# Phase Equilibria in Hydrocarbon Systems 

# Volumetric and Phase Behavior of the Propane-n-Decane System 

H. H. REAMER and B. H. SAGE<br>Chemical Engineering Laboratory, California Institute of Technology, Pasadena, Calif.


#### Abstract

The volumetric behavior of four mixtures of propane and $n$-decane was investigated at pressures up to 10,000 p.s.i.a. in the temperature interval between $40^{\circ}$ and $460^{\circ} \mathrm{F}$. The compositions of the coexisting gas and liquid phases were established throughout the two-phase region within the above-indicated range of temperature. The results are in good agreement with the volumetric and phase behavior of propane and of $n$-decane.


TTHERE CONTINUES to be a need for measurements of the volumetric and phase behavior of binary mixtures of the aliphatic hydrocarbons, since entirely adequate methods of prediction of such properties of binary systems are not as yet available. Furthermore, such measurements add to the background of experimental data necessary for the more extensive studies associated with ternary and more complicated systems. Experimental measurements concerning the volumetric or the phase behavior of the propane- $n$-decane system are not available, and for this reason, studies of the behavior of this binary system were carried out at temperatures between $40^{\circ}$ and $460^{\circ} \mathrm{F}$. for pressures from 200 to 10,000 p.s.i.a. The volumetric behavior of propane has been studied in detail by Beattie, Kay, and Kaminsky (2), and the critical properties have also been determined by Beattie and others (3). These measurements are in good agreement with studies of the volumetric and phase behavior of propane (10) which were carried out somewhat later than the studies of Beattie, and are also in fair agreement with the earlier studies $(6,13)$ upon this hydrocarbon. The available information concerning the volumetric and phase behavior of propane is believed to be adequate for the purposes of this study. $n$-Decane has also been studied in some detail (9). The volumetric behavior and the vapor pressure of this compound were investigated as part of the study of the binary system methane and $n$-decane $(9,12)$. Since difficulty in


Figure 1. Volumetric measurements at $220^{\circ} \mathrm{F}$.
determining the specific volume of the dew-point gas is experienced with materials of low volatility such as $n$-decane, measurements of the latent heat of vaporization (5) together with the vapor pressure of $n$-decane and appropriate thermodynamic relations were employed to establish the specific volume of the dew-point gas. Young (14) was perhaps the first to investigate the volumetric and phase behavior of decane, and the more recent measurements $(9,12)$ are in good agreement with his earlier classical studies. No further reference to the behavior of propane or $n$-decane will be made except to indicate in a semi-quantitative way the agreement of the volumetric and phase behavior for the components with the mixtures investigated.

## APPARATUS AND METHODS

The equipment employed in this study has been described in detail (11). Mixtures of propane and $n$-decane were confined in a stainless steel vessel over mercury. The effective volume of the system could be varied by the introduction and withdrawal of mercury. Mechanical agitation was provided to hasten the attainment of physical equilibrium. The molal volume and associated pressure were determined for a series of states at each of eight systematically chosen temperatures between $40^{\circ}$ and $460^{\circ} \mathrm{F}$.

Figure 2. Pressure-composition diagram for the propane-n-decane system

The quantity of $n$-decane introduced into the equipment was established from volumetric measurements made upon the sample after its introduction, and the results agreed within $0.15 \%$ of the quantity of decane measured into the apparatus. The quantity of propane introduced into the variable-volume vessel was determined by weighing bomb techniques (11) with a probable uncertainty of not more than $0.05 \%$.

The pressure of the system was established by means of a piston-cylinder combination utilized in connection with a balance (11). The over-all device has been periodically calibrated against the vapor pressure of carbon dioxide at the ice point (4). Experience over nearly three decades with this equipment indicates that the pressure of the sample was established with a probable uncertainty of $0.05 \%$ or 0.1 p.s.i., whichever is the larger measure of uncertainty. Temperatures of the sample
under investigation were determined from that of a vigorously stirred oil bath surrounding the stainless steel pressure vessel. A strain-free, platinum resistance thermometer (7) measured the temperature of the oil bath. This instrument was periodically compared with the indications of a similar platinum resistance thermometer which had been calibrated by the National Bureau of Standards. Comparisons of at least three such calibrated resistance thermometers indicated that the temperature of the sample was related to that of the international platinum scale with an uncertainty of less than $0.03^{\circ} \mathrm{F}$. The total volume of the pressure vessel filled with hydrocarbon was established within $0.1 \%$ at pressures up to approximately 5000 p.s.i. and within $0.25 \%$ at the higher pressures. Variations in the calibration of the equipment at pressures above 5000 p.s.i. with respect to time introduced the uncertainty enumerated.

