STATION		BOEING CARD NO.
TAIL NO.	SAS BOEING 767 TASK CARD	71-600-00
DATE		AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	ENGINE 1		S 1C	11212	008	APR 22/08

TASK	TITLE	STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE	APPLICABILITY ENGINE
DETAILED IN	NO. 1 POWERPLANT	4-1 4-23 4-3 4-4	ALL	NOTE

ZONES	ACCESS PANELS
410	415AL 416AR 417AL 418AR

MECH	INSP	MPD ITEM NUMBER

DETAILED - INTERNAL: 71-600-00
FORWARD ENGINE MOUNT BULKHEAD FITTING, VISIBLE PORTION OF DIAGONAL BRACE ATTACH FITTING AT AFT END OF FIREWALL, FORWARD AND AFT ENGINE MOUNTS, LINKS, AND ATTACHMENTS.

ENGINE NOTE: APPLICABLE TO AIRPLANES EQUIPPED WITH THE 80A, 80C, 7R4, AND 4000 ENGINES.

NOTE: THIS TASK SATISFIES THE MRB TASK 71-400-00.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
   IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

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MECH	INSP	

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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STATION		BOEING CARD NO.
TAIL NO.	SAS BOEING 767 TASK CARD	71-602-00
DATE		AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	ENGINE 1		ENG CNG	99XXX	008	APR 22/08

TASK	TITLE	STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE	APPLICABILITY ENGINE
GEN VISUAL	NO. 1 POWERPLANT	4-1 4-23 4-3 4-4 4-6	ALL	NOTE

ZONES	ACCESS PANELS
410	NOTE

MECH	INSP	MPD ITEM NUMBER

GENERAL VISUAL - INTERNAL: 71-602-00
FORWARD ENGINE MOUNT BULKHEAD FITTING, VISIBLE PORTION OF DIAGONAL BRACE ATTACH FITTING AT AFT END OF FIREWALL, FORWARD AND AFT ENGINE MOUNTS, LINKS, ATTACHMENTS, AND STRUT LOWER SURFACE (FIREWALL).

ACCESS NOTE: AT ENGINE REMOVAL.

ENGINE NOTE: APPLICABLE TO AIRPLANES EQUIPPED WITH THE 80A, 80C, 7R4, AND 4000 ENGINES.

NOTE: THIS TASK SATISFIES THE MRB TASK 71-402-00.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
   IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL

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MECH	INSP	

FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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STATION		BOEING CARD NO.
TAIL NO.	SAS BOEING 767 TASK CARD	71-604-00
DATE		AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	ENGINE 2		S 1C	11212	008	APR 22/08

TASK	TITLE	STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE	APPLICABILITY ENGINE
DETAILED IN	NO. 2 POWERPLANT	4-1 4-23 4-3 4-4	ALL	NOTE

ZONES	ACCESS PANELS
420	425AL 426AR 427AL 428AR

MECH	INSP	MPD ITEM NUMBER
		71-604-00

DETAILED - INTERNAL:
FORWARD ENGINE MOUNT BULKHEAD FITTING, VISIBLE PORTION OF DIAGONAL BRACE ATTACH FITTING AT AFT END OF FIREWALL, FORWARD AND AFT ENGINE MOUNTS, LINKS, AND ATTACHMENTS.

ENGINE NOTE: APPLICABLE TO AIRPLANES EQUIPPED WITH THE 80A, 80C, 7R4, AND 4000 ENGINES.

NOTE: THIS TASK SATISFIES THE MRB TASK 71-404-00.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
   IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.
5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.

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MECH	INSP	

6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR
- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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STATION		BOEING CARD NO.
TAIL NO.	SAS BOEING 767 TASK CARD	71-606-00
DATE		AIRLINE CARD NO.

SKILL	WORK AREA	RELATED TASK	INTERVAL	PHASE	MPD REV	TASK CARD REVISION
AIRPL	ENGINE 2		ENG CNG	99XXX	008	APR 22/08

TASK	TITLE	STRUCTURAL ILLUSTRATION REFERENCE	APPLICABILITY AIRPLANE	APPLICABILITY ENGINE
GEN VISUAL	NO. 2 POWERPLANT	4-1 4-23 4-3 4-4 4-6	ALL	NOTE

ZONES	ACCESS PANELS
420	NOTE

MECH	INSP	MPD ITEM NUMBER
		71-606-00

GENERAL VISUAL - INTERNAL:
FORWARD ENGINE MOUNT BULKHEAD FITTING, VISIBLE PORTION OF DIAGONAL BRACE ATTACH FITTING AT AFT END OF FIREWALL, FORWARD AND AFT ENGINE MOUNTS, LINKS, ATTACHMENTS, AND STRUT LOWER SURFACE (FIREWALL).

ACCESS NOTE: AT ENGINE REMOVAL.

ENGINE NOTE: APPLICABLE TO AIRPLANES EQUIPPED WITH THE 80A, 80C, 7R4, AND 4000 ENGINES.

NOTE: THIS TASK SATISFIES THE MRB TASK 71-406-00.

