL
ZONES 200		ACCESS PANELS 831 832 833 834 835 836 837 838 841 842 843 844 845 846				

MECH	INSP	MPD ITEM NUMBER 53-650-00
<p>DETAILED - EXTERNAL: DOOR STOPS AND VISIBLE ADJACENT CUT-OUT STRUCTURE WITHIN AIRCRAFT CONTOUR AT ALL DOOR LOCATIONS (CABIN ENTRY/SERVICE, AND EMERGENCY) INCLUDING SECTION 41 ACCESS DOORS AND VISIBLE ADJACENT SUPPORT AND CUTOUT STRUCTURE AT NO. 2 COCKPIT WINDOW.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

EFFECTIVITY	DETAILED EX	DOOR STOPS AND CUTOUTS
		53-650-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED EX

DOOR STOPS AND CUTOUTS

53-650-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-652-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	MPD REV 24848	TASK CARD REVISION 012	APR 22/08
TASK GEN VISUAL	TITLE MAIN DECK DOOR & HATCH INTERIORS			STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 200		ACCESS PANELS NOTE					

MECH	INSP	MPD ITEM NUMBER 53-652-00
		<p>GENERAL VISUAL - INTERNAL: MAIN DECK DOOR AND HATCH INTERIORS.</p> <p>ACCESS NOTE: REMOVAL OF SEATS, CABIN EQUIPMENT, AND FLOOR PANELS TO GAIN ACCESS MAY BE REQUIRED.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

EFFECTIVITY	GEN VISUAL	MAIN DECK DOOR & HATCH INTERIORS
		53-652-00 PAGE 1 OF 2 APR 22/08

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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	MAIN DECK DOOR & HATCH INTERIORS
		53-652-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-654-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CREW CABIN	RELATED TASK	INTERVAL	PHASE 24848	MPD REV 009	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE CONTROL CABIN WINDSHIELD STRUCTURE		STRUCTURAL ILLUSTRATION REFERENCE 2-1 2-2 2-3		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 211 212			ACCESS PANELS 2111 2112 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-654-00
<p>GENERAL VISUAL - INTERNAL: INTERIOR STRUCTURE ABOVE CREW CABIN FLOOR INCLUDING CONTROL CABIN WINDSHIELDS AND VISIBLE PORTIONS OF WINDOW POSTS AND SILLS AFTER ACCESSED.</p> <p>INTERVAL NOTE: THRESHOLD - S 8C AND REPEAT INTERVAL - S 4C. THIS TASK SHOULD ALSO BE PERFORMED AT WINDSHIELD AND WINDOW REPLACEMENT.</p> <p>ACCESS NOTE: REMOVAL OF READILY REMOVED SIDEWALL PANELS REQUIRED. REMOVAL OF OVERHEAD DRIPSHIELD, GLARE PANEL, WINDOWS, OR PERMANENTLY ATTACHED WALL PANELS NOT REQUIRED TO ACCOMPLISH INTENT OF INSPECTION.</p> <p>ACCESS NOTE: INSPECT FLIGHT DECK FROM BS 132.5 TO BS 246, WITHOUT REMOVAL OF FLIGHT PANELS/CONSOLES OR ELECTRICAL PANELS. INSPECTION OF STRUCTURE WITH PANEL AND CONSOLE REMOVAL IS LIMITED TO OPPORTUNITY INSPECTIONS AND SELECTED OUT-OF-SERVICE/RETIRED AIRPLANES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2112 REQUIRES REMOVAL OF THE WINDSHIELD POST PADDING.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE 		

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			53-654-00 PAGE 1 OF 3 APR 22/08

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CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL

EFFECTIVITY

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GEN VISUAL

CONTROL CABIN WINDSHIELD STRUCTURE

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TASK CARD

BOEING CARD NO. 53-654-00
AIRLINE CARD NO.

MECH	INSP

- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



GEN VISUAL

CONTROL CABIN WINDSHIELD STRUCTURE

53-654-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-656-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CREW CABIN	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE CONTROL CABIN FLOOR STRUCTURE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 211 212			ACCESS PANELS 2111			

MECH	INSP	MPD ITEM NUMBER 53-656-00
<p>GENERAL VISUAL - INTERNAL: CONTROL CABIN FLOOR STRUCTURE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2111 REQUIRES REMOVAL OF THE CREW SEATS PER MAINTENANCE MANUAL REFERENCE 25-11-01.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

EFFECTIVITY	GEN VISUAL	CONTROL CABIN FLOOR STRUCTURE
		53-656-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP
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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	CONTROL CABIN FLOOR STRUCTURE
		53-656-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-662-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INT STRUCTURE		STRUCTURAL ILLUSTRATION REFERENCE 2-5 2-6		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 221 222 223 224			ACCESS PANELS 2003 2215 2221 2223 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-662-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE AND INCLUDING MAIN DECK FLOOR STRUCTURE, BS 243.5/246 TO BS 434, BETWEEN S-20 AND S-12, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAMES AND SILLS WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, AND DOOR CUTOUT STRUCTURE, AND HORIZONTAL SHEAR WEB.</p> <p>ACCESS NOTE: OUTBOARD EQUIPMENT, SIDEWALL PANELS AND MOVABLE CEILING PANELS TO BE REMOVED (FOR FREIGHTERS ONLY).</p> <p>ACCESS NOTE: FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION (EXCEPT UNDER GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES).</p> <p>ACCESS NOTE: CARGO COMPARTMENT FLOOR PANEL REMOVAL REQUIRED. REMOVAL OF RIGID BARRIER, FITTINGS, AND DRAIN TROUGH NOT REQUIRED (FOR FREIGHTERS ONLY).</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2221 REQUIRES THE REMOVAL OF THE FORWARD CABIN EQUIPMENT - STA 276 TO 368.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <p>1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE</p>		

5 8 9 7	EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INT STRUCTURE
			53-662-00 PAGE 1 OF 3 APR 22/08

MECH	INSP

- BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)

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EFFECTIVITY	_____	GEN VISUAL	FUSELAGE UPPER LOBE INT STRUCTURE
			53-662-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-662-00
AIRLINE CARD NO.

MECH	INSP

- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY



GEN VISUAL

FUSELAGE UPPER LOBE INT STRUCTURE

53-662-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-664-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FLIGHT DECK SECURITY DOOR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 221 222			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-664-00
<p>DETAILED - INTERNAL: FLIGHT DECK DOOR PANEL ASSEMBLY, BREAK AWAY PANEL, DOOR POST ASSEMBLY (RIGHT), ARMORED POST COVER ASSEMBLY (RIGHT & LEFT), PARTITION ASSEMBLY (LEFT POST), AND FLIGHT DECK DOOR AND DOOR FRAME TRIM.</p> <p>AIRPLANE NOTE: APPLICABLE TO AIRPLANES PRODUCTION LINE NUMBER 895 AND ON, OR INCORPORATION OF BOEING SB 767-25-0325, SB 767-25-0327 OR SB 767-25-0332.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

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			53-664-00 PAGE 1 OF 2 APR 22/08

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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

FLIGHT DECK SECURITY DOOR

53-664-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-666-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD ENTRY/SERVICE DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 221		ACCESS PANELS 2215 831				

MECH	INSP	MPD ITEM NUMBER 53-666-00
		<p>DETAILED - INTERNAL: FWD ENTRY/SERVICE DOOR CUTOUT LOWER SILL UPPER SURFACE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES REMOVAL OF THE ENTRY/SERVICE DOORWAY SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

EFFECTIVITY	DETAILED IN	FWD ENTRY/SERVICE DOOR LOWER SILL
		53-666-00 PAGE 1 OF 2 APR 22/08

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- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	FWD ENTRY/SERVICE DOOR LOWER SILL
			53-666-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-674-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE FWD ENTRY/SERVICE DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 222			ACCESS PANELS 2215 841			

MECH	INSP	MPD ITEM NUMBER 53-674-00
<p>DETAILED - INTERNAL: FWD ENTRY/SERVICE DOOR CUTOUT LOWER SILLS UPPER SURFACE.</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>NOTE: INCLUDES DEACTIVATED DOOR.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES REMOVAL OF THE ENTRY/SERVICE DOORWAY SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

5 9 0 4	EFFECTIVITY	DETAILED IN	FWD ENTRY/SERVICE DOOR LOWER SILL
			53-674-00 PAGE 1 OF 2 APR 22/08

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6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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FWD ENTRY/SERVICE DOOR LOWER SILL

53-674-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-678-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 12C	PHASE S 4C	34848	MPD REV 012	TASK CARD REVISION DEC 22/08
TASK GEN VISUAL	TITLE UPPER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL		
ZONES 223 224		ACCESS PANELS 2003 2222 2231					

MECH	INSP	MPD ITEM NUMBER 53-678-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, BS 243.5/246 TO BS 434, BETWEEN S-12 AND BL 0.0, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, BONDED TEAR STRAPS, STRINGER SPLICES, AND RIGID CARGO BARRIER ATTACHMENT STRUCTURE TO FUSELAGE FROM BL 0 TO STRINGER S-9.</p> <p>AIRPLANE NOTE: INSPECTION IS APPLICABLE TO ALL AIRCRAFT. HOWEVER, RIGID CARGO BARRIER IS INSTALLED ONLY ON FRTR.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

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			53-678-00 PAGE 1 OF 2 DEC 22/08

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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

UPPER LOBE INTERIOR

53-678-00

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-682-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE 2-7		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 231		ACCESS PANELS 2003 2215 2221 2312 2313 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-682-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, INCLUDING THE MAIN DECK FLOOR STRUCTURE, BS 434 TO BS 785.9, BETWEEN S-20L AND S-12L, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAME AND SILLS WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, STUB BEAM TO WING UPPER PANEL INTERFACE, DOOR CUTOUT STRUCTURE. ACCESS NOTE: 1) FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION. 2) OUTBOARD EQUIPMENT AND SIDEWALL PANELS. ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05. ACCESS NOTE: SPECIAL ACCESS 2312 REQUIRES THE REMOVAL OF THE MID CABIN EQUIPMENT STA 764 TO 785.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS 		

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-682-00 PAGE 1 OF 3 APR 22/08

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FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)

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EFFECTIVITY

GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

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SAS  **BOEING**
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TASK CARD

BOEING CARD NO. 53-682-00
AIRLINE CARD NO.

MECH	INSP	
		<ul style="list-style-type: none"> - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-682-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-684-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE MID-CABIN ENTRY/SVC DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 231			ACCESS PANELS 2215 835			

MECH	INSP	MPD ITEM NUMBER 53-684-00
<p>DETAILED - INTERNAL: MID-CABIN ENTRY/SERVICE DOOR CUTOUT LOWER SILLS UPPER SURFACE.</p> <p>AIRPLANE NOTE: AIRPLANES WITH MID-CABIN DOORS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES THE REMOVAL OF ENTRY/SERVICE DOOR SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

EFFECTIVITY	DETAILED IN	MID-CABIN ENTRY/SVC DOOR LOWER SILL
		53-684-00 PAGE 1 OF 2 APR 22/08

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AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	MID-CABIN ENTRY/SVC DOOR LOWER SILL
			53-684-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-692-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR	STRUCTURAL ILLUSTRATION REFERENCE 2-7			APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 232		ACCESS PANELS 2003 2215 2221 2312 2322 2323 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-692-00					
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE AND INCLUDING MAIN DECK FLOOR STRUCTURE, BS 434 TO BS 785.9, BETWEEN S-20R AND S-12R INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAMES AND SILLS WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, STUB BEAM TO WING UPPER PANEL INTERFACE, DOOR CUTOUT STRUCTURE.</p> <p>ACCESS NOTE: OUTBOARD SEAT SETS, CARPET RISERS AND GRILLES FROM BODY STA 456 TO STA 654.</p> <p>ACCESS NOTE: 1) FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION. 2) OUTBOARD EQUIPMENT AND SIDEWALL PANELS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2322 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 456 TO 785.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., 							

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-692-00 PAGE 1 OF 3 APR 22/08

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GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)

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EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
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BOEING CARD NO. 53-692-00
AIRLINE CARD NO.