Table I. Molal Volumes for Mixtures of

| Pressure, P.S.I.A. | Mole Fraction Propane |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| $40^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | (8) ${ }^{\text {a }}$ | (16) | (25) | (33) | (41) | (49) | (56) | (64) | (72) |
| BP | 2.878 | 2.701 | 2.526 | 2.351 | 2.177 | 2.005 | 1.835 | 1.662 | 1.497 |
| 200 | $2.874^{\text {b }}$ | 2.697 | 2.521 | 2.347 | 2.173 | 2.001 | 1.831 | 1.659 | 1.494 |
| 400 | 2.870 | 2.693 | 2.516 | 2.341 | 2.168 | 1.997 | 1.826 | 1.656 | 1.491 |
| 600 | 2.865 | 2.689 | 2.511 | 2.336 | 2.164 | 1.993 | 1.822 | 1.653 | 1.488 |
| 800 | 2.861 | 2.685 | 2.507 | 2.332 | 2.159 | 1.989 | 1.818 | 1.649 | 1.484 |
| 1000 | 2.857 | 2.680 | 2.502 | 2.327 | 2.155 | 1.985 | 1.814 | 1.645 | 1.480 |
| 1250 | 2.853 | 2.676 | 2.497 | 2.322 | 2.150 | 1.980 | 1.809 | 1.640 | 1.476 |
| 1500 | 2.848 | 2.671 | 2.493 | 2.317 | 2.145 | 1.975 | 1.805 | 1.636 | 1.472 |
| 1750 | 2.844 | 2.667 | 2.488 | 2.312 | 2.140 | 1.971 | 1.801 | 1.632 | 1.467 |
| 2000 | 2.839 | 2.662 | 2.484 | 2.308 | 2.135 | 1.966 | 1.796 | 1.628 | 1.462 |
| 2250 | 2.834 | 2.658 | 2.480 | 2.304 | 2.131 | 1.962 | 1.792 | 1.624 | 1.457 |
| 2500 | 2.830 | 2.653 | 2.476 | 2.299 | 2.127 | 1.958 | 1.788 | 1.620 | 1.452 |
| 2750 | 2.826 | 2.649 | 2.471 | 2.295 | 2.123 | 1.954 | 1.784 | 1.616 | 1.448 |
| 3000 | 2.822 | 2.645 | 2.467 | 2.292 | 2.120 | 1.950 | 1.781 | 1.612 | 1.443 |
| 3500 | 2.814 | 2.637 | 2.459 | 2.284 | 2.113 | 1.943 | 1.773 | 1.603 | 1.434 |
| 4000 | 2.807 | 2.629 | 2.452 | 2.277 | 2.105 | 1.936 | 1.765 | 1.595 | 1.425 |
| 4500 | 2.799 | 2.621 | 2.444 | 2.271 | 2.099 | 1.928 | 1.757 | 1.586 | 1.416 |
| 5000 | 2.793 | 2.614 | 2.438 | 2.265 | 2.092 | 1.921 | 1.749 | 1.577 | 1.408 |
| 6000 | 2.779 | 2.601 | 2.426 | 2.254 | 2.079 | 1.908 | 1.733 | 1.562 | 1.392 |
| 7000 | 2.766 | 2.590 | 2.416 | 2.244 | 2.069 | 1.896 | 1.721 | 1.549 | 1.379 |
| 8000 | 2.756 | 2.580 | 2.406 | 2.233 | 2.060 | 1.886 | 1.711 | 1.539 | 1.367 |
| 9000 | 2.747 | 2.570 | 2.396 | 2.223 | 2.051 | 1.876 | 1.701 | 1.527 | 1.355 |
| 10,000 | 2.739 | 2.562 | 2.388 | 2.215 | 2.042 | 1.867 | 1.692 | 1.518 | 1.344 |
| $100^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | $(16)^{a}$ | (34) | (51) | (68) | (86) | (105) | (124) | (145) | (166) |
| BP | 2.986 | 2.805 | 2.625 | 2.447 | 2.273 | 2.103 | 1.935 | 1.771 | 1.620 |
| 200 | $2.980^{\text {b }}$ | 2.799 | 2.620 | 2.443 | 2.270 | 2.100 | 1.933 | 1.769 | 1.619 |
| 400 | 2.975 | 2.793 | 2.614 | 2.439 | 2.265 | 2.094 | 1.927 | 1.763 | 1.611 |
| 600 | 2.969 | 2.788 | 2.609 | 2.433 | 2.260 | 2.088 | 1.921 | 1.757 | 1.604 |
| 800 | 2.964 | 2.783 | 2.604 | 2.428 | 2.254 | 2.083 | 1.915 | 1.752 | 1.596 |
| 1000 | 2.959 | 2.777 | 2.599 | 2.423 | 2.249 | 2.077 | 1.910 | 1.747 | 1.589 |
| 1250 | 2.952 | 2.772 | 2.593 | 2.417 | 2.243 | 2.070 | 1.903 | 1.739 | 1.580 |
| 1500 | 2.947 | 2.766 | 2.587 | 2.412 | 2.236 | 2.064 | 1.896 | 1.731 | 1.571 |
| 1750 | 2.941 | 2.761 | 2.581 | 2.406 | 2.230 | 2.057 | 1.888 | 1.724 | 1.562 |
| 2000 | 2.935 | 2.755 | 2.575 | 2.400 | 2.224 | 2.051 | 1.882 | 1.716 | 1.553 |
| 2250 | 2.928 | 2.748 | 2.570 | 2.394 | 2.218 | 2.045 | 1.877 | 1.709 | 1.546 |
| 2500 | 2.924 | 2.743 | 2.564 | 2.388 | 2.212 | 2.039 | 1.869 | 1.702 | 1.538 |
| 2750 | 2.918 | 2.737 | 2.559 | 2.382 | 2.207 | 2.034 | 1.863 | 1.695 | 1.531 |
| 3000 | 2.912 | 2.732 | 2.554 | 2.377 | 2.201 | 2.029 | 1.858 | 1.689 | 1.525 |
| 3500 | 2.901 | 2.721 | 2.544 | 2.368 | 2.192 | 2.019 | 1.847 | 1.677 | 1.512 |
| 4000 | 2.891 | 2.711 | 2.534 | 2.359 | 2.183 | 2.009 | 1.837 | 1.667 | 1.500 |
| 4500 | 2.881 | 2.702 | 2.525 | 2.351 | 2.174 | 2.000 | 1.828 | 1.658 | 1.489 |
| 5000 | 2.874 | 2.694 | 2.516 | 2.343 | 2.165 | 1.992 | 1.820 | 1.649 | 1.478 |
| 6000 | 2.859 | 2.680 | 2.501 | 2.327 | 2.150 | 1.977 | 1.804 | 1.632 | 1.460 |
| 7000 | 2.845 | 2.667 | 2.489 | 2.312 | 2.136 | 1.962 | 1.789 | 1.617 | 1.442 |
| 8000 | 2.832 | 2.655 | 2.477 | 2.300 | 2.124 | 1.949 | 1.775 | 1.603 | 1.427 |
| 9000 | 2.819 | 2.642 | 2.465 | 2.287 | 2.112 | 1.936 | 1.762 | 1.591 | 1.412 |
| 1.0,000 | 2.806 | 2.628 | 2.451 | 2.273 | 2.098 | 1.922 | 1.748 | 1.575 | 1.399 |

Measurements upon each of the samples were made at a series of ascending temperatures at intervals of $60^{\circ}$ throughout the previously indicated temperature interval. Experience indicates that the molal volumes do not involve uncertainties greater than $0.25 \%$ at temperatures below $300^{\circ} \mathrm{F}$. and may be as large as $0.3 \%$ at the higher temperatures.

The composition of the dew-point gas withdrawn from heterogeneous mixtures under isobaric, isothermal conditions was established by partial condensation procedures (8). The condenser was maintained near the temperature of solid carbon dioxide and acetone. The propane, carried as overhead, was condensed in a weighing bomb (11) at the temperature of liquid nitrogen. After the completion of the condensation procedure, the $n$-decane was permitted to warm to room temperature and then recooled again several times to ensure a relatively complete
separation of the propane from the $n$-decane which was condensed in the above-mentioned weighing bomb. The gains in weight of the partial condenser and of the weighing bomb were employed to determine the composition of the gas phase sample. Duplicate samples withdrawn at the same equilibrium state indicate a probable error of the order of 0.002 mole fraction of $n$-decane in these procedures.

