

A Dispersion Analysis for Difference Schemes: Tables of Generalized Airy Functions*

By R. C. Y. Chin and G. W. Hedstrom

Abstract. This paper contains graphs and tables of the function

$$Ai_{p,q}(\alpha, x) = \int_{-\infty}^{\infty} (2\pi)^{-1} \exp\{iy^p/p - \alpha y^q/q + ixy\} dy$$

and its indefinite integral for $p = 3, 5, 7$, for $q = 2, 4, 6$, and for several values of α with $\alpha \geq 0$. It is shown how these tables should influence the choice of an artificial viscosity for a difference scheme for a linear hyperbolic equation.

1. Introduction. This paper consists primarily of graphs and tables of values of the generalized Airy function

$$(1.1) \quad Ai_{p,q}(\alpha, x) = (2\pi)^{-1} \int_{-\infty}^{\infty} \exp\{iy^p/p - \alpha y^q/q + ixy\} dy$$

and its integral

$$(1.2) \quad Aii_{p,q}(\alpha, x) = (2\pi)^{-1} \int_{-\infty}^{\infty} y^{-1} \exp\{iy^p/p - \alpha y^q/q + ixy\} dy.$$

Here, $\alpha \geq 0$, x is real, and p and q are natural numbers such that q is even and p is odd, $p \geq 3$. The improper integral (1.2) is the Cauchy principal value. We computed these tables because the solution of a linear hyperbolic equation with constant coefficients behaves near a discontinuity like the integral $Aii_{p,q}$. See [4] and [5]. Our tables show the magnitude and extent of the numerically induced oscillations.

Our results show that it is better to add an artificial viscosity to a nondissipative difference scheme than to use a difference scheme with the viscosity built in. The reason is that if $q < p$ as in the upstream-difference method, the viscosity becomes predominant as t/h increases, and if $q > p$ as in the Lax-Wendroff method, the viscosity weakens as t/h increases. On the other hand, for a nondissipative scheme with order of accuracy $p - 1$, if the artificial viscosity is a discretization of

$$(1.3) \quad Q = -(-1)^{q/2} \delta h^{q(p-1)/p} q^{-1} \partial^q u / \partial x^q$$

(q even), then the amount of viscosity as measured by the parameter α in (1.1) and (1.2) is independent of h ,

$$\alpha = \delta t^{(p-q)/p}.$$

We see that α gives a measure of the effective diffusive time scale. Note, though, that with the artificial viscosity (1.3) α is an increasing function of t if $p > q$ and a decreasing function of t if $p < q$.

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The behavior of a high-order difference scheme near a discontinuity is as follows. Suppose we use a nondissipative difference scheme with order of accuracy $p - 1$ and artificial viscosity (1.3). Then for fixed t the oscillations near a discontinuity extend over a distance

$$O(h^{(p-1)/p}).$$

We see that there is a slight improvement as p increases. Our tables show that as p increases ($p = 3, 5, 7$) there is also a decrease in the amplitude of the oscillations. As we increase q , though, the damping near the wave front is decreased. In fact, it is only for $q = 2$ that there is no overshoot.

Let us emphasize that we consider here only linear equations with linear artificial viscosity. The behavior of difference schemes for nonlinear equations with shocks is different. Nonlinear artificial viscosities of the form

$$(1.4) \quad Q = -(-1)^m \delta h^{2m(p-1)/p} (2m)^{-1} D^m (|Du|^r D^m u)$$

are in common use [9, p. 313], [7], where Du denotes $\partial u / \partial x$. It is not usual, though, to use this particular power of h . The reason for using a nonlinear viscosity of the form (1.4) is that one wants the effect to be greatest when $|Du|$ is large. If r is even, the computation is more efficient because we do not need to check $\text{sgn}(Du)$. Finally, let us remark that the clipping algorithm of Boris and Book [1] is very good for discontinuities of the sort $u = \text{sgn}(x)$, but for $u = \text{sgn}(x) - x$ it produces a staircase.

We have concentrated on the behavior of a difference scheme near a discontinuity of the solution of the differential equation. It is also of interest to know the behavior near a discontinuity of a derivative. If the discontinuity were in the derivative of order r , we would need to compute tables of the $(r + 1)$ -fold indefinite integral of the generalized Airy function $Ai_{p,q}$.

Our graphs and tables are in the Appendix which appears in the microfiche section of this issue. In Section 2 we describe the graphs and tables and tell how we computed them. In Section 3 we show how the generalized Airy functions apply to various difference schemes.

2. Description of the Tables. In the Appendix are three sets of graphs with adjoining tables. The first set (Figures 1–17 and Tables 1–9) gives four local extrema of $Aii_{p,q}(\alpha, x)$. The second set (Figures 18–23 and Tables 10–12) shows the function

$$(2.1) \quad F_q(x) = (2\pi)^{-1} \int_{-\infty}^{\infty} \exp\{-y^q/q + ixy\} dy$$

and its integral. Finally, the third set (Figures 24–105 and Tables 13–53) consists of graphs and tables for the functions $Ai_{p,q}$ and $Aii_{p,q}$.

The function $Aii_{p,q}(\alpha, x)$ is known [5] to oscillate about the value $-1/2$ for negative values of x . If $p = 3$ and either $\alpha = 0$ or $q = 2$, we have

$$\lim_{x \rightarrow \infty} Aii_{p,q}(\alpha, x) = 1/2$$

monotonically. Otherwise, the function $Aii_{p,q}(\alpha, x)$ oscillates about the value 1/2 for positive values of x . The amplitudes of these oscillations are important in numerical analysis. The first set of graphs and tables shows the values for $p = 3, 5, 7$ and $q = 2, 4, 6$ of $Aii_{p,q}(\alpha, x)$ at the first three negative local extrema and at the first positive local maximum. Of the three negative local extrema the first is a local minimum and usually the absolute minimum, the second is a local maximum, and the third is a local minimum.

We point out some peculiarities. For $q = 2$ and $p = 5$ or $p = 7$ the first two negative local extrema coalesce and disappear near $\alpha = 3$ or $\alpha = 3.2$, respectively. In fact, we see values of α for which the first negative local minimum is greater than $-1/2$. Similarly, for $q = 2$ and $p = 5$ the third and fourth negative extrema coalesce near $\alpha = 4.8$, and for $q = 2$ and $p = 7$ the first and second positive local extrema coalesce near $\alpha = 4.1$. For $p = 3$ and $q = 2$ there is no positive local maximum. For $p = 3, q = 4$, and $\alpha < 0.3$ the first positive local maximum is too flat to compute accurately. For $p = 3$ and $q = 6$ the first two positive local extrema coalesce as α decreases to some value between 0.2 and 0.3.

The second set of graphs and tables is labeled "P-term missing" and shows the function $F_q(x)$ defined by Eq. (2.1) and its integral, the Cauchy principal value of

$$Fi_q(x) = (2\pi i)^{-1} \int_{-\infty}^{\infty} y^{-1} \exp\{-y^q/q + ix y\} dy.$$

The importance of these functions is that they show the limiting behavior of the generalized Airy functions for large viscosity,

$$(2.2) \quad \lim_{\alpha \rightarrow \infty} \alpha^{1/q} Ai_{p,q}(\alpha, \alpha^{1/q} x) = F_q(x), \quad \lim_{\alpha \rightarrow \infty} Aii_{p,q}(\alpha, \alpha^{1/q} x) = Fi_q(x).$$

For this reason we include the case $q = 2$, even though F_2 is the Gaussian function

$$F_2(x) = (2\pi)^{-1/2} \exp\{-x^2/2\},$$

and $Fi_2(x)$ differs from the error function by 1/2. It is clear that F_q is an even function, so that it would have been sufficient to plot and print its values for positive x . For ease of comparison with the generalized Airy functions we used for both F_q and $Ai_{p,q}$ a format appropriate for the Airy functions. We remark that since $\exp\{-y^q/q\}$ and e^{ixy} are G-functions [8], it follows that $F_q(x)$ is also a G-function. In fact, it is not hard to show that in terms of the hypergeometric function ${}_0F_{q-2}$ we have

$$F_q(x) = \sum_{m=0}^{q-2} b_m x^m {}_0F_{q-2}(; \delta_{m1}, \delta_{m2}, \dots, \delta_{m,q-2}; (-1)^{q/2} x^q / q^{q-1}),$$

$$b_m = 0 \quad (m \text{ odd}),$$

$$b_m = (-1)^{m/2} q^{(m+1)/q} \Gamma((m+1)/q) / (m! q \pi) \quad (m \text{ even}).$$

Here, $\delta_{m1}, \delta_{m2}, \dots, \delta_{m,q-2}$ are the consecutive numbers $(m+2)/q, (m+3)/q, \dots, (m+q)/q$ with q/q omitted.

The first three negative local extrema for Fi_4 and Fi_6 are as follows

x	$Fi_4(x)$
-2.44196790	-0.55220867
-4.79724416	-0.49384599
-6.81358116	-0.50082421
x	$Fi_6(x)$
-2.50081385	-0.57059517
-4.93258336	-0.47901803
-7.23239945	-0.50617746

Hence, by (2.2) we have the limiting behavior of the first three local extrema of $Aii_{p,q}$ for $q = 4, 6$. It is obvious that Fi_2 is monotone.

The final set of graphs and tables is of values of $Ai_{p,q}(\alpha, x)$ and $Aii_{p,q}(\alpha, x)$ for $p = 3, 5, 7$ and $q = 2, 4, 6$. They appear in the order $(p, q) = (3, 2), (3, 4), (3, 6), (5, 2), \dots$. For all values of p and q we give the generalized Airy function and its integral for $\alpha = 0.5, 1, 2, 4$. The case $\alpha = 0$ is given only for $q = 2$, because

$Ai_{p,q}(0, x)$ is independent of q . For $q = 2$ and $p = 5, 7$ we include the case $\alpha = 3$ in order to show the coalescence of the first two negative extrema.

We close this section with a description of the methods of computation. The derivative of order r of the generalized Airy function may be written

$$(2.3) \quad Ai_{p,q}^{(r)}(\alpha, x) = \pi^{-1} \operatorname{Re} \int_0^\infty (iy)^r \exp\{iy^p/p - \alpha y^q/q + ixy\} dy,$$

$r = -1, 0, 1, \dots$. In order to make the exponential decay as $|y| \rightarrow \infty$, we replace the path of integration by the ray in the complex plane

$$y = ze^{i\theta}, \quad 0 < z < \infty, \theta = \pi/(2(p+q)).$$

For $r = -1$ we must add the contribution $2\theta/\pi$ from the pole at the origin. For any given positive number ϵ it is easy to find an R such that

$$\pi^{-1} \left| \int_R^\infty z^r \exp\{ie^{ip\theta} z^p/p - \alpha e^{iq\theta} z^q/q + ixe^{i\theta} z\} dz \right| < \epsilon.$$

We now use Romberg integration to compute the integral on the interval $0 \leq z \leq R$. In fact, the trapezoid rule for this integral is a finite complex Fourier series which is evaluated by the fast Fourier transform. The values of $Ai_{p,q}$ and $Aii_{p,q}$ were computed by this method. The values of F_q and Fi_q were computed in the same way but without the rotation of the path of integration. In order to find the local extrema of $Aii_{p,q}$, we used Newton's method, which requires evaluation of the integrals (2.3) for $r = -1, 0, 1$.

We checked our results as follows. First, it is easy to see that

$$Ai_{3,2}(\alpha, x) = \exp\{\alpha x/2\} Ai(x + \alpha^2/4),$$

where Ai is the usual Airy function. Second, the function $Aii_{p,q}$ is a solution of the ordinary differential equation

$$(2.4) \quad (-1)^{(p-1)/2} u^{(p)} - \alpha(-1)^{q/2} u^{(q)} + xu' = 0,$$

which may be solved numerically once we have evaluated the integral (2.3) at a starting

value x for $r = -1, 0, 1, \dots, \max(p, q) - 2$. The difficulty with this method is that for $q \geq 6$ and for $p \geq 7$ (and also for $p = 5$ if $x < 0$) the differential equation (2.4) has a parasitic solution which dominates $Aii_{p,q}$ as x increases and another parasitic solution which dominates $Aii_{p,q}$ as x decreases. Hence, we must restart the computation fairly often. We found that it is better to start at a maximal value of $|x|$ and go toward the origin, using a stiff equation solver [6].

We make a remark regarding our method of computing values of the integral (2.3). The method is adequate for the values of x we used, but for larger values of $|x|$ there would be roundoff error from cancellation. We would then need to use a contour which better approximates the optimal path through the saddle points [4], [5].

3. Applications to Difference Schemes. We consider difference schemes for the equation

$$(3.1) \quad u_t + u_x = 0 \quad (-\infty < x < \infty, t > 0)$$

with $u(x, 0) = \text{sgn}(x)/2$. Our applications are based on the amplification function [9, p. 67] for the difference scheme. We assume that the difference scheme is stable. The amplification function has the form

$$G(\xi) = \exp\{\lambda(-i\xi + iA(\xi) - B(\xi))\},$$

where A and B are real, periodic functions such that

$$\begin{aligned} A(\xi) &= c\xi^p/p + O(|\xi|^{p+2}) \quad (\xi \rightarrow 0), \\ B(\xi) &= \epsilon\xi^q/q + O(\epsilon|\xi|^{q+2}) \quad (\xi \rightarrow 0). \end{aligned}$$

Here, p is odd, q is even, $c \neq 0$, and λ is the mesh ratio $\lambda = \Delta t/h$, $h = \Delta x$. It may happen that $B = 0$, in which case the difference scheme is nondissipative. Otherwise, we have $\epsilon > 0$.

It was shown in [4] that if v_k^n denotes the solution of the difference scheme at the grid point $x = kh$, $t = n\Delta t$ with $v_k^0 = \text{sgn}(kh)/2$, then near the wave front we have the asymptotic estimate

$$(3.2) \quad \begin{aligned} V_k^n &\sim Aii_{p,q}(\alpha, y), \\ \alpha &= \epsilon|c|^{-q/p}(\lambda t/h)^{(p-q)/p}, \\ y &= \omega \text{sgn}(c)|c|^{-1/p}(t/h)^{(p-1)/p}. \end{aligned}$$

Here, ω is the scaled distance from the wave front $\omega = (x-t)/t$. It was shown in [4] that the region of validity of (3.2) if $q < p$ is

$$(3.3) \quad (\epsilon t/h)|c|^{-(pq+2)/(p^2-p)}|\omega|^{(q+2)/(p-1)} \rightarrow 0,$$

$$(3.4) \quad |\omega||c|^{-1/p}(t/h)^{(p-1)/(p+2)} \rightarrow 0.$$

If $p < q$, the region of validity of (3.2) is [4]

$$(3.5) \quad |\omega||c|^{-1/q}(t/h)^{(q-1)/(q+2)} \rightarrow 0$$

together with (3.3). Actually, the condition (3.3) was not needed in [4] because ϵ was taken to be constant there; it is not hard to show by the methods of [4] that (3.3) is needed if ϵ is allowed to vary. We remark that the numbers p , q , c , and ϵ for a difference scheme may be obtained from the Maclaurin series for $\log(G(\xi))$.

If p and q are known, c and ϵ may alternatively be obtained by a modification of the method used to find the local truncation error. This method is easier. One substitutes into the difference scheme a smooth solution u of the equation

$$(3.6) \quad u_t + u_x = (-1)^{(p-1)/2} cp^{-1} h^{p-1} \partial^p u / \partial x^p - (-1)^{q/2} \epsilon q^{-1} h^{q-1} \partial^q u / \partial x^q.$$

One then makes Taylor expansions, using Eq. (3.6) to eliminate derivatives with respect to t , and one chooses c and ϵ so that the low-order terms cancel. We give an example in a moment. It is remarkable that this procedure is valid for a differential equation with a discontinuous solution, but it was shown in [4] and [5] that it is valid under conditions (3.3), (3.4) if $q < p$ and under conditions (3.3), (3.5) if $p < q$. Chin [2] has found the error of the approximation in a special case.

We now give some examples.

Example 1. Upstream differences. For the upstream-difference method

$$(3.7) \quad v_k^{n+1} - v_k^n = \lambda(v_k^n - v_{k-1}^n)$$

we know that $q = 2$ and $p = 3$. Upon substituting a smooth function u into Eq. (3.7) and using Taylor expansions, we obtain

$$(3.8) \quad \begin{aligned} u_t + (\lambda h/2)u_{tt} + (\lambda^2 h^2/6)u_{ttt} + O(h^3) \\ = -u_x + (h/2)u_{xx} - (h^2/6)u_{xxx} + O(h^3). \end{aligned}$$

In our case Eq. (3.6) becomes

$$u_t = -u_x + (\epsilon h/2)u_{xx} - (ch^2/3)u_{xxx},$$

so that

$$u_{tt} = u_{xx} - \epsilon hu_{xxx} + O(h^2), \quad u_{ttt} = -u_{xxx} + O(h).$$

Consequently, the terms of (3.8) cancel to within $O(h^3)$ if and only if

$$\epsilon = 1 - \lambda, \quad c = (1 - \lambda)(1 - 2\lambda)/2.$$

Thus, if $\lambda \neq 1/2$, we have obtained ϵ and c for the estimate (3.2) under conditions (3.3), (3.4). Note that $p \geq 5$ if $\lambda = 1/2$.

Example 2. The Lax-Wendroff method. The Lax-Wendroff method [9, p. 331], when used for (3.1) has the amplification factor

$$G(\xi) = 1 - i\lambda \sin \xi - \lambda^2(1 - \cos \xi),$$

so that

$$\log G(\xi) = \lambda(i\xi + i(1 - \lambda^2)2^{-1}\xi^3/3 - \lambda(1 - \lambda^2)2^{-1}\xi^4/4 + O(\xi^5))$$

as $\xi \rightarrow 0$. Thus, we have (3.2) under conditions (3.3), (3.5) with $p = 3, q = 4$.

$$c = (1 - \lambda^2)/2, \quad \epsilon = \lambda(1 - \lambda^2)/2.$$

Example 3. Fromm's method. Fromm's method [3] when applied to (3.1) has the amplification factor

$$G(\xi) = 1 - i(\lambda/2) \sin \xi (1 + e^{-i\xi}) - (\lambda/2)(\lambda + (\lambda - 2)e^{-i\xi})(1 - \cos \xi).$$

Consequently, we have

$$\log G(\xi) = \lambda(-i\xi + ic\xi^3/3 - \epsilon\xi^4/4 + O(\xi^5))$$

as $\xi \rightarrow 0$ with

$$c = -(1 - \lambda)(1 - 2\lambda)/4, \quad \epsilon = (1 - \lambda)(1 - \lambda + \lambda^2)/2.$$

If $0 < \lambda < 1/2$ or $1/2 < \lambda < 1$, we have (3.2) under conditions (3.3), (3.5) with $p = 3$ and $q = 4$. If $\lambda = 1/2$, we have to determine p ($p \geq 5$) and c by keeping higher order terms.

Example 4. The leapfrog scheme. The leapfrog scheme for (3.1) is

$$(3.9) \quad v_k^{n+1} - v_k^{n-1} = \lambda(v_{k+1}^n - v_{k-1}^n).$$

The solution to difference equation (3.9) has two waves, one with amplification factor

$$G(\xi) = -i\lambda \sin \xi + (1 - \lambda^2 \sin^2 \xi)^{1/2},$$

the other with amplification factor

$$G_2(\xi) = -i\lambda \sin \xi - (1 - \lambda^2 \sin^2 \xi)^{1/2}.$$

The wave generated by G_2 has a front with speed -1 , but it is easy to show that this wave has amplitude $O(h^2)$ if the starting values v_k^0 and v_k^1 are correct to within $O(h^2)$. Consequently, we focus our attention on G_1 . It is clear that for $0 < \lambda < 1$ we have

$$|G_1(\xi)| = 1 \quad (-\infty < \xi < \infty).$$

Hence, there is no q -term, or equivalently, $\alpha = 0$ in (3.2). The Maclaurin series for $\log G_1(\xi)$ has the form

$$\log G_1(\xi) = \lambda(-i\xi + i(1 - \lambda^2)2^{-1}\xi^3/3 + O(\xi^5)).$$

Consequently, we have $p = 3$ and

$$(3.10) \quad c = (1 - \lambda^2)/2.$$

Our asymptotic relation (3.2) is valid only if (3.4) holds, but because the difference scheme is not dissipative, the oscillations extend over a long distance. It is known [10] that the solution of (3.9) with initial data

$$v_k^0 = \operatorname{sgn}(kh)/2, \quad v_k^1 = \operatorname{sgn}(kh - \lambda h)/2$$

differs from the exact solution to (3.1) by $O((h/t)^{1/2})$ over every interval

$$|x| < \theta t \quad (0 < \theta < 1).$$

In order to reduce this error, one usually adds an artificial viscosity. If this artificial viscosity is linear, the model equation (3.6) takes the form

$$u_t + u_x = -(ch^2/3)u_{xxx} - \epsilon(-1)^{q/2} q^{-1} h^{q-1} \partial^q u / \partial x^q.$$

One ordinarily chooses $q = 2$ or $q = 4$. If $q = 2$ and $\epsilon = \delta h^{1/3}$ for a fixed constant δ , then (3.2) takes the form

$$\begin{aligned} v_k^n &\sim Aii_{3,2}(\alpha, y), \\ \alpha &= \delta c^{-2/3} (\lambda t)^{1/3}, \\ y &= (x - t)(ct)^{-1/3} h^{2/3} \end{aligned}$$

with c as in (3.10). Note that this choice of ϵ makes α independent of h . The artificial viscosity is a discretization of $(\delta h^{4/3}/2)u_{xx}$. Similarly, if we want $q = 4$ and

$$v_k^n \sim Aii_{3,4}(\alpha, y)$$

with α independent of h , we take $\epsilon = \delta h^{1/3}$. The artificial viscosity is then a discretization of $-(\delta h^{8/3}/4)u_{xxxx}$.

Example 5. Fourth-order schemes. For a fourth-order, nondissipative scheme Eq. (3.2) takes the form

$$v_k^n \sim Aii_{5,q}(0, y)$$

under condition (3.4) with $p = 5$. Here,

$$y = (x - t)(|c|t)^{-1/5} \operatorname{sgn}(c)h^{-4/5}.$$

If we want positive α independent of h , we add to the difference scheme an artificial viscosity which is a difference approximation to one of the following

$$\begin{aligned} \delta h^{1.6} \partial^2 u / \partial x^2 & \quad (q = 2), \\ -\delta h^{3.2} \partial^4 u / \partial x^4 & \quad (q = 4), \\ \delta h^{4.8} \partial^6 u / \partial x^6 & \quad (q = 6), \dots \end{aligned}$$

Other examples of model equations (3.6) are given in [11].

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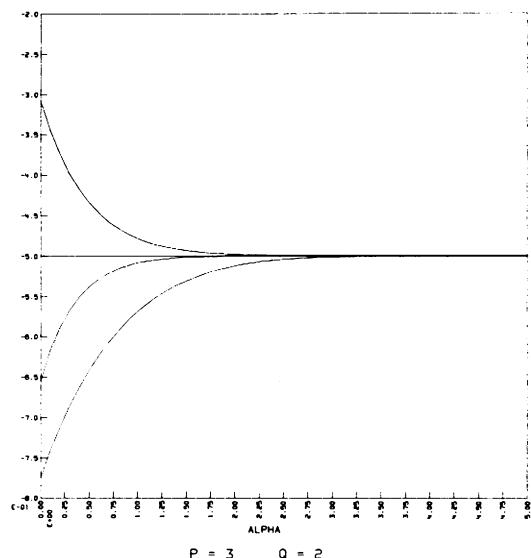
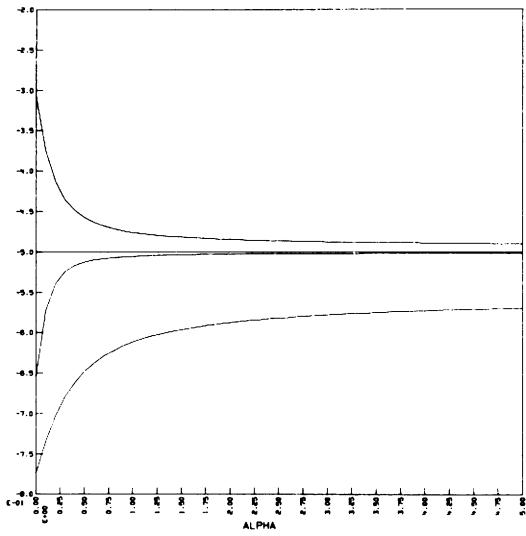