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		<ul style="list-style-type: none"> - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY



GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-694-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE MID-CABIN ENTRY/SVC DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 232			ACCESS PANELS 2215 845			

MECH	INSP	MPD ITEM NUMBER 53-694-00
<p>DETAILED - INTERNAL: MID-CABIN ENTRY/SERVICE DOOR CUTOUT LOWER SILLS UPPER SURFACE. AIRPLANE NOTE: AIRPLANES WITH MID-CABIN DOORS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES THE REMOVAL OF ENTRY/SERVICE DOOR SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE 		

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			53-694-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
 WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	MID-CABIN ENTRY/SVC DOOR LOWER SILL
			53-694-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-698-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE 2-8		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 233		ACCESS PANELS 2003 2222 2331					

MECH	INSP	MPD ITEM NUMBER 53-698-00
		<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE BS 434 TO BS 785.9, BETWEEN S-12L AND BL O.O, LAP JOINTS, CIRCUMFERENTIAL JOINTS, FRAMES, BONDED TEAR STRAPS, STRINGER SPLICES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2331 REQUIRES THE REMOVAL OF THE SCULPTURED CEILING PANEL STA 434 TO 785.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
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6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

53-698-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-700-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE 2-8		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 234		ACCESS PANELS 2003 2222 2331					

MECH	INSP	MPD ITEM NUMBER 53-700-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, BS 434 TO BS 785.9, BETWEEN S-12R AND BL O.O, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, FRAMES, BONDED TEAR STRAPS, STRINGER SPLICES.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2331 REQUIRES THE REMOVAL OF THE SCULPTURED CEILING PANEL STA 434 TO 785.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. 		

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			53-700-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-704-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INT		STRUCTURAL ILLUSTRATION REFERENCE 1-21 2-10 2-9			APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 241 242		ACCESS PANELS 2003 2411 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-704-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, INCLUDING MAIN DECK FLOOR STRUCTURE, BS 785.9 TO BS 1065, BETWEEN S-20 AND S-12, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAMES AND SILLS WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, STUB BEAM TO WING UPPER PANEL INTERFACE, AND HATCH CUTOUT STRUCTURE, AND BODY WINDOW FORGINGS IN AREAS OF ATTACH FASTENERS BETWEEN BODY STA. 785.9 AND STA. 955.1.</p> <p>ACCESS NOTE: FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION (EXCEPT UNDER GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES).</p> <p>ACCESS NOTE: CABIN EQUIPMENT BETWEEN BODY STA 785.9 AND 999.</p> <p>ACCESS NOTE: CABIN EQUIPMENT AND SIDEWALL PANELS AT BS 785, 955, AND 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH 		

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INT
		53-704-00 PAGE 1 OF 3 APR 22/08

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MECH	INSP
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- STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER

5 9 2 3	EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INT
			53-704-00 PAGE 2 OF 3 APR 22/08

MECH	INSP

- REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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4

EFFECTIVITY



GEN VISUAL

FUSELAGE UPPER LOBE INT

53-704-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-706-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE PASS CABIN ABOVE CEILING 785-1065	STRUCTURAL ILLUSTRATION REFERENCE 2-8	APPLICABILITY AIRPLANE ALL ENGINE ALL				
ZONES 243 244		ACCESS PANELS 2003 2431					

MECH	INSP	MPD ITEM NUMBER 53-706-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, BS 785.9 TO BS 1065, BETWEEN S-12L AND S-12R, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, BONDED TEAR STRAPS, STRINGERS SPLICES. BODY BULKHEAD AND ADJACENT SKIN AND STRINGER SPLICES AT BODY STA. 785.9, 955, AND 1065 (FORWARD AND AFT SIDES) BETWEEN STRINGER S-12L AND S-12R.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> 1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. 2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. 3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. 4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

EFFECTIVITY	GEN VISUAL	PASS CABIN ABOVE CEILING 785-1065
		53-706-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

PASS CABIN ABOVE CEILING 785-1065

53-706-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-710-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR	STRUCTURAL ILLUSTRATION REFERENCE 2-11			APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 251		ACCESS PANELS 2003 2215 2511 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-710-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, INCLUDING MAIN DECK FLOOR STRUCTURE, BS 1065 TO BS 1636, BETWEEN S-20L AND S-12L, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAMES AND SILLS WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, FORWARD SIDE OF BS 1582 AFT PRESSURE BULKHEAD, DOOR CUTOUT STRUCTURE.</p> <p>ACCESS NOTE: FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION (EXCEPT UNDER GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES).</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE 		

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-710-00 PAGE 1 OF 3 APR 22/08

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- ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
 5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F

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EFFECTIVITY

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GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

MECH	INSP	
		<p>- SELECTED AREAS NOTED. 7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-710-00 PAGE 3 OF 3 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-712-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT ENTRY/SERVICE DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE 2-13	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 251			ACCESS PANELS 2215 833			

MECH	INSP	MPD ITEM NUMBER 53-712-00
<p>DETAILED - INTERNAL: AFT ENTRY/SERVICE DOOR CUTOUT LOWER SILL UPPER SURFACE.</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>NOTE: INCLUDES DEACTIVATED DOORS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES THE REMOVAL OF THE ENTRY/SERVICE DOORWAY SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

5 9 3 0	EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR LOWER SILL
			53-712-00 PAGE 1 OF 2 APR 22/08

MECH	INSP	
		<p>6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):</p> <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23). <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35). <p>B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):</p> <ul style="list-style-type: none"> - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II) <p>OR</p> <ul style="list-style-type: none"> - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35). <p>C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:</p> <ul style="list-style-type: none"> - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS) - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1) - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED) - AREAS WITH ELECTRICAL ARC POTENTIAL - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES) - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

5 9 3 1	EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR LOWER SILL
			53-712-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-714-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE FUSELAGE UPPER LOBE INTERIOR		STRUCTURAL ILLUSTRATION REFERENCE 2-11		APPLICABILITY AIRPLANE ENGINE ALL ALL		
ZONES 252		ACCESS PANELS 2003 2215 2511 NOTE					

MECH	INSP	MPD ITEM NUMBER 53-714-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, INCLUDING MAIN DECK FLOOR STRUCTURE, BS 1065 TO BS 1636, BETWEEN S-20R AND S-12R, INCLUDING LAP JOINTS, CIRCUMFERENTIAL JOINTS, WINDOW FORGINGS, BONDED TEAR STRAPS, STRINGER SPLICES, SEAT TRACKS AND FLOOR SUPPORT STRUCTURE (EXCEPT UNDER GALLEYS AND LAVS), FRAMES AND SILL WITHIN 22 INCHES OF DOORS AND ESCAPE HATCHES, FORWARD SIDE OF BS 1582 AFT PRESSURE BULKHEAD, AND DOOR CUTOUT STRUCTURE.</p> <p>ACCESS NOTE: SEAT SETS, CARPET RISER, GRILLES STA 1263 TO 1417.</p> <p>ACCESS NOTE: FLOOR PANEL REMOVAL IS REQUIRED FOR SEAT TRACK AND FLOOR SUPPORT STRUCTURE INSPECTION (EXCEPT UNDER GALLEYS AND LAVS, AND NEAR DOORS AND ESCAPE HATCHES).</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2511 REQUIRES THE REMOVAL OF AFT CABIN EQUIPMENT STA 1065 TO 1569.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., 		

EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-714-00 PAGE 1 OF 3 APR 22/08

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MECH	INSP
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GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.

4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)

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EFFECTIVITY	GEN VISUAL	FUSELAGE UPPER LOBE INTERIOR
		53-714-00 PAGE 2 OF 3 APR 22/08

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TASK CARD

BOEING CARD NO. 53-714-00
AIRLINE CARD NO.

MECH	INSP	
		<ul style="list-style-type: none"> - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES) - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F - SELECTED AREAS NOTED. <p>7. REINSTALL DRY INSULATION BLANKETS.</p>

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EFFECTIVITY



GEN VISUAL

FUSELAGE UPPER LOBE INTERIOR

53-714-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-716-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT ENTRY/SERVICE DOOR LOWER SILL		STRUCTURAL ILLUSTRATION REFERENCE 2-13	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 252			ACCESS PANELS 2215 843			

MECH	INSP	MPD ITEM NUMBER 53-716-00
<p>DETAILED - INTERNAL: AFT ENTRY/SERVICE DOOR CUTOUT LOWER SILLS UPPER SURFACE.</p> <p>AIRPLANE NOTE: APPLICABLE TO PASS, SF, AND BCF AIRPLANES.</p> <p>NOTE: INCLUDES DEACTIVATED DOORS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2215 REQUIRES THE REMOVAL OF THE ENTRY/SERVICE DOORWAY SCUFF PLATE.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

EFFECTIVITY	DETAILED IN	AFT ENTRY/SERVICE DOOR LOWER SILL
		53-716-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

AFT ENTRY/SERVICE DOOR LOWER SILL

53-716-00

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-720-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 003	TASK CARD REVISION APR 22/08
TASK IN SURVEILL	TITLE FUSELAGE UPPER LOBE INT	STRUCTURAL ILLUSTRATION REFERENCE 2-8	APPLICABILITY AIRPLANE ALL	APPLICABILITY ENGINE ALL			
ZONES 253 254	ACCESS PANELS 2003 2222 2531 NOTE						

MECH	INSP	MPD ITEM NUMBER 53-720-00
<p>GENERAL VISUAL - INTERNAL: FUSELAGE UPPER LOBE INTERIOR STRUCTURE, BS 1065 TO BS 1636, BETWEEN S-12L AND S-12R, INCLUDING LAP SPLICES, CIRCUMFERENTIAL JOINTS, FRAMES, BONDED TEAR STRAPS, STRINGER SPLICES, AND FORWARD SIDE OF BS 1582 AFT PRESSURE BULKHEAD.</p> <p>ACCESS NOTE: CEILING PANELS FROM STA 1065 TO 1263.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2531 REQUIRES REMOVAL OF SCUPLTURED CEILING PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL 		

5 9 3 7	EFFECTIVITY	IN SURVEILL	FUSELAGE UPPER LOBE INT
			53-720-00 PAGE 1 OF 2 APR 22/08

MECH	INSP
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- FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
 6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
 - A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
 - B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
 - C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	IN SURVEILL	FUSELAGE UPPER LOBE INT
		53-720-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-726-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT BODY, STA. 1702 - 1725.5		STRUCTURAL ILLUSTRATION REFERENCE 3-3 3-5		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 311 312 313 314			ACCESS PANELS 312AR			

MECH	INSP	MPD ITEM NUMBER 53-726-00
<p>DETAILED - INTERNAL: FORWARD AND AFT SIDES OF BODY STA. 1702 AND STA. 1725.5 BULKHEADS BETWEEN LBL 15 AND RBL 15.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 9 3 9	EFFECTIVITY	DETAILED IN	AFT BODY, STA. 1702 - 1725.5
			53-726-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