## MATERIALS

The propane and $n$-decane were obtained from the Phillips Petroleum Co. The propane was of research grade and was reported to contain not more than 0.001 mole fraction of materials other than propane. A mass spectrographic analysis indicated that the sample employed in this investigation contained as much as 0.004 mole fraction of materials other than

Propane and $n$-Decane in the Liquid Phase

| Pressure, P.S.I.A. | Mole Fraction Propane |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| $160^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | $(30)^{\text {a }}$ | (61) | (94) | (129) | (166) | (204) | (244) | (288) | (333) |
| BP | 3.107 | 2.920 | 2.738 | 2.559 | 2.384 | 2.215 | 2.053 | 1.901 | 1.793 |
| 200 | $3.100^{\text {b }}$ | 2.915 | 2.734 | 2.556 | 2.382 |  |  |  |  |
| 400 | 3.091 | 2.908 | 2.726 | 2.548 | 2.374 | 2.206 | 2.044 | 1.894 | 1.785 |
| 600 | 3.083 | 2.901 | 2.719 | 2.541 | 2.366 | 2.198 | 2.034 | 1.883 | 1.761 |
| 800 | 3.076 | 2.895 | 2.712 | 2.534 | 2.359 | 2.189 | 2.024 | 1.872 | 1.742 |
| 1000 | 3.068 | 2.888 | 2.706 | 2.527 | 2.352 | 2.181 | 2.014 | 1.862 | 1.725 |
| 1250 | 3.060 | 2.880 | 2.698 | 2.519 | 2.344 | 2.171 | 2.004 | 1.849 | 1.708 |
| 1500 | 3.052 | 2.872 | 2.691 | 2.511 | 2.335 | 2.161 | 1.993 | 1.837 | 1.692 |
| 1750 | 3.044 | 2.865 | 2.683 | 2.503 | 2.327 | 2.152 | 1.984 | 1.826 | 1.679 |
| 2000 | 3.036 | 2.857 | 2.676 | 2.496 | 2.319 | 2.143 | 1.975 | 1.816 | 1.666 |
| 2250 | 3.028 | 2.849 | 2.668 | 2.489 | 2.311 | 2.135 | 1.965 | 1.806 | 1.653 |
| 2500 | 3.021 | 2.841 | 2.661 | 2.482 | 2.302 | 2.127 | 1.957 | 1.796 | 1.641 |
| 2750 | 3.014 | 2.834 | 2.654 | 2.476 | 2.294 | 2.120 | 1.950 | 1.788 | 1.629 |
| 3000 | 3.007 | 2.827 | 2.647 | 2.469 | 2.287 | 2.112 | 1.943 | 1.780 | 1.619 |
| 3500 | 2.992 | 2.814 | 2.635 | 2.457 | 2.274 | 2.099 | 1.928 | 1.764 | 1.599 |
| 4000 | 2.980 | 2.801 | 2.622 | 2.444 | 2.263 | 2.087 | 1.916 | 1.750 | 1.583 |
| 4500 | 2.968 | 2.789 | 2.610 | 2.431 | 2.251 | 2.076 | 1.904 | 1.737 | 1.568 |
| 5000 | 2.958 | 2.778 | 2.598 | 2.419 | 2.240 | 2.066 | 1.894 | 1.725 | 1.554 |
| 6000 | 2.940 | 2.760 | 2.580 | 2.399 | 2.221 | 2.044 | 1.874 | 1.705 | 1.529 |
| 7000 | 2.923 | 2.743 | 2.563 | 2.384 | 2.205 | 2.027 | 1.856 | 1.685 | 1.509 |
| 8000 | 2.908 | 2.727 | 2.548 | 2.367 | 2.190 | 2.012 | 1.840 | 1.666 | 1.490 |
| 9000 | 2.892 | 2.710 | 2.533 | 2.352 | 2.174 | 1.996 | 1.823 | 1.648 | 1.473 |
| 10,000 | 2.874 | 2.694 | 2.517 | 2.337 | 2.158 | 1.982 | 1.807 | 1.630 | 1.453 |
| $220^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | (46 ${ }^{\text {a }}$ ) | (94) | (146) | (202) | (262) | (327) | (400) | (482) | (580) |
| BP | 3.239 | 3.049 | 2.865 | 2.684 | 2.512 | 2.353 | 2.205 | 2.082 | 2.135 |
| 200 | $3.230^{\text {b }}$ | 3.044 | 2.863 |  |  |  |  |  |  |
| 400 | 3.219 | 3.034 | 2.854 | 2.675 | 2.505 | 2.347 | 2.205 |  |  |
| 600 | 3.208 | 3.024 | 2.845 | 2.666 | 2.494 | 2.331 | 2.186 | 2.065 | 2.115 |
| 800 | 3.197 | 3.015 | 2.836 | 2.657 | 2.484 | 2.317 | 2.168 | 2.041 | 2.001 |
| 1000 | 3.188 | 3.005 | 2.827 | 2.649 | 2.474 | 2.304 | 2.151 | 2.019 | 1.941 |
| 1250 | 3.176 | 2.995 | 2.816 | 2.639 | 2.462 | 2.289 | 2.133 | 1.996 | 1.892 |
| 1500 | 3.166 | 2.984 | 2.805 | 2.627 | 2.450 | 2.276 | 2.117 | 1.975 | 1.857 |
| 1750 | 3.155 | 2.975 | 2.795 | 2.616 | 2.439 | 2.263 | 2.102 | 1.956 | 1.827 |
| 2000 | 3.145 | 2.965 | 2.785 | 2.606 | 2.427 | 2.251 | 2.088 | 1.939 | 1.802 |
| 2250 | 3.135 | 2.955 | 2.775 | 2.595 | 2.416 | 2.240 | 2.075 | 1.923 | 1.781 |
| 2500 | 3.125 | 2.945 | 2.765 | 2.585 | 2.406 | 2.228 | 2.063 | 1.909 | 1.763 |
| 2750 | 3.116 | 2.935 | 2.756 | 2.575 | 2.396 | 2.218 | 2.051 | 1.895 | 1.746 |
| 3000 | 3.108 | 2.926 | 2.746 | 2.565 | 2.385 | 2.208 | 2.040 | 1.882 | 1.731 |
| 3500 | 3.091 | 2.910 | 2.730 | 2.549 | 2.368 | 2.190 | 2.020 | 1.860 | 1.703 |
| 4000 | 3.076 | 2.895 | 2.714 | 2.534 | 2.352 | 2.174 | 2.002 | 1.839 | 1.679 |
| 4500 | 3.062 | 2.880 | 2.700 | 2.520 | 2.337 | 2.159 | 1.987 | 1.821 | 1.657 |
| 5000 | 3.048 | 2.867 | 2.685 | 2.505 | 2.323 | 2.145 | 1.973 | 1.805 | 1.638 |
| 6000 | 3.025 | 2.843 | 2.660 | 2.479 | 2.299 | 2.120 | 1.948 | 1.778 | 1.606 |
| 7000 | 3.004 | 2.821 | 2.639 | 2.457 | 2.277 | 2.098 | 1.926 | 1.755 | 1.581 |
| 8000 | 2.984 | 2.803 | 2.620 | 2.438 | 2.258 | 2.078 | 1.905 | 1.733 | 1.558 |
| 9000 | 2.965 | 2.784 | 2.602 | 2.420 | 2.239 | 2.060 | 1.884 | 1.710 | 1.534 |
| 10,000 | 2.945 | 2.763 | 2.583 | 2.401 | 2.227 | 2.041 | 1.864 | 1.687 | 1.511 |