Figure 1



P = 3 Q = 4

Figure 2

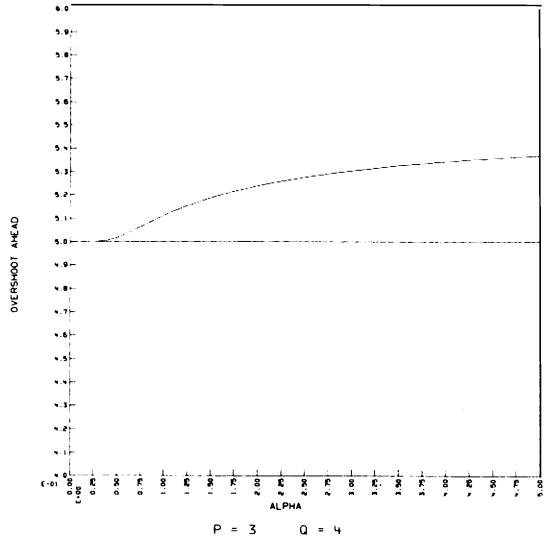


Figure 3

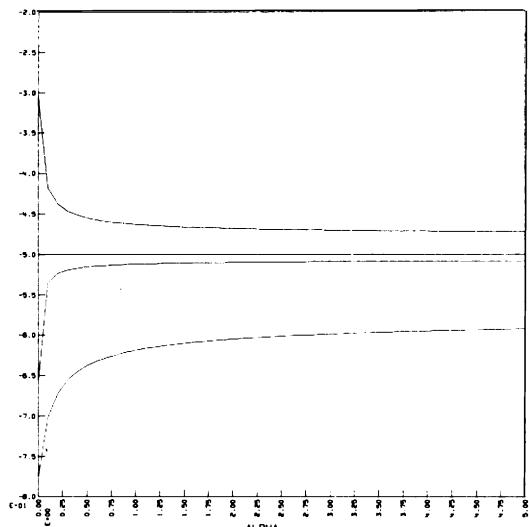


Figure 4

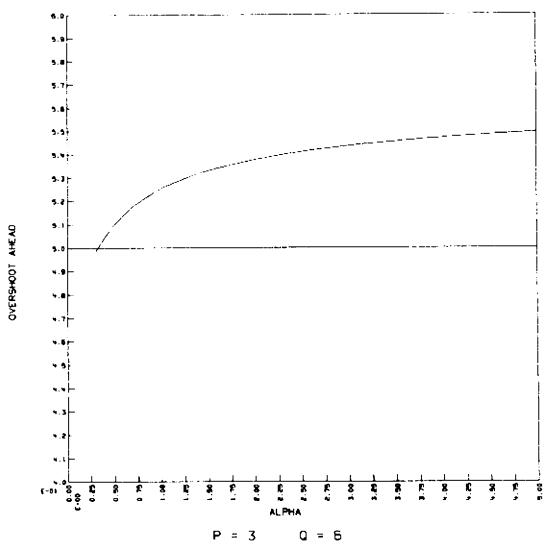


Figure 5

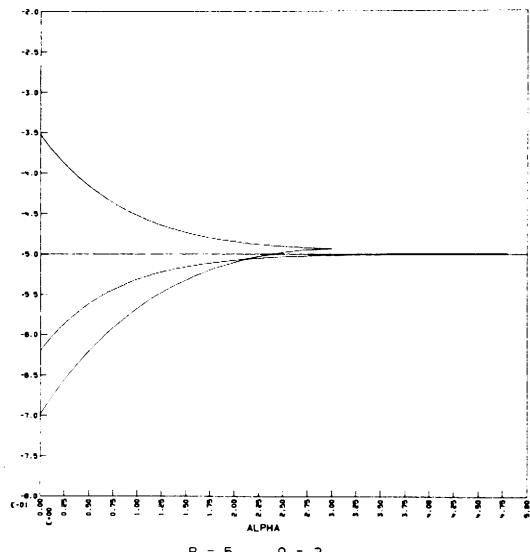


Figure 6

OVERSHOOT AHEAD

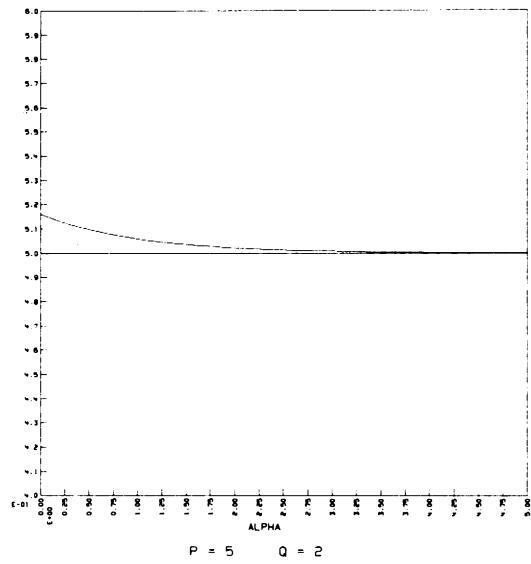


Figure 7

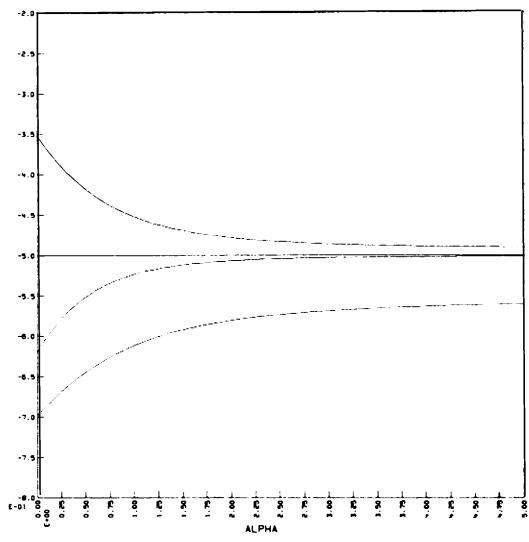


Figure 8

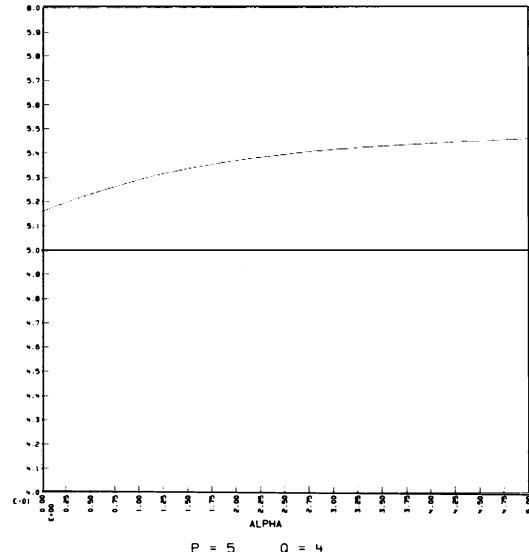


Figure 9

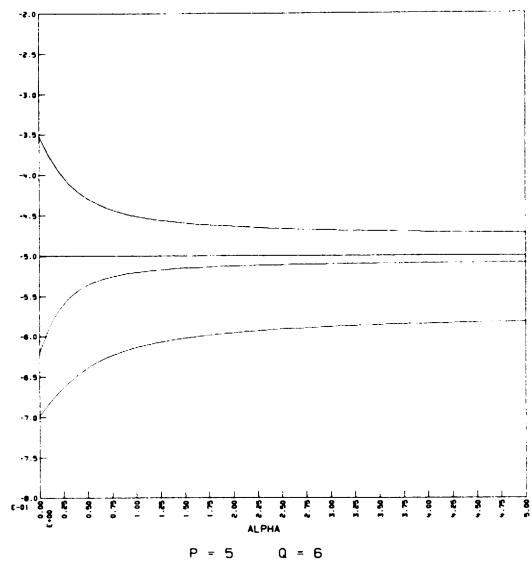


Figure 10

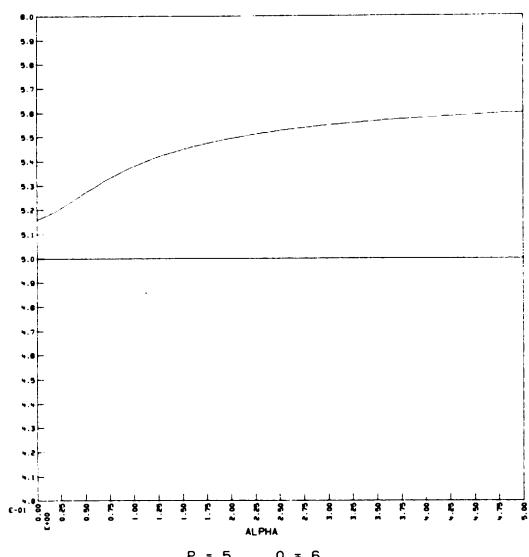
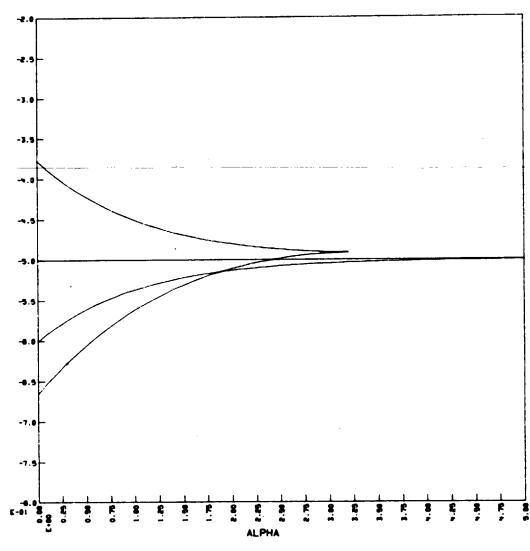


Figure 11



P = 7 Q = 2

Figure 12

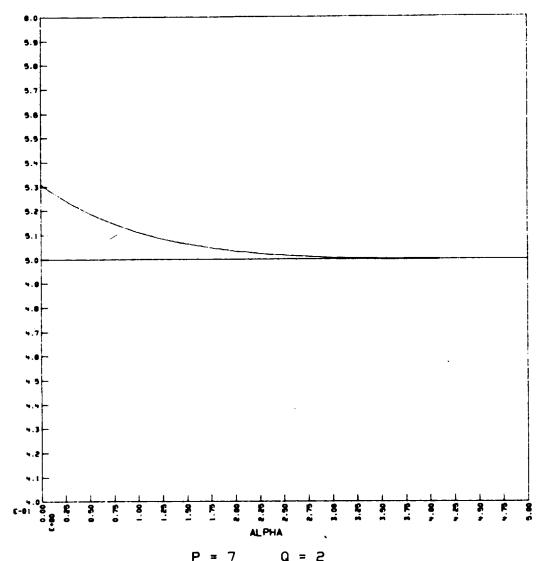


Figure 13

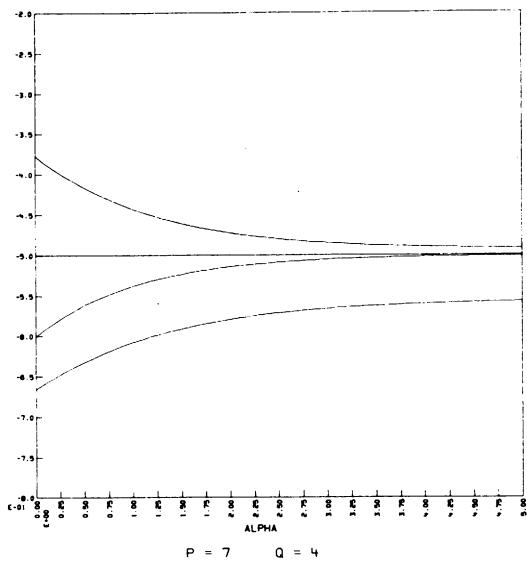
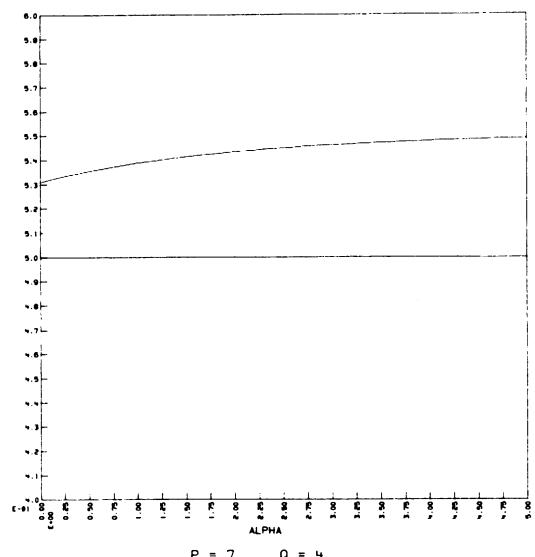


Figure 14



P = 7 Q = 4

Figure 15

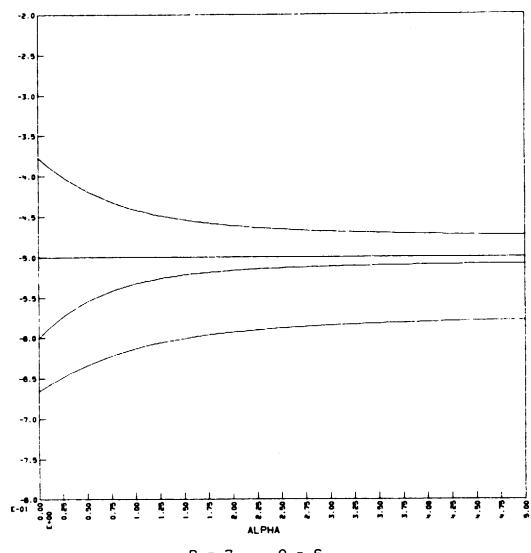


Figure 16

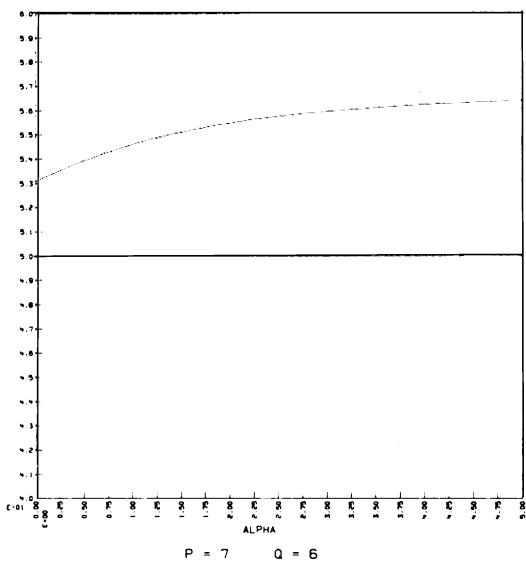


Figure 17

Table 1

EXTREMA FOR THE INTEGRAL OF THE GENERALIZED AIRY FUNCTION		P	S	Z	X	A111X	ALPHA
-1	1.111111111111111	A1111	A1111	A1111	A1111	A1111	A1111
-2	-3.180170000000000	-7779395000000000	-1.067939500000000	-0.3086600000000000	-5.520000000000000	-0.9750000000000000	-0.0000000000000000
-3	-2.941070000000000	-7118031000000000	-1.057950000000000	-0.3271950000000000	-5.309500000000000	-0.9650000000000000	-0.0000000000000000
-4	-2.702070000000000	-6562960000000000	-1.047950000000000	-0.3461700000000000	-5.098500000000000	-0.9550000000000000	-0.0000000000000000
-5	-2.463070000000000	-6017483000000000	-1.037950000000000	-0.3651450000000000	-4.887500000000000	-0.9450000000000000	-0.0000000000000000
-6	-2.224070000000000	-5500000000000000	-1.027950000000000	-0.3841200000000000	-4.676500000000000	-0.9350000000000000	-0.0000000000000000
-7	-1.985070000000000	-4992600000000000	-1.017950000000000	-0.4030950000000000	-4.465500000000000	-0.9250000000000000	-0.0000000000000000
-8	-1.746070000000000	-4585200000000000	-1.007950000000000	-0.4220700000000000	-4.254500000000000	-0.9150000000000000	-0.0000000000000000
-9	-1.507070000000000	-4177800000000000	-0.9979500000000000	-0.4410450000000000	-4.043500000000000	-0.9050000000000000	-0.0000000000000000
-10	-1.268070000000000	-3770400000000000	-0.9879500000000000	-0.4600200000000000	-3.832500000000000	-0.8950000000000000	-0.0000000000000000
-11	-1.029070000000000	-3363000000000000	-0.9779500000000000	-0.4789950000000000	-3.621500000000000	-0.8850000000000000	-0.0000000000000000
-12	-7.860070000000000	-2955600000000000	-0.9679500000000000	-0.4979700000000000	-3.410500000000000	-0.8750000000000000	-0.0000000000000000
-13	-5.501070000000000	-2548200000000000	-0.9579500000000000	-0.5169450000000000	-3.199500000000000	-0.8650000000000000	-0.0000000000000000
-14	-3.142070000000000	-2140800000000000	-0.9479500000000000	-0.5359200000000000	-2.988500000000000	-0.8550000000000000	-0.0000000000000000
-15	-0.783070000000000	-1733400000000000	-0.9379500000000000	-0.5548950000000000	-2.777500000000000	-0.8450000000000000	-0.0000000000000000
-16	-2.424070000000000	-1326000000000000	-0.9279500000000000	-0.5738700000000000	-2.566500000000000	-0.8350000000000000	-0.0000000000000000
-17	-4.065070000000000	-9186000000000000	-0.9179500000000000	-0.5928450000000000	-2.355500000000000	-0.8250000000000000	-0.0000000000000000
-18	-5.706070000000000	-5113200000000000	-0.9079500000000000	-0.6118200000000000	-2.144500000000000	-0.8150000000000000	-0.0000000000000000
-19	-7.347070000000000	-1024900000000000	-0.8979500000000000	-0.6307950000000000	-1.933500000000000	-0.8050000000000000	-0.0000000000000000
-20	-8.988070000000000	-6176500000000000	-0.8879500000000000	-0.6497700000000000	-1.722500000000000	-0.7950000000000000	-0.0000000000000000
-21	-10.62907000000000	-2110200000000000	-0.8779500000000000	-0.6687450000000000	-1.511500000000000	-0.7850000000000000	-0.0000000000000000
-22	-12.27007000000000	-8036800000000000	-0.8679500000000000	-0.6877200000000000	-1.299500000000000	-0.7750000000000000	-0.0000000000000000
-23	-13.91107000000000	-4970400000000000	-0.8579500000000000	-0.7066950000000000	-1.088500000000000	-0.7650000000000000	-0.0000000000000000
-24	-15.55207000000000	-1913000000000000	-0.8479500000000000	-0.7256700000000000	-8.777500000000000	-0.7550000000000000	-0.0000000000000000
-25	-17.19307000000000	-7046600000000000	-0.8379500000000000	-0.7446450000000000	-6.666500000000000	-0.7450000000000000	-0.0000000000000000
-26	-18.83407000000000	-3079200000000000	-0.8279500000000000	-0.7636200000000000	-4.555500000000000	-0.7350000000000000	-0.0000000000000000
-27	-20.47507000000000	-1191800000000000	-0.8179500000000000	-0.7825950000000000	-2.444500000000000	-0.7250000000000000	-0.0000000000000000
-28	-22.11607000000000	-4035400000000000	-0.8079500000000000	-0.8015700000000000	-0.3335000000000000	-0.7150000000000000	-0.0000000000000000
-29	-23.75707000000000	-1118000000000000	-0.7979500000000000	-0.8205450000000000	-0.1225000000000000	-0.7050000000000000	-0.0000000000000000
-30	-25.41107000000000	-3051600000000000	-0.7879500000000000	-0.8395200000000000	-0.0015000000000000	-0.6950000000000000	-0.0000000000000000
-31	-27.05207000000000	-9987200000000000	-0.7779500000000000	-0.8584950000000000	-0.1115000000000000	-0.6850000000000000	-0.0000000000000000
-32	-28.69307000000000	-2922300000000000	-0.7679500000000000	-0.8774700000000000	-0.2205000000000000	-0.6750000000000000	-0.0000000000000000
-33	-30.33407000000000	-8856900000000000	-0.7579500000000000	-0.8964450000000000	-0.3295000000000000	-0.6650000000000000	-0.0000000000000000
-34	-31.97507000000000	-2880500000000000	-0.7479500000000000	-0.9154200000000000	-0.4385000000000000	-0.6550000000000000	-0.0000000000000000
-35	-33.61607000000000	-8814100000000000	-0.7379500000000000	-0.9343950000000000	-0.5475000000000000	-0.6450000000000000	-0.0000000000000000
-36	-35.25707000000000	-2817000000000000	-0.7279500000000000	-0.9533700000000000	-0.6565000000000000	-0.6350000000000000	-0.0000000000000000
-37	-36.89807000000000	-8050600000000000	-0.7179500000000000	-0.9723450000000000	-0.7655000000000000	-0.6250000000000000	-0.0000000000000000
-38	-38.53907000000000	-2083200000000000	-0.7079500000000000	-0.9913200000000000	-0.8745000000000000	-0.6150000000000000	-0.0000000000000000
-39	-40.18007000000000	-6015800000000000	-0.6979500000000000	-0.0000000000000000	-0.9835000000000000	-0.6050000000000000	-0.0000000000000000
-40	-41.82107000000000	-1929400000000000	-0.6879500000000000	-0.0000000000000000	-1.092500000000000	-0.5950000000000000	-0.0000000000000000
-41	-43.46207000000000	-5962000000000000	-0.6779500000000000	-0.0000000000000000	-2.191500000000000	-0.5850000000000000	-0.0000000000000000
-42	-45.10307000000000	-1889600000000000	-0.6679500000000000	-0.0000000000000000	-4.180500000000000	-0.5750000000000000	-0.0000000000000000
-43	-46.74407000000000	-5823200000000000	-0.6579500000000000	-0.0000000000000000	-6.169500000000000	-0.5650000000000000	-0.0000000000000000
-44	-48.38507000000000	-1865800000000000	-0.6479500000000000	-0.0000000000000000	-8.158500000000000	-0.5550000000000000	-0.0000000000000000
-45	-50.02607000000000	-5859400000000000	-0.6379500000000000	-0.0000000000000000	-10.14750000000000	-0.5450000000000000	-0.0000000000000000
-46	-51.66707000000000	-1849000000000000	-0.6279500000000000	-0.0000000000000000	-12.13650000000000	-0.5350000000000000	-0.0000000000000000
-47	-53.30807000000000	-5852600000000000	-0.6179500000000000	-0.0000000000000000	-14.12550000000000	-0.5250000000000000	-0.0000000000000000
-48	-54.94907000000000	-1848200000000000	-0.6079500000000000	-0.0000000000000000	-16.11450000000000	-0.5150000000000000	-0.0000000000000000
-49	-56.59007000000000	-5856800000000000	-0.5979500000000000	-0.0000000000000000	-18.10350000000000	-0.5050000000000000	-0.0000000000000000
-50	-58.23107000000000	-1846400000000000	-0.5879500000000000	-0.0000000000000000	-19.99250000000000	-0.4950000000000000	-0.0000000000000000
-51	-59.87207000000000	-5857000000000000	-0.5779500000000000	-0.0000000000000000	-21.98150000000000	-0.4850000000000000	-0.0000000000000000
-52	-61.51307000000000	-1846000000000000	-0.5679500000000000	-0.0000000000000000	-23.97050000000000	-0.4750000000000000	-0.0000000000000000
-53	-63.15407000000000	-5857600000000000	-0.5579500000000000	-0.0000000000000000	-25.95950000000000	-0.4650000000000000	-0.0000000000000000
-54	-64.79507000000000	-1845600000000000	-0.5479500000000000	-0.0000000000000000	-27.94850000000000	-0.4550000000000000	-0.0000000000000000
-55	-66.43607000000000	-5858200000000000	-0.5379500000000000	-0.0000000000000000	-29.93750000000000	-0.4450000000000000	-0.0000000000000000
-56	-68.07707000000000	-1844600000000000	-0.5279500000000000	-0.0000000000000000	-31.92650000000000	-0.4350000000000000	-0.0000000000000000
-57	-69.71807000000000	-5859800000000000	-0.5179500000000000	-0.0000000000000000	-33.91550000000000	-0.4250000000000000	-0.0000000000000000
-58	-71.35907000000000	-1843600000000000	-0.5079500000000000	-0.0000000000000000	-35.90450000000000	-0.4150000000000000	-0.0000000000000000
-59	-72.99907000000000	-5860400000000000	-0.4979500000000000	-0.0000000000000000	-37.89350000000000	-0.4050000000000000	-0.0000000000000000
-60	-74.64007000000000	-1842200000000000	-0.4879500000000000	-0.0000000000000000	-39.88250000000000	-0.3950000000000000	-0.0000000000000000
-61	-76.28107000000000	-5861000000000000	-0.4779500000000000	-0.0000000000000000	-41.87150000000000	-0.3850000000000000	-0.0000000000000000
-62	-77.92207000000000	-1839400000000000	-0.4679500000000000	-0.0000000000000000	-43.86050000000000	-0.3750000000000000	-0.0000000000000000
-63	-79.56307000000000	-5861800000000000	-0.4579500000000000	-0.0000000000000000	-45.84950000000000	-0.3650000000000000	-0.0000000000000000
-64	-81.20407000000000	-1837200000000000	-0.4479500000000000	-0.0000000000000000	-47.83850000000000	-0.3550000000000000	-0.0000000000000000
-65	-82.84507000000000	-5862600000000000	-0.4379500000000000	-0.0000000000000000	-49.82750000000000	-0.3450000000000000	-0.0000000000000000
-66	-84.48607000000000	-1832600000000000	-0.4279500000000000	-0.0000000000000000	-51.81650000000000	-0.3350000000000000	-0.0000000000000000
-67	-86.12707000000000	-5863400000000000	-0.4179500000000000	-0.0000000000000000	-53.80550000000000	-0.3250000000000000	-0.0000000000000000
-68	-87.76807000000000	-1828400000000000	-0.4079500000000000	-0.0000000000000000	-55.79450000000000	-0.3150000000000000	-0.0000000000000000
-69	-89.40907000000000	-5864200000000000	-0.3979500000000000	-0.0000000000000000	-57.78350000000000	-0.3050000000000000	-0.0000000000000000
-70	-91.05007000000000	-1822200000000000	-0.3879500000000000	-0.0000000000000000	-59.77250000000000	-0.2950000000000000	-0.0000000000000000
-71	-92.69107000000000	-5865000000000000	-0.3779500000000000	-0.0000000000000000	-61.76150000000000	-0.2850000000000000	-0.0000000000000000
-72	-94.33207000000000	-1816600000000000	-0.3679500000000000	-0.0000000000000000	-63.75050000000000	-0.2750000000000000	-0.0000000000000000
-73	-9						