THE FOLLOWING DESCRIBES THE BASIC TASK TO BE ACCOMPLISHED IN EACH DEFINED AIRPLANE AREA AS PART OF THE STRUCTURAL AND ZONAL REQUIREMENTS:

1. REMOVE ALL SYSTEMS, EQUIPMENT, AND INTERIOR FURNISHINGS, ETC (E.G. TOILETS, GALLEYS, LINING, INSULATION) AS REQUIRED TO ACCOMPLISH STEP 3).
   IT IS NOT NECESSARY TO REMOVE BUSHINGS UNLESS SPECIFIED IN THE TASK DESCRIPTION OR, THERE IS AN INDICATION OF CORROSION OR, THE BUSHING HAS MIGRATED.
2. PRIOR TO INSPECTION, CLEAN THE AREA AS REQUIRED TO ACCOMPLISH STEP 3) OF THE BASIC TASK.
3. VISUALLY INSPECT ALL STRUCTURE LISTED FROM A DISTANCE CONSIDERED NECESSARY TO DETECT EARLY STAGES OF CORROSION OR INDICATIONS OF OTHER DISCREPANCIES SUCH AS CRACKING (E.G., GENERAL VISUAL INSPECTION). AREAS REQUIRING A DETAILED INSPECTION ARE NOTED IN THE APPROPRIATE TASK DESCRIPTION. ADDITIONAL NON-DESTRUCTIVE INSPECTIONS OR VISUAL INSPECTIONS FOLLOWING PARTIAL DISASSEMBLY ARE REQUIRED IF THERE ARE INDICATIONS OF HIDDEN CORROSION SUCH AS BULGING SKINS, CORROSION RUNNING INTO SPLICES, OR UNDER FITTINGS, ETC.
4. REMOVE ALL CORROSION, EVALUATE DAMAGE AND REPAIR OR REPLACE ALL DISCREPANT STRUCTURE, INCLUDING APPLICATION OF PROTECTIVE FINISHES, AS APPROPRIATE. SURFACE OXIDATION OF FERROUS METAL

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MECH	INSP	

FASTENERS MAY BE HANDLED BY NORMAL OR EXISTING MAINTENANCE PRACTICES.

5. CLEAR ANY BLOCKED HOLES OR GAPS THAT MAY HINDER DRAINAGE.
6. APPLY SUITABLE APPROVED WATER DISPLACING/ANTI-CORROSION COMPOUNDS AS FOLLOWS: (RECOMMENDED PROCEDURES FOR APPLYING COMPOUNDS ARE GIVEN IN REFERENCE (OPTIONAL)):

A. MINIMUM REQUIREMENT FOR ALL AREAS EXCEPT AS NOTED 6 C):
- APPLY A SINGLE COAT THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE (E.G., BMS 3-23).

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G., BMS 3-35).

B. RECOMMENDED APPLICATION FOR AREAS WITH HIGH POTENTIAL FOR SEVERE CORROSION (OPTIONAL):
- APPLY A DUAL APPLICATION CONSISTING OF A SINGLE COAT THAT PENETRATES AND DISPLACES MOISTURE COVERED BY A SINGLE COAT THAT FORMS A DURABLE BARRIER (E.G., A SINGLE COAT PER BMS 3-23 WITH A TOP-COAT PER BMS 3-26 TYPE II)

OR

- APPLY A SINGLE COAT OF DUAL ACTING COMPOUND THAT PENETRATES FAYING SURFACES AND DISPLACES MOISTURE AND AFTER DRYING FORMS A DURABLE BARRIER (E.G. BMS 3-35).

C. LIST OF AREAS/ITEMS WHERE WATER DISPLACING/ANTI-CORROSION COMPOUNDS (STEP 6) SHOULD NOT BE APPLIED.
WATER DISPLACING/ANTI-CORROSION COMPOUNDS SHOULD NOT BE APPLIED IN THE FOLLOWING AREAS:
- CABLES, PULLEYS, WIRING, PLASTICS, ELASTOMERS, OXYGEN SYSTEMS
- LUBRICATED OR TEFLON SURFACES (E.G., GREASED JOINTS, SEALED BEARINGS)
- OVER COSMOLINE 1058 (OR EQUIVALENT PER MIL-C-16173 GRADE 1)
- ADJACENT TO TEARS OR HOLES IN INSULATION BLANKETS (WATER REPELLING CHARACTERISTICS ARE DIMINISHED)
- AREAS WITH ELECTRICAL ARC POTENTIAL
- INTERIOR MATERIALS, INCLUDING CARGO LINERS (CHANGE OF FLAMMABILITY PROPERTIES)
- ENGINE STRUT CAVITIES, APU/APU SHROUD, COWLING PANELS OR POD (INCOMPATIBILITY WITH BMS 5-63 SEALANT AND/OR HIGH TEMPERATURES)
- FIBERGLASS DUCTS WHERE TEMPERATURE EXCEEDS 220 DEGREES F
- SELECTED AREAS NOTED.

7. REINSTALL DRY INSULATION BLANKETS.

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