AFT BODY, STA. 1702 - 1725.5

53-726-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-728-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE H STABILIZER TRIM ACTUATOR SUPPORT		STRUCTURAL ILLUSTRATION REFERENCE 3-4	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 311 312			ACCESS PANELS 312AR			

MECH	INSP	MPD ITEM NUMBER 53-728-00
<p>DETAILED - INTERNAL: HORIZONTAL STABILIZER TRIM ACTUATOR SUPPORT FITTING IN AREA OF LUGS AND ATTACHMENT TO BODY STA 1702 BULKHEAD.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 9 4 1	EFFECTIVITY	DETAILED IN	H STABILIZER TRIM ACTUATOR SUPPORT
			53-728-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
- WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	_____	DETAILED IN	H STABILIZER TRIM ACTUATOR SUPPORT
			53-728-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-730-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE EXT SECTION 48, LEFT SIDE		STRUCTURAL ILLUSTRATION REFERENCE 3-9	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 311 313			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-730-00
<p>GENERAL VISUAL - EXTERNAL: EXTERIOR OF SECTION 48, LEFT SIDE, BS 1582 THROUGH BS 1832/1843</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	GEN VISUAL	EXT SECTION 48, LEFT SIDE
		53-730-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

EXT SECTION 48, LEFT SIDE

53-730-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-732-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL NOTE	PHASE 12424	MPD REV 008	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE AFT PRESSURE BULKHEAD - STA 1582		STRUCTURAL ILLUSTRATION REFERENCE 3-1 3-2 3-32		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 311 312			ACCESS PANELS 312AR			

MECH	INSP	MPD ITEM NUMBER 53-732-00
<p>DETAILED - INTERNAL: AFT PRESSURE BULKHEAD, BODY STA 1582, AFT SURFACE FOR DAMAGE; DENTS, SCRATCHES, NICKS, GOUGES OR CORROSION, INCLUDING CRACKS AT SPLICES, DOUBLERS, AND AROUND ALL CUTOUTS.</p> <p>INTERVAL NOTE: AIRPLANE LINE NUMBERS 1 THROUGH 175 UPON REACHING 25,000 CYCLES REPEAT EVERY 1800 CYCLES PER SB 767-53-0026.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. 		

EFFECTIVITY	DETAILED IN	AFT PRESSURE BULKHEAD - STA 1582
		53-732-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):
- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
- OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

AFT PRESSURE BULKHEAD - STA 1582

53-732-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-734-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL S 4C	PHASE 14848	MPD REV 003	TASK CARD REVISION DEC 22/08
TASK CHECK/INSP		TITLE VERTICAL STABILIZER ATTACH BOLT		STRUCTURAL ILLUSTRATION REFERENCE 3-5	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 311 312 313 314			ACCESS PANELS 311AZ 312AR			

MECH	INSP	MPD ITEM NUMBER 53-734-00
<p>CHECK/INSP - INTERNAL: VERTICAL STABILIZER ATTACH BOLTS (14 BOLTS IN ZONE 311/312 AND 2 BOLTS IN ZONE 313/314). INSPECT BY TORQUE CHECK WITH PER SERVICE BULLETIN 767-53A0085 INSTRUCTIONS.</p> <p>NOTE: REPORT TO BOEING IF ANY FASTENER DOES NOT MEET THIS TORQUE REQUIREMENT WITHIN A TOLERANCE OF 10 PERCENT.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): 		

5	EFFECTIVITY	CHECK/INSP	VERTICAL STABILIZER ATTACH BOLT
9			53-734-00 PAGE 1 OF 2 DEC 22/08
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MECH	INSP

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	CHECK/INSP	VERTICAL STABILIZER ATTACH BOLT
		53-734-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-740-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE EXT OF SECTION 48, RIGHT SIDE		STRUCTURAL ILLUSTRATION REFERENCE 3-9	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 312 314			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-740-00
<p>GENERAL VISUAL - EXTERNAL: EXTERIOR OF SECTION 48, RIGHT SIDE, BS 1582 THROUGH BS 1832/1843.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5	EFFECTIVITY	GEN VISUAL	EXT OF SECTION 48, RIGHT SIDE
9			53-740-00 PAGE 1 OF 2 APR 22/08
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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
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WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

EXT OF SECTION 48, RIGHT SIDE

53-740-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-746-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 011	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE H STABILIZER PIVOT FITTINGS		STRUCTURAL ILLUSTRATION REFERENCE 3-6	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 313 314			ACCESS PANELS 312AR 313AL			

MECH	INSP	MPD ITEM NUMBER 53-746-00
<p>DETAILED - INTERNAL: HORIZONTAL STABILIZER PIVOT FITTINGS IN AREAS OF ATTACHMENT TO THE BODY STA. 1809.5 BULKHEAD, LUGS FOR PIVOT PIN, THE BACKUP FITTINGS, AND THE PIVOT NUT LOCKWIRE.</p> <p>NOTE: SEVERAL OPERATORS HAVE REPORTED INSTANCES OF BEARING RETAINER ROTATION. SERVICE BULLETIN 767-55-0008 IS ASSOCIATED WITH THIS CONDITION.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS 		

5 9 5 1	EFFECTIVITY	DETAILED IN	H STABILIZER PIVOT FITTINGS
			53-746-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

- A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
 - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
 - APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
 - OR
 - APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED. WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
 - CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
 - LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
 - OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
 - ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
 - AREAS WITH ELECTRICAL ARC POTENTIAL
 - INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
 - ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
 - FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
 - SELECTED AREAS NOTED.

- 7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

DETAILED IN

H STABILIZER PIVOT FITTINGS

53-746-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-748-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL S 2C	PHASE 12424	MPD REV 002	TASK CARD REVISION APR 22/08
TASK DETAILED IN		TITLE BODY LONGERONS AT HORIZ STAB CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE 3-7 3-8		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 313 314			ACCESS PANELS 312AR 313AL			

MECH	INSP	MPD ITEM NUMBER 53-748-00
<p>DETAILED - INTERNAL: BODY UPPER AND LOWER LONGERONS ADJACENT TO THE HORIZONTAL STABILIZER CUTOUT.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 9 5 3	EFFECTIVITY	DETAILED IN	BODY LONGERONS AT HORIZ STAB CUTOUT
			53-748-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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DETAILED IN

BODY LONGERONS AT HORIZ STAB CUTOUT

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-750-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 12C	PHASE S 4C	34848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE SECTION 48 EXTERIOR - LEFT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL		
ZONES 311 330		ACCESS PANELS 313BLX 313CLX					

MECH	INSP	MPD ITEM NUMBER 53-750-00
<p>GENERAL VISUAL - EXTERNAL: EXTERNAL SECTION 48 SKIN AND HORIZONTAL STABILIZER SKIN UNDER FAIRINGS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

5 9 5 5	EFFECTIVITY	GEN VISUAL	SECTION 48 EXTERIOR - LEFT
			53-750-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

SECTION 48 EXTERIOR - LEFT
53-750-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-756-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 12C	PHASE S 4C	34848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE SECTION 48 EXTERIOR - RIGHT			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 312 340			ACCESS PANELS 314BRX 314CRX				

MECH	INSP	MPD ITEM NUMBER 53-756-00	
		<p>GENERAL VISUAL - EXTERNAL: EXTERNAL SECTION 48 SKIN AND HORIZONTAL STABILIZER SKIN UNDER FAIRINGS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>	

5 9 5 7	EFFECTIVITY	GEN VISUAL	SECTION 48 EXTERIOR - RIGHT
			53-756-00 PAGE 1 OF 2 APR 22/08

MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

SECTION 48 EXTERIOR - RIGHT

53-756-00

PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-758-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA APU COMPT	RELATED TASK	INTERVAL S 1C	PHASE 11212	MPD REV 003	TASK CARD REVISION APR 22/08
TASK GEN VISUAL		TITLE APU COMPARTMENT EXTERNAL SURFACE		STRUCTURAL ILLUSTRATION REFERENCE 3-9 3-10		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 315 316			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-758-00
<p>GENERAL VISUAL - EXTERNAL: APU COMPARTMENT EXTERNAL SURFACE BETWEEN BODY STA 1832/1843 AND STA 1952.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	GEN VISUAL	APU COMPARTMENT EXTERNAL SURFACE
		53-758-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY	GEN VISUAL	APU COMPARTMENT EXTERNAL SURFACE
		53-758-00 PAGE 2 OF 2 APR 22/08

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-762-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA	RELATED TASK	INTERVAL S 8C	PHASE S 4C	24848	MPD REV 012	TASK CARD REVISION APR 22/08
TASK GEN VISUAL	TITLE SECTION 48 EXTERIOR			STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 320		ACCESS PANELS 321BL 321BR 322AL 322AR 323AL 323AR 324AL 324AR					

MECH	INSP	MPD ITEM NUMBER 53-762-00
<p>GENERAL VISUAL - INTERNAL: EXTERIOR OF SECTION 48 SKIN AND VERTICAL STABILIZER SKIN UNDER FAIRINGS / ACCESS PANELS.</p> <p>THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:</p> <ol style="list-style-type: none"> REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3). IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)): <p>A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C): - APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND</p>		

EFFECTIVITY	GEN VISUAL	SECTION 48 EXTERIOR
		53-762-00 PAGE 1 OF 2 APR 22/08

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MECH	INSP

- DISPLACES MOISTURE (E.G., BMS 3-23).
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).
- B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)
OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).
- C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.
7. REINSTALL DRY INSULATION BLANKETS.

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EFFECTIVITY

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GEN VISUAL

SECTION 48 EXTERIOR

53-762-00

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-778-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL 25000 CYC	PHASE S 2C	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK DETAILED IN		TITLE BODY STA 1809.5 BHD OUTBD CHORD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 313 314		ACCESS PANELS 313AL				

MECH	INSP	MPD ITEM NUMBER 53-778-12
		<p>DETAILED - INTERNAL: BS 1809.5 BULKHEAD OUTBOARD CHORD (TYPE 1 STRUCTURE).</p> <p>AIRPLANE NOTE: APPLICABLE TO ALL. FOR AIRPLANE LINE NUMBERS 1-709 THIS INSPECTION IS SATISFIED BY THE INSPECTIONS IN SERVICE BULLETIN 767-53-0078.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-80-I15, -I16, -I17 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: INSPECTION TO BE DONE AT THE FOLLOWING AREAS: FWD OF STA 1809, WL240 TO 250 AFT OF STA 1809, WL240 TO 260</p>

EFFECTIVITY	DETAILED IN	BODY STA 1809.5 BHD OUTBD CHORD
		53-778-12 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-778-14
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	STABLIZR BX		25000 CYC	S 2C	329XX	014 AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		BODY STA 1809.5 BHD OUTBD CHORD				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
313 314		313AL				

MECH	INSP		MPD ITEM NUMBER
		<p>SPECIAL DETAILED - INTERNAL: BS 1809.5 BULKHEAD OUTBOARD CHORD (TYPE 1 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO ALL. FOR AIRPLANE LINE NUMBERS 1-709 THIS INSPECTION IS SATISFIED BY THE INSPECTIONS IN SERVICE BULLETIN 767-53-0078</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-80-I15, -I16, -I17 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-80-04.</p> <p>NOTE: INSPECTION TO BE DONE AT THE FOLLOWING AREAS: FWD OF STA 1809, WL240 TO 250 - HFEC, WL250 TO 260 - LFEC AFT OF STA 1809, WL240 TO 260 - HFEC</p>	53-778-14