Table 2

Table 3

Table 4

Table 5

EXTREMA FOR THE INTEGRAL OF THE GENERALIZED AIRY FUNCTION $P = 3.0$									
ALPHA	S	A11(X)	K	A11(X)	S	A11(X)	K	A11(X)	ALPHA
0.00	-2.47426795	-0.4843073	-0.7030775	-0.4526763	-0.7030775	-0.4526763	-0.50491895	-0.50491895	-0.50491895
0.10	-2.48013773	-0.4859651	-0.7107485	-0.4546162	-0.7208150	-0.4546162	-0.50491895	-0.50491895	-0.50491895
0.20	-2.48601773	-0.4876651	-0.7184395	-0.4562671	-0.7308739	-0.4562671	-0.50491895	-0.50491895	-0.50491895
0.30	-2.49189295	-0.4893652	-0.7261305	-0.4579180	-0.7409328	-0.4579180	-0.50491895	-0.50491895	-0.50491895
0.40	-2.49772626	-0.4910652	-0.7338215	-0.4595689	-0.7509917	-0.4595689	-0.50491895	-0.50491895	-0.50491895
0.50	-2.51389089	-0.4927652	-0.7415125	-0.4612198	-0.7609506	-0.4612198	-0.50491895	-0.50491895	-0.50491895
0.60	-2.52985462	-0.4944652	-0.7492035	-0.4628617	-0.7709095	-0.4628617	-0.50491895	-0.50491895	-0.50491895
0.70	-2.54571825	-0.4961652	-0.7568945	-0.4645136	-0.7799684	-0.4645136	-0.50491895	-0.50491895	-0.50491895
0.80	-2.56158188	-0.4978652	-0.7645855	-0.4661655	-0.7889273	-0.4661655	-0.50491895	-0.50491895	-0.50491895
0.90	-2.59361331	-0.5016665	-0.7812765	-0.4773310	-0.8712629	-0.4773310	-0.50491895	-0.50491895	-0.50491895
1.00	-2.61787338	-0.5033665	-0.8019672	-0.4883975	-0.9611930	-0.4883975	-0.50491895	-0.50491895	-0.50491895
1.10	-2.64217291	-0.5050665	-0.8216582	-0.4990505	-0.9501930	-0.4990505	-0.50491895	-0.50491895	-0.50491895
1.20	-2.67647259	-0.5067665	-0.8413492	-0.5096135	-0.9391900	-0.5096135	-0.50491895	-0.50491895	-0.50491895
1.30	-2.71077227	-0.5084665	-0.8610402	-0.5192665	-0.9181890	-0.5192665	-0.50491895	-0.50491895	-0.50491895
1.40	-2.74507195	-0.5095665	-0.8807312	-0.5298195	-0.8971880	-0.5298195	-0.50491895	-0.50491895	-0.50491895
1.50	-2.77937163	-0.5112665	-0.9004222	-0.5394225	-0.8761870	-0.5394225	-0.50491895	-0.50491895	-0.50491895
1.60	-2.81367131	-0.5129665	-0.9191132	-0.5490255	-0.8550860	-0.5490255	-0.50491895	-0.50491895	-0.50491895
1.70	-2.84796999	-0.5146665	-0.9388042	-0.5586285	-0.8339849	-0.5586285	-0.50491895	-0.50491895	-0.50491895
1.80	-2.88226867	-0.5163665	-0.9574952	-0.5682315	-0.8130846	-0.5682315	-0.50491895	-0.50491895	-0.50491895
1.90	-2.91656735	-0.5180665	-0.9761862	-0.5778345	-0.7917759	-0.5778345	-0.50491895	-0.50491895	-0.50491895
2.00	-2.95086603	-0.5197665	-0.9948772	-0.5874375	-0.7698656	-0.5874375	-0.50491895	-0.50491895	-0.50491895
2.10	-2.98516471	-0.5214665	-0.1011682	-0.5970405	-0.7478543	-0.5970405	-0.50491895	-0.50491895	-0.50491895
2.20	-3.01946339	-0.5231665	-0.1198572	-0.6066435	-0.7268410	-0.6066435	-0.50491895	-0.50491895	-0.50491895
2.30	-3.05376207	-0.5248665	-0.1385462	-0.6162465	-0.7058317	-0.6162465	-0.50491895	-0.50491895	-0.50491895
2.40	-3.08776075	-0.5265665	-0.1572352	-0.6258495	-0.6848214	-0.6258495	-0.50491895	-0.50491895	-0.50491895
2.50	-3.12205943	-0.5282665	-0.1759242	-0.6354525	-0.6638062	-0.6354525	-0.50491895	-0.50491895	-0.50491895
2.60	-3.15635811	-0.5299665	-0.1946132	-0.6450555	-0.6427549	-0.6450555	-0.50491895	-0.50491895	-0.50491895
2.70	-3.19065679	-0.5316665	-0.2133022	-0.6546585	-0.6204537	-0.6546585	-0.50491895	-0.50491895	-0.50491895
2.80	-3.22495547	-0.5333665	-0.2320012	-0.6642615	-0.5982507	-0.6642615	-0.50491895	-0.50491895	-0.50491895
2.90	-3.25925415	-0.5350665	-0.2506902	-0.6738645	-0.5760477	-0.6738645	-0.50491895	-0.50491895	-0.50491895
3.00	-3.29355283	-0.5367665	-0.2693892	-0.6834675	-0.5538447	-0.6834675	-0.50491895	-0.50491895	-0.50491895
3.10	-3.32785151	-0.5384665	-0.2880782	-0.6930705	-0.5310417	-0.6930705	-0.50491895	-0.50491895	-0.50491895
3.20	-3.36214919	-0.5401665	-0.3067672	-0.7026735	-0.5082385	-0.7026735	-0.50491895	-0.50491895	-0.50491895
3.30	-3.39644787	-0.5418665	-0.3254562	-0.7122765	-0.4854355	-0.7122765	-0.50491895	-0.50491895	-0.50491895
3.40	-3.43073655	-0.5435665	-0.3441452	-0.7218795	-0.4626325	-0.7218795	-0.50491895	-0.50491895	-0.50491895
3.50	-3.46502523	-0.5452665	-0.3628342	-0.7314825	-0.4398295	-0.7314825	-0.50491895	-0.50491895	-0.50491895
3.60	-3.49931391	-0.5469665	-0.3815232	-0.7410855	-0.4170265	-0.7410855	-0.50491895	-0.50491895	-0.50491895
3.70	-3.53360259	-0.5486665	-0.3992122	-0.7506885	-0.3942235	-0.7506885	-0.50491895	-0.50491895	-0.50491895
3.80	-3.56789127	-0.5503665	-0.4179012	-0.7602915	-0.3714205	-0.7602915	-0.50491895	-0.50491895	-0.50491895
3.90	-3.60217995	-0.5520665	-0.4365902	-0.7698945	-0.3486175	-0.7698945	-0.50491895	-0.50491895	-0.50491895
4.00	-3.63646863	-0.5537665	-0.4552792	-0.7794975	-0.3258145	-0.7794975	-0.50491895	-0.50491895	-0.50491895
4.10	-3.67075731	-0.5554665	-0.4739682	-0.7891005	-0.3030115	-0.7891005	-0.50491895	-0.50491895	-0.50491895
4.20	-3.70504609	-0.5571665	-0.4926472	-0.7987035	-0.2802085	-0.7987035	-0.50491895	-0.50491895	-0.50491895
4.30	-3.73933477	-0.5588665	-0.5113362	-0.8083065	-0.2574055	-0.8083065	-0.50491895	-0.50491895	-0.50491895
4.40	-3.77362345	-0.5605665	-0.5300252	-0.8179095	-0.2346025	-0.8179095	-0.50491895	-0.50491895	-0.50491895
4.50	-3.80791213	-0.5622665	-0.5487142	-0.8275125	-0.2118095	-0.8275125	-0.50491895	-0.50491895	-0.50491895
4.60	-3.84219981	-0.5639665	-0.5673932	-0.8371155	-0.1890065	-0.8371155	-0.50491895	-0.50491895	-0.50491895
4.70	-3.87648849	-0.5656665	-0.5860822	-0.8467185	-0.1662035	-0.8467185	-0.50491895	-0.50491895	-0.50491895
4.80	-3.91077717	-0.5673665	-0.6047712	-0.8563215	-0.1434005	-0.8563215	-0.50491895	-0.50491895	-0.50491895
4.90	-3.94506585	-0.5690665	-0.6234592	-0.8659245	-0.1206075	-0.8659245	-0.50491895	-0.50491895	-0.50491895
5.00	-3.97935453	-0.5707665	-0.6421482	-0.8755275	-0.0978045	-0.8755275	-0.50491895	-0.50491895	-0.50491895

Table 6

Table 7

ESTIMATION FOR THE INTEGRAL OF THE GENERALIZED AIRY FUNCTION $P_{\nu}(T) = \int_0^{\infty} e^{-x^{\nu}} T^x dx$									
ALPHA	A111X1	X	A111X2	T	A111X3	L	A111X4	Z	A111X5
-1.10	-0.2132662	0	0.9544110	-0.2379949	2.7799399	-0.8606691	2.8954029	0.5109675	0.10
-1.05	-0.2162661	0	0.9544110	-0.2381795	2.8624970	-0.5612217	2.7316949	0.5209565	0.10
-1.00	-0.2192739	0	0.9544110	-0.2383645	2.8799399	-0.2612217	2.7516949	0.5309555	0.10
-0.95	-0.2222817	0	0.9544110	-0.2385500	2.8999399	-0.0612217	2.7716949	0.5409545	0.10
-0.90	-0.2252895	0	0.9544110	-0.2387352	2.9099399	0.1612217	2.7916949	0.5509535	0.10
-0.85	-0.2282973	0	0.9544110	-0.2389204	2.9299399	0.3612217	2.8116949	0.5609525	0.10
-0.80	-0.2312951	0	0.9544110	-0.2391056	2.9499399	0.5612217	2.8316949	0.5709515	0.10
-0.75	-0.2342929	0	0.9544110	-0.2392908	2.9699399	0.7612217	2.8516949	0.5809505	0.10
-0.70	-0.2372907	0	0.9544110	-0.2394760	2.9899399	0.9612217	2.8716949	0.5909495	0.10
-0.65	-0.2402885	0	0.9544110	-0.2396612	3.0099399	1.1612217	2.8916949	0.6009485	0.10
-0.60	-0.2432863	0	0.9544110	-0.2398464	3.0299399	1.3612217	2.9116949	0.6109475	0.10
-0.55	-0.2462841	0	0.9544110	-0.2400316	3.0499399	1.5612217	2.9316949	0.6209465	0.10
-0.50	-0.2492819	0	0.9544110	-0.2402168	3.0699399	1.7612217	2.9516949	0.6309455	0.10
-0.45	-0.2522797	0	0.9544110	-0.2404020	3.0899399	1.9612217	2.9716949	0.6409445	0.10
-0.40	-0.2552775	0	0.9544110	-0.2405872	3.1099399	2.1612217	2.9916949	0.6509435	0.10
-0.35	-0.2582753	0	0.9544110	-0.2407724	3.1299399	2.3612217	3.0116949	0.6609425	0.10
-0.30	-0.2612731	0	0.9544110	-0.2409576	3.1499399	2.5612217	3.0316949	0.6709415	0.10
-0.25	-0.2642709	0	0.9544110	-0.2411428	3.1699399	2.7612217	3.0516949	0.6809405	0.10
-0.20	-0.2672687	0	0.9544110	-0.2413280	3.1899399	2.9612217	3.0716949	0.6909395	0.10
-0.15	-0.2702665	0	0.9544110	-0.2415132	3.2099399	3.1612217	3.0916949	0.7009385	0.10
-0.10	-0.2732643	0	0.9544110	-0.2416984	3.2299399	3.3612217	3.1116949	0.7109375	0.10
-0.05	-0.2762621	0	0.9544110	-0.2418836	3.2499399	3.5612217	3.1316949	0.7209365	0.10
0.00	-0.2792600	0	0.9544110	-0.2420688	3.2699399	3.7612217	3.1516949	0.7309355	0.10
0.05	-0.2822578	0	0.9544110	-0.2422540	3.2899399	3.9612217	3.1716949	0.7409345	0.10
0.10	-0.2852556	0	0.9544110	-0.2424392	3.3099399	4.1612217	3.1916949	0.7509335	0.10
0.15	-0.2882534	0	0.9544110	-0.2426244	3.3299399	4.3612217	3.2116949	0.7609325	0.10
0.20	-0.2912512	0	0.9544110	-0.2428096	3.3499399	4.5612217	3.2316949	0.7709315	0.10
0.25	-0.2942490	0	0.9544110	-0.2430948	3.3699399	4.7612217	3.2516949	0.7809305	0.10
0.30	-0.2972468	0	0.9544110	-0.2432800	3.3899399	4.9612217	3.2716949	0.7909295	0.10
0.35	-0.3002446	0	0.9544110	-0.2434652	3.4099399	5.1612217	3.2916949	0.8009285	0.10
0.40	-0.3032424	0	0.9544110	-0.2436504	3.4299399	5.3612217	3.3116949	0.8109275	0.10
0.45	-0.3062402	0	0.9544110	-0.2438356	3.4499399	5.5612217	3.3316949	0.8209265	0.10
0.50	-0.3092380	0	0.9544110	-0.2440208	3.4699399	5.7612217	3.3516949	0.8309255	0.10
0.55	-0.3122358	0	0.9544110	-0.2442060	3.4899399	5.9612217	3.3716949	0.8409245	0.10
0.60	-0.3152336	0	0.9544110	-0.2443912	3.5099399	6.1612217	3.3916949	0.8509235	0.10
0.65	-0.3182314	0	0.9544110	-0.2445764	3.5299399	6.3612217	3.4116949	0.8609225	0.10
0.70	-0.3212292	0	0.9544110	-0.2447616	3.5499399	6.5612217	3.4316949	0.8709215	0.10
0.75	-0.3242270	0	0.9544110	-0.2449468	3.5699399	6.7612217	3.4516949	0.8809205	0.10
0.80	-0.3272248	0	0.9544110	-0.2451320	3.5899399	6.9612217	3.4716949	0.8909195	0.10
0.85	-0.3302226	0	0.9544110	-0.2453172	3.6099399	7.1612217	3.4916949	0.9009185	0.10
0.90	-0.3332204	0	0.9544110	-0.2455024	3.6299399	7.3612217	3.5116949	0.9109175	0.10
0.95	-0.3362182	0	0.9544110	-0.2456876	3.6499399	7.5612217	3.5316949	0.9209165	0.10
1.00	-0.3392160	0	0.9544110	-0.2458728	3.6699399	7.7612217	3.5516949	0.9309155	0.10
1.05	-0.3422138	0	0.9544110	-0.2460580	3.6899399	7.9612217	3.5716949	0.9409145	0.10
1.10	-0.3452116	0	0.9544110	-0.2462432	3.7099399	8.1612217	3.5916949	0.9509135	0.10
1.15	-0.3482094	0	0.9544110	-0.2464284	3.7299399	8.3612217	3.6116949	0.9609125	0.10
1.20	-0.3512072	0	0.9544110	-0.2466136	3.7499399	8.5612217	3.6316949	0.9709115	0.10
1.25	-0.3542050	0	0.9544110	-0.2467988	3.7699399	8.7612217	3.6516949	0.9809105	0.10
1.30	-0.3572028	0	0.9544110	-0.2469840	3.7899399	8.9612217	3.6716949	0.9909095	0.10
1.35	-0.3602006	0	0.9544110	-0.2471692	3.8099399	9.1612217	3.6916949	1.0009085	0.10
1.40	-0.3632084	0	0.9544110	-0.2473544	3.8299399	9.3612217	3.7116949	1.0109075	0.10
1.45	-0.3662062	0	0.9544110	-0.2475396	3.8499399	9.5612217	3.7316949	1.0209065	0.10
1.50	-0.3692040	0	0.9544110	-0.2477248	3.8699399	9.7612217	3.7516949	1.0309055	0.10
1.55	-0.3722018	0	0.9544110	-0.2479100	3.8899399	9.9612217	3.7716949	1.0409045	0.10
1.60	-0.3752000	0	0.9544110	-0.2480952	3.9099399	1.01612217	3.7916949	1.0509035	0.10
1.65	-0.3781978	0	0.9544110	-0.2482804	3.9299399	1.06612217	3.8116949	1.0609025	0.10
1.70	-0.3811956	0	0.9544110	-0.2484656	3.9499399	1.11612217	3.8316949	1.0709015	0.10
1.75	-0.3841934	0	0.9544110	-0.2486508	3.9699399	1.16612217	3.8516949	1.0809005	0.10
1.80	-0.3871912	0	0.9544110	-0.2488360	3.9899399	1.21612217	3.8716949	1.0909005	0.10
1.85	-0.3901890	0	0.9544110	-0.2490212	4.0099399	1.26612217	3.8916949	1.1009005	0.10
1.90	-0.3931868	0	0.9544110	-0.2492064	4.0299399	1.31612217	3.9116949	1.1109005	0.10
1.95	-0.3961846	0	0.9544110	-0.2493916	4.0499399	1.36612217	3.9316949	1.1209005	0.10
2.00	-0.4001824	0	0.9544110	-0.2495768	4.0699399	1.41612217	3.9516949	1.1309005	0.10
2.05	-0.4031802	0	0.9544110	-0.2497620	4.0899399	1.46612217	3.9716949	1.1409005	0.10
2.10	-0.4061780	0	0.9544110	-0.2499472	4.1099399	1.51612217	3.9916949	1.1509005	0.10
2.15	-0.4091758	0	0.9544110	-0.2501324	4.1299399	1.56612217	4.0116949	1.1609005	0.10
2.20	-0.4121736	0	0.9544110	-0.2503176	4.1499399	1.61612217	4.0316949	1.1709005	0.10
2.25	-0.4151714	0	0.9544110	-0.2505028	4.1699399	1.66612217	4.0516949	1.1809005	0.10
2.30	-0.4181692	0	0.9544110	-0.2506880	4.1899399	1.71612217	4.0716949	1.1909005	0.10
2.35	-0.4211670	0	0.9544110	-0.2508732	4.2099399	1.76612217	4.0916949	1.2009005	0.10
2.40	-0.4241648	0	0.9544110	-0.2510584	4.2299399	1.81612217	4.1116949	1.2109005	0.10
2.45	-0.4271626	0	0.9544110	-0.2512436	4.2499399	1.86612217	4.1316949	1.2209005	0.10
2.50	-0.4301604	0	0.9544110	-0.2514288	4.2699399	1.91612217	4.1516949	1.2309005	0.10
2.55	-0.4331582	0	0.9544110	-0.2516140	4.2899399	1.96612217	4.1716949	1.2409005	0.10
2.60	-0.4361560	0	0.9544110	-0.2517992	4.3099399	2.01612217	4.1916949	1.2509005	0.10
2.65	-0.4391538	0	0.9544110	-0.2519844	4.3299399	2.06612217	4.2116949	1.2609005	0.10
2.70	-0.4421516	0	0.9544110	-0.2521696	4.3499399	2.11612217	4.2316949	1.2709005	0.10
2.75	-0.4451494	0	0.9544110	-0.2523548	4.3699399	2.16612217	4.2516949	1.2809005	0.10
2.80	-0.4481472	0	0.9544110	-0.2525400	4.3899399	2.21612217	4.2716949	1.2909005	0.10
2.85	-0.4511450	0	0.9544110	-0.2527252	4.4099399	2.26612217	4.2916949	1.3009005	0.10
2.90	-0.4541428	0	0.9544110	-0.2529104	4.4299399	2.31612217	4.3116949	1.3109005	0.10
2.95	-0.4571406	0	0.9544110	-0.2530956	4.4499399	2.36612217	4.3316949	1.3209005	0.10
3.00	-0.4601384	0	0.9544110	-0.2532808	4.4699399	2.41612217	4.3516949	1.3309005	0.10
3.05	-0.4631362	0	0.9544110	-0.2534660	4.4899399	2.46612217	4.3716949	1.3409005	0.10
3.10	-0.4661340	0	0.9544110	-0.2536512	4.5099399	2.51612217	4.3916949	1.3509005	0.10
3.15	-0.4691318	0	0.9544110	-0.2538364	4.5299399	2.56612217	4.4116949	1.3609005	0.10
3.20	-0.4721296	0	0.9544110	-0.2540216	4.5499399	2.61612217	4.4316949	1.3709005	0.10
3.25	-0.4751274	0	0.9544110	-0.2542068	4.5699399	2.66612217	4.4516949	1.3809005	0.10
3.30	-0.4781252	0	0.9544110	-0.2543920	4.5899399	2.71612217	4.4716949	1.3909005	0.10
3.35	-0.4811230	0	0.9544110	-0.2545772	4.6099399	2.76612217	4.4916949	1.4009005	0.10
3.40	-0.4841208	0	0.9544110	-0.2547624	4.6299399	2.81612217	4.5116949	1.4109005	0.10
3.45	-0.4871186	0	0.9544110	-0.2549476	4.64993				