| Pressure, P.S.I.A. | Mole Fraction Propane |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| $280^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | (68) ${ }^{\text {a }}$ | (136) | (211) | (295) | (386) | (487) | (598) | (718) | (850) |
| BP | $3.386$ | $3.195$ | 3.014 | 2.841 | 2.676 | 2.523 | 2.403 | 2.343 | $2.78$ |
| 200 | $3.375{ }^{\text {b }}$ | 3.191 | ... |  |  |  | . . |  |  |
| 400 | 3.359 | 3.177 | 3.001 | 2.832 | 2.674 |  |  |  | , $\cdot$ |
| 600 | 3.344 | 3.164 | 2.988 | 2.818 | 2.655 | 2.506 | 2.402 |  |  |
| 800 | 3.330 | 3.152 | 2.976 | 2.803 | 2.636 | 2.480 | 2.358 | 2.309 |  |
| 1000 | 3.317 | 3.140 | 2.963 | 2.790 | 2.619 | 2.458 | 2.323 | 2.248 | 2.42 |
| 1250 | 3.302 | 3.122 | 2.947 | 2.772 | 2.599 | 2.433 | 2.290 | 2.193 | 2.248 |
| 1500 | 3.287 | 3.108 | 2.931 | 2.755 | 2.581 | 2.412 | 2.263 | 2.152 | 2.143 |
| 1750 | 3.273 | 3.093 | 2.916 | 2.739 | 2.564 | 2.393 | 2.241 | 2.117 | 2.066 |
| 2000 | 3.261 | 3.080 | 2.902 | 2.724 | 2.547 | 2.376 | 2.220 | 2.088 | 2.007 |
| 2250 | 3.248 | 3.067 | 2.888 | 2.709 | 2.532 | 2.360 | 2.201 | 2.060 | 1.959 |
| 2500 | 3.237 | 3.055 | 2.875 | 2.695 | 2.517 | 2.345 | 2.183 | 2.036 | 1.921 |
| 2750 | 3.226 | 3.043 | 2.863 | 2.681 | 2.503 | 2.331 | 2.167 | 2.014 | 1.890 |
| 3000 | 3.216 | 3.032 | 2.851 | 2.669 | 2.491 | 2.318 | 2.151 | 1.995 | 1.863 |
| 3500 | 3.196 | 3.011 | 2.829 | 2.646 | 2.466 | 2.292 | 2.122 | 1.962 | 1.819 |
| 4000 | 3.178 | 2.992 | 2.810 | 2.625 | 2.445 | 2.270 | 2.096 | 1.934 | 1.784 |
| 4500 | 3.162 | 2.976 | 2.792 | 2.608 | 2.425 | 2.247 | 2.074 | 1.910 | 1.756 |
| 5000 | 3.145 | 2.960 | 2.775 | 2.592 | 2.408 | 2.230 | 2.056 | 1.889 | 1.731 |
| 6000 | 3.117 | 2.931 | 2.744 | 2.561 | 2.379 | 2.198 | 2.024 | 1.856 | 1.691 |
| 7000 | 3.090 | 2.904 | 2.719 | 2.534 | 2.352 | 2.173 | 1.999 | 1.828 | 1.658 |
| 8000 | 3.065 | 2.879 | 2.696 | 2.511 | 2.328 | 2.149 | 1.973 | 1.802 | 1.628 |
| 9000 | 3.041 | 2.856 | 2.674 | 2.488 | 2.306 | 2.126 | 1.947 | 1.774 | 1.599 |
| 10,000 | 3.018 | 2.834 | 2.652 | 2.467 | 2.283 | 2.102 | 1.923 | 1.749 | 1.572 |
| $340^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |  |
|  | (96) ${ }^{\text {a }}$ | (188) | (289) | (402) | (522) | (654) | (794) | (934) | (962) ${ }^{\text {c }}$ |
| BP | 3.558 | 3.376 | 3.201 | 3.031 | 2.867 | 2.750 | 2.706 | 2.799 | 3.72 |
| 200 | $3.546^{\text {b }}$ | 3.374 | . $\cdot$ |  |  |  |  |  |  |
| 400 | 3.524 | 3.350 | 3.186 |  |  |  |  | ... | . . |
| 600 | 3.503 | 3.328 | 3.161 | 3.002 | 2.855 |  |  |  |  |
| 800 | 3.484 | 3.308 | 3.139 | 2.976 | 2.824 | 2.711 | 2.703 | $\cdots$ |  |
| 1000 | 3.467 | 3.288 | 3.117 | 2.952 | 2.797 | 2.667 | 2.612 | 2.713 | 3.45 |
| 1250 | 3.447 | 3.367 | 3.093 | 2.925 | 2.766 | 2.623 | 2.533 | 2.529 | 2.810 |
| 1500 | 3.428 | 3.246 | 3.071 | 2.901 | 2.738 | 2.587 | 2.476 | 2.420 | 2.550 |
| 1750 | 3.409 | 3.228 | 3.051 | 2.879 | 2.713 | 2.555 | 2.428 | 2.347 | 2.391 |
| 2000 | 3.395 | 3.210 | 3.032 | 2.858 | 2.688 | 2.528 | 2.388 | 2.291 | 2.275 |
| 2250 | 3.380 | 3.193 | 3.014 | 2.837 | 2.666 | 2.503 | 2.355 | 2.242 | 2.188 |
| 2500 | 3.365 | 3.178 | 2.997 | 2.819 | 2.644 | 2.479 | 2.327 | 2.203 | 2.122 |
| 2750 | 3.352 | 3.163 | 2.981 | 2.800 | 2.626 | 2.459 | 2.302 | 2.168 | 2.069 |
| 3000 | 3.339 | 3.149 | 2.966 | 2.783 | 2.608 | 2.439 | 2.280 | 2.138 | 2.026 |
| 3500 | 3.312 | 3.124 | 2.937 | 2.753 | 2.578 | 2.406 | 2.239 | 2.087 | 1.961 |
| 4000 | 3.288 | 3.101 | 2.912 | 2.726 | 2.550 | 2.376 | 2.204 | 2.047 | 1.913 |
| 4500 | 3.266 | 3.078 | 2.890 | 2.703 | 2.524 | 2.348 | 2.176 | 2.013 | 1.871 |
| 5000 | 3.246 | 3.058 | 2.870 | 2.682 | 2.502 | 2.324 | 2.152 | 1.985 | 1.837 |
| 6000 | 3.210 | 3.019 | 2.832 | 2.647 | 2.462 | 2.283 | 2.110 | 1.941 | 1.781 |
| 7000 | 3.178 | 2.988 | 2.801 | 2.616 | 2.431 | 2.250 | 2.074 | 1.904 | 1.738 |
| 8000 | 3.148 | 2.962 | 2.774 | 2.590 | 2.401 | 2.219 | 2.043 | 1.872 | 1.701 |
| 9000 | 3.120 | 2.935 | 2.748 | 2.564 | 2.375 | 2.192 | 2.015 | 1.842 | 1.667 |
| 10,000 | 3.097 | 2.909 | 2.722 | 2.537 | 2.349 | 2.167 | 1.989 | 1.814 | 1.637 |