EFFECTIVITY		SPCL DETAIL	BODY STA 1809.5 BHD OUTBD CHORD
			53-778-14 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-778-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE HORIZONTAL STABILIZER PIVOT FITTINGS		STRUCTURAL ILLUSTRATION REFERENCE 3-6	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 313 314			ACCESS PANELS 313AL			

MECH	INSP	MPD ITEM NUMBER 53-778-16
		<p>SPECIAL DETAILED - INTERNAL: SECTION 48- PIVOT FITTING LUG. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-80-113 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-80-02.</p> <p>NOTE: INSPECT BOTH BACK-TO-BACK LUGS.</p>

EFFECTIVITY	SPCL DETAIL	HORIZONTAL STABILIZER PIVOT FITTINGS
		53-778-16 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS **BOEING**
767
TASK CARD

BOEING CARD NO. 53-778-18
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	STABLIZR BX		25000 CYC	S 4C	329XX	014 DEC 22/06
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		HORIZONTAL STABILIZER PIVOT PINS		3-6		AIRPLANE ENGINE
ZONES		ACCESS PANELS				
313 314		3131 313AL				

MECH	INSP		MPD ITEM NUMBER
		<p>SPECIAL DETAILED - INTERNAL: HORIZONTAL STABILIZER FITTING HINGE PINS - STA 1809, BL41.50 (TYPE 1 STRUCTURE.)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-125 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: SB 767-55-0005. THRESHOLD APPLIES AFTER TERMINATING ACTION IS ACCOMPLISHED PER SB.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-80-03. THIS TASK REQUIRES REMOVAL OF BOTH INNER AND OUTER PINS. VISUAL INSPECTION OF OUTER SURFACES IS REQUIRED.</p> <p>NOTE: REMOVE AND INSPECT THE INNER SURFACE OF THE OUTER PIN AND INNER PIN WITH HFEC PROBE AND VISUALLY INSPECT OUTER SURFACE OF EACH PIN. BOTH INNER AND OUTER PINS NEED TO BE REMOVED. NDI PROCEDURE IS FOR NDI PART OF THE INSPECTION. CARRY OUT VISUAL INSPECTION OF OUTER SURFACES OF BOTH PINS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 3131 REQUIRES REMOVAL OF HORIZONTAL STABILIZER INNER AND OUTER HINGE PINS AS PER MAINTENANCE MANUAL REFERENCE 55-17-51.</p>	53-778-18

EFFECTIVITY		SPCL DETAIL	HORIZONTAL STABILIZER PIVOT PINS
			53-778-18 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-778-20
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL 25000 CYC	PHASE S 2C	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK DETAILED EX		TITLE BODY STA 1809.5 BHD OUTBD CHORD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 313 314			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-778-20
		<p>DETAILED - EXTERNAL: BS 1809.5 BULKHEAD OUTBOARD CHORD (TYPE 1 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO ALL. FOR AIRPLANE LINE NUMBERS 1-709 THIS INSPECTION IS SATISFIED BY THE INSPECTIONS IN SERVICE BULLETIN 767-53-0078.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-80-I15, -I16, -I17 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: INSPECTION TO BE DONE AT THE FOLLOWING AREAS: FWD OF STA 1809, WL240 TO 250 AFT OF STA 1809, WL240 TO 260</p>

EFFECTIVITY	DETAILED EX	BODY STA 1809.5 BHD OUTBD CHORD
		53-778-20 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-780-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STABLIZR BX	RELATED TASK	INTERVAL 25000 CYC	PHASE S 2C	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE BODY STA 1809.5 BHD OUTBD CHORD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 313 314			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-780-00
		<p>SPECIAL DETAILED - EXTERNAL: BS 1809.5 BULKHEAD OUTBOARD CHORD (TYPE 1 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO ALL. FOR AIRPLANE LINE NUMBERS 1-709 THIS INSPECTION IS SATISFIED BY THE INSPECTIONS IN SERVICE BULLETIN 767-53-0078.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-80-I15, -I16, -I17 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-80-04.</p> <p>NOTE: INSPECTION TO BE DONE AT THE FOLLOWING AREAS: FWD OF STA 1809, WL250 TO 260 - LFEC</p>

EFFECTIVITY	SPCL DETAIL	BODY STA 1809.5 BHD OUTBD CHORD
		53-780-00 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-780-02
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 2C	99XXX	014	APR 22/08
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		AFT PRESSURE DOME BS 1582				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
311 312		312AR				

MECH	INSP	MPD ITEM NUMBER				
		<p>SPECIAL DETAILED - INTERNAL: 53-780-02</p> <p>AFT PRESSURE BULKHEAD - CIRCUMFERENTIAL TEAR STRAP SPLICE. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-101E FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6 SUBJECT 53-80-01.</p>				

EFFECTIVITY	SPCL DETAIL	AFT PRESSURE DOME BS 1582
		53-780-02 PAGE 1 OF 1 APR 22/08

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STATION
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BOEING CARD NO. 53-780-04
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION APR 22/08
TASK SPCL DETAIL		TITLE AFT PRESSURE DOME BS 1582		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 311 312		ACCESS PANELS 312AR				

MECH	INSP	MPD ITEM NUMBER 53-780-04
		<p>SPECIAL DETAILED - INTERNAL: AFT PRESSURE BULKHEAD - CIRCUMFERENTIAL TEAR STRAP SPLICE. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-101D FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6 SUBJECT 53-80-02.</p>

EFFECTIVITY	SPCL DETAIL	AFT PRESSURE DOME BS 1582
		53-780-04 PAGE 1 OF 1 APR 22/08

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-780-06
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE AFT PRESSURE DOME BS 1582		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 311 312			ACCESS PANELS 312AR			

MECH	INSP	MPD ITEM NUMBER 53-780-06
		<p>SPECIAL DETAILED - INTERNAL: AFT PRESSURE BULKHEAD - RADIAL WEB LAP SPLICES. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-101C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6 SUBJECT 53-80-01.</p>

EFFECTIVITY	SPCL DETAIL	AFT PRESSURE DOME BS 1582
		53-780-06 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-780-08
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 2C	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		AFT PRESSURE DOME BS 1582				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
311 312		312AR				

MECH	INSP	MPD ITEM NUMBER				
		53-780-08				
<p>SPECIAL DETAILED - INTERNAL: AFT PRESSURE BULKHEAD - RADIAL WEB LAP SPLICES. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-101B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-80-02</p>						

EFFECTIVITY	SPCL DETAIL	AFT PRESSURE DOME BS 1582
		53-780-08 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-780-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA STAB COMPT	RELATED TASK	INTERVAL NOTE	PHASE S 4C 1/5 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE H STABILIZ TRIM ACTUATOR		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 311 312		ACCESS PANELS 312AR NOTE				

MECH	INSP	MPD ITEM NUMBER 53-780-10
<p>SPECIAL DETAILED - INTERNAL: SECTION 48- JACK SCREW FITTING LUG- UPPER BULKHEAD AT STA 1702. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>ACCESS NOTE: REQUIRES ACTUATOR ATTACH PIN REMOVAL.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-103 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-80-01.</p>		

EFFECTIVITY	SPCL DETAIL	H STABILIZ TRIM ACTUATOR
		53-780-10 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-780-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE AFT ENTRY/SERVICE DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 251 252		ACCESS PANELS 2003 833 843				

MECH	INSP	MPD ITEM NUMBER 53-780-16
		<p>SPECIAL DETAILED - INTERNAL: AFT PASSENGER ENTRY/SERVICE DOOR CUTOUT- UPPER MAIN SILL. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-I14F FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-60-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>NOTE: HFEC INSPECT FROM BOTH DIRECTIONS.</p>

EFFECTIVITY	SPCL DETAIL	AFT ENTRY/SERVICE DOOR CUTOUT
		53-780-16 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-780-18
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 2C	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		AFT ENTRY/SERVICE DOOR CUTOUT				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
251 252		2003 833 843				

MECH	INSP	MPD ITEM NUMBER				
		53-780-18				
<p>SPECIAL DETAILED - INTERNAL: AFT PASSENGER ENTRY/SERVICE DOOR CUTOUT- EDGE FRAMES- FRAMES AT DOOR STA 876.3, S-9 TO S-11 AND S-15 TO S-19 AND AT STA 923.7, S-8 TO S-13. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-I14E FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>NOTE: INSPECTIONS ARE REQUIRED ARE REQUIRED OF BOTH THE INNER CHORD AND THE STRAP.</p>						

EFFECTIVITY	SPCL DETAIL	AFT ENTRY/SERVICE DOOR CUTOUT
		53-780-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-782-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 1/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE OVERWING EMERGENCY EXIT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-782-12
		<p>SPECIAL DETAILED - INTERNAL: SINGLE EMERGENCY EXIT CUTOUT- SILL OUTER CHORD AT FRAME INTERSECT. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCHES.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-106F FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-50-04.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS: UPPER SILL: STA 859.5. LOWER SILL: STA 883.5.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

5 9 7 6	EFFECTIVITY	SPCL DETAIL	OVERWING EMERGENCY EXIT
			53-782-12 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-782-14
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK DETAILED IN		TITLE OVERWING EMERGENCY EXIT CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-782-14
<p>DETAILED- INTERNAL: SINGLE EMERGENCY EXIT CUTOUT-SILL OUTER CHORD. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCH.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-106E FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p> <p>NOTE: INSPECT STRAP SPECIALLY AROUND CORNERS OF CUTOUTS. INSPECT BOTH UPPER AND LOWER SILL ALONG ENTIRE LENGTH OF CUTOUTS.</p>		

EFFECTIVITY	DETAILED IN	OVERWING EMERGENCY EXIT CUTOUT
		53-782-14 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-782-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 7/10 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE OVERWING EMERGENCY EXIT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-782-16
		<p>SPECIAL DETAILED - INTERNAL: SINGLE EMERGENCY EXIT CUTOUT- SILL INNER CHORDS AT FRAME INTERSECT AT STA 859.5 AND 883.5. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCH.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-106D FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-08.</p> <p>NOTE: INSPECTION AT: UPPER SILL: STA 859.5. LOWER SILL: STA 859.5 AND STA 883.5.</p> <p>NOTE: PERFORM IN CONJUNCTION WITH 5350-241/242-60I (DTR 53-50-106B)</p> <p>NOTE: FASTENER REMOVAL REQUIRED.</p> <p>NOTE: RENEW THRESHOLD OF INSPECTION AFTER FASTENER HOLE OVERSIZE. ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

EFFECTIVITY	SPCL DETAIL	OVERWING EMERGENCY EXIT
		53-782-16 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-782-18
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 2/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL	TITLE OVERWING EMERGENCY EXIT CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-782-18
		<p>SPECIAL DETAILED - INTERNAL: SINGLE EMERGENCY EXIT CUTOUT- SILL INNER CHORD- UPPER SILL AT STA 859.5 - STA 871.5. (TYPE 2 STRUCTURE).</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCH.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-50-106C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-07.</p> <p>NOTE: INSPECTION REQUIRED FOR BOTH CHORD AND STRAP.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