Table 8

INTEGRAL OF THE AIRY FUNCTION

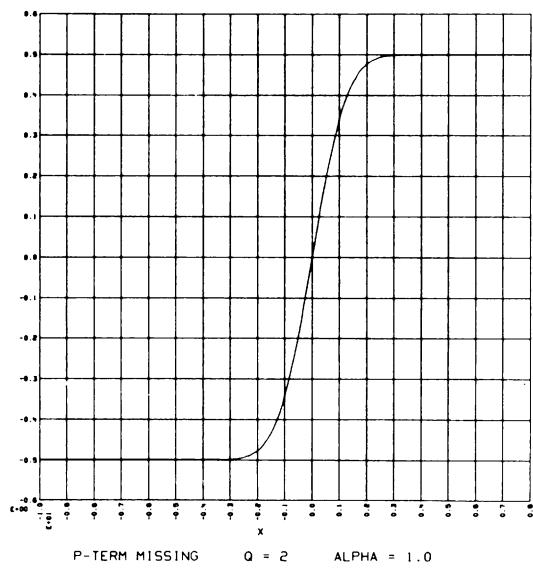


Figure 18

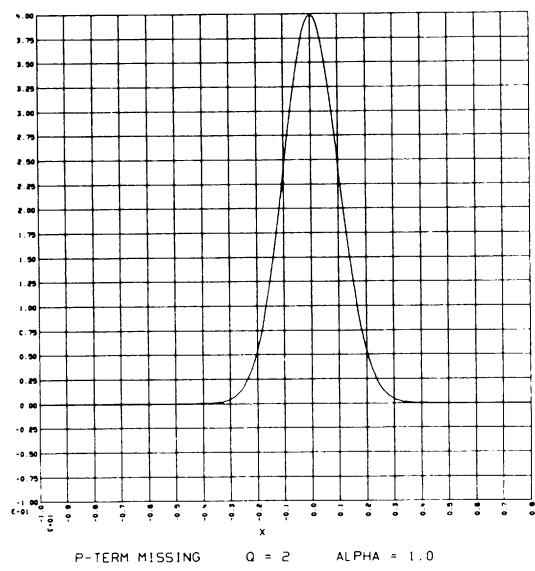


Figure 19

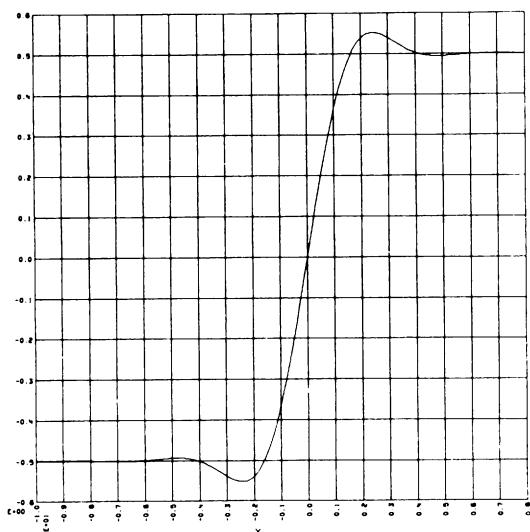


Figure 20

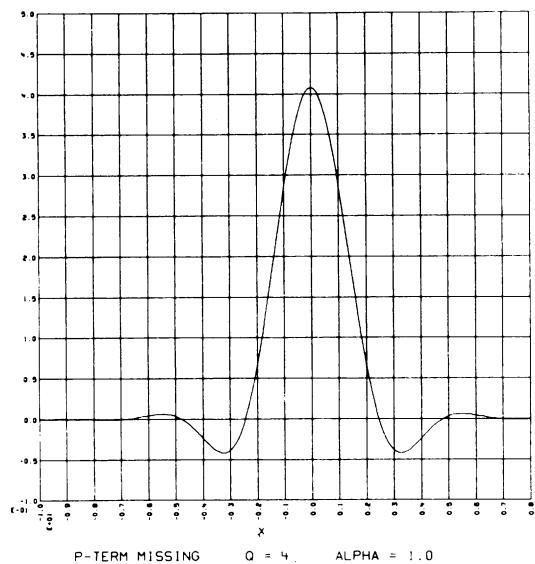


Figure 21

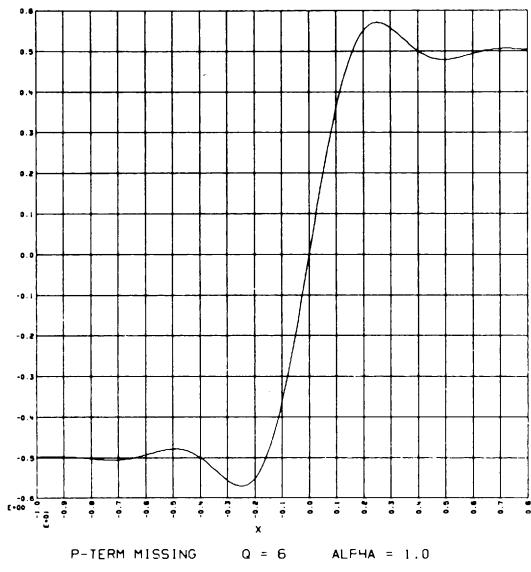


Figure 22

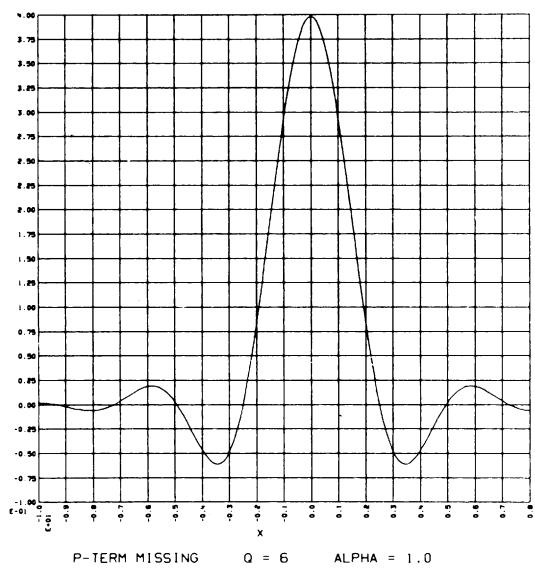


Figure 23

Table 10

P-TERM MISSING Q = ? ALPHA = 1.0

Table 11

P-TERM MISSING Q = 4 ALPHA = 1.0

Table 12

P TERM MISSING Q = 8 ALPHA = 1.0

X	A[11]	A[12]	A[13]	A[14]	A[15]	X	A[16]	A[17]	A[18]	X	A[19]	A[20]	X	A[21]	A[22]	A[23]
-1	-0.9875	-0.02367	5	0.40444	0.05664	-1	-0.31816	0.29379	3	0.46218	0.02367	0	-0.9875	-0.02367	0.00000	0.00000
-1	-0.9875	-0.02367	5	0.40444	0.05664	-1	-0.31816	0.29379	3	0.46218	0.02367	0	-0.9875	-0.02367	0.00000	0.00000
-1	-0.9869	0.00919	0	0.0125	5	-0.81503	-0.01293	-6	-0.29952	0.12912	3	0.51611	0.05763	-0.00000	0.00000	0.00000
-1	-0.9869	0.00919	0	0.0125	5	-0.81503	-0.01293	-6	-0.29952	0.12912	3	0.51611	0.05763	-0.00000	0.00000	0.00000
-1	-0.9861	0.00938	0	0.00988	5	-0.79512	-0.00662	-6	-0.21307	0.13055	3	0.53026	0.03990	-0.00000	0.00000	0.00000
-1	-0.9861	0.00938	0	0.00988	5	-0.79512	-0.00662	-6	-0.21307	0.13055	3	0.53026	0.03990	-0.00000	0.00000	0.00000
-1	-0.9852	0.00920	0	0.00507	5	-0.79112	-0.02771	-6	-0.19155	0.30505	3	0.56096	0.00000	-0.00000	0.00000	0.00000
-1	-0.9852	0.00920	0	0.00507	5	-0.79112	-0.02771	-6	-0.19155	0.30505	3	0.56096	0.00000	-0.00000	0.00000	0.00000
-1	-0.9842	0.00862	0	0.00503	5	-0.79703	-0.05619	-6	-0.11893	0.38779	4	0.59217	0.03711	-0.00000	0.00000	0.00000
-1	-0.9842	0.00862	0	0.00503	5	-0.79703	-0.05619	-6	-0.11893	0.38779	4	0.59217	0.03711	-0.00000	0.00000	0.00000
-1	-0.9832	0.00795	0	0.00165	5	-0.81626	-0.10460	-6	-0.19769	0.39919	4	0.62502	0.00000	-0.00000	0.00000	0.00000
-1	-0.9832	0.00795	0	0.00165	5	-0.81626	-0.10460	-6	-0.19769	0.39919	4	0.62502	0.00000	-0.00000	0.00000	0.00000
-1	-0.9820	0.00747	0	0.00220	5	-0.84911	-0.21817	-7	-0.20189	0.38689	4	0.65796	0.00000	-0.00000	0.00000	0.00000
-1	-0.9820	0.00747	0	0.00220	5	-0.84911	-0.21817	-7	-0.20189	0.38689	4	0.65796	0.00000	-0.00000	0.00000	0.00000
-1	-0.9810	0.00690	0	0.00327	5	-0.87877	-0.31389	-8	-0.79137	0.39740	4	0.69182	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9810	0.00690	0	0.00327	5	-0.87877	-0.31389	-8	-0.79137	0.39740	4	0.69182	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9809	0.00693	0	0.00361	5	-0.87813	-0.21713	-8	-0.15673	0.37965	4	0.73094	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9809	0.00693	0	0.00361	5	-0.87813	-0.21713	-8	-0.15673	0.37965	4	0.73094	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9802	0.00652	0	0.00529	5	-0.86950	-0.06950	-9	-0.19353	0.36996	5	0.76249	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9802	0.00652	0	0.00529	5	-0.86950	-0.06950	-9	-0.19353	0.36996	5	0.76249	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9799	0.00611	0	0.00511	5	-0.81789	-0.07013	-9	-0.26948	0.34244	5	0.79839	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9799	0.00611	0	0.00511	5	-0.81789	-0.07013	-9	-0.26948	0.34244	5	0.79839	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9797	0.00607	0	0.00687	5	-0.52293	-0.99883	-10	-0.31192	0.31708	5	0.82671	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9797	0.00607	0	0.00687	5	-0.52293	-0.99883	-10	-0.31192	0.31708	5	0.82671	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9794	0.00617	0	0.00540	5	-0.54943	-0.99896	-10	-1.21686	0.29362	5	0.85999	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9794	0.00617	0	0.00540	5	-0.54943	-0.99896	-10	-1.21686	0.29362	5	0.85999	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9793	0.00616	0	0.00541	5	-0.55195	-0.99323	-10	-0.63611	0.21193	5	0.88189	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9793	0.00616	0	0.00541	5	-0.55195	-0.99323	-10	-0.63611	0.21193	5	0.88189	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9785	0.00578	0	0.00358	5	-0.59961	-0.04338	-11	-0.18313	0.18529	6	0.91204	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9785	0.00578	0	0.00358	5	-0.59961	-0.04338	-11	-0.18313	0.18529	6	0.91204	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9784	0.00516	0	0.00482	5	-0.56249	-0.35309	-11	-0.15649	0.14782	6	0.94710	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9784	0.00516	0	0.00482	5	-0.56249	-0.35309	-11	-0.15649	0.14782	6	0.94710	-0.02189	-0.00000	0.00000	0.00000
-1	-0.9781	0.00507	0	0.00207	5	-0.56993	-0.20161	-11	-0.51931	0.13074	6	0.98287	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9781	0.00507	0	0.00207	5	-0.56993	-0.20161	-11	-0.51931	0.13074	6	0.98287	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9775	0.00565	0	0.00180	5	-0.57599	-0.01019	-12	-0.25012	0.08775	5	0.81224	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9775	0.00565	0	0.00180	5	-0.57599	-0.01019	-12	-0.25012	0.08775	5	0.81224	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9774	0.00568	0	0.00180	5	-0.56769	-0.02616	-12	-0.25863	0.06642	6	0.83793	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9774	0.00568	0	0.00180	5	-0.56769	-0.02616	-12	-0.25863	0.06642	6	0.83793	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9773	0.00568	0	0.00178	5	-0.55932	-0.06767	-12	-0.56986	0.01210	5	0.85331	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9773	0.00568	0	0.00178	5	-0.55932	-0.06767	-12	-0.56986	0.01210	5	0.85331	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9772	0.00568	0	0.00179	5	-0.50912	-0.08777	-12	-0.51995	0.00105	5	0.88263	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9772	0.00568	0	0.00179	5	-0.50912	-0.08777	-12	-0.51995	0.00105	5	0.88263	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9769	0.00562	0	0.00133	5	-0.51013	-0.16782	-13	-0.56165	-0.02386	7	0.90810	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9769	0.00562	0	0.00133	5	-0.51013	-0.16782	-13	-0.56165	-0.02386	7	0.90810	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9768	0.00529	0	0.00184	5	-0.50995	-0.16799	-13	-0.56174	-0.01676	7	0.95799	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9768	0.00529	0	0.00184	5	-0.50995	-0.16799	-13	-0.56174	-0.01676	7	0.95799	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9767	0.00565	0	0.00165	5	-0.48132	-0.19699	-13	-0.55681	-0.04913	7	0.98716	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9767	0.00565	0	0.00165	5	-0.48132	-0.19699	-13	-0.55681	-0.04913	7	0.98716	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9765	0.00563	0	0.00191	5	-0.44267	-0.24267	-13	-0.54530	-0.05729	7	0.99049	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9765	0.00563	0	0.00191	5	-0.44267	-0.24267	-13	-0.54530	-0.05729	7	0.99049	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9764	0.00567	0	0.00187	1	-0.39020	-0.27437	-13	-0.53507	-0.08639	7	0.95384	-0.00000	-0.00000	0.00000	0.00000
-1	-0.9764	0.00567	0	0.00187	1	-0.39020	-0.27437	-13	-0.53507	-0.08639	7	0.95384	-0.00000	-0.00000	0.00000	0.00000

INTEGRAL OF THE AIRY FUNCTION

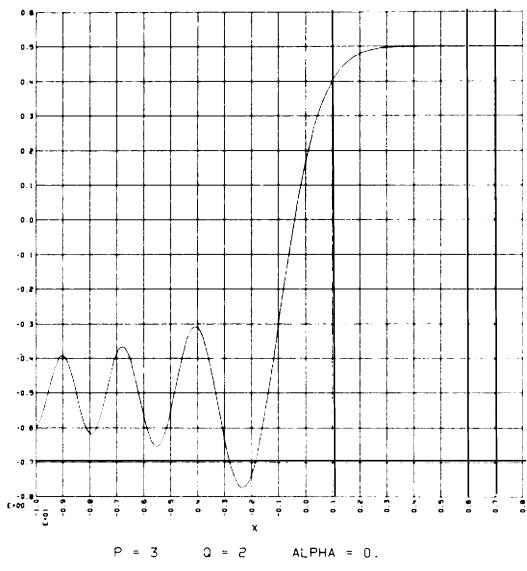


Figure 24

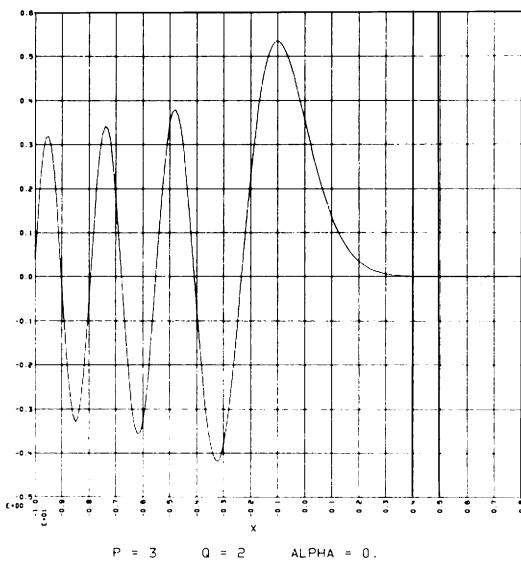


Figure 25

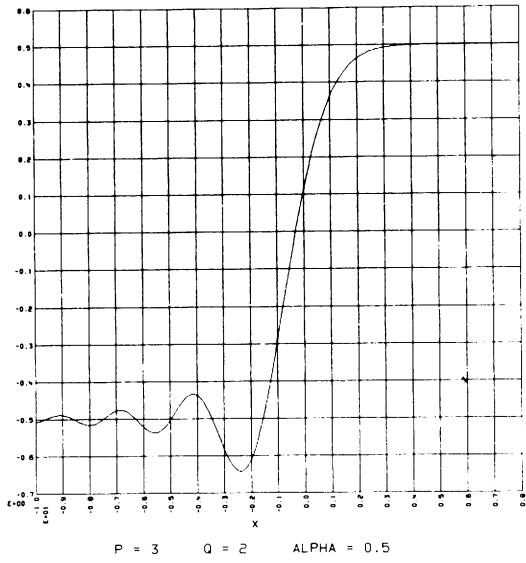


Figure 26

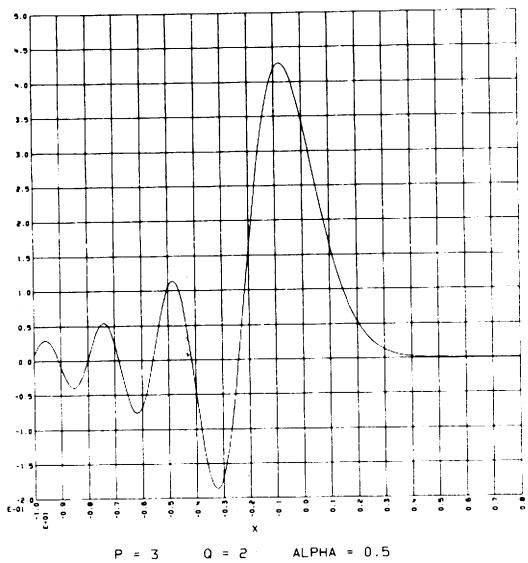


Figure 27

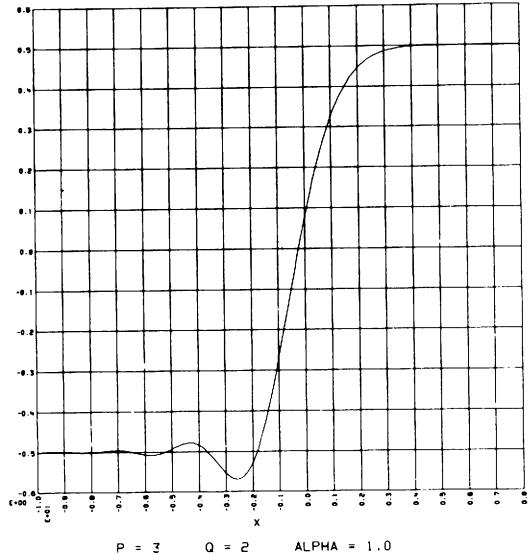


Figure 28

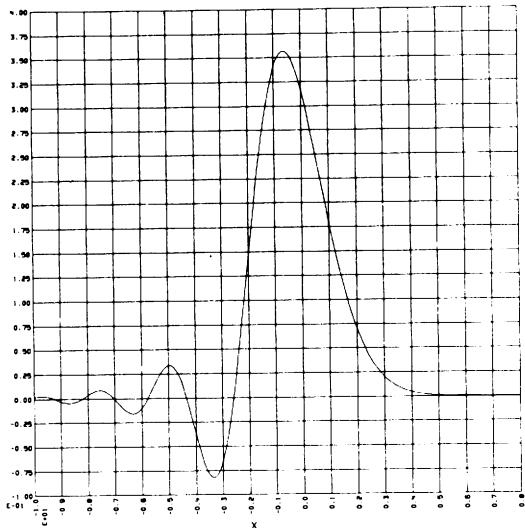


Figure 29

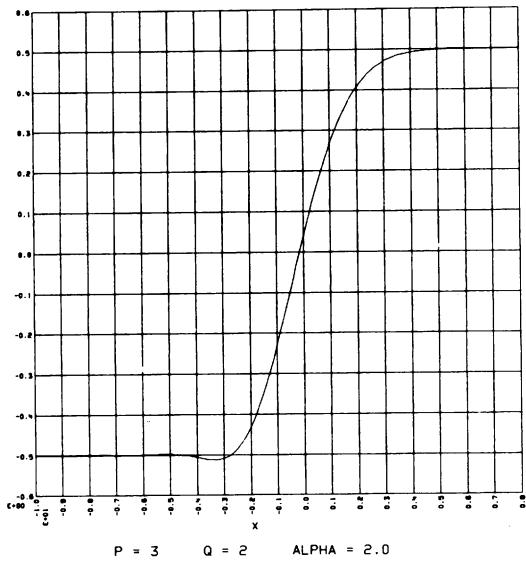


Figure 30

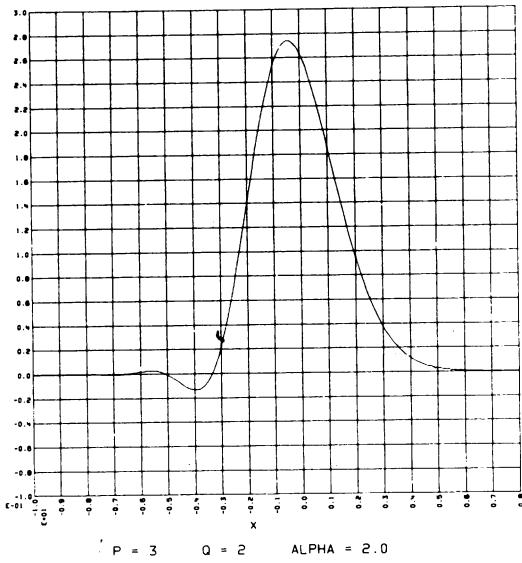


Figure 31

INTEGRAL OF THE AIRY FUNCTION

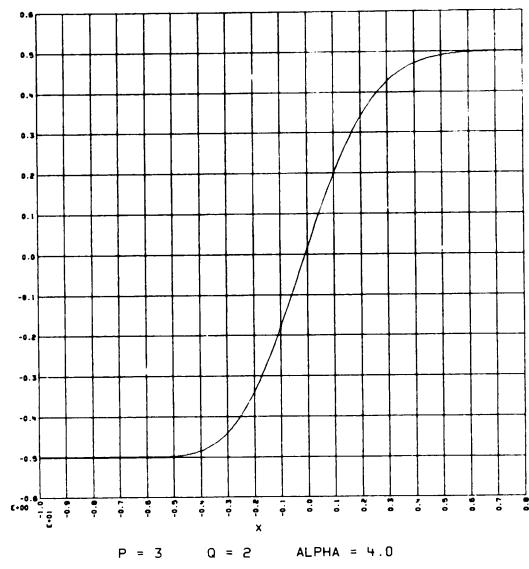


Figure 32

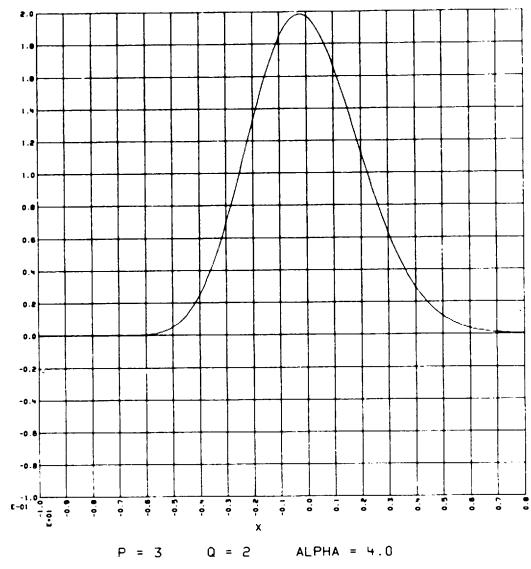


Figure 33

Table 13

P = 3 Q = ? ALPHA = 0.

Table 14

P = 3 Q = 2 ALPHA = 0.5

Table 15

Table 16

$\rho = 3$ $Q = 2$ $\text{ALPHA} = 2.1$

Table 17

P = 3 Q = 2 ALPHA = 4.0

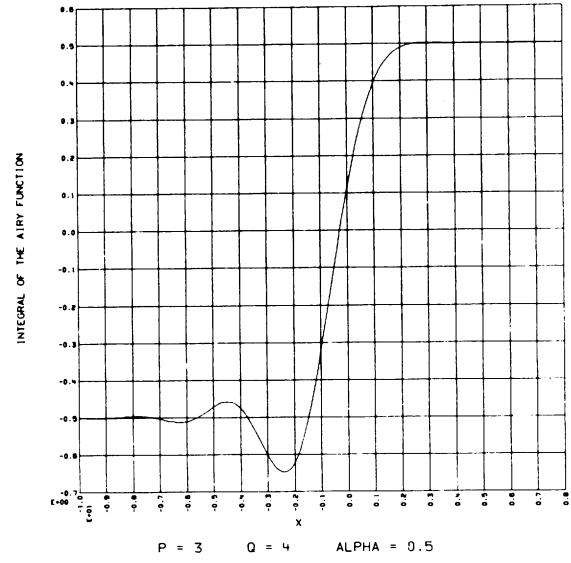
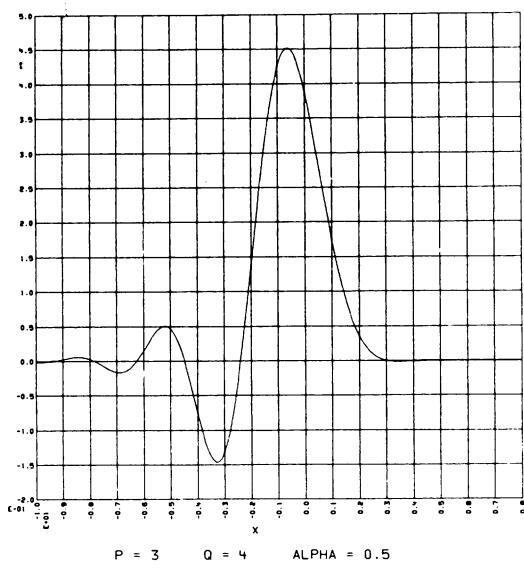


Figure 34



$P = 3 \quad Q = 4 \quad \text{ALPHA} = 0.5$

Figure 35

INTEGRAL OF THE AIRY FUNCTION

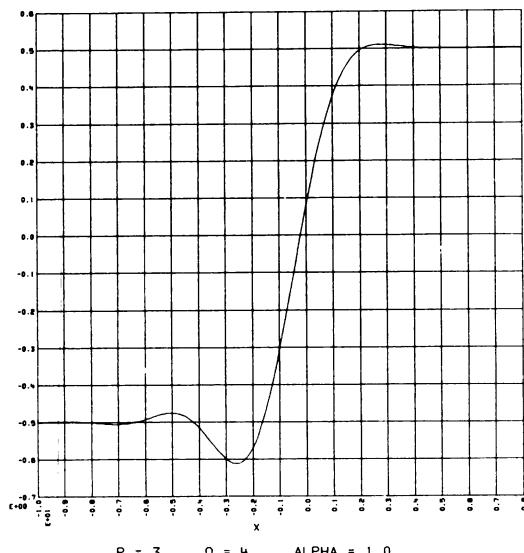


Figure 36

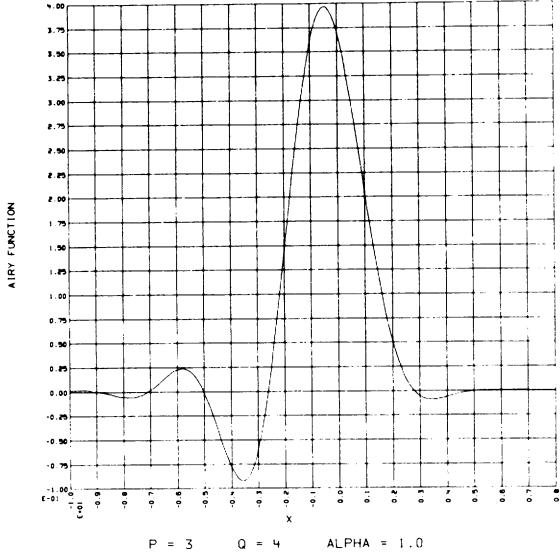


Figure 37

INTEGRAL OF THE AIRY FUNCTION

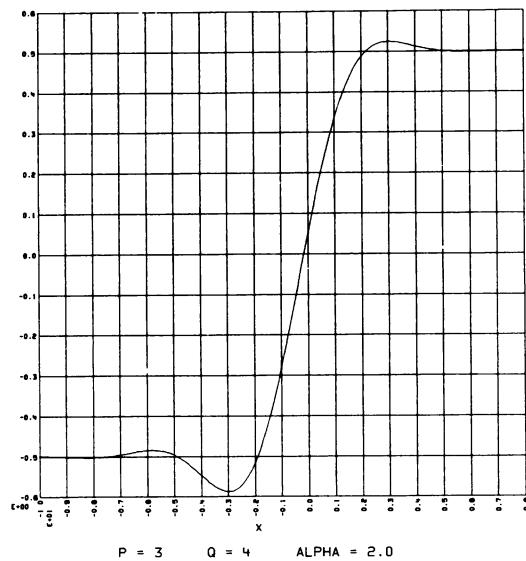


Figure 38

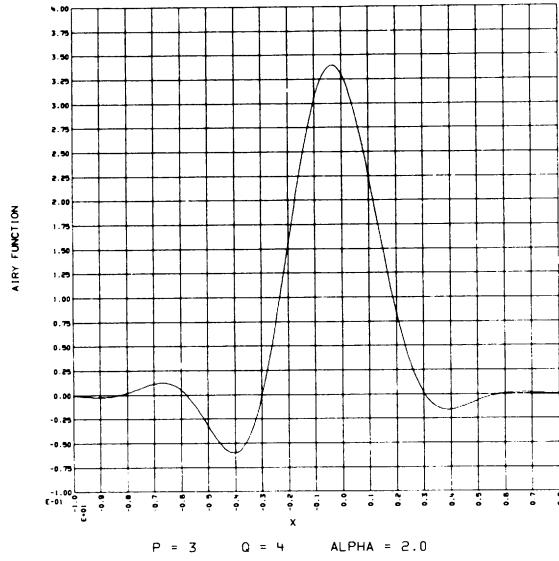


Figure 39

INTEGRAL OF THE AIRY FUNCTION

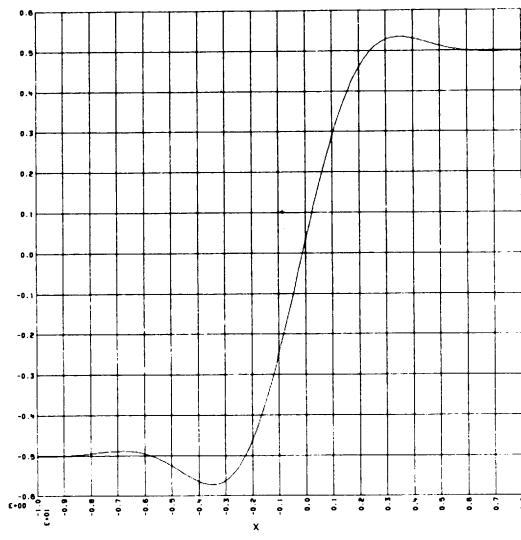


Figure 40

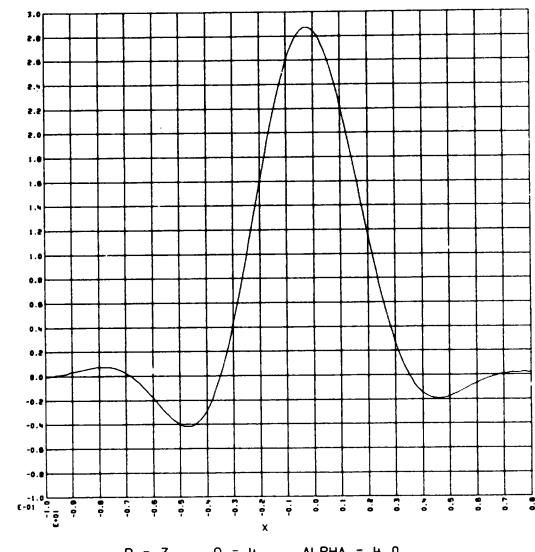


Figure 41

Table 18

P = 3 Q = 4 ALPHA = 0.5

Table 19

Table 20

P = 3 Q = 4 Δ RNA = 20

Table 21

INTEGRAL OF THE AIRY FUNCTION

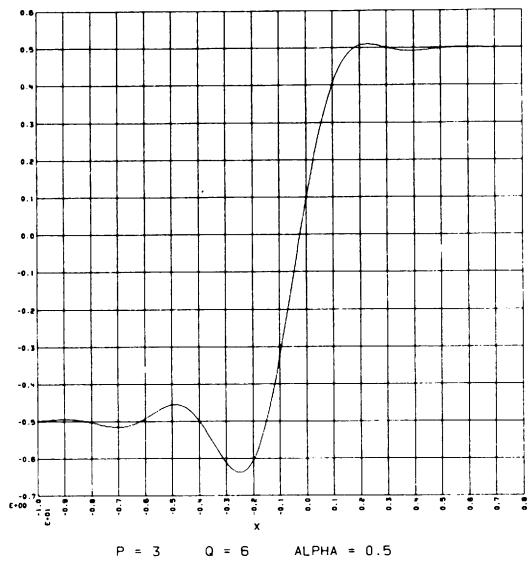


Figure 42

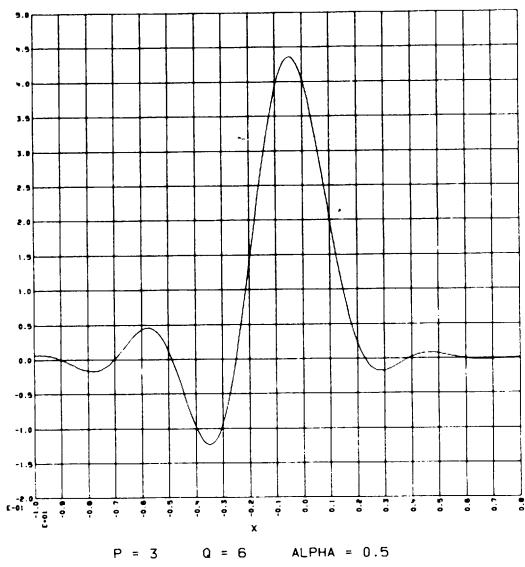


Figure 43

INTEGRAL OF THE AIRY FUNCTION

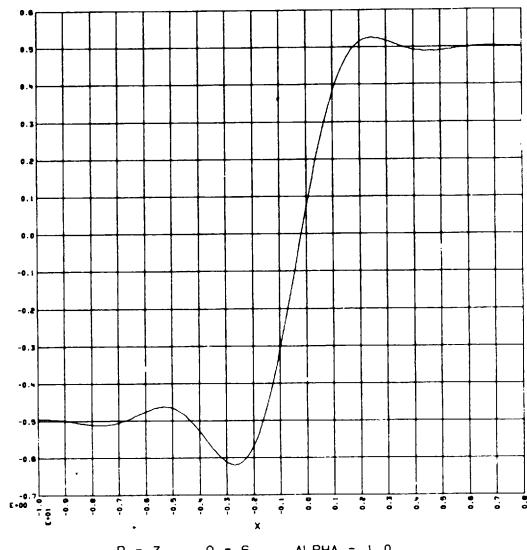
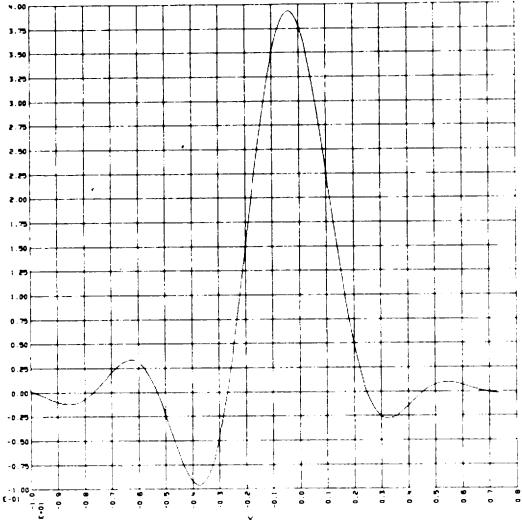