propane. The sample employed was subjected to partial condensation followed by several evacuations at liquid nitrogen temperature. It was introduced into the volumetric equipment by conventional weighing-bomb techniques.

The $n$-decane was reported to contain not more than 0.0062 mole fraction of impurities. After deaeration and drying over metallic sodium, the $n$-decane had a specific weight of 45.3356 pounds per cubic foot at $77^{\circ} \mathrm{F}$. and atmospheric pressure. This value compares with 45.337 pounds per cubic foot reported by API 44 (1) for an air-saturated sample at the same temperature. A refractive index of 1.4097 relative to the D-lines of sodium was obtained at $77^{\circ} \mathrm{F}$. for the deaerated sample. This value compares favorably with a value of 1.40967 reported (1) for
air-saturated $n$-decane at the same temperature. On the basis of the above comparisons with critically chosen values of the properties of $n$-decane at atmospheric pressure, it appears that the sample of $n$-decane utilized contained less than 0.0062 mole fraction of materials other than $n$-decane and that these impurities are probably saturated hydrocarbons involving 10 carbon atoms per molecule.

## EXPERIMENTAL RESULTS

Illustrative volumetric measurements obtained upon each of the four mixtures investigated are shown in Figure 1. For comparison, the corresponding influence of pressure on the molal