EFFECTIVITY	SPCL DETAIL	OVERWING EMERGENCY EXIT CUTOUT
		53-782-18 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-782-20
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 7/10 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE OVERWING EMERGENCY EXIT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-782-20
		<p>SPECIAL DETAILED - INTERNAL: SINGLE EMERGENCY EXIT CUTOUT- FRAME INNER CHORDS AT SILL INTERSECT AT STA 859.5 AND 883.5. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCH.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-106B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-08.</p> <p>NOTE: INSPECTION AT: UPPER SILL: STA 859.5. LOWER SILL: STA 859.5 AND STA 883.5.</p> <p>NOTE: PERFORM IN CONJUNCTION WITH 5350-241/242-61I (DTR 53-50-106D).</p> <p>NOTE: FASTENER REMOVAL REQUIRED.</p> <p>NOTE: RENEW THRESHOLD OF INSPECTION AFTER FASTENER HOLE OVERSIZE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

5 9 8 0	EFFECTIVITY	SPCL DETAIL	OVERWING EMERGENCY EXIT
			53-782-20 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-784-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 7/10 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE OVERWING EMERGENCY EXIT CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 842			

MECH	INSP	MPD ITEM NUMBER 53-784-00
		<p>SPECIAL DETAILED - INTERNAL: SINGLE EMERGENCY EXIT CUTOUT- FRAME INNER CHORD STA 859.5 STRINGERS S-18 TO S-20 AND STA 883.5 STRINGERS S-16 TO S-20. (TYPE 2 STRUCTURE).</p> <p>AIRPLANE NOTE: 200 AIRPLANES WITH SINGLE OVERWING ESCAPE HATCH.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-50-106A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-07.</p> <p>NOTE: INSPECTION REQUIRED FOR BOTH CHORD AND STRAP.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

5 9 8 1	EFFECTIVITY	SPCL DETAIL	OVERWING EMERGENCY EXIT CUTOUT
			53-784-00 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-784-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 1/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE DUAL OVERWING EMERGENCY EXIT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 834 842 844			

MECH	INSP	MPD ITEM NUMBER 53-784-02
		<p>SPECIAL DETAILED - INTERNAL: DUAL EMERGENCY EXIT CUTOUT- SILL OUTER CHORD AT FRAME INTERSECT. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22G FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-50-04.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS: UPPER SILL: STA 859.5 AND STA 903.5 LOWER SILL: STA 883.5 AND STA 927.5</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

5 9 8 2	EFFECTIVITY	SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
			53-784-02 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-784-04
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK DETAILED IN		TITLE DUAL OVERWING EMERGENCY EXIT			STRUCTURAL ILLUSTRATION REFERENCE	
ZONES 241 242		ACCESS PANELS 2003 2411 2431 832 834 842 844				
MECH	INSP	MPD ITEM NUMBER 53-784-04				

**DETAILED - INTERNAL:
DUAL EMERGENCY EXIT CUTOUT- SILL OUTER CHORD.
(TYPE 2 STRUCTURE).**

**AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH
DUAL OVERWING ESCAPE HATCHES.**

**NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22F
FOR ALTERNATIVE REPEAT INSPECTIONS**

**NOTE: INSPECT STRAP ESPECIALLY AROUND CORNERS OF CUTOUTS.
INSPECT BOTH UPPER AND LOWER SILL ALONG ENTIRE LENGTH OF
CUTOUTS.**

**ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE
CREW CABIN/PASSENGER COMPARTMENT INSULATION
PER MAINTENANCE MANUAL REFERENCE 25-21-05.**

**ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF
MID CABIN EQUIPMENT STA 785 TO 1065.**

**ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF
SCULPTURED CEILING PANELS STA 785 TO 1065.**

EFFECTIVITY

DETAILED IN

DUAL OVERWING EMERGENCY EXIT

53-784-04

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STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-784-06
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION	
AIRPL	FUSELAGE		50000 CYC	S 4C	610XX	014 DEC 22/06	
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY	
SPCL DETAIL		DUAL OVERWING EMERGENCY EXIT				AIRPLANE ENGINE	
						NOTE ALL	
ZONES			ACCESS PANELS				
241 242			2003 2411 2431 832 834 842 844				

MECH	INSP	MPD ITEM NUMBER
		53-784-06
<p>SPECIAL DETAILED - INTERNAL: DUAL EMERGENCY EXIT CUTOUT- UPPER SILL INNER CHORD AT STA 859.5 AND 903.5. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES ONLY.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22E FOR ALTERNATIVE REPEAT INSPECTION.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-09.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>		

5 9 8 4	EFFECTIVITY	SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
			53-784-06 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-784-08
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 7/10 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE DUAL OVERWING EMERGENCY EXIT			STRUCTURAL ILLUSTRATION REFERENCE	
ZONES 241 242		ACCESS PANELS 2003 2411 2431 832 834 842 844				
MECH	INSP	MPD ITEM NUMBER 53-784-08				

SPECIAL DETAILED - INTERNAL:
DUAL EMERGENCY EXIT CUTOUT- SILL INNER CHORD AT FRAME INTERSECT.
(TYPE 2 STRUCTURE)

AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES.

NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22D FOR ALTERNATIVE REPEAT INSPECTIONS.

NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-08.

NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS:
 UPPER SILL: STA 859.5 AND STA 903.5
 LOWER SILL: STA 883.5, STA 903.5 AND STA 927.5.

NOTE: PERFORM IN CONJUNCTION WITH 53-784-12 (DTR 53-50-I22B).

NOTE: FASTENER REMOVAL REQUIRED.

NOTE: RENEW THRESHOLD OF INSPECTION AFTER FASTENER HOLE OVERSIZE.

ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.

ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.

ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.

EFFECTIVITY

SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
53-784-08	PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-784-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 2/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06	
TASK SPCL DETAIL		TITLE DUAL OVERWING EMERGENCY EXIT			STRUCTURAL ILLUSTRATION REFERENCE		
ZONES 241 242		ACCESS PANELS 2003 2411 2431 832 834 842 844					
MECH	INSP					MPD ITEM NUMBER	

SPECIAL DETAILED - INTERNAL:
DUAL EMERGENCY EXIT CUTOUT- SILL INNER CHORD.
(TYPE 2 STRUCTURE)

AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES.

NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22C FOR ALTERNATIVE REPEAT INSPECTION.

NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-07.

NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS:
LOWER SILL: STA 875 TO 894.(ALL)
UPPER SILL: STA 840 TO 871.5 AND STA 885 TO 916 (-200 ONLY)
(FOR -300 UPPER SILL SEE 5350-241/242-56I)

ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.

ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.

ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.

53-784-10

EFFECTIVITY	SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
		53-784-10 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-784-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 7/10 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE DUAL OVERWING EMERGENCY EXIT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 241 242			ACCESS PANELS 2003 2411 2431 832 834 842 844			

MECH	INSP	MPD ITEM NUMBER 53-784-12
		<p>SPECIAL DETAILED - INTERNAL: DUAL EMERGENCY EXIT CUTOUT- FRAME INNER CHORD AT SILL INTERSECT. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-08.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS: UPPER SILL: STA 859.5 AND STA 903.5. LOWER SILL: STA 883.5, STA 903.5 AND STA 927.5.</p> <p>NOTE: PERFORM IN CONJUNCTION WITH 53-784-08 (DTR 53-50-I22D).</p> <p>NOTE: FASTENER REMOVAL REQUIRED.</p> <p>NOTE: RENEW THRESHOLD OF INSPECTION AFTER FASTENER HOLE OVERSIZE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>

5 9 8 7	EFFECTIVITY	SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
			53-784-12 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-784-14
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		50000 CYC	S 4C 7/10 610XX	014	DEC 22/06
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		DUAL OVERWING EMERGENCY EXIT				AIRPLANE ENGINE
						NOTE ALL
ZONES			ACCESS PANELS			
241 242			2003 2411 2431 832 834 842 844			

MECH	INSP	MPD ITEM NUMBER
		53-784-14
<p>SPECIAL DETAILED - INTERNAL: DUAL EMERGENCY EXIT CUTOUT- FRAME INNER CHORDS. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH DUAL OVERWING ESCAPE HATCHES.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I22A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-07.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING AREAS : STA 859.5 FROM STRINGER S-12 TO S-14; STA 883.5 FROM STRINGER S-15 TO S-20; STA 903.5 FROM STRINGER S-11 TO S-20; STA 927.5 FROM STRINGER S-14 TO S-20;</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2431 REQUIRES THE REMOVAL OF SCULPTURED CEILING PANELS STA 785 TO 1065.</p>		

EFFECTIVITY	SPCL DETAIL	DUAL OVERWING EMERGENCY EXIT
		53-784-14 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-784-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA PASS CABIN	RELATED TASK	INTERVAL S 4C 1/5	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE PASS CABIN BODY STA 955.1 BLKHD		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ALL ENGINE ALL
ZONES 241 242			ACCESS PANELS 2411			

MECH	INSP	MPD ITEM NUMBER 53-784-16
		<p>SPECIAL DETAILED - INTERNAL: FUSELAGE REAR SPAR BULKHEAD- BULKHEAD FITTING INNER CHORD STRINGER S-19 TO S-20. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I13C FOR ALTERNATIVE REPEAT INSPECTION.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-04.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2411 REQUIRES THE REMOVAL OF MID CABIN EQUIPMENT STA 785 TO 1065.</p>

EFFECTIVITY	SPCL DETAIL	PASS CABIN BODY STA 955.1 BLKHD
		53-784-16 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-786-08
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		NOTE	S 4C	99XXX	014 AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
GEN VISUAL		CROWN SKIN STRINGERS				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
230 240 250						

MECH	INSP	MPD ITEM NUMBER	
		53-786-08	
<p>GENERAL VISUAL - INTERNAL: FUSELAGE CROWN STRINGERS, STA 654 TO STA 1395, STRINGER S-8L TO S-8R. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I01/53-50-I01/53-60-I01 FOR ALTERNATIVE REPEAT INSPECTIONS.</p>			

EFFECTIVITY	GEN VISUAL	CROWN SKIN STRINGERS
		53-786-08 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-786-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 1/5	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE FRAMES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 230 240 250			ACCESS PANELS 2003 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-786-16
		<p>GENERAL VISUAL - INTERNAL FRAMES ABOVE STRINGER S-20. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYCLES (200, 300PASS) 40000 CYCLES (300F, 400E)</p> <p>ACCESS NOTE: SIDEWALL PANEL, CEILING PANEL AND INSULATION REMOVAL/DISPLACEMENT IS REQUIRED.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I24A/53-50-I01A/53-50-I02A/53-60-I08A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING LOCATIONS: 767-200: FROM STA 786 TO STA 1582. 767-300: FROM STA 654 TO STA 1582. 767-400: FROM STA 434 TO STA 1582.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p>

EFFECTIVITY	GEN VISUAL	FRAMES	53-786-16	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE



SAS **BOEING**

 767

TASK CARD

BOEING CARD NO. 53-786-20
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 2/5	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		CROWN SKIN CIRC SPLICES				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
230 240 250						