Figure 44



P = 3 Q = 6 ALPHA = 1.0

Figure 45

INTEGRAL OF THE AIRY FUNCTION

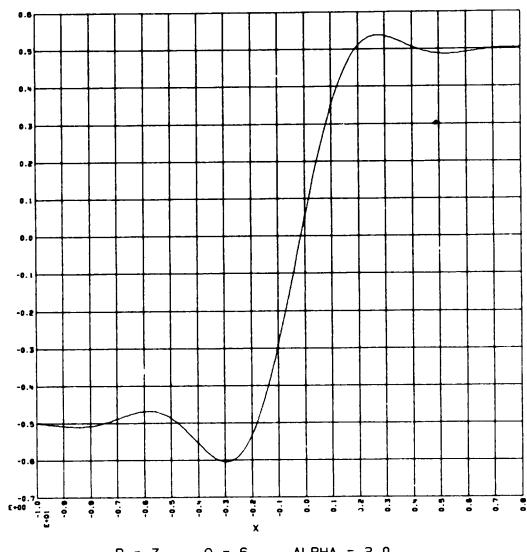


Figure 46

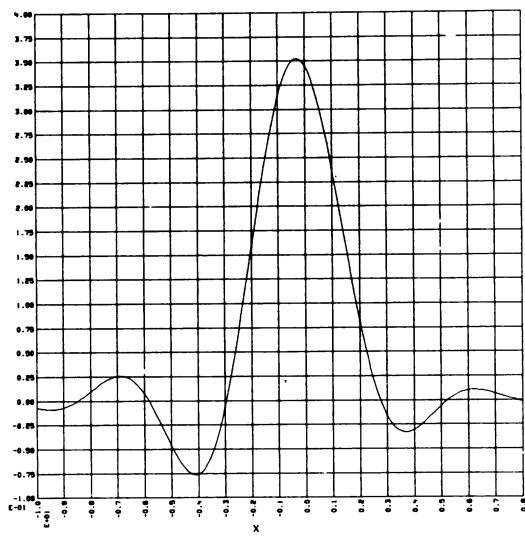
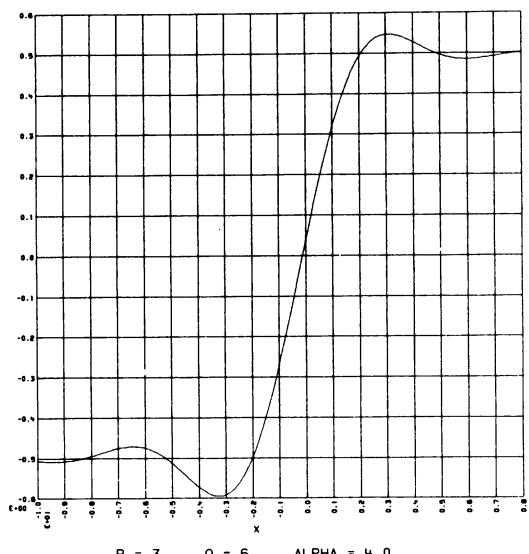


Figure 47

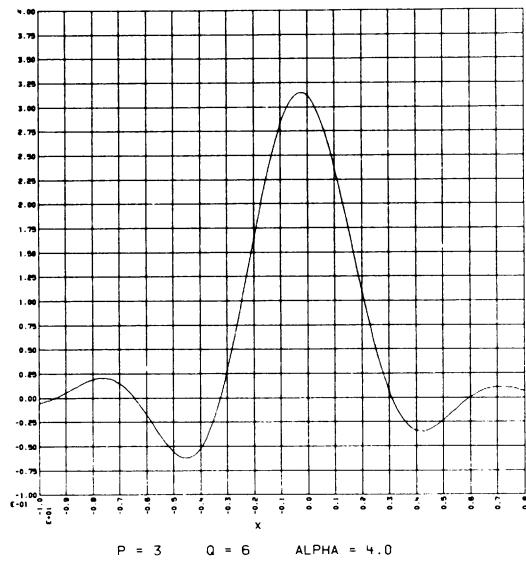
INTEGRAL OF THE AIRY FUNCTION



$P = 3 \quad Q = 6 \quad \text{ALPHA} = 4.0$

Figure 48

AIRY FUNCTION



$P = 3$ $Q = 6$ $\text{ALPHA} = 4.0$

Figure 49

Table 22

P = 3 Q = 8 ALPHA = 0.5

Table 23

P = 3 Q = 6 ALPHA = 1.0

Table 24

Table 25

$\alpha = 3$	$\alpha = 6$	$\alpha \text{ PHA} = \sim 0$
-1.0 - 0.507402	0.025600	1.3 - 0.499209
-1.0 - 0.507402	0.025600	1.3 - 0.499209
-0.9 - 0.507402	0.025600	1.3 - 0.499209
-0.8 - 0.507402	0.025600	1.3 - 0.499209
-0.7 - 0.507402	0.025600	1.3 - 0.499209
-0.6 - 0.507402	0.025600	1.3 - 0.499209
-0.5 - 0.507402	0.025600	1.3 - 0.499209
-0.4 - 0.507402	0.025600	1.3 - 0.499209
-0.3 - 0.507402	0.025600	1.3 - 0.499209
-0.2 - 0.507402	0.025600	1.3 - 0.499209
-0.1 - 0.507402	0.025600	1.3 - 0.499209
0.0 - 0.507402	0.025600	1.3 - 0.499209
0.1 + 0.507402	0.025600	1.3 - 0.499209
0.2 + 0.507402	0.025600	1.3 - 0.499209
0.3 + 0.507402	0.025600	1.3 - 0.499209
0.4 + 0.507402	0.025600	1.3 - 0.499209
0.5 + 0.507402	0.025600	1.3 - 0.499209
0.6 + 0.507402	0.025600	1.3 - 0.499209
0.7 + 0.507402	0.025600	1.3 - 0.499209
0.8 + 0.507402	0.025600	1.3 - 0.499209
0.9 + 0.507402	0.025600	1.3 - 0.499209
1.0 + 0.507402	0.025600	1.3 - 0.499209

INTEGRAL OF THE AIRY FUNCTION

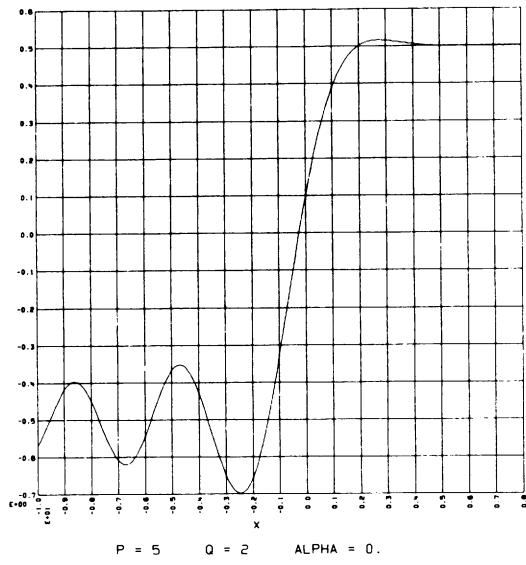


Figure 50

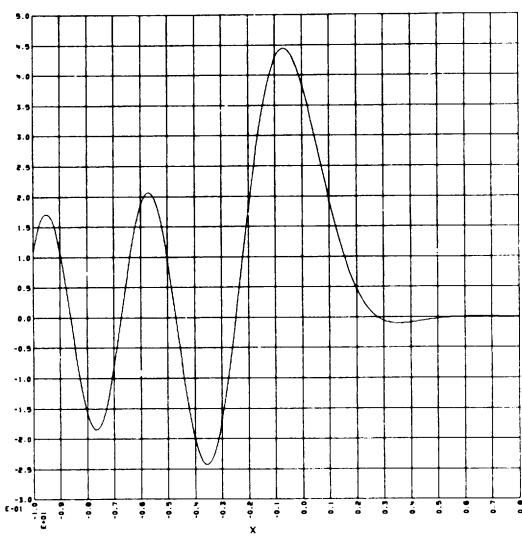


Figure 51

INTEGRAL OF THE AIRY FUNCTION

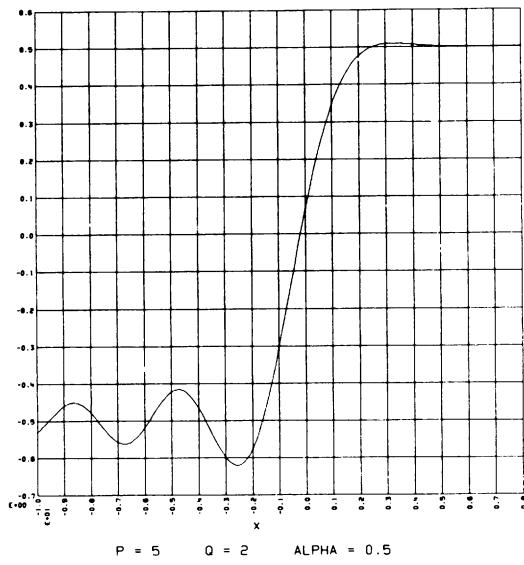


Figure 52

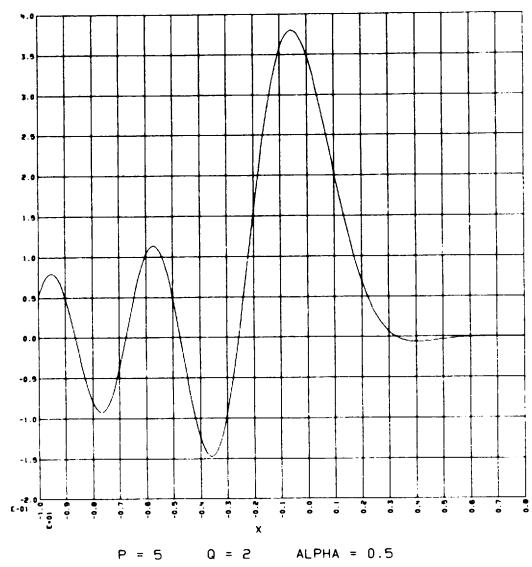
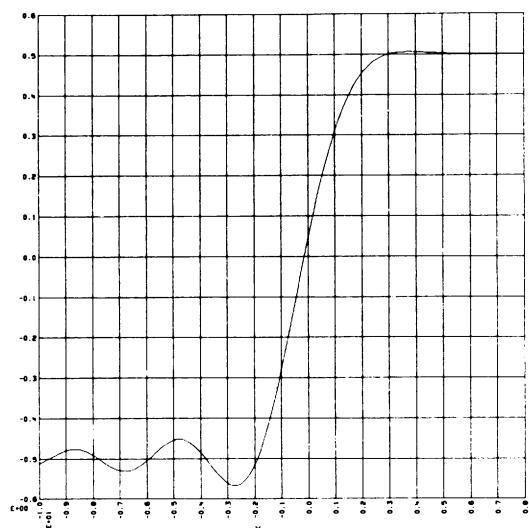


Figure 53

INTEGRAL OF THE AIRY FUNCTION



$P = 5$ $Q = 2$ $\text{ALPHA} = 1.0$

Figure 54

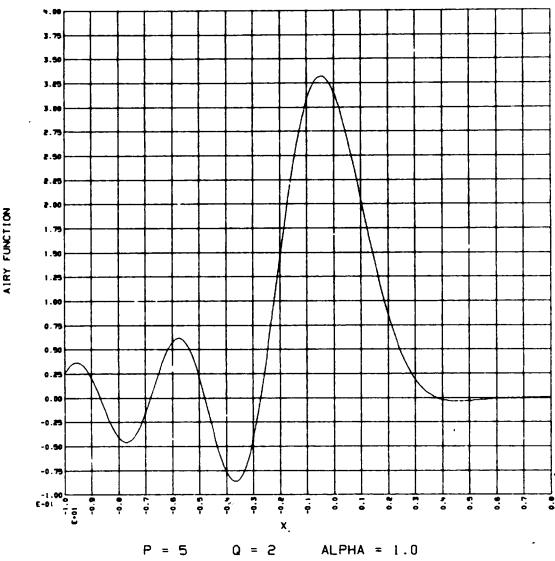


Figure 55

INTEGRAL OF THE AIRY FUNCTION

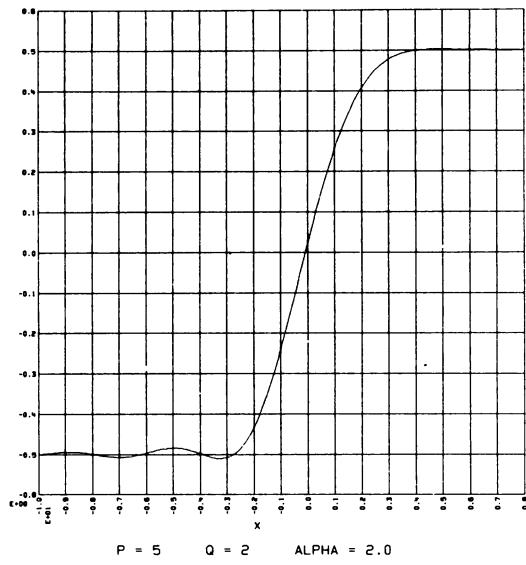


Figure 56

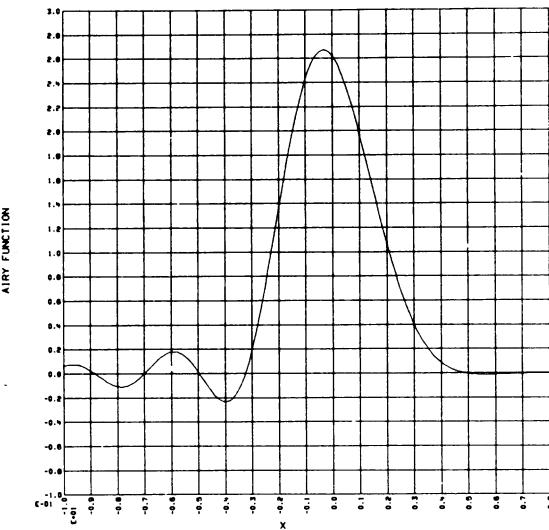


Figure 57

INTEGRAL OF THE AIRY FUNCTION

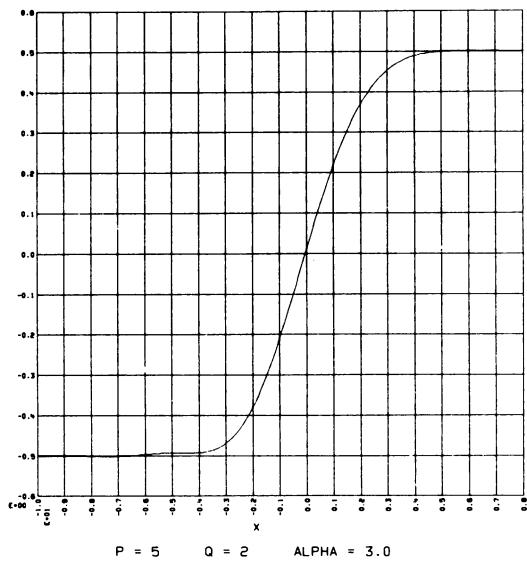


Figure 58

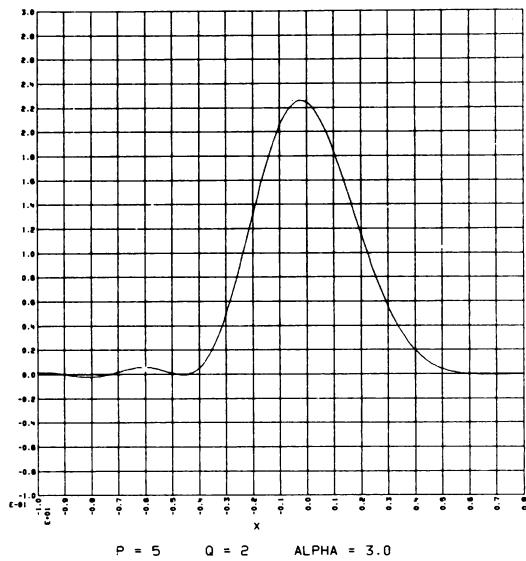


Figure 59

INTEGRAL OF THE AIRY FUNCTION

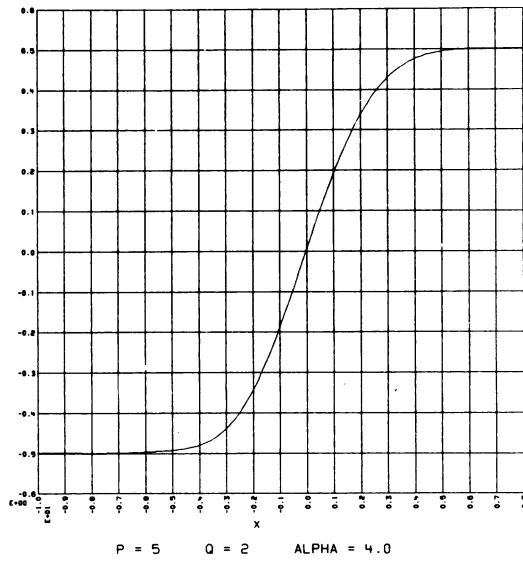
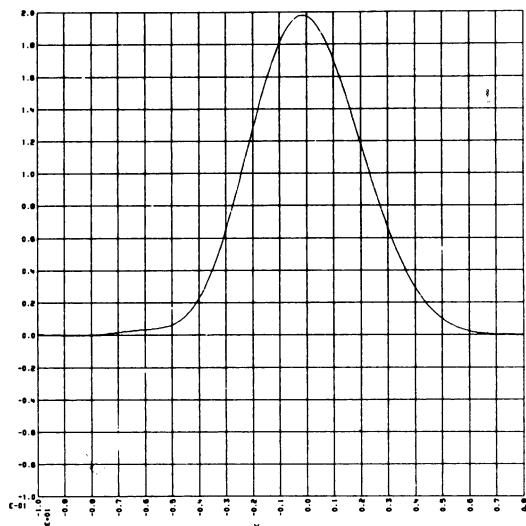