## Propane and $n$-Decane in the Liquid Phase (Continued)

| Pressure, P.S.I.A. | Mole Fraction Propane |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
|  | $400^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |
|  | (134) ${ }^{\text {a }}$ | (249) | (375) | (512) | (655) | (802) | (948) | (1028) ${ }^{\text {e }}$ |  |
| BP | 3.779 | 3.601 | 3.433 | 3.273 | 3.136 | 3.085 | 3.169 | 3.81 |  |
| 200 | $3.765^{\text {b }}$ |  |  |  |  | $\ldots$ | $\ldots$ |  | $\ldots$ |
| 400 | 3.728 | 3.567 | 3.426 |  |  | $\ldots$ | $\ldots$ |  |  |
| 600 | 3.695 | 3.529 | 3.379 | 3.250 |  |  |  |  |  |
| 800 | 3.667 | 3.497 | 3.340 | 3.202 | 3.092 |  |  |  |  |
| 1000 | 3.642 | 3.469 | 3.306 | 3.160 | 3.036 | 2.980 | 3.094 |  |  |
| 1250 | 3.615 | 3.439 | 3.271 | 3.115 | 2.979 | 2.888 | 2.876 | 3.18 |  |
| 1500 | 3.590 | 3.414 | 3.240 | 3.075 | 2.930 | 2.819 | 2.760 | 2.861 | $3.24{ }^{\text {d }}$ |
| 1750 | 3.568 | 3.388 | 3.210 | 3.040 | 2.887 | 2.763 | 2.679 | 2.691 | 2.873 |
| 2000 | 3.547 | 3.363 | 3.182 | 3.009 | 2.850 | 2.715 | 2.614 | 2.571 | 2.639 |
| 2250 | 3.526 | 3.340 | 3.156 | 2.981 | 2.817 | 2.675 | 2.558 | 2.490 | 2.501 |
| 2500 | 3.505 | 3.318 | 3.132 | 2.955 | 2.789 | 2.640 | 2.509 | 2.424 | 2.396 |
| 2750 | 3.487 | 3.298 | 3.111 | 2.932 | 2.764 | 2.610 | 2.471 | 2.369 | 2.314 |
| 3000 | 3.468 | 3.279 | 3.092 | 2.911 | 2.741 | 2.582 | 2.435 | 2.322 | 2.248 |
| 3500 | 3.435 | 3.244 | 3.056 | 2.872 | 2.699 | 2.532 | 2.376 | 2.248 | 2.149 |
| 4000 | 3.406 | 3.214 | 3.024 | 2.840 | 2.663 | 2.490 | 2.328 | 2.189 | 2.075 |
| 4500 | 3.379 | 3.187 | 2.996 | 2.810 | 2.631 | 2.454 | 2.288 | 2.140 | 2.013 |
| 5000 | 3.354 | 3.163 | 2.971 | 2.785 | 2.603 | 2.423 | 2.256 | 2.101 | 1.960 |
| 6000 | 3.309 | 3.117 | 2.925 | 2.739 | 2.553 | 2.373 | 2.201 | 2.038 | 1.881 |
| 7000 | 3.273 | 3.079 | 2.888 | 2.699 | 2.512 | 2.332 | 2.157 | 1.989 | 1.824 |
| 8000 | 3.238 | 3.047 | 2.856 | 2.666 | 2.478 | 2.296 | 2.119 | 1.947 | 1.778 |
| 9000 | 3.207 | 3.018 | 2.825 | 2.636 | 2.446 | 2.264 | 2.084 | 1.910 | 1.738 |
| 10,000 | 3.179 | 2.988 | 2.795 | 2.606 | 2.416 | 2.235 | 2.056 | 1.878 | 1.700 |
|  | $460^{\circ} \mathrm{F}$. |  |  |  |  |  |  |  |  |
|  | (183) ${ }^{\text {a }}$ | (315) | (463) | (621) | (775) | (908) | (987) | (866) |  |
| BP | 4.058 | 3.887 | 3.727 | 3.615 | 3.619 | 3.741 |  |  |  |
| 200 | $4.052^{\text {b }}$ |  |  | $\ldots$ | ... | $\ldots$ | $\ldots$ | $\ldots$ |  |
| 400 | 3.995 | 3.856 |  |  | ... | $\ldots$ | $\ldots$ | $\ldots$ |  |
| 600 | 3.944 | 3.792 | 3.668 |  |  | $\cdots$ | ... | ... | $\ldots$ |
| 800 | 3.898 | 3.739 | 3.598 | 3.525 | 3.594 |  |  |  |  |
| 1000 | 3.859 | 3.692 | 3.542 | 3.446 | 3.440 | 3.539 | $3.88{ }^{\text {d }}$ |  |  |
| 1250 | 3.815 | 3.643 | 3.485 | 3.367 | 3.302 | 3.286 | 3.452 | $4.27^{\text {d }}$ |  |
| 1500 | 3.776 | 3.600 | 3.436 | 3.304 | 3.200 | 3.137 | 3.200 | 3.54 | $4.41{ }^{\text {d }}$ |
| 1750 | 3.741 | 3.563 | 3.394 | 3.249 | 3.121 | 3.030 | 3.034 | 3.186 | 3.580 |
| 2000 | 3.710 | 3.529 | 3.356 | 3.201 | 3.060 | 2.949 | 2.915 | 2.960 | 3.176 |
| 2250 | 3.681 | 3.497 | 3.320 | 3.157 | 3.007 | 2.885 | 2.816 | 2.806 | 2.909 |
| 2500 | 3.655 | 3.468 | 3.287 | 3.119 | 2.963 | 2.831 | 2.735 | 2.692 | 2.740 |
| 2750 | 3.631 | 3.442 | 3.257 | 3.084 | 2.925 | 2.784 | 2.668 | 2.603 | 2.616 |
| 3000 | 3.608 | 3.417 | 3.230 | 3.055 | 2.891 | 2.743 | 2.615 | 2.535 | 2.520 |
| 3500 | 3.565 | 3.372 | 3.184 | 3.005 | 2.831 | 2.672 | 2.531 | 2.427 | 2.366 |
| 4000 | 3.528 | 3.334 | 3.147 | 2.965 | 2.784 | 2.615 | 2.468 | 2.347 | 2.251 |
| 4500 | 3.495 | 3.300 | 3.111 | 2.928 | 2.743 | 2.570 | 2.416 | 2.280 | 2.163 |
| 5000 | 3.468 | 3.271 | 3.081 | 2.895 | 2.709 | 2.533 | 2.372 | 2.226 | 2.093 |
| 6000 | 3.416 | 3.220 | 3.024 | 2.834 | 2.646 | 2.470 | 2.302 | 2.141 | 1.991 |
| 7000 | 3.372 | 3.176 | 2.979 | 2.786 | 2.597 | 2.420 | 2.246 | 2.078 | 1.916 |
| 8000 | 3.332 | 3.134 | 2.940 | 2.747 | 2.557 | 2.378 | 2.200 | 2.027 | 1.859 |
| 9000 | 3.296 | 3.097 | 2.904 | 2.713 | 2.523 | 2.342 | 2.160 | 1.984 | 1.811 |
| 10,000 | 3.260 | 3.062 | 2.869 | 2.681 | 2.492 | 2.309 | 2.125 | 1.945 | 1.767 |

a Values in parentheses represent bubble-point pressures expressed in p.s.i.a.
${ }^{b}$ Volume expressed in cubic feet per pound-mole.

- Retrograde dew point.
${ }^{d}$ Estimated.
volume of propane and $n$-decane at a temperature of $220^{\circ} \mathrm{F}$. has been included in the figure. Experimental information in all respects comparable with that depicted in Figure 1 was obtained for temperatures between $40^{\circ}$ and $460^{\circ} \mathrm{F}$. The detailed experimental data are available through ADI. Smoothed values of the molal volume for even values of composition, pressure, and temperature are recorded in Table I. The standard error of estimate of these experimental values of the molal volume from the smoothed data recorded in Table I was 0.003 cubic foot per pound mole, corresponding to an error of $0.12 \%$ as related to the average value of the molal volumes.

Figure 2 shows a pressure-composition diagram for the propane- $n$-decane system. The solid points represent values of the bubble-point pressure as established from discontinuities in the first derivative of the isothermal change in molal volume with pressure under conditions of constant composition. The open circles correspond to values of the composition of the coexisting gas phase as measured by the partial condensation techniques described (8). The estimated loci of the maxcondentherm, critical, and maximum pressure have been included in Figure 2. A record of the experimentally determined compositions of the coexisting gas phase is available through ADI.