MECH	INSP	MPD ITEM NUMBER
		53-786-20
<p>SPECIAL DETAILED - EXTERNAL: FUSELAGE CIRCUMFERENTIAL SPLICES IN THE BODY CROWN- AT SPECIFIED LOCATIONS. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200,300PASS AND 300F AIRPLANES</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I05/ 53-30-I21/53-30-I27/53-50-I03/53-60-I03/53-60-I26 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-30-06.</p> <p>NOTE: INSPECTION TO BE DONE AT FOLLOWING LOCATIONS:</p> <p style="margin-left: 40px;">767-200 : S-7L TO S-7R AT BODY STA 786 S-9L TO S-9R AT BODY STA 1065 S-8L TO S-8R AT BODY STA 1395</p> <p style="margin-left: 40px;">767-300 : S-4L TO S-4R AT BODY STA 654 S-7L TO S-7R AT BODY STA 786 S-9L TO S-9R AT BODY STA 1065 S-4L TO S-4R AT BODY STA 1219</p> <p style="margin-left: 40px;">767-300F: S-4L TO S-4R AT BODY STA 654+22 S-7L TO S-7R AT BODY STA 786 S-9L TO S-9R AT BODY STA 1065 S-4L TO S-4R AT BODY STA 1219 S-4L TO S-4R AT BODY STA 1395</p>		

EFFECTIVITY	SPCL DETAIL	CROWN SKIN CIRC SPLICES
		53-786-20 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-788-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE FWD ENTRY/SERVICE DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 221 222		ACCESS PANELS 2003 831 841				

MECH	INSP	MPD ITEM NUMBER 53-788-00
		<p>SPECIAL DETAILED - INTERNAL: FORWARD PASSENGER ENTRY/SERVICE DOOR CUTOUT- UPPER MAIN SILL (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I10B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-08.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p>

EFFECTIVITY	SPCL DETAIL	FWD ENTRY/SERVICE DOOR CUTOUT
		53-788-00 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-788-02
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 7/10	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		SKIN LONGITUDINAL LAP SPLICES				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
220 230 250		NOTE				

MECH	INSP	MPD ITEM NUMBER				
		53-788-02				
<p>SPECIAL DETAILED - INTERNAL: SKIN LONGITUDINAL LAP SPLICES- LOWER ROW- CROWN AND UPPER LOBE, SECTION 41, 43 AND 46. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I07B/ 53-30-I04B/53-60-I05B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SIDEWALL PANEL, CEILING PANEL AND INSULATION REMOVAL/DISPLACEMENT IS REQUIRED.</p> <p>NOTE: INSPECTION TO BE DONE AT SKIN SPLICES AT STRINGERS S-2R, S-8L/R, S-17L/R FROM STA 246 TO 786 AND FROM 1065 TO 1582.</p>						

EFFECTIVITY	SPCL DETAIL	SKIN LONGITUDINAL LAP SPLICES
		53-788-02 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-788-04
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 7/10	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE FRAMES, LONGITUDINAL SKIN LAP SPLICE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 220 230			ACCESS PANELS 2003 NOTE			

MECH	INSP	MPD ITEM NUMBER 53-788-04
		<p>GENERAL VISUAL - INTERNAL FRAMES, ABOVE S-20. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I08B/53-30-I24B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SIDEWALL PANEL, CEILING PANEL AND INSULATION REMOVAL/DISPLACEMENT IS REQUIRED.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p> <p>NOTE: 767-200: FROM STA 246 TO 786. 767-300: FROM STA 246 TO 654.</p>

EFFECTIVITY	GEN VISUAL	FRAMES, LONGITUDINAL SKIN LAP SPLICE
		53-788-04 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-788-06
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 1/5	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		SKIN LONGITUDINAL LAP SPLICES				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
220 230 250						

MECH	INSP		MPD ITEM NUMBER
		<p>SPECIAL DETAILED - EXTERNAL: SKIN LONGITUDINAL LAP SPLICES- UPPER ROW- CROWN AND UPPER LOBE, SECTION 41, 43 AND 46. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I07A/ 53-30-I04A/53-60-I05A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: SKIN SPLICES AT S-2R, S-8L/R AND S-17L/R FROM STA 246 TO 786 AND 1065 TO 1582.</p>	53-788-06

5 9 9 6	EFFECTIVITY	SPCL DETAIL	SKIN LONGITUDINAL LAP SPLICES
			53-788-06 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE



BOEING CARD NO. 53-788-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CREW CABIN	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL	TITLE CABIN WINDOW NO. 2 POST		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 211 212		ACCESS PANELS 2112 NOTE				

MECH	INSP	MPD ITEM NUMBER 53-788-10
<p>SPECIAL DETAILED - INTERNAL: WINDOW NUMBER 2 CUTOUT STRUCTURE- EF POST. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I16 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-12.</p> <p>ACCESS NOTE: COCKPIT SIDE PANELS REMOVAL IS REQUIRED TO GAIN ACCESS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2112 REQUIRES REMOVAL OF WINDSHIELD POST PADDING.</p>		

EFFECTIVITY	SPCL DETAIL	CABIN WINDOW NO. 2 POST
		53-788-10 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-788-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA CREW CABIN	RELATED TASK	INTERVAL S 4C 3/10	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE CABIN WINDOW NO. 1 POST		STRUCTURAL ILLUSTRATION REFERENCE 2-3	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 211 212			ACCESS PANELS 2112 NOTE			

MECH	INSP		MPD ITEM NUMBER
		<p>SPECIAL DETAILED - INTERNAL: CONTROL CABIN WINDOW NUMBER 1 CUT OUT STRUCTURE- AB POST. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I14A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-11.</p> <p>ACCESS NOTE: COCKPIT GLARE SHIELD REMOVAL AND/OR WINDOW REMOVAL IS REQUIRED. SEE DTR FORM 53-10-I14A FOR DETAILS.</p> <p>NOTE: INSPECT BOTH LEFT AND RIGHT WINDOW NUMBER 1 FRAMES AT POST AB.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2112 REQUIRES REMOVAL OF WINDSHIELD POST PADDING.</p>	53-788-12

EFFECTIVITY		SPCL DETAIL	CABIN WINDOW NO. 1 POST
			53-788-12 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-788-14
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE AFT PRESSURE DOME BS 1582		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 165 166 251 252 253 254			ACCESS PANELS 811			

MECH	INSP	MPD ITEM NUMBER 53-788-14
		<p>SPECIAL DETAILED - INTERNAL: AFT PRESSURE BULKHEAD - CIRCUMFERENTIAL ATTACHMENT OF WEB TO Y-RING. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>AIRPLANE NOTE: APPLICABLE TO ALL. FOR AIRPLANE LINE NUMBERS 1-423 THIS INSPECTION IS SATISFIED BY THE INSPECTIONS IN SERVICE BULLETIN 767-53A0087.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-80-101A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-80-05.</p>

EFFECTIVITY	SPCL DETAIL	AFT PRESSURE DOME BS 1582
		53-788-14 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-788-18
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE			S 4C	99XXX	014 AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		BULK CARGO DOOR CUTOUT				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
161		1611 1612 1631 811				

MECH	INSP	MPD ITEM NUMBER
		53-788-18
<p>SPECIAL DETAILED - INTERNAL: BULK CARGO DOOR CUTOUT- FRAME SPLICE- AFT EDGE FRAME STA 1461 INNER CHORD AT STRINGER S-26. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-I10B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-60-01.</p> <p>NOTE: DETAIL COVERS THE FOLLOWING AREA: BULK CARGO DOOR CUTOUT EDGE FRAME AT STA 1461 AT FRAME SPLICE, S-24L TO S-26L. INSPECT BOTH THE CHORD AND STRAP AT END FASTENERS OF SPLICE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1611 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT LEFT SIDE PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1612 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1631 REQUIRES REMOVAL OF THE BULK CARGO COMPARTMENT FLOOR PANELS.</p>		

EFFECTIVITY	SPCL DETAIL	BULK CARGO DOOR CUTOUT
		53-788-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-788-20
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE BULK CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 161 163		ACCESS PANELS 811				

MECH	INSP	MPD ITEM NUMBER 53-788-20
		<p>SPECIAL DETAILED - INTERNAL: BULK CARGO DOOR CUTOUT- EDGE FRAMES INNER CHORD/STRAP. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-I10A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-60-02.</p> <p>NOTE: DETAIL COVERS THE FOLLOWING AREA: BULK CARGO DOOR CUTOUT EDGE FRAME AT STA 1417, INNER CHORD/STRAP, S-26L TO S-33L. EDGE FRAME AT STA 1461, INNER CHORD, S-21L TO S-24L AND S-27L TO S-30L. INSPECT BOTH INNER CHORD AND STRAP.</p>

EFFECTIVITY	SPCL DETAIL	BULK CARGO DOOR CUTOUT
		53-788-20 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-790-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 3C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE AFT CARGO DOOR CUTOUT LWR MAIN SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 154			ACCESS PANELS 822			

MECH	INSP	MPD ITEM NUMBER 53-790-12
		<p>SPECIAL DETAILED - INTERNAL: AFT MAIN CARGO DOOR CUTOUT- LOWER MAIN SILLS- STA 1263-1270 . (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>AIRPLANE NOTE: PASSENGER AIRPLANES WITH STANDARD AFT CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-106E/ 53-30-114E FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-60-07.</p>

6 0 0 3	EFFECTIVITY	SPCL DETAIL	AFT CARGO DOOR CUTOUT LWR MAIN SILL
			53-790-12 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-790-14
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 2/5	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE AFT CARGO DOOR CUTOUT UPR MAIN SILL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 154			ACCESS PANELS 1532 1541 822			

MECH	INSP	MPD ITEM NUMBER 53-790-14
		<p>SPECIAL DETAILED - INTERNAL: AFT MAIN CARGO DOOR CUTOUT- UPPER MAIN SILLS- STA 1346-1395. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>AIRPLANE NOTE: PASSENGER AIRPLANES WITH STANDARD AFT CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-106D/ 53-30-114D FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-60-06.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p>

EFFECTIVITY	SPCL DETAIL	AFT CARGO DOOR CUTOUT UPR MAIN SILL
		53-790-14 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-790-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK DETAILED IN		TITLE AFT CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 154		ACCESS PANELS 1541 822				

MECH	INSP	MPD ITEM NUMBER 53-790-16
		<p>DETAILED - INTERNAL: AFT MAIN CARGO DOOR CUTOUT- EDGE FRAMES-FRAMES AT STA 1270 AND 1346 BETWEEN UPPER AND LOWER SILLS. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>AIRPLANE NOTE: PASSENGER AIRPLANES WITH STANDARD AFT CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-106C/ 53-30-114C FOR ALTERNATIVE REPEAT INTERVALS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p>

EFFECTIVITY		DETAILED IN	AFT CARGO DOOR CUTOUT
			53-790-16 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
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TASK CARD

BOEING CARD NO. 53-790-18
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 7/10	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE AFT CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 154		ACCESS PANELS 1532 1541 1551				

MECH	INSP	MPD ITEM NUMBER 53-790-18
		<p>SPECIAL DETAILED - INTERNAL: AFT MAIN CARGO DOOR CUTOUT- EDGE FRAMES- FRAMES AT STA 1270 AND STA 1346 BETWEEN UPPER AND LOWER SILLS. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-60-106B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-60-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1532 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT CEILING PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-02.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1541 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT RIGHT SIDEWALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1551 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FLOOR PANELS PER MAINTENANCE MANUAL REFERENCE 25-53-03.</p> <p>NOTE: INSPECT BOTH THE INNER CHORD AND STRAP.</p>

EFFECTIVITY	SPCL DETAIL	AFT CARGO DOOR CUTOUT
		53-790-18 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-02
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 3/10	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE BODY STA 1065 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151 152		ACCESS PANELS 1001 1511 822				