Figure 60

AIRY FUNCTION



P = 5 Q = 2 ALPHA = 4.0

Figure 61

Table 26

P = 5 Q = ? ALPHA = 0

Table 27

$$p = 5 \quad q = 2 \quad \Delta P_{\text{NA}} = 0.5$$

Table 28

$$P = 5 \quad Q = 2 \quad \Delta PMA = 10$$

Table 29

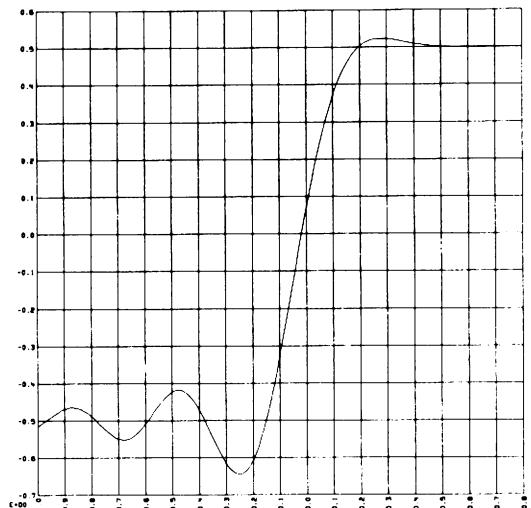
P = 3 Q = 2 ALPHAB = 2.0

Table 30

P = 5 Q = 2 ALPHA = 3.0

Table 31

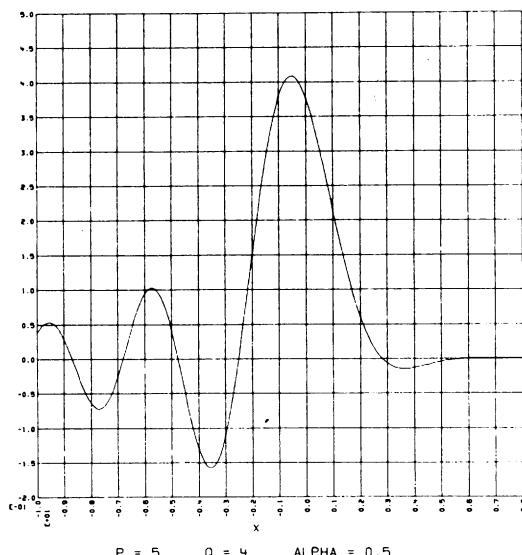
INTEGRAL OF THE AIRY FUNCTION



$P = 5 \quad Q = 4 \quad \text{ALPHA} = 0.5$

Figure 62

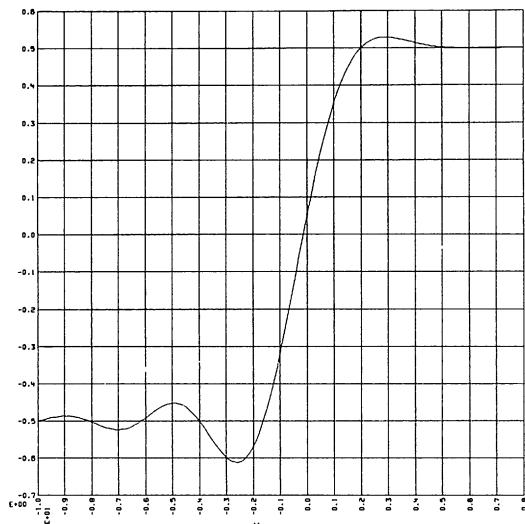
AIRY FUNCTION



$P = 5 \quad Q = 4 \quad \text{ALPHA} = 0.5$

Figure 63

INTEGRAL OF THE AIRY FUNCTION



$P = 5 \quad Q = 4 \quad \text{ALPHA} = 1.0$

Figure 64

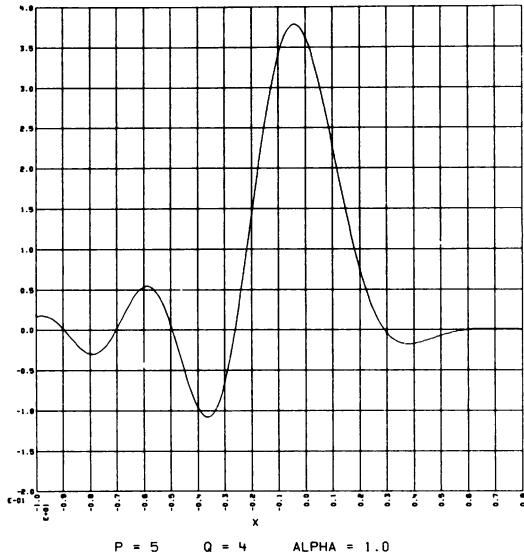


Figure 65

INTEGRAL OF THE AIRY FUNCTION

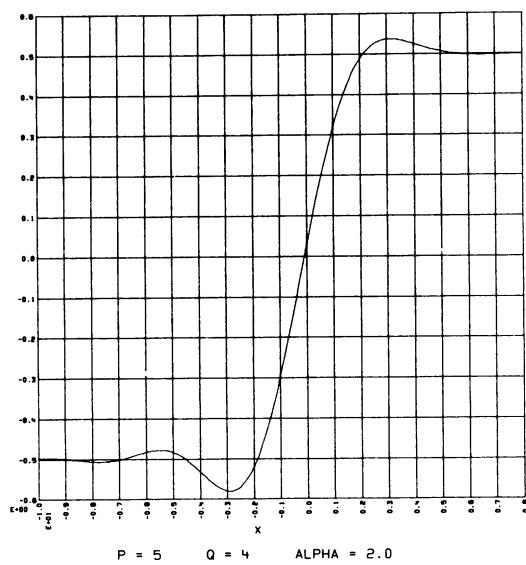


Figure 66

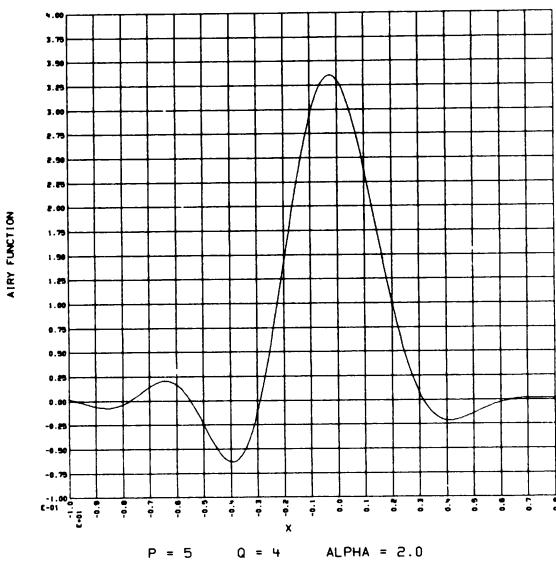


Figure 67

INTEGRAL OF THE AIRY FUNCTION

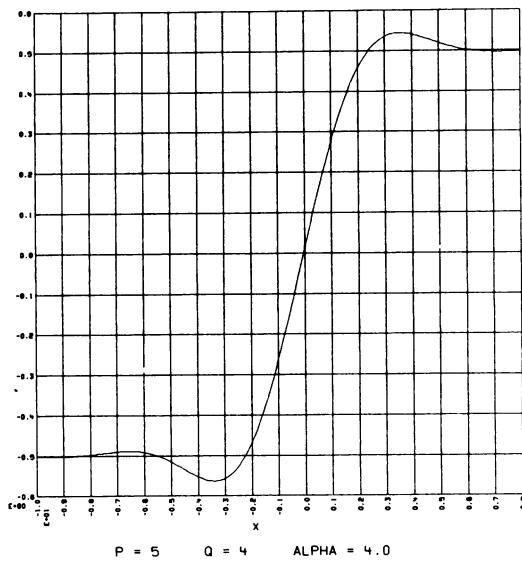


Figure 68

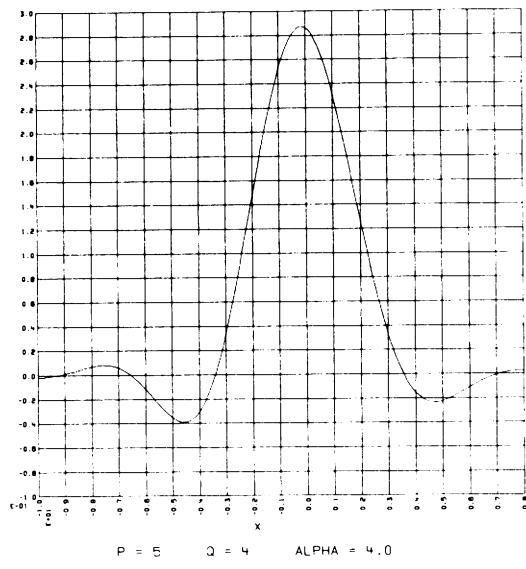


Figure 69

Table 32

Table 33

P = 5	Q = Y	ALPHA = 1.0
5	A11111	A11121
5	X-A11111	A11111
-1.0	-0.500308	0.017306
-0.9	-0.498878	0.017306
-0.8	-0.497447	0.017306
-0.7	-0.496017	0.017306
-0.6	-0.494587	0.017306
-0.5	-0.493157	0.017306
-0.4	-0.491727	0.017306
-0.3	-0.490297	0.017306
-0.2	-0.488867	0.017306
-0.1	-0.487437	0.017306
0	-0.486007	0.017306
0.1	-0.484577	0.017306
0.2	-0.483147	0.017306
0.3	-0.481717	0.017306
0.4	-0.480287	0.017306
0.5	-0.478857	0.017306
0.6	-0.477427	0.017306
0.7	-0.475997	0.017306
0.8	-0.474567	0.017306
0.9	-0.473137	0.017306
1.0	-0.471707	0.017306
1.1	-0.470277	0.017306
1.2	-0.468847	0.017306
1.3	-0.467417	0.017306
1.4	-0.465987	0.017306
1.5	-0.464557	0.017306
1.6	-0.463127	0.017306
1.7	-0.461697	0.017306
1.8	-0.460267	0.017306
1.9	-0.458837	0.017306
2.0	-0.457407	0.017306
2.1	-0.455977	0.017306
2.2	-0.454547	0.017306
2.3	-0.453117	0.017306
2.4	-0.451687	0.017306
2.5	-0.450257	0.017306
2.6	-0.448827	0.017306
2.7	-0.447397	0.017306
2.8	-0.445967	0.017306
2.9	-0.444537	0.017306
3.0	-0.443107	0.017306
3.1	-0.441677	0.017306
3.2	-0.440247	0.017306
3.3	-0.438817	0.017306
3.4	-0.437387	0.017306
3.5	-0.435957	0.017306
3.6	-0.434527	0.017306
3.7	-0.433097	0.017306
3.8	-0.431667	0.017306
3.9	-0.430237	0.017306
4.0	-0.428807	0.017306
4.1	-0.427377	0.017306
4.2	-0.425947	0.017306
4.3	-0.424517	0.017306
4.4	-0.423087	0.017306
4.5	-0.421657	0.017306
4.6	-0.420227	0.017306
4.7	-0.418797	0.017306
4.8	-0.417367	0.017306
4.9	-0.415937	0.017306
5.0	-0.414507	0.017306
5.1	-0.413077	0.017306
5.2	-0.411647	0.017306
5.3	-0.410217	0.017306
5.4	-0.408787	0.017306
5.5	-0.407357	0.017306
5.6	-0.405927	0.017306
5.7	-0.404497	0.017306
5.8	-0.403067	0.017306
5.9	-0.401637	0.017306
6.0	-0.400207	0.017306
6.1	-0.398777	0.017306
6.2	-0.397347	0.017306
6.3	-0.395917	0.017306
6.4	-0.394487	0.017306
6.5	-0.392957	0.017306
6.6	-0.391527	0.017306
6.7	-0.390097	0.017306
6.8	-0.388667	0.017306
6.9	-0.387237	0.017306
7.0	-0.385807	0.017306
7.1	-0.384377	0.017306
7.2	-0.382947	0.017306
7.3	-0.381517	0.017306
7.4	-0.380087	0.017306
7.5	-0.378657	0.017306
7.6	-0.377227	0.017306
7.7	-0.375797	0.017306
7.8	-0.374367	0.017306
7.9	-0.372937	0.017306
8.0	-0.371507	0.017306
8.1	-0.370077	0.017306
8.2	-0.368647	0.017306
8.3	-0.367217	0.017306
8.4	-0.365787	0.017306
8.5	-0.364357	0.017306
8.6	-0.362927	0.017306
8.7	-0.361497	0.017306
8.8	-0.360067	0.017306
8.9	-0.358637	0.017306
9.0	-0.357207	0.017306
9.1	-0.355777	0.017306
9.2	-0.354347	0.017306
9.3	-0.352917	0.017306
9.4	-0.351487	0.017306
9.5	-0.350057	0.017306
9.6	-0.348627	0.017306
9.7	-0.347197	0.017306
9.8	-0.345767	0.017306
9.9	-0.344337	0.017306
10.0	-0.342907	0.017306
10.1	-0.341477	0.017306
10.2	-0.340047	0.017306
10.3	-0.338617	0.017306
10.4	-0.337187	0.017306
10.5	-0.335757	0.017306
10.6	-0.334327	0.017306
10.7	-0.332897	0.017306
10.8	-0.331467	0.017306
10.9	-0.330037	0.017306
11.0	-0.328607	0.017306
11.1	-0.327177	0.017306
11.2	-0.325747	0.017306
11.3	-0.324317	0.017306
11.4	-0.322887	0.017306
11.5	-0.321457	0.017306
11.6	-0.319027	0.017306
11.7	-0.317597	0.017306
11.8	-0.316167	0.017306
11.9	-0.314737	0.017306
12.0	-0.313307	0.017306
12.1	-0.311877	0.017306
12.2	-0.310447	0.017306
12.3	-0.308017	0.017306
12.4	-0.306587	0.017306
12.5	-0.305157	0.017306
12.6	-0.303727	0.017306
12.7	-0.302297	0.017306
12.8	-0.300867	0.017306
12.9	-0.299437	0.017306
13.0	-0.297997	0.017306
13.1	-0.296567	0.017306
13.2	-0.295137	0.017306
13.3	-0.293707	0.017306
13.4	-0.292277	0.017306
13.5	-0.290847	0.017306
13.6	-0.289417	0.017306
13.7	-0.287987	0.017306
13.8	-0.286557	0.017306
13.9	-0.285127	0.017306
14.0	-0.283697	0.017306
14.1	-0.282267	0.017306
14.2	-0.280837	0.017306
14.3	-0.279407	0.017306
14.4	-0.277977	0.017306
14.5	-0.276547	0.017306
14.6	-0.275117	0.017306
14.7	-0.273687	0.017306
14.8	-0.272257	0.017306
14.9	-0.270827	0.017306
15.0	-0.269397	0.017306
15.1	-0.267967	0.017306
15.2	-0.266537	0.017306
15.3	-0.265107	0.017306
15.4	-0.263677	0.017306
15.5	-0.262247	0.017306
15.6	-0.260817	0.017306
15.7	-0.259387	0.017306
15.8	-0.257957	0.017306
15.9	-0.256527	0.017306
16.0	-0.255097	0.017306
16.1	-0.253667	0.017306
16.2	-0.252237	0.017306
16.3	-0.250807	0.017306
16.4	-0.249377	0.017306
16.5	-0.247947	0.017306
16.6	-0.246517	0.017306
16.7	-0.245087	0.017306
16.8	-0.243657	0.017306
16.9	-0.242227	0.017306
17.0	-0.240797	0.017306
17.1	-0.239367	0.017306
17.2	-0.237937	0.017306
17.3	-0.236507	0.017306
17.4	-0.235077	0.017306
17.5	-0.233647	0.017306
17.6	-0.232217	0.017306
17.7	-0.230787	0.017306
17.8	-0.229357	0.017306
17.9	-0.227927	0.017306
18.0	-0.226497	0.017306
18.1	-0.225067	0.017306
18.2	-0.223637	0.017306
18.3	-0.222207	0.017306
18.4	-0.220777	0.017306
18.5	-0.219347	0.017306
18.6	-0.217917	0.017306
18.7	-0.216487	0.017306
18.8	-0.215057	0.017306
18.9	-0.213627	0.017306
19.0	-0.212197	0.017306
19.1	-0.210767	0.017306
19.2	-0.209337	0.017306
19.3	-0.207907	0.017306
19.4	-0.206477	0.017306
19.5	-0.204047	0.017306
19.6	-0.202617	0.017306
19.7	-0.201187	0.017306
19.8	-0.199757	0.017306
19.9	-0.198327	0.017306
20.0	-0.196897	0.017306
20.1	-0.195467	0.017306
20.2	-0.194037	0.017306
20.3	-0.192607	0.017306
20.4	-0.191177	0.017306
20.5	-0.189747	0.017306
20.6	-0.188317	0.017306
20.7	-0.186887	0.017306
20.8	-0.185457	0.017306
20.9	-0.184027	0.017306
21.0	-0.182597	0.017306
21.1	-0.181167	0.017306
21.2	-0.179737	0.017306
21.3	-0.178307	0.017306
21.4	-0.176877	0.017306
21.5	-0.175447	0.017306
21.6	-0.174017	0.017306
21.7	-0.172587	0.017306
21.8	-0.171157	0.017306
21.9	-0.169727	0.017306
22.0	-0.168297	0.017306
22.1	-0.166867	0.017306
22.2	-0.165437	0.017306
22.3	-0.163907	0.017306
22.4	-0.162477	0.017306
22.5	-0.160947	0.017306
22.6	-0.159517	0.017306
22.7	-0.158087	0.017306
22.8	-0.156657	0.017306
22.9	-0.155227	0.017306
23.0	-0.153797	0.017306
23.1	-0.152367	0.017306
23.2	-0.150937	0.017306
23.3	-0.149507	0.017306
23.4	-0.148077	0.017306
23.5	-0.146647	0.017306
23.6	-0.145217	0.017306
23.7	-0.143787	0.017306
23.8	-0.142357	0.017306
23.9	-0.140927	0.017306
24.0	-0.139497	0.017306
24.1	-0.137967	0.017306
24.2	-0.136537	0.017306
24.3	-0.135107	0.017306
24.4	-0.133677	0.017306
24.5	-0.132247	0.017306
24.6	-0.130817	0.017306
24.7	-0.129387	0.017306
24.8	-0.127957	0.017306
24.9	-0.126527	0.017306
25.0	-0.125097	0.017306
25.1	-0.123667	0.017306
25.2	-0.122237	0.017306
25.3	-0.120807	0.017306
25.4	-0.119377	0.017306
25.5	-0.117947	0.017306
25.6	-0.116517	0.017306
25.7	-0.115087	0.017306
25.8	-0.113657	0.017306
25.9	-0.112227	0.017306
26.0	-0.110797	0.017306
26.1	-0.109367	0.017306
26.2	-0.107937	0.017306
26.3	-0.106507	0.017306
26.4	-0.105077	0.017306
26.5	-0.103647	0.017306
26.6	-0.102217	0.017306
26.7	-0.100787	0.017306
26.8	-0.099357	0.017306
26.9	-0.097927	0.017306
27.0	-0.096497	0.017306
27.1	-0.095067	0.017306
27.2	-0.093637	0.017306
27.3	-0.092207	0.017306
27.4	-0.090777	0.017306
27.5	-0.089347	0.017306
27.6	-0.087917	0.017306
27.7	-0.086487	0.017306
27.8	-0.085057	0.017306
27.9	-0.083627	0.017306
28.0	-0.082197	0.017306
28.1	-0.080767	0.017306
28.2	-0.079337	0.017306
28.3	-0.077907	0.017306
28.4	-0.076477	0.017306
28.5	-0.075047	0.017306
28.6	-0.073617	0.017306
28.7	-0.072187	0.017306
28.8	-0.070757	0.017306
28.9	-0.069327	0.017306
29.0	-0.067897	0.017306
29.1	-0.066467	0.017306
29.2	-0.065037	0.017306
29.3	-0.063607	0.017306
29.4	-0.062177	0.017306
29.5	-0.060747	0.017

Table 34

$$P = 3 \quad Q = 4 \quad \text{ALPHA} = 2.0$$

Table 35

P = 3 Q = 4 ALPHA = 4.0

INTEGRAL OF THE AIRY FUNCTION

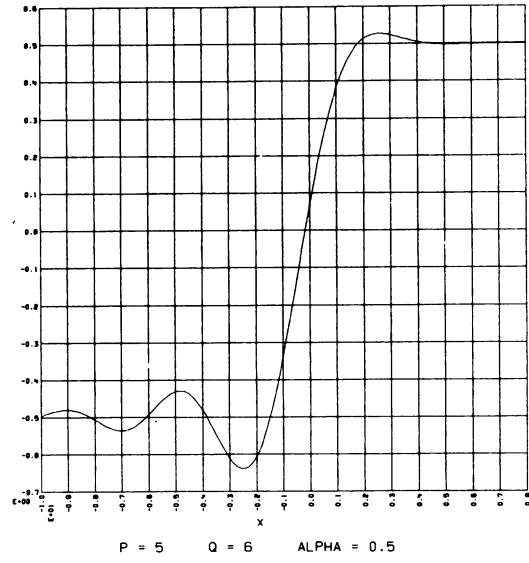


Figure 70

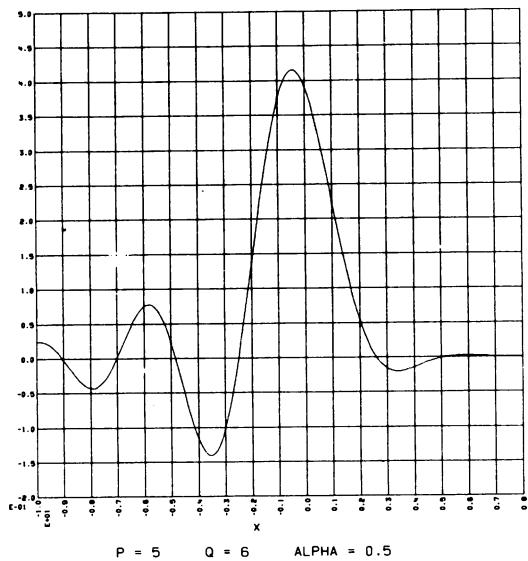


Figure 71

INTEGRAL OF THE AIRY FUNCTION

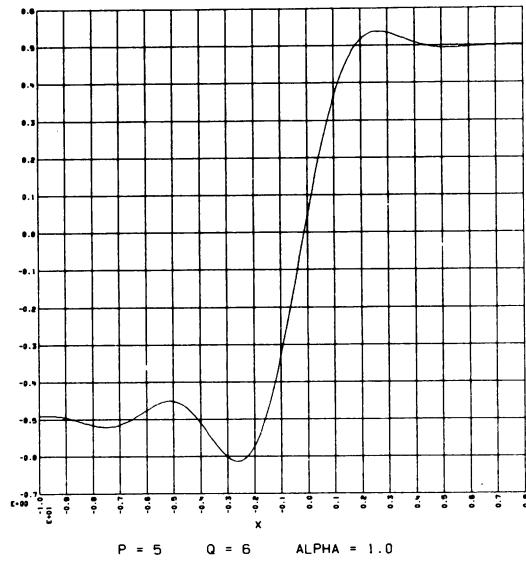


Figure 72

AIRY FUNCTION

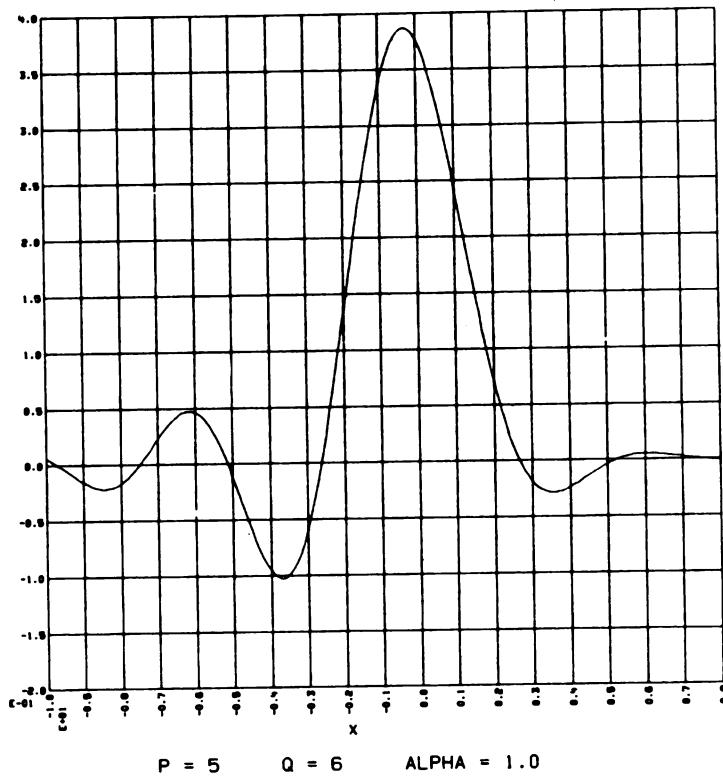


Figure 73

INTEGRAL OF THE AIRY FUNCTION

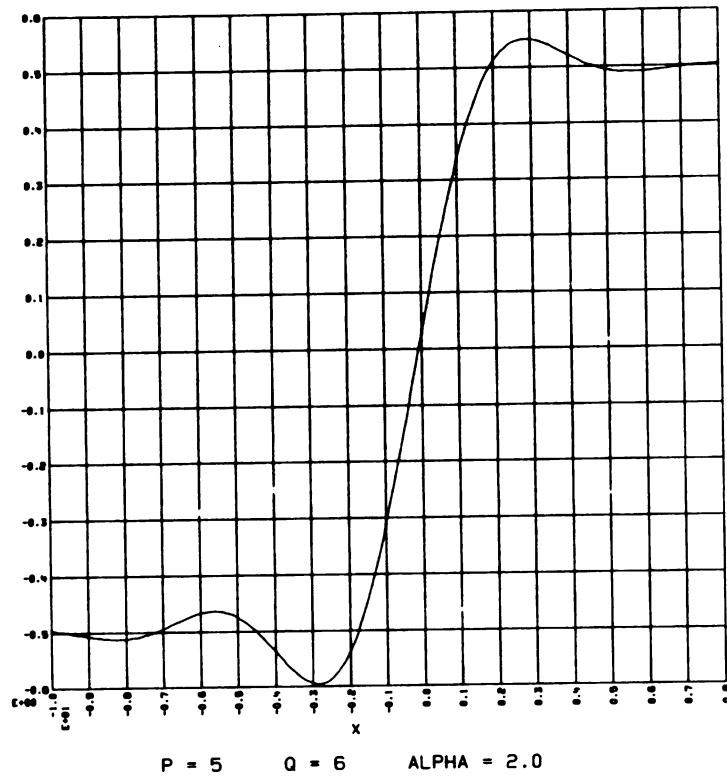


Figure 74

AIRY FUNCTION

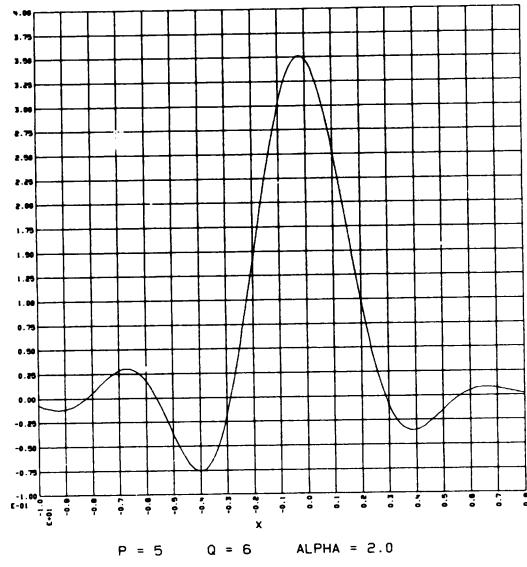


Figure 75

INTEGRAL OF THE AIRY FUNCTION

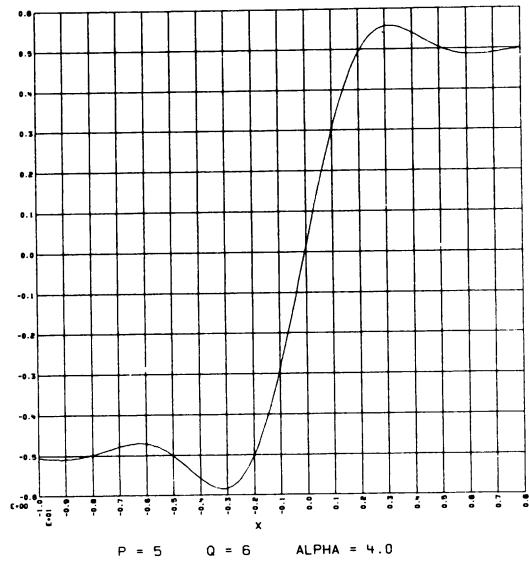


Figure 76

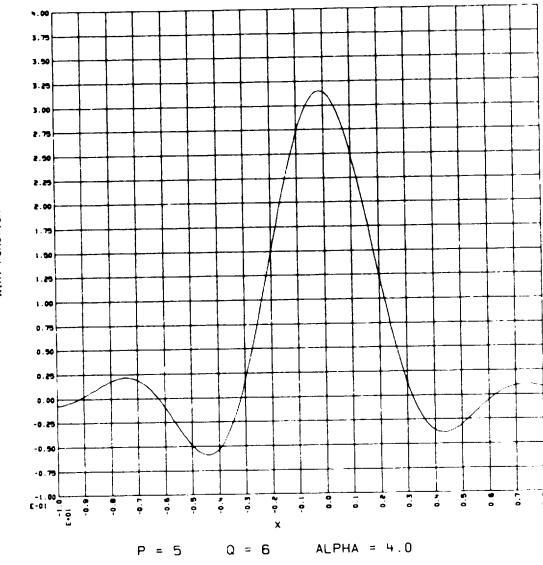


Figure 77

Table 36

	P = 5	Q = 6	ALPHA = 0.5
-1.0	-0.11111	0.11111	0.11111
-0.9	-0.10000	0.10000	0.10000
-0.8	-0.08926	0.08926	0.08926
-0.7	-0.07857	0.07857	0.07857
-0.6	-0.06807	0.06807	0.06807
-0.5	-0.05767	0.05767	0.05767
-0.4	-0.04737	0.04737	0.04737
-0.3	-0.03717	0.03717	0.03717
-0.2	-0.02707	0.02707	0.02707
-0.1	-0.01697	0.01697	0.01697
0.0	-0.00687	0.00687	0.00687
0.1	0.01697	-0.01697	0.01697
0.2	0.02707	-0.02707	0.02707
0.3	0.03717	-0.03717	0.03717
0.4	0.04737	-0.04737	0.04737
0.5	0.05767	-0.05767	0.05767
0.6	0.06807	-0.06807	0.06807
0.7	0.07857	-0.07857	0.07857
0.8	0.08926	-0.08926	0.08926
0.9	0.10000	-0.10000	0.10000
1.0	0.11111	-0.11111	0.11111

Table 37

P = 5 Q = 6 ALPHA = 1.0

Table 38

P = 5 Q = 6 ALPHA = 20

Table 39

P = 5 Q = 5 ALPHA = 4.3

INTEGRAL OF THE AIRY FUNCTION

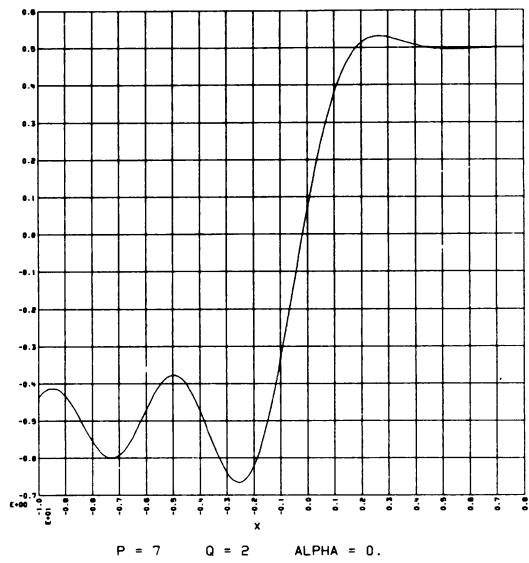


Figure 78

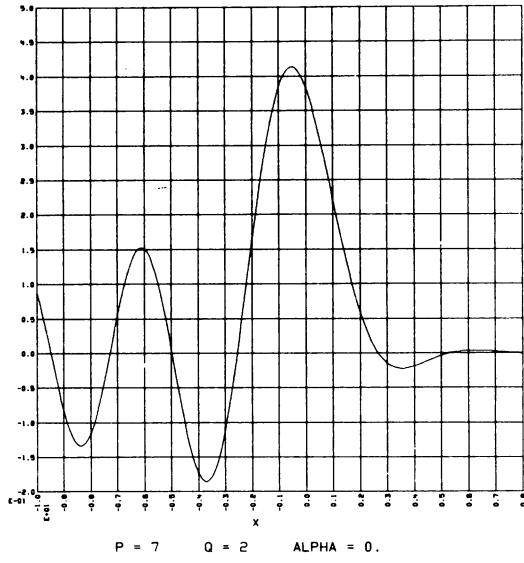


Figure 79

INTEGRAL OF THE AIRY FUNCTION

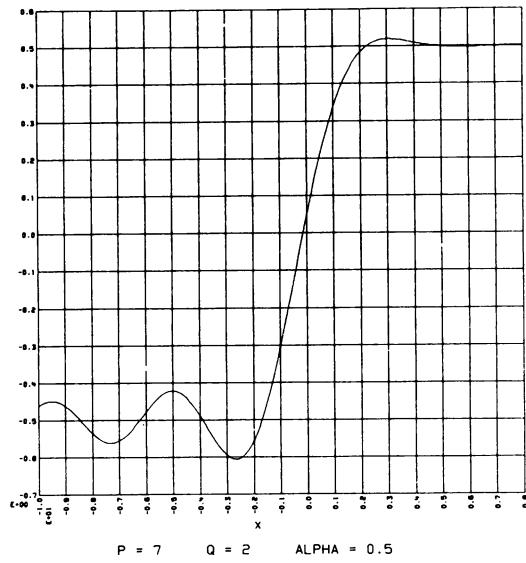


Figure 80

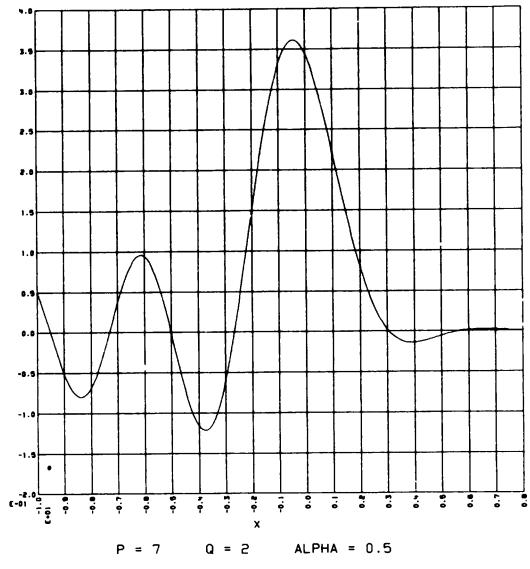


Figure 81

INTEGRAL OF THE AIRY FUNCTION

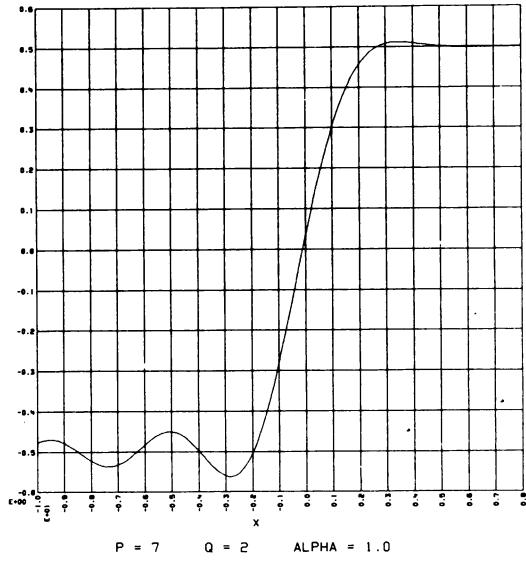
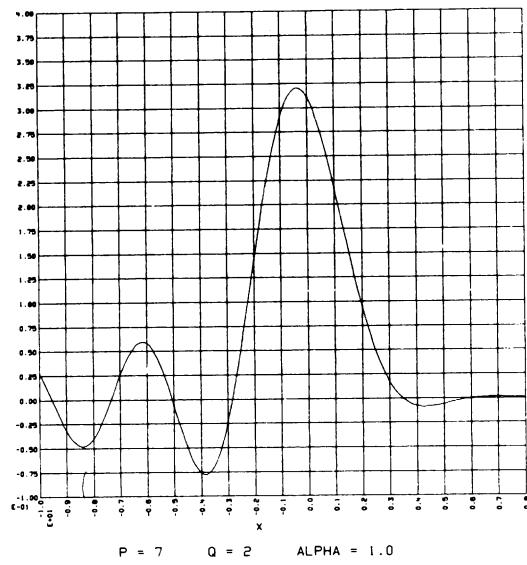


