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Letter to the Editor

An almost semicentennial formula for a simple approximation of the natural frequencies of Bernoulli–Euler beams

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Even though its applications were rarely discussed, the brief note which is referred to in the title should be cited in the present investigations on the subject. The reason is that the mentioned study is probably the earliest investigation developed on the adoption of approximate formulae for the determination of frequencies of free transverse vibrations of particularly restrained beams.

As a part of a program initiated in 1952 by the Structural Research Laboratory, Department of Civil Engineering, of the University of Illinois (USA), the professors Newmark and Veletsos [1] pioneered the employment of the basic formulae that would be of use to structural engineers to estimate, at the preliminary design stage, a simple approximation of the natural frequencies in the case of beams elastically restrained against rotation at the ends.

Consider a uniform beam to be oscillating in one of its natural modes of free vibration. EI is the modulus of flexural rigidity, m is its mass per unit length, and L denotes the span length between supports which are non-deflecting but both of them offer a linear resistance to end rotations of the beam (see Fig. 1).

The restraining moments at the corresponding ends are related to the end rotations by the equations

$$M_1 = K_{r1}\theta_1 \quad \text{and} \quad M_2 = K_{r2}\theta_2. \quad (1,2)$$

Here K_{r1} and K_{r2} are the rotational spring constants and they are in connection to the characteristics of the beam by the dimensionless coefficients as follows:

$$\beta_1 = \frac{K_{r1}L}{EI} \quad \text{and} \quad \beta_2 = \frac{K_{r2}L}{EI}. \quad (3,4)$$

It can be pointed out that for a hinged end $\beta = 0$ and for a clamped end $\beta \rightarrow \infty$.

The frequencies f_n of the elastically restrained beam have been stated as the product of a dimensionless coefficient C_n multiplied by the fundamental frequency of same beam hinged at the

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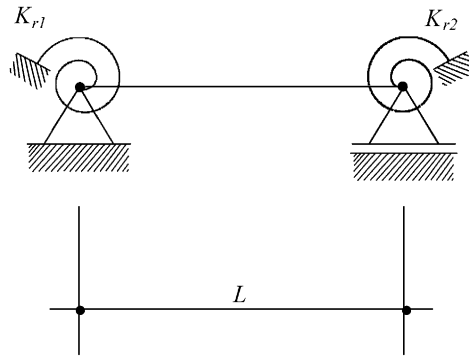


Fig 1. Beam elastically restrained at the ends.

ends

$$f_n = C_n \frac{\pi}{2L^2} \sqrt{\frac{EI}{m}}. \quad (5)$$

Furthermore, the following simple approximation for the coefficient C_n , for the fundamental frequency ($n = 1$) and the higher natural frequencies, was also reported in Ref. [1]:

$$C_n = \left[n + \frac{1}{2} \left(\frac{\beta_1}{5n + \beta_1} \right) \right] \left[n + \frac{1}{2} \left(\frac{\beta_2}{5n + \beta_2} \right) \right]. \quad (6)$$

Additionally, when the end restraints are very small, i.e., for values of the coefficients β approaching zero, the value of C_n is proposed under the following form:

$$C_n \cong n^2 + 0.1(\beta_1 + \beta_2). \quad (7)$$

These approximate equations are similar to those developed by the same authors in a previous paper [2] in relation to buckling loads for partially restrained beams.

On the other hand, the general equation of frequencies for generally restrained Bernoulli–Euler beams presented by Maurizi et al. [3], and also quoted in Refs. [4–7], can be used for a wide range of combinations of boundary conditions. In the present study the transcendental equation is as follows [6]:

$$2R_1 R_2 \varphi_1(y_n) y_n^2 + (R_1 + R_2) \varphi_6(y_n) y_n - \varphi_4(y_n) = 0, \quad (8)$$

where

$$(y_n/L)^4 = k_n^4 = \omega_n^2(\rho A/EI), \quad (9)$$

$$\varphi_1(y_n) = \sin y_n \sinh y_n, \quad (10a)$$

$$\varphi_4(y) = \cos y_n \cosh y_n - 1, \quad (10b)$$

$$\varphi_6(y_n) = \sin y_n \cosh y_n - \sinh y_n \cos y_n \quad (10c)$$

Table 1
Comparison of the frequency parameters between the exact values and the N–V method

β_1	β_2	Mode number					
		1	2	3	4	5	
0	0	3.14159265	6.28318531	9.42477796	12.56637061	15.70796327	Exact
		3.14159265	6.28318531	9.42477796	12.56637061	15.70796327	N–V
0.01	10	3.66600954	6.68815630	9.75207435	12.84001779	15.94263705	Exact
		3.62940846	6.66515656	9.73440922	12.82589889	15.93110087	N–V
0	100	3.88918500	7.00322722	10.11854568	13.23541270	16.35372435	Exact
		3.81699044	6.96065983	10.08463369	13.20465869	16.32419428	N–V
1	10,000	4.04143802	7.13313310	10.24653159	13.37459282	16.50651642	Exact
		4.00442698	7.10348394	10.23171197	13.36677608	16.50443133	N–V
10,000	10,000	4.72909537	7.85163553	10.99341157	14.13434261	17.27531056	Exact
		4.71160397	7.85241241	10.99322162	14.13403162	17.27484240	N–V
∞	∞	4.73004074	7.85320462	10.99560784	14.13716549	17.27875966	Exact
		4.73004074	7.85320462	10.99560784	14.13716549	17.27875966	N–V

and

$$R_1 = EI/K_{r1}L, \quad R_2 = EI/K_{r2}L. \tag{11a, b}$$

Obviously, $R_1 \equiv 1/\beta_1$ and $R_2 \equiv 1/\beta_2$.

In Table 1 the first five natural frequencies of a beam elastically restrained against rotation at the ends are presented. Numerical results were determined by means of expression (6) (henceforth N–V) and subsequent application of Eq. (5) and are tabulated for various dimensionless coefficients β . Additionally, the free vibration frequencies were computed according to the Bernoulli–Euler theory by employing Eq. (8) (henceforth “exact”).

It appears that, the maximum difference between the exact solution and the approximations does not exceed 4% for the fundamental frequency and a little over 1% for the higher frequencies [1]. It was found that the absolute maximum error of Eq. (6) is obtained for a hinged–clamped beam [1] and the results show that errors greater than 2% in the fundamental frequency occur only for beams having one end either hinged or practically unrestrained and the other end clamped or approximately fixed.

On the other hand, the results obtained by using relation (7) are in error by less than 1.5% when the sum $(\beta_1 + \beta_2)$ is less than 1.0.

Clearly, the published work of Newmark and Veletsos [1] reveals that the calculations of the parameters, which significantly affect the dynamic performance of beams, are simplified greatly with the use of these approximate formulations.

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