MECH	INSP	MPD ITEM NUMBER 53-792-02
		<p>SPECIAL DETAILED - INTERNAL: AFT WHEEL WELL BULKHEAD STA 1065- HIDDEN DETAILS AT CHORD SPLICES STRINGER S-27 TO S-28 AND S-35 TO S-36. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I14B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-50-03.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FORWARD WALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p>

EFFECTIVITY	SPCL DETAIL	BODY STA 1065 BULKHEAD
		53-792-02 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-04
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 25000 CYC	PHASE S 4C 9/10 329XX	MPD REV 014	TASK CARD REVISION APR 22/07
TASK SPCL DETAIL		TITLE BODY STA 1065 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 151 152			ACCESS PANELS 1001 1511 822			

MECH	INSP	MPD ITEM NUMBER 53-792-04
		<p>SPECIAL DETAILED - INTERNAL: AFT WHEEL WELL BULKHEAD STA 1065- HIDDEN DETAILS AT CHORD BETWEEN STRINGERS S-29 AND S-34. (TYPE 1 STRUCTURE)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I14C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-50-02 AND IN PART 6, SUBJECT 53-50-02. BOTH NDI PROCEDURES ARE REQUIRED TO ADEQUATELY EXAMINE ALL INSPECTION AREAS AND MEET THE DTR.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1511 REQUIRES REMOVAL OF THE AFT CARGO COMPARTMENT FORWARD WALL PANELS PER MAINTENANCE MANUAL REFERENCE 25-52-01.</p>

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EFFECTIVITY	SPCL DETAIL	BODY STA 1065 BULKHEAD
		53-792-04 PAGE 1 OF 1 APR 22/07

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-06
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE			99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY	
SPCL DETAIL		BODY STA 1065 BULKHEAD			AIRPLANE	ENGINE
					ALL	ALL
ZONES			ACCESS PANELS			
143 144			732 742			

MECH	INSP	MPD ITEM NUMBER
		53-792-06
<p>SPECIAL DETAILED - INTERNAL: AFT WHEEL WELL BULKHEAD STA 1065- VERTICAL BEAM BL 26 INTERCOSTAL UPPER CHORD SPLICE. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-50-I14A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-06.</p>		

EFFECTIVITY	SPCL DETAIL	BODY STA 1065 BULKHEAD
		53-792-06 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-08
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MLG W/W	RELATED TASK	INTERVAL S 4C 3/5	PHASE 99XXX	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE TRANSVERSE BEAM AT MAIN WHEEL WELL		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE PASS ALL	
ZONES 143 144			ACCESS PANELS 732 742			

MECH	INSP	MPD ITEM NUMBER 53-792-08
		<p>SPECIAL DETAILED - INTERNAL: MLG WHEEL WELL TRANSVERSE FLOOR BEAM BS 1021 LOWER CHORD- FROM LBL 33 TO LBL 60 AND RBL 33 TO RBL 60. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I19 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-05.</p>

6 0 1 0	EFFECTIVITY	SPCL DETAIL	TRANSVERSE BEAM AT MAIN WHEEL WELL
			53-792-08 PAGE 1 OF 1 AUG 22/05

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 3/5	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE STA 1043 SPLICE AT LNG GEAR FITTING		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 141 142			ACCESS PANELS			

MECH	INSP	MPD ITEM NUMBER 53-792-10
		<p>SPECIAL DETAILED - INTERNAL: STA 1043 FRAME SPLICE WITH THE LANDING GEAR FITTING - STA 1043, BL 492, WL 193 TO WL 199. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I11 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-10.</p>

EFFECTIVITY	SPCL DETAIL	STA 1043 SPLICE AT LNG GEAR FITTING
		53-792-10 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-14
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 25000 CYC	PHASE S 4C	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE BODY STA 955.1 BULKHEAD INNER FLANGE		STRUCTURAL ILLUSTRATION REFERENCE 1-25	APPLICABILITY AIRPLANE ALL ENGINE ALL	
ZONES 131 132		ACCESS PANELS 1311 1312				

MECH	INSP	MPD ITEM NUMBER 53-792-14
		<p>SPECIAL DETAILED - INTERNAL: FUSELAGE REAR SPAR BULKHEAD- BULKHEAD FITTING INNER CHORD S-20 TO S-21 AT SEAT TRACK BRACKET. (TYPE 1 STRUCTURE.)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I13B FOR ALTERNATIVE REPEAT INSPECTION.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-50-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1311 REQUIRES REMOVAL OF THE CABIN FLOOR COVERING STA 785 TO 955 PER MAINTENANCE MANUAL REFERENCE 25-27-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1312 REQUIRES REMOVAL OF THE CABIN FLOOR PANELS STA 785 TO 955 PER MAINTENANCE MANUAL REFERENCE 53-01-01.</p>

6 0 1 2	EFFECTIVITY	SPCL DETAIL	BODY STA 955.1 BULKHEAD INNER FLANGE
			53-792-14 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-792-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION APR 22/08
TASK SPCL DETAIL		TITLE BODY STA 808-933 FRAME AT STUB BEAM		STRUCTURAL ILLUSTRATION REFERENCE 1-21 1-23 1-24		APPLICABILITY AIRPLANE ENGINE NOTE ALL
ZONES 131 132			ACCESS PANELS 1311 1312			

MECH	INSP	MPD ITEM NUMBER 53-792-16
		<p>SPECIAL DETAILED - INTERNAL: SECTION 45 STUB FRAMES- STA 808 - 933 FRAME INNER CHORD AT STUB BEAM. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200, 300PASS, AND 300F AIRPLANES.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-50-I18/ 53-50-I24/53-50-I28 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-50-11. OPTIONAL PROCEDURES TO INSPECT THE OPEN HOLE ARE FOUND IN PART 6, SUBJECT 53-50-12.</p>

EFFECTIVITY	SPCL DETAIL	BODY STA 808-933 FRAME AT STUB BEAM
		53-792-16 PAGE 1 OF 1 APR 22/08

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-792-18
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C	99TBD	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		LARGE FWD CARGO DOOR LATCH				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
124		821				

MECH	INSP	MPD ITEM NUMBER
		53-792-18
<p>SPECIAL DETAILED - INTERNAL: LARGE FORWARD CARGO DOOR CUTOUT- LOWER LATCH BACKUP STRUCTURE. LATCH SUPPORT FITTINGS AT STA 487.7 AND STA 608.3. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>AIRPLANE NOTE: AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-120A FOR ALTERNATIVE REPEAT INSPECTIONS.</p>		

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR LATCH
		53-792-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-794-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 2/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122		ACCESS PANELS 821				

MECH	INSP	MPD ITEM NUMBER 53-794-10
		<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- UPPER SILL: STA 482-493. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO -200 AND -300 PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19G FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-30-10.</p>

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
		53-794-10 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-794-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 2/5 610XX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122		ACCESS PANELS 821				

MECH	INSP	MPD ITEM NUMBER 53-794-12
		<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- UPPER SILL: STA 480. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19F FOR ALTERNATIVE REPEAT INSPECTIONS.</p>

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
		53-794-12 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-794-14
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122		ACCESS PANELS 821				

MECH	INSP	MPD ITEM NUMBER 53-794-14
		<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- UPPER SILL: STA 478-482. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19E FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-30-10.</p>

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
		53-794-14 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-794-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 2/5 610XX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122 124			ACCESS PANELS 821			

MECH	INSP	MPD ITEM NUMBER 53-794-16
		<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- LOWER SILL: STA 610-632, UPPER SILL: STA 456-478 AND STA 482-493. (TYPE 2 STRUCTURE)</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19D FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-30-10.</p>

6 0 1 8	EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
			53-794-16 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-794-18
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 2C	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE	
ZONES 122		ACCESS PANELS 821				
MECH	INSP	MPD ITEM NUMBER				

53-794-18

SPECIAL DETAILED - INTERNAL:
FORWARD LARGE CARGO DOOR CUTOUT- EDGE FRAME AT STA 618
BETWEEN STRINGER S-28R AND S-29R.
(TYPE 2 STRUCTURE)

AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH
FORWARD LARGE CARGO DOOR.

NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19C
FOR ALTERNATIVE REPEAT INSPECTIONS.

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
		53-794-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-794-20
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C 1/2 610XX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122			ACCESS PANELS 821			

MECH	INSP	MPD ITEM NUMBER 53-794-20
		<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- EDGE FRAME INNER CHORD AT STA 478, S-23R TO S-28R AND AT STA 618, S-29R TO S-30R. (TYPE 2 STRUCTURE).</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19B FOR ALTERNATIVE REPEAT INSPECTIONS.</p>

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT
		53-794-20 PAGE 1 OF 1 AUG 22/05

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STATION
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DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-796-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL 50000 CYC	PHASE S 4C	610XX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE LARGE FWD CARGO DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 122		ACCESS PANELS 821					

MECH	INSP	MPD ITEM NUMBER 53-796-00
<p>SPECIAL DETAILED - INTERNAL: FORWARD LARGE CARGO DOOR CUTOUT- EDGE FRAME AT STA 478 STRINGER S-23R TO S-28R AND EDGE FRAME AT STA 618 STRINGER S-27R TO S-28R, S-29R TO S-30R. (TYPE 2 STRUCTURE).</p> <p>AIRPLANE NOTE: APPLICABLE TO 200 AND 300PASS AIRPLANES WITH FORWARD LARGE CARGO DOOR.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-30-I19A FOR ALTERNATIVE REPEAT INSPECTIONS.</p>		

EFFECTIVITY	SPCL DETAIL	LARGE FWD CARGO DOOR CUTOUT	53-796-00	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-796-12
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 018	TASK CARD REVISION APR 22/07
TASK GEN VISUAL		TITLE MAIN EQUIPMENT CENTER DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 119		ACCESS PANELS 119AL				

MECH	INSP	MPD ITEM NUMBER 53-796-12
		<p>GENERAL VISUAL - INTERNAL: MAIN EQUIPMENT CENTER ACCESS DOOR CUTOUT, STA 303 - STA 325, LBL 13.1 - RBL 13.1. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I12 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: INSPECT THE DOUBLER FOR CRACKS AT ALL FOUR CORNERS OF THE CUTOUT. CHECK FOR CRACKS EMANATING FROM THE EDGE OF THE CUTOUT AND THE FASTENER HOLES.</p>

EFFECTIVITY	GEN VISUAL	MAIN EQUIPMENT CENTER DOOR CUTOUT
		53-796-12 PAGE 1 OF 1 APR 22/07

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STATION
TAIL NO.
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BOEING CARD NO. 53-796-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 2C	PHASE 99XXX	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE FWD ENTRY/SERVICE DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 119 221 222 223 224			ACCESS PANELS 2003 831 841			

MECH	INSP	MPD ITEM NUMBER 53-796-16
<p>SPECIAL DETAILED - INTERNAL: FORWARD PASSENGER ENTRY/SERVICE DOOR CUTOUT- DOOR STA 76.3, S-7 TO S-12 AND S-19 TO S-23. DOOR STA 123.7, S-8 TO S-12 AND S-19 TO S-23.</p> <p>(TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I10A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-07.</p> <p>ACCESS NOTE: SPECIAL ACCESS 2003 REQUIRES REMOVAL OF THE CREW CABIN/PASSENGER COMPARTMENT INSULATION PER MAINTENANCE MANUAL REFERENCE 25-21-05.</p>		