Table II. Properties of Coexisting Gas and Liquid Phases

| Pressure, P.S.I.A. | Dew Point |  | Bubble Point |  | Equilibrium Ratio |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Mole fraction propane | Volume $\mathrm{cu} . \mathrm{ft} . / \mathrm{lb}$. mole | Mole fraction propane | Volume $\mathrm{cu} . \mathrm{ft} . / \mathrm{lb}$. mole |  |  |
|  |  |  |  |  | Propane | $n$-Decane |
|  |  |  | $40^{\circ} \mathrm{F}$. |  |  |  |
| $0.004{ }^{\text {a }}$ | 0.0000 |  | 0.0000 | 3.055 |  | $1.000000^{\circ}$ |
| 25 | 0.9996 |  | 0.3042 | 2.517 | 3.286 | 0.000568 |
| 50 | 0.9998 |  | 0.6172 | 1.977 | 1.620 | 0.000538 |
| 75 | 0.9999 |  | 0.9463 | 1.425 | 1.057 | 0.000596 |
| $79^{\circ}$ | 1.0000 | 63.0 | 1.0000 | 1.347 | 1.000 | 1.000000 |
|  |  |  | $100^{\circ} \mathrm{F}$. |  |  |  |
| $0.073{ }^{\text {a }}$ | 0.0000 | $\ldots{ }^{\text {d }}$ | 0.0000 | 3.166 |  | $1.00000^{6}$ |
| 50 | 0.9979 |  | 0.2973 | 2.630 | 3.356 | 0.00300 |
| 100 | 0.9989 |  | 0.5746 | 2.146 | 1.738 | 0.00249 |
| 150 | 0.9996 | 33. | 0.8253 | 1.732 | 1.211 | 0.00252 |
| $189{ }^{\circ}$ | 1.0000 | 24.5 | 1.0000 | 1.494 | 1.000 | 1.00000 |
|  |  |  | $160^{\circ} \mathrm{F}$. |  |  |  |
| $0.40^{\text {a }}$ | 0.0000 | ${ }^{\text {d }}$ | 0.0000 | 3.302 |  | $1.0000^{\text {b }}$ |
| 50 | 0.9910 |  | 0.1652 | 2.985 | 5.999 | 0.0108 |
| 100 | 0.9948 |  | 0.3178 | 2.705 | 3.130 | 0.00767 |
| 150 | 0.9961 | $39 .{ }^{\text {e }}$ | 0.4584 | 2.456 | 2.173 | 0.00718 |
| 200 | 0.9969 | 28.0 | 0.5899 | 2.232 | 1.690 | 0.00749 |
| 300 | 0.9985 | 16.14 | 0.8275 | 1.865 | 1.207 | 0.00864 |
| $384{ }^{\text {c }}$ | 1.0000 | 10.70 | 1.0000 | 1.761 | 1.000 | 1.0000 |
|  |  |  | $220^{\circ} \mathrm{F}$. |  |  |  |
| $1.59{ }^{\text {a }}$ | 0.0000 | $\ldots{ }^{\text {d }}$ | 0.0000 | 3.443 |  | $1.00000^{\text {b }}$ |
| 50 | 0.9652 | ... | 0.1077 | 3.223 | 8.962 | 0.0390 |
| 100 | 0.9810 |  | 0.2110 | 3.028 | 4.649 | 0.0241 |
| 150 | 0.9863 | $45 .{ }^{\text {e }}$ | 0.3070 | 2.852 | 3.213 | 0.0197 |
| 200 | 0.9890 | 33. | 0.3971 | 2.690 | 2.490 | 0.0183 |
| 300 | 0.9915 | 20.3 | 0.5591 | 2.416 | 1.773 | 0.0193 |
| 400 | 0.9923 | 13.59 | 0.7003 | 2.204 | 1.417 | 0.0257 |
| 600 | 0.9929 | 6.78 | 0.9167 | 2.182 | 1.083 | 0.0850 |
| 678 | 0.9870 | 3.21 | 0.9870 | 3.21 | 1.000 | 1.0000 |
| $618{ }^{\circ}$ | 0.993 | ... |  |  |  |  |
|  |  |  | $280^{\circ} \mathrm{F}$. |  |  |  |
| $5.08^{a}$ | 0.0000 | $\ldots{ }^{\text {d }}$ | 0.0000 | 3.585 |  | 1.0000 |
| 50 | 0.8903 | . . . | 0.0732 | 3.439 | 12.17 | 0.1184 |
| 100 | 0.9403 |  | 0.1488 | 3.292 | 6.320 | 0.0701 |
| 150 | 0.9571 | $50 .{ }^{\text {e }}$ | 0.2195 | 3.160 | 4.360 | 0.0550 |
| 200 | 0.9654 | 36. | 0.2856 | 3.041 | 3.380 | 0.0484 |
| 300 | 0.9737 | 22.8 | 0.4057 | 2.831 | 2.400 | 0.0442 |
| 400 | 0.9766 | 16.07 | 0.5139 | 2.654 | 1.900 | 0.0481 |
| 600 | 0.9770 | 8.94 | 0.7023 | 2.400 | 1.391 | 0.0773 |
| 800 | 0.9671 | 5.02 | 0.8629 | 2.464 | 1.121 | 0.2400 |
| $873{ }^{\prime}$ | 0.9283 | 3.22 | 0.9283 | 3.22 | 1.000 | 1.0000 |
| $622^{\circ}$ | 0.977 |  |  |  |  |  |
|  |  |  | $340^{\circ} \mathrm{F} .{ }^{\text {h }}$ |  |  |  |
| $13.49^{\circ}$ | 0.0000 | $\ldots{ }^{\text {d }}$ | 0.0000 | 3.742 |  | 1.0000 |
| 50 | 0.7147 | . . . | 0.0459 | 3.657 | 15.57 | 0.2990 |
| 100 | 0.8469 | . $\cdot$ | 0.1051 | 3.548 | 8.058 | 0.1711 |
| 150 | 0.8914 | $52 .{ }^{\text {e }}$ | 0.1606 | 3.446 | 5.550 | 0.1294 |
| 200 | 0.9128 | 39. | 0.2128 | 3.353 | 4.290 | 0.1108 |
| 300 | 0.9343 | 24.8 | 0.3099 | 3.184 | 3.015 | 0.09520 |
| 400 | 0.9427 | 17.85 | 0.3988 | 3.034 | 2.364 | 0.09531 |
| 600 | 0.9463 | 10.54 | 0.5595 | 2.790 | 1.691 | 0.1219 |
| 800 | 0.9420 | 6.77 | 0.7043 | 2.707 | 1.338 | 0.1961 |
| $980{ }^{\prime}$ | 0.8673 | 3.27 | 0.8673 | 3.27 | 1.000 | 1.0000 |
| $628^{\circ}$ | 0.946 | . . |  |  |  |  |
|  |  |  | $400^{\circ} \mathrm{F}$. |  |  |  |
| $31.19{ }^{\text {c }}$ | 0.0000 | . $\cdot$ | 0.0000 | 3.962 |  | 1.0000 |
| 50 | 0.3637 | . $\cdot \cdot$ | 0.0190 | 3.926 | 19.12 | 0.6486 |
| 100 | 0.6647 |  | 0.0679 | 3.838 | 9.789 | 0.3597 |
| 150 | 0.7645 | 55. | 0.1144 | 3.753 | 6.683 | 0.2659 |
| 200 | 0.8145 | 41. | 0.1588 | 3.674 | 5.129 | 0.2205 |