Figure 82

AIRY FUNCTION



$P = 7 \quad Q = 2 \quad \text{ALPHA} = 1.0$

Figure 83

INTEGRAL OF THE AIRY FUNCTION

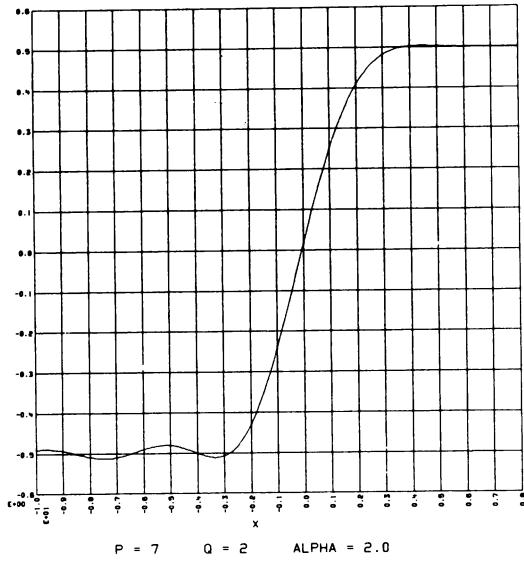


Figure 84

AIRY FUNCTION

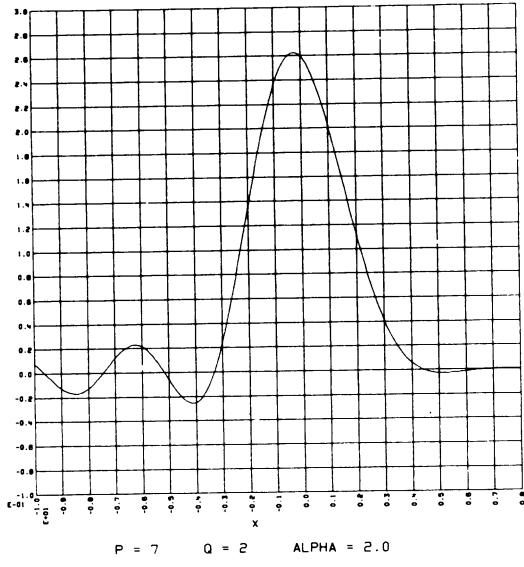


Figure 85

INTEGRAL OF THE AIRY FUNCTION

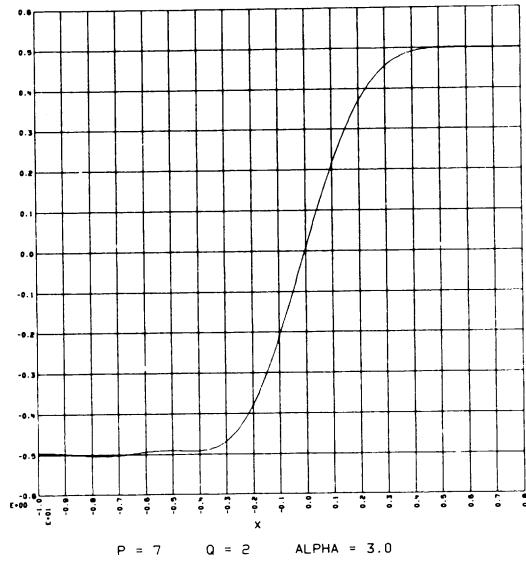


Figure 86

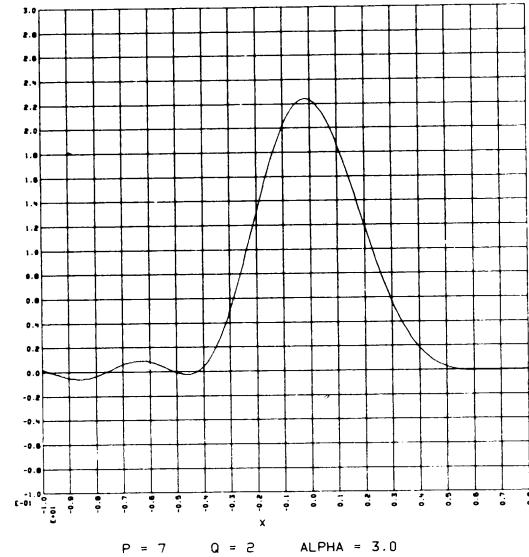


Figure 87

INTEGRAL OF THE AIRY FUNCTION

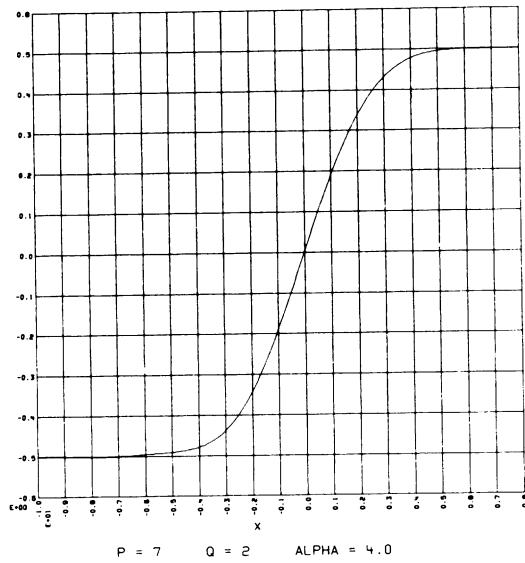
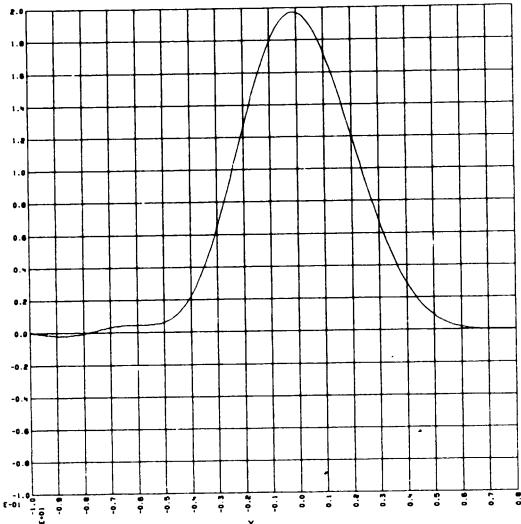


Figure 88



$P = 7$ $Q = 2$ $\text{ALPHA} = 4.0$

Figure 89

Table 40

Table 41

P = 7	Q = 2	ALPHA = 0.5
-1.0	0.462120	0.048866
-0.9	-0.578510	0.048866
-0.8	0.578510	0.048866
-0.7	-0.513510	0.021667
-0.6	0.513510	0.021667
-0.5	-0.449132	0.003232
-0.4	0.449132	0.003232
-0.3	-0.385292	-0.031190
-0.2	0.385292	-0.031190
-0.1	-0.324205	0.1
0.0	0.324205	0.1
0.1	-0.263205	0.051133
0.2	0.263205	0.051133
0.3	-0.208305	-0.097767
0.4	0.208305	-0.097767
0.5	-0.153024	0.003232
0.6	0.153024	0.003232
0.7	-0.098324	-0.031190
0.8	0.098324	-0.031190
0.9	-0.043324	0.1
1.0	0.043324	0.1
-1.0	0.515950	0.011470
-0.9	-0.515950	0.011470
-0.8	0.515950	0.011470
-0.7	-0.513432	0.011470
-0.6	0.513432	0.011470
-0.5	-0.502477	0.013069
-0.4	0.502477	0.013069
-0.3	-0.496869	0.359191
-0.2	0.496869	0.359191
-0.1	-0.492689	0.359191
0.0	0.492689	0.359191
0.1	-0.489910	0.361192
0.2	0.489910	0.361192
0.3	-0.487120	0.361192
0.4	0.487120	0.361192
0.5	-0.484330	0.361192
0.6	0.484330	0.361192
0.7	-0.481540	0.361192
0.8	0.481540	0.361192
0.9	-0.478750	0.361192
1.0	0.478750	0.361192
-1.0	0.542926	0.010631
-0.9	-0.542926	0.010631
-0.8	0.542926	0.010631
-0.7	-0.540136	0.010631
-0.6	0.540136	0.010631
-0.5	-0.537346	0.010631
-0.4	0.537346	0.010631
-0.3	-0.534556	0.010631
-0.2	0.534556	0.010631
-0.1	-0.531766	0.010631
0.0	0.531766	0.010631
0.1	-0.528976	0.010631
0.2	0.528976	0.010631
0.3	-0.526186	0.010631
0.4	0.526186	0.010631
0.5	-0.523406	0.010631
0.6	0.523406	0.010631
0.7	-0.520616	0.010631
0.8	0.520616	0.010631
0.9	-0.517826	0.010631
1.0	0.517826	0.010631
-1.0	0.549296	0.009575
-0.9	-0.549296	0.009575
-0.8	0.549296	0.009575
-0.7	-0.546506	0.009575
-0.6	0.546506	0.009575
-0.5	-0.543716	0.009575
-0.4	0.543716	0.009575
-0.3	-0.540926	0.009575
-0.2	0.540926	0.009575
-0.1	-0.538136	0.009575
0.0	0.538136	0.009575
0.1	-0.535346	0.009575
0.2	0.535346	0.009575
0.3	-0.532556	0.009575
0.4	0.532556	0.009575
0.5	-0.529766	0.009575
0.6	0.529766	0.009575
0.7	-0.526976	0.009575
0.8	0.526976	0.009575
0.9	-0.524186	0.009575
1.0	0.524186	0.009575
-1.0	0.550496	0.008495
-0.9	-0.550496	0.008495
-0.8	0.550496	0.008495
-0.7	-0.547706	0.008495
-0.6	0.547706	0.008495
-0.5	-0.544916	0.008495
-0.4	0.544916	0.008495
-0.3	-0.542126	0.008495
-0.2	0.542126	0.008495
-0.1	-0.539336	0.008495
0.0	0.539336	0.008495
0.1	-0.536546	0.008495
0.2	0.536546	0.008495
0.3	-0.533756	0.008495
0.4	0.533756	0.008495
0.5	-0.530966	0.008495
0.6	0.530966	0.008495
0.7	-0.528176	0.008495
0.8	0.528176	0.008495
0.9	-0.525386	0.008495
1.0	0.525386	0.008495
-1.0	0.557696	0.007415
-0.9	-0.557696	0.007415
-0.8	0.557696	0.007415
-0.7	-0.554906	0.007415
-0.6	0.554906	0.007415
-0.5	-0.552116	0.007415
-0.4	0.552116	0.007415
-0.3	-0.549326	0.007415
-0.2	0.549326	0.007415
-0.1	-0.546536	0.007415
0.0	0.546536	0.007415
0.1	-0.543746	0.007415
0.2	0.543746	0.007415
0.3	-0.540956	0.007415
0.4	0.540956	0.007415
0.5	-0.538166	0.007415
0.6	0.538166	0.007415
0.7	-0.535376	0.007415
0.8	0.535376	0.007415
0.9	-0.532586	0.007415
1.0	0.532586	0.007415
-1.0	0.565906	0.006364
-0.9	-0.565906	0.006364
-0.8	0.565906	0.006364
-0.7	-0.563116	0.006364
-0.6	0.563116	0.006364
-0.5	-0.560326	0.006364
-0.4	0.560326	0.006364
-0.3	-0.557536	0.006364
-0.2	0.557536	0.006364
-0.1	-0.554746	0.006364
0.0	0.554746	0.006364
0.1	-0.551956	0.006364
0.2	0.551956	0.006364
0.3	-0.549166	0.006364
0.4	0.549166	0.006364
0.5	-0.546376	0.006364
0.6	0.546376	0.006364
0.7	-0.543586	0.006364
0.8	0.543586	0.006364
0.9	-0.540796	0.006364
1.0	0.540796	0.006364
-1.0	0.564006	0.005312
-0.9	-0.564006	0.005312
-0.8	0.564006	0.005312
-0.7	-0.561216	0.005312
-0.6	0.561216	0.005312
-0.5	-0.558426	0.005312
-0.4	0.558426	0.005312
-0.3	-0.555636	0.005312
-0.2	0.555636	0.005312
-0.1	-0.552846	0.005312
0.0	0.552846	0.005312
0.1	-0.550056	0.005312
0.2	0.550056	0.005312
0.3	-0.547266	0.005312
0.4	0.547266	0.005312
0.5	-0.544476	0.005312
0.6	0.544476	0.005312
0.7	-0.541686	0.005312
0.8	0.541686	0.005312
0.9	-0.538896	0.005312
1.0	0.538896	0.005312
-1.0	0.562206	0.004291
-0.9	-0.562206	0.004291
-0.8	0.562206	0.004291
-0.7	-0.559416	0.004291
-0.6	0.559416	0.004291
-0.5	-0.556626	0.004291
-0.4	0.556626	0.004291
-0.3	-0.553836	0.004291
-0.2	0.553836	0.004291
-0.1	-0.551046	0.004291
0.0	0.551046	0.004291
0.1	-0.548256	0.004291
0.2	0.548256	0.004291
0.3	-0.545466	0.004291
0.4	0.545466	0.004291
0.5	-0.542676	0.004291
0.6	0.542676	0.004291
0.7	-0.539886	0.004291
0.8	0.539886	0.004291
0.9	-0.537096	0.004291
1.0	0.537096	0.004291
-1.0	0.565506	0.003232
-0.9	-0.565506	0.003232
-0.8	0.565506	0.003232
-0.7	-0.562716	0.003232
-0.6	0.562716	0.003232
-0.5	-0.559926	0.003232
-0.4	0.559926	0.003232
-0.3	-0.557136	0.003232
-0.2	0.557136	0.003232
-0.1	-0.554346	0.003232
0.0	0.554346	0.003232
0.1	-0.551556	0.003232
0.2	0.551556	0.003232
0.3	-0.548766	0.003232
0.4	0.548766	0.003232
0.5	-0.545976	0.003232
0.6	0.545976	0.003232
0.7	-0.543186	0.003232
0.8	0.543186	0.003232
0.9	-0.540396	0.003232
1.0	0.540396	0.003232
-1.0	0.564806	0.002212
-0.9	-0.564806	0.002212
-0.8	0.564806	0.002212
-0.7	-0.562016	0.002212
-0.6	0.562016	0.002212
-0.5	-0.559226	0.002212
-0.4	0.559226	0.002212
-0.3	-0.556436	0.002212
-0.2	0.556436	0.002212
-0.1	-0.553646	0.002212
0.0	0.553646	0.002212
0.1	-0.550856	0.002212
0.2	0.550856	0.002212
0.3	-0.548066	0.002212
0.4	0.548066	0.002212
0.5	-0.545276	0.002212
0.6	0.545276	0.002212
0.7	-0.542486	0.002212
0.8	0.542486	0.002212
0.9	-0.539696	0.002212
1.0	0.539696	0.002212
-1.0	0.565206	0.001190
-0.9	-0.565206	0.001190
-0.8	0.565206	0.001190
-0.7	-0.562416	0.001190
-0.6	0.562416	0.001190
-0.5	-0.559626	0.001190
-0.4	0.559626	0.001190
-0.3	-0.556836	0.001190
-0.2	0.556836	0.001190
-0.1	-0.554046	0.001190
0.0	0.554046	0.001190
0.1	-0.551256	0.001190
0.2	0.551256	0.001190
0.3	-0.548466	0.001190
0.4	0.548466	0.001190
0.5	-0.545676	0.001190
0.6	0.545676	0.001190
0.7	-0.542886	0.001190
0.8	0.542886	0.001190
0.9	-0.539996	0.001190
1.0	0.539996	0.001190
-1.0	0.565686	0.000176
-0.9	-0.565686	0.000176
-0.8	0.565686	0.000176
-0.7	-0.562896	0.000176
-0.6	0.562896	0.000176
-0.5	-0.559906	0.000176
-0.4	0.559906	0.000176
-0.3	-0.557016	0.000176
-0.2	0.557016	0.000176
-0.1	-0.554226	0.000176
0.0	0.554226	0.000176
0.1	-0.551436	0.000176
0.2	0.551436	0.000176
0.3	-0.548646	0.000176
0.4	0.548646	0.000176
0.5	-0.545856	0.000176
0.6	0.545856	0.000176
0.7	-0.543066	0.000176
0.8	0.543066	0.000176
0.9	-0.540176	0.000176
1.0	0.540176	0.000176
-1.0	0.566186	0.000116
-0.9	-0.566186	0.000116
-0.8	0.566186	0.000116
-0.7	-0.563396	0.000116
-0.6	0.563396	0.000116
-0.5	-0.560606	0.000116
-0.4	0.560606	0.000116
-0.3	-0.557816	0.000116
-0.2	0.557816	0.000116
-0.1	-0.555026	0.000116
0.0	0.555026	0.000116
0.1	-0.552236	0.000116
0.2	0.552236	0.000116
0.3	-0.549446	0.000116
0.4	0.549446	0.000116
0.5	-0.546656	0.000116
0.6	0.546656	0.000116
0.7	-0.543866	0.000116
0.8	0.543866	0.000116
0.9	-0.541076	0.000116
1.0	0.541076	0.000116
-1.0	0.566786	0.000116
-0.9	-0.566786	0.000116
-0.8	0.566786	0.000116
-0.7	-0.563996	0.000116
-0.6	0.563996	0.000116
-0.5	-0.561206	0.000116
-0.4	0.561206	0.000116
-0.3	-0.558416	0.000116
-0.2	0.558416	0.000116
-0.1	-0.555626	0.000116
0.0	0.555626	0.000116
0.1	-0.552836	0.000116
0.2	0.552836	0.000116
0.3	-0.549446	0.000116
0.4	0.549446	0.000116
0.5	-0.546656	0.000116
0.6	0.546656	0.000116
0.7	-0.543866	0.000116
0.8	0.543866	0.000116

Table 42

P = 7 Q = 2 ALPHA = 1.0

Table 43

P = ? Q = ? ALPHIA = 2.0

Table 44

P - 7	Q - 2	ALPHA = 3.0
-10.0	0.000001	0.001785
-9.9	0.000001	0.001785
-9.8	0.000001	0.001785
-9.7	0.000001	0.001785
-9.6	0.000001	0.001785
-9.5	0.000001	0.001785
-9.4	0.000001	0.001785
-9.3	0.000001	0.001785
-9.2	0.000001	0.001785
-9.1	0.000001	0.001785
-9.0	0.000001	0.001785
-8.9	0.000001	0.001785
-8.8	0.000001	0.001785
-8.7	0.000001	0.001785
-8.6	0.000001	0.001785
-8.5	0.000001	0.001785
-8.4	0.000001	0.001785
-8.3	0.000001	0.001785
-8.2	0.000001	0.001785
-8.1	0.000001	0.001785
-8.0	0.000001	0.001785
-7.9	0.000001	0.001785
-7.8	0.000001	0.001785
-7.7	0.000001	0.001785
-7.6	0.000001	0.001785
-7.5	0.000001	0.001785
-7.4	0.000001	0.001785
-7.3	0.000001	0.001785
-7.2	0.000001	0.001785
-7.1	0.000001	0.001785
-7.0	0.000001	0.001785
-6.9	0.000001	0.001785
-6.8	0.000001	0.001785
-6.7	0.000001	0.001785
-6.6	0.000001	0.001785
-6.5	0.000001	0.001785
-6.4	0.000001	0.001785
-6.3	0.000001	0.001785
-6.2	0.000001	0.001785
-6.1	0.000001	0.001785
-6.0	0.000001	0.001785
-5.9	0.000001	0.001785
-5.8	0.000001	0.001785
-5.7	0.000001	0.001785
-5.6	0.000001	0.001785
-5.5	0.000001	0.001785
-5.4	0.000001	0.001785
-5.3	0.000001	0.001785
-5.2	0.000001	0.001785
-5.1	0.000001	0.001785
-5.0	0.000001	0.001785
-4.9	0.000001	0.001785
-4.8	0.000001	0.001785
-4.7	0.000001	0.001785
-4.6	0.000001	0.001785
-4.5	0.000001	0.001785
-4.4	0.000001	0.001785
-4.3	0.000001	0.001785
-4.2	0.000001	0.001785
-4.1	0.000001	0.001785
-4.0	0.000001	0.001785
-3.9	0.000001	0.001785
-3.8	0.000001	0.001785
-3.7	0.000001	0.001785
-3.6	0.000001	0.001785
-3.5	0.000001	0.001785
-3.4	0.000001	0.001785
-3.3	0.000001	0.001785
-3.2	0.000001	0.001785
-3.1	0.000001	0.001785
-3.0	0.000001	0.001785
-2.9	0.000001	0.001785
-2.8	0.000001	0.001785
-2.7	0.000001	0.001785
-2.6	0.000001	0.001785
-2.5	0.000001	0.001785
-2.4	0.000001	0.001785
-2.3	0.000001	0.001785
-2.2	0.000001	0.001785
-2.1	0.000001	0.001785
-2.0	0.000001	0.001785
-1.9	0.000001	0.001785
-1.8	0.000001	0.001785
-1.7	0.000001	0.001785
-1.6	0.000001	0.001785
-1.5	0.000001	0.001785
-1.4	0.000001	0.001785
-1.3	0.000001	0.001785
-1.2	0.000001	0.001785
-1.1	0.000001	0.001785
-1.0	0.000001	0.001785
-0.9	0.000001	0.001785
-0.8	0.000001	0.001785
-0.7	0.000001	0.001785
-0.6	0.000001	0.001785
-0.5	0.000001	0.001785
-0.4	0.000001	0.001785
-0.3	0.000001	0.001785
-0.2	0.000001	0.001785
-0.1	0.000001	0.001785
0.0	0.000001	0.001785