EFFECTIVITY	SPCL DETAIL	FWD ENTRY/SERVICE DOOR CUTOUT
		53-796-16 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-796-18
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 4C 1/2	PHASE 99XXX	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE BODY STA 287 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 119		ACCESS PANELS 1001 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-796-18
		<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL STA 287 BULKHEAD DETAILS- OUTER CHORD- STRINGER SPLICES BETWEEN S-24L AND S-24R. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORMS 53-10-I21C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-10.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>

EFFECTIVITY	SPCL DETAIL	BODY STA 287 BULKHEAD
		53-796-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-796-20
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 4C 1/5	PHASE 99XXX	MPD REV 018	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE BODY STA 287 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 119		ACCESS PANELS 1001 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-796-20
		<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING WHEEL WELL STA 287 BULKHEAD DETAILS- OUTER CHORD TYPICAL LOCATIONS BETWEEN S-24L AND S-24R. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-121B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-10.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>

EFFECTIVITY	SPCL DETAIL	BODY STA 287 BULKHEAD
		53-796-20 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
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BOEING CARD NO. 53-798-00
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA MAIN EE CTR	RELATED TASK	INTERVAL S 4C 1/5	PHASE 99XXX	MPD REV 018	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE BODY STA 287 BULKHEAD		STRUCTURAL ILLUSTRATION REFERENCE 1-11	APPLICABILITY AIRPLANE ENGINE NOTE ALL	
ZONES 119		ACCESS PANELS 1001 1003 119AL				

MECH	INSP	MPD ITEM NUMBER 53-798-00
<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL STA 287 BULKHEAD DETAILS- WL 159 BEAM AND BULKHEAD WEB FROM LBL 27 TO LBL 23, LBL 10 TO RBL 10, RBL 23 TO RBL 27. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>AIRPLANE NOTE: APPLICABLE TO AIRPLANE LINE NUMBERS 1 THROUGH 757.</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I21A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-04.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>		

6 0 2 6	EFFECTIVITY	SPCL DETAIL	BODY STA 287 BULKHEAD
			53-798-00 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-798-02
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	MAIN EE CTR		S 4C	99XXX	018	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		BODY STA 287 BULKHEAD				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
119						

MECH	INSP	MPD ITEM NUMBER				
		53-798-02				
<p>SPECIAL DETAILED - EXTERNAL: NOSE LANDING WHEEL WELL STA 287 BULKHEAD DETAILS- OUTER CHORD HIDDEN LOCATIONS AT S-36L,S-36R AND S-27R. (TYPE 1 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I21D FOR ALTERNATIVE REPEAT INSPECTIONS.</p>						

EFFECTIVITY	SPCL DETAIL	BODY STA 287 BULKHEAD	53-798-02	PAGE 1 OF 1	AUG 22/05
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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-798-04
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 3/5	99XXX	014	DEC 22/06
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		NOSE WHEEL WELL SIDE PANEL BEAM				AIRPLANE ENGINE
						ALL ALL
ZONES			ACCESS PANELS			
117 118			1001 1003 113AL 119AL			

MECH	INSP	MPD ITEM NUMBER
		53-798-04
<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL - TRUNNION SUPPORT FITTING AT STRAP ATTACHMENT, STA 276, WL 165 TO WL 169, WHERE THE STRAP ATTACHES TO THE FREE FLANGE OF THE TRUNNION SUPPORT FITTING. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-120C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-15.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>		

EFFECTIVITY	SPCL DETAIL	NOSE WHEEL WELL SIDE PANEL BEAM
		53-798-04 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-798-06
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 3/5	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		NOSE WHEEL WELL SIDE PANEL BEAM				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
117 118		1001 1003 113AL 119AL				

MECH	INSP	MPD ITEM NUMBER	
		53-798-06	
<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL - TRUNNION SUPPORT FITTING AT TOP PANEL BEAM ATTACHMENT. STA 276, BL 30, WL 181 TO WL 187, WHERE THE TOP PANEL BEAM ATTACHES TO THE TRUNNION SUPPORT FITTING. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-120B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>			

EFFECTIVITY	SPCL DETAIL	NOSE WHEEL WELL SIDE PANEL BEAM
		53-798-06 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-798-08
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION DEC 22/06
TASK DETAILED IN		TITLE NOSE WHEEL WELL SIDE PANEL VER BEAMS		STRUCTURAL ILLUSTRATION REFERENCE 1-11	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 117 118			ACCESS PANELS 1001 1003 113AL 119AL			

MECH	INSP		MPD ITEM NUMBER
		<p>DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL BL 25 SIDE PANEL DETAILS- VERTICAL BEAMS AND PANEL WEB FROM STA 180.5 TO STA 276. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I20A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1003 NOTES THAT EQUIPMENT REMOVAL IS NOT REQUIRED.</p>	53-798-08

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EFFECTIVITY		DETAILED IN	NOSE WHEEL WELL SIDE PANEL VER BEAMS
			53-798-08 PAGE 1 OF 1 DEC 22/06

STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-798-10
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 1/2	PHASE 99XXX	MPD REV 018	TASK CARD REVISION DEC 22/06
TASK SPCL DETAIL		TITLE FORWARD EQUIPMENT BAY DOOR CUTOUT		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 113 114		ACCESS PANELS 113AL				

MECH	INSP		MPD ITEM NUMBER
		<p>SPECIAL DETAILED - INTERNAL: FORWARD EQUIPMENT BAY ACCESS DOOR CUTOUT, STA 144.5 - STA 163.2, LBL 9 - RBL 9. (TYPE 2 STRUCTURE)</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I13 FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-16.</p> <p>NOTE: INSPECT THE DOUBLER FOR CRACKS AT ALL FOUR CORNERS OF THE CUTOUT. CHECK FOR CRACKS EMANATING FROM THE EDGE OF THE CUTOUT AND THE FASTENER HOLES.</p>	53-798-10

EFFECTIVITY		SPCL DETAIL	FORWARD EQUIPMENT BAY DOOR CUTOUT
			53-798-10 PAGE 1 OF 1 DEC 22/06

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-798-12
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		NOSE WHEEL WELL CANTED BULKHEAD				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
113 114		1001				

MECH	INSP	MPD ITEM NUMBER
		53-798-12
<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL CANTED BULKHEAD DETAILS- OUTER CHORD TYPICAL LOCATIONS BETWEEN S-32L AND S-32R. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I18A FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 6, SUBJECT 53-10-05.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p>		

EFFECTIVITY	SPCL DETAIL	NOSE WHEEL WELL CANTED BULKHEAD
		53-798-12 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-798-14
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 2C	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		NOSE WHEEL WELL CANTED BULKHEAD				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
113 114		1001				

MECH	INSP	MPD ITEM NUMBER
		53-798-14
<p>SPECIAL DETAILED - INTERNAL: NOSE LANDING GEAR WHEEL WELL CANTED BULKHEAD DETAILS- OUTER CHORD HIDDEN LOCATIONS L/RBL 10.6 AND L/RBL 25.0. (TYPE 1 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I18B FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: THE NDI METHOD(S) NECESSARY TO ACCOMPLISH THE INTENT OF THIS INSPECTION IS CONTAINED IN THE 767 NON-DESTRUCTIVE TEST MANUAL D634T301. THE INSPECTION PROCEDURES ARE CONTAINED IN PART 4, SUBJECT 53-10-01.</p> <p>ACCESS NOTE: SPECIAL ACCESS 1001 REQUIRES REMOVAL OR DISPLACEMENT OF INSULATION IN THE LOWER HALF OF THE FUSELAGE.</p>		

EFFECTIVITY	SPCL DETAIL	NOSE WHEEL WELL CANTED BULKHEAD
		53-798-14 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE



BOEING CARD NO. 53-798-16
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C 7/10	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK SPCL DETAIL		TITLE SKIN LONGITUDINAL LAP SPLICES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 100		ACCESS PANELS NOTE				

MECH	INSP	MPD ITEM NUMBER 53-798-16
		<p>SPECIAL DETAILED - INTERNAL: SKIN LONGITUDINAL LAP SPLICES- LOWER ROW- BILGE AND LOWER LOBE, SECTIONS 41, 43 AND 46. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I07D/ 53-30-I04D/53-60-I05D FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SIDEWALL PANEL, FLOOR PANEL AND INSULATION REMOVAL/DISPLACEMENT IS REQUIRED.</p> <p>NOTE: SKIN SPLICES AT: S-21 L/R FROM STA 132.5 TO 246 S-26 L/R FROM STA 246 TO 786 AND 1065 TO 1582 S-30 L/R FROM STA 132.5 TO 246 S-36 L/R FROM STA 434 TO 786 AND 1065 TO 1582 S-37 L/R FROM STA 287 TO 434</p>

EFFECTIVITY	SPCL DETAIL	SKIN LONGITUDINAL LAP SPLICES
		53-798-16 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
767
TASK CARD

BOEING CARD NO. 53-798-18
AIRLINE CARD NO.

SKILL AIRPL	WORK AREA FUSELAGE	RELATED TASK	INTERVAL S 4C	PHASE 99XXX	MPD REV 014	TASK CARD REVISION AUG 22/05
TASK GEN VISUAL		TITLE FRAMES, LONGTDNAL SKIN LAP SPLICES		STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE ENGINE ALL ALL	
ZONES 100			ACCESS PANELS NOTE			

MECH	INSP	MPD ITEM NUMBER 53-798-18
		<p>GENERAL VISUAL - INTERNAL: FRAMES, S-20 AND BELOW. BS 246 TO BS 786 AND BS 1065 TO BS 1582 (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I08C/53-30-I24C/53-60-I08C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>ACCESS NOTE: SIDEWALL PANEL, FLOOR PANEL AND INSULATION REMOVAL/DISPLACEMENT IS REQUIRED.</p>

EFFECTIVITY	GEN VISUAL	FRAMES, LONGTDNAL SKIN LAP SPLICES
		53-798-18 PAGE 1 OF 1 AUG 22/05

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STATION
TAIL NO.
DATE

SAS  **BOEING**
 767
TASK CARD

BOEING CARD NO. 53-798-20
AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	FUSELAGE		S 4C 1/2	99XXX	014	AUG 22/05
TASK		TITLE		STRUCTURAL ILLUSTRATION REFERENCE		APPLICABILITY
SPCL DETAIL		SKIN LONGITUDINAL LAP SPLICES				AIRPLANE ENGINE
ZONES		ACCESS PANELS				
100						

MECH	INSP	MPD ITEM NUMBER				
		53-798-20				
<p>SPECIAL DETAILED - EXTERNAL: SKIN LONGITUDINAL LAP SPLICES- UPPER ROW- BILGE AND LOWER LOBE, SECTIONS 41, 43 AND 46. (TYPE 2 STRUCTURE).</p> <p>INTERVAL NOTE: THRESHOLD: 50000 CYC (200, 300PASS) 40000 CYC (300F, 400E)</p> <p>NOTE: SEE DOC D622T001-DTR, DTR CHECK FORM 53-10-I07C/53-30-I04C/53-60-I05C FOR ALTERNATIVE REPEAT INSPECTIONS.</p> <p>NOTE: SKIN SPLICES AT: S-21 L/R FROM STA 132.5 TO 246 S-26 L/R FROM STA 246 TO 786 AND 1065 TO 158 S-30 L/R FROM STA 132.5 TO 246 S-36 L/R FROM STA 434 TO 786 AND 1065 TO 158 S-37 L/R FROM STA 287 TO 434</p>						

EFFECTIVITY	SPCL DETAIL	SKIN LONGITUDINAL LAP SPLICES
		53-798-20 PAGE 1 OF 1 AUG 22/05

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