Table II. Properties of Coexisting Gas and Liquid Phases (Continued)

| Pressure, P.S.I.A. | Dew Point |  | Bubble Point |  | Equilibrium Ratio |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Mole fraction propane | Volume cu. ft./lb. mole | Mole fraction propane | Volume cu. ft./lb. mole |  |  |
|  |  |  |  |  | Propane | $n$-Decane |
| 300 | 0.8624 | 26.7 | 0.2419 | 3.530 | 3.565 | 0.1815 |
| 400 | 0.8827 | 19.32 | 0.3188 | 3.402 | 2.769 | 0.1722 |
| 600 | 0.8990 | 11.66 | 0.4615 | 3.182 | 1.948 | 0.1876 |
| 800 | 0.8945 | 7.72 | 0.5988 | 3.085 | 1.494 | 0.2630 |
| 1000 | 0.8456 | 4.013 | 0.7450 | 3.274 | 1.135 | 0.6055 |
| $1028{ }^{\prime}$ | 0.7993 | 3.78 | 0.7993 | 3.78 | 1.000 | 1.0000 |
| $640^{\circ}$ | 0.900 |  |  |  |  |  |
|  |  |  | $460^{\circ} \mathrm{F}$. |  |  |  |
| $64.72^{a}$ | 0.0000 |  | 0.0000 | 4.229 |  | 1.0000 |
| 100 | 0.3382 |  | 0.0311 | 4.177 | 10.88 | 0.6830 |
| 150 | 0.5416 | $55 .{ }^{\text {e }}$ | 0.0731 | 4.105 | 7.407 | 0.4946 |
| 200 | 0.6432 | 41. | 0.1136 | 4.035 | 5.662 | 0.4025 |
| 300 | 0.7379 | 27.1 | 0.1890 | 3.906 | 3.904 | 0.3232 |
| 400 | 0.7803 | 19.86 | 0.2586 | 3.791 | 3.017 | 0.2963 |
| 600 | 0.8129 | 12.23 | 0.3867 | 3.623 | 2.102 | 0.3051 |
| 800 | 0.8104 | 8.04 | 0.5168 | 3.630 | 1.568 | 0.3924 |
| $988{ }^{\prime}$ | 0.7120 | 4.50 | 0.7120 | 4.50 | 1.000 | 1.0000 |
| $650{ }^{\circ}$ | 0.814 |  |  |  |  |  |

a Vapor pressure of $n$-decane ( 9 ).
${ }^{b}$ As a result of the number of significant figures reported for dew-point gas compositions, some discrepancies in values of the equilibrium ratio, computed from the compositions reported and depicted in Figure 4, may exist at this temperature.
c Vapor pressure of propane (10).
${ }^{d}$ Dew-point volumes of $n$-decane expressed in cubic feet per pound-mole: at $100^{\circ} \mathrm{F}$. $=82,200$; at $160^{\circ} \mathrm{F} .=$ 16,600 ; at $220^{\circ} \mathrm{F}=4339$; at $280^{\circ} \mathrm{F} .=1447$; at $340^{\circ} \mathrm{F} .=591$. Values based upon calorimetric vaporization measurements ( 5 ).
${ }^{e}$ Dew-point volumes involve a somewhat larger uncertainty since they were established from volumetric measurements in the two-phase region.
${ }^{f}$ Estimated critical state.

- Estimated maxcondentherm.
${ }^{\circ}$ Data at $340^{\circ} \mathrm{F}$. interpolated.


Figure 3. Experimental composition of dew-point gas

In the interest of depicting the behavior in somewhat greater detail at compositions near pure propane, an enlarged portion of Figure 2 is presented in Figure 3. From the information depicted in Figures 2 and 3, the equilibrium ratios have been


Figure 4. Equilibrium ratios for propane and $n$-decane

|  | Critical |  | Maxcondentherm |  | Maximum Pressure |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Fraction Propane | Pressure, p.s.i.a. | Temp., ${ }^{\circ} \mathrm{F}$. | Pressure, p.s.i,a. | Temp., ${ }^{\circ} \mathrm{F} .$ | Pressure, p.s.i.a. | Temp., ${ }^{\circ} \mathrm{F}$. |
| $0.0{ }^{\text {a }}$ | 304.0 | 655.0 | 304.0 | 655.0 | 304.0 | 655.0 |
| 0.1 | 387.5 | 641.3 | 361.0 | 642.8 | 406.0 | 633.5 |
| 0.2 | 477.0 | 625.1 | 418.0 | 628.5 | 511.5 | 610.0 |
| 0.3 | 580.5 | 604.8 | 472.0 | 611.7 | 618.0 | 582.8 |
| 0.4 | 688.0 | 580.4 | 521.5 | 592.0 | 724.0 | 552.6 |
| 0.5 | 793.5 | 551.7 | 562.0 | 570.0 | 824.0 | 519.6 |
| 0.6 | 892.5 | 515.6 | 600.0 | 543.8 | 918.0 | $483.0^{\text {b }}$ |
| 0.7 | 979.0 | $466.8{ }^{\text {b }}$ | 632.0 | 510.7 | 996.0 | 444.2 |
| 0.8 | 1028.0 | 399.5 | 649.5 | $467.0^{\text {b }}$ | 1028.0 | 400.2 |
| 0.9 | 932.0 | 308.5 | 639.0 | 399.5 | 962.0 | 342.0 |
| $1.0{ }^{\text {c }}$ | 617.4 | 206.3 | 617.4 | 206.3 | 617.4 | 206.3 |

${ }^{a}$ Critical state of $n$-decane (1).
${ }^{b}$ Values at this and higher temperatures are subject to greater uncertainty.
c Critical state of propane (1).


Figure 5. Pressure-temperature diagram for the propane-n-decane system
standard error of estimate of the experimentally determined composition data from the information presented in Table II was 0.0028 mole fraction.
As a matter of interest, Figure 4 shows the product of the pressure and the equilibrium ratio for both components as a function of pressure for each of the several temperatures investigated. The critical states of the components have been included also. There exists a larger measure of uncertainty in the equilibrium ratios for propane when present in dilute solution in $n$-decane than at other states. This has been indicated by the dashed boundary curve shown in Figure 4. The behavior depicted in this diagram is similar to that found for other paraffin hydrocarbon systems containing propane.
A pressure-temperature diagram showing the behavior of the bubble-point liquid and dew-point gas for each of the mixtures experimentally investigated constitutes Figure 5. The loci of the unique states have been included. The diagram has been extended to the critical temperature of $n$-decane. Since uncertainty exists as to the behavior of this system at temperatures significantly above those covered by this investigation, the curves have been dashed at temperatures beyond the range of experimental investigation. Estimated values of the unique states which include the critical, maxcondentherm, and loci of maximum pressure are set forth in Table III. Much larger uncertainties exist in the values reported for these states than in the case of the information presented in Tables I and II. Extensive interpolation of the volumetric and phase equilibrium data was required to arrive at the pressures and temperatures recorded in Table III. For this reason, uncertainties as large as
$4 \%$ in pressure and $5^{\circ} \mathrm{F}$. in temperature are to be expected. The probable error in the temperatures and pressures is somewhat smaller but was not established with certainty.

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