Table 45

$P = \gamma$ $O = P$ $\Delta P_{\text{PA}} = 4.8$

INTEGRAL OF THE AIRY FUNCTION

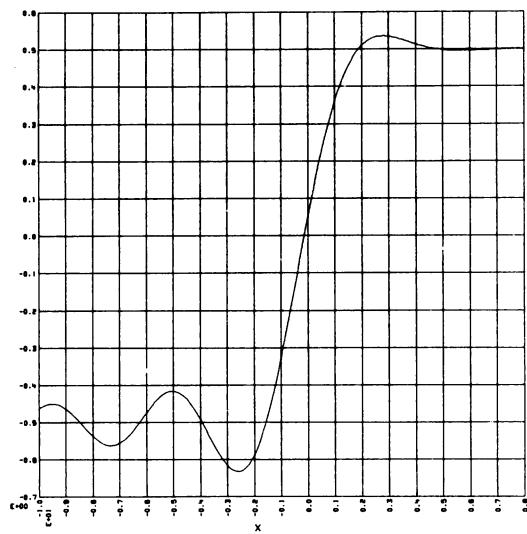


Figure 90

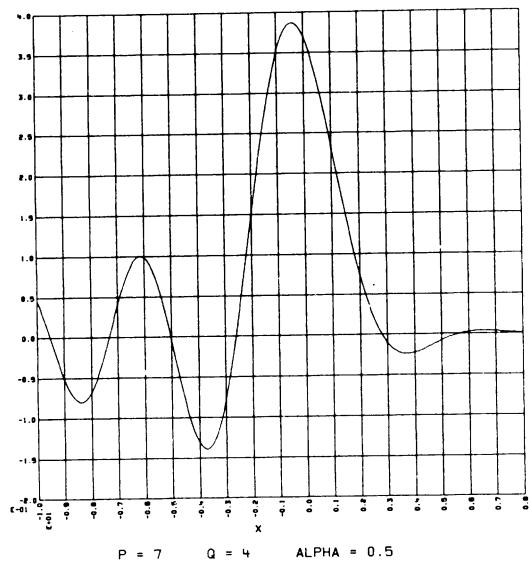


Figure 91

INTEGRAL OF THE AIRY FUNCTION

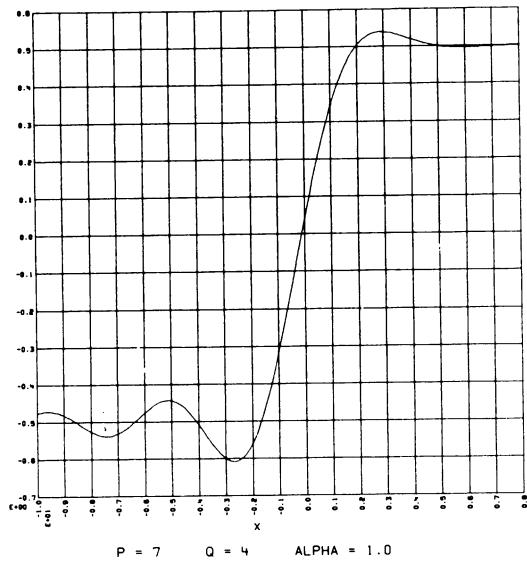


Figure 92

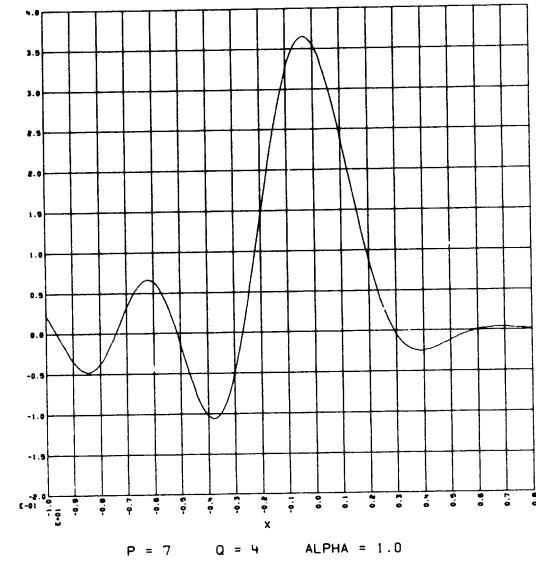


Figure 93

INTEGRAL OF THE AIRY FUNCTION

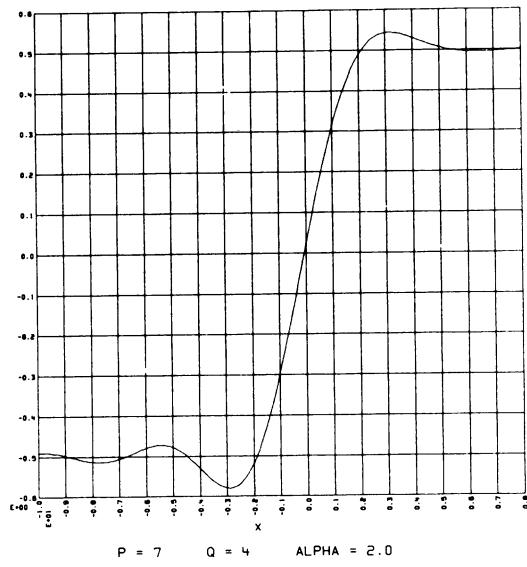


Figure 94

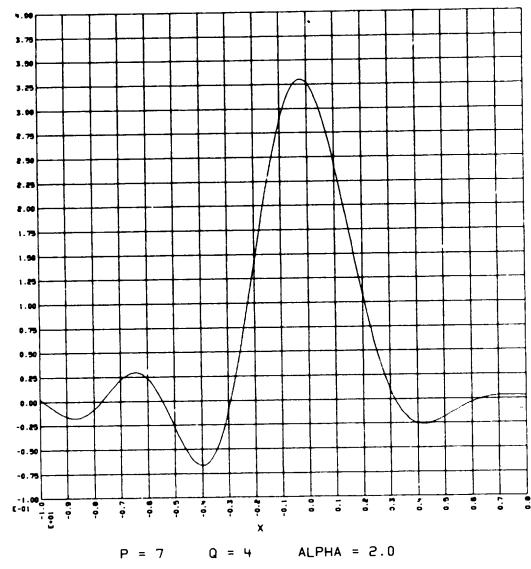


Figure 95

INTEGRAL OF THE AIRY FUNCTION

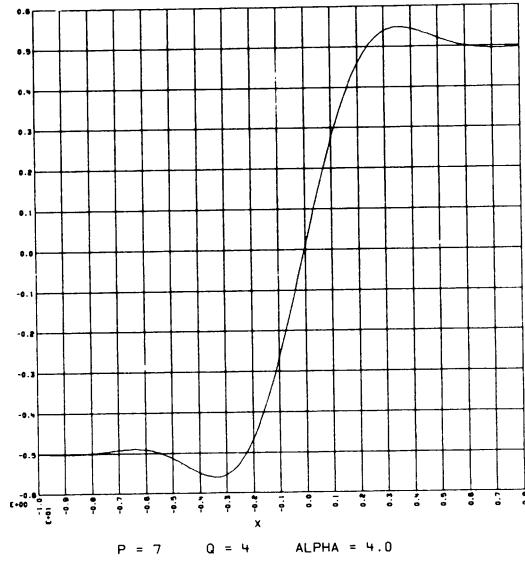


Figure 96

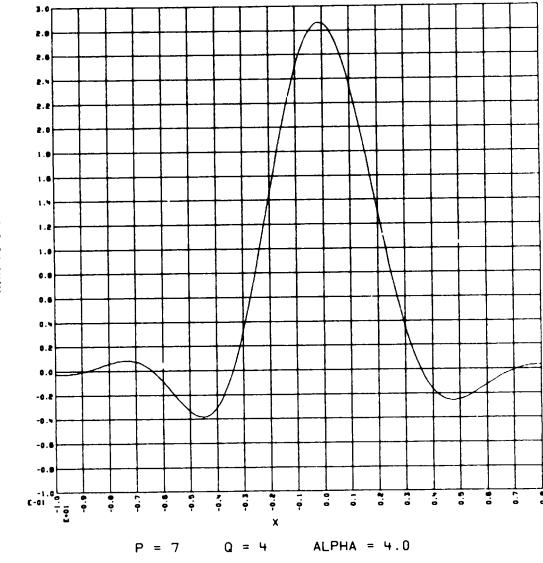


Figure 97

Table 46

P = 7 Q = 4 ALPHA = 0.5

Table 47

P = 7	Q = 4	ALPHA = 1.0
-5	A11(x)	A11(x)
-10	0.476590	0.222171
-9	0.474611	0.217317
-8	0.473460	0.212557
-7	0.472229	0.206249
-6	0.471000	0.199395
-5	0.472183	0.192689
-4	0.473079	0.185775
-3	0.473974	0.178861
-2	0.474664	0.171947
-1	0.475256	0.165033
0	0.475726	0.159120
1	0.476077	0.153207
2	0.476325	0.147294
3	0.476565	0.141380
4	0.476795	0.135467
5	0.477007	0.129554
6	0.477195	0.123641
7	0.477375	0.117728
8	0.477545	0.111815
9	0.477705	0.105892
10	0.477855	0.100000
-9	0.503900	0.279375
-8	0.503900	0.279375
-7	0.503900	0.279375
-6	0.503900	0.279375
-5	0.503900	0.279375
-4	0.503900	0.279375
-3	0.503900	0.279375
-2	0.503900	0.279375
-1	0.503900	0.279375
0	0.503900	0.279375
1	0.503900	0.279375
2	0.503900	0.279375
3	0.503900	0.279375
4	0.503900	0.279375
5	0.503900	0.279375
6	0.503900	0.279375
7	0.503900	0.279375
8	0.503900	0.279375
9	0.503900	0.279375
10	0.503900	0.279375
-8	0.526160	0.236037
-7	0.526160	0.236037
-6	0.526160	0.236037
-5	0.526160	0.236037
-4	0.526160	0.236037
-3	0.526160	0.236037
-2	0.526160	0.236037
-1	0.526160	0.236037
0	0.526160	0.236037
1	0.526160	0.236037
2	0.526160	0.236037
3	0.526160	0.236037
4	0.526160	0.236037
5	0.526160	0.236037
6	0.526160	0.236037
7	0.526160	0.236037
8	0.526160	0.236037
9	0.526160	0.236037
10	0.526160	0.236037
-7	0.533018	0.227983
-6	0.533018	0.227983
-5	0.533018	0.227983
-4	0.533018	0.227983
-3	0.533018	0.227983
-2	0.533018	0.227983
-1	0.533018	0.227983
0	0.533018	0.227983
1	0.533018	0.227983
2	0.533018	0.227983
3	0.533018	0.227983
4	0.533018	0.227983
5	0.533018	0.227983
6	0.533018	0.227983
7	0.533018	0.227983
8	0.533018	0.227983
9	0.533018	0.227983
10	0.533018	0.227983
-6	0.537777	0.219511
-5	0.537777	0.219511
-4	0.537777	0.219511
-3	0.537777	0.219511
-2	0.537777	0.219511
-1	0.537777	0.219511
0	0.537777	0.219511
1	0.537777	0.219511
2	0.537777	0.219511
3	0.537777	0.219511
4	0.537777	0.219511
5	0.537777	0.219511
6	0.537777	0.219511
7	0.537777	0.219511
8	0.537777	0.219511
9	0.537777	0.219511
10	0.537777	0.219511
-5	0.558583	0.207151
-4	0.558583	0.207151
-3	0.558583	0.207151
-2	0.558583	0.207151
-1	0.558583	0.207151
0	0.558583	0.207151
1	0.558583	0.207151
2	0.558583	0.207151
3	0.558583	0.207151
4	0.558583	0.207151
5	0.558583	0.207151
6	0.558583	0.207151
7	0.558583	0.207151
8	0.558583	0.207151
9	0.558583	0.207151
10	0.558583	0.207151
-4	0.558747	0.198942
-3	0.558747	0.198942
-2	0.558747	0.198942
-1	0.558747	0.198942
0	0.558747	0.198942
1	0.558747	0.198942
2	0.558747	0.198942
3	0.558747	0.198942
4	0.558747	0.198942
5	0.558747	0.198942
6	0.558747	0.198942
7	0.558747	0.198942
8	0.558747	0.198942
9	0.558747	0.198942
10	0.558747	0.198942
-3	0.559177	0.190452
-2	0.559177	0.190452
-1	0.559177	0.190452
0	0.559177	0.190452
1	0.559177	0.190452
2	0.559177	0.190452
3	0.559177	0.190452
4	0.559177	0.190452
5	0.559177	0.190452
6	0.559177	0.190452
7	0.559177	0.190452
8	0.559177	0.190452
9	0.559177	0.190452
10	0.559177	0.190452
-2	0.561122	0.182032
-1	0.561122	0.182032
0	0.561122	0.182032
1	0.561122	0.182032
2	0.561122	0.182032
3	0.561122	0.182032
4	0.561122	0.182032
5	0.561122	0.182032
6	0.561122	0.182032
7	0.561122	0.182032
8	0.561122	0.182032
9	0.561122	0.182032
10	0.561122	0.182032
-1	0.561471	0.173917
0	0.561471	0.173917
1	0.561471	0.173917
2	0.561471	0.173917
3	0.561471	0.173917
4	0.561471	0.173917
5	0.561471	0.173917
6	0.561471	0.173917
7	0.561471	0.173917
8	0.561471	0.173917
9	0.561471	0.173917
10	0.561471	0.173917
-1	0.562126	0.165722
0	0.562126	0.165722
1	0.562126	0.165722
2	0.562126	0.165722
3	0.562126	0.165722
4	0.562126	0.165722
5	0.562126	0.165722
6	0.562126	0.165722
7	0.562126	0.165722
8	0.562126	0.165722
9	0.562126	0.165722
10	0.562126	0.165722
-1	0.562471	0.157537
0	0.562471	0.157537
1	0.562471	0.157537
2	0.562471	0.157537
3	0.562471	0.157537
4	0.562471	0.157537
5	0.562471	0.157537
6	0.562471	0.157537
7	0.562471	0.157537
8	0.562471	0.157537
9	0.562471	0.157537
10	0.562471	0.157537
-1	0.562821	0.149353
0	0.562821	0.149353
1	0.562821	0.149353
2	0.562821	0.149353
3	0.562821	0.149353
4	0.562821	0.149353
5	0.562821	0.149353
6	0.562821	0.149353
7	0.562821	0.149353
8	0.562821	0.149353
9	0.562821	0.149353
10	0.562821	0.149353
-1	0.563177	0.141177
0	0.563177	0.141177
1	0.563177	0.141177
2	0.563177	0.141177
3	0.563177	0.141177
4	0.563177	0.141177
5	0.563177	0.141177
6	0.563177	0.141177
7	0.563177	0.141177
8	0.563177	0.141177
9	0.563177	0.141177
10	0.563177	0.141177
-1	0.563532	0.132997
0	0.563532	0.132997
1	0.563532	0.132997
2	0.563532	0.132997
3	0.563532	0.132997
4	0.563532	0.132997
5	0.563532	0.132997
6	0.563532	0.132997
7	0.563532	0.132997
8	0.563532	0.132997
9	0.563532	0.132997
10	0.563532	0.132997
-1	0.563877	0.124822
0	0.563877	0.124822
1	0.563877	0.124822
2	0.563877	0.124822
3	0.563877	0.124822
4	0.563877	0.124822
5	0.563877	0.124822
6	0.563877	0.124822
7	0.563877	0.124822
8	0.563877	0.124822
9	0.563877	0.124822
10	0.563877	0.124822
-1	0.564232	0.116647
0	0.564232	0.116647
1	0.564232	0.116647
2	0.564232	0.116647
3	0.564232	0.116647
4	0.564232	0.116647
5	0.564232	0.116647
6	0.564232	0.116647
7	0.564232	0.116647
8	0.564232	0.116647
9	0.564232	0.116647
10	0.564232	0.116647
-1	0.564587	0.108472
0	0.564587	0.108472
1	0.564587	0.108472
2	0.564587	0.108472
3	0.564587	0.108472
4	0.564587	0.108472
5	0.564587	0.108472
6	0.564587	0.108472
7	0.564587	0.108472
8	0.564587	0.108472
9	0.564587	0.108472
10	0.564587	0.108472
-1	0.564942	0.100297
0	0.564942	0.100297
1	0.564942	0.100297
2	0.564942	0.100297
3	0.564942	0.100297
4	0.564942	0.100297
5	0.564942	0.100297
6	0.564942	0.100297
7	0.564942	0.100297
8	0.564942	0.100297
9	0.564942	0.100297
10	0.564942	0.100297
-1	0.565303	0.092122
0	0.565303	0.092122
1	0.565303	0.092122
2	0.565303	0.092122
3	0.565303	0.092122
4	0.565303	0.092122
5	0.565303	0.092122
6	0.565303	0.092122
7	0.565303	0.092122
8	0.565303	0.092122
9	0.565303	0.092122
10	0.565303	0.092122
-1	0.565658	0.083947
0	0.565658	0.083947
1	0.565658	0.083947
2	0.565658	0.083947
3	0.565658	0.083947
4	0.565658	0.083947
5	0.565658	0.083947
6	0.565658	0.083947
7	0.565658	0.083947
8	0.565658	0.083947
9	0.565658	0.083947
10	0.565658	0.083947
-1	0.566013	0.075772
0	0.566013	0.075772
1	0.566013	0.075772
2	0.566013	0.075772
3	0.566013	0.075772
4	0.566013	0.075772
5	0.566013	0.075772
6	0.566013	0.075772
7	0.566013	0.075772
8	0.566013	0.075772
9	0.566013	0.075772
10	0.566013	0.075772
-1	0.566377	0.067597
0	0.566377	0.067597
1	0.566377	0.067597
2	0.566377	0.067597
3	0.566377	0.067597
4	0.566377	0.067597
5	0.566377	0.067597
6	0.566377	0.067597
7	0.566377	0.067597
8	0.566377	0.067597
9	0.566377	0.067597
10	0.566377	0.067597
-1	0.566742	0.060422
0	0.566742	0.060422
1	0.566742	0.060422
2	0.566742	0.060422
3	0.566742	0.060422
4	0.566742	0.060422
5	0.566742	0.060422
6	0.566	

Table 48

P = 7 Q = 4 ALPHAB = 2.0

Table 49

P = 7 Q = 4 ALPHA = 4.0

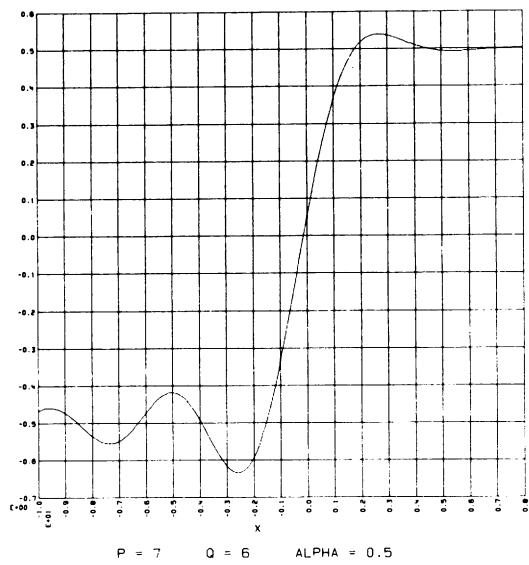


Figure 98

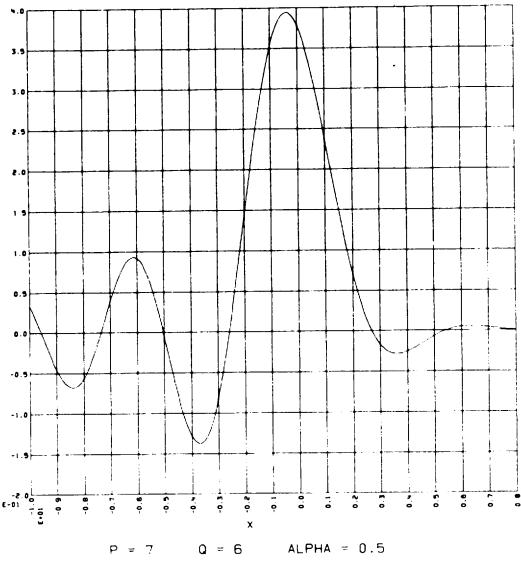


Figure 99

INTEGRAL OF THE AIRY FUNCTION

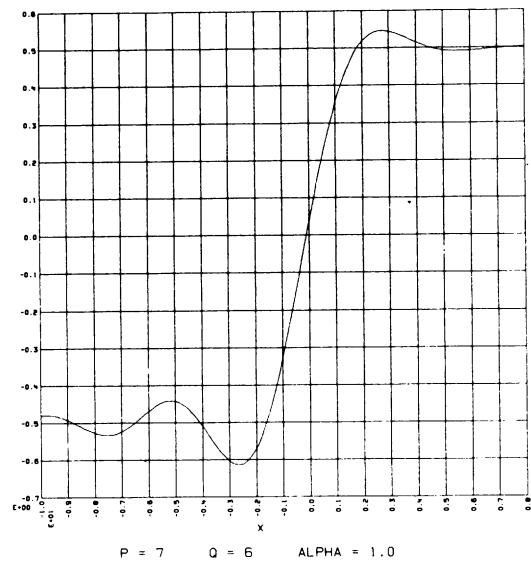


Figure 100

Table 53

P = 7 Q = 6 ALPHA = 4.0

INTEGRAL OF THE AIRY FUNCTION

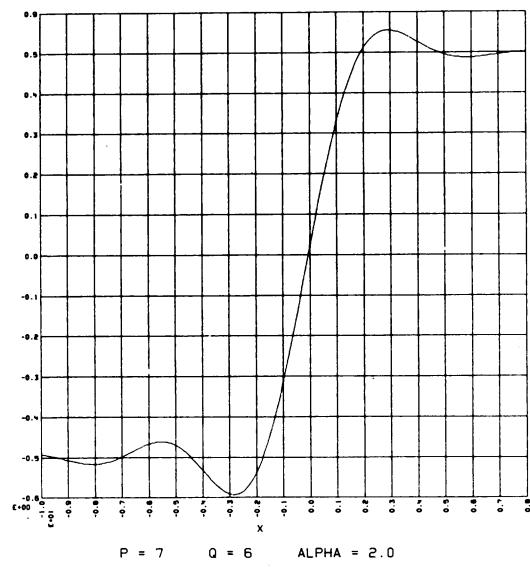


Figure 102

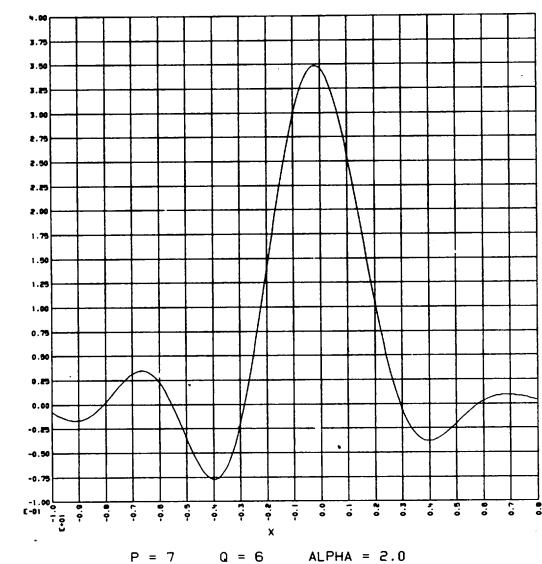


Figure 103

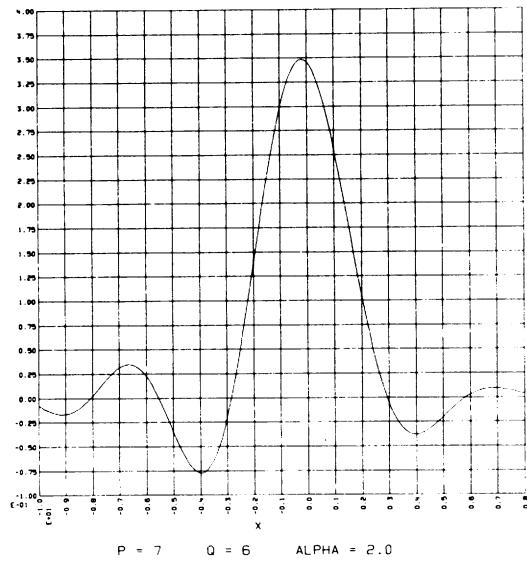


Figure 103

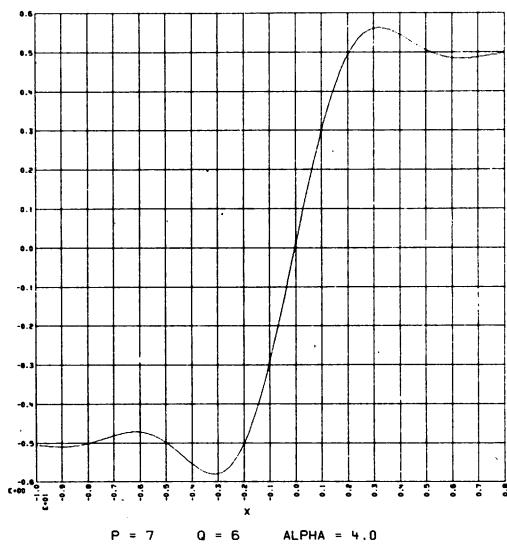


Figure 104

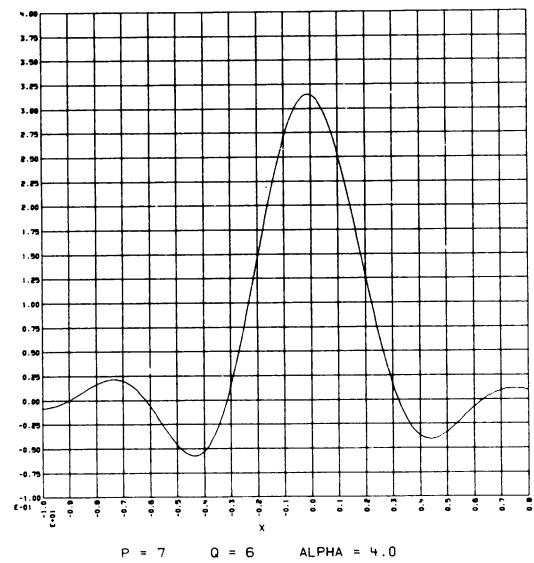


Figure 105

Table 50

P = 7 Q = 6 ALPHA = 0.5

Table 52

P = 7 Q = 6 ALPHA = 2.0

Table 53

X	A11(1)	A11(2)	A11(3)	A11(4)	A11(5)	A11(6)	A11(7)	A11(8)	A11(9)	A11(10)	A11(11)
-10.0	-0.50773	-0.008304	-5.5	-0.480771	-0.026971	-1.0	-0.200370	0.270661	3.5	0.559035	-0.019207
-9.9	-0.505959	-0.008097	-5.4	-0.480575	-0.031104	-0.9	-0.202648	0.270661	3.5	0.559035	-0.019207
-9.8	-0.504186	-0.007890	-5.3	-0.480379	-0.035137	-0.8	-0.205175	0.270661	3.5	0.559035	-0.019207
-9.7	-0.502410	-0.007684	-5.2	-0.480173	-0.039166	-0.7	-0.205135	0.294782	3.8	0.551227	-0.019168
-9.6	-0.500634	-0.007477	-5.1	-0.480077	-0.043195	-0.6	-0.205135	0.294782	3.8	0.551227	-0.019168
-9.5	-0.498858	-0.007270	-5.0	-0.480029	-0.048275	-0.5	-0.14202	0.309336	4.0	0.547296	-0.017053
-9.4	-0.497083	-0.006963	-4.9	-0.480070	-0.048275	-0.4	-0.14201	0.309336	4.0	0.547296	-0.017053
-9.3	-0.495307	-0.006756	-4.8	-0.480070	-0.048275	-0.3	-0.14201	0.309336	4.0	0.547296	-0.017053
-9.2	-0.493531	-0.006549	-4.7	-0.480070	-0.048275	-0.2	-0.14201	0.309336	4.0	0.547296	-0.017053
-9.1	-0.491755	-0.006342	-4.6	-0.480070	-0.048275	-0.1	-0.14201	0.309336	4.0	0.547296	-0.017053
-9.0	-0.490079	-0.006135	-4.5	-0.480070	-0.048275	0	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.9	-0.488303	-0.005928	-4.4	-0.480070	-0.048275	-1	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.8	-0.486527	-0.005721	-4.3	-0.480070	-0.048275	-2	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.7	-0.484751	-0.005514	-4.2	-0.480070	-0.048275	-3	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.6	-0.482975	-0.005307	-4.1	-0.480070	-0.048275	-4	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.5	-0.481200	-0.005100	-4.0	-0.480070	-0.048275	-5	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.4	-0.479424	-0.004893	-3.9	-0.480070	-0.048275	-6	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.3	-0.477648	-0.004686	-3.8	-0.480070	-0.048275	-7	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.2	-0.475872	-0.004479	-3.7	-0.480070	-0.048275	-8	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.1	-0.474106	-0.004272	-3.6	-0.480070	-0.048275	-9	-0.14201	0.309336	4.0	0.547296	-0.017053
-8.0	-0.472330	-0.004065	-3.5	-0.480070	-0.048275	-10	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.9	-0.470554	-0.003858	-3.4	-0.480070	-0.048275	-11	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.8	-0.468778	-0.003651	-3.3	-0.480070	-0.048275	-12	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.7	-0.467002	-0.003444	-3.2	-0.480070	-0.048275	-13	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.6	-0.465226	-0.003237	-3.0	-0.480070	-0.048275	-14	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.5	-0.463450	-0.003030	-2.9	-0.480070	-0.048275	-15	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.4	-0.461674	-0.002823	-2.8	-0.480070	-0.048275	-16	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.3	-0.459898	-0.002616	-2.7	-0.480070	-0.048275	-17	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.2	-0.458122	-0.002409	-2.6	-0.480070	-0.048275	-18	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.1	-0.456346	-0.002202	-2.5	-0.480070	-0.048275	-19	-0.14201	0.309336	4.0	0.547296	-0.017053
-7.0	-0.454570	-0.002095	-2.4	-0.480070	-0.048275	-20	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.9	-0.452794	-0.001888	-2.3	-0.480070	-0.048275	-21	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.8	-0.451018	-0.001681	-2.2	-0.480070	-0.048275	-22	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.7	-0.449242	-0.001474	-2.1	-0.480070	-0.048275	-23	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.6	-0.447466	-0.001267	-2.0	-0.480070	-0.048275	-24	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.5	-0.445690	-0.001060	-1.9	-0.480070	-0.048275	-25	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.4	-0.443914	-0.000853	-1.8	-0.480070	-0.048275	-26	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.3	-0.442138	-0.000646	-1.7	-0.480070	-0.048275	-27	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.2	-0.440362	-0.000439	-1.6	-0.480070	-0.048275	-28	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.1	-0.438586	-0.000232	-1.5	-0.480070	-0.048275	-29	-0.14201	0.309336	4.0	0.547296	-0.017053
-6.0	-0.436810	-0.000025	-1.4	-0.480070	-0.048275	-30	-0.14201	0.309336	4.0	0.547296	-0.017053
-5.9	-0.435034	-0.019152	-1.3	-0.480070	-0.048275	-31	-0.14201	0.309336	4.0	0.547296	-0.017053
-5.8	-0.433258	-0.018125	-1.2	-0.480070	-0.048275	-32	-0.14201	0.309336	4.0	0.547296	-0.017053
-5.7	-0.431482	-0.017098	-1.1	-0.480070	-0.048275	-33	-0.14201	0.309336	4.0	0.547296	-0.017053
-5.6	-0.429706	-0.016071	-1.0	-0.480070	-0.048275	-34	-0.14201	0.309336	4.0	0.547296	-